Exam in Automatic Control II Reglerteknik II 5hp (1RT495)

Date: October 21, 2022

Time: 14:00 – 19:00

Venue: Fyrislundsgatan 80, sal 1

Responsible teacher: Hans Rosth

Aiding material: Calculator, mathematical handbooks, textbooks by Glad & Ljung (Reglerteori/Control theory & Reglerteknik). Additional notes in the textbooks are allowed.

Preliminary grades: 23p for grade 3, 33p for grade 4, 43p for grade 5.

Use separate sheets for each problem, i.e. no more than one problem per sheet. Write your exam code on every sheet.

Important: Your solutions should be well motivated unless else is stated in the problem formulation! Vague or lacking motivations may lead to a reduced number of points.

Problem 6 is an alternative to the homework assignments from the autumn semester 2022. (In case you choose to hand in a solution to Problem 6 you will be accounted for the best performance of the homework assignments and Problem 6.)

Good luck!

Laplacetransformtabell

Definition: För en reellvärd funktion f(t), definierad för $t \geq 0$, ges Laplacetransformen av

$$\mathcal{L}[f(t)] = \int_{0}^{\infty} f(t)e^{-st}dt, \quad s \in \mathbb{C}$$

Det är en konvention att använda gemener¹ för tidsfunktioner, och versaler² för Laplace-transformer. T.ex. betecknar man Laplacetransformen av f(t) med F(s), d.v.s. $F(s) = \mathcal{L}[f(t)]$. I strik mening existerar bara Laplacetransformen för $s \in \mathbb{C}$ sådana att integralen i definitionen konvergerar. Om integralen konvergerar för $s = a \in \mathbb{C}$, så konvergerar den för alla s sådana att Re $s \geq \text{Re}\,a$. Den *inversa* Laplacetransformen ges då av

$$f(t) = \frac{1}{2\pi i} \int_{\alpha - i\infty}^{\alpha + i\infty} e^{st} F(s) ds \quad \text{for} \quad t \ge 0$$

där $\alpha \in \mathbb{R}$ och $\alpha \geq \text{Re}\alpha$.

Operationslexikon

Ŗ	Laplacetransform	Funktion i tidsplanet	
Н	F(s)	f(t)	
ଦ	F(s+a)	$e^{-at}f(t)$	Dämpningssatsen
60	$e^{-as}F(s)$	(t-a)	Förskjutningssatsen
4	$\frac{1}{a}F\left(\frac{s}{a}\right)$	$f(at) \qquad t-a < 0$	Täining
ıo	F(as)	$\frac{1}{2}f(\underline{t})$	Smiles
9	$\frac{d^n F(s)}{ds^n}$	$(-t)^n f(t)$	Derivering i s-nlanet
7	$\int_{\kappa}^{\infty}F(\sigma)d\sigma$	T(a)	Integration i s-planet
00	$\frac{1}{2\pi i}\int_{c-i\infty}^{c+i\infty}F_1(\sigma)F_2(s-\sigma)d\sigma$	$f_1(t)f_2(t)$	Faltning i s-planet
0	$F_1(s)F_2(s)$	$\int_0^t f_1(\tau) f_2(t-\tau) d\tau$	Faltning i tidsplanet
10	$sF(s) - f(0_{-})$	dr(t)	Derivering i tidsmlanet
11	$s^2F(s) - [sf(0) + \frac{d}{dt}f(0)]$	$\frac{d^2f(t)}{dt^2}$	omorden Guine
12	$s^n F(s) - \left[s^{n-1} f(0) + \dots + \frac{q^{n-1}}{dt^{n-1}} f(0) \right]$	d ⁿ f(t)	
13	$\frac{1}{s}F(s)$	$\int_{0_{-}}^{t} f(\tau) d\tau$	Integration i tidsplanet
14	$\lim_{s\to 0} sF(s)$	$\lim_{t\to\infty} f(t)$	Slutvärdesteoremet
15	$\lim_{s\to\infty} sF(s)$	$\lim_{t\to 0} f(t)$	

¹Små bokstäver. ²Stora bokstäver.

Transformlexikon

Ŗ	Laplacetransform	Funktion i tidsplanet	
\mathbf{H}	F(s)	f(t)	Notation
2		$\delta(t)$	Diracpuls
က	rii s	п	Konstant eller stegfunktion
4	-t-10	+2	Rampfunktion
S	r ₀ -0	$\frac{1}{2}t^{2}$	Acceleration
9	1 11 11 11 11 11 11 11 11 11 11 11 11 1	$\frac{1}{2}t^{T}$	
1	1 s+a	e-at	Exponentialfunktioner
00	$\frac{1}{(s+a)^2}$	te-at	
6	2(s+a)	$(1-at)e^{-at}$	
10	$\frac{1}{1+as}$	10-t	
11	$\frac{1}{s(s+a)}$	$\frac{1}{a}\left(1-e^{-at}\right)$	
12	$\frac{1}{s(1+as)}$	$1 - e^{-\frac{t}{a}}$	
13	S = 02	sinh at	Hyperboliska funktioner
14	\$ 82 42	cosh at	
12	\$2+a2	sin at	Trigonometriska funktioner
16	S2+G2	cosat	
17	$\frac{a}{2a+2(a+s)}$	$e^{-bt}\sin at$	Dämpade trig.funktioner
18	2+p 2+r 2+r)	e-bt cos at	

nämnarpolynom av gradtal högre än två kan man utnyttja att Laplacetransformen är linjär, genom att använda partialbråksuppdelning. Man kan sedan använda transformlexikonet och Om man vill bestämma den inversa Laplacetransformen för en rationell funktion med inverstransformera varje term för sig.

$$F(s) = \frac{2}{s(s+1)(s+2)} = \frac{1}{s} - \frac{2}{s+1} + \frac{1}{s+2} \iff f(t) = \mathcal{L}^{-1}\left[F(s)\right] = 1 - 2e^{-t} + e^{-2t}$$

Problem 1 The function of a hot air balloon is based on the fact that warm air has lower density than cold air. A simple model of the vertical motion of a hot air balloon, i.e. the change of altitude, is as follows¹:

i. The rate of the change in altitude, i.e. the vertical speed, denoted with r(t), is modelled as

$$\frac{dr}{dt} = \frac{1}{\tau_1}(w(t) - r(t)) + k\theta(t), \quad k > 0,$$

where w(t) is the vertical wind speed, and $\theta(t)$ is the temperature of the air in the balloon. Here $\tau_1 > 0$ is a time constant.

ii. The temperature is modelled as

$$\frac{d\theta}{dt} = -\frac{1}{\tau_2}\theta(t) + u(t),$$

where u(t) is the amount of heat produced by the burner. Again, $\tau_2 > 0$ is a time constant.

 We assume that the vertical wind speed is slowly varying and described by

$$w(t) = \frac{1}{n+\delta}v(t), \qquad \delta > 0,$$

where v(t) is zero mean white noise with intensity $\Phi_v(\omega) = R_v$.

iv. Finally, the altitude h(t) is governed by $\frac{dh}{dt} = r(t)$.

(a) Introduce the state vector $x(t) = \begin{bmatrix} \theta(t) & r(t) & h(t) & w(t) \end{bmatrix}^T$ and give the state space model in the form

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) + Nv(t), \\ z(t) = Mx(t), \end{cases}$$

i.e. determine the matrices and vectors A, B, N and M. We regard

$$z(t) = \begin{bmatrix} z_1(t) & z_2(t) \end{bmatrix}^T = \begin{bmatrix} r(t) & h(t) \end{bmatrix}^T$$

as the output/performance variable.

(5p)

(b) Assume that u(t) = 0. Show that $z_1(t) = r(t)$ is a stationary stochastic process, but that $z_2(t) = h(t)$ is not. (2p)

(c) Assume again that u(t) = 0. Determine the spectral density, $\Phi_r(\omega)$, of $z_1(t) = r(t)$.

¹All signals represent deviations from an operating point.

Problem 2 The block diagram below shows the implementation af a position servo, based on feedback control of a DC motor. The DC motor is described by the state space model (1).

Initially a continuous-time controller was used, with the control law

$$u(t) = K(r(t) - y(t)), \qquad K \in \mathbb{R}.$$
 (2)

(a) For which $K \in \mathbb{R}$ is the closed loop system stable when the continuous-time controller (2) is used? (1p)

Due to malfunction the original continuous-time controller had to be replaced, and a sampling controller was employed instead. The control law then changed to

$$u(t) = u(kh)$$
 for $kh \le t < kh + h$, $k \in \mathbb{Z}$,
and $u(kh) = K(r(kh) - y(kh))$. (3)

Here h is the sampling period, and $K \in \mathbb{R}$ is a pure gain.

(b) Determine the zero-order-hold sampled, discrete-time version of the state space model (1). (3p)

For a certain choice of sampling period the discrete-time representation in (b) will correspond to the following model of the system:

$$Y(z) = G(z)U(z), \qquad G(z) = \frac{(2\alpha - 1)z + 1 - \alpha - \alpha^2}{(z - 1)(z - \alpha)}, \qquad \alpha \approx 0.567$$
 (4)

- (c) For what values of $K \in \mathbb{R}$ is the closed loop system stable when the sampling controller (3) is used on (4)? (3p)
- (d) Assume that (3) is used on (4), and that K is chosen so that the closed loop system is stable. What is then the static gain from reference r to output y? (1p)

Problem 3 A system of radio telescopes is used for observations of an interstellar space probe on its way out of the solar system. The space probe is travelling along a straight line, on a radius starting in the sun, see the figure below. Let z(t) represent the deviation from the calculated trajectory of the space probe. This deviation is modeled with Newton's second law, and is here represented by the state space model (5).

$$\begin{cases} \dot{x}(t) = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} x(t) + \begin{bmatrix} 0 \\ 1 \end{bmatrix} v_1(t), \\ z(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} x(t), \\ y(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} x(t) + v_2(t), \end{cases}$$
(5)

The process noise, v_1 , is caused for example by fluctuations in the solar wind, and is considered as a zero mean white noise process with intensity $R_1 = 1$. The measurement noise, v_2 , is also considered as zero mean white noise with intensity $R_2 = 1$, and it is uncorrelated with v_1 .

Based on the observations the position (in terms of z(t)) is estimated with an observer:

$$\dot{\hat{x}}(t) = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \hat{x}(t) + \begin{bmatrix} k_1 \\ k_2 \end{bmatrix} (y(t) - \begin{bmatrix} 1 & 0 \end{bmatrix} \hat{x}(t))$$
 (6)

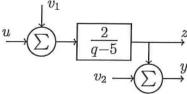
- (a) Give the state equation for the estimation error $\tilde{x}(t) = x(t) \hat{x}(t)$, based on (5) and (6). (2p)
- (b) Compute the covariance matrix $\Pi_{\tilde{x}} = E\tilde{x}\tilde{x}^T$ when the observer gain $K = \begin{bmatrix} k_1 & k_2 \end{bmatrix}^T = \begin{bmatrix} 1 & 1 \end{bmatrix}^T$ is used in (6). (3p)
- (c) What is the smallest value $\Pi_{\tilde{x}}$ can attain, that is, what is the covariance matrix of \tilde{x} for the optimal observer? (4p)

Problem 4 Specify for each of the following statements whether it is true or false. No motivations required — only answers "true"/"false" are considered!

- (a) $\Phi(\omega) = \frac{\omega^2}{\omega^4 4}$ is the spectrum of a continuous-time stochastic process.
- (b) For white noise the covariance function is $r(\tau) = 0$ for $\tau \neq 0$.
- (c) A Kalman filter is always stable.
- (d) The Nyquist frequency is 50% of the sampling frequency.
- (e) MPC controllers are always linear and time invariant.
- (f) In MPC the prediction horizon > the control horizon should hold.

Each correct answer scores +1, each incorrect answer scores -1, and omitted answers score 0 points. (Minimal total score is 0 points.) (6p)

Problem 5 Consider the following block diagram associated with a discrete-time system.



The system is affected by process and measurement noise denoted by v_1 and v_2 , respectively. The noises v_1 and v_2 are uncorrelated zero mean white noises with respective intensities $R_1 = 1$ and $R_2 = 2$.

We seek to control the system by the standard LQG controller

$$u(k) = -L\hat{x}(k|k-1),\tag{7}$$

that minimizes the cost function

$$V = E[z^2 + \rho u^2], \qquad \rho \ge 0.$$

(a) Determine the matrices that describe the state-space representation of the system depicted above in the following standard form

$$\begin{cases} x(k+1) = Fx(k) + Gu(k) + Nv_1(k), \\ z(k) = Mx(k), \\ y(k) = Hx(k) + v_2(k). \end{cases}$$
(8)

(2p)

For questions (b)-(d), use your results from (a). In case you have not determined the matrices in (a) you may (for simplicity) consider these to be F = G = N = M = H = 1.

- (b) Determine the control gain L in (7). (3p)
- (c) Determine the Kalman filter for obtaining the estimate $\hat{x}(k|k-1)$. (3p)
- (d) Determine the covariance $\Pi_{\hat{x}}$ associated with $\hat{x}(k|k-1)$ when (7) is used on (8). (3p)

Problem 6 The HW bonus points (from the autumn 2022) are exchangeable for this problem.

The block diagram below represents a stationary discrete-time stochastic process. The transfer operator G(q) is minimum phase (with $G(1) \ge 0$), and w is zero mean white noise.

$$\Phi_w(\omega) = 0.5$$
 $\frac{w}{q - 0.7}$
 $\frac{1.4}{q - 0.7}$
 Z
 $G(q)$
 $\Phi_y(\omega) = \frac{98}{1.64 - 1.6 \cos \omega}$

- (a) Determine the spectral density for z. (3p)
- (b) Determine the transfer operator G(q). (4p)

A short glossary of automatic control

ENGELSKA	
	į
SVENSKA	
	•

SVENSKA

closed loop system automatic control

slutet system, återkopplat system

bandbredd reglerteknik

mean lower limit linear limit cycle input gain error minimum phase loop gain impulse response expectation estimation error lag filter gain margin frequency estimate describing function ead filter expected value eigenvalue disturbance decoupled cross-over frequency covariance controller canonical form eedforward eedback damping ratio dampen controllable control variable control system control engineering complementary sensitivity function återkoppling framkoppling väntevärde, medelvärde kretsförstärkning linjar nedre gräns impulssvar amplitudmarginal störning, störsignal skärfrekvens (stabil) självsvängning fasavancerande länk fasretarderande länk insignal förstärkning frekvens väntevärde estimeringsfel, skattningsfel fel, reglerfel skattning beskrivande funktion relativ dämpning styrbar kanonisk form väntevärde egenvärde dämpa frikopplad styrsignal komplementär känslighetsfunktion kovarians styrbar reglersystem reglerteknik

multivariable

flervariabel

weighting function white noise state space form state space description state small gain theorem spectral density variance unstable transient response transfer function time delay static gain state space state feedback stable settling time sensitivity function time-invariant step response singular value singular value setpoint scalar root locus reference signal overshoot output open loop system rise time phase margin observer observable observer canonical form decompostion överföringsfunktion tidsfördröjning statisk förstärkning tillståndsform tillståndsbeskrivning stigtid översläng noliställe vitt brus viktfunktion varians instabil transientsvar tidsinvariant stegsvar tillståndsåterkoppling singulärvärdesuppdelning singulärt värde rotort tillstånd spektraltäthet lågförstärkningssatsen börvärde, referenssignal känslighetsfunktion referenssignal illståndsrum insvängningstid skalärt . observerbar kanonisk form öppet system observatör observerbar asmarginal utsignal

