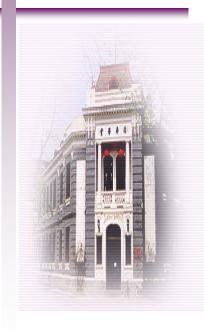
### State Feedback Control Systems

**Chapter 6** 







- Definition of controllability
- Controllability condition and its demonstration
- Controllability canonical form and its controllability

#### **Outlines to this Lesson**



- Definition of observability
- Observability condition and its demonstration
- Observability canonical form and its observability
- Controllability and Observability versus Zero-Pole Cancellation
- Controllability and Observability Decomposition
- State-feedback control
- Output-feedback control





#### Definition:

Given a linear time-invariant system that is described by the following dynamic equations, the state  $X(t_0)$  is said to be observable if given any input U(t), there exists a finite time  $t_f \geq t_0$  such that the knowledge of U(t) for  $t_0 \leq t < t_f$ ; matrices A, B, C, and D; and the output Y(t) for  $t_0 \leq t < t_f$  are sufficient to determine  $X(t_0)$ . If every state of the system is observable for a finite  $t_f$ , we say that the system is completely observable, or simply observable.

$$\dot{X}(t) = AX(t) + BU(t)$$

$$Y(t) = CX(t) + DU(t)$$

$$X(t): n \times 1$$

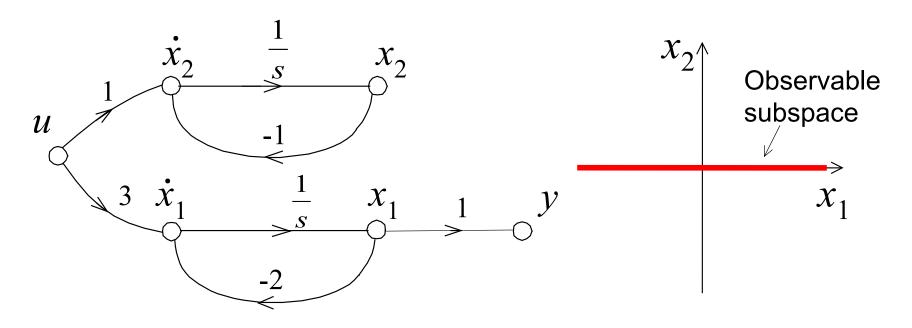
$$U(t): r \times 1$$

$$Y(t): p \times 1$$

### **Examples of Unobservable**



Example 6.7 Find the given system is observable or unobservable.

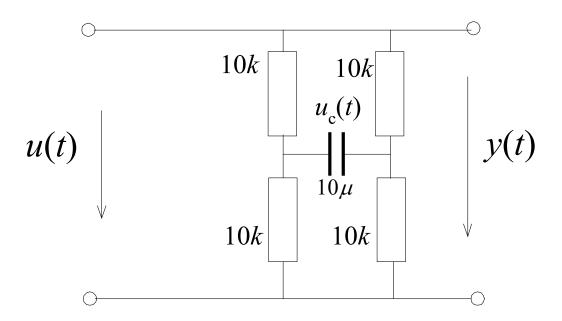


y is independent of  $x_2$ , system is unobservable.

### **Examples of Unobservable**



Example 6.8 Find the given system is observable or unobservable.



y(t) does not have any information of  $u_c(t)$ , system is unobservable.

### **Observability Condition**



For the system described by the given state equation

$$\dot{X}(t) = AX(t) + BU(t)$$

$$Y(t) = CX(t) + DU(t)$$

$$X(t): n \times 1$$

$$U(t): r \times 1$$

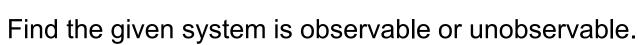
$$Y(t): p \times 1$$

to be completely observable, it is necessary and sufficient that the following  $np \times n$  observability matrix V has a rank of n:

$$V = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}_{np \times n}$$

### **Observability Example**

Example 6.9





$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -2 & 0 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 3 \\ 1 \end{bmatrix} u(t) \qquad y = \begin{bmatrix} 1 & 2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

A: 
$$C = \begin{bmatrix} 1 & 2 \end{bmatrix}$$
  $CA = \begin{bmatrix} -2 & -2 \end{bmatrix}$ 

$$V = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 1 & 2 \\ -2 & -2 \end{bmatrix}$$

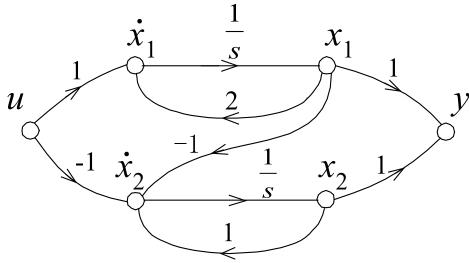
$$rank(V) = 2$$
 system is observable

### Observability Example

Example 6.10

Find the given system is observable or unobservable.





A:

$$\dot{X} = \begin{bmatrix} 2 & 0 \\ -1 & 1 \end{bmatrix} X + \begin{bmatrix} 1 \\ -1 \end{bmatrix} u$$

$$y = \begin{bmatrix} 1 & 1 \end{bmatrix} X$$

$$\dot{X} = \begin{bmatrix} 2 & 0 \\ -1 & 1 \end{bmatrix} X + \begin{bmatrix} 1 \\ -1 \end{bmatrix} u \qquad CA = \begin{bmatrix} 1 & 1 \end{bmatrix} \qquad V = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 1 & 1 \\ 1 & 1 \end{bmatrix}$$

$$rank(V) = 1 < 2$$

system is unobservable

### **Proof of the Observability Condition**

To prove the observability of a system, we need to demonstrate that by the knowledge of U(t), A, B, C, D, and Y(t), we can determine  $X(t_0)$ .

For a LTI system: 
$$\dot{X}(t) = AX(t) + BU(t)$$
  $X(t): n \times 1$   $Y(t) = CX(t) + DU(t)$   $Y(t): p \times 1$ 

Set 
$$U(t) = 0$$

$$Y = CX$$

$$\dot{Y} = C\dot{X} = CAX$$

$$\dot{Y} = CA\dot{X} = CA^{2}X$$

$$\vdots$$

$$Y$$

$$\vdots$$

$$Y$$

$$CA$$

$$\vdots$$

$$Y$$

$$CA$$

$$\vdots$$

$$Y$$

$$CA$$

$$\vdots$$

$$CA^{n-1}$$

$$Y^{(n-1)} = CA^{n-1}Y$$

#### For an single output system:



$$\begin{bmatrix} Y(t_0) \\ \dot{Y}(t_0) \\ \vdots \\ Y^{(n-1)}(t_0) \end{bmatrix} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{(n-1)} \end{bmatrix}_{n \times n} \begin{bmatrix} x_1(t_0) \\ x_2(t_0) \\ \vdots \\ x_3(t_0) \end{bmatrix}_{n \times 1}$$

There are n unknown variables and n equations.

If rank(V) = n, the inverse of V exists, the equation has unique solution, and the system is observable.

If rank(V) < n, the inverse of V does not exist, we can not find a unique solution through the above equation.

For a multiple-output system, assume there are q outputs:



$$\begin{bmatrix} Y(t_0) \\ \dot{Y}(t_0) \\ \vdots \\ Y^{(n-1)}(t_0) \end{bmatrix} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}_{nq \times n} \begin{bmatrix} x_1(t_0) \\ x_2(t_0) \\ \vdots \\ x_n(t_0) \end{bmatrix}_{n \times 1}$$

There are n unknown variables and nq equations.

If rank(V) = n, the equations have solutions, and the system is observable.

If rank(V) < n , we can not find a unique solution through the above equation.

Observability does not change after non-singular linear transformation

### **Observability Condition 2**

For a linear time-invariant system with distinct eigenvalues to be observable, it is necessary and sufficient that the C matrix in its diagonal canonical form does not have a column with all elements to be zero.

For example:

A single output system

$$\dot{X} = \begin{bmatrix} \lambda_1 & & & & 0 \\ & \lambda_2 & & \\ & & \ddots & & \\ 0 & & & \lambda_n \end{bmatrix} X + \begin{bmatrix} b_1 \\ b_2 \\ \vdots \\ b_n \end{bmatrix} u$$

$$y = \begin{bmatrix} c_1 & c_2 & \cdots & c_n \end{bmatrix} X$$

if C does not have any zero element, the system is observable.

$$\lambda_i \neq \lambda_j$$
  $i, j = 1, 2, \dots, n$   $c_i \neq 0$ ,  $i = 1, 2, \dots, n$ 

$$C = \begin{bmatrix} c_1 & c_2 & \cdots & c_n \end{bmatrix}$$

$$C = \begin{bmatrix} c_1 & c_2 & \cdots & c_n \end{bmatrix}$$

$$CA = \begin{bmatrix} c_1 & c_2 & \cdots & c_n \end{bmatrix} \begin{bmatrix} \lambda_1 & & & 0 \\ & \lambda_2 & & \\ & & \ddots & \\ & & & \lambda_n \end{bmatrix} = \begin{bmatrix} \lambda_1 c_1 & \lambda_2 c_2 & \cdots & \lambda_n c_n \end{bmatrix}$$

$$CA^{2} = \begin{bmatrix} c_{1} & c_{2} & \cdots & c_{n} \end{bmatrix}^{2} \begin{bmatrix} \lambda_{1} & & & & \\ & \lambda_{2} & & \\ & & \ddots & \\ 0 & & & \lambda_{n} \end{bmatrix}^{2} = \begin{bmatrix} \lambda_{1}^{2}c_{1} & \lambda_{2}^{2}c_{2} & \cdots & \lambda_{n}^{2}c_{n} \end{bmatrix}$$

$$V = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix} = \begin{bmatrix} c_1 & c_2 & \cdots & c_n \\ \lambda_1 c_1 & \lambda_2 c_2 & \cdots & \lambda_n c_n \\ \vdots & \vdots & \vdots & \vdots \\ \lambda_1^{n-1} c_1 & \lambda_2^{n-1} c_2 & \cdots & \lambda_n^{n-1} c_n \end{bmatrix} \text{ if } c_i \neq 0, \quad i = 1, 2, \cdots, n$$

$$\lambda_i \neq \lambda_j \quad i \neq j$$

$$i, j = 1, 2, \cdots, n$$
then  $rank(V) = n$ 

Example 6.11 Use Matlab tools to find out the observability of the give system.



$$\dot{X} = \begin{bmatrix} -1 & 0 \\ 0 & -2 \end{bmatrix} X + \begin{bmatrix} 1 \\ -1 \end{bmatrix} u$$

$$y = \begin{bmatrix} 1 & 1 \end{bmatrix} X$$

A: 
$$a=[-1\ 0;0\ -2]$$
  $v=$ 
 $c=[1\ 1]$   $1\ 1$ 
 $v=[c\ ;c*a]$   $-1\ -2$ 
 $rank\ (v)$   $rank\ (v)=2$ 

### Observability Canonical Form



$$\dot{X} = \begin{bmatrix} 0 & 0 & \cdots & 0 & -a_1 \\ 1 & 0 & \cdots & 0 & -a_2 \\ 0 & 1 & \cdots & 0 & -a_3 \\ \vdots & \vdots & \ddots & \vdots & \vdots \\ 0 & 0 & \cdots & 1 & -a_n \end{bmatrix} X + \begin{bmatrix} b_1 \\ b_2 \\ \vdots \\ 0 \\ 0 \end{bmatrix} U$$

$$y = \begin{bmatrix} 0 & 0 & \cdots & 0 & 1 \end{bmatrix} X$$

Systems in Observability canonical form must be observable.

$$V = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix} = \begin{bmatrix} 0 & 0 & \cdots & 1 \\ 0 & 1 & -a_n \\ \vdots & \ddots & \ddots & \vdots \\ 1 & -a_n & \cdots & \cdots \end{bmatrix} \qquad rank(V) = n$$

$$rank(V) = n$$

# **Controllability and Observability versus Zero-Pole Cancellation**



Why can a transfer function always be transformed into a state space model in CCF or OCF?

# Controllability and Observability versus Zero-Pole Cancellation

Example 6.12 Please convert the transfer function of the given system into diagonal canonical form, then discuss its controllability and observability.

$$G(s) = \frac{(s+a)}{(s+2)(s+3)}$$

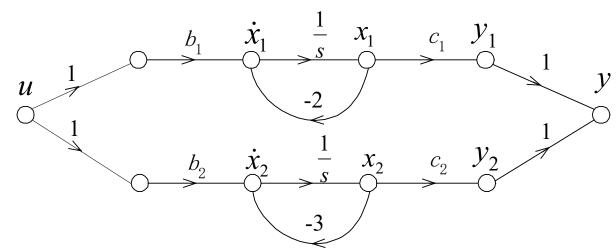
A: 
$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} \lambda_1 & 0 \\ 0 & \lambda_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} b_1 \\ b_2 \end{bmatrix} u \qquad y = \begin{bmatrix} c_1 & c_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

$$Y(s) = G(s)U(s) = \frac{a-2}{s+2}U(s) + \frac{-a+3}{s+3}U(s) = Y_1(s) + Y_2(s)$$

### **Controllability and Observability** versus Zero-Pole Cancellation



$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} \lambda_1 & 0 \\ 0 & \lambda_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} b_1 \\ b_2 \end{bmatrix} u \qquad y = \begin{bmatrix} c_1 & c_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

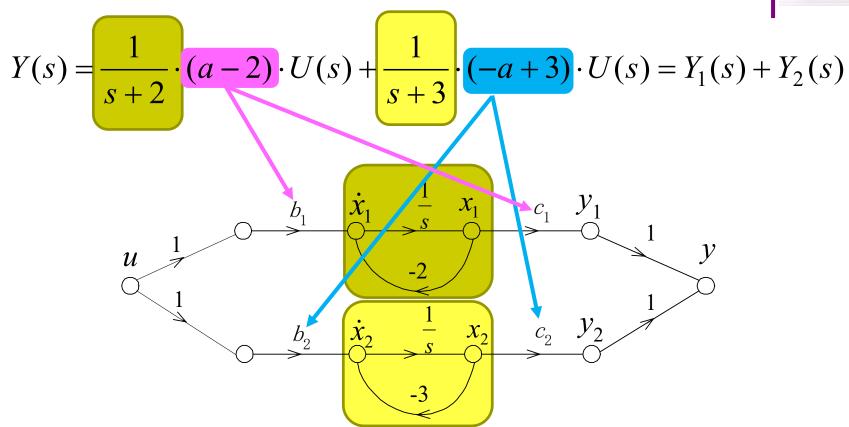


$$Y(s) = G(s)U(s) = \frac{a-2}{s+2}U(s) + \frac{-a+3}{s+3}U(s) = Y_1(s) + Y_2(s)$$

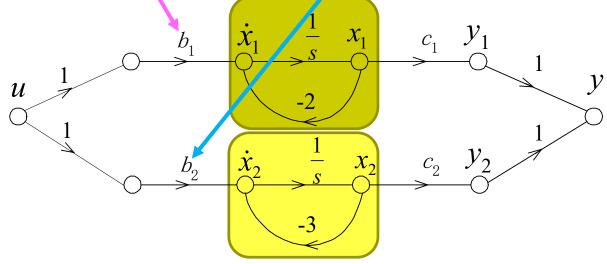
$$Y(s) = G(s)U(s) = \frac{a-2}{s+2}U(s) + \frac{-a+3}{s+3}U(s) = Y_1(s) + Y_2(s)$$
$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

# Controllability and Observability versus Zero-Pole Cancellation





$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$



If 
$$b_1 = a - 2$$
,  $b_2 = -a + 3$   $c_1 = 1$ ,  $c_2 = 1$ 

when  $a \neq 2$   $a \neq 3$ 

system is both controllable and observable

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -2 & 0 \\ 0 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} a-2 \\ -a+3 \end{bmatrix} u$$

$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

$$u = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

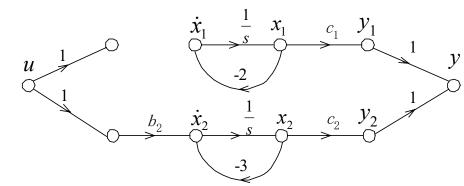
$$u = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

-3

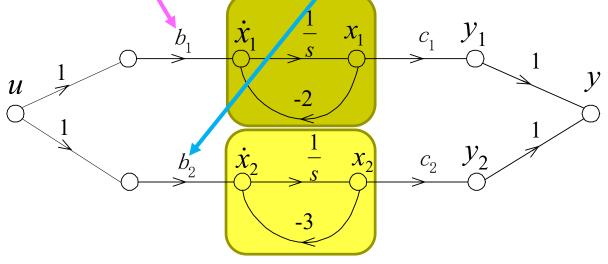
If 
$$b_1 = a - 2$$
,  $b_2 = -a + 3$   $c_1 = 1$ ,  $c_2 = 1$ 

$$a = 2$$

$$G(s) = \frac{1}{s + 3}$$
uncontrollable
$$\frac{\dot{x}_1 - \frac{1}{s} x}{\dot{x}_2 - \frac{1}{s}}$$



$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$



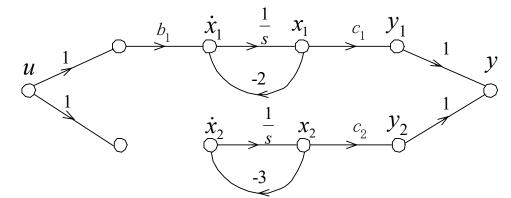
If 
$$b_1 = a - 2$$
,  $b_2 = -a + 3$   $c_1 = 1$ ,  $c_2 = 1$ 

$$a = 3$$

$$G(s) = \frac{1}{s+2}$$

$$\frac{\dot{x}_1}{s+2} = \frac{\dot{x}_2}{s}$$

uncontrollable



$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

$$u = \frac{b_1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + \frac{1}{s+3} \cdot U(s) = Y_1(s) +$$

If 
$$b_1 = 1$$
,  $b_2 = 1$   $c_1 = a - 2$ ,  $c_2 = -a + 3$ 

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -2 & 0 \\ 0 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 1 \end{bmatrix} u$$

$$y = \begin{bmatrix} a - 2 & -a + 3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

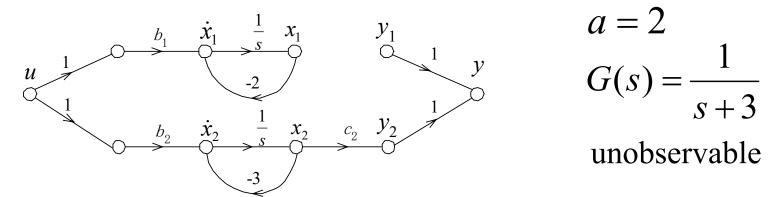
when  $a \neq 2$   $a \neq 3$ 

system is both controllable and observable

$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

$$u = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + \frac{1}{s+3} \cdot U(s) =$$

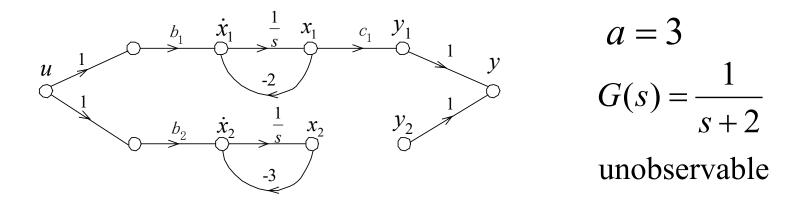
If 
$$b_1 = 1$$
,  $b_2 = 1$   $c_1 = a - 2$ ,  $c_2 = -a + 3$ 



$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

$$u = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + \frac{1}{s+3}$$

If 
$$b_1 = 1$$
,  $b_2 = 1$   $c_1 = a - 2$ ,  $c_2 = -a + 3$ 



$$Y(s) = \frac{1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + Y_2(s)$$

$$u = \frac{b_1}{s+2} \cdot (a-2) \cdot U(s) + \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_1(s) + \frac{b_1}{s+3} \cdot U(s) = Y_1(s) + \frac{b_$$

If choose 
$$b_1 = (a-2)$$
 and  $a = 2$ , then  $b_1 = 0$ 

$$Y(s) = \frac{1}{s+3} \cdot (-a+3) \cdot U(s) = Y_2(s)$$

and further 
$$c_1 = 0$$
,  $c_2 = 1$ 

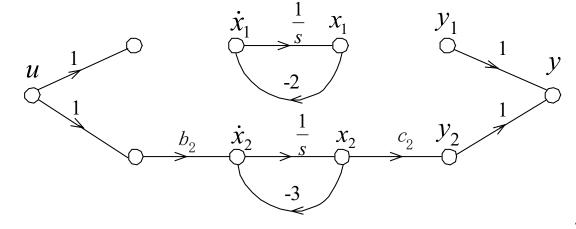
$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -2 & 0 \\ 0 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u \qquad y = \begin{bmatrix} 0 \\ 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

$$y = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$



$$G(s) = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} s+2 & 0 \\ 0 & s+3 \end{bmatrix}^{-1} \begin{bmatrix} 0 \\ 1 \end{bmatrix} = \frac{\begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} s+3 & 0 \\ 0 & s+2 \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix}}{(s+2)(s+3)}$$
$$= \frac{s+2}{(s+2)(s+3)} = \frac{1}{s+3}$$

neither controllable nor observable







 For a SISO system (A, B, C) to be controllable and observable, it is sufficient and necessary that its transfer function does not have zero-pole cancellation.

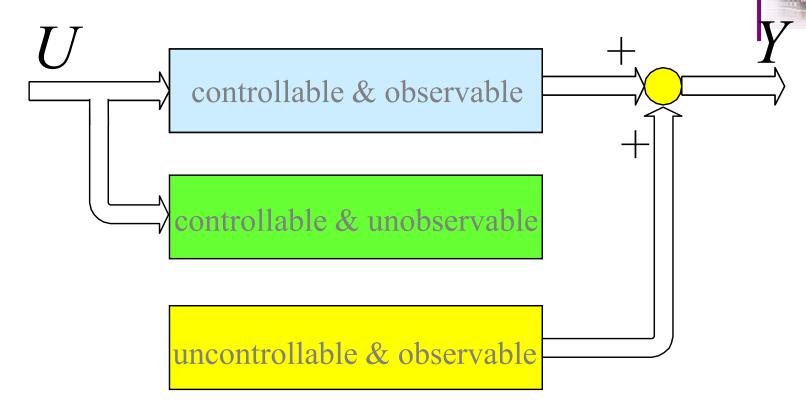
$$G(s) = C(sI - A)^{-1}B = \frac{C \cdot \operatorname{adj}(sI - A) \cdot B}{|sI - A|}$$



#### Remarks

 In the transfer function of a SISO system, a cancelled pole could relate to an uncontrollable, or an unobservable, or an uncontrollable and unobservable mode. The transfer function only depicts the controllable and observable modes.

# Controllability and Observability Decomposition



uncontrollable & unobservable

## Controllability and Observability Decomposition



For a LTI system (A, B, C), rank(S) = m < n, there must exist a linear non-singular transformation P, X=PX', which can transform the system into a new one (A',B',C') that explicitly comprise a controllable part and an uncontrollable part.

$$\begin{bmatrix} \dot{X}_1' \\ \dot{X}_2' \end{bmatrix} = \begin{bmatrix} A_{11}' & A_{12}' \\ 0 & A_{22}' \end{bmatrix} \begin{bmatrix} X_1' \\ X_2' \end{bmatrix} + \begin{bmatrix} B_1' \\ 0 \end{bmatrix} U$$

$$Y = \begin{bmatrix} C_1' & C_2' \end{bmatrix} \begin{bmatrix} X_1' \\ X_2' \end{bmatrix}$$

$$\dot{X}_{1}' = A_{11}' X_{1}' + A_{12}' X_{2}' + B_{1}' U$$



controllable states

$$\dot{X}_{2}' = A_{22}' X_{2}'$$



uncontrollable states

### **Controllability and Observability Decomposition**



How to construct the linear non-singular transformation P?

$$P = [p_1, \dots, p_m, p_{m+1}, \dots, p_n]$$

where:

$$[p_1, \cdots, p_m]$$

 $p_1, \dots, p_m$  are m linearly independent columns in S

$$p_{m+1}, \cdots, p_n$$

 $[p_{m+1}, \dots, p_n]$  are n-m linearly independent columns in  $\mathbb{R}^n$ , which are also independent of  $p_1, ..., p_m$ .

### **Example – 6.13**

$$\dot{X} = \begin{bmatrix} -2 & 1 & -1 \\ 0 & -2 & 0 \\ 1 & -1 & 0 \end{bmatrix} X + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$$

$$y = \begin{bmatrix} 1 & -1 & 1 \end{bmatrix} X$$

$$S = \begin{bmatrix} B & AB & A^2B \end{bmatrix} = \begin{bmatrix} 0 & -1 & 2 \\ 0 & 0 & 0 \\ 1 & 0 & 1 \end{bmatrix}$$
 rank $(S) = 2 < 3$ ,  
System is uncontrollable

$$\begin{bmatrix} 0 & -1 & 0 \\ 0 & -1 & 0 \end{bmatrix}$$

$$\operatorname{rank}(S) = 2 < 3,$$

$$P = \begin{bmatrix} 0 & -1 & 0 \\ 0 & 0 & 1 \\ 1 & 0 & 0 \end{bmatrix} \qquad P^{-1} = \begin{bmatrix} 0 & 0 & 1 \\ -1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix}$$

$$\dot{X}' = P^{-1}APX' + P^{-1}Bu$$

$$\dot{X}' = P^{-1}APX' + P^{-1}Bu$$

$$= \begin{bmatrix} 0 & 0 & 1 \\ -1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} -2 & 1 & -1 \\ 0 & -2 & 0 \\ 1 & -1 & 0 \end{bmatrix} \begin{bmatrix} 0 & -1 & 0 \\ 0 & 0 & 1 \\ 1 & 0 & 0 \end{bmatrix} X' + \begin{bmatrix} 0 & 0 & 1 \\ -1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} 0 & u \\ 0 & 1 \end{bmatrix}$$

$$\begin{bmatrix} \dot{x}_{1}' \\ \dot{x}_{2}' \\ \dot{x}_{3}' \end{bmatrix} = \begin{bmatrix} 0 & -1 & -1 \\ 1 & -2 & -1 \\ 0 & 0 & -2 \end{bmatrix} \begin{bmatrix} x_{1}' \\ x_{2}' \\ x_{3}' \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} u$$

$$y = CPX' = \begin{bmatrix} 1 & -1 & 1 \end{bmatrix} \begin{bmatrix} 0 & -1 & 0 \\ 0 & 0 & 1 \\ 1 & 0 & 0 \end{bmatrix} X' = \begin{bmatrix} 1 & -1 & -1 \end{bmatrix} \begin{bmatrix} x_1' \\ x_2' \\ x_3' \end{bmatrix}$$

## Controllability and Observability Decomposition



For a LTI system (A, B, C), rank(V) = m < n, there must exist a linear non-singular transformation Q, X=QX', which can transform the system into a new one (A',B',C') that explicitly comprise an observable part and an unobservable part.

$$\begin{bmatrix} \dot{X}_1' \\ \dot{X}_2' \end{bmatrix} = \begin{bmatrix} A_{11}' & 0 \\ A_{21}' & A_{22}' \end{bmatrix} \begin{bmatrix} X_1' \\ X_2' \end{bmatrix} + \begin{bmatrix} B_1' \\ B_2' \end{bmatrix} U \qquad Y = \begin{bmatrix} C_1' & 0 \end{bmatrix} \begin{bmatrix} X_1' \\ X_2' \end{bmatrix}$$

$$\dot{X}_1' = A_{11}' X_1' + B_1' U$$
 observable states

$$\dot{X}_{2}^{'} = A_{21}^{'} X_{1}^{'} + A_{22}^{'} X_{2}^{'} + B_{2}^{'} U \qquad \text{unobservable states}$$

$$Y = C_1' X_1'$$

# Controllability and Observability Decomposition



How to construct the linear non-singular transformation Q?

$$Q^{-1} = \left[ q_1^T, \dots, q_n^T, q_{m+1}^T, \dots, q_n^T \right]^T$$

where:

$$\begin{bmatrix} q_1^T, \dots, q_m^T \end{bmatrix}^T \text{ are m linearly independent rows in V}$$

$$\begin{bmatrix} q_{m+1}^T, \dots, q_n^T \end{bmatrix}^T \text{ are n-m linearly independent rows in R}^n, \\ \text{which are also independent of } q_1, \dots, q_m.$$

### **Example**

$$\dot{X} = \begin{bmatrix} 1 & 2 & -1 \\ 0 & 1 & 0 \\ 1 & -4 & 3 \end{bmatrix} X + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$$



$$\begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 1 & -1 & 1 \\ 2 & -3 & 2 \end{bmatrix}$$

$$y = \begin{bmatrix} 1 & -1 & 1 \end{bmatrix} X$$

$$V = \begin{bmatrix} C \\ CA \\ CA^2 \end{bmatrix} = \begin{bmatrix} 1 & -1 & 1 \\ 2 & -3 & 2 \\ 4 & -7 & 4 \end{bmatrix}$$

$$\operatorname{rank}(V) = 2 < 3,$$

System is unobservable

$$Q^{-1} = \begin{bmatrix} 1 & -1 & 1 \\ 2 & -3 & 2 \\ 0 & 0 & 1 \end{bmatrix} \qquad Q = \begin{bmatrix} 3 & -1 & -1 \\ 2 & -1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$Q = \begin{vmatrix} 3 & -1 & -1 \\ 2 & -1 & 0 \\ 0 & 0 & 1 \end{vmatrix}$$

$$\dot{X}' = Q^{-1}AQX' + Q^{-1}Bu$$

$$X = Q^{T}AQX + Q^{T}Bu$$

$$= \begin{bmatrix} 1 & -1 & 1 \\ 2 & 3 & 2 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 1 & 2 & -1 \\ 0 & 1 & 0 \\ 1 & -4 & 3 \end{bmatrix} \begin{bmatrix} 3 & -1 & -1 \\ 2 & -1 & 0 \\ 0 & 0 & 1 \end{bmatrix} X' + \begin{bmatrix} 1 & -1 \\ 2 & 3 & 2 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 0 & u \\ 1 & -1 \end{bmatrix}$$

$$\begin{bmatrix} \dot{x}_1' \\ \dot{x}_2' \\ \dot{x}_3' \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ -2 & 3 & 0 \\ -5 & 3 & 2 \end{bmatrix} \begin{bmatrix} x_1' \\ x_2' \\ x_3' \end{bmatrix} + \begin{bmatrix} 1 \\ 2 \\ 1 \end{bmatrix} u$$

$$y = CQX' = \begin{bmatrix} 1 & -1 & 1 \end{bmatrix} \begin{bmatrix} 3 & -1 & -1 \\ 2 & -1 & 0 \\ 0 & 0 & 1 \end{bmatrix} X' = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x'_1 \\ x'_2 \\ x'_3 \end{bmatrix}$$

### Controllability and Observability Decomposition

If a LTI system (A, B, C) is neither controllable nor observable, there must exist a linear non-singular transformation P, X=PX', which can transform the system into a new one (A',B',C') that has the following form:

$$\begin{bmatrix} \dot{x}_{1}' \\ \dot{x}_{2}' \\ \dot{x}_{3}' \\ \dot{x}_{4}' \end{bmatrix} = \begin{bmatrix} A_{11}' & 0 & A_{13}' & 0 \\ A_{21}' & A_{22}' & A_{23}' & A_{24}' \\ 0 & 0 & A_{33}' & 0 \\ 0 & 0 & A_{43}' & A_{44}' \end{bmatrix} \begin{bmatrix} x_{1}' \\ x_{2}' \\ x_{3}' \\ x_{4}' \end{bmatrix} + \begin{bmatrix} B_{1}' \\ B_{2}' \\ 0 \\ 0 \end{bmatrix} U$$

$$Y = \begin{bmatrix} C_1' & 0 & C_3' & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2' \\ x_3' \\ x_4' \end{bmatrix}$$



Controllable & observable



Controllable & unobservable



Uncontrollable & observable



Uncontrollable & unobservable

### State-Feedback Control

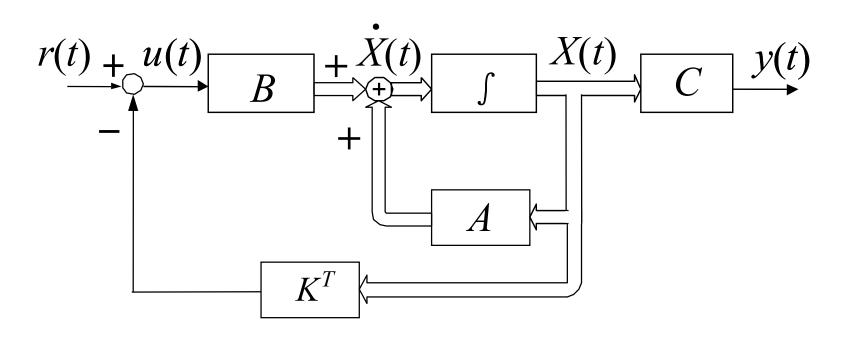
Instead of using controllers with fixed configurations in the forward or feedback path, control is achieved by feeding back the state variables through real constant gains.

SISO: X = AX + Bu

y = CX

state feedback gain

If choose:  $u = r - K^T X$  where  $K^T = [k_1, k_2, \dots, k_n]$ 



### Pole-Placement Design through State Feedback

Knowing the relation between the closed-loop poles and the system performance, we can effectively carry out the design by specifying the location of these poles.

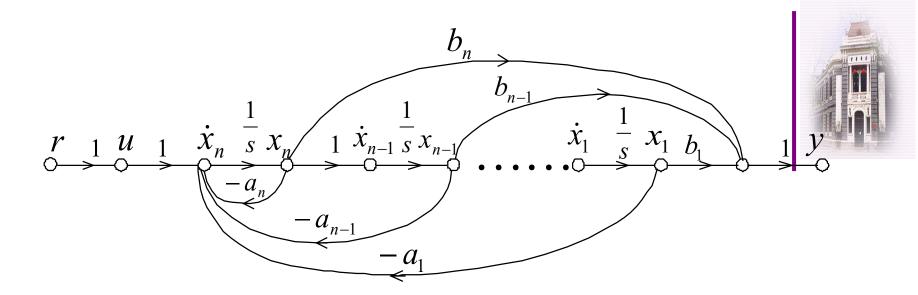
Theorem: For a SISO system (A,B,C), if its poles can be arbitrarily placed, it is a sufficient and necessary condition that the system is controllable.

#### Proof:

sufficiency

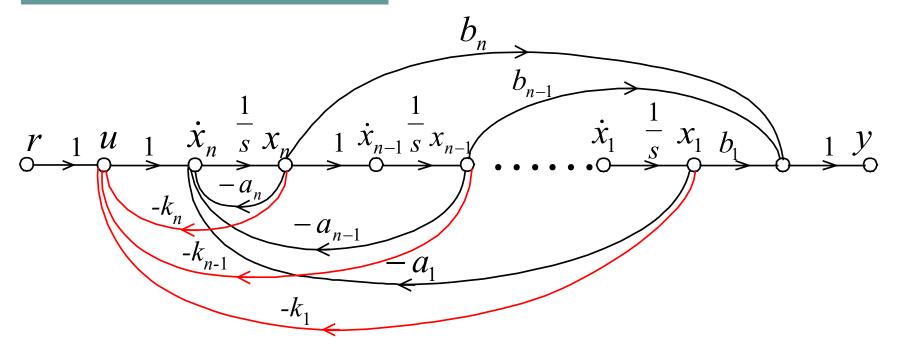
If a system is controllable, it can be transformed into CCF

$$\dot{X} = \begin{bmatrix}
0 & 1 & 0 & \cdots & 0 \\
0 & 0 & 1 & \cdots & 0 \\
\vdots & \vdots & \vdots & \ddots & \vdots \\
0 & 0 & 0 & \cdots & 1 \\
-a_1 & -a_2 & -a_3 & \cdots & -a
\end{bmatrix} X + \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} u \quad y = \begin{bmatrix} b_1 & b_2 & \cdots & b_n \end{bmatrix} X$$



### Introducing state feedback:

$$u = r - K^T X$$



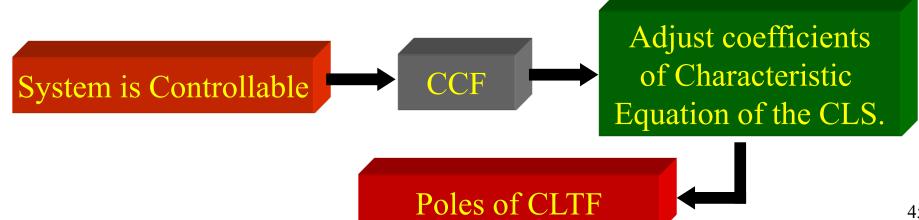
$$A - BK^{T} = A - \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 1 \end{bmatrix} \begin{bmatrix} k_{1} & k_{2} & \cdots & k_{n} \end{bmatrix} = A + \begin{bmatrix} 0 & 0 & \cdots & 0 \\ \vdots & \vdots & \cdots & 0 \\ 0 & 0 & \cdots & 0 \\ -k_{1} & -k_{2} & \cdots & -k_{n} \end{bmatrix}$$

$$\dot{X} = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \cdots & 1 \\ -k_1 - a_1 & -k_2 - a_2 & -k_3 - a_3 & \cdots & -k_n - a_n \end{bmatrix} X + \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} r$$

$$G(s) = \frac{\frac{b_n}{s} + \frac{b_{n-1}}{s^2} + \dots + \frac{b_1}{s^n}}{1 + \frac{k_n}{s} + \frac{a_n}{s} + \frac{k_{n-1}}{s^2} + \frac{a_{n-1}}{s^2} \dots + \frac{k_1}{s^n} + \frac{a_1}{s^n}}$$

$$G(s) = \frac{Y(s)}{R(s)} = \frac{b_n s^{n-1} + b_{n-1} s^{n-2} + \dots + b_1}{s^n + (a_n + k_n) s^{n-1} + \dots + (a_1 + k_1)}$$

n arbitrary poles can determine n coefficients.





#### Necessity:

If the system is uncontrollable, it must have uncontrollable states. The poles corresponding to these states can't be arbitrarily placed.

$$\begin{bmatrix} \dot{X}_1' \\ \dot{X}_2' \end{bmatrix} = \begin{bmatrix} A_{11}' & A_{12}' \\ 0 & A_{22}' \end{bmatrix} \begin{bmatrix} X_1' \\ X_2' \end{bmatrix} + \begin{bmatrix} B_1' \\ 0 \end{bmatrix} U$$

U can't affect  $X_2$ 

# How to determine the Feedback Gain

Method of undetermined coefficient:

Step 1: Substitute  $u = r - K^T X$  into the state equation

$$\dot{X} = AX + B(r - K^T X) = (A - BK^T)X + Br$$

Characteristic equation:

$$\Delta(s) = |sI - (A - BK^T)| = 0$$

Step 2: Calculate the characteristic equation of the desired system

$$\Delta'(s) = (s + p_1)(s + p_2) \cdots (s + p_n) = s^n + a_n s^{n-1} + \cdots + a_2 s + a_1 = 0$$

Compare the two characteristic equations:

$$|sI - (A - BK^{t})| = s^{n} + a_{n}s^{n-1} + \dots + a_{2}s + a_{1}$$

# **Example – 6.14**

Q: Giving a system with the following state equation, please find the feedback gain that can make the poles of the modified system at  $-2 \pm j2$ 

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u$$

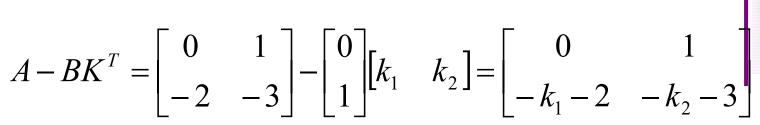
A: check the controllability of the original system

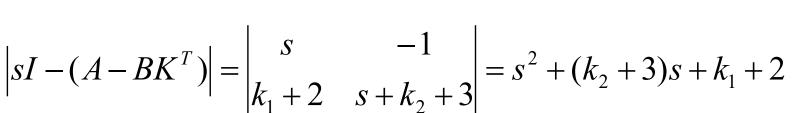
$$S = \begin{bmatrix} B & AB \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 1 & -3 \end{bmatrix}$$

$$rank(S) = 2$$

system is controllable, so its poles can be arbitrarily placed.

The characteristic equation after introducing state feedback:





The characteristic equation of the desired system:

$$(s+2-j2)(s+2+j2) = s^2 + 4s + 8$$

Compare the above two equations, we get:

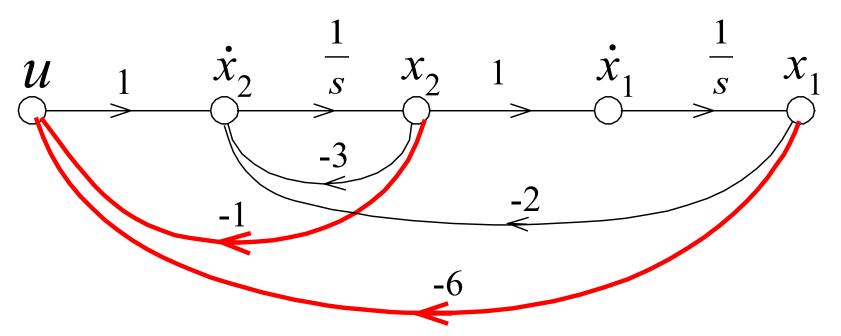
$$k_1 = 6, \quad k_2 = 1$$

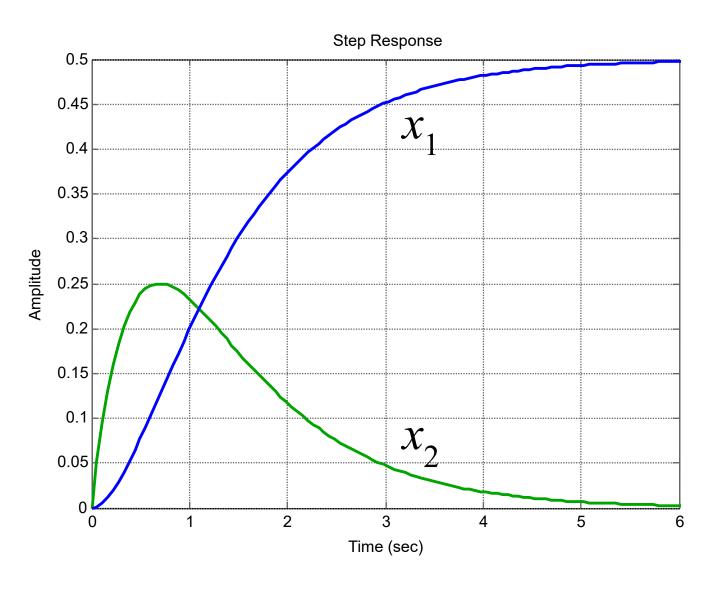
The original system:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u$$

The state feedback gain:

$$K^T = [6, 1]$$

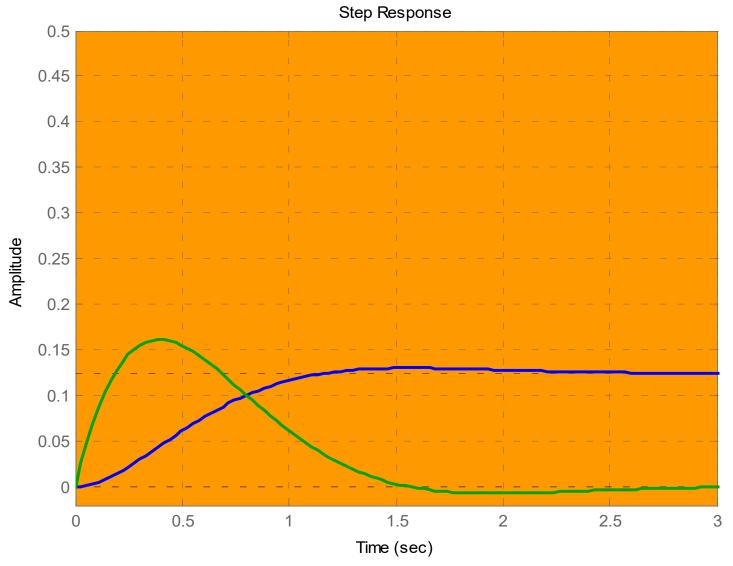






step(g1,g2)





# **Example – 6.15**

Q: Given a system with the following state equation, please find a feedback gain that can make the poles of the modified system at -2,  $-1 \pm j$ 

$$\dot{X} = \begin{bmatrix} 0 & 0 & 0 \\ 1 & -6 & 0 \\ 0 & 1 & -12 \end{bmatrix} X + \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} u$$

A: check the controllability of the original system

$$S = [B, AB, A^{2}B] = \begin{vmatrix} 1 & 0 & 0 \\ 0 & 1 & -6 \\ 0 & 0 & 1 \end{vmatrix}$$

$$rank(S) = 3$$

system is controllable, so its poles can be arbitrarily placed.

The characteristic equation after introducing state feedback:

$$A - BK^{T} = \begin{bmatrix} 0 & 0 & 0 \\ 1 & -6 & 0 \\ 0 & 1 & -12 \end{bmatrix} - \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} \begin{bmatrix} k_{1} & k_{2} & k_{3} \end{bmatrix} = \begin{bmatrix} -k_{1} & -k_{2} \\ 1 & -6 \\ 0 & 1 & -12 \end{bmatrix}$$

$$\begin{vmatrix} sI - (A - BK^{T}) \end{vmatrix} = \begin{vmatrix} s + k_{1} & k_{2} & k_{3} \\ -1 & s + 6 & 0 \\ 0 & -1 & s + 12 \end{vmatrix}$$
$$= s^{3} + (k_{1} + 18)s^{2} + (18k_{1} + k_{2} + 72)s + 72k_{1} + 12k_{2} + k_{3}$$

The characteristic equation of the desired system:

$$(s+2)(s+1-j)(s+1+j) = s^3 + 4s^2 + 6s + 4$$



Compare the above two equations, we get:

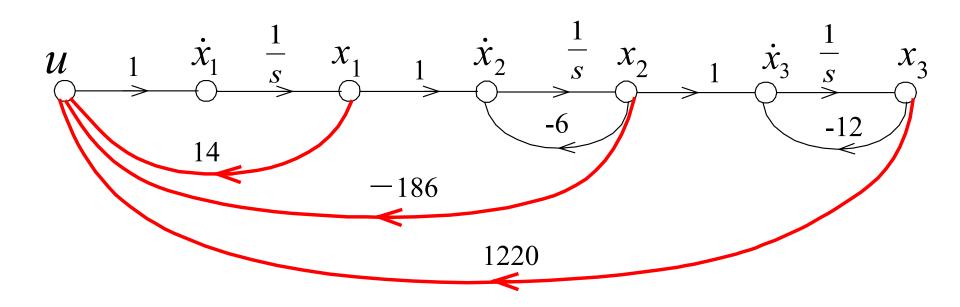
$$k_1 + 18 = 4$$
  
 $18k_1 + k_2 + 72 = 6$   
 $72k_1 + 12k_2 + k_3 = 4$ 

$$k_1 = -14$$
 $k_2 = 186$ 
 $k_3 = -1220$ 

The original system:

$$\dot{X} = \begin{bmatrix} 0 & 0 & 0 \\ 1 & -6 & 0 \\ 0 & 1 & -12 \end{bmatrix} X + \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} u$$

The state feedback gain:  $K^T = \begin{bmatrix} -14 & 186 & -1220 \end{bmatrix}$ 





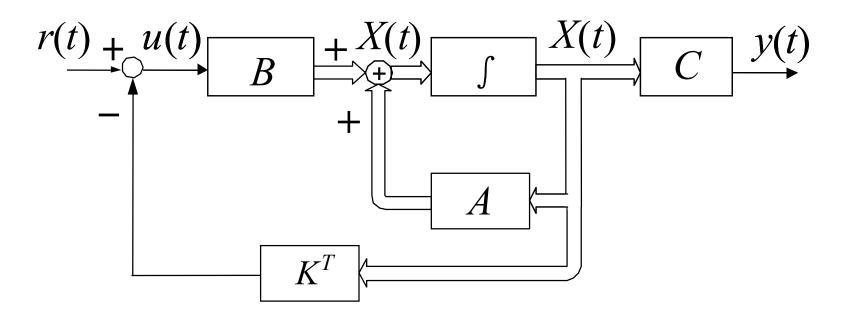
### Can state feedback change the controllability of a system?

- A Yes
- B No

### State Feedback and Controllability



State feedback can not change the controllability of a system



$$\dot{X} = AX + Bu$$

$$y = CX$$

$$u = r - K^{T}X$$

$$\dot{X} = AX + B(r - K^{T}X)$$
$$= (A - BK^{T})X + Br$$

The controllability matrix of the original system:

$$S = [B \quad AB \quad \cdots \quad A^{n-1}B]$$



The controllability matrix of the compensated system:

$$S' = [B \quad (A - BK^T)B \quad \cdots \quad (A - BK^T)^{n-1}B]$$

$$(A - BK^{T})B = AB - B \cdot \underline{\underline{K}^{T}B}$$

$$(A - BK^{T})^{2}B = (A^{2} - ABK^{T} - BK^{T}A + BK^{T}BK^{T})B$$
$$= A^{2}B - AB\underline{\underline{K}^{T}B} - B\underline{\underline{K}^{T}AB} + B\underline{\underline{K}^{T}B}\underline{\underline{K}^{T}B}\underline{\underline{K}^{T}B}$$

Notice that the double underscored is real numbers. So each column of S' can be linearly represented by the columns of S. Therefore:

$$rank(S) = rank(S')$$



#### Can state feedback change the observability of a system?

- A Yes
- B No

# State Feedback and Observability



#### Can state feedback change the observability of a system?

Because the system poles can be arbitrarily placed through state feedback, when a replaced pole cancels a zero, the system observability will be changed.



#### Will state feedback change the order of a system?

- A Yes
- B No

### State Feedback and Observability

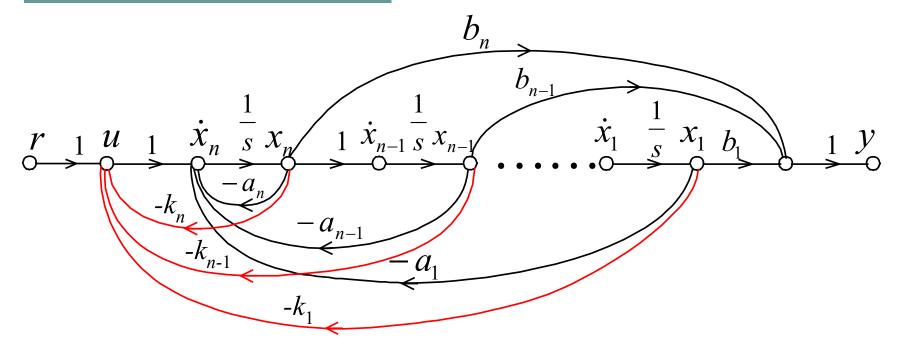


Will state feedback change the order of a system?

State feedback will not change the order of a system.

Introducing state feedback:

$$u = r - K^T X$$



# **Output Feedback**

Control is achieved by feeding back the outputs through real constant gains.

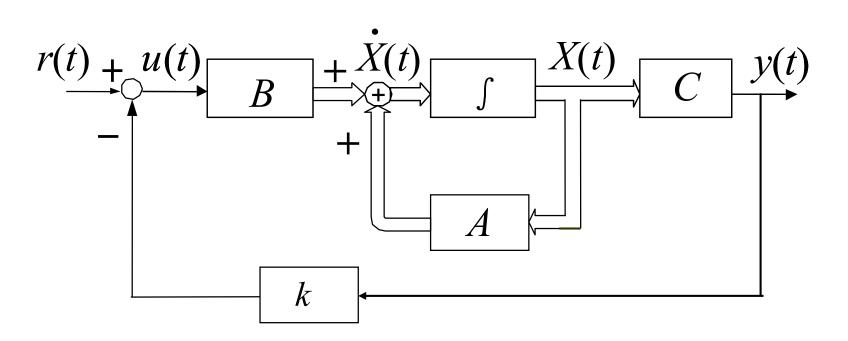
SISO:  $\dot{X} = AX + Bu$ 

$$y = CX$$

output feedback gain

If choose: 
$$u = r - ky$$

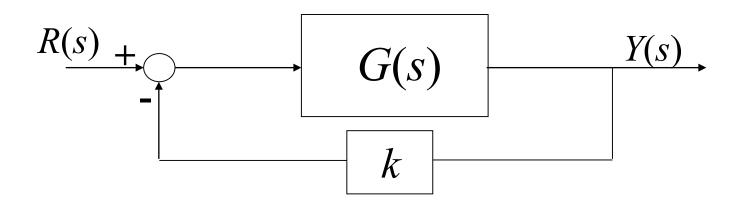
where 
$$k$$







$$\dot{X} = AX + B(r - ky) = AX + Br - BkCX$$
$$= (A - BkC)X + Br$$



### General Conclusions about Output Feedback

If a SISO system is both controllable and observable, after introducing the output feedback, the new system is also controllable and observable.

When there is not output feedback: 
$$G(s) = \frac{N(s)}{D(s)}$$

Because the system is both controllable and observable, there is no common factor in numerator and denominator.

When there is output feedback: 
$$M(s) = \frac{G(s)}{1 + kG(s)} = \frac{\frac{N(s)}{D(s)}}{1 + k\frac{N(s)}{D(s)}} = \frac{N(s)}{D(s) + kN(s)}$$

there is no common factor in numerator and denominator either. So, there is no zero-pole cancellation.

### General Conclusions about Output Feedback

If a SISO system is uncontrollable (unobservable), after introducing the output feedback, the new system is still uncontrollable (unobservable).

There exists zero-pole cancellation in  $\frac{Cadj(sI-A)B}{|sI-A|}$ 

When there is output feedback:

$$\frac{Cadj(sI - A)B}{\left|sI - A\right|} = \frac{Cadj(sI - A)B}{\left|sI - A\right|} = \frac{Cadj(sI - A)B}{\left|sI - A\right| + kCadj(sI - A)B}$$

The common factors don't change, zero-pole cancellation does not disappear.





- Definition of observability
- Observability condition and its demonstration
- Observability canonical form and its observability
- Controllability and Observability versus Zero-Pole Cancellation
- Controllability and Observability Decomposition
- State-feedback control
- Output-feedback control

# **Assignment**

Page 150

Page 150~151

6

9, (1)



