# TTK4115 Lecture 3/4

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## This lecture

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- 2. Realizations
- 3. Discretization
- 4. Controllability

Controllability Gramians

Eigenvector tests

Controllability in practice

Controllability indices

## **Topic**

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2 Realizations

3. Discretization

#### 4. Controllability

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## **Canonical Forms**

#### Canonical Forms

- Using the change of basis  $\mathbf{x} = \mathbf{T}\bar{\mathbf{x}}$  we can change a system into infinitely many similar forms.
- Some of these forms are more useful than others.
- Some of these are called canonical.

## Equivalence/Similarity transform

$$\dot{x} = Ax + Bu$$
 $y = Cx + Du$ 

$$\dot{\bar{\mathbf{x}}} = \mathbf{T}^{-1}\mathbf{A}\mathbf{T}\bar{\mathbf{x}} + \mathbf{T}^{-1}\mathbf{B}\mathbf{u}$$
 $\mathbf{y} = \mathbf{C}\mathbf{T}\bar{\mathbf{x}} + \mathbf{D}\mathbf{u}$ 

## **Canonical Forms**

## Canonical Forms<sup>1</sup>

- Jordan Form
- Modal Form
- Companion form
- Controllable form
- Observable form

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<sup>&</sup>lt;sup>1</sup>This list is not exhaustive.

## **Canonical Forms**

#### Jordan Form

The Jordan form is the most convenient to use when solving the system. We have seen that this form is very practical for finding solutions to LTI systems:

$$\mathbf{y}(t) = \mathbf{C}e^{\mathbf{A}t}\mathbf{x}_0 + \mathbf{C}\int_0^t e^{\mathbf{A}(t- au)}\mathbf{B}\mathbf{u}( au)d au + \mathbf{D}\mathbf{u}(t)$$

## Diagonal matrix

$$\mathbf{A} = \mathbf{\Lambda} = \begin{bmatrix} \lambda_1 & 0 & 0 \\ 0 & \lambda_2 & 0 \\ 0 & 0 & \lambda_3 \end{bmatrix} \quad \Rightarrow \mathbf{e}^{\mathbf{A}t} = \begin{bmatrix} \mathbf{e}^{t\lambda_1} & 0 & 0 \\ 0 & \mathbf{e}^{t\lambda_2} & 0 \\ 0 & 0 & \mathbf{e}^{t\lambda_3} \end{bmatrix}$$

## Jordan Block

$$\mathbf{A} = \mathbf{J} = \begin{bmatrix} \lambda & 1 & 0 & 0 \\ 0 & \lambda & 1 & 0 \\ 0 & 0 & \lambda & 1 \\ 0 & 0 & 0 & \lambda \end{bmatrix} \quad \Rightarrow e^{\mathbf{A}t} = \begin{bmatrix} e^{t\lambda} & e^{t\lambda}t & \frac{1}{2!}e^{t\lambda}t^2 & \frac{1}{3!}e^{t\lambda}t^3 \\ 0 & e^{t\lambda} & e^{t\lambda}t & \frac{1}{2!}e^{t\lambda}t^2 \\ 0 & 0 & e^{t\lambda} & e^{t\lambda}t \\ 0 & 0 & 0 & e^{t\lambda} \end{bmatrix}$$

## **Topic**

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#### 2. Realizations

#### 3 Discretization

#### 4. Controllability

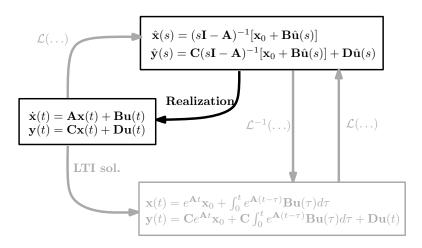
Controllability Gramians

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## LTI systems overview



## Realizations

The final piece in the diagram

Key points

#### Realization

- We have seen that a transformation  $\bar{\mathbf{x}} = \mathbf{T}\mathbf{x}$  can change the state equation..
- but the transfer function remains the same.
- When we realize, we start with a transfer function  $\mathbf{H}(s)$ ..
- and generate a state-space {A, B, C, D}
- that yields  $\mathbf{H}(s) = \mathbf{C}(s\mathbb{I} \mathbf{A})^{-1}\mathbf{B} + \mathbf{D}$

#### Note

There are infinitely many state-spaces we could realize to!

## Note

We usually go for a canonical form.

Conditions

## **Proper**

A transfer function must be proper to have a realization:

$$h(s) = \frac{n(s)}{d(s)} \quad \Rightarrow \deg d(s) \ge n(s)$$

$$|h_{\!p}(j\infty)|<\infty,\quad |h_{\!sp}(j\infty)|=0$$

#### Rational

A transfer function must be rational to have a realization.

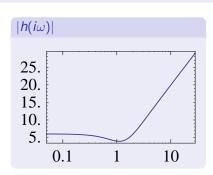
- The degrees of the numerator and denominator must be finite.
- All lumped LTI systems are rational.

## Proper transfer functions

We must have a proper transfer function for realization.

## Example

$$h(s) = \frac{2 + 2s + s^2}{1 + s}$$



#### Question:

Is this a proper transfer function?

## Signals are amplified

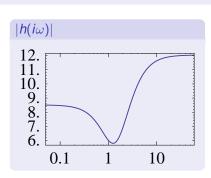
at infinite frequencies.. no device can do this

## Proper transfer functions

Proper transfer functions behave nicely at high frequencies

## Example

$$h(s) = \frac{4(2+2s+s^2)}{3+4s+s^2}$$



## Question:

Is this a proper transfer function?

#### **Answer**

Yes, but not strictly proper. The transferfunction is finite at infinite frequencies:

$$\lim_{\omega\to\infty}h(j\omega)=h_\infty\neq 0$$

## Quiz

Are these transfer functions realizable?

- $g_1(s) = \frac{1}{s}$ 
  - $g_2(s) = s$
  - $g_3(s) = \frac{1}{s+1}$
  - $g_4(s) = \frac{1}{s-1}$
  - $g_5(s) = \frac{s}{s+1}$
  - $g_6(s) = e^{-\tau s}, \quad \tau > 0$
  - $g_7(s) = \frac{1 \frac{\tau}{2}s}{1 + \frac{\tau}{2}s}, \quad \tau > 0$

# Improper/Proper/Strictly proper

Improper

$$H_{i.p.}(s) = k_P + k_D s + \frac{k_I}{s} = \frac{k_D s^2 + k_P s + k_I}{s}$$
 PID regulator

Proper

$$H_p(s) = \frac{s}{T_{s+1}}$$
 Band-limited differentiator

Strictly proper

$$H_{s.p.}(s) = \frac{1}{s^2 m + ds + k}$$
 Msd.

Strictly proper transfer functions

## Decomposition

We decompose the proper transfer function as:

strictly proper constant

$$\mathbf{G}(s) = \mathbf{G}_{sp}(s) + \mathbf{G}_{\infty}$$

Question: where is D?

$$\hat{\mathbf{y}}(s) = \mathbf{G}_{sp}(s)\hat{\mathbf{u}}(s) + \mathbf{G}_{\infty}\hat{\mathbf{u}}(s)$$

$$\hat{\mathbf{y}}(s) = \mathbf{C}(s\mathbb{I} - \mathbf{A})^{-1}\mathbf{B}\hat{\mathbf{u}}(s) + \mathbf{D}\hat{\mathbf{u}}(s)$$

Canonical forms

## Matching

The crucial next step is to select a state-space model with unknown coefficients:

$$\Sigma_r:~\{\textbf{A},\textbf{B},\textbf{C},\textbf{D}\}$$

that can represent our transfer-function.

## Matching

We shall use the **Controllable Canonical Form** today. This is one of many choices.

Controllable form

## Let's pick a nice A for the realization

Four states  $\rightarrow$  up to  $s^4$  in the denominator of g(s).

$$\mathbf{A} = \begin{bmatrix} -\alpha_1 & -\alpha_2 & -\alpha_3 & -\alpha_4 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix}$$

$$(sI - A)^{-1}$$

$$(\mathbf{s}\mathbb{I} - \mathbf{A})^{-1} = \frac{1}{\mathbf{s}^4 + \mathbf{s}^3 \alpha_1 + \mathbf{s}^2 \alpha_2 + \mathbf{s} \alpha_3 + \alpha_4} \begin{bmatrix} s^3 & -s^2 \alpha_2 - s \alpha_3 - \alpha_4 & -s^2 \alpha_3 - s \alpha_4 & -s^2 \alpha_4 \\ s^2 & s^3 + s^2 \alpha_1 & -s \alpha_3 - \alpha_4 & -s \alpha_4 \\ s & s^2 + s \alpha_1 & s^3 + s^2 \alpha_1 + s \alpha_2 & -\alpha_4 \\ 1 & s + \alpha_1 & s^2 + s \alpha_1 + \alpha_2 & s^3 + s^2 \alpha_1 + s \alpha_4 \end{bmatrix}$$

Controllable form

## Let's pick a nice B for the realization too

Four states  $\rightarrow$  up to  $s^4$  in the denominator of g(s).

$$\mathbf{A} = \left[ \begin{array}{cccc} -\alpha_1 & -\alpha_2 & -\alpha_3 & -\alpha_4 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{array} \right] \quad \mathbf{B} = \left[ \begin{array}{c} 1 \\ 0 \\ 0 \\ 0 \end{array} \right]$$

$$(sI - A)^{-1}B$$

$$(s\mathbb{I} - \mathbf{A})^{-1}\mathbf{B} = \begin{bmatrix} s^3 \\ s^2 \\ s \\ 1 \end{bmatrix} \frac{1}{s^4 + s^3\alpha_1 + s^2\alpha_2 + s\alpha_3 + \alpha_4}$$

## What about C?

Four states  $\rightarrow$  up to  $s^4$  in the denominator of g(s).

$$\mathbf{A} = \begin{bmatrix} -\alpha_1 & -\alpha_2 & -\alpha_3 & -\alpha_4 \\ 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \quad \mathbf{B} = \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix} \quad \mathbf{C} = \begin{bmatrix} n_1 & n_2 & n_3 & n_4 \end{bmatrix}$$

$$C(sI - A)^{-1}B$$

$$\mathbf{C}(s\mathbb{I} - \mathbf{A})^{-1}\mathbf{B} = [ n_1 \quad n_2 \quad n_3 \quad n_4 ] \begin{bmatrix} s^3 \\ s^2 \\ s \\ 1 \end{bmatrix} \frac{1}{s^4 + s^3\alpha_1 + s^2\alpha_2 + s\alpha_3 + \alpha_4}$$

$$C(sI - A)^{-1}B$$

$$\mathbf{G}_{sp}(s) = \mathbf{C}(s\mathbb{I} - \mathbf{A})^{-1}\mathbf{B} = \frac{s^3 n_1 + s^2 n_2 + s n_3 + n_4}{s^4 + s^3 \alpha_1 + s^2 \alpha_2 + s \alpha_3 + \alpha_4}$$

$$\frac{x(s)}{f(s)} = \frac{y(s)}{u(s)} = \frac{1}{ms^2 + sd + k} = \frac{1/m}{s^2 + s(d/m) + (k/m)}$$

Controllable canonical form, n = 2

$$\mathbf{A} = \left[ \begin{array}{cc} -\alpha_1 & -\alpha_2 \\ 1 & 0 \end{array} \right] \quad \mathbf{B} = \left[ \begin{array}{c} 1 \\ 0 \end{array} \right] \quad \mathbf{C} = \left[ \begin{array}{c} n_1 & n_2 \end{array} \right]$$

 $C(sI - A)^{-1}B$ 

$$\mathbf{G}_{sp}(s) = \mathbf{C}(s\mathbb{I} - \mathbf{A})^{-1}\mathbf{B} = \frac{s \underbrace{n_1}^{0} + \underbrace{n_2}^{1/m}}{s^2 + s \underbrace{\alpha_1}_{d/m} + \underbrace{\alpha_2}_{k/m}}$$

#### Realization

The mass spring damper back on state-space form:

$$\mathbf{A} = \left[ \begin{array}{cc} -d/m & -k/m \\ 1 & 0 \end{array} \right] \quad \mathbf{B} = \left[ \begin{array}{c} 1 \\ 0 \end{array} \right] \quad \mathbf{C} = \left[ \begin{array}{c} 0 & 1/m \end{array} \right]$$

Controllable form

## Controllable canonical form: *p* inputs, *q* outputs

$$\mathbf{A} = \left[ \begin{array}{cccc} -\alpha_1 \mathbb{I}_{\rho} & -\alpha_2 \mathbb{I}_{\rho} & -\alpha_3 \mathbb{I}_{\rho} & -\alpha_4 \mathbb{I}_{\rho} \\ \mathbb{I}_{\rho} & \mathbf{0} & \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \mathbb{I}_{\rho} & \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} & \mathbb{I}_{\rho} & \mathbf{0} \end{array} \right] \quad \mathbf{B} = \left[ \begin{array}{c} \mathbb{I}_{\rho} \\ \mathbf{0} \\ \mathbf{0} \\ \mathbf{0} \end{array} \right] \quad \mathbf{C} = \left[ \begin{array}{c} \mathbf{N}_1 & \mathbf{N}_2 & \mathbf{N}_3 & \mathbf{N}_4 \end{array} \right]$$

$$\mathbf{C}(s\mathbb{I} - \mathbf{A})^{-1}\mathbf{B}$$

$$\mathbf{G}_{sp}(s) = \mathbf{C}(s\mathbb{I} - \mathbf{A})^{-1}\mathbf{B} = \frac{s^3\mathbf{N}_1 + s^2\mathbf{N}_2 + s\mathbf{N}_3 + \mathbf{N}_4}{s^4 + s^3\alpha_1 + s^2\alpha_2 + s\alpha_3 + \alpha_4}$$

## d(s)

We have to find the common denominator of  $\mathbf{G}_{sp}(s)$ :  $d(s) = s^4 + s^3\alpha_1 + s^2\alpha_2 + s\alpha_3 + \alpha_4$ 

## Example 1

Realize G(s) to controllable canonical form:

$$\mathbf{G}(s) = \begin{bmatrix} \frac{-10+4s}{1+2s} & \frac{3}{2+s} \\ \frac{1}{(2+s)(1+2s)} & \frac{1+s}{(2+s)^2} \end{bmatrix}$$

Find **G**<sub>∞</sub>

$$\mathbf{D} = \mathbf{G}_{\infty} = \lim_{s \to \infty} \begin{bmatrix} \frac{-10+4s}{1+2s} & \frac{3}{2+s} \\ \frac{1}{(2+s)(1+2s)} & \frac{1+s}{(2+s)^2} \end{bmatrix} = \begin{bmatrix} 2 & 0 \\ 0 & 0 \end{bmatrix}$$

Find 
$$\mathbf{G}_{sp} = \mathbf{G}(s) - \mathbf{G}_{\infty}$$

$$\mathbf{G}_{SP} = \left[ \begin{array}{cc} \frac{-10+4s}{1+2s} & \frac{3}{2+s} \\ \frac{1}{(2+s)(1+2s)} & \frac{1+s}{(2+s)^2} \end{array} \right] - \left[ \begin{array}{cc} 2 & 0 \\ 0 & 0 \end{array} \right] = \left[ \begin{array}{cc} -\frac{12}{1+2s} & \frac{3}{2+s} \\ \frac{1}{2+5s+2s^2} & \frac{1+s}{(2+s)^2} \end{array} \right]$$

Find common denominator d(s)

$$\mathbf{G}_{sp} = \frac{1}{s^3 + (9/2)s^2 + 6s + 2} \begin{bmatrix} -6(2+s)^2 & 3(1+s/2)(1+2s) \\ 1+s/2 & (1/2+s)(1+s) \end{bmatrix}$$

## Example 1

Realize G(s) to controllable canonical form:

$$\mathbf{G}_{Sp} = \frac{1}{d(s)} \left( \begin{bmatrix} -24 - 24s & 3 + \frac{15s}{2} \\ 1 + \frac{s}{2} & \frac{1}{2} + \frac{3s}{2} \end{bmatrix} + s^2 \begin{bmatrix} -6 & 3 \\ 0 & 1 \end{bmatrix} \right)$$
$$d(s) = s^3 + (9/2)s^2 + 6s + 2$$

Find numerator matrices N<sub>i</sub>

$$\mathbf{G}_{Sp} = \frac{1}{d(s)} \left( \begin{array}{c|c} \mathbf{N}_3 & \mathbf{N}_2 & \mathbf{N}_1 \\ \hline -24 & 3 \\ 1 & \frac{1}{2} \end{array} \right) + s \left[ \begin{array}{c|c} -24 & \frac{15}{2} \\ \frac{1}{2} & \frac{3}{2} \end{array} \right] + s^2 \left[ \begin{array}{c|c} -6 & 3 \\ 0 & 1 \end{array} \right] \right)$$
$$d(s) = s^3 + (9/2)s^2 + 6s + 2$$

## Example 1

Realize G(s) to controllable canonical form:

$$\mathbf{G}_{sp} = \frac{1}{d(s)} \begin{bmatrix} \mathbf{N}_3 + s\mathbf{N}_2 + s^2\mathbf{N}_1 \end{bmatrix} \quad \mathbf{G}_{\infty} = \mathbf{D} = \begin{bmatrix} 2 & 0 \\ 0 & 0 \end{bmatrix}$$

$$\mathbf{N}_1 = \begin{bmatrix} -6 & 3 \\ 0 & 1 \end{bmatrix} \quad \mathbf{N}_2 = \begin{bmatrix} -24 & \frac{15}{2} \\ \frac{1}{2} & \frac{3}{2} \end{bmatrix} \quad \mathbf{N}_3 = \begin{bmatrix} -24 & 3 \\ 1 & \frac{1}{2} \end{bmatrix}$$

$$d(s) = s^3 + \alpha_1 s^2 + \alpha_2 s + \alpha_3 = s^3 + (9/2)s^2 + 6s + 2$$

#### Realize:

$$\mathbf{A} = \left[ \begin{array}{ccc} -\alpha_1 \mathbb{I}_\rho & -\alpha_2 \mathbb{I}_\rho & -\alpha_3 \mathbb{I}_\rho \\ \mathbb{I}_\rho & \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \mathbb{I}_\rho & \mathbf{0} \end{array} \right] \quad \mathbf{B} = \left[ \begin{array}{c} \mathbb{I}_\rho \\ \mathbf{0} \\ \mathbf{0} \end{array} \right] \quad \mathbf{C} = \left[ \begin{array}{ccc} \mathbf{N}_1 & \mathbf{N}_2 & \mathbf{N}_3 \end{array} \right]$$

## Topic

1. Canonical Forms

#### 2 Realizations

#### 3. Discretization

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## Analog state space model

The continuous state space model is analog<sup>2</sup>:

$$\dot{x} = Ax + Bu$$
 $y = Cx + Du$ 

To simulate it as is would require an analog computer.

#### Discretization

Discretization is a necessary step for computer simulation. A recursive model is sought:

$$\mathbf{x}[k+1] = \mathbf{A}_d \mathbf{x}[k] + \mathbf{B}_d \mathbf{u}[k]$$
  
 $\mathbf{y}[k] = \mathbf{C}\mathbf{x}[k] + \mathbf{D}\mathbf{u}[k]$ 

Many methods are available for obtaining a discretized model. We examine the two most common methods: **Exact** and **Euler** discretization<sup>3</sup>.

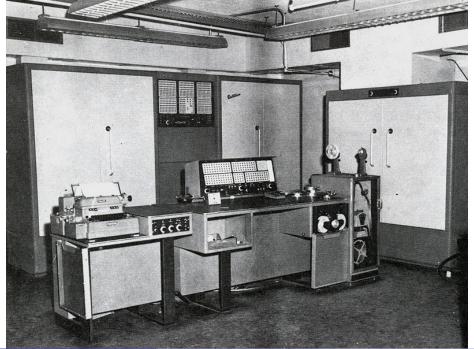
## Approach

$$\mathsf{TF} \xrightarrow{\mathsf{Realize}} \mathsf{CLTI} \xrightarrow{\mathsf{Discretize}} \mathsf{DLTI} \xrightarrow{\mathsf{Recursion}} \mathsf{Solution}$$

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<sup>&</sup>lt;sup>2</sup>Some systems are discrete by nature, such as financial systems or discrete filters. Most plants will however be continuous as they are based on a physical model.

<sup>&</sup>lt;sup>3</sup>These are repectively the best and worst of the common methods.



#### LTI solution

The exact solution of the LTI system forms the theoretical basis of **exact** discretization.

$$\dot{x} = Ax + Bu$$
 $y = Cx + Du$ 

$$\mathbf{y}(t) = \mathbf{C}e^{\mathbf{A}t}\mathbf{x}_0 + \mathbf{C}\int_0^t e^{\mathbf{A}(t-\tau)}\mathbf{B}\mathbf{u}(\tau)d\tau + \mathbf{D}\mathbf{u}(t)$$

## Sampling

Time is discretized into intervals of duration T: t = kT. Sample index is denoted k. The state solution from one sample to the next is:

$$\mathbf{x}((k+1)T) = e^{\mathbf{A}T}\mathbf{x}(kT) + \int_{kT}^{(k+1)T} e^{\mathbf{A}[(k+1)T - \tau]} \mathbf{B}\mathbf{u}(\tau) d\tau$$

Here  $\mathbf{x}[k] \triangleq \mathbf{x}(t)|_{t=kT}$  serves as an initial condition

The resulting solution is evaluated at  $\mathbf{x}[k+1] \triangleq \mathbf{x}(t)|_{t=(k+1)T}$ .

## Piecewise constant input

The input is assumed to stay approximately constant between samples:

$$\mathbf{u}[k] \simeq \mathbf{u}(t), \quad kT \leq t \leq (k+1)T$$

## Sampled model

$$\mathbf{x}[k+1] = e^{\mathbf{A}T}\mathbf{x}[k] + \left(\int_{kT}^{(k+1)T} e^{\mathbf{A}[(k+1)T - \tau]} d\tau\right) \mathbf{B}\mathbf{u}[k]$$

#### Substitution rule

$$\int_{y(a)}^{y(b)} F(x) dx = \int_{a}^{b} F(y(x)) \frac{dy}{dx} dx$$

Change of variable:  $\alpha(\tau) \triangleq (k+1)T - \tau$ ,  $d\tau = -d\alpha$ 

Integration limits are simplified:

$$\tau_0 = kT \rightarrow \alpha_0 = T$$
,  $\tau_1 = (k+1)T \rightarrow \alpha_1 = 0$ 

along with integrand:

$$e^{\mathbf{A}[(k+1)T-\tau]} \rightarrow e^{\mathbf{A}\alpha}$$

$$\mathbf{B}_d = \left( \int_{kT}^{(k+1)T} e^{\mathbf{A}[(k+1)T - \tau]} d\tau \right) \mathbf{B} = \left( \int_0^T e^{\mathbf{A}\alpha} d\alpha \right) \mathbf{B}$$

## Discretization

## Exactly discretized model

$$\mathbf{x}[k+1] = \underbrace{e^{\mathbf{A}T}}_{\mathbf{A}_d} \mathbf{x}[k] + \underbrace{\left(\int_0^T e^{\mathbf{A}\alpha} d\alpha\right)}_{\mathbf{D}_d} \mathbf{B} \mathbf{u}[k]$$

$$\mathbf{y}[k] = \underbrace{\mathbf{C}}_{\mathbf{C}_d} \mathbf{x}[k] + \underbrace{\mathbf{D}}_{\mathbf{D}_d} \mathbf{u}[k]$$

## Discrete time system

This model is exact under the assumption:

$$\mathbf{u}[k] = \mathbf{u}(t), \quad kT \le t \le (k+1)T$$

It is recursive, and very efficient in implementation:

$$\mathbf{x}[k+1] = \mathbf{A}_{d}\mathbf{x}[k] + \mathbf{B}_{d}\mathbf{u}[k]$$
  
 $\mathbf{y}[k] = \mathbf{C}_{d}\mathbf{x}[k] + \mathbf{D}_{d}\mathbf{u}[k]$ 

## Discretization

#### Euler discretization

Euler's method proceeds via the definition of the derivative<sup>4</sup>:

$$\dot{\mathbf{x}}[k] \approx \frac{\mathbf{x}[k+1] - \mathbf{x}[k]}{T}$$

Thus:

$$\dot{\mathbf{x}}[k] = \mathbf{A}\mathbf{x}[k] + \mathbf{B}\mathbf{u}[k] \quad \Rightarrow \mathbf{x}[k+1] = \mathbb{I}\mathbf{x}[k] + T\mathbf{A}\mathbf{x} + T\mathbf{B}\mathbf{u}$$

## Stability

Euler's method may be unstable although the underlying plant is stable. This problem gets worse with larger timesteps. A mathematical criterion for first order systems may be stated as:

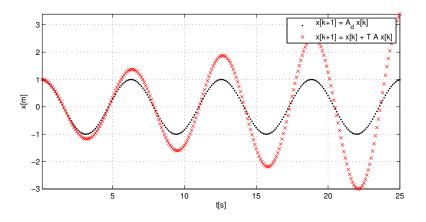
$$|1 + T\lambda| \le 1$$

Insufficiently stable systems or large timesteps will result in a divergent solution.

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 $<sup>^4</sup>x[k] = x(kT)$ 

$$\mathbf{x}[k+1] = \mathbf{A}_{d}\mathbf{x}[k] + \mathbf{B}_{d}\mathbf{u}[k]$$
  
 $\mathbf{x}_{e}[k+1] = \mathbf{x}_{e}[k] + T\mathbf{A}\mathbf{x}_{e}[k] + T\mathbf{B}\mathbf{u}[k]$ 



# **Topic**

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2. Realizations

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## Example

Consider the single input system:

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{b}u \quad \mathbf{A} \in \mathbb{R}^{4 \times 4}$$

## 4 steps forward

We step the system forward by infinitesimally small time steps  $\Delta t = \epsilon$ :

$$\mathbf{x}_{k+1} = \mathbf{x}_k + \epsilon \mathbf{A} \mathbf{x}_k + \epsilon \mathbf{b} u_k$$

Starting point is x<sub>0</sub>

$$\begin{array}{rcl} \mathbf{x}_1 & = & \mathbf{x}_0 + \epsilon \mathbf{A} \mathbf{x}_0 + \epsilon \mathbf{b} u_0 \\ \mathbf{x}_2 & = & \mathbf{x}_1 + \epsilon \mathbf{A} \mathbf{x}_1 + \epsilon \mathbf{b} u_1 \\ \mathbf{x}_3 & = & \mathbf{x}_2 + \epsilon \mathbf{A} \mathbf{x}_2 + \epsilon \mathbf{b} u_2 \\ \mathbf{x}_4 & = & \mathbf{x}_3 + \epsilon \mathbf{A} \mathbf{x}_3 + \epsilon \mathbf{b} u_3 \end{array}$$

Last step may be written as:

$$\mathbf{x}_4 = (\mathbb{I} + \epsilon \mathbf{A})^4 \, \mathbf{x}_0 + \epsilon \, (\mathbb{I} + \epsilon \mathbf{A})^3 \, \mathbf{b} u_0 + \epsilon \, (\mathbb{I} + \epsilon \mathbf{A})^2 \, \mathbf{b} u_1 + \epsilon \, (\mathbb{I} + \epsilon \mathbf{A}) \, \mathbf{b} u_2 + \epsilon \mathbf{b} u_3$$

## The *n*'th step is linear in the initial condition and the sequence of inputs:

$$\begin{aligned} \mathbf{x}_4 &= & \left( \mathbb{I} + \epsilon \mathbf{A} \right)^4 \mathbf{x}_0 + \epsilon \left( \mathbb{I} + \epsilon \mathbf{A} \right)^3 \mathbf{b} u_0 + \epsilon \left( \mathbb{I} + \epsilon \mathbf{A} \right)^2 \mathbf{b} u_1 + \epsilon \left( \mathbb{I} + \epsilon \mathbf{A} \right) \mathbf{b} u_2 + \epsilon \mathbf{b} u_3 \\ &= & \left( \mathbb{I} + \epsilon \mathbf{A} \right)^4 \mathbf{x}_0 + \epsilon \left[ & \left( \mathbb{I} + \epsilon \mathbf{A} \right)^3 \mathbf{b} & \left( \mathbb{I} + \epsilon \mathbf{A} \right)^2 \mathbf{b} & \left( \mathbb{I} + \epsilon \mathbf{A} \right) \mathbf{b} & \mathbf{b} & \right] \begin{bmatrix} u_0 \\ u_1 \\ u_2 \\ u_3 \end{bmatrix} \end{aligned}$$

## Gather linearly dependent columns:

## The *n*'th step is linear in the initial condition and the sequence of inputs

$$\mathbf{x}_{4} = \overbrace{\left(\mathbb{I} + \epsilon \mathbf{A}\right)^{4} \mathbf{x}_{0}}^{\text{zir}} + \underbrace{\left(\left[\begin{array}{ccc} \epsilon^{4} \mathbf{A}^{3} \mathbf{b} & \epsilon^{3} \mathbf{A}^{2} \mathbf{b} & \epsilon^{2} \mathbf{A} \mathbf{b} & \epsilon \mathbf{b} \end{array}\right] + LDC\right) \left[\begin{array}{c} u_{0} \\ u_{1} \\ u_{2} \\ u_{3} \end{array}\right]}_{l}$$

## Key idea

Iff  $[\epsilon^4 \mathbf{A}^3 \mathbf{b} \quad \epsilon^3 \mathbf{A}^2 \mathbf{b} \quad \epsilon^2 \mathbf{A} \mathbf{b} \quad \epsilon \mathbf{b}]$  has *full rank*, we can choose  $\mathbf{x}_4$  as we like with our inputs.

## Controllability matrix

Iff the controllability matrix has full rank: rank(C) = n

$$C \triangleq [ B AB \dots A^{(n-1)}b ]$$

the state can be placed anywhere with the right sequence of inputs.

-This is controllability.

## Topic

- 1. Canonical Forms
- 2. Realizations
- 3. Discretization
- 4. Controllability

#### Controllability Gramians

Eigenvector tests

Controllability in practice

Controllability indices

## Controllability Gramian

#### Given an LTI system:

$$\mathbf{x}(t) = e^{\mathbf{A}t}\mathbf{x}_0 + \int_0^t e^{\mathbf{A}(t- au)}\mathbf{B}\mathbf{u}( au)d au$$

#### Attempt to place the system at $\mathbf{x}_1$ at $t = t_1$ :

$$\mathbf{x}_1 = \mathbf{x}(t_1) = e^{\mathbf{A}t_1}\mathbf{x}_0 + \int_0^{t_1} e^{\mathbf{A}(t_1- au)}\mathbf{B}\mathbf{u}( au)d au$$

#### We clearly need the proper input signal $\mathbf{u}(t)$ to do this.

If we can find such an input, the system is controllable.

#### Controllability Gramian: definition

$$\mathbf{W}_{\mathcal{C}}(t) = \int_{0}^{t} e^{\mathbf{A}(t- au)} \mathbf{B} \mathbf{B}^{\mathsf{T}} e^{\mathbf{A}^{\mathsf{T}}(t- au)} d au$$

The existence of a nonsingular Controllability Gramian is important because it guarantees that a sufficient  $\mathbf{u}(t)$  exists.

## Place the system at $\mathbf{x}_1$ at $t = t_1$

$$\mathbf{x}_1 = \mathbf{x}(t_1) = e^{\mathbf{A}t_1}\mathbf{x}_0 + \int_0^{t_1} e^{\mathbf{A}(t_1-\tau)}\mathbf{B}\mathbf{u}(\tau)d\tau$$

#### We need an input $\mathbf{u}(t)$ to do this.

Educated guess:

$$\mathbf{u}(t) = -\mathbf{B}^{\mathsf{T}} e^{\mathbf{A}^{\mathsf{T}}(t_1 - t)} \mathbf{W}_c^{-1}(t_1) \left[ e^{\mathbf{A}t_1} \mathbf{x}_0 - \mathbf{x}_1 \right]$$

#### Result

$$\begin{split} \mathbf{x}_1 &= e^{\mathbf{A}t_1}\mathbf{x}_0 - \int_0^{t_1} e^{\mathbf{A}(t_1-\tau)}\mathbf{B} \left(\mathbf{B}^\mathsf{T} e^{\mathbf{A}^\mathsf{T}(t_1-\tau)}\mathbf{W}_c^{-1}(t_1) \left[e^{\mathbf{A}t_1}\mathbf{x}_0 - \mathbf{x}_1\right]\right) d\tau \\ &= e^{\mathbf{A}t_1}\mathbf{x}_0 - \overbrace{\left(\int_0^{t_1} e^{\mathbf{A}(t_1-\tau)}\mathbf{B}\mathbf{B}^\mathsf{T} e^{\mathbf{A}^\mathsf{T}(t_1-\tau)} d\tau\right)}^{\mathbf{W}_c^{-1}(t_1) \left[e^{\mathbf{A}t_1}\mathbf{x}_0 - \mathbf{x}_1\right]} \\ &= e^{\mathbf{A}t_1}\mathbf{x}_0 - \left[e^{\mathbf{A}t_1}\mathbf{x}_0 - \mathbf{x}_1\right] = \underline{\mathbf{x}_1} \end{split}$$

Iff  $\mathbf{W}_c(t)$  is invertible, the system is controllable.

$$\mathbf{W}_{\mathcal{C}}(t) = \int_{0}^{t} e^{\mathbf{A}(t- au)} \mathbf{B} \mathbf{B}^{\mathsf{T}} e^{\mathbf{A}^{\mathsf{T}}(t- au)} d au$$

#### When is $\mathbf{W}_c(t)$ invertible?

- $e^{\mathbf{A}t}$  may be expressed as a linear combination of  $\{\mathbb{I}, \mathbf{A}, \dots, \mathbf{A}^{n-1}\}$
- $e^{\mathbf{A}t}\mathbf{B}$  may be expressed as a linear combination of  $\{\mathbf{B}, \mathbf{AB}, \dots, \mathbf{A}^{n-1}\mathbf{B}\}$

#### Consider the effect of the vector v

$$\mathbf{v}^{\mathsf{T}}\mathbf{W}_{c}(t)\mathbf{v} = \int_{0}^{t} \overbrace{\mathbf{v}^{\mathsf{T}}e^{\mathbf{A}\tau}\mathbf{B}}^{\mathbf{0}} \mathbf{B}^{\mathsf{T}}e^{\mathbf{A}^{\mathsf{T}}\tau}\mathbf{v} d\tau$$

#### Key point:

$$\overbrace{v^{T}e^{A\tau}B}^{0} \Rightarrow \overbrace{v^{T} \begin{bmatrix} B & AB & \cdots & A^{n-1}B \end{bmatrix}}^{0}$$

The Gramian is invertible iff the controllability matrix has full rank!

## **Equivalent Statements on Controllability**

#### Controllability Gramian

$$\mathbf{W}_c(t) = \int_0^t e^{\mathbf{A}(t- au)} \mathbf{B} \mathbf{B}^\mathsf{T} e^{\mathbf{A}^\mathsf{T}(t- au)} d au$$

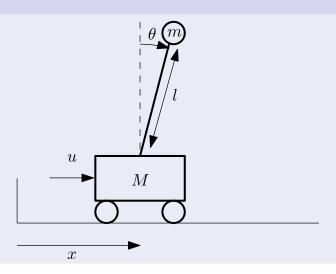
Iff  $\mathbf{W}_c(t)$  is invertible, the system is controllable.

#### Controllability Matrix

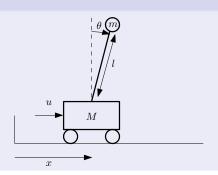
$$C = \overbrace{\begin{bmatrix} \mathbf{B} & \mathbf{AB} & \cdots & \mathbf{A}^{n-1}\mathbf{B} \end{bmatrix}}^{np} n$$

Iff the controllability matrix has full rank: rank(C) = n, the system is controllable.

## Example



#### Example



#### Linearized EOM

$$\begin{bmatrix} \dot{x} \\ \dot{\theta} \\ \ddot{x} \\ \ddot{\theta} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & -\frac{gm}{M} & 0 & 0 \\ 0 & \frac{g(m+M)}{M} & 0 & 0 \end{bmatrix} \begin{bmatrix} x \\ \theta \\ \dot{x} \\ \dot{\theta} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{M} \\ -\frac{1}{M} \end{bmatrix} u$$

#### Linearized EOM

$$\begin{bmatrix} \dot{x} \\ \dot{\theta} \\ \ddot{x} \\ \ddot{\theta} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & -\frac{gm}{M} & 0 & 0 \\ 0 & \frac{g(m+M)}{M} & 0 & 0 \end{bmatrix} \begin{bmatrix} x \\ \theta \\ \dot{x} \\ \dot{\theta} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{M} \\ -\frac{1}{M} \end{bmatrix} u$$

Linearized EOM: 
$$M = 2kg$$
,  $m = 1kg$ ,  $l = 1m$ ,  $g = 10\frac{m}{s^2}$ 

$$\begin{bmatrix} \dot{x} \\ \dot{\theta} \\ \ddot{x} \\ \ddot{\theta} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & -5 & 0 & 0 \\ 0 & 15 & 0 & 0 \end{bmatrix} \begin{bmatrix} x \\ \theta \\ \dot{x} \\ \dot{\theta} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{2} \\ -\frac{1}{2} \end{bmatrix} u$$

#### Controllability Matrix: Full row rank

#### Controllability Gramian: Invertible

$$\mathbf{W}_c(t) = \int_0^t e^{\mathbf{A}(t-\tau)} \mathbf{B} \mathbf{B}^\mathsf{T} e^{\mathbf{A}^\mathsf{T}(t-\tau)} d\tau$$

Linearized EOM: 
$$M = 2kg$$
,  $m = 1kg$ ,  $l = 1m$ ,  $g = 10\frac{m}{s^2}$ 

$$\begin{bmatrix} \dot{x} \\ \dot{\theta} \\ \ddot{x} \\ \ddot{\theta} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & -5 & 0 & 0 \\ 0 & 15 & 0 & 0 \end{bmatrix} \begin{bmatrix} x \\ \theta \\ \dot{x} \\ \dot{\theta} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{2} \\ -\frac{1}{2} \end{bmatrix} u$$

## Move the cart from $\mathbf{x}_0 = \mathbf{0}$ to $\mathbf{x}_1 = [1, 0, 0, 0]^T$ , $t_1 = 3s$

Let's use the Gramian:

$$\mathbf{u}(t) = -\mathbf{B}^\mathsf{T} e^{\mathbf{A}^\mathsf{T} (t_1 - t)} \mathbf{W}_c^{-1}(t_1) \left[ e^{\mathbf{A} t_1} \mathbf{x}_0 - \mathbf{x}_1 \right]$$

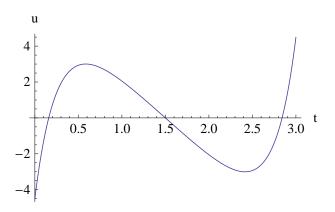
#### Controllability Gramian

$$\mathbf{W}_{c}(t) = \int_{0}^{t} e^{\mathbf{A}(t-\tau)} \mathbf{B} \mathbf{B}^{\mathsf{T}} e^{\mathbf{A}^{\mathsf{T}}(t-\tau)} d\tau$$

Move the cart from 
$$\mathbf{x}_0 = \mathbf{0}$$
 to  $\mathbf{x}_1 = [1, 0, 0, 0]^T$ ,  $t_1 = 3s$ 

Let's use the Gramian:

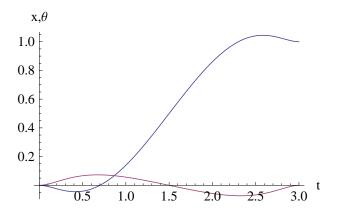
$$\mathbf{u}(t) = -\mathbf{B}^{\mathsf{T}} e^{\mathbf{A}^{\mathsf{T}}(t_1 - t)} \mathbf{W}_c^{-1}(t_1) \left[ e^{\mathbf{A}t_1} \mathbf{x}_0 - \mathbf{x}_1 \right]$$



Move the cart from 
$$\mathbf{x}_0 = \mathbf{0}$$
 to  $\mathbf{x}_1 = [1, 0, 0, 0]^T$ ,  $t_1 = 3s$ 

Let's use the Gramian:

$$\mathbf{u}(t) = -\mathbf{B}^{\mathsf{T}} e^{\mathbf{A}^{\mathsf{T}}(t_1 - t)} \mathbf{W}_c^{-1}(t_1) \left[ e^{\mathbf{A}t_1} \mathbf{x}_0 - \mathbf{x}_1 \right]$$



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#### Eigenvector test

#### Basic idea

If the Controllability matrix has full rank, there is no vector **v** such that:

$$\mathcal{C}^{T}\boldsymbol{v} = \left[ \begin{array}{c} \boldsymbol{B}^{T} \\ \boldsymbol{B}^{T}\boldsymbol{A}^{T} \\ \dots \\ \boldsymbol{B}^{T}(\boldsymbol{A}^{T})^{(n-1)} \end{array} \right] \boldsymbol{v} = \boldsymbol{0}, \quad \forall \boldsymbol{v} \neq \boldsymbol{0}$$

## Let **v** be an eigenvector of $\mathbf{A}^{\mathsf{T}}$ : $\mathbf{A}^{\mathsf{T}}\mathbf{q} = \lambda \mathbf{v}$

$$\begin{bmatrix} \mathbf{B}^T \\ \mathbf{B}^T \mathbf{A}^T \\ \dots \\ \mathbf{B}^T (\mathbf{A}^T)^{(n-1)} \end{bmatrix} \mathbf{q} = \begin{bmatrix} \mathbf{B}^T \\ \lambda \mathbf{B}^T \\ \vdots \\ \lambda^{n-1} \mathbf{B}^T \end{bmatrix} \mathbf{q}$$

#### Warning

Controllability is only possible if every eigenvector of  $\mathbf{A}^T$  is not in the null-space of  $\mathbf{B}^T$ .

## Eigenvector test

#### Warning

Controllability is only possible if every eigenvector of  $\mathbf{A}^T$  is not in the null-space of  $\mathbf{B}^T$ .

#### Popov-Belevitch-Hautus - test

The PBH test gives an elegant test based on this insight. An LTI system is controllable iff:

$$rank[\mathbf{A} - \lambda \mathbb{I} \ \mathbf{B}] = n, \quad \forall \lambda \in \mathbb{C}$$

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## Controllable systems

#### Three important points

- If the pair  $\{A, B\}$  is controllable, so is  $\{A BK, B\}$ .
- If the system is controllable we can place the eigenvalues of the system exactly as desired.
- A controllable system can always be transformed to the controllable canonical form.

## Controllability in practice

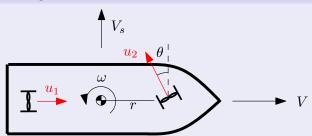
#### Caveats

Even if the controllability matrix has full rank, this does not mean that the system is easy to control in practice.

- The controller may require too large inputs.
- The closed loop response may be highly sensitive to modeling errors in  $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$ .
- The closed loop eigenvalues may have been chosen unrealistically fast.
- Fast response requires powerful actuators and an accurate model.
- The system may be "almost uncontrollable" in practice.

#### Controllable?

#### Dynamic positioning

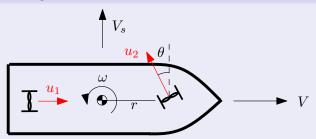


#### State equation

$$\left[ \begin{array}{c} \dot{V} \\ \dot{V}_{s} \\ \dot{\omega} \end{array} \right] = \left[ \begin{array}{ccc} -\frac{d}{m} & 0 & 0 \\ 0 & -\frac{d_{s}}{m} & 0 \\ 0 & 0 & -\frac{d_{\omega}}{J} \end{array} \right] \left[ \begin{array}{c} V \\ V_{s} \\ \omega \end{array} \right] + \left[ \begin{array}{ccc} 1/m & -\sin(\theta)/m \\ 0 & \cos(\theta)/m \\ 0 & \cos(\theta)r/J \end{array} \right] \left[ \begin{array}{c} u_{1} \\ u_{2} \end{array} \right]$$

#### Controllable?

#### Dynamic positioning



### Controllability matrix

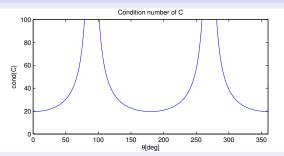
$$\mathcal{C} = \left[ \begin{array}{cccc} \frac{1}{m} & -\frac{\sin(\theta)}{m} & -\frac{d}{m^2} & \frac{d\sin(\theta)}{m^2} & \frac{d^2}{m^3} & -\frac{d^2\sin(\theta)}{m^3} \\ 0 & \frac{\cos(\theta)}{m} & 0 & -\frac{\cos(\theta)d_s}{m^2} & 0 & \frac{\cos(\theta)d_s^2}{m^3} \\ 0 & \frac{r\cos(\theta)}{J} & 0 & -\frac{r\cos(\theta)d_\omega}{J^2} & 0 & \frac{r\cos(\theta)d_\omega^2}{J^3} \end{array} \right]$$

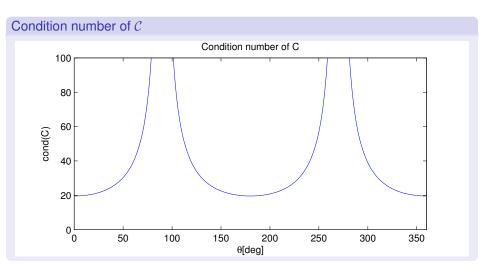
#### Controllable?

## Dynamic positioning



## Condition number of $\mathcal C$





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## Controllability Controllability indices

# What's wrong with this **B**?

$$\mathbf{B} = \left[ \begin{array}{ccc} 1 & 0 & 0 \\ 0 & 1 & 1 \\ 0 & 0 & 0 \\ 0 & 1 & 1 \end{array} \right]$$

#### **Answer**

Linear dependency in the columns. We disregard redundant inputs.

#### We have *p* inputs. Let:

#### Controllability matrix

#### Controllability index

The controllability indices of  $\mathbf{b}_{\rho}$ :  $\mu_{\rho}$ , are the number of linearly independent columns associated with  $\mathbf{b}_{\rho}$ . These indices sum to:

$$\mu_1 + \mu_2 + \cdots + \mu_p = n$$

The largest  $\mu_i$  is the controllability index.

#### Multi-input controllability

Using these indices, we can show that it is sufficient to check the rank of:

$$C = [ \mathbf{B} \mid \mathbf{AB} \mid \dots \mid \mathbf{A}^{n-p}\mathbf{B} ]$$

#### Final notes

## Property 1

Controllability is not affected by an equivalence transformation.

#### Property 2

Controllability is not affected by reordering the columns of B.