

$$m_1 = 4 \cdot 10^9 \text{ kg}$$

$$m_2 = 2m_1$$

Sono a dist. molto grande
= sono a dist. infinita

$$R = 6.67 \cdot 10^3 \text{ km}$$

$$K_{\text{TOT}} = ?$$

$$V_{\text{CM}} = ?$$

$$V_1 = ?$$

$$(1) E_{\infty} = K_{\infty} + U_{\infty} = 0 + 0 = 0$$

$$E_f = K_f + U_f$$

$$E_{\infty} = E_f \quad \rightsquigarrow \quad 0 = E_f \quad K_f + U_f = 0 \quad \rightsquigarrow \quad K_f = -U_f$$

$$K_f = - \left(-G \frac{m_1 m_2}{R} \right) = G \frac{2m_1^2}{R} \approx 320 \text{ J}$$

$$(2) V_{\text{CM}} = ? \quad \vec{p}_{\text{TOT}} = m_{\text{TOT}} \cdot \vec{V}_{\text{CM}}$$

$$\text{la qdm. si conserva.} \quad \vec{p}_{\text{TOT}} = m_1 \vec{v}_{1,\infty} + m_2 \vec{v}_{2,\infty} = 0$$

$$0 = m_{\text{TOT}} \cdot \vec{V}_{\text{CM}} \Rightarrow \vec{V}_{\text{CM}} = 0$$

$$(3) \text{ Si conserva } E: \quad E_{\infty} = 0 = E_f = \overbrace{\frac{1}{2} m_1 v_1^2 + \frac{1}{2} m_2 v_2^2}^{K_f} + U_f$$

$$\text{Si conserva qdm:} \quad p_{\infty} = 0 = p_f = m_1 v_1 - m_2 v_2$$

$$\text{So da } 2m_1 = m_2$$

$$\left\{ \begin{array}{l} m_1 v_1 = 2m_1 v_2 \\ -U_f = \frac{1}{2} m_1 v_1^2 + \frac{1}{2} 2m_1 v_2^2 \end{array} \right.$$

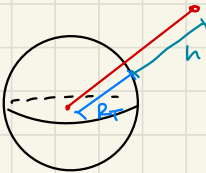
$$\left\{ \begin{array}{l} v_1 = 2v_2 \\ \frac{G 2m_1^2}{R} = \frac{1}{2} m_1 \cdot 4v_2^2 + m_1 v_2^2 \end{array} \right.$$

$$\begin{cases} V_1 = 2V_2 \rightarrow [V_1^2 = 4V_2^2] \rightsquigarrow V_1^2 = \frac{8}{3} \frac{Gm_1}{R} \\ 3V_2^2 = \frac{2Gm_1}{R} \rightsquigarrow V_2^2 = \frac{2}{3} \frac{Gm_1}{R} \end{cases}$$

$$V_1^2 = \frac{8}{3} \cdot \frac{6.67 \cdot 10^{-11} \cdot 4 \cdot 10^3}{6.67 \cdot 10^6} \frac{\text{m}^2}{\text{s}^2} = \frac{32}{3} \cdot 10^{-8} \frac{\text{m}^2}{\text{s}^2} = 10,6 \cdot 10^{-8} \frac{\text{m}^2}{\text{s}^2}$$

$$V_1 \approx 3,3 \cdot 10^{-4} \frac{\text{m}}{\text{s}}$$

Pag 313 n 3



$$h = 408 \text{ km}$$

Quante orbite in $24 \text{ h} = d$

Se calcolo il periodo T , ho tutto.

Per la Terza Legge di Keplero

$$\frac{a^3}{T^2} = k = \frac{GM_T}{4\pi^2}$$

dove a è il semiasse² dell'orbita del satellite. Ma l'orbita è circolare quindi $a = R_T + h$

$$T^2 = \frac{4\pi^2}{GM_T} \cdot (R_T + h)^3$$

$$T = 2\pi(R_T + h) \sqrt{\frac{R_T + h}{GM_T}}$$

orbite in un giorno $\frac{d}{T} = \frac{d}{2\pi(R_T + h)} \cdot \sqrt{\frac{GM_T}{R_T + h}} \approx 15$

