



Indian Institute of Technology Bhubaneswar

School of Infrastructure

Subject Name : Solid Mechanics

Subject Code: CE2L001

Tutorial No. 4

1. Different simple cases of transformations are illustrated in Fig. 1, where α and β are arbitrary scalar positive values. Consider the two-dimensional context and determine the deformation map, $\chi(\mathbf{x})$, deformation gradient, \mathbf{F} , the strain tensor, $\mathbf{E} = (1/2)(\mathbf{F}^T \mathbf{F} - \mathbf{I})$, displacement vector, \mathbf{u} and the linearized strain tensor, \mathbf{E}_s , for each case.

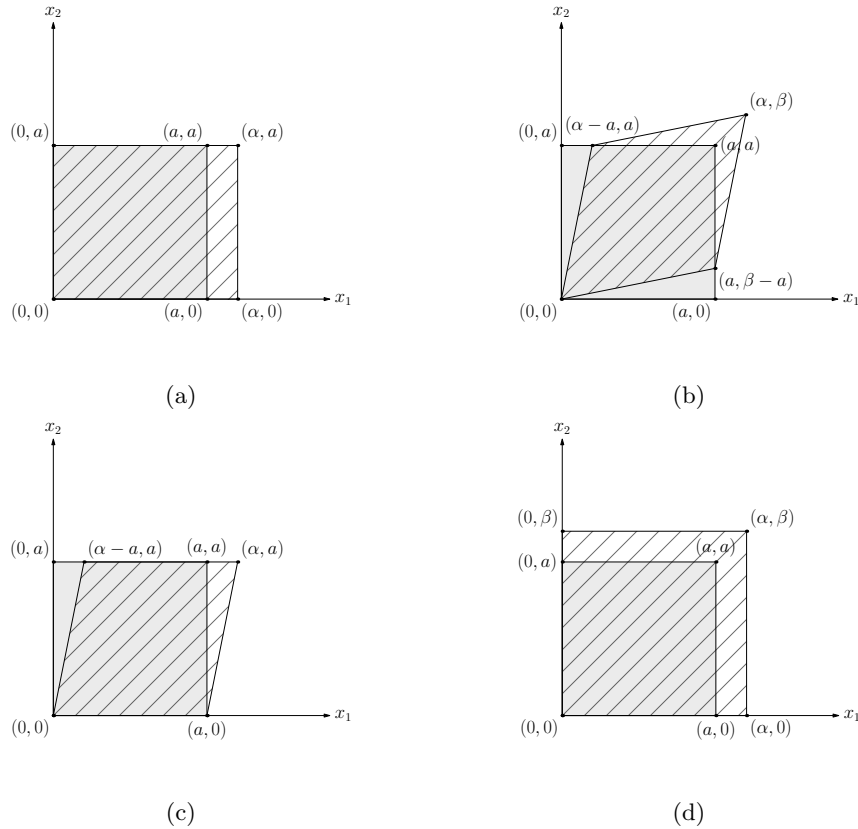


Figure 1: The undeformed and deformed configurations of a body under different cases of transformations with a , α , β denoting arbitrary scalar positive constants. The hatched area illustrates the deformed configurations.

Solution: Consider the displacement components to be a function of the coordinates x_1 and x_2 . Then, a general expression for the displacement components can be written as

$$\begin{aligned}u_1(x_1, x_2) &= a + bx_1 + cx_2 + dx_1x_2 \\u_2(x_1, x_2) &= p + qx_1 + rx_2 + sx_1x_2\end{aligned}$$

(a) From the deformed shape of the solid block in Fig.1 (a), the displacement along x_1 at each four corners of the solid block can be written as

$$\begin{aligned}u_1(0, 0) &= a = 0 \\u_1(a, 0) &= a + ba = \alpha - a \implies b = (\alpha - a)/a \\u_1(0, a) &= a + ca = 0 \implies c = 0 \\u_1(a, a) &= a + ba + ca + da^2 = \alpha - a \\&= (\alpha - a) + da^2 = (\alpha - a) \\\implies d &= 0\end{aligned}\tag{1}$$

The displacement along x_2 and x_3 at each four corners of the solid block is zero. Thus the expression for the displacement components can be written as

$$\begin{aligned}u_1(x_1, x_2) &= \frac{(\alpha - a)}{a}x_1 \\u_2(x_1, x_2) &= 0 \\u_3(x_1, x_2) &= 0\end{aligned}\tag{2}$$

The linear elasticity tensor \mathbf{F} can be written as

$$\mathbf{E}_s = \frac{1}{2}(\nabla \mathbf{u} + (\nabla \mathbf{u})^T)$$

The components of the strain tensor can be obtained as

$$\begin{aligned}E_{11} &= \frac{\partial u_1}{\partial x_1} \\E_{22} &= \frac{\partial u_2}{\partial x_2} \\E_{12} &= \frac{1}{2} \left(\frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \right)\end{aligned}\tag{3}$$

Thus from Eq.(2) and Eq.(3), the components of strain can be written as

$$\begin{aligned}E_{11} &= \frac{(\alpha - a)}{a} \\E_{22} &= 0 \\E_{12} &= 0\end{aligned}\tag{4}$$

Thus the strain tensor in component form can be written as

$$[\mathbf{E}] = \begin{bmatrix} \frac{(\alpha-a)}{a} & 0 \\ 0 & 0 \end{bmatrix} \quad (5)$$

(b) From the deformed shape of the solid block in Fig.1(b), the displacement along x_1 at each four corners of the solid block can be written as

$$\begin{aligned} u_1(0,0) &= a = 0 \\ u_1(a,0) &= a + ba = 0 \implies b = 0 \\ u_1(0,a) &= a + ca = \alpha - a \implies c = (\alpha - a)/a \\ u_1(a,a) &= a + ba + ca + da^2 = \alpha - a \\ &= (\alpha - a) + da^2 = (\alpha - a) \\ &\implies d = 0 \end{aligned} \quad (6)$$

The displacement along x_2 at each four corners of the solid block can be written as

$$\begin{aligned} u_2(0,0) &= p = 0 \\ u_2(a,0) &= p + qa = \beta - a \implies q = (\beta - a)/a \\ u_2(0,a) &= p + ra = 0 \implies r = 0 \\ u_2(a,a) &= p + qa + ra + sa^2 = \beta - a \\ &= (\beta - a) + sa^2 = (\beta - a) \\ &\implies s = 0 \end{aligned} \quad (7)$$

The displacement along x_3 at each four corners of the solid block is zero. Thus the expression for the displacement components can be written as

$$\begin{aligned} u_1(x_1, x_2) &= \frac{(\alpha - a)}{a} x_2 \\ u_2(x_1, x_2) &= \frac{(\beta - a)}{a} x_1 \\ u_3(x_1, x_2) &= 0 \end{aligned} \quad (8)$$

The linear elasticity tensor \mathbf{F} can be written as

$$\mathbf{E}_s = \frac{1}{2}(\nabla \mathbf{u} + (\nabla \mathbf{u})^T)$$

The components of the strain tensor can be obtained as

$$\begin{aligned} E_{11} &= \frac{\partial u_1}{\partial x_1} \\ E_{22} &= \frac{\partial u_2}{\partial x_2} \end{aligned}$$

$$E_{12} = \frac{1}{2} \left(\frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \right) \quad (9)$$

Thus from Eq.(8) and Eq.(9), the components of strain can be written as

$$\begin{aligned} E_{11} &= 0 \\ E_{22} &= 0 \\ E_{12} &= \frac{1}{2a}((\alpha - a) + (\beta - a)) \end{aligned} \quad (10)$$

Thus the strain tensor in component form can be written as

$$[\mathbf{E}] = \begin{bmatrix} 0 & \frac{1}{2a}((\alpha - a) + (\beta - a)) \\ \frac{1}{2a}((\alpha - a) + (\beta - a)) & 0 \end{bmatrix} \quad (11)$$

(c) From the deformed shape of the solid block in Fig.1(c), the displacement along x_1 at each four corners of the solid block can be written as

$$\begin{aligned} u_1(0, 0) &= a = 0 \\ u_1(a, 0) &= a + ba = 0 \implies b = 0 \\ u_1(0, a) &= a + ca = (\alpha - a) \implies c = (\alpha - a)/a \\ u_1(a, a) &= a + ba + ca + da^2 = \alpha - a \\ &= (\alpha - a) + da^2 = (\alpha - a) \\ \implies d &= 0 \end{aligned} \quad (12)$$

The displacement along x_2 and x_3 at each four corners of the solid block is zero. Thus the expression for the displacement components can be written as

$$\begin{aligned} u_1(x_1, x_2) &= \frac{(\alpha - a)}{a} x_2 \\ u_2(x_1, x_2) &= 0 \end{aligned}$$

The linear elasticity tensor \mathbf{F} can be written as

$$\mathbf{E}_s = \frac{1}{2}(\nabla \mathbf{u} + (\nabla \mathbf{u})^T)$$

The components of the strain tensor can be obtained as

$$\begin{aligned} E_{11} &= \frac{\partial u_1}{\partial x_1} \\ E_{22} &= \frac{\partial u_2}{\partial x_2} \\ E_{12} &= \frac{1}{2} \left(\frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \right) \end{aligned} \quad (13)$$

Thus from Eq.(??) and Eq.(13), the components of strain can be written as

$$\begin{aligned} E_{11} &= 0 \\ E_{22} &= 0 \\ E_{12} &= \frac{(\alpha - a)}{a} \end{aligned} \quad (14)$$

Thus the strain tensor in component form can be written as

$$[\mathbf{E}] = \begin{bmatrix} 0 & \frac{(\alpha - a)}{a} \\ \frac{(\alpha - a)}{a} & 0 \end{bmatrix} \quad (15)$$

(d) From the deformed shape of the solid block in Fig.1(d), the displacement along x_1 at each four corners of the solid block can be written as

$$\begin{aligned} u_1(0, 0) &= a = 0 \\ u_1(a, 0) &= a + ba = \alpha - a \implies b = (\alpha - a)/a \\ u_1(0, a) &= a + ca = 0 \implies c = 0 \\ u_1(a, a) &= a + ba + ca + da^2 = \alpha - a \\ &= (\alpha - a) + da^2 = (\alpha - a) \\ \implies d &= 0 \end{aligned} \quad (16)$$

The displacement along x_2 at each four corners of the solid block can be written as

$$\begin{aligned} u_2(0, 0) &= p = 0 \\ u_2(a, 0) &= p + qa = 0 \implies b = 0 \\ u_2(0, a) &= p + ra = \beta - a \implies r = (\beta - a)/a \\ u_2(a, a) &= p + qa + ra + sa^2 = \beta - a \\ &= (\beta - a) + sa^2 = (\beta - a) \\ \implies s &= 0 \end{aligned} \quad (17)$$

The displacement along x_3 at each four corners of the solid block is zero. Thus the expression for the displacement components can be written as

$$\begin{aligned} u_1(x_1, x_2) &= \frac{(\alpha - a)}{a} x_1 \\ u_2(x_1, x_2) &= \frac{(\beta - a)}{a} x_2 \\ u_3(x_1, x_2) &= 0 \end{aligned} \quad (18)$$

The linear elasticity tensor \mathbf{F} can be written as

$$\mathbf{E}_s = \frac{1}{2}(\nabla \mathbf{u} + (\nabla \mathbf{u})^T)$$

The components of the strain tensor can be obtained as

$$\begin{aligned} E_{11} &= \frac{\partial u_1}{\partial x_1} \\ E_{22} &= \frac{\partial u_2}{\partial x_2} \\ E_{12} &= \frac{1}{2} \left(\frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \right) \end{aligned} \quad (19)$$

Thus from Eq.(18) and Eq.(19), the components of strain can be written as

$$\begin{aligned} E_{11} &= (\alpha - a)/a \\ E_{22} &= (\beta - a)/a \\ E_{12} &= 0 \end{aligned} \quad (20)$$

Thus the strain tensor in component form can be written as

$$[\mathbf{E}] = \begin{bmatrix} (\alpha - a)/a & 0 \\ 0 & (\beta - a)/a \end{bmatrix} \quad (21)$$

2. Consider a homogeneous deformation corresponds to a strain field where the strain is the same at all points in a material body. Consider a prismatic, uniform thickness bar of initial length l_o undergoing a homogeneous deformation as shown in Fig. 2. Determine (a) deformation gradient, \mathbf{F} , (b) finite strain, \mathbf{E} and (c) linearized strain, \mathbf{E}_s (d) Demonstrate how the finite strain component reduces to linearized strain component through a numerical example.

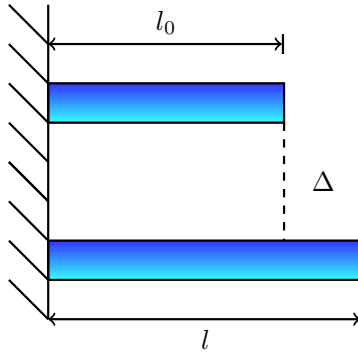


Figure 2: Undeformed and deformed element in the homogenous strain field in the bar.

Solution: (a) From Fig. 2, one can write that

$$\mathbf{F}(l_0 \mathbf{e}_1) = l \mathbf{e}_1 \implies \mathbf{F} \mathbf{e}_1 = \left(\frac{l}{l_0}\right) \mathbf{e}_1, \quad \mathbf{F} \mathbf{e}_2 = \mathbf{e}_2, \quad \mathbf{F} \mathbf{e}_3 = \mathbf{e}_3. \quad (22)$$

One can find the deformation gradient, \mathbf{F} , using any one of the following methods.

Method I: For $\mathbf{F} = F_{ij} \mathbf{e}_i \otimes \mathbf{e}_j$, with $\{\mathbf{e}_i\}$ denoting the orthonormal basis, one can find the components as

$$F_{ij} = \mathbf{e}_i \cdot (\mathbf{F} \mathbf{e}_j) \quad (23)$$

By using Eq. (22) and Eq. (23),

$$F_{11} = \mathbf{e}_1 \cdot (\mathbf{F} \mathbf{e}_1) = l/l_0, \quad F_{12} = \mathbf{e}_1 \cdot (\mathbf{F} \mathbf{e}_2) = 0, \quad F_{13} = \mathbf{e}_1 \cdot (\mathbf{F} \mathbf{e}_3) = 0.$$

$$F_{21} = \mathbf{e}_2 \cdot (\mathbf{F} \mathbf{e}_1) = 0, \quad F_{22} = \mathbf{e}_2 \cdot (\mathbf{F} \mathbf{e}_2) = 1, \quad F_{23} = \mathbf{e}_2 \cdot (\mathbf{F} \mathbf{e}_3) = 0,$$

$$F_{31} = \mathbf{e}_3 \cdot (\mathbf{F} \mathbf{e}_1) = 0, \quad F_{32} = \mathbf{e}_3 \cdot (\mathbf{F} \mathbf{e}_2) = 0, \quad F_{33} = \mathbf{e}_3 \cdot (\mathbf{F} \mathbf{e}_3) = 1.$$

Hence,

$$\mathbf{F} = \left(\frac{l}{l_0}\right) \mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + \mathbf{e}_3 \otimes \mathbf{e}_3$$

Method II: From Eq. (22), one can write that

$$\mathbf{F} \mathbf{e}_1 \otimes \mathbf{e}_1 = \left(\frac{l}{l_0}\right) \mathbf{e}_1 \otimes \mathbf{e}_1, \quad \mathbf{F} \mathbf{e}_2 \otimes \mathbf{e}_2 = \mathbf{e}_2 \otimes \mathbf{e}_2, \quad \mathbf{F} \mathbf{e}_3 \otimes \mathbf{e}_3 = \mathbf{e}_3 \otimes \mathbf{e}_3.$$

Adding all the above three components,

$$\begin{aligned} \mathbf{F}(\mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + \mathbf{e}_3 \otimes \mathbf{e}_3) &= \left(\frac{l}{l_0}\right) \mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + \mathbf{e}_3 \otimes \mathbf{e}_3 \\ \implies \mathbf{F} \mathbf{I} &= \left(\frac{l}{l_0}\right) \mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + \mathbf{e}_3 \otimes \mathbf{e}_3 \quad (\text{since } \mathbf{I} = \mathbf{e}_i \otimes \mathbf{e}_i) \\ \implies \mathbf{F} &= \left(\frac{l}{l_0}\right) \mathbf{e}_1 \otimes \mathbf{e}_1 + \mathbf{e}_2 \otimes \mathbf{e}_2 + \mathbf{e}_3 \otimes \mathbf{e}_3 \quad (\text{since } \mathbf{F} \mathbf{I} = \mathbf{F}) \end{aligned}$$

Method III: A general expression for the displacement components in a one-dimensional (1D) bar can be written as,

$$x'_1(x_1) = a + bx_1, \quad x'_2(x_J) = x_2, \quad x'_3(x_1) = x_3.$$

Applying boundary conditions,

$$x'_1(0) = 0 \implies a = 0.$$

$$x'_1(l_0) = l \implies b l_0 = l \implies b = \frac{l}{l_0}.$$

Hence,

$$x'_1 = \left(\frac{l}{l_0}\right) x_1, \quad x'_2(x_1) = x_2, \quad x'_3(x_1) = x_3.$$

The deformation gradient, \mathbf{F} , can be written in its component form as

$$[\mathbf{F}] = \begin{bmatrix} \frac{\partial x'_1}{\partial x_1} & \frac{\partial x'_1}{\partial x_2} & \frac{\partial x'_1}{\partial x_3} \\ \frac{\partial x'_2}{\partial x_1} & \frac{\partial x'_2}{\partial x_2} & \frac{\partial x'_2}{\partial x_3} \\ \frac{\partial x'_3}{\partial x_1} & \frac{\partial x'_3}{\partial x_2} & \frac{\partial x'_3}{\partial x_3} \end{bmatrix} = \begin{bmatrix} \left(\frac{l}{l_0}\right) & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

Hence, $\mathbf{F} = \left(\frac{l}{l_0}\right) \mathbf{e}_1 \otimes \mathbf{e}_1 + 1 \mathbf{e}_2 \otimes \mathbf{e}_2 + 1 \mathbf{e}_3 \otimes \mathbf{e}_3$.

(b) Finite Cauchy-Green strain, $\mathbf{E} = (1/2) (\mathbf{F}^T \mathbf{F} - \mathbf{I})$ can be computed using its component form as follows.

$$[\mathbf{F}]^T [\mathbf{F}] = \begin{bmatrix} \left(\frac{l}{l_0}\right) & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} \left(\frac{l}{l_0}\right) & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} = \begin{bmatrix} \left(\frac{l}{l_0}\right)^2 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$[\mathbf{E}] = \frac{1}{2} \left(\begin{bmatrix} \left(\frac{l}{l_0}\right)^2 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} - \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \right) = \begin{bmatrix} \left(\frac{l^2 - l_0^2}{2l_0^2}\right) & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

So, $\mathbf{E} = \left(\frac{l^2 - l_0^2}{2l_0^2}\right) \mathbf{e}_1 \otimes \mathbf{e}_1 = \epsilon_{11}^{\text{Cauchy}} \mathbf{e}_1 \otimes \mathbf{e}_1$, where $\epsilon_{11}^{\text{Cauchy}} = \frac{l^2 - l_0^2}{2l_0^2}$. (c) Linearized strain, $\mathbf{E}_s = \frac{1}{2} (\mathbf{F}^T + \mathbf{F} - 2\mathbf{I})$ can be computed using its component form as

$$[\mathbf{E}_s] = \frac{1}{2} ([\mathbf{F}^T] + [\mathbf{F}] - 2[\mathbf{I}]).$$

For the present problem, $[\mathbf{F}^T] = [\mathbf{F}]$.

$$[\mathbf{E}_s] = \frac{1}{2} \left(\begin{bmatrix} 2\left(\frac{l}{l_0}\right) & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix} - \begin{bmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 2 \end{bmatrix} \right) = \begin{bmatrix} \left(\frac{l - l_0}{l_0}\right) & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

So, $\mathbf{E}_s = \left(\frac{l - l_0}{l_0}\right) \mathbf{e}_1 \otimes \mathbf{e}_1 = \epsilon_{11}^{\text{eng}} \mathbf{e}_1 \otimes \mathbf{e}_1$, where engineering strain component, $\epsilon_{11}^{\text{eng}} = \left(\frac{l - l_0}{l_0}\right)$. One can see how the Cauchy-Green finite strain component reduces to a linearized strain component as

$$\begin{aligned} \epsilon_{11}^{\text{Cauchy}} &= \frac{l^2 - l_0^2}{2l_0^2} = \frac{(l - l_0)(l + l_0)}{2l_0^2} \\ &= \frac{(l - l_0)(2l_0)}{2l_0^2} \quad (\text{if } \Delta \text{ is very less, then } l + l_0 \approx 2l_0) \\ &= \left(\frac{l - l_0}{l_0}\right) = \epsilon_{11}^{\text{eng}} \end{aligned}$$

(d) Let's take $l_o = 10$ mm and $l = 15$ mm, and find the Engineering and Cauchy-Green Strain.

Engineering strain component,

$$\epsilon_{11}^{\text{eng}} = \left(\frac{l - l_o}{l_o} \right) = \frac{15 - 10}{10} = 0.5$$

Cauchy-Green finite strain component,

$$\epsilon_{11}^{\text{Cauchy}} = \frac{1}{2} \left(\frac{l^2 - l_o^2}{l_o^2} \right) = \frac{1}{2} \left(\frac{(15)^2 - (10)^2}{(10)^2} \right) = 0.625$$

Observation- In case of large deformation (Δ is significantly large), $\epsilon_{11}^{\text{eng}}$ will be significantly different than $\epsilon_{11}^{\text{Cauchy}}$. As seen from Fig. 3, both the strain components may provide identical results only when the deformation is minimal.

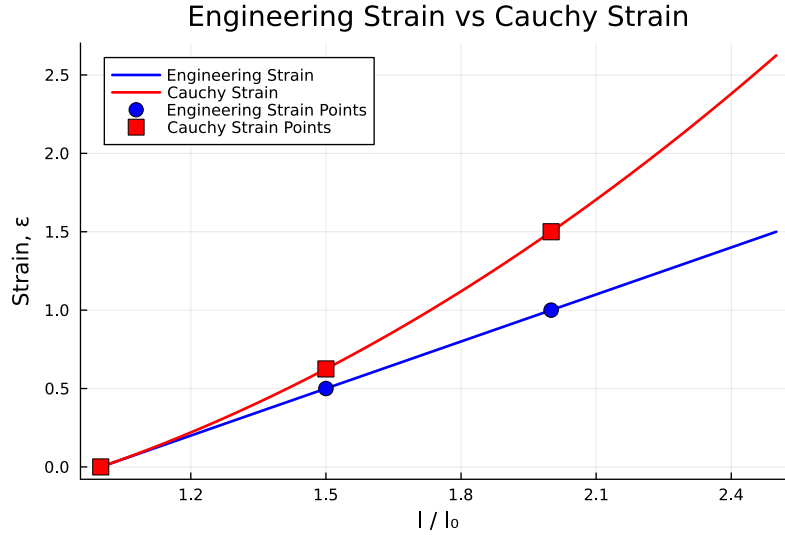


Figure 3: Relationship between engineering strain and Cauchy-Green strain component.

3. Consider a two-dimensional (2D) square infinitesimal element in the $x_1 - x_2$ plane as shown in Fig. 4. The displacement field within the element is defined as,

$$[\mathbf{u}] = \begin{bmatrix} 0.1x_1 + 0.2x_2 \\ 0.2x_2 \end{bmatrix}.$$

- (a) Plot the displacement field. (b) Compute the divergence of the displacement field \mathbf{u} . (c) Determine the strain \mathbf{E} and linearized strain \mathbf{E}_s .

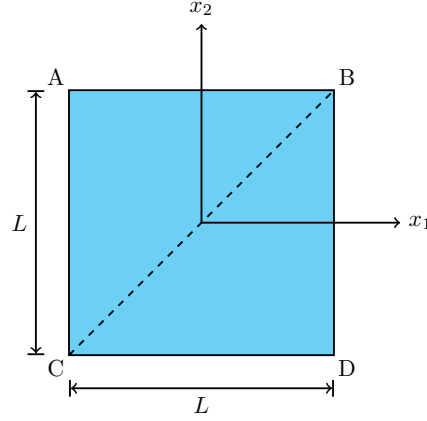


Figure 4: Inhomogeneous strain field

Solution: (a) One can plot the displacement field \mathbf{u} considering different points such as $[1, 1], [-1, 1], [-1, -1], [1, 0], [-1, 0], [0, 1], [0, -1]$ etc.

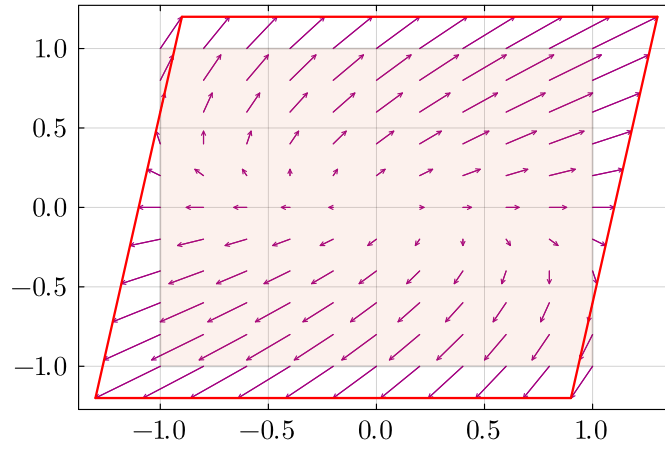


Figure 5: Resulting displacement field is shown.

(b) The divergence of the given vector field can be determined as

$$\begin{aligned}
 \nabla \cdot \mathbf{u} &= \mathbf{e}_i \left(\frac{\partial}{\partial x_i} \right) \cdot (u_j \mathbf{e}_j) \\
 &= \frac{\partial u_j}{\partial x_i} \delta_{ij} \quad (\text{since } \mathbf{e}_i \cdot \mathbf{e}_j = \delta_{ij}) \\
 \Rightarrow \nabla \cdot \mathbf{u} &= \frac{\partial u_i}{\partial x_i}.
 \end{aligned}$$

Thus, for the given problem, the divergence of a vector field can be determined as

$$\nabla \cdot \mathbf{u} = \frac{\partial u_1}{\partial x_1} + \frac{\partial u_2}{\partial x_2}$$

$$= 0.1 + 0.2$$

$$\implies \nabla \cdot \mathbf{u} = 0.3.$$

(c) The finite strain tensor, \mathbf{E} , can be expressed as

$$\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I}),$$

where $\mathbf{F} = \nabla \mathbf{u} + \mathbf{I}$, which can be determined as

$$\begin{aligned} \mathbf{F} &= \nabla \mathbf{u} + \mathbf{I} \\ &= \left(\frac{\partial u_i}{\partial x_j} + \delta_{ij} \right) \mathbf{e}_i \otimes \mathbf{e}_j \\ \implies F_{ij} &= \frac{\partial u_i}{\partial x_j} + \delta_{ij}. \end{aligned}$$

Thus the components of \mathbf{F} can be determined as

$$[\mathbf{F}] = \begin{bmatrix} \frac{\partial u_1}{\partial x_1} + 1 & \frac{\partial u_1}{\partial x_2} \\ \frac{\partial u_2}{\partial x_1} & \frac{\partial u_2}{\partial x_2} + 1 \end{bmatrix} = \begin{bmatrix} 1.1 & 0.2 \\ 0.0 & 1.2 \end{bmatrix}.$$

The strain, \mathbf{E} , can be computed using its component form as

$$\begin{aligned} [\mathbf{E}] &= \frac{1}{2} ([\mathbf{F}]^T [\mathbf{F}] - [\mathbf{I}]) \\ &= \frac{1}{2} \left(\begin{bmatrix} 1.1 & 0.0 \\ 0.2 & 1.2 \end{bmatrix} \begin{bmatrix} 1.1 & 0.2 \\ 0.0 & 1.2 \end{bmatrix} - \begin{bmatrix} 1.0 & 0.0 \\ 0.0 & 1.0 \end{bmatrix} \right) = \begin{bmatrix} 0.105 & 0.11 \\ 0.11 & 0.24 \end{bmatrix}. \end{aligned}$$

The linearized strain, $\mathbf{E}_s = \frac{1}{2}(\mathbf{F}^T + \mathbf{F} - 2\mathbf{I})$, can be computed using its component form as

$$\begin{aligned} [\mathbf{E}_s] &= \frac{1}{2} ([\mathbf{F}^T] + [\mathbf{F}] - 2[\mathbf{I}]) \\ &= \frac{1}{2} \left(\begin{bmatrix} 1.1 & 0.0 \\ 0.2 & 1.2 \end{bmatrix} + \begin{bmatrix} 1.1 & 0.2 \\ 0.0 & 1.2 \end{bmatrix} - \begin{bmatrix} 2.0 & 0.0 \\ 0.0 & 2.0 \end{bmatrix} \right) = \begin{bmatrix} 0.1 & 0.1 \\ 0.1 & 0.2 \end{bmatrix}. \end{aligned}$$

4. Consider the motion given by $[\mathbf{F}] = \begin{bmatrix} \lambda(t) & 0 & 0 \\ 0 & \lambda(t) & 0 \\ 0 & 0 & \lambda(t) \end{bmatrix}$, where $\lambda(t)$ is a time-dependent function.

Determine the values of $\lambda(t)$ for which the motion is a rigid body motion.

Solution: A motion is a rigid body motion if it preserves the distance between any two material points. This is true if and only if the deformation gradient \mathbf{F} is an orthogonal tensor, which requires

the right Cauchy-Green deformation tensor $\mathbf{C} = \mathbf{F}^T \mathbf{F}$ to be equal to the identity tensor \mathbf{I} .

The deformation gradient tensor \mathbf{F} is given in matrix form as:

$$[\mathbf{F}] = \begin{bmatrix} \lambda(t) & 0 & 0 \\ 0 & \lambda(t) & 0 \\ 0 & 0 & \lambda(t) \end{bmatrix} = \lambda(t)[\mathbf{I}]$$

The transpose of \mathbf{F} is identical:

$$[\mathbf{F}]^T = \begin{bmatrix} \lambda(t) & 0 & 0 \\ 0 & \lambda(t) & 0 \\ 0 & 0 & \lambda(t) \end{bmatrix}$$

The right Cauchy-Green deformation tensor \mathbf{C} is $\mathbf{C} = \mathbf{F}^T \mathbf{F}$. In matrix form:

$$\begin{aligned} [\mathbf{C}] &= [\mathbf{F}]^T [\mathbf{F}] = \begin{bmatrix} \lambda(t) & 0 & 0 \\ 0 & \lambda(t) & 0 \\ 0 & 0 & \lambda(t) \end{bmatrix} \begin{bmatrix} \lambda(t) & 0 & 0 \\ 0 & \lambda(t) & 0 \\ 0 & 0 & \lambda(t) \end{bmatrix} \\ &= \begin{bmatrix} \lambda^2(t) & 0 & 0 \\ 0 & \lambda^2(t) & 0 \\ 0 & 0 & \lambda^2(t) \end{bmatrix} = \lambda^2(t)[\mathbf{I}] \end{aligned}$$

For a rigid body motion, we must have $\mathbf{C} = \mathbf{I}$.

$$[\mathbf{C}] = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

By comparing the components, we get:

$$\lambda^2(t) = 1 \implies \lambda(t) = \pm 1$$

However, a motion is typically required to be a proper rigid body motion, which means it must also satisfy the condition $\det(\mathbf{F}) = +1$.

We calculate the determinant of \mathbf{F} :

$$\det(\mathbf{F}) = \det \begin{pmatrix} \lambda(t) & 0 & 0 \\ 0 & \lambda(t) & 0 \\ 0 & 0 & \lambda(t) \end{pmatrix} = \lambda^3(t)$$

We check our two possible values for $\lambda(t)$:

- If $\lambda(t) = 1$, then $\det(\mathbf{F}) = (1)^3 = 1$. This satisfies the condition. This corresponds to the identity transformation (no motion).

- If $\lambda(t) = -1$, then $\det(\mathbf{F}) = (-1)^3 = -1$. This does not satisfy the condition. This motion is an inversion (a reflection through the origin), which is an improper rigid body motion.

Therefore, for the motion to be a proper rigid body motion, the only possible value is:

$$\lambda(t) = 1$$

5. Consider the motion given by $\mathbf{x}' = \mathbf{x} + t^2 \mathbf{e}_1 + \sin(t) \mathbf{e}_2$. Determine if the motion is a rigid body motion by computing the strain tensor.

Solution: The given motion is a pure translation. The position components are:

$$\begin{aligned} x'_1 &= x_1 + t^2 \\ x'_2 &= x_2 + \sin(t) \end{aligned}$$

A motion is a rigid body motion if the strain tensor is zero. We first compute the deformation gradient tensor \mathbf{F} , where $F_{ij} = \frac{\partial x'_i}{\partial x_j}$.

$$[\mathbf{F}] = \begin{bmatrix} \frac{\partial x'_1}{\partial x_1} & \frac{\partial x'_1}{\partial x_2} \\ \frac{\partial x'_2}{\partial x_1} & \frac{\partial x'_2}{\partial x_2} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} = [\mathbf{I}]$$

The deformation gradient is the identity tensor. Now, we compute the Green-Lagrange strain tensor, $\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I})$.

$$\begin{aligned} [\mathbf{F}]^T [\mathbf{F}] &= [\mathbf{I}]^T [\mathbf{I}] = [\mathbf{I}] \\ [\mathbf{E}] &= \frac{1}{2}([\mathbf{I}] - [\mathbf{I}]) = \frac{1}{2} \begin{bmatrix} 0 & 0 \\ 0 & 0 \end{bmatrix} = [\mathbf{0}] \end{aligned}$$

Since the strain tensor \mathbf{E} is the zero tensor, the motion is a rigid body motion (specifically, a pure translation).

6. Consider the motion given by $\mathbf{x}' = \mathbf{Q}(t) \mathbf{x} + \mathbf{c}(t)$, where $\mathbf{Q}(t)$ is a time-dependent rotation matrix and $\mathbf{c}(t)$ is a time-dependent translation vector. Determine if the motion is a rigid body motion.

Solution: A motion is a rigid body motion if the associated strain tensor is zero. The given motion in component form is $x'_i = Q_{ij}(t)x_j + c_i(t)$.

First, we compute the deformation gradient tensor \mathbf{F} , where $F_{ik} = \frac{\partial x'_i}{\partial x_k}$.

$$\begin{aligned} F_{ik} &= \frac{\partial}{\partial x_k} (Q_{ij}(t)x_j + c_i(t)) \\ &= Q_{ij}(t) \frac{\partial x_j}{\partial x_k} + \frac{\partial c_i(t)}{\partial x_k} \\ &= Q_{ij}(t) \delta_{jk} + 0 \\ &= Q_{ik}(t) \end{aligned}$$

Thus, the deformation gradient tensor is the rotation tensor, $[\mathbf{F}] = [\mathbf{Q}(t)]$.

Next, we compute the Green-Lagrange strain tensor, $\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I})$.

$$[\mathbf{E}] = \frac{1}{2} ([\mathbf{Q}(t)]^T [\mathbf{Q}(t)] - [\mathbf{I}])$$

By definition, a rotation matrix \mathbf{Q} is an orthogonal matrix, which satisfies the property $\mathbf{Q}^T \mathbf{Q} = \mathbf{I}$.

Substituting this property into the strain equation:

$$[\mathbf{E}] = \frac{1}{2} ([\mathbf{I}] - [\mathbf{I}]) = [\mathbf{0}]$$

Since the Green-Lagrange strain tensor \mathbf{E} is the zero tensor, the motion is a rigid body motion.

7. Consider the motion given by $\mathbf{u} = a x_1 \mathbf{e}_1 + b x_2 \mathbf{e}_2 + c x_3 \mathbf{e}_3$. Determine the values of a , b , and c for which the motion is a rigid body motion by computing the strain tensor.

Solution: The displacement vector \mathbf{u} is given. The relationship between the deformed position \mathbf{x}' and the original position \mathbf{x} is $\mathbf{x}' = \mathbf{x} + \mathbf{u}$.

The components of the deformed position are:

$$\begin{aligned} x'_1 &= x_1 + u_1 = x_1 + a x_1 = (1 + a)x_1 \\ x'_2 &= x_2 + u_2 = x_2 + b x_2 = (1 + b)x_2 \\ x'_3 &= x_3 + u_3 = x_3 + c x_3 = (1 + c)x_3 \end{aligned}$$

First, we compute the deformation gradient tensor \mathbf{F} , where $F_{ij} = \frac{\partial x'_i}{\partial x_j}$.

$$[\mathbf{F}] = \begin{bmatrix} \frac{\partial x'_1}{\partial x_1} & \frac{\partial x'_1}{\partial x_2} & \frac{\partial x'_1}{\partial x_3} \\ \frac{\partial x'_2}{\partial x_1} & \frac{\partial x'_2}{\partial x_2} & \frac{\partial x'_2}{\partial x_3} \\ \frac{\partial x'_3}{\partial x_1} & \frac{\partial x'_3}{\partial x_2} & \frac{\partial x'_3}{\partial x_3} \end{bmatrix} = \begin{bmatrix} (1 + a) & 0 & 0 \\ 0 & (1 + b) & 0 \\ 0 & 0 & (1 + c) \end{bmatrix}$$

Next, we compute the Green-Lagrange strain tensor, $\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I})$.

$$\begin{aligned} [\mathbf{F}]^T [\mathbf{F}] &= \begin{bmatrix} (1+a) & 0 & 0 \\ 0 & (1+b) & 0 \\ 0 & 0 & (1+c) \end{bmatrix} \begin{bmatrix} (1+a) & 0 & 0 \\ 0 & (1+b) & 0 \\ 0 & 0 & (1+c) \end{bmatrix} \\ &= \begin{bmatrix} (1+a)^2 & 0 & 0 \\ 0 & (1+b)^2 & 0 \\ 0 & 0 & (1+c)^2 \end{bmatrix} \end{aligned}$$

Substituting this into the strain equation:

$$\begin{aligned} [\mathbf{E}] &= \frac{1}{2} \left(\begin{bmatrix} (1+a)^2 & 0 & 0 \\ 0 & (1+b)^2 & 0 \\ 0 & 0 & (1+c)^2 \end{bmatrix} - \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \right) \\ &= \frac{1}{2} \begin{bmatrix} (1+a)^2 - 1 & 0 & 0 \\ 0 & (1+b)^2 - 1 & 0 \\ 0 & 0 & (1+c)^2 - 1 \end{bmatrix} \end{aligned}$$

A motion is a rigid body motion if the strain tensor \mathbf{E} is the zero tensor, $[\mathbf{E}] = [\mathbf{0}]$. This requires all diagonal components to be zero.

- $(1+a)^2 - 1 = 0 \implies (1+a)^2 = 1 \implies 1+a = \pm 1$. This gives $a = 0$ or $a = -2$.
- $(1+b)^2 - 1 = 0 \implies (1+b)^2 = 1 \implies 1+b = \pm 1$. This gives $b = 0$ or $b = -2$.
- $(1+c)^2 - 1 = 0 \implies (1+c)^2 = 1 \implies 1+c = \pm 1$. This gives $c = 0$ or $c = -2$.

Therefore, the motion is a rigid body motion (i.e., preserves distances) if and only if a, b , and c are chosen from the set $\{0, -2\}$.

8. Consider the motion given by $\mathbf{x}' = \mathbf{x} + a \sin(\omega t) \mathbf{e}_1 + b \cos(\omega t) \mathbf{e}_2$. Determine if the motion is a rigid body motion by computing the strain tensor.

Solution: The given motion is a pure translation, where the translation vector $\mathbf{c}(t) = a \sin(\omega t) \mathbf{e}_1 + b \cos(\omega t) \mathbf{e}_2$ depends only on time.

The components of the deformed position \mathbf{x}' are:

$$\begin{aligned} x'_1 &= x_1 + a \sin(\omega t) \\ x'_2 &= x_2 + b \cos(\omega t) \end{aligned}$$

First, we compute the deformation gradient tensor \mathbf{F} , where $F_{ij} = \frac{\partial x'_i}{\partial x_j}$.

$$[\mathbf{F}] = \begin{bmatrix} \frac{\partial x'_1}{\partial x_1} & \frac{\partial x'_1}{\partial x_2} \\ \frac{\partial x'_2}{\partial x_1} & \frac{\partial x'_2}{\partial x_2} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} = [\mathbf{I}]$$

Next, we compute the Green-Lagrange strain tensor, $\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I})$.

$$\begin{aligned} [\mathbf{F}]^T [\mathbf{F}] &= [\mathbf{I}]^T [\mathbf{I}] = [\mathbf{I}] \\ [\mathbf{E}] &= \frac{1}{2} ([\mathbf{I}] - [\mathbf{I}]) = [\mathbf{0}] \end{aligned}$$

Since the strain tensor \mathbf{E} is the zero tensor, the motion is a rigid body motion. This is expected, as a pure translation (even time-dependent) does not cause any deformation.

9. Consider the motion given by $\mathbf{F} = \mathbf{I} + \gamma(t) \mathbf{e}_1 \otimes \mathbf{e}_2$. Determine the values of $\gamma(t)$ for which the motion is a rigid body motion by computing the strain tensor.

Solution: The deformation gradient tensor \mathbf{F} is given. In matrix form, it can be written as:

$$[\mathbf{F}] = [\mathbf{I}] + \gamma(t)[\mathbf{e}_1 \otimes \mathbf{e}_2] = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} + \gamma(t) \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} = \begin{bmatrix} 1 & \gamma(t) \\ 0 & 1 \end{bmatrix}$$

This motion is a simple shear.

Next, we compute the Green-Lagrange strain tensor, $\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I})$.

First, find \mathbf{F}^T :

$$[\mathbf{F}]^T = \begin{bmatrix} 1 & 0 \\ \gamma(t) & 1 \end{bmatrix}$$

Now, compute $\mathbf{F}^T \mathbf{F}$:

$$\begin{aligned} [\mathbf{F}]^T [\mathbf{F}] &= \begin{bmatrix} 1 & 0 \\ \gamma(t) & 1 \end{bmatrix} \begin{bmatrix} 1 & \gamma(t) \\ 0 & 1 \end{bmatrix} \\ &= \begin{bmatrix} 1 & \gamma(t) \\ \gamma(t) & \gamma(t)^2 + 1 \end{bmatrix} \end{aligned}$$

Now, compute \mathbf{E} :

$$\begin{aligned} [\mathbf{E}] &= \frac{1}{2} \left(\begin{bmatrix} 1 & \gamma(t) \\ \gamma(t) & \gamma(t)^2 + 1 \end{bmatrix} - \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \right) \\ &= \frac{1}{2} \begin{bmatrix} 0 & \gamma(t) \\ \gamma(t) & \gamma(t)^2 \end{bmatrix} \end{aligned}$$

A motion is a rigid body motion if the strain tensor \mathbf{E} is the zero tensor, $[\mathbf{E}] = [\mathbf{0}]$. This requires all components of $[\mathbf{E}]$ to be zero.

- $\frac{1}{2}\gamma(t) = 0 \implies \gamma(t) = 0$
- $\frac{1}{2}\gamma(t)^2 = 0 \implies \gamma(t) = 0$

Therefore, the motion is a rigid body motion if and only if $\gamma(t) = 0$ for all time t . A non-zero shear $\gamma(t)$ always causes deformation.

10. Consider the motion given by $\mathbf{u} = \beta(t)(x_2 \mathbf{e}_1 - x_1 \mathbf{e}_2)$, where $\beta(t)$ is a time-dependent function. Determine the values of $\beta(t)$ for which the motion is a rigid body motion.

Solution: The displacement vector \mathbf{u} has components:

$$\begin{aligned} u_1 &= \beta(t)x_2 \\ u_2 &= -\beta(t)x_1 \end{aligned}$$

The relationship between the deformed position \mathbf{x}' and the original position \mathbf{x} is $\mathbf{x}' = \mathbf{x} + \mathbf{u}$. The components of the deformed position are:

$$\begin{aligned} x'_1 &= x_1 + u_1 = x_1 + \beta(t)x_2 \\ x'_2 &= x_2 + u_2 = x_2 - \beta(t)x_1 \end{aligned}$$

First, we compute the deformation gradient tensor \mathbf{F} , where $F_{ij} = \frac{\partial x'_i}{\partial x_j}$.

$$[\mathbf{F}] = \begin{bmatrix} \frac{\partial x'_1}{\partial x_1} & \frac{\partial x'_1}{\partial x_2} \\ \frac{\partial x'_2}{\partial x_1} & \frac{\partial x'_2}{\partial x_2} \end{bmatrix} = \begin{bmatrix} 1 & \beta(t) \\ -\beta(t) & 1 \end{bmatrix}$$

Next, we compute the Green-Lagrange strain tensor, $\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I})$.

First, find \mathbf{F}^T :

$$[\mathbf{F}]^T = \begin{bmatrix} 1 & -\beta(t) \\ \beta(t) & 1 \end{bmatrix}$$

Now, compute $\mathbf{F}^T \mathbf{F}$:

$$\begin{aligned} [\mathbf{F}]^T [\mathbf{F}] &= \begin{bmatrix} 1 & -\beta(t) \\ \beta(t) & 1 \end{bmatrix} \begin{bmatrix} 1 & \beta(t) \\ -\beta(t) & 1 \end{bmatrix} \\ &= \begin{bmatrix} 1 + \beta(t)^2 & 0 \\ 0 & 1 + \beta(t)^2 \end{bmatrix} \end{aligned}$$

Now, compute \mathbf{E} :

$$\begin{aligned} [\mathbf{E}] &= \frac{1}{2} \left(\begin{bmatrix} 1 + \beta(t)^2 & 0 \\ 0 & 1 + \beta(t)^2 \end{bmatrix} - \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \right) \\ &= \frac{1}{2} \begin{bmatrix} \beta(t)^2 & 0 \\ 0 & \beta(t)^2 \end{bmatrix} \end{aligned}$$

A motion is a rigid body motion if the strain tensor \mathbf{E} is the zero tensor, $[\mathbf{E}] = [\mathbf{0}]$. This requires all components of $[\mathbf{E}]$ to be zero.

- $\frac{1}{2}\beta(t)^2 = 0 \implies \beta(t) = 0$

Therefore, the motion is a rigid body motion if and only if $\beta(t) = 0$ for all time t . This means no motion is occurring. Any non-zero $\beta(t)$ introduces a uniform stretch (since $\mathbf{F}^T \mathbf{F} \neq \mathbf{I}$), so it is not a rigid motion.