



Sandia
National
Laboratories

Exceptional service in the national interest

ME469: Common Discretization Approaches

Stefan P. Domino^{1,2}

¹ Computational Thermal and Fluid Mechanics, Sandia National Laboratories

² Institute for Computational and Mathematical Engineering, Stanford

This presentation has been authored by an employee of National Technology & Engineering Solutions of Sandia, LLC under Contract No. DE-NA0003525 with the U.S. Department of Energy (DOE). The employee owns all right, title and interest in and to the presentation and is solely responsible for its contents. The United States Government retains and the publisher, by accepting the article for publication, acknowledges that the United States Government retains a non-exclusive, paid-up, irrevocable, world-wide license to publish or reproduce the published form of this article or allow others to do so, for United States Government purposes. The DOE will provide public access to these results of federally sponsored research in accordance with the DOE Public Access Plan.

SAND2018-4536 PE





Lecture Objectives

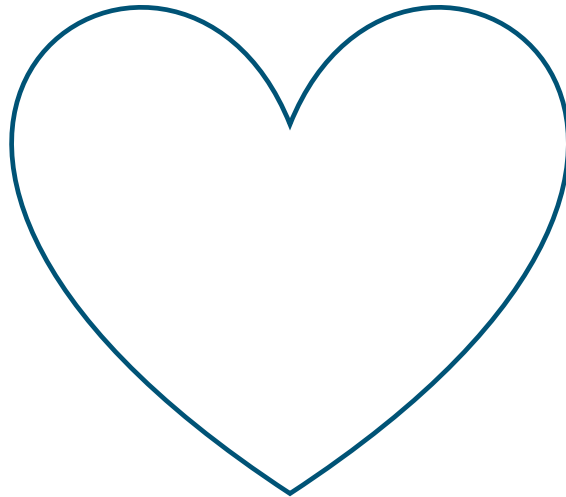
- The Concept of Meshing
- Why Unstructured?
- Unstructured Element Types
- Cell-centered Finite Volume (FV)
- Edge-based Vertex-Centered (EBVC)
- Control-Volume Finite Element Method (CVFEM)
- Finite Element Method (FEM)
- Staggered arrangement



Introducing a Mesh over Heart Domain, Ω

Geometry is:

- Complex
- Curved
- Sharp

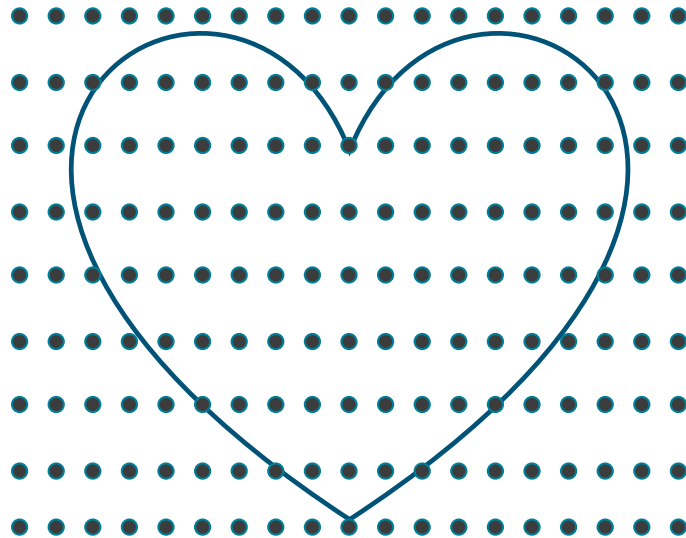




Re-introducing a [Finite Difference] Mesh over Heart Domain, Ω

Geometry is:

- Complex
- Curved
- Sharp

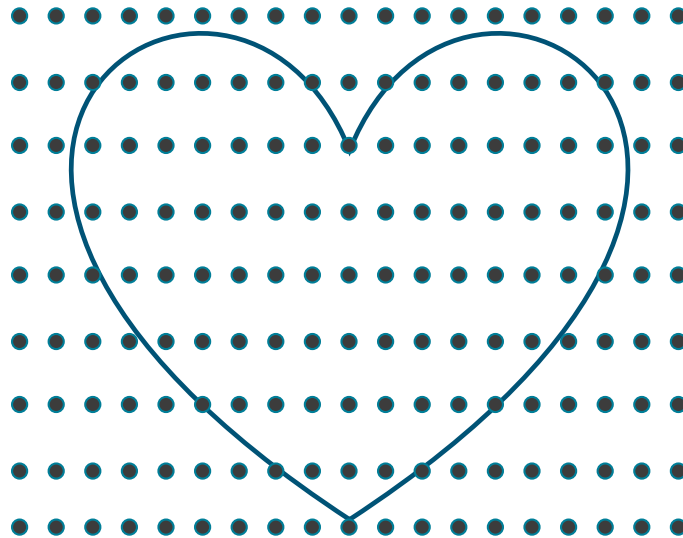




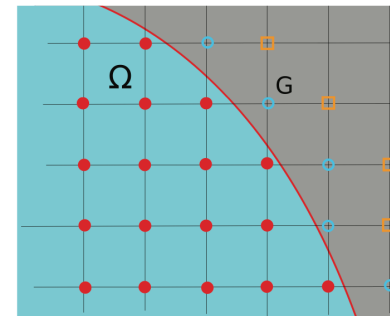
Re-introducing a [Finite Difference] Mesh over Heart Domain, Ω

Geometry is:

- Complex
- Curved
- Sharp



Not impossible: Chertock, et al., "A Second-Order Finite-Difference Method for Compressible Fluids in Domains with Moving Boundaries", Commun. Comput. Phys., 2018

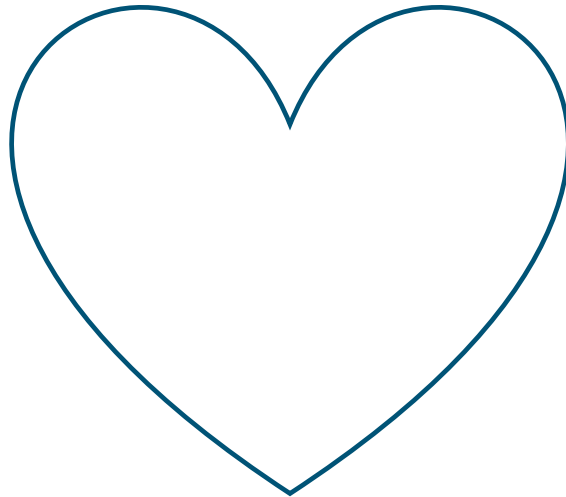




Introducing a [Structured Mesh] over Heart Domain, Ω ;

Geometry is:

- Complex
- Curved
- Sharp

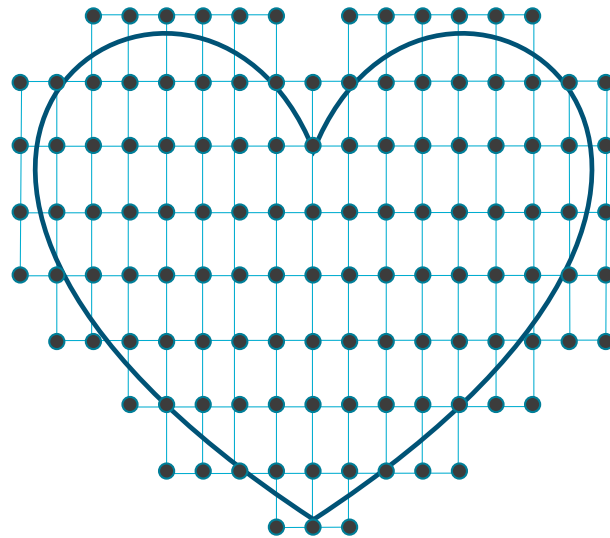




Introducing a [Structured Mesh] over Heart Domain, Ω ;

Geometry is:

- Complex
- Curved
- Sharp

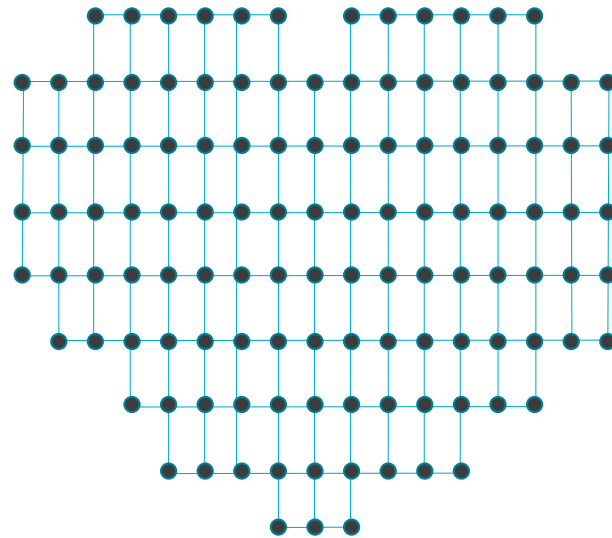




Introducing a [Structured Mesh] over Heart Domain, Ω ;

Geometry is:

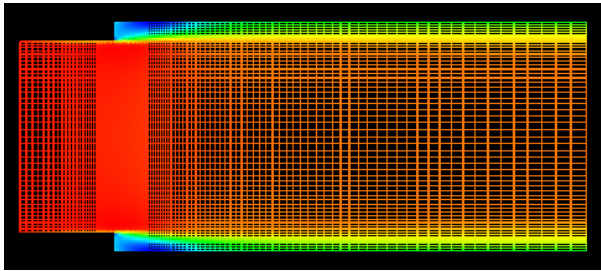
- Complex
- Curved
- Sharp



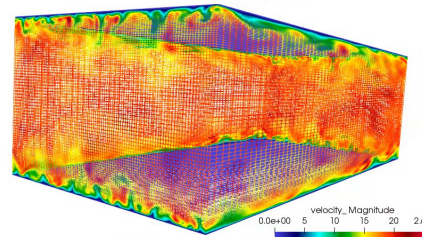


Structured vs Unstructured

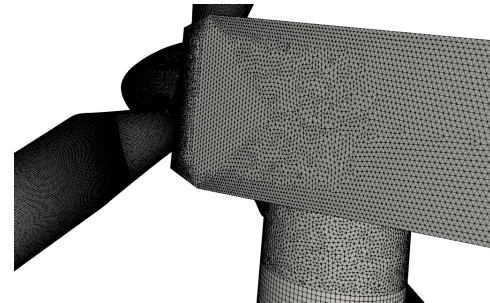
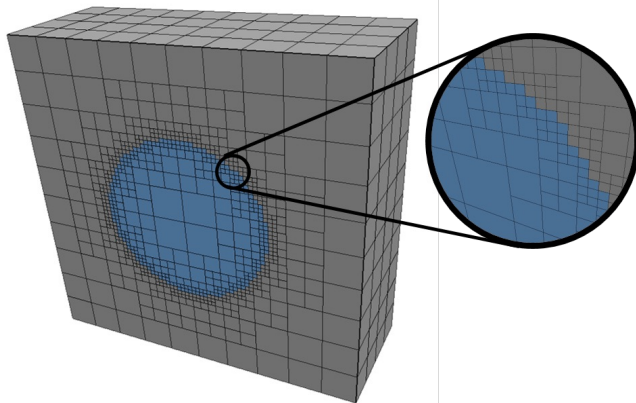
- Many times, canonical flows of interest are represented by simplified geometries that allow for cartesian meshes – with “stair-stepping”



RANS-based backward facing step (Domino, 2012)



Re = 395 plane-channel (Jofre, Domino, Iaccarino, 2018)

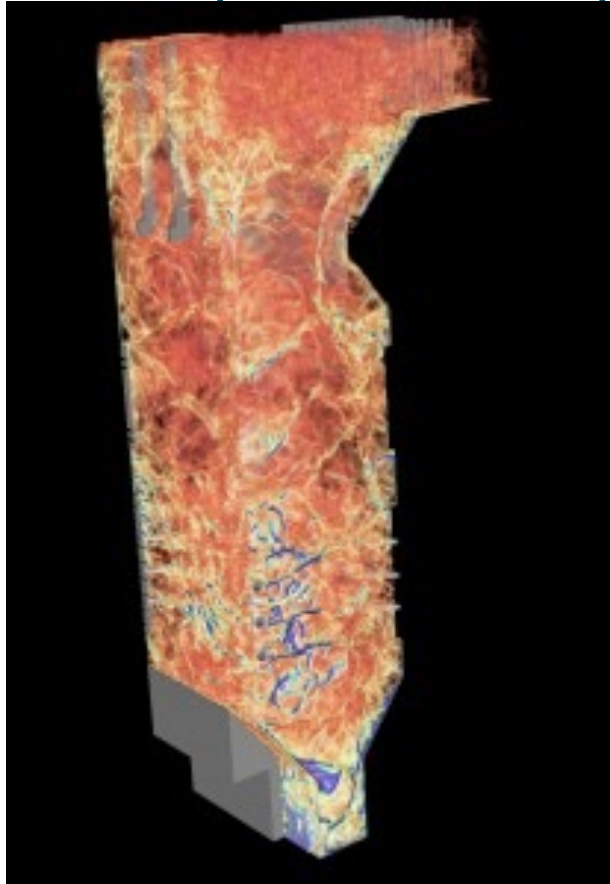


Often times, not!

<https://www.itscainternational.com/software/introduction-to-meshing>



Example: The Carbon-Capture Multidisciplinary Simulation Center



15MW coal-fired boiler volume rendered
image of large (90 μm) particles

Staggered schemes have been
demonstrated to support complex
applications

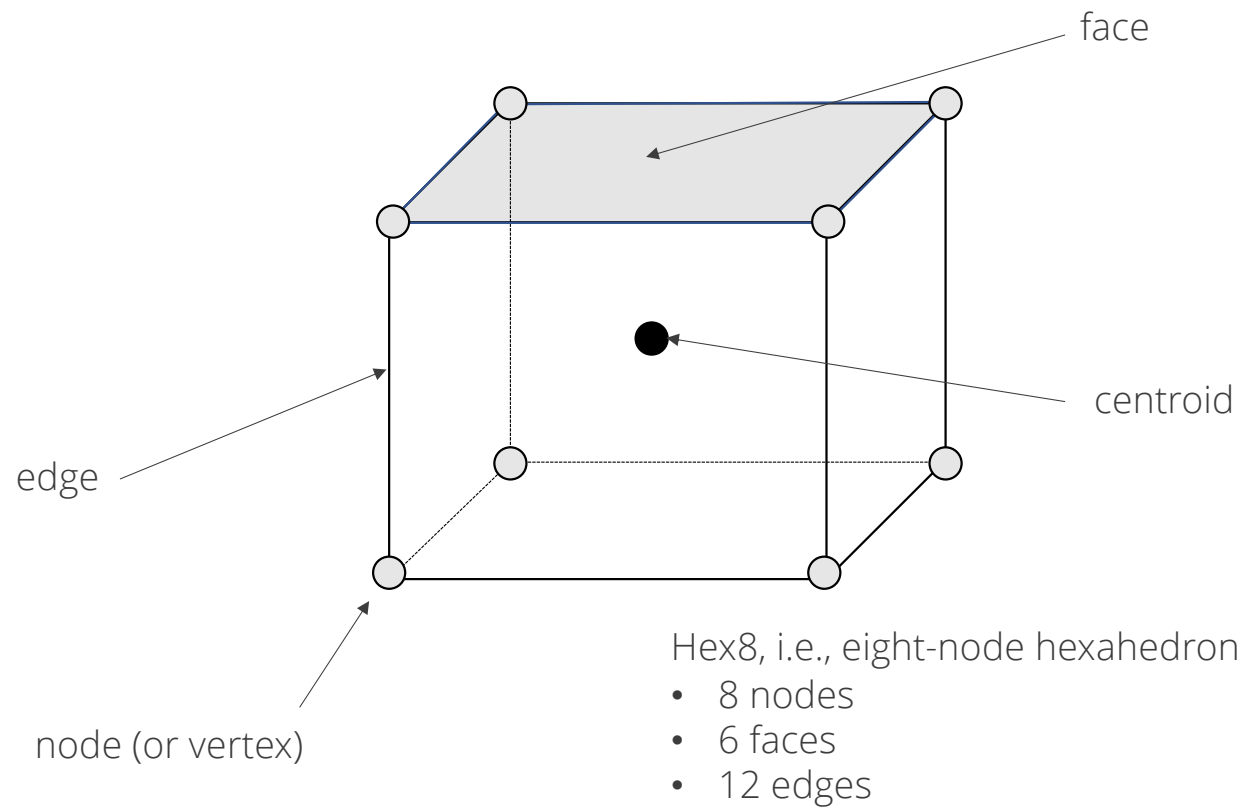
Cut-cells and embedded
approaches help

<http://ccmsc.utah.edu/about.html>



Attributes of an Element

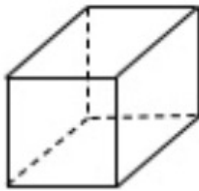
An element consists of nodes, edges, and faces





Examples of Various Topologies

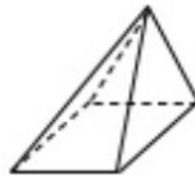
Hex8



Tet4



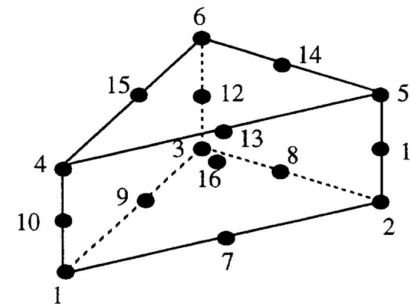
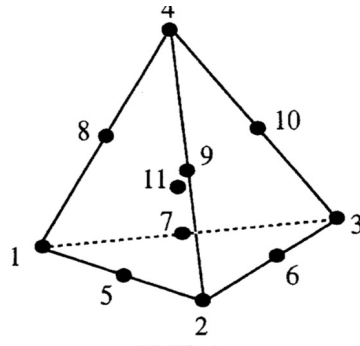
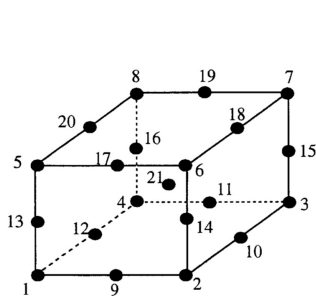
Pyramid5



Wedge6



Arbitrary

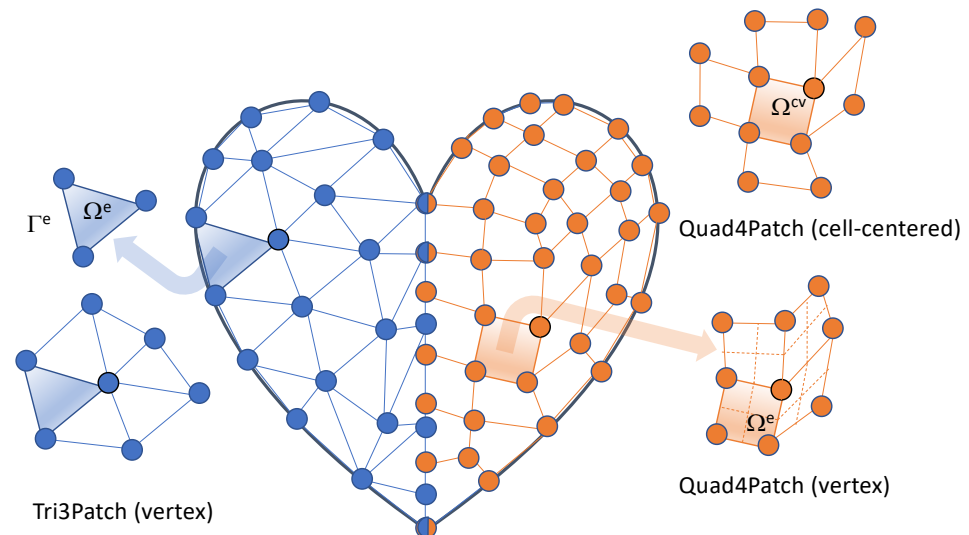
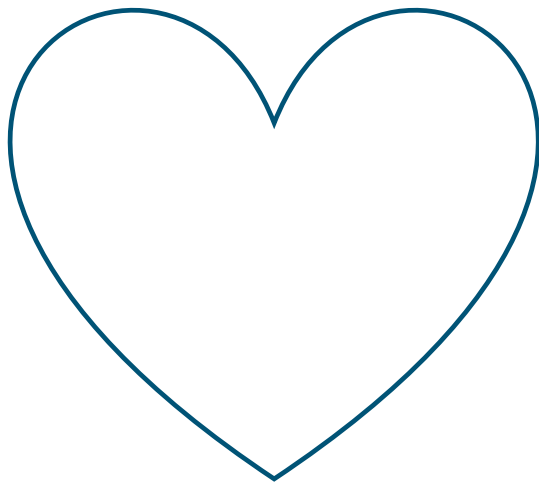


Higher-order promoted elements (Hex27, Tet10, Wedge16, Hex64, etc.)



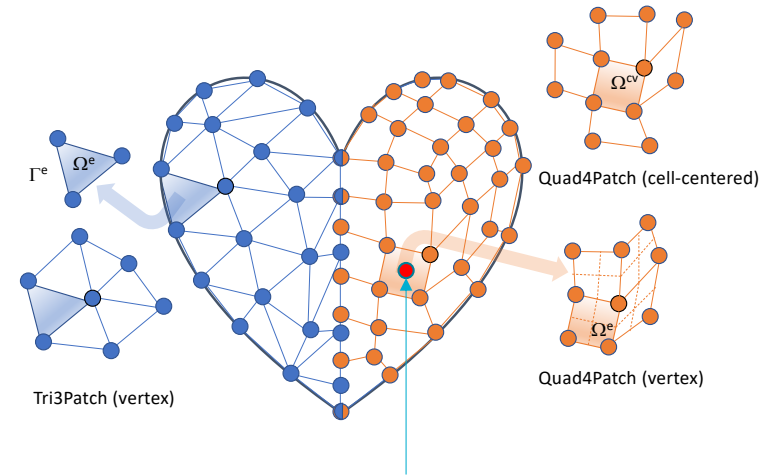
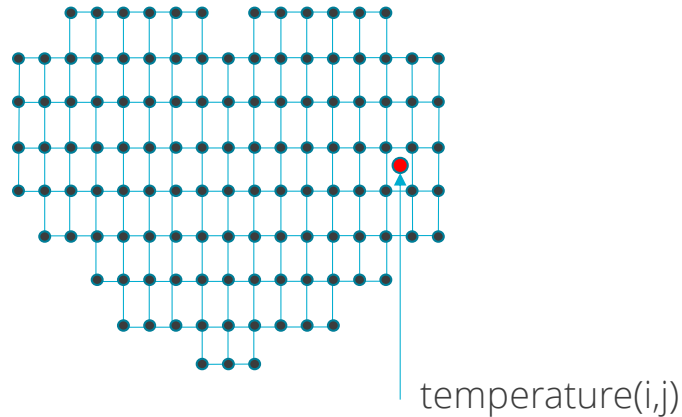
Introducing a Mesh over Heart Domain, Ω

- Elements of size 4 (Quad4) or 3 (Tri3) have been introduced
- Exterior domain is faceted
- Non-conformal interface between the Tri3 and Quad4 block
- Two types of connectivity have been presented: node:element and element:face:element
- Two types of integration: Ω^e vs Ω^{cv}





Data Structure Ramifications: A bit more complex...



- Element and associated data structures are indexed directly via i^{th} and j^{th} location, e.g., **temperature**(i,j), over the range: **temperature**(0:nX-1,0:nY-1)
- Neighbors are directly indexed, e.g., “north” neighbor of (i,j) is ($i,j+1$)
- Element and associated data structures are indexed indirectly via a data structure, e.g., **temperature**(k), over the range: **temperature**(0:nElem-1)
- Nodes of element(k) are obtained via connectivity relationship mappings
 - `std::vector<mesh_type> nodes = elem_nodes(k)`
- Nodal fields, for element k via:
 - `pressure = field_data(nodes[0,...,numElem])`



Integration Over the Domain: The "Finite" in Finite-Volume and Finite Element

- Consider a simple model equation with the heart domain in mind:

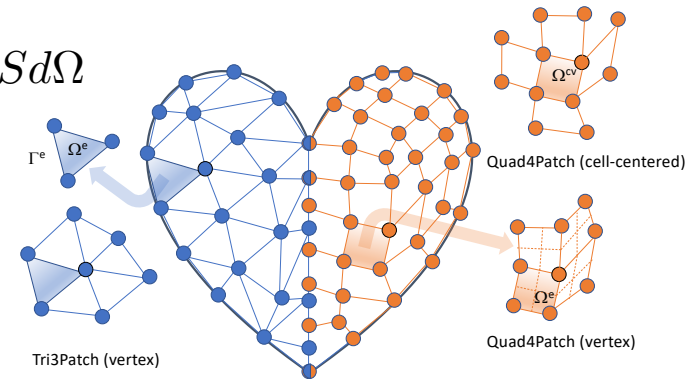
$$\frac{\partial F_j}{\partial x_j} = S$$

Where F_j is a flux and S is a source term

- Integrating over the entire domain, Ω :
$$\int_{\Omega} \frac{\partial F_j}{\partial x_j} d\Omega = \int_{\Omega} S d\Omega$$
- Without loss of generality, let us define a set of subdomains, Ω_k :

$$\sum_k \int_{\Omega_k} \frac{\partial F_j}{\partial x_j} d\Omega_k = \sum_k \int_{\Omega_k} S d\Omega_k$$

- As present, only volumetric integrals appear



Note:

- The formality of Σ_k and Ω_k is implied to exist over the full domain and is often times dropped – integral type implied by dV (volume) and dS (domain boundary)



Fundamentals of Discretization: Surface vs Volume Integrations

- Given a partial differential equation (PDE) and associated volumetric form:

$$\int \frac{\partial F_j}{\partial x_j} dV = \int S dV$$

- Applying Gauss Divergence provides the standard finite volume form for fluxes in surface integral form (no distinction between internal control volume faces and boundary faces):

$$\sum_k \int_{\Omega_k} \frac{\partial F_j}{\partial x_j} d\Omega_k = \sum_k \int_{\Omega_k} S d\Omega_k \longrightarrow \sum_k \int_{\Gamma_k} F_j n_j d\Gamma_k = \sum_k \int_{\Omega_k} S d\Omega \longrightarrow \int F_j n_j dS = \int S dV$$

- We can also multiple PDE by an arbitrary test function, w , and integrate over a volume,

$$\int w \frac{\partial F_j}{\partial x_j} dV = \int w S dV$$

Next, integrate by parts and apply Gauss-Divergence. Note, that test function must be differentiable – shown here, at least once..

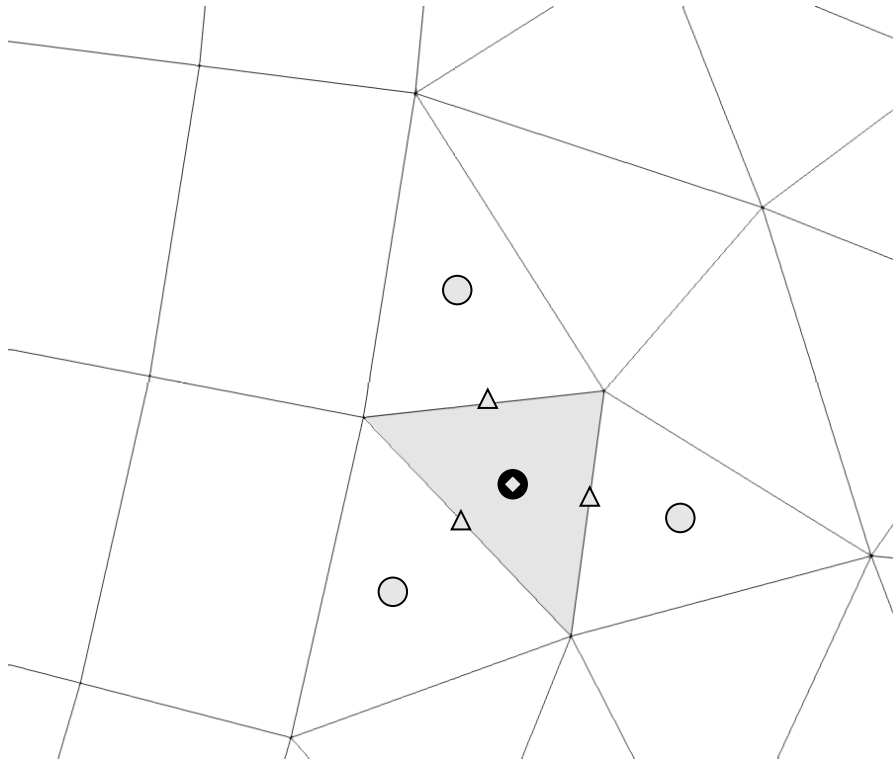
$$-\int F_j \frac{\partial w}{\partial x_j} dV + \int \frac{\partial w F_j}{\partial x_j} dV = -\int \underset{\substack{\uparrow \\ \text{Interior}}}{F_j \frac{\partial w}{\partial x_j}} dV + \int \underset{\substack{\uparrow \\ \text{Boundary}}}{w F_j n_j} dS$$

$$\frac{\partial w F_j}{\partial x_j} = w \frac{\partial F_j}{\partial x_j} + \frac{\partial w}{\partial x_j} F_j$$



Define a Stencil: Element:Face:Element

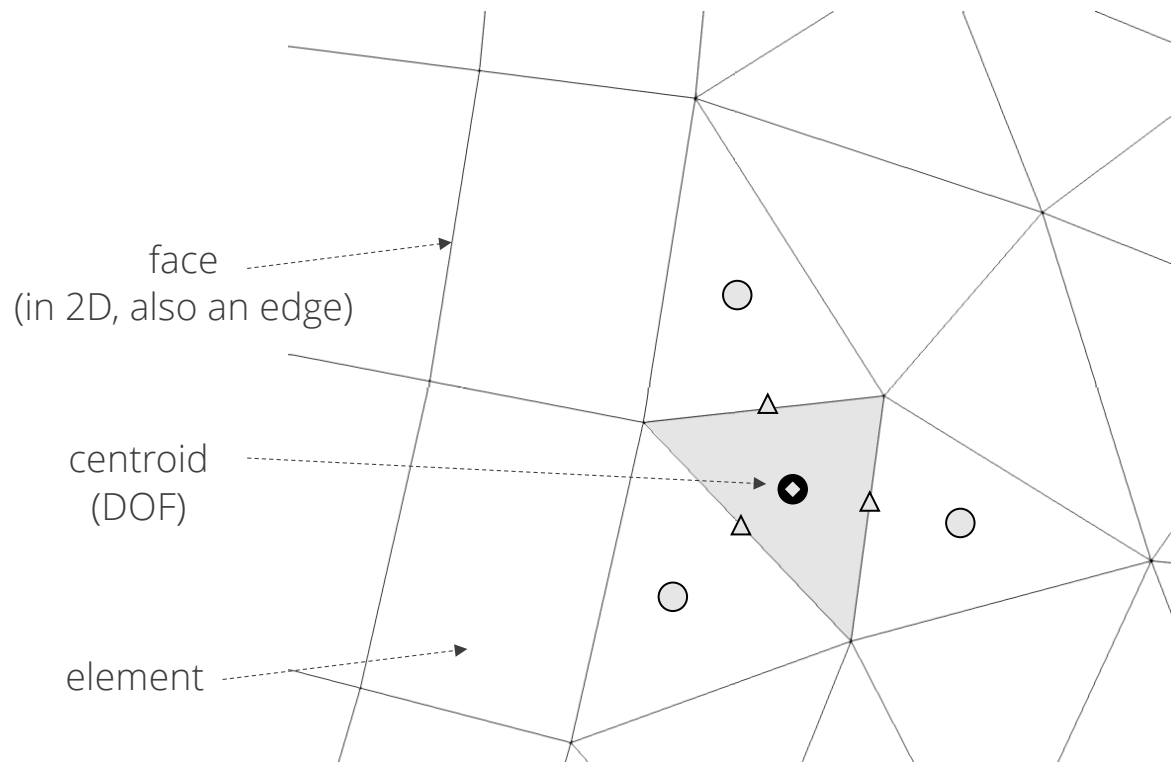
- Cell-centered, finite volume (shaded region)
 - Degree of freedom (DOF), i.e., solution, resides at the centroid of the element





Define a Stencil: Element:Face:Element

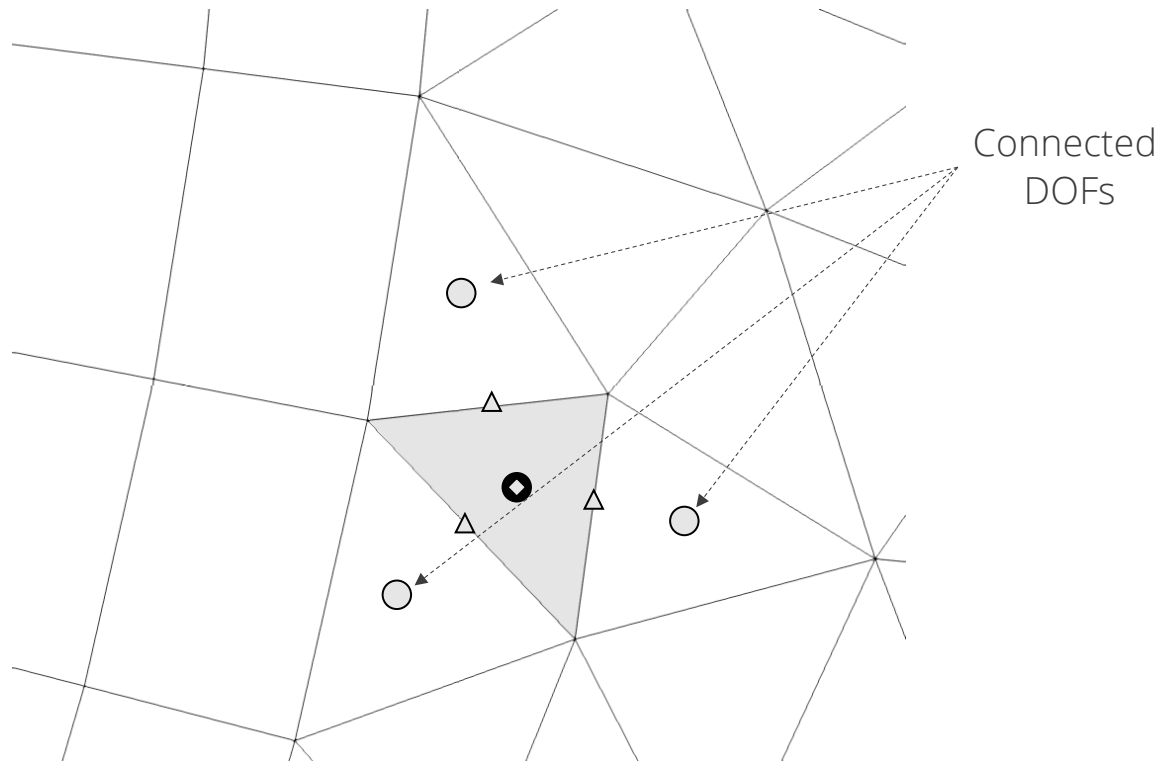
- Cell-centered, finite volume (shaded region)
 - Degree of freedom (DOF), i.e., solution, resides at the centroid of the element





Define a Stencil: Element:Face:Element

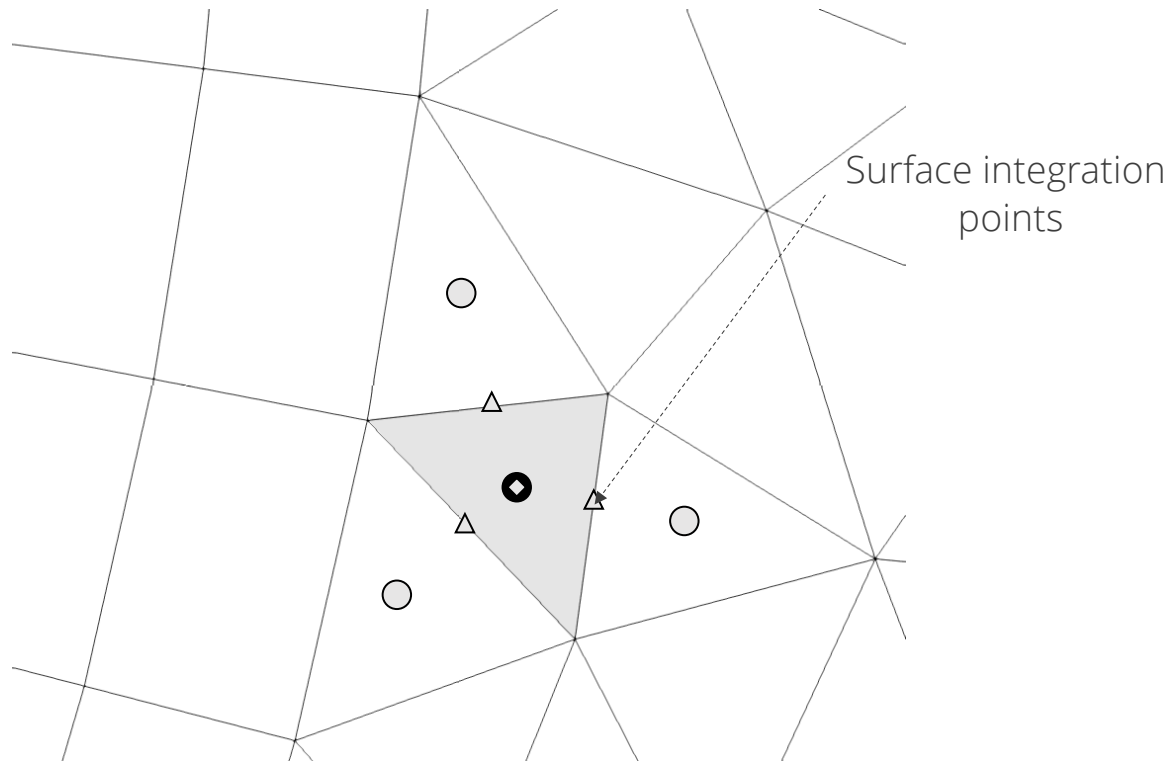
- Cell-centered, finite volume (shaded region)
 - Degree of freedom (DOF), i.e., solution, resides at the centroid of the element





Define a Stencil: Element:Face:Element

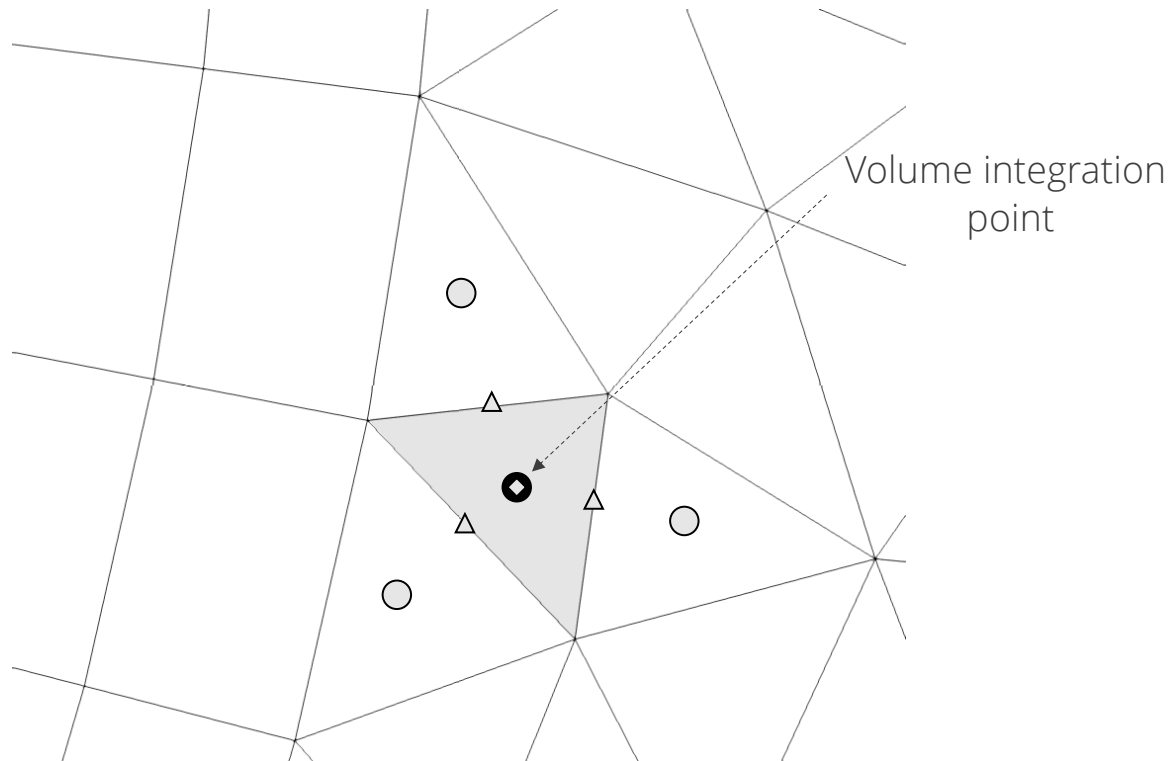
- Cell-centered, finite volume (shaded region)
 - Degree of freedom (DOF), i.e., solution, resides at the centroid of the element





Define a Stencil: Element:Face:Element

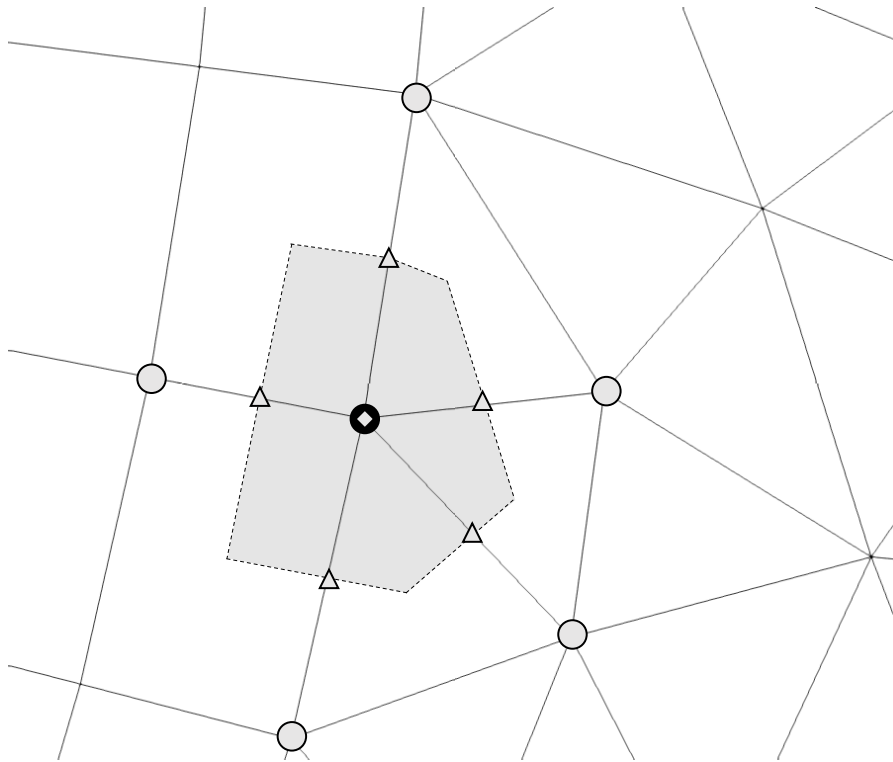
- Cell-centered, finite volume (shaded region)
 - Degree of freedom (DOF), i.e., solution, resides at the centroid of the element





Define a Stencil: Node:Edge:Node

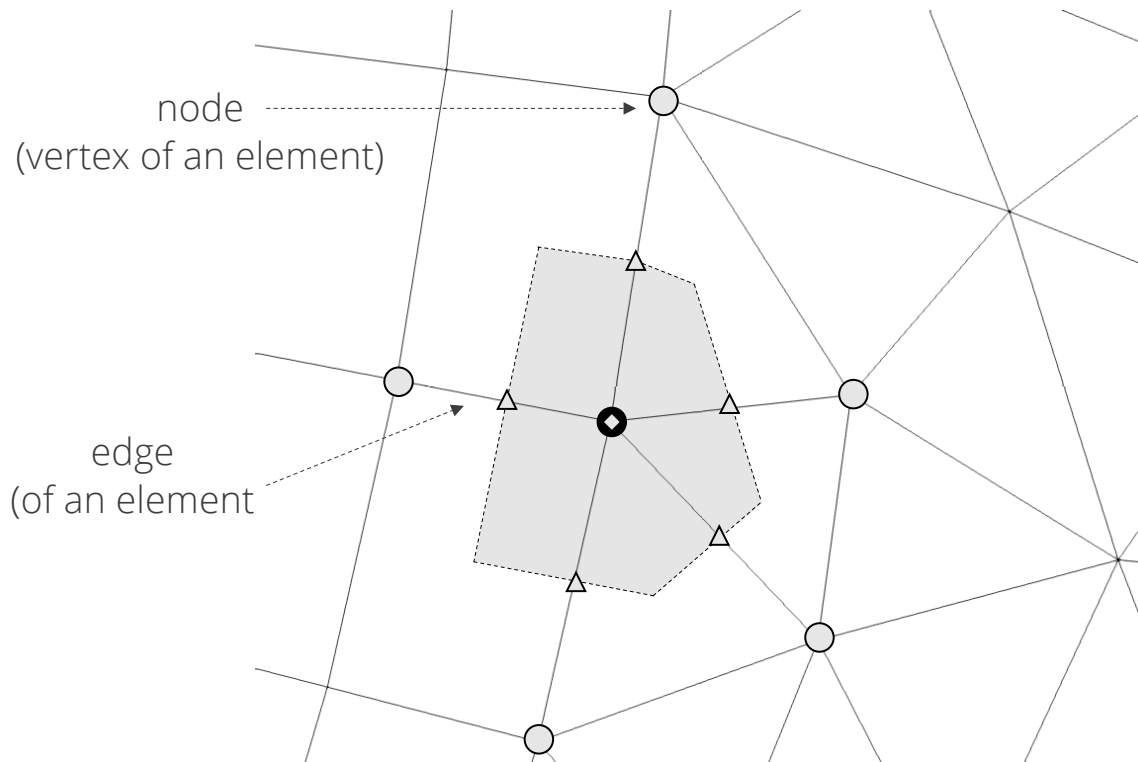
- Edge-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Edge:Node

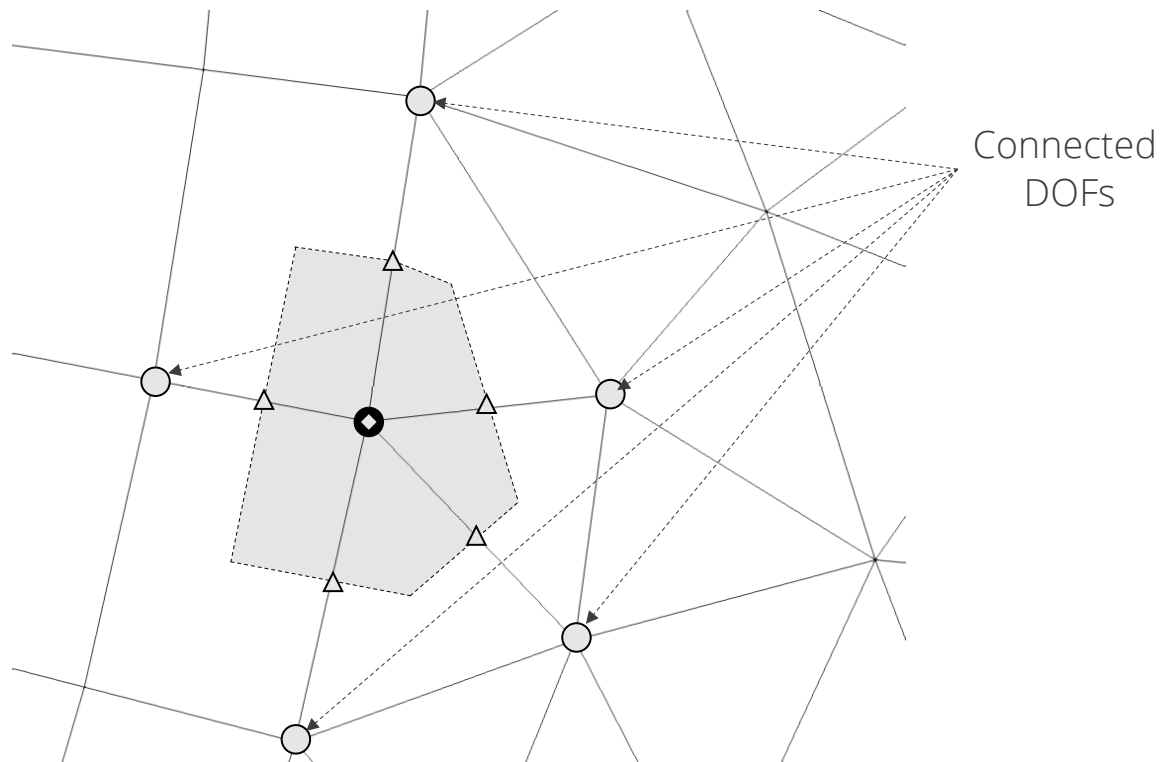
- Edge-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Edge:Node

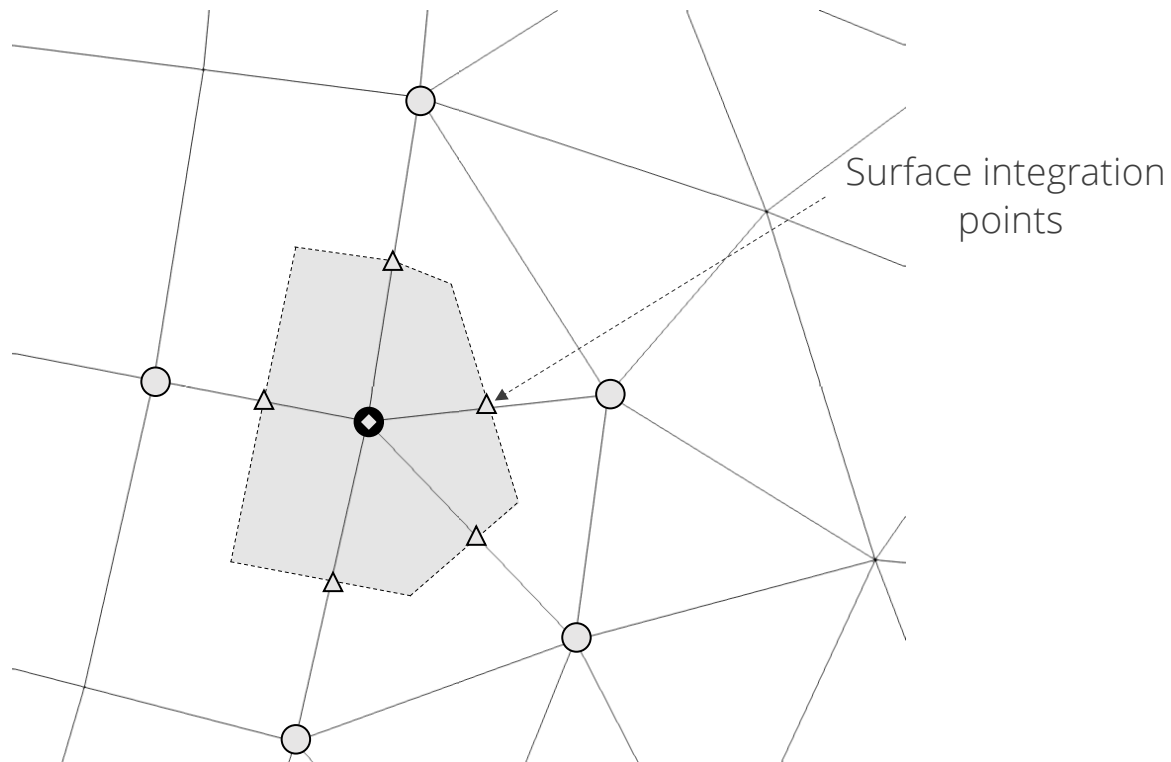
- Edge-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Edge:Node

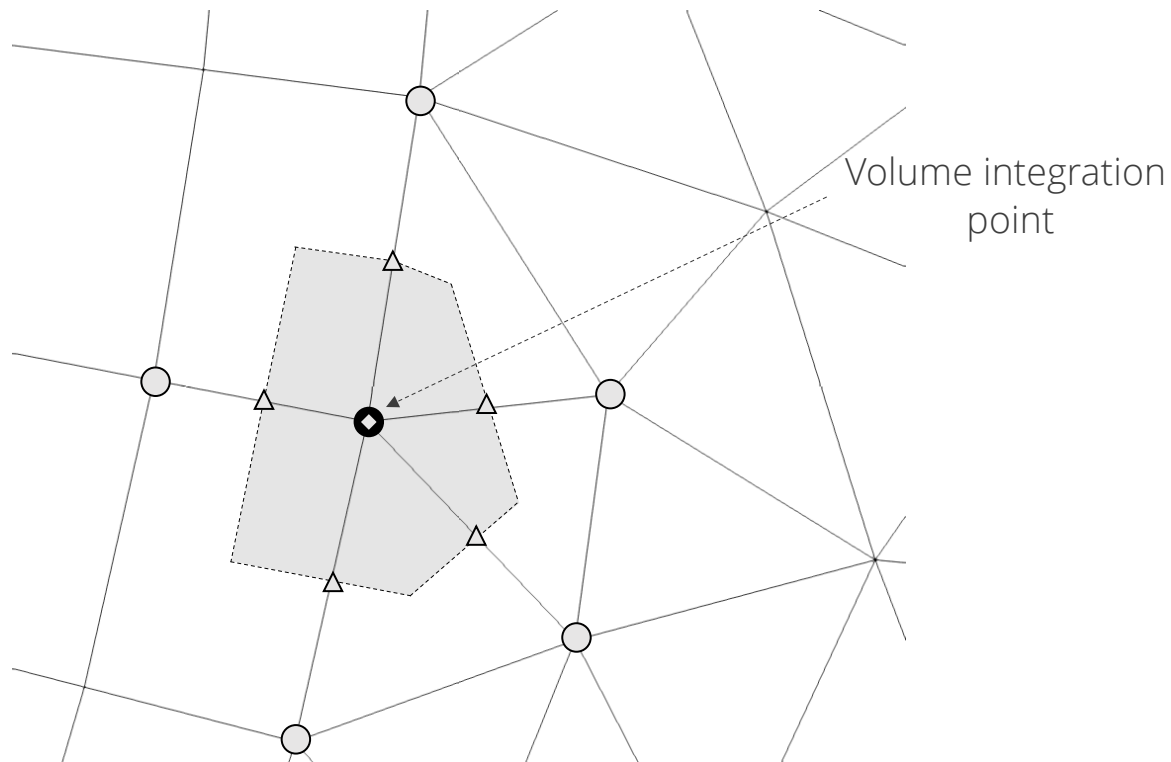
- Edge-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Edge:Node

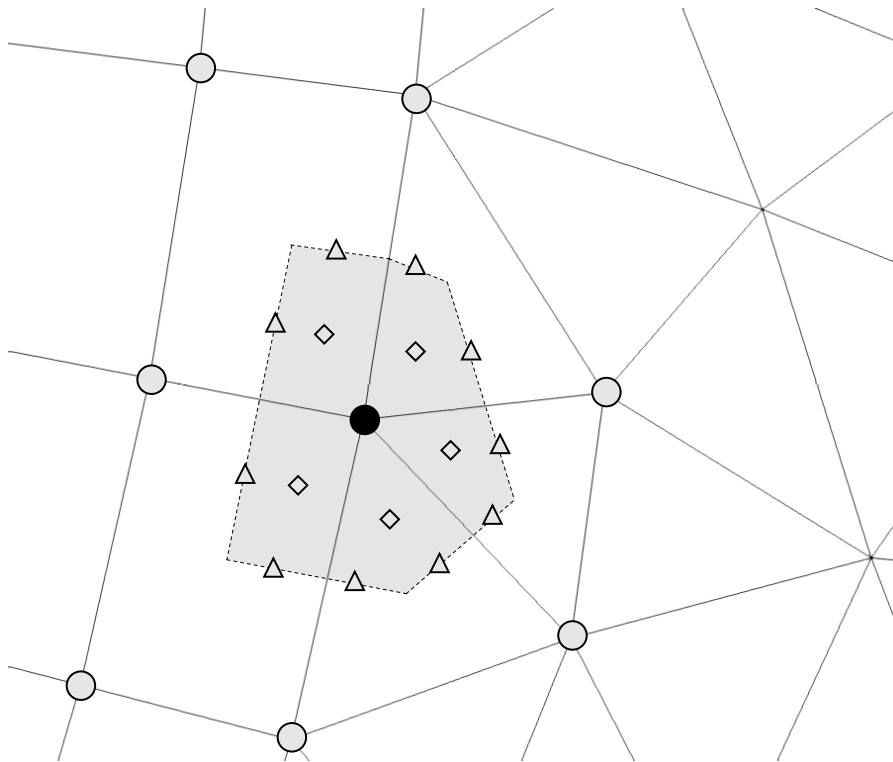
- Edge-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Element:Node

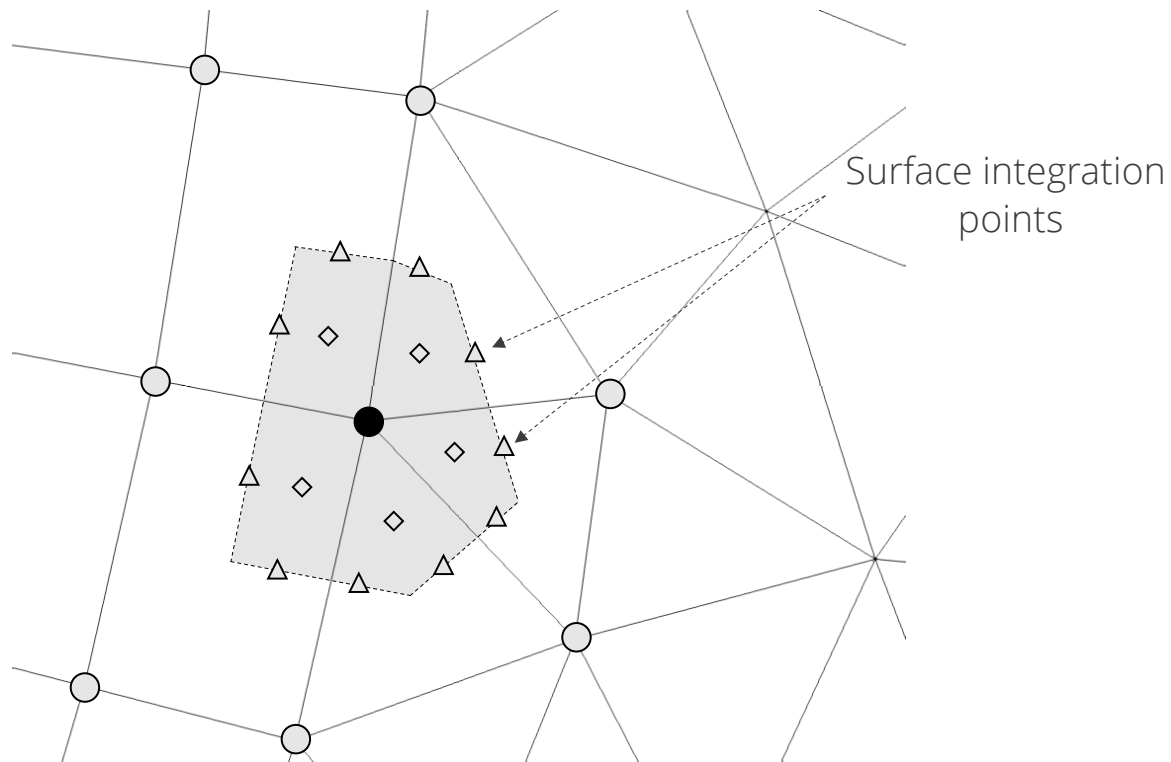
- Choice #1, Element-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Element:Node

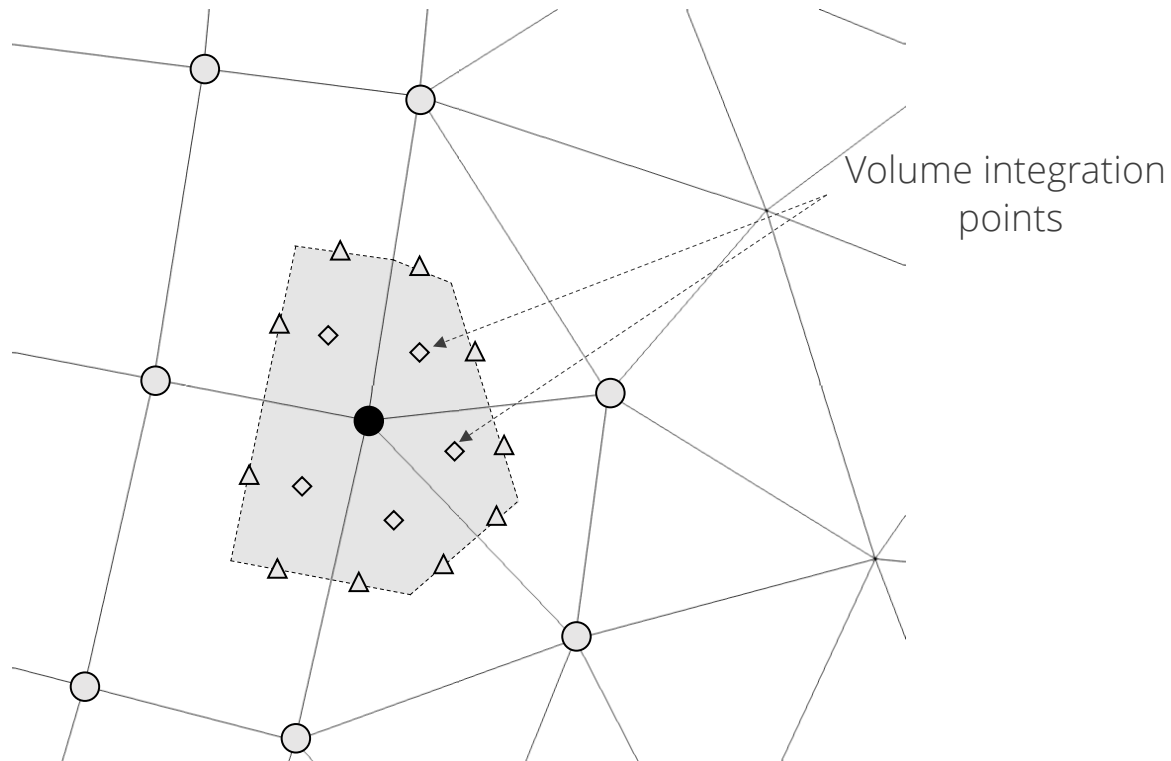
- Choice #1, Element-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Element:Node

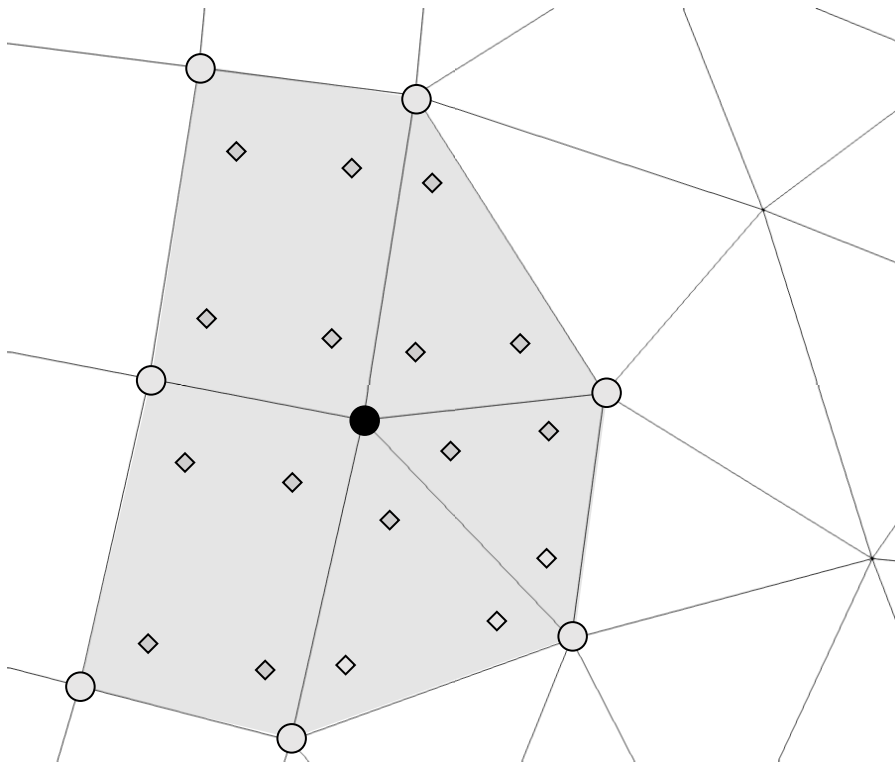
- Choice #1, Element-based, vertex (or node)-centered finite volume (shaded region)
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Element:Node

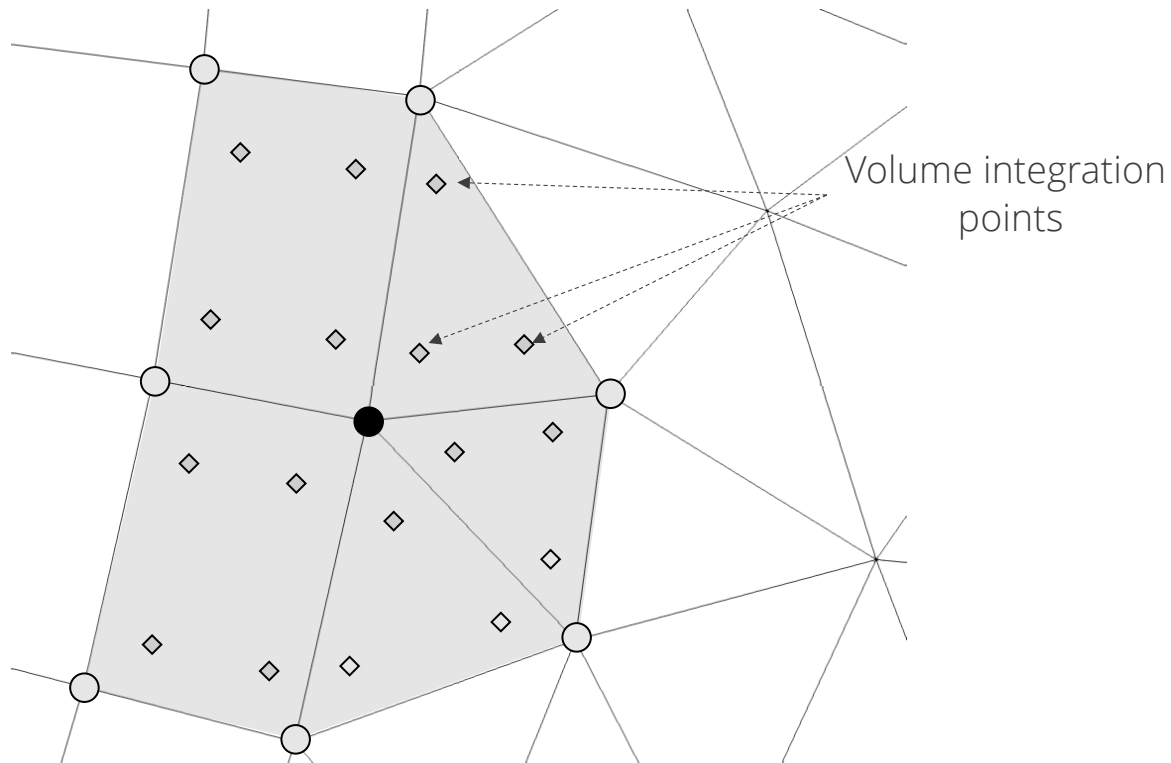
- Choice #2, Element-based, finite element
 - Degree of freedom, i.e., solution, resides at the node, or vertex





Define a Stencil: Node:Element:Node

- Choice #2, Element-based, finite element
 - Degree of freedom, i.e., solution, resides at the node, or vertex



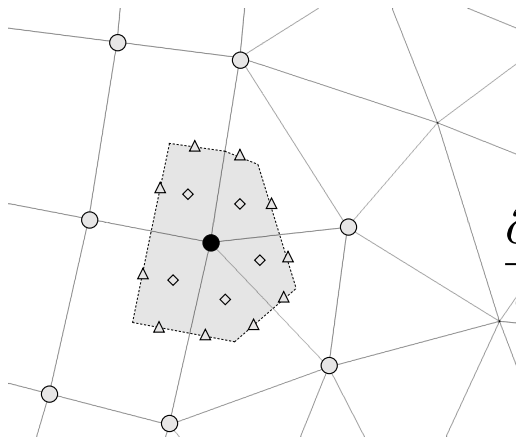


VOF Transport Discretization Nuance: Volume- or Surface-based?

- Simple enough, define the volume (fraction) of fluid (absent evaporation): $\frac{\partial \alpha}{\partial t} + u_j \frac{\partial \alpha}{\partial x_j} = 0$

Option 1: volumetric-form: $\int \frac{\partial \alpha}{\partial t} dV + \int u_j \frac{\partial \alpha}{\partial x_j} dV = 0$

CVFEM/FEM (really, any element-based approach)
Evaluated as a volumetric-contribution (diamonds)



$$\phi_{ip} = \sum_n N_n^{ip} \phi_n$$

$$\frac{\partial \phi_{ip}}{\partial x_j} = \sum_n \frac{\partial N_n^{ip}}{\partial x_j} \phi_n$$

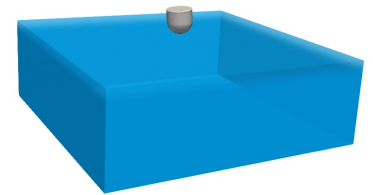
Option 2: divergence-form:

$$\int \frac{\partial \alpha}{\partial t} dV + \int \frac{\partial \alpha u_j}{\partial x_j} dV - \int \alpha \frac{\partial u_j}{\partial x_j} dV = 0$$

Gauss-Divergence $\int \alpha u_j n_j dS$

Traditional finite volume (element, edge, cell-centered)
Evaluated as a surface integral (triangle)

Allows for a consistent advecting velocity (mass conserving)
that is obtained from the continuity equation





Linking to 1d_quad4_adv_diff

Recall, the transport equation for the 1d_quad4_adv_diff laboratory exercise is as follows:

$$\frac{\partial \phi}{\partial t} + u_j \frac{\partial \phi}{\partial x_j} + \frac{\partial q_j}{\partial x_j} = 0 \quad \text{where} \quad q_j = -\nu \frac{\partial \phi}{\partial x_j}$$

For now, let's focus on the advection term. We know that we can integrate over the volume and simply compute this term at the volume integration points,

$$\int u_j \frac{\partial \phi}{\partial x_j} dV \quad \text{Option: } \mathbf{scv_advection_np}$$

As with the volume of fluid equation on the previous slide, we can also write this term as:

$$\int u_j \frac{\partial \phi}{\partial x_j} dV = \int \frac{\partial u_j \phi}{\partial x_j} dV - \int \phi \frac{\partial u_j}{\partial x_j} dV \quad \text{that can be simplified (for constant velocity)}$$

$$\int u_j \frac{\partial \phi}{\partial x_j} dV = \int u_j \phi n_j dS \quad \text{Option: } \mathbf{scs_advection_np} \text{ (or } \mathbf{scs_upw_advection_np})$$



Linking to 1d_quad4_adv_diff

Recall, the transport equation for the 1d_quad4_adv_diff laboratory exercise is as follows:

$$\frac{\partial \phi}{\partial t} + u_j \frac{\partial \phi}{\partial x_j} + \frac{\partial q_j}{\partial x_j} = 0 \quad \text{where} \quad q_j = -\nu \frac{\partial \phi}{\partial x_j}$$

For now, let's focus on the advection term. We know that we can integrate over the volume and simply compute this term at the volume integration points,

$$\int u_j \frac{\partial \phi}{\partial x_j} dV$$

Option: **scv_advection_np**

Q: Is this really simple for all of the schemes we have discussed?

As with the volume of fluid equation on the previous slide, we can also write this term as:

$$\int u_j \frac{\partial \phi}{\partial x_j} dV = \int \frac{\partial u_j \phi}{\partial x_j} dV - \int \phi \frac{\partial u_j}{\partial x_j} dV \quad \text{that can be simplified (for constant velocity)}$$

$$\int u_j \frac{\partial \phi}{\partial x_j} dV = \int u_j \phi n_j dS \quad \text{Option: } \mathbf{scs_advection_np} \text{ (or } \mathbf{scs_upw_advection_np})$$

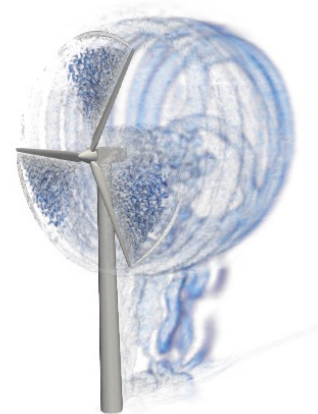
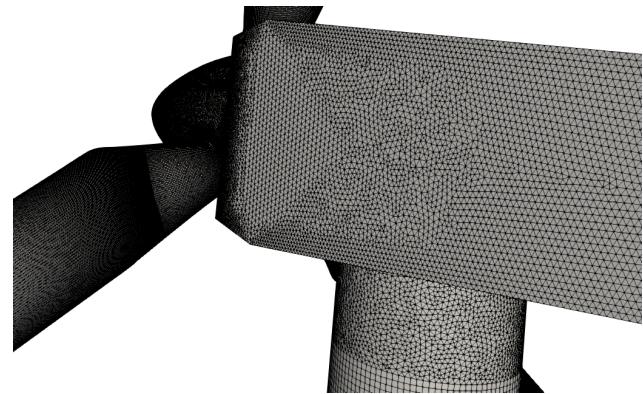


Reality: Meshing time for complex applications remains a significant bottleneck!

- Many applications of interest contain complex geometries low-Mach fluids users interested in high-quality simulation results tend towards hexahedral-based topologies (if possible)
- However, if a scheme is “design-order” accurate, any topology may suffice as it is simply a matter of mesh size and efficiency – not unlike the active discussion on low- vs higher-order
- Sometimes, the penetration of a low-Mach fluids physics addition in common analysis is high as the meshing can be prohibitively complex



Very complex world – stair-stepped!



Example: Vestas V27 225
kw hybrid low-order
Hex8/Tet4/Pyr5/Wedge6



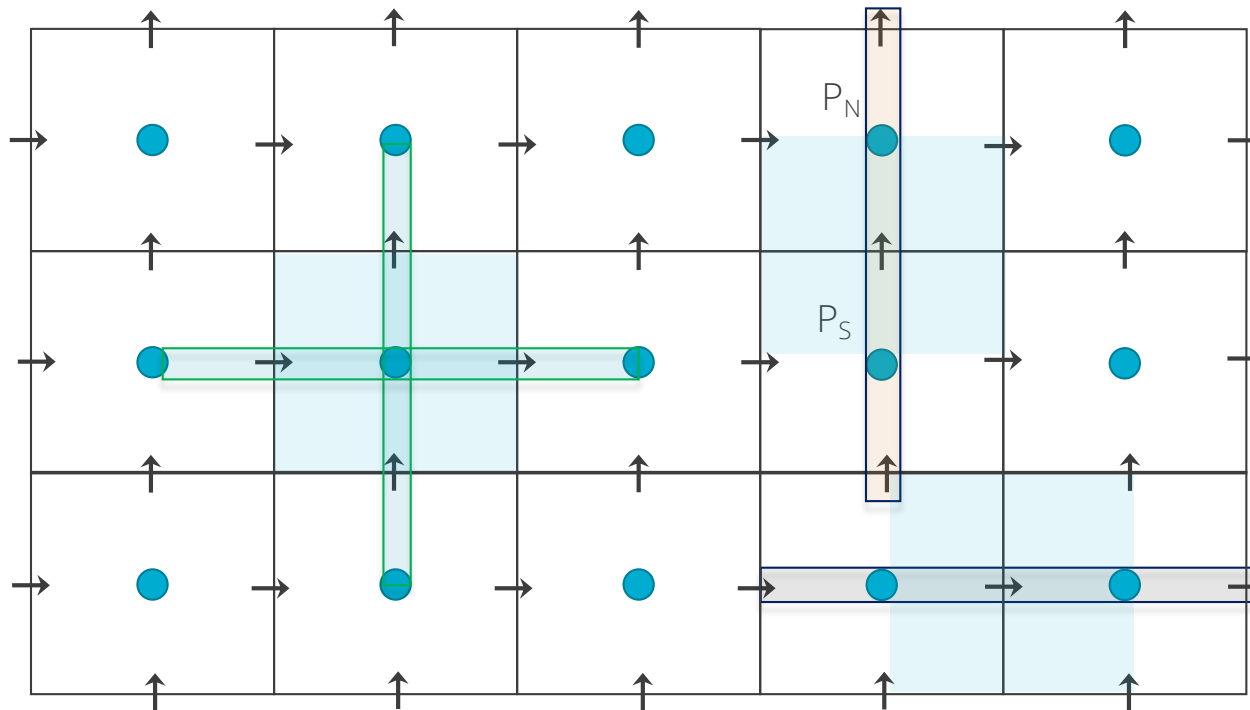
Classic Staggered Finite Volume

- Velocity degree-of-freedom is staggered relative to pressure and other primitives, e.g., enthalpy, mixture fraction, etc.

Stencil for CC-quantities ●

Stencil for x-velocity →

Stencil for y-velocity ↑





Attributes of a Staggered Scheme

- By design, non-orthogonality is absent, however, complex geometry will be stair-stepped
- From a fluids perspective, the operators are ideal, i.e., pressure gradient for momentum is compact, e.g., $(P_E - P_W)\Delta x^{-1}$
- As will be seen in future lecture topics, the skew-adjoint nature of the Divergence operator, **D**, and Gradient operator, **G**, allows for a Laplace operator, **L = DG**
- Can be extended to higher-order
- Frequently, meshing complex geometries can be extremely difficult (consider our V27 example)



An Informal Survey....

- Cell-Centered: (Sometimes generalized Polyhedra)
 - Ansys Fluent, OpenFOAM (FireFOAM, NavyFOAM), CD-Adapco (Star-CCM), Soleil-X (Stanford)
- EBVC: (Most typical in the acoustically compressible space)
 - SU2, FUN3D, CHAD, Nalu-Wind (production), Nalu (option to explore discretizations)
- CVFEM
 - Fluent (originally!), CFX-TASC-FLOW, Sierra Fuego (SNL), Nalu
- FEM
 - FIDAP, COMSOL, EDDY (NASA), AcuSim, PHASTA
- Staggered
 - Arches (Utah)



An Informal Survey....

- Cell-Centered: (Sometimes generalized Polyhedra)
 - Ansys Fluent, OpenFOAM (FireFOAM, NavyFOAM), CD-Adapco (Star-CCM), Soleil-X (Stanford)
- EBVC: (Most typical in the acoustically compressible space)
 - SU2, FUN3D, CHAD, Nalu-Wind (production), Nalu (option to explore discretizations)
- CVFEM
 - Fluent (originally!), CFX-TASC-FLOW, Sierra Fuego (SNL), Nalu
- FEM
 - FIDAP, COMSOL, EDDY (NASA), AcuSim, PHASTA
- Staggered
 - Arches (Utah)

Common Water-Cooler CFD Arguments:

- Structured vs. Un-structured
- FEM vs. Finite Volume
- Node-centered vs. Cell-centered
- Monolithic vs. Operator Split
- Compressible vs. (acoustically) Incompressible
- Explicit vs. Implicit



An Informal Survey....

- Cell-Centered: (Sometimes generalized Polyhedra)
 - Ansys Fluent, OpenFOAM (FireFOAM, NavyFOAM), CD-Adapco (Star-CCM), Soleil-X (Stanford)
- EBVC: (Most typical in the acoustically compressible space)
 - SU2, FUN3D, CHAD, Nalu-Wind (production), Nalu (option to explore discretizations)
- CVFEM
 - Fluent (originally!), CFX-TASC-FLOW, Sierra Fuego (SNL), Nalu
- FEM
 - FIDAP, COMSOL, EDDY (NASA), AcuSim, PHASTA
- Staggered
 - Arches (Utah)

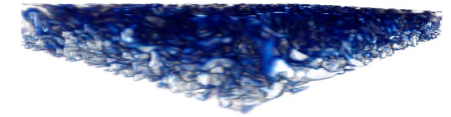
Common Water-Cooler CFD Arguments:

- Structured vs. Un-structured
- FEM vs. Finite Volume
- Node-centered vs. Cell-centered
- Monolithic vs. Operator Split
- Compressible vs. (acoustically) Incompressible
- Explicit vs. Implicit

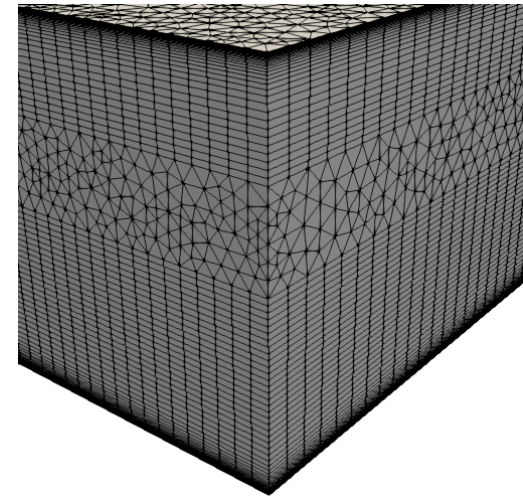
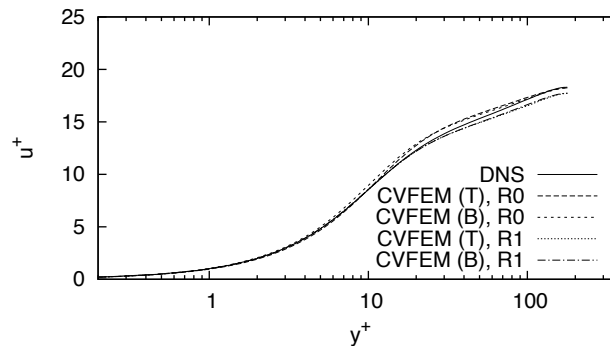
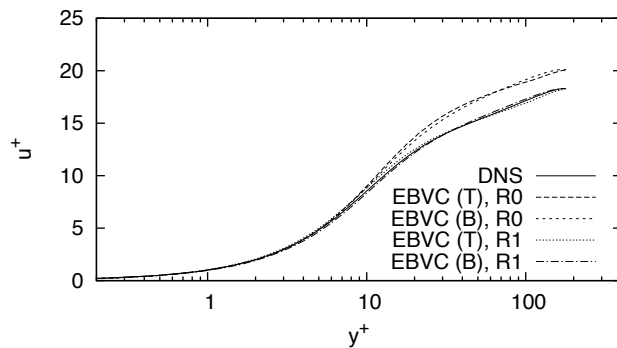




Hybrid Meshes, Even for LES!



- Hybrid mesh study based on Ham and Iaccarino, *CTR Annual Brief*, 2006, found that simulations were extremely sensitive to mesh topology
- Non-symmetric time mean flow found for cell-centered; better for the CTR node-centered formulation
- Native CVFEM and EBVC are both symmetric in mean quantities



Domino, et. al, "The suitability of hybrid meshes for low-Mach large-eddy simulation" Stanford CTR Summer Program, 2018



Recent Generalized Unstructured Findings

- Domino, et. al, "An assessment of atypical mesh topologies for low-Mach large-eddy simulation", Comput. Fluids (2019)

