

Problems:

= With what velocity should a body be projected vertically upwards from the surface of the earth so that it may just attain a height of $R/2$. where R is the radius of the earth. ($R = 6400 \text{ km}$) .

Sol: Work done in moving the body from the surface of the earth to a height $R/2$ above the earth's surface

$$(R + R/2) = 3R/2$$

$$W = \int_R^{3R/2} \left(-\frac{GMm}{r^2} \right) dr$$

$$= GMm \left[-\frac{1}{r} \right]_R^{3R/2}$$

$$= -GMm \left[\frac{1}{3R/2} - \frac{1}{R} \right]$$

$$\therefore W = \frac{GMm}{3R}$$

$$K.E. = \frac{1}{2}mv^2$$

$$\text{or, } \frac{1}{2}mv^2 = \frac{GMm}{3R} \quad \text{--- (1)}$$

At the surface

of the earth

$$mg = \frac{GMm}{R^2} \quad \text{--- (2)}$$

Dividing (1) \div (2)

$$\frac{v^2}{2g} = \frac{R}{3}$$

$$v = \sqrt{\frac{2gR}{3}}$$

$$= \sqrt{\frac{2 \times 9.8 \times 6.4 \times 10^9}{3}}$$

$$= 6.467 \text{ km/s.}$$

The mass and satellite of Jupiter are $1.9 \times 10^{27} \text{ kg}$ and $7 \times 10^7 \text{ m}$ respectively. Calculate the escape velocity.

We know:

$$V_e = \sqrt{2gR}$$

$$= \sqrt{2 \times \frac{GM}{R^2} \times R}$$

$$= \sqrt{2 \frac{GM}{R}}$$

$$= \left[\frac{2 \times 6.67 \times 10^{-11} \times 1.9 \times 10^{27}}{7 \times 10^7} \right]^{\frac{1}{2}}$$

$$= 6.03 \times 10^4 \text{ m/s.}$$

Here,

$$G = 6.67 \times 10^{-11} \text{ Nm}^2/\text{kg}^2$$

M = Mass of Jupiter
 $= 1.9 \times 10^{27} \text{ kg}$

R = radius of the earth
 $= 7 \times 10^7 \text{ m.}$

In At what height above the surface of the Earth the value of acceleration due to gravity will be 1% of its value at the Earth's surface : [$R = 6.4 \times 10^3$ km]

We know,

$$g' = g \left(\frac{R}{R+h} \right)^2$$

$$\text{But } g' = \frac{1}{100} g$$

$$\therefore \frac{g}{100} = g \left(\frac{R}{R+h} \right)^2$$

$$\text{or, } \frac{1}{100} = \left(\frac{R}{R+h} \right)^2$$

$$\text{or, } \frac{1}{100} = \frac{R}{R+h}$$

$$\text{or, } R+h = 10R$$

$$\text{or, } h = 9R = 9 \times 6.4 \times 10^3 \text{ km} \\ = 5.76 \times 10^4 \text{ km.}$$

$$\text{Ans: } 5.76 \times 10^4 \text{ km}$$



Q An artificial satellite circles of the earth at a distance of 4000 km.

(a) Calculate its orbital velocity
and (b) Time Period of revolution.

[Radius of the earth = 6400 Km ; $g = 9.8 \text{ m/s}^2$]

Solⁿ:

$$\text{orbital velocity } v = \sqrt{\frac{gR}{(R+h)}}$$

Here, R = radius of earth.

$$= 6400 \text{ Km} = 64 \times 10^5 \text{ m}$$

h = distance = 4000 Km

$$= 40 \times 10^5 \text{ m.}$$

$$\therefore R+h = 104 \times 10^5 \text{ m}$$

$$\therefore v = \sqrt{\frac{9.8 \text{ m/s}^2 (64 \times 10^5 \text{ m})}{104 \times 10^5 \text{ m}}}$$

$$= \sqrt{\frac{9.8 \times 64 \times 64 \times 10^5 \times 10^5}{104 \times 10^5}}$$

$$= 6.212 \times 10^3 \text{ m/s}$$

$$\left[g = \frac{GM}{R^2}$$

and $\frac{v^2}{(R+h)} = g$

$$\left[a = \frac{v^2}{R} \right]$$
$$\left[g = \frac{v^2}{R+h} \right]$$

Time Period,

$$T = 2\pi \sqrt{\frac{(R+h)^3}{gR}}$$

$$= 2\pi \sqrt{\frac{(6.4 \times 10^6)^3}{9.8 \times (6.4 \times 10^6)}}$$

$$= 10512 \text{ s}$$

$$= 175.21 \text{ min.}$$

Ans: (a) 6.212 km s^{-1} (b) 175.21 min.

To Prove that the least velocity with which a particle must be projected from the surface of a planet of radius R and density ρ in order that it may escape completely is

$$\sqrt{\frac{8\pi G \rho R v}{3}}$$

To calculate the limiting velocity required by an artificial satellite for orbiting round the Earth. Given that the radius of earth = 6.4×10^6 m and $g = 9.8 \text{ ms}^{-2}$.

$$\text{Here, } v = \sqrt{gR} \quad \left[g = \frac{v^2}{R} \right]$$

$$g = 9.8 \text{ ms}^{-2}, R = 6.4 \times 10^6 \text{ m.}$$

$$v = \sqrt{9.8 \times 6.4 \times 10^6}$$

$$v = 7.92 \times 10^3 \text{ m/s}$$

$$v = 7.92 \text{ km/s.}$$