

$$R_{air} = 287.05 J/kg^\circ K \quad 1 atm = 1.013 \times 10^5 N/m^2 \quad \gamma_{air} = 1.4 \quad a = \sqrt{\gamma RT}$$

### Isentropic flow formulas

$$\frac{T_0}{T} = \left(1 + \frac{1}{5} M^2\right)$$

$$\frac{\rho_0}{\rho} = \left(\frac{T_0}{T}\right)^{C_v/R} = \left(1 + \frac{1}{5} M^2\right)^{5/2}$$

$$\frac{P_0}{P} = \left(\frac{T_0}{T}\right)^{C_p/R} = \left(1 + \frac{1}{5} M^2\right)^{7/2}$$

$$\frac{A}{A_*} = \frac{1}{M} \left(\frac{5 + M^2}{6}\right)^3$$

$$M = \sqrt{5 \left(\frac{P_0}{P}\right)^{2/7} - 5}$$

### Normal shock formulas

$$M_2^2 = \frac{5 + M_1^2}{7M_1^2 - 1} \geq \frac{1}{7}$$

$$\frac{\rho_2}{\rho_1} = \frac{6M_1^2}{5 + M_1^2}$$

$$\frac{P_2}{P_1} = \frac{7M_1^2 - 1}{6}$$

$$\frac{T_2}{T_1} = \frac{(7M_1^2 - 1)(5 + M_1^2)}{36M_1^2}$$

$$loss : \frac{P_{02}}{P_{01}} = \frac{P_2}{P_1} \left(\frac{T_1}{T_2}\right)^{7/2}$$

These are the general form of the equations

$$\frac{M^2(\gamma-1)+2}{2\gamma M^2-(\gamma-1)}$$

$$\frac{(\gamma+1)M^2}{(\gamma-1)M^2+2}$$

$$\frac{2\gamma M^2-(\gamma-1)}{\gamma+1}$$

$$\frac{P_2}{P_1} \frac{\rho_1}{\rho_2}$$

Pitot tube formula

$$\frac{P_{02}}{P_1} = \frac{6^6 M^7}{5^{7/2}(7M^2 - 1)^{5/2}}$$

### Duct flow

$$\dot{m} = P_0 A \sqrt{\frac{\gamma}{RT_0}} \frac{M}{(1 + \frac{1}{5} M^2)^3} \quad \left( = 0.6847 \frac{P_0 A_*}{\sqrt{RT_0}} \text{ when choked, } P_a/P_0 \leq 0.5283 \right)$$

### Oblique shock

$$M_{n1} = M_1 \sin \beta \quad M_2 = \frac{M_{n2}}{\sin(\beta - \theta)} \quad \tan \theta = \frac{2}{\tan \beta} \left[ \frac{M^2 \sin^2 \beta - 1}{M^2(1.4 + \cos 2\beta) + 2} \right]$$