Input-State Linearization Two Examples

John Y. Hung

Visiting Professor

Department of Electrical Engineering
National Taiwan University of Science and Technology – "Taiwan Tech"

27 May 2010

1 / 1

Example 1: A 2nd Order Problem

Consider the plant

$$\dot{x}_1 = a \sin x_2 \tag{1}$$

$$\dot{x}_2 = -x_1^3 + u \tag{2}$$

or

$$\dot{x} = f(x) + g(x)u$$

where

$$f(x) = \begin{bmatrix} a \sin x_2 \\ -x_1^3 \end{bmatrix}, \qquad g(x) = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

2 / 1

The Problem

Find a transformation z = T(x) such that:

• the plant dynamics appear as:

$$\dot{z} = Az + B(\text{nonlinear function including } u)$$
 (3)

where (A, B) are in controllable canonic form.

 Then the input u can be designed to cancel undesirable terms of the nonlinear function.

Set Up the Problem

The problem is second order, so the transformation T(x) consists of two functions:

$$T(x) = \left[\begin{array}{c} T_1(x) \\ T_2(x) \end{array} \right]$$

Since (A, B) are in controllable canonic form, the transformation must satisfy these conditions:

$$T_2(x) = \mathcal{L}_f T_1(x)$$

= $\nabla_x T_1(x) f(x)$ (4)

$$\mathcal{L}_{g} T_{1}(x) = \nabla_{x} T_{1}(x) g(x)$$

$$= 0$$
(5)

$$\mathcal{L}_g T_2(x) = \mathcal{L}_g \mathcal{L}_f T_1(x)$$

$$\neq 0.$$
(6)

Study condition (5)

$$\mathcal{L}_{g} T_{1}(x) = 0$$

$$\nabla_{x} T_{1}(x) g(x) = 0$$

$$\begin{bmatrix} \frac{\partial T_{1}}{\partial x_{1}} & \frac{\partial T_{1}}{\partial x_{2}} \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix} = 0$$

$$\frac{\partial T_{1}}{\partial x_{2}} = 0$$
(7)

Therefore, the function $T_1(x)$ must be independent of the variable x_2 .

Study condition (6)

$$\mathcal{L}_{g}\mathcal{L}_{f}T_{1}(x) \neq 0$$

$$\mathcal{L}_{g}\left(\nabla_{x}T_{1}(x)f(x)\right) \neq 0$$

$$\mathcal{L}_{g}\left(\left[\begin{array}{cc} \frac{\partial T_{1}}{\partial x_{1}} & \frac{\partial T_{1}}{\partial x_{2}} \end{array}\right] \left[\begin{array}{c} a\sin x_{2} \\ -x_{1}^{3} \end{array}\right]\right) \neq 0$$

$$\mathcal{L}_{g}\left(\left[\begin{array}{cc} \frac{\partial T_{1}}{\partial x_{1}} a\sin x_{2} \end{array}\right] \neq 0$$

$$\nabla_{x}\left(\frac{\partial T_{1}}{\partial x_{1}} a\sin x_{2}\right) \left[\begin{array}{c} 0 \\ 1 \end{array}\right] \neq 0$$

$$\frac{\partial}{\partial x_{2}}\left(\frac{\partial T_{1}}{\partial x_{1}} a\sin x_{2}\right) \neq 0$$

$$\frac{\partial}{\partial x_{2}}\left(\frac{\partial T_{1}}{\partial x_{1}} a\sin x_{2}\right) \neq 0$$

$$\frac{\partial}{\partial x_{2}}\left(\frac{\partial T_{1}}{\partial x_{1}} a\cos x_{2} \neq 0$$
(8)

Summary of conditions for $T_1(x)$

From (7) and (8):

- **1** $T_1(x)$ must not be a function of x_2 .
- ② $T_1(x)$ should not be zero when $\cos x_2$ is nonzero.

An acceptable solution is $T_1(x) = x_1$.

Another solution is $T_1(x) = \sin x_1$. Solutions are not unique!

Finish the transformation T(x)

The function $T_2(x)$ is computed after choosing a function $T_1(x)$, using the condition (4):

$$T_{2}(x) = \mathcal{L}_{f} T_{1}(x)$$

$$= \begin{bmatrix} \frac{\partial T_{1}}{\partial x_{1}} & \frac{\partial T_{1}}{\partial x_{2}} \end{bmatrix} f(x)$$

$$= \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} a \sin x_{2} \\ -x_{1}^{3} \end{bmatrix}$$

$$= a \sin x_{2}$$

The complete transformation is given by:

$$z = T(x) = \begin{bmatrix} T_1(x) \\ T_2(x) \end{bmatrix} = \begin{bmatrix} x_1 \\ a\sin x_2 \end{bmatrix}. \tag{9}$$

Plant dynamics in new coordinate z

$$\dot{z}_1 = \dot{x}_1
= a \sin x_2
= z_2$$
(10)

$$\dot{z}_2 = a \cos x_2 \dot{x}_2
= a \cos x_2 (-x_1^3 + u)$$
(11)

Indeed,

$$\dot{z} = Az + B(\text{nonlinear function including } u)$$

where

$$A = \left[\begin{array}{cc} 0 & 1 \\ 0 & 0 \end{array} \right] \qquad B = \left[\begin{array}{c} 0 \\ 1 \end{array} \right]$$

9 / 1

The control u

From (11), the control u that linearizes the plant dynamics (in z coordinates) can be determined. Furthermore, a new input variable v can be introduced. The control is given by:

$$u = x_1^3 + \frac{1}{a\cos x_2}v\tag{12}$$

The input-state linearized dynamics (in z coordinates) are described by:

$$\dot{z} = \left[\begin{array}{cc} 0 & 1 \\ 0 & 0 \end{array} \right] z + \left[\begin{array}{c} 0 \\ 1 \end{array} \right] v$$

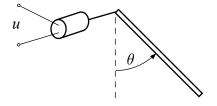
Linear system techniques can now be used to design a linear feedback law for v, so that the closed loop dynamics of are asymptotically stable!

Input-State Linearization Comments

- The math can be quite complicated.
- The linearization may not be global see the singularity in (12)
- The transformed coordinates z may not be physically meaningful, which means state feedback may not be feasible. A state estimator becomes necessary.

Example 2: A 3rd Order Problem

Robot arm driven by electric motor



Nomenclature

 θ : arm angle, u: motor input voltage

J: moment of inertia, B: friction coefficient

L: motor winding inductance, R: winding resistance

K: motor voltage and torque constant, g: gravity constant

Robot arm model

Plant dynamics are given by:

$$J\ddot{ heta} + B\dot{ heta} + MgL\sin{ heta} = Ki$$

$$L\frac{di}{dt} + Ri + K\dot{ heta} = u$$

State variable model

Let
$$x_1=\theta$$
, $x_2=\dot{\theta}$, $x_3=i$. Then

$$\begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= -\frac{MgL}{J} \sin x_1 - \frac{B}{J} x_2 + \frac{K}{J} x_3 \\ \dot{x}_3 &= -\frac{K}{L} x_2 - \frac{R}{L} x_3 + \frac{1}{L} u \end{aligned}$$

or (the simplified form):

$$\dot{x}_1 = x_2
\dot{x}_2 = -\sin x_1 - x_2 + x_3
\dot{x}_3 = -x_2 - x_3 + u$$
(13)

State variable model (continued)

In the form $\dot{x} = f(x) + g(x)u$

$$f(x) = \begin{bmatrix} x_2 \\ -\sin x_1 - x_2 + x_3 \\ -x_2 - x_3 \end{bmatrix}, \quad g(x) = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

The nonlinear coordinate transformation

$$z = T(x) = \begin{bmatrix} T_1(x) \\ T_2(x) \\ T_3(x) \end{bmatrix}$$

 $T_1(x)$ must satisfy:

$$\mathcal{L}_g T_1 = 0$$
 $\mathcal{L}_g \mathcal{L}_f T_1 = 0$
 $\mathcal{L}_g \mathcal{L}_f^2 T_1 \neq 0$

After finding T_1 , the remaining functions are given by:

$$T_2 = \mathcal{L}_f T_1, \qquad T_3 = \mathcal{L}_f T_2 = \mathcal{L}_f^2 T_1$$

Study conditions on $T_1(x)$

From the first condition on T_1 :

$$\mathcal{L}_g T_1 = 0 \quad \rightarrow \quad \frac{\partial T_1}{\partial x_3} = 0$$

Conclusion: T_1 must not be a function of x_3 .

Continue search for $T_1(x)$

From the second condition on T_1 :

$$\mathcal{L}_{g}\mathcal{L}_{f}T_{1} = 0 \quad \rightarrow \quad \mathcal{L}_{g}\left[\begin{array}{cc} \frac{\partial T_{1}}{\partial x_{1}} & \frac{\partial T_{1}}{\partial x_{2}} & 0 \end{array}\right]f(x) = 0$$

$$\rightarrow \quad \frac{\partial}{\partial x_{3}}\left(\frac{\partial T_{1}}{\partial x_{1}}x_{2} + \frac{\partial T_{1}}{\partial x_{2}}(-\sin x_{1} - x_{2} + x_{3})\right) = 0$$

Oh, this looks really tough!

Another way to look at the second condition:

$$\mathcal{L}_g T_2 = 0 \quad \rightarrow \quad \frac{\partial T_2}{\partial x_3} = 0$$

So, $T_2(x)$ must not be a function of x_3 .

Solving the conditions

Suggestion:

- Make some simplifying choices, and
- Check that all conditions are satisfied

For example, consider letting

$$\frac{\partial T_1}{\partial x_2} = 0.$$

Then the second condition simplifies to

$$\frac{\partial}{\partial x_3} \left(\frac{\partial T_1}{\partial x_1} x_2 \right) = 0$$

One solution is $T_1(x) = x_1$ (the arm angle).

Finishing the coordinate transformation T(x)

$$z_1 = T_1 = x_1$$

 $z_2 = T_2 = \mathcal{L}_f T_1 = x_2$
 $z_3 = T_3 = \mathcal{L}_f T_2 = -\sin x_1 - x_2 + x_3$

or

$$z = \begin{bmatrix} \text{arm position (angle)} \\ \text{angular rate} \\ \text{angular acceleration} \end{bmatrix}$$

Plant dynamics in new coordinate z

$$\dot{z}_1 = z_2$$
 $\dot{z}_2 = z_3$
 $\dot{z}_3 = -(\cos x_1)\dot{x}_1 - \dot{x}_2 + \dot{x}_3$

which has the desired form

$$\dot{z} = Az + B(\text{nonlinear function including } u)$$

where

$$A = \left[\begin{array}{ccc} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{array} \right] \qquad B = \left[\begin{array}{c} 0 \\ 0 \\ 1 \end{array} \right]$$

Try these exercises. . .

- **①** Complete the problem by finding the nonlinear u(x) that
 - cancels the nonlinear dynamics, and
 - ▶ introduces a new input variable v.
- The output of the original model is

$$y = x_1$$
 (arm angle).

Show that input-output linearization yields the same result for this application.

Comments

- Input-state linearizing solutions are generally not unique for a given problem. In other words, there are many transformations T(x) that are mathematically satisfactory. The challenge is to find a solution that meets engineering constraints.
- Input-output linearization may be easier to perform than input-state linearization, but internal dynamics (zero dynamics) must be checked.
- If the relative degree equals the system order, then input-output linearization is also achieves input-state linearization.

Gain Scheduling interpretation

- These techniques can be interpreted as "continuous" gain scheduling:
 - ▶ The linear feedback algorithm for *v* is "fixed."
 - ▶ The nonlinear control *u* represents a "scheduled" modification of *v*, performed smoothly over all values of the scheduling variable(s).
- One difference: Classical gain scheduling rules are based on auxiliary variables, not feedback variables.
 - ► Thought: Treat auxiliary variables as feedback variables just a matter of interpretation?
 - Then classical gain scheduling is a type of feedback linearization, with possibly discrete changes.