

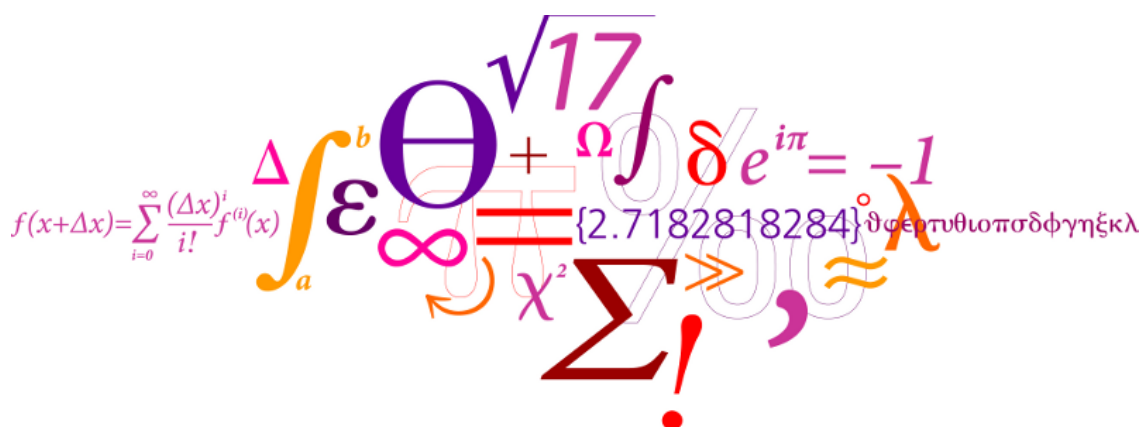
# 31310 LINEAR CONTROL DESIGN 2

COMPULSORY ASSIGNMENT 2015 : LOUDSPEAKER CONTROL

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## 0.1 Problem 1

First of all, the irradiance  $F$  of the light from the sun falling at the top of the atmosphere of Mars can be calculated as following : Conservation of energy :

$$4\pi R_{\odot}^2 F_{\odot} = 4\pi R^2 F \quad (1)$$

with

$R_{\odot} = 6,956.10^8 \text{ m}$  : solar radius

$F_{\odot} = 6,45.10^7 \text{ W.m}^{-2}$  : energy flow of the surface of the sun

$R \in [2.06644, 2.49228].10^{11} \text{ m}$  : distance Mars-Sun (Aphelion and Perihelion)

$$F = F_{\odot} \left( \frac{R_{\odot}}{R} \right)^2 \in [502, 730] \text{ W/m}^2 \quad (2)$$

In this report, we will consider that the rover is working on a specific date and we will chose the one when  $R$  corresponds to the semi-major axis. In this case  $R = 2,27936.10^{11} \text{ km}$  and

$$F = 589 \text{ W/m}^2 \quad (3)$$

Moreover, we can assume that a part of the irradiance is absorbed by the atmosphere. Knowing that the atmosphere of Earth absorbs and scatters to space around 30% of the incident irradiance of the Sun[3], and knowing that the atmosphere of Mars is thinner than the one of the Earth, we will postulate that 10% of the incident irradiance is absorbed. Thus, using 3 the actual irradiance  $F_a$  of the light from the sun falling on the surface of Mars is

$$F_a = \frac{90}{100} F = \frac{90 * 589}{100} = 530 \text{ W/m}^2 \quad (4)$$

However, this irradiance is the one of surface exposed perpendicular to the sun's beams. As Mars is a sphere, the projection need to be considered. Knowing that the weather is better into the northern hemisphere of Mars[2] and the fact that a latitude between 30 and 70 degrees is favored for a landing[1], we will assume that the rover has a latitude of  $50^\circ$ . This latitude corresponds to an angle of  $40^\circ$  between the surface of Mars and the sun's beams. Moreover, suppose that the rover stop working when this angle is inferior to  $10^\circ$ . Thus, the irradiance  $F_{50}$  at a latitude of  $50^\circ$  is

$$F_{50} = F_a \sin(\text{angleBeams}) \in [92, 341] \text{ W/m}^2 \quad (5)$$

with  $\text{angleBeams} = [10, 90 - \text{latitude}] = [10, 40]^\circ$ .

Then, considering the trajectory of the Sun into the sky of Mars and knowing that the rock target is more or less vertical to the surface of Mars, the angle  $\theta$  between the

target's normal and the sun's beam is considered to be included in  $[10, 50]^\circ$ . In addition, in the optimal case (when all the optimal conditions are provided to have the maximal radiance), the BRDF of the surface of the target is assumed to be 90% Lambertian and 10% Glossy while in the worst case the BRDF will be only Lambertian. In this way, the radiance of the target  $R_T$  is

$$R_T = \begin{cases} \frac{F_{50}\alpha}{\pi} \cos \theta & \text{optimal case} \\ F_{50}\alpha(\frac{9}{10\pi} \cos \theta + \frac{1}{10}) & \text{worst case} \end{cases} \quad (6)$$

with

$\alpha \in [0.05, 0.45]$ , the albedo of the target??

$\theta \in [10, 50]^\circ$ , the angle between the target's normal and the sun's beam

Thus,

$$R_T \in [92, 340] \text{ W/m}^2 \quad (7)$$

## References

- [1] John Leif Jørgensen, 2015.
- [2] Wikipedia. Mars — Wikipedia, the free encyclopedia, 2004. [Online; accessed November 18, 2015].
- [3] Giichi Yamamoto. Direct absorption of solar radiation by atmospheric water vapor, carbon dioxide and molecular oxygen. *Journal of the Atmospheric Sciences*, 19(2):182–188, 1962.