

The sum of the aircraft motion relative to the air ( $\mathbf{V}$ ) and the motion of the air relative to the ground ( $\mathbf{w}$ ) must give the motion relative to the ground ( $\mathbf{s}$ ). The equations below then constrain the five unknowns  $\{a, b, \alpha, \gamma, \tau\}$  where  $a$  and  $b$  are the respective magnitudes of  $\mathbf{V}$  and  $\mathbf{w}$ ,  $\alpha$  is the angle of offset between the flight direction and the direction of motion relative to the ground or the orientation of  $\mathbf{s}$ ,  $\gamma$  is the angle between the heading (or orientation of  $\mathbf{V}$ ) and the wind direction, and  $\tau$  is the flight time to cover the distance  $|\mathbf{s}|$ . The known quantities are  $s$ ,  $\beta$ = angle between the ground track and the wind direction, the wind speed  $w_s$ , and the flight speed  $V_t$ . The last two equations come from the law of sines:

$$\alpha + \beta + \gamma = \pi \quad (1)$$

$$a = V_t \tau \quad (2)$$

$$b = w_s \tau \quad (3)$$

$$\frac{\sin \beta}{a} = \frac{\sin \alpha}{a} \quad (4)$$

$$\frac{\sin \gamma}{s} = \frac{\sin \beta}{a} \quad (5)$$

From (2), (3) and (4),

$$\sin \alpha = \frac{b}{a} \sin \beta = \frac{w_s}{V_t} \sin \beta$$

which gives  $\alpha$  in terms of the known quantities  $w_s, V_t, \beta$ :

$$\alpha = \arcsin \left( \frac{w_s}{V_t} \sin \beta \right) .$$

Equation (1) then gives the angle  $\gamma$ . Equation (5) then gives

$$a = s \frac{\sin \beta}{\sin \gamma}$$

and so defines the time required for the flight segment:

$$\tau = \frac{a}{V_t}$$