# MAE5403 (Linear System Control Design Problem)

## Unmanned Helicopter Controller Design

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#### 1. Abstract

In this Unmanned Helicopter case. We consider 9 state space variable. There are three angle(Roll,Pitch,Yaw) angular rates and three unmeasured variable  $(a_s, b_s, \delta_{ped,int})$ . This system also has three input and three dsiturbance  $\delta_{lat}$ ,  $\delta_{lon}$ ,  $\delta_{ped}$  and  $u_{wind}$ ,  $u_{wind}$ ,  $v_{wind}$ , respectively. I choose LQG technique in this case for Inner-loop Flight control. I also try  $H_{\infty}$ , but I have some problem about reduced order Controller design, I can not caculate the right nums. So I decide to use LQG for this case. And I use simulink to simulation my controller, It have good performance that I will show in Part2(Results analysis).

### 2. Part1: Mathmactics and Fligh dynamics model analysis

(a) Here, We Let

$$\dot{x} = Ax + Bu + Ew$$
$$y = C_1 x + D_1 v$$

And the noise v is the measurement output noise, In this case i choose the white noise for output

measurement noise. The state variable is 
$$x = \begin{cases} \rho \\ p \\ q \\ a_s \\ b_s \\ r \\ \delta_{ped,int} \\ \psi \end{cases}$$
;  $u = \begin{cases} \delta_{lat} \\ \delta_{lon} \\ \delta_{ped} \end{cases}$ ;  $w = \begin{cases} u_{wind} \\ v_{wind} \\ v_{wind} \end{cases}$ ;

(b) And from dynamics linearization model, we also obtain the A,B and E martix, They showed the fellowing.

(c) Choose the LQG technique for this. Before we start the design we need to know the pair (A,B) and (A,C1) is controllable and observable. It's obvious full rank with the matlab command rank(ctrb(A,B)) and rank(obsv(A,C1)). So we can do the job now. we need to three steps, they show as fellowing: **First:**, We need to design a LQR control law u = -Fx, and we need to choose the Matirx Q and  $R(Q \ge 0, R \ge 0)$ . Here In my simulation I choose the Q and R matrix as fellow.

Then, use Matlab function icare to solve the Riccati Equation  $PA + A^TP - PBR^{-1}B^TP + Q = 0$ ,

 $P \geq 0, F = R^{-1}B^TP$  The matlab function is

Then, We obtain the F matrix as fellow:

$$F = 1.0e + 04* \begin{bmatrix} 0.0943 & 0.0035 & 0.0652 & 0.0039 & 0.1251 & 1.5944 & 0.0356 & -0.0087 & 0.0332 \\ -0.0046 & 0.0999 & -0.0025 & 0.0724 & 1.0974 & -0.0017 & -0.0012 & 0.0003 & 0.0028 \\ 0.0331 & 0.0042 & 0.0354 & 0.0004 & -0.0045 & -0.0391 & -0.0721 & -0.1079 & -0.0943 \end{bmatrix};$$

And the matrix  $\underline{A} - BF$  is show:

$$Abk = 1.0e + 04* \\ \begin{bmatrix} 0 & 0 & 0.0001 & 0 & 0 & 0.0000 & 0 & 0 \\ 0 & 0 & 0 & 0.0001 & 0 & 0 & -0.0000 & 0 & 0 \\ -1.4344 & -0.1806 & -1.5341 & -0.0177 & 0.1934 & 1.7547 & 3.1298 & 4.6733 & 4.08 \\ 0 & 0 & 0 & -0.0000 & 0.0268 & -0.0000 & 0 & 0 & 0 \\ -0.0071 & -0.2591 & -0.0067 & -0.1883 & -2.8654 & -0.3184 & -0.0041 & 0.0010 & -0.0 \\ -0.2442 & -0.0023 & -0.1689 & -0.0053 & -0.2506 & -4.1263 & -0.0922 & 0.0225 & -0.0 \\ 2.7517 & 0.3465 & 2.9430 & 0.0339 & -0.3711 & -3.2539 & -6.0041 & -8.9653 & -7.8 \\ 0.1273 & 0.0160 & 0.1362 & 0.0016 & -0.0172 & -0.1506 & -0.2779 & -0.4154 & -0.3 \\ 0 & 0 & 0 & 0.0000 & 0 & 0 & 0.0001 & 0 & 0 \end{bmatrix}$$

And the eigenues is show

And the eigvalues is show 
$$\begin{bmatrix} -7.9558 + 0.0000i \\ -4.1816 + 0.0000i \\ -2.8032 + 0.0000i \\ -0.0023 + 0.0000i \\ -0.0016 + 0.0000i \\ -0.0005 + 0.0000i \\ -0.0001 + 0.0000i \\ -0.0001 - 0.0000i \\ -0.0001 + 0.0000i \\ -0.0001 + 0.0000i \\ -0.0001 + 0.0000i \\ \end{bmatrix}; \text{ We can see all eigvalue have negative real part.}$$

**Second:**, We need to use Kalman filter for the plant,  $\dot{\hat{x}} = A\hat{x} + K_e(y - \hat{y}), \hat{y} = C_1\hat{x}$ We also use icare to sove Riccati Equtaion:  $P_eA^T + AP_e - P_eC^TR_e^{-1}CP_e + Q_e = 0, P_e \ge 0, Ke = P_eC^TR_e^{-1}$ 

And here we choose the  $Q_e$  and  $R_e$  matrix is:

0.004

Using Matlab code as fellow:

And we obtain Ke and  $Ack = A - K_eC_1$  and Ack eigenvalues: Then, We obtain the F matrix as fellow:

$$K_e = \begin{bmatrix} 0.2500 & 0.0000 & 0.0014 & -0.0023 & 0 & 0 & 0.0007 & 0 & 0.0000 \\ 0.0000 & 0.2500 & -0.0009 & 0.0031 & 0 & 0 & -0.0005 & 0 & -0.0000 \\ 0.0014 & -0.0009 & 0.0001 & -0.0000 & 0 & 0.0000 & 0 & -0.0000 \\ -0.0023 & 0.0031 & -0.0000 & 0.0001 & 0 & 0 & -0.0000 & 0 & 0.0001 \\ 0.0002 & 0.2496 & -0.0012 & 0.0033 & 0 & 0 & -0.0005 & 0 & 0.0097 \\ 0.2500 & 0.0000 & 0.0016 & -0.0021 & 0 & 0 & 0.0011 & 0 & 0.0000 \\ 0.0007 & -0.0005 & 0.0000 & -0.0000 & 0 & 0.0000 & 0 & -0.0000 \\ 0.0002 & -0.0097 & -0.0001 & -0.0000 & 0 & 0 & -0.0000 & 0 & 0.2498 \\ 0.0000 & -0.0000 & -0.0000 & 0.0001 & 0 & 0 & -0.0000 & 0 & 0.2500 \end{bmatrix}$$

And the matrix  $A - K_eC_1$  is show:

	-0.2500	-0.0000	0.9986	0.0023	0	0	0.0002	0	_
Ack = 1.0e + 04*	-0.0000	-0.2500	0.0009	0.9961	0	0	-0.0384	0	C
	-0.0014	0.0009	-0.0303	-0.0056	-0.0003	585.1165	11.4448	-59.5290	C
	0.0023	-0.0031	0.0000	-0.0708	267.7499	-0.0003	0.0000	0	_
	-0.0002	-0.2496	0.0012	-1.0033	-3.3607	2.2223	0.0005	0	_
	-0.2500	-0.0000	-1.0016	0.0021	2.4483	-3.3607	-0.0011	0	_
	-0.0007	0.0005	0.0579	0.0108	0.0049	0.0037	-21.9557	114.2000	C
	-0.0002	0.0097	0.0001	0.0000	0	0	-1.0000	0	_
	[-0.0000]	0.0000	0.0000	0.0388	0	0	0.9992	0	_

And the eigenutes is show

And the eigvalues is show 
$$\begin{bmatrix} -1.5367 + 23.9237i \\ -1.5367 - 23.9237i \\ -1.6190 + 16.4397i \\ -13.8925 + 0.0000i \\ -7.7978 + 0.0000i \\ -0.5274 + 0.0000i \\ -0.4993 + 0.0000i \\ -0.4999 + 0.0000i \end{bmatrix}; \text{ We can also see all eigvalue have negative real part.}$$

The last:, The plant LQG control law is  $u = -\hat{F(x)}$ 

$$\begin{cases} \dot{\hat{x}} = (A - BF - K_eC_1) + K_ey, \\ u = -F\hat{x} \end{cases}$$
 And,  $G = [C_1(A - BF)^{-1}]^{-1}$ , Using the Matlab we obtain the G:

$$G = \begin{bmatrix} -942.5693 & -34.5139 & -332.2226 \\ 46.3757 & -998.5361 & -27.8396 \\ -330.7754 & -41.6478 & 942.7901 \end{bmatrix};$$

Clearly, the closed-loop system is characterized by the following state space equation we can 
$$\begin{cases} \left( \dot{x} \right) = \begin{bmatrix} A - BF & BF \\ 0 & A - K_eC \end{bmatrix} \begin{pmatrix} x \\ e \end{pmatrix} - \begin{pmatrix} BG \\ 0 \end{pmatrix} r + \tilde{v}, \tilde{v} = \begin{pmatrix} v \\ v - K_eW \end{pmatrix} \end{cases}$$
 Then, We let 
$$y = \begin{bmatrix} C & 0 \end{bmatrix} \begin{pmatrix} x \\ e \end{pmatrix} + w$$
 
$$A_c = \begin{bmatrix} A - BF & BF \\ 0 & A - K_eC \end{bmatrix}, B_c = \begin{pmatrix} BG \\ 0 \end{pmatrix}, C_c = \begin{bmatrix} C & 0 \end{bmatrix},$$
 Cacculating From Matlab We Obatin the

$$A_c = \begin{bmatrix} A - BF & BF \\ 0 & A - K_eC \end{bmatrix}, B_c = \begin{pmatrix} BG \\ 0 \end{pmatrix}, C_c = \begin{bmatrix} C & 0 \end{bmatrix}$$
, Caculating From Matlab We Obatin the

$$A_c = 1.0e + 04 *$$

0	0		$-0.0000 \\ -0.1883$	-2.8654 $-0.2506$ $-0.3711$	$     \begin{array}{r}       -0 \\       -0 \\       -4 \\       -3     \end{array} $	.1263 $.2539$	-0.0922 $-6.0041$	$\begin{array}{c} 4.6733 \\ 0 \\ 0.0010 \\ 0.0225 \\ -8.9653 \end{array}$		0 39 0.0071 558 0.2442 529 -2.751 630 -0.127 0 -0.000 -0.000 0.0000 -0.000 -0.000 -0.000 -0.000 -0.000		$\begin{array}{c} -0.1362 \\ 0 \\ 0.0001 \\ 0.0000 \\ -0.0000 \\ 0.0000 \\ 0.0000 \end{array}$	
							0	0		0			
							0	0		0			
						-1.	4344	-0.180	06	4.0883			
							0	0		0			
							.0071	-0.259		-0.0139			
						-0.	2442	-0.002	23 -	-0.0858			
						2.7	7517	0.346	5 -	-7.8429			
						0.1	273	0.016	0 -	-0.3630			
The $B_c$ is equal: $B_c = 1.0e + 04$			*		0	0		0	;				
							0	0		0	'		
							0	0		0			
							0	0		0			
							0	0		0			
							0	0		0			
							0	0		0			
							0	0		0			
							0	0		0			
						_	0	0		0			

 $\begin{array}{c}
0 \\
-0.1934 \\
0 \\
2.8651 \\
0.2509 \\
0.3711 \\
0.172
\end{array}$ 

0.0172

-0.0003 0.0002 0.0000

 $\begin{array}{r}
-1.6961 \\
0 \\
0.3186 \\
4.1260 \\
3.2539
\end{array}$ 

0.1506

 $0.0002 \\ -0.0003 \\ 0.0000$ 

-0.0000 -0.0000

- (d) Summary Here, We use LQG caculating all required martix, those matrix will all use in the Part2.
- 3. **Part2:** Results First,I will give the simulink block diagram show figure 1,It has 6 main block,the detail can be found in figure 1:

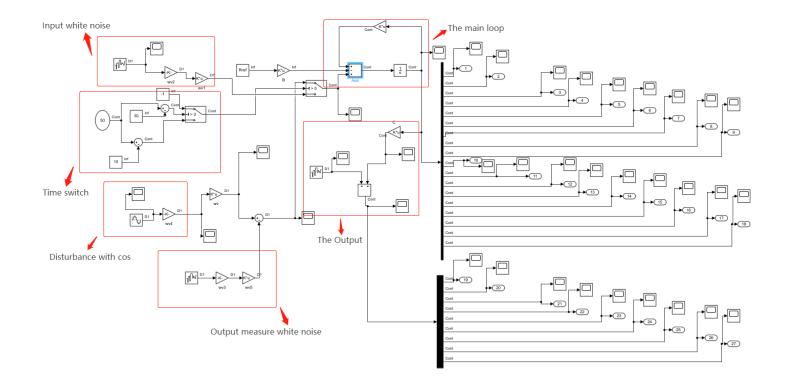


Figure 1: The Unmanned Helicopter LQG Controller diagram

Then,I will show the state and error ,and y output from figure 2 and figure 3:

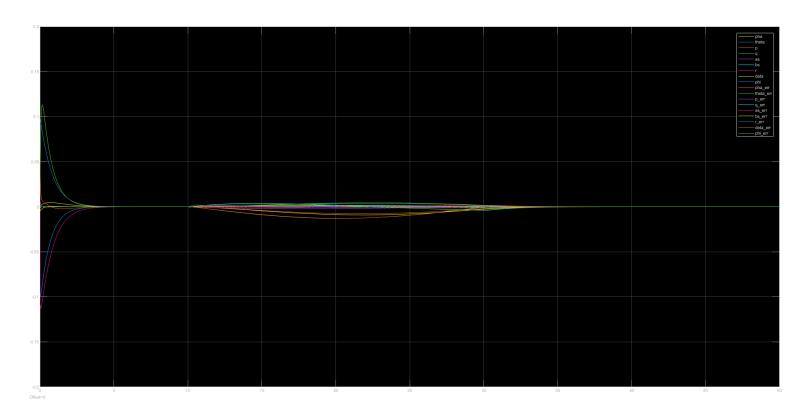


Figure 2: The Plant state  $\mathbf{x}$  and error

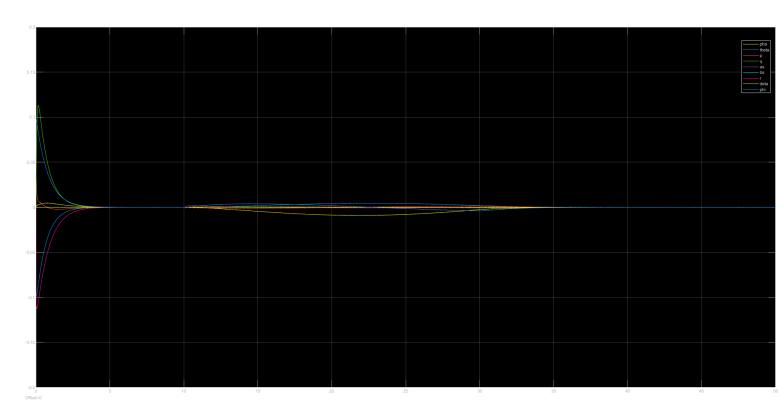


Figure 3: The Plant y output

At last,I will show all state and output respectively.

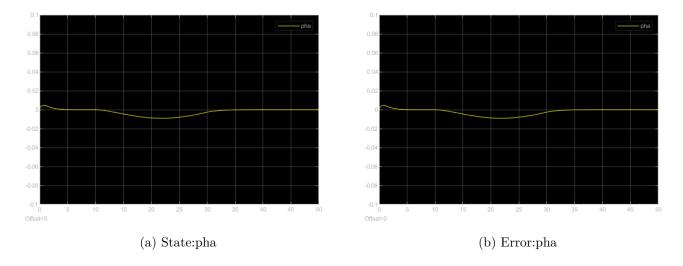


Figure 4: Pha state and error

## References

[1] G. O. Young, "Synthetic structure of industrial plastics," in *Plastics*, 2<sup>nd</sup> ed., vol. 3, J. Peters, Ed. New York, NY, USA: McGraw-Hill, 1964, pp. 15–64.