



Suppose we get longitude  $d_1^{\circ} m_1' s_1''$  and latitude  $d_2^{\circ} m_2' s_2''$ , plus a NS and an EW bit.

If NS > 0, then  $\theta \in (0, \frac{\pi}{2})$

— otherwise,  $\theta \in (\frac{\pi}{2}, \pi)$

If EW > 0, then  $\phi \in (0, \pi)$

— otherwise,  $\phi \in (0, -\pi)$

Convert  $d_1^{\circ} m_1' s_1''$  to radians:  $\theta^{\circ} \rightarrow$  E/W of Greenwich  
Convert  $d_2^{\circ} m_2' s_2''$  to radians:  $\phi^{\circ} \rightarrow$  N/S of Equator

If NS > 0, then  $\theta = \frac{\pi}{2} - \theta^{\circ}$

— otherwise,  $\theta = \frac{\pi}{2} + \theta^{\circ}$

If EW > 0, then  $\phi = \phi^{\circ}$

— otherwise,  $\phi = 2\pi - \phi^{\circ}$

$$z = r \cos \theta$$

$$x = r \sin \theta \cos \phi$$

$$y = r \sin \theta \sin \phi$$

$$\theta = \cos^{-1} \frac{z}{\sqrt{x^2 + y^2 + z^2}}$$

$$\phi = \tan^{-1} \frac{y}{x}$$

Note that for the program, we set  $r = R + h$ , since the vehicle has an altitude component.