## **Problem statement:**



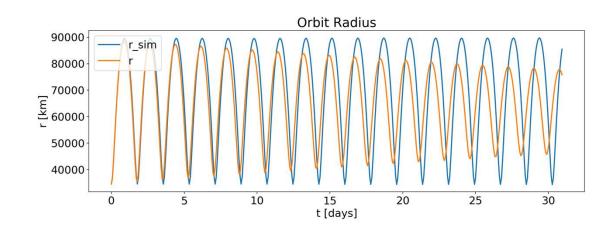
Monthly Number of Objects in Earth Orbit by Object Type -Total Objects -Fragmentation Debris -Mission-related Debris -Rocket Bodies 

Inaccurate prediction of the Earth satellite orbits can lead to dangerous collisions

Task: predict real coordinates from simulated values

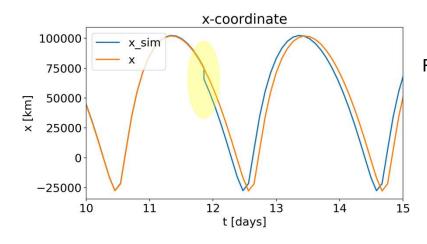
$$\begin{bmatrix} V_{x} - sim \\ V_{y} - sim \\ V_{z} - sim \\ x_{z} - sim \\ y_{z} - sim \\ y_{z} - sim \end{bmatrix} \rightarrow \begin{bmatrix} V_{x} \\ V_{y} \\ V_{z} \\ x \\ y \\ z \end{bmatrix}$$

Train data (January 2014):  $[r\_sim]$ ,  $[r\_real]$  600 satellites

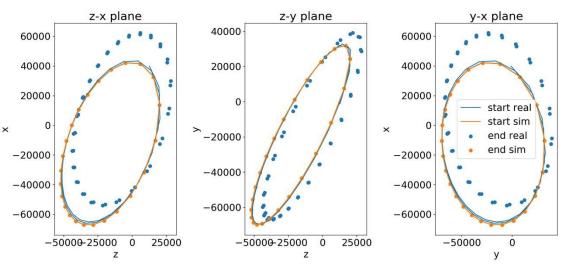


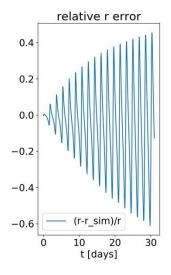
Images: isa.int, nasa.gov

## **Problem statement:**



Remove sudden jumps in simulated signal due to nonuniform time grid  $\downarrow$ Mapping signal to the nonuniform time grid using interpolation





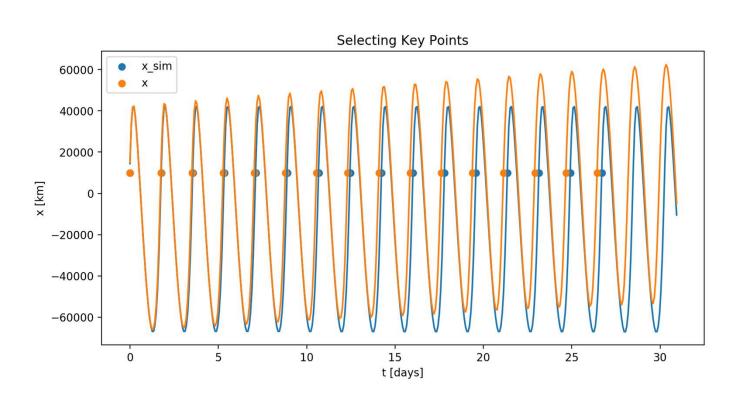
Address gradual dephasing and slow orbit precession

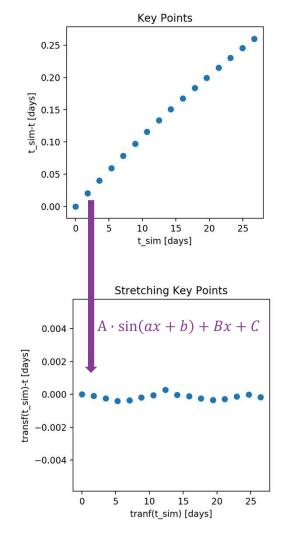
V

Nonlinear alignment

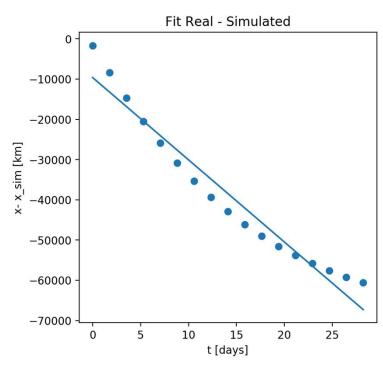
Challenge: finding uniform approach for all satellite

# Nonlinear alignment: matching key time points





### Nonlinear alignment: fit coordinate difference at key points



Linear fit  $X_{real} - X_{sim} = A \cdot t + B$ 

#### Full algorithm:

- I. Remove time jumps (shift signal using spline interpolation)
- II. Perform nonlinear alignment for each generalized coordinate X
  - 1. Identify families of key time points (by offset, 100 points per period)
  - 2. For each family of key points:
    - a) Transform simulation key points to real key points
    - b) Linearly stretch  $\Delta X$  at key points
  - 3. Combine stretched coordinates and transformed key points into sparse solution
- III. Map the sparse solution onto the initial time grid using 2<sup>nd</sup> order EOM:

$$a[i] = -\frac{G M_E}{r[i]^2}$$

$$v[i+1] = v[i] + a[i] \cdot dt$$

$$r[i+1] = r[i] + (v[i+1] + v[i])dt$$

# Nonlinear alignment: combining the result from all key points

