

Radiation- Hydrodynamics with MPI-AMVRAC: Massive-Star Atmospheres and Winds

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Preface

Thank you everybody

Summary

Something something modeling stellar winds

Summary in layman's terms

Somethin something shiny point in sky

List of abbreviations and symbols

Abbreviations

HD	Hydrodynamics
RHD	Radiohydrodynamics
LTE	local thermal equilibrium
CAK	Castor, Abott and Klein
FLD	Flux limited diffusion
MPI	message passing interface
AMR	adaptive mesh refinement
VAC	versatile advection code
PDE	partial differential equations
BH	black hole
NS	neutron star
WD	white dwarf

General

c	speed of light
M_{\odot}	solar mass
R_{\odot}	solar radius
L_{\odot}	solar luminosity

Hydrodynamics

ρ	density
\vec{v}	velocity
e	gas energy density
p	gas pressure
γ	adiabatic index
a_{adiab}	adiabatic soundspeed
T	gas temperature
∇	gradient
$\vec{\nabla}$	divergent
$S_{\rho, \vec{v}, e}$	sourceterms for HD-equations
g_{grav}	gravitational acceleration
g_{rad}	radiative acceleration

Radiation hydrodynamics

Γ	Eddington factor
ν	frequency
Ω	solid angle
κ_{ν}	opacity at frequency ν
κ	Rosseland mean opacity
χ_{ν}	absorption coefficient at frequency ν
η_{ν}	emission coefficient at frequency ν
\vec{n}	unit vector along line of sight
S_{ν}	source function
B_{ν}	Planck function
I_{ν}, I	frequency dependent and integrated intensity
J_{ν}, J	mean intensity
E_{ν}, E	radiation energy density
\vec{F}_{ν}, \vec{F}	radiation flux
H_{ν}, H	1 st moment of intensity
P_{ν}, P	radiation pressure
K_{ν}, K	2 nd moment of intensity
H_{eff}	scale height
g_{eff}	effective acceleration
τ	optical depth

CAK and FLD

l_{sob}	sobolev length
g_{line}	line acceleration
κ_e	electron opacity
q	
t	
λ	flux limiter
l_{γ}	photon mean free path
R	ratio of l_{γ} and radiation scale height
D	diffusion coefficient

Numerics

Δx	cell width in the x -direction
$\Delta t, dt$	time step
N, M	dimension of grid
\tilde{x}	quantity x in dimensionless units
x_0	the normalised value of quantity x
$x_{i,j}^n$	quantity x at timestep n in cell i, j

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Chapter 1

Introduction

High mass stars are the drivers of dynamical and chemical evolutions of galaxies throughout the universe, including our own Milky Way. Massive stars live short but exciting lives, finally ending in giant supernova explosions. After their death, they leave behind exotic remnants such as black holes (BH) or neutron stars (NS). Whether a massive star ends as either NS or BH depends a lot on the amounts of mass expelled during their lifetime due to stellar winds. Research in stellar winds and atmospheres of massive stars will give clues about the unexpected BH mass distribution observed by the LIGO collaboration [1], to which data is being added as we speak. This thesis focusses on making radiation-hydrodynamical models of the highly structured stellar winds and atmospheres of massive stars. Next to modelling stellar winds and atmospheres, a main result of this thesis are computer codes written to be used with mpi-AMRVAC which will be used in research on radiation dominated processes.

In various systems it is important to take into account a radiation field when considering gas dynamics ([10], ...). Radiation can exert forces to, for example, accelerate stellar winds and transport energy like in the non-convective stellar radiation zones. The equations describing gas dynamics with a radiation field will evolve on two different timescales, a radiation field generally evolves much faster than gas, so solving together them isn't trivial.

The importance of good models for radiation hydrodynamics is clear in a multitude of astrophysical systems, a few are listed below. An important concept in these sort of systems is the Eddington factor Γ , this is the ratio between the radiative and gravitational acceleration.

$$\Gamma = \frac{\kappa F_r}{c g_{grav}} \quad (1.1)$$

Where F_r is the radiation energy flux component parallel to the gravitational field, κ the absorption opacity, c the speed of light in vacuum and g_{grav} the magnitude of the local gravitational acceleration. If the Eddington factor describing a system gets bigger, the importance for taking into account radiation grows as well.

- **stellar winds**

Stellar winds can be subdivided in three types: pressure driven for solar type stars,

dust driven for cool massive stars and radiation driven for hot massive stars. The driving force behind these radiation driven winds is, as the name suggests, radiation force. Outwards travelling photons get absorbed by atomic line transitions and the photon's momentum is transferred to the ion or atom. Hot massive stars can undergo mass loss of $\sim 10^{-6} M_{\odot}/\text{yr}$ due to radiation driven winds, this of course will have an impact on their evolution.

Stars with a luminosity of $10^6 L_{\odot}$ and a stellar mass of $80 M_{\odot}$ will have an Eddington factor $\Gamma \sim 50$ near the surface, radiation plays an essential role in their dynamics. CAK-theory is an analytical model describing these winds, and the first half of this thesis was spent on modelling this.

These dust driven winds are also driven by radiation, however in contrast to the hot stellar radiative winds, the opacity source in the system is caused by the dust instead of atomic absorption lines or electron scattering.

- **stellar atmospheres**

In stellar atmospheres like that from our sun, the radiation field is directly responsible for setting the temperature structure. Energy created by fusion processes in the stellar core is transported outward by the radiation field. The atmosphere is said to be in radiative equilibrium, this means that the radiation energy absorbed by the atmospheric gas is equal to the energy that is radiated.

- **stellar or galactic disks**

Accretion disks appear in all sorts of objects: young stellar objects, pre-main sequence objects, various types of stars, around BH's, NS's, active galactic nuclei (AGN) [9] and so on. The gas in these disks is often heated due to internal collisions or accretion on the central object, which gives rise to a radiation field. This radiation field can highly impact the shape and energy distribution within the disk due to ablation and radiative heating [3], [7]. In turn, depending on the type of disk, this can have an impact on the end products of star and planetary formation. AGN disks also drive winds on the surface of their disks.

- **star formation**

During the collapse of interstellar gas clouds during the process of star formation, energy transport due to radiation has effects on the formation process of pre-stellar cores [2]

- **very massive stars**

- **interstellar medium**

The interstellar medium (ISM) is home to numerous types of gas clouds. A prominent example of a regime where radiation plays an important role are so called HII regions. These hydrogen clouds surrounding hot massive stars are ionised by the stellar UV-photons. Due to the radiative heating of the gas, these clouds are ever expanding [?], [4].

- **Eruptive variables**

Systems that exceed the Eddington limit ($\Gamma > 1$) are called super-Eddington systems. A prime example for such a system is η Carinae. η Carinae is what is known

as an eruptive variable. It is a star which underwent an enormous mass loss in the middle of the 19th century, losing $10M_{\odot}$ over the course of 10 years. One of the hypothesised drivers of this huge mass loss is radiation.

This thesis describes radiation hydrodynamics (RHD) in two different regimes: CAK-theory in hot stellar winds and flux limited diffusion in stellar atmospheres. Methods used and codes developed in this project are applicable in a very broad range of scenarios. Before we move to the mathematical framework of radiative gasses, let us first have a short review of the equations describing gasses and liquids.

1.1 Hydrodynamics

The movement of non-isothermal, isotropic gasses and fluids, in the absence of magnetic fields and radiation, can be described by ideal hydrodynamics. The hydrodynamical equations describe conservation of mass, conservation of momentum and conservation of energy in the following equations:

$$\partial_t (\rho) + \vec{\nabla} \cdot (\rho \vec{v}) = S_{\rho} \quad (1.2)$$

$$\partial_t (\rho \vec{v}) + \vec{\nabla} \cdot (\vec{v} \rho \vec{v} + p) = S_{\rho \vec{v}} \quad (1.3)$$

$$\partial_t (e) + \vec{\nabla} \cdot (\vec{v} e + \vec{v} p) = S_e \quad (1.4)$$

These three partial differential equations are called the continuity equation (describing mass conservation), the momentum equation (describing momentum conservation) and the energy equation (describing energy conservation). Above, the equations were written in their conservative form, they all have the same shape:

$$\partial_t \underbrace{u}_{\text{density}} + \vec{\nabla} \cdot \overbrace{\vec{f}_u}^{\text{density flux}} = \underbrace{S_u}_{\text{source term}} \quad (1.5)$$

There are only 3 PDE's describing 4 primitive variables: mass density ρ , velocity \vec{v} , gas energy density e and gas pressure p , $\rho \vec{v}$ is the momentum density. This means that there is need for an additional closure relation, an equation of state like for example the ideal gas law:

$$p = (\gamma - 1) \left(e - \frac{\rho \vec{v}^2}{2} \right) \quad (1.6)$$

Where γ is the adiabatic index.

The source terms in the equations describe adding or subtracting to one of the densities: S_{ρ} describes mass being added or subtracted from the system, $S_{\rho \vec{v}}$ describes external forces such as gravity or radiative forces and S_e describes work exerted on the system, this can be for example due to heating or cooling of the fluid by a radiation field. A good

description of radiation hydrodynamics needs to formulate the correct source terms for the momentum equation (1.3) and gas energy equation (1.4). These source terms can depend on the other primitive variables ρ , \vec{v} and e as well as free parameters describing the source of the radiation field such as the luminosity and mass of a central star.

Predicting fluid dynamics means solving these aforementioned equations simultaneously, this is done using numerical computer codes such as `mpi-AMRVAC` [8]. Solving these equations is a tricky business, and there are numerous codes and numerical schemes available. The Evolution of a simulation will depend not only on both initial and boundary conditions, but also on which physics taken into consideration (sourceterms) and even a little on which numerical methods are used, more on this in chapter 2.1.

In section ??, an analytic expression for these source terms are calculated to simulate the conditions in line driven stellar winds. This calculation was first done by Castor, Abbott and Klein [?] CITE HERE and carries their initials in its name: CAK-theory.

1.2 Radiation Hydrodynamics

The last section gave a small recap of hydrodynamics. In this section, a mathematical framework is set up to combine the hydrodynamics equations with the effects of a time dependent radiation field. At the end of this we will be left with a system of partial differential equations describing the dynamics of both the gas and radiation field.

The main governing equation describing the evolution of radiation through a medium is the radiative transfer equation:

$$\left(\frac{1}{c} \frac{\partial}{\partial t} + \vec{n} \cdot \vec{\nabla} \right) I_\nu = \eta_\nu + \chi_\nu I_\nu \quad (1.7)$$

Where I_ν is the intensity at frequency ν along the direction of unit vector \vec{n} . η_ν And χ_ν are the emissivity and total opacity. This equation describes how the value of the intensity I_ν changes when propagating through a medium with given emissivity and opacity. The literature can be confusing and inconsistent when handling the opacity and the absorption coefficient. So to avoid any confusion: the absorption coefficient χ , generally given in units of $\frac{1}{cm}$ is the product of the opacity κ given in $\frac{cm^2}{g}$ and the density ρ .

Equation (1.7) is known as the 0th order of the radiative transfer equation. The 1st order radiative transfer equations are obtained by integrating over all solid angles and dividing by 4π , the 2nd order one by first multiplying with \vec{n} before doing the integration.

$$\frac{1}{c} \frac{\partial J_\nu}{\partial t} + \vec{\nabla} \cdot \vec{H}_\nu = \frac{1}{4\pi} \int_\Omega \eta_\nu + \chi_\nu I_\nu d\Omega \quad (1.8)$$

$$\frac{1}{c} \frac{\partial \vec{H}_\nu}{\partial t} + \vec{\nabla} \cdot \vec{K}_\nu = \frac{1}{4\pi} \int_\Omega (\eta_\nu + \chi_\nu I_\nu) \vec{n} d\Omega \quad (1.9)$$

Where J_ν , \vec{H}_ν and \vec{K}_ν are the higher order moments of the intensity. J_ν is the mean intensity, which is I_ν averaged over all solid angles and thus a scalar. \vec{H}_ν Is a vector and

K_ν is a tensor of order 2. The following relations exist between the moments of intensity and some more physical quantities:

$$E_\nu = \frac{4\pi}{c} j_\nu = \frac{1}{c} \int_{\Omega} I_\nu d\Omega \quad (1.10)$$

$$\vec{F}_\nu = \frac{4\pi}{c} \vec{H}_\nu = \frac{1}{c} \int_{\Omega} I_\nu \vec{n} d\Omega \quad (1.11)$$

$$P_\nu = \frac{4\pi}{c} K_\nu = \frac{1}{c} \int_{\Omega} I_\nu \vec{n}^2 d\Omega \quad (1.12)$$

These are the radiative energy E_ν , the radiation flux \vec{F}_ν and the radiation pressure tensor P_ν . When assuming an isotropic radiation field, just like with the gas pressure tensor, P_ν can be written as a scalar times the unit tensor. The fraction of emissivity and total opacity is often written as the source function $S_\nu = \frac{\eta_\nu}{\chi_\nu}$.

When doing full radiative transfer, these equations are solved for an enormous amount of frequencies, corresponding to the indices ν . The situation is simplified when assuming a frequency independent absorption and emission. This simplifies the equations because the radiation quantities ($I_\nu, \vec{F}_\nu, E_\nu, \dots$) can now be integrated over all frequencies, dropping their index. The LTE, frequency integrated radiation energy and radiation flux equations can be written in a conservative form, similar to the hydrodynamics equations:

$$\frac{\partial E}{\partial t} + \vec{\nabla} \cdot \vec{F} = \int_{\nu} \int_{\Omega} \chi_\nu (S_\nu + I_\nu) d\nu d\Omega \quad (1.13)$$

$$\frac{\partial \vec{F}}{\partial t} + c^2 \vec{\nabla} P = \int_{\nu} \int_{\Omega} \chi_\nu (S_\nu + I_\nu) \vec{n} d\nu d\Omega \quad (1.14)$$

Assuming LTE, the source function is equal to the Planck function $S_\nu = B_\nu(T)$ and both $B_\nu(T)$ and χ_ν are independent of solid angle. The relations between flux, energy and the moments of intensity can be used again to replace I_ν in the source terms. Because of symmetry, $S_\nu \vec{n}$ integrated over the total solid angle will return $\vec{0}$ in the source term of the flux equation.

The absorption coefficient is the product of the gas density and the frequency dependent opacity: $\chi_\nu = \rho \kappa_\nu$. To make life easier, an Energy-meaned and Plack opacity can be defined, and in first approximation they are equal to the Rosseland mean opacity (SOURCE?!?!?) which is meaned over the radiative flux:

$$\frac{\int_{\nu} E_\nu \kappa_\nu d\nu}{\int_{\nu} E_\nu d\nu} = \frac{\int_{\nu} B_\nu \kappa_\nu d\nu}{\int_{\nu} B_\nu d\nu} = \kappa = \frac{\int_{\nu} \vec{F}_\nu \kappa_\nu d\nu}{\int_{\nu} \vec{F}_\nu d\nu} \quad (1.15)$$

Integrating the left hand sides of equations 1.13 and ?? over frequency and solid angle and replacing opacities by the Rosseland mean opacity leads to:

$$\frac{\partial E}{\partial t} + \vec{\nabla} \cdot \vec{F} = 4\pi \kappa \rho B(T) - c \kappa \rho E \quad (1.16)$$

$$\frac{\partial \vec{F}}{\partial t} + c^2 \vec{\nabla} P = c \kappa \rho \vec{F} \quad (1.17)$$

These equations are written here in the co-moving frame. Transforming to a static frame, the same frame as used in aforementioned hydro equations, is done in [6] by complicated Lorentz transformations. Effectively an advection term is added to the density flux in both equations, $\vec{v}E$ and $\vec{v} \cdot \vec{F}$. The two source terms in the radiation energy equation are interpreted as energy exchange between the gas and the radiation field, energy leaving the radiation field heats the gas and gas is cooled by energy entering the radiation field. These so called heating and cooling source terms, $4\pi\kappa\rho B(T)$ and $c\kappa\rho E$, must be added to the gas energy equation as well. In a steady state radiative equilibrium regime ($B = J = \frac{c}{4\pi}E$), these terms cancel each other out.

Another source term in the radiative energy equation that needs to be added is the photon tiring term $\vec{\nabla} \cdot \vec{v}P$. This term expresses the work done by the photons in the radiation field to accelerate the gas. It is essentially the radiation energy equivalent of the momentum radiation force term. The origin of this term is also explained more carefully in [6].

The source term in the Flux energy equation has the same units as the gas momentum source term, this is the expression for radiation force. If momentum leaves the radiation flux there is work being done, this must also mean that there is energy leaving the radiation field. This is translated in the photon tiring term $\vec{\nabla} \cdot \vec{v}P$, which must be subtracted from the radiation energy source term.

$$\partial_t(\rho) + \vec{\nabla} \cdot (\rho\vec{v}) = 0 \quad (1.18)$$

$$\partial_t(\rho\vec{v}) + \vec{\nabla} \cdot (\vec{v}\rho\vec{v} + p) = \frac{\kappa\rho}{c}\vec{F} \quad (1.19)$$

$$\partial_t(e) + \vec{\nabla} \cdot (\vec{v}e + \vec{v}p) = -4\pi\kappa\rho B + c\kappa\rho E \quad (1.20)$$

$$\partial_t(E) + \vec{\nabla} \cdot (\vec{v}E + \vec{F}) = -\vec{\nabla} \cdot \vec{v}P + 4\pi\kappa\rho B - c\kappa\rho E \quad (1.21)$$

$$\partial_t\left(\frac{\vec{F}}{c^2}\right) + \vec{\nabla} \cdot \left(\frac{\vec{v} \cdot \vec{F}}{c^2} + P\right) = -\frac{\kappa\rho}{c}\vec{F} \quad (1.22)$$

These are the final radiation hydrodynamics equations. There are only 5 equations for 7 primitive variables: ρ , \vec{v} , e , p , E , \vec{F} and P . Two closing relations are necessary to close the system. A first closing relation is obtained by re-using equation (1.6), a second one can be obtained by for example the *Flux limited diffusion* approximation (FLD) described in section 1.4.

1.3 Sobolev and CAK-theory

CAK-theory is a formalism developed by Castor Abbot and Klein [?] describing the radiative acceleration of the gas g_{rad} . Gas is accelerated in the line of sight of a radiation source, an ensemble of absorption lines leads to a significant radiative acceleration. The theory is applied in the radiation driven winds of for example hot massive stars and active galactic nuclei. Let's begin by writing down the acceleration caused by free electron scattering. Consider a star with mass M_* and luminosity L_* , if only electron scattering

is assumed, light gets absorbed and re-emitted equally. If κ_e is defined as the electron scattering opacity, the radiative acceleration at any radial distance r is given by:

$$g_e = \frac{\kappa_e L_*}{4\pi r^2 c} \quad (1.23)$$

The gravitational attraction of the gas is given by Newtons gravitation law and is equal to $g_{grav} = G \frac{M_*}{r^2}$. Both accelerations vary as $\frac{1}{r^2}$, so their ratio is constant as a function of radius. An Eddington parameter for a purely electron scattering radiation force is defined as Γ_e .

$$\Gamma_e = \frac{\kappa_e L_*}{4\pi G M_* c} \quad (1.24)$$

Let's now have a look at radiative acceleration of a single absorption line in a line driven wind. Consider a set of ions absorbing at λ_0 , these ions have very high velocities and the doppler shifting of the line becomes important for absorbing photons. The first ions closest to the star will absorb photons at λ_0 so the photons further from the star are in the lines' shadow. However, the photons absorbing these photons pick up a velocity δv so they can now pick up photons with a slightly higher wavelength. This mechanism leads to a steady state monotonously increasing velocity as function of distance from the stellar surface $v(r)$. The line profile isn't infinitesimally thin, photons can be absorbed for a frequency range surrounding the central wavelength λ_0 , this is described by the profile function. In velocity-space this width is equal to the thermal velocity of the ions. In radius space, an exact wavelength can be absorbed in a part of space with geometrical width $l_{sob} dv/dr$, the Sobolev length. With this Sobolev length comes a Sobolev optical depth $\tau_{sob} = \rho \kappa l_{sob}$. With $q \kappa_e c$ describing the lines' effectiveness scaled to electron scattering opacity, and tdv/dr describing the optical depth for a purely electron scattering opacity, τ_{sob} can be written as:

$$\tau_{sob} = \frac{\rho \kappa v_{th}}{dv/dr} = qt \quad (1.25)$$

In the optically thin limit, where the amount of absorption is considered to be insignificant, the line acceleration g_{line} can be related to g_e via q . Note however that the luminosity of the star has to be weighted with the frequency at which absorption occurs.

$$g_{thin} = w_{\nu,0} q g_e = \frac{\kappa v_{th} \nu_0 L_*}{4\pi r^2 c^2} \quad (1.26)$$

Where $w_{\nu,0} = \nu_0 L_\nu / L_*$ is the weight for the frequency at which the line is absorbed. Using solutions for the radiative transfer equation, the total line acceleration can be formulated analytically.

$$g_{line} = g_{thin} \frac{1 - e^{-qt}}{qt} \quad (1.27)$$

A radiation driven wind is accelerated by a whole load of lines, so one has to sum over all line-accelerations to come to the total radiative acceleration. The density distribution of lines N as function of their strength q was approximated by Castor Abbott and Klein to be dependent on a few free parameters and the Riemann- Γ function.

$$q \frac{dN}{dq} = \frac{1}{\Gamma} \left(\frac{q}{\bar{Q}} \right)^{\alpha-1} \quad (1.28)$$

With this continuous function, the summation over all lines is transformed to an integration over all line strengths, and the final CAK acceleration can be written down:

$$g_{CAK} = g_e \int_0^\infty q \frac{dN}{dq} \frac{1 - e^{-qt}}{qt} dq \quad (1.29)$$

$$= \frac{1}{1 - \alpha} \frac{\kappa_e L_* \bar{Q}}{4\pi r^2 c} \left(\frac{dv/dr}{\rho c \bar{Q} \kappa_e} \right)^\alpha \quad (1.30)$$

When applying the CAK-formalism to the winds produced by massive stars, there is one last addition to be made. The acceleration described above assumes the star to be a point source, for material close to the stellar surface this is of course not the case. One can correct for this assumption using the finite disk correction factor f_{fd} , which depends on the distance to the stellar surface, the velocity field and the velocity gradient.

$$f_{fd}(r, v, dv/dr) = \frac{(1 + \sigma)^{1+\alpha} - (1 + \sigma \mu_*^2)^{1+\alpha}}{(1 + \alpha) \sigma (1 + \sigma)^\alpha (1 - \mu_*^2)} \quad (1.31)$$

$$g_{CAK}^{fd} = f_{fd} g_{CAK} \quad (1.32)$$

Where $\mu_* = \sqrt{1 - \frac{R_*^2}{r^2}}$ and $\sigma = \frac{d \ln(v)}{d \ln(r)} - 1$. The derivation of this finite disk correction factor can be found in [1].

WHERE? CAN I CITE COURSENOTES?

CAK-theory is a nice, easy to implement theory. In this thesis, the main application will be simulating the wind of Massive stars. Research performed here is not exactly new, but it's a great way to introduce radiation forces in **MPI-AMRVAC**. For me personally it was the first step in learning Fortran and getting to know the workings of an advanced (M)HD-code.

1.4 Flux Limited Diffusion

As mentioned before, the full RHD equations leave us with 5 partial differential equations and one HD closure relation for 7 variables. Several methods and approximations exist for closing the system and one of them is flux limited diffusion. (GIVE EXAMPLES, CITE)

Consider the steady state solution of the radiation flux equation (1.22), $\partial_t \left(\frac{\vec{F}}{c^2} \right) = \vec{0}$. In first order $\frac{v}{c^2} \ll 1$, so the relation between P and F is given by.

$$\vec{\nabla} P = -\frac{\kappa \rho}{c} \vec{F} \quad (1.33)$$

In the optically thick limit the Eddington approximation gives us $P = \frac{1}{3} E$, so (1.33) can be written as $\vec{F} = -\frac{1}{3} \frac{c}{\kappa \rho} \vec{\nabla} E$. This is the diffusion limit in radiative transfer. However, in the optically thin, free streaming limit, the density goes to zero and thus the radiation flux should go to infinity, this is unphysical. At all times, the radiation flux \vec{F} should be smaller than cE .

A solution lies in introducing the flux limiter λ , which is a factor varying between $\frac{1}{3}$ in the optically thick regime, and 0 in the optically thin. The extra necessary closure relation can now be written as:

$$\vec{F} = -\frac{c\lambda}{\kappa\rho}\vec{\nabla}E \quad (1.34)$$

Different formalisms for expressing λ exist, the one used in this work has been worked out in [5].

$$R = \frac{|\nabla E|}{\rho\kappa E} \quad (1.35)$$

$$\lambda = \frac{2 + R}{6 + 3R + R^2} \quad (1.36)$$

R Is the ratio between the photon mean free path $l_\gamma = \frac{1}{\kappa\rho}$ and the radiation field scale height $H_{rad} = \frac{E}{|\nabla E|}$. When the density is high and the medium is optically thick, l_γ will approach zero. In a low density, optically thin environment, l_γ will approach infinity. A small l_γ gives a small R and a large l_γ a large R .

$$R = \frac{\frac{1}{\kappa\rho}}{\frac{E}{|\nabla E|}}$$

λ And R also relate the radiation pressure P to the radiation energy density E via the Eddington tensor f , which is approximated as a scalar in this situation.

$$P = fE \quad (1.37)$$

$$f = \lambda + \lambda^2 R^2 \quad (1.38)$$

This means we have 7 unknowns, 5 PDE's and 3 closure relations. The momentum flux equation can be dropped and the system is self consistent within equations (1.18), (1.19), (1.20), (1.21), (1.6), (1.34) and (1.37).

The radiation flux is eliminated from the radiation gas equation 1.21:

$$\partial_t(E) + \vec{\nabla} \cdot \left(\vec{v}E - \frac{c\lambda}{\kappa\rho}\vec{\nabla}E \right) = -\vec{\nabla} \cdot \vec{v}P + 4\pi\kappa\rho B - c\kappa\rho E \quad (1.39)$$

The power of this method relies in it's simplicity with respect to solving the full radiative transfer equation (1.7) every time step for every frequency. Instead of solving the global radiative transfer equation one only needs to solve equation (1.39) which depends on local quantities. However, there are some disadvantages as well. This approach assumes the radiation field to be angle independent. So in certain regimes near the stellar surface, FLD results might be sub-optimal [12]. For these systems one is better off with a much more computationally expensive Monte-Carlo radiative transfer solver [?]. CITE HERE!!!!

Flux limited diffusion is a useful tool which can be used to probe the 2D or even 3D structure of for example stellar winds and atmospheres. Not only the energy and momentum source terms are calculated, but also a direct observable: the radiation flux. In this thesis, the main application for FLD will lie within simulating instabilities in an isothermal atmosphere of a massive star.

Chapter 2

Methodology

The CAK and FLD equations described in the introduction are not to be solved analytically. Instead, they are solved using advanced computer codes. This chapter aims to describe how RHD the equations are being solved in a computer code. The code at hand is `mpi-AMRVAC` which, except for optically thin radiative cooling, had until now been incompatible with any radiation effects. Additions made during this thesis are a novelty for the code and can in the future be used in the simulations of a multitude of astrophysical processes.

2.1 `mpi-amrvac`

Modelling of the winds and atmospheres in this thesis is done in `mpi-AMRVAC` (Message Passing Interface - Adaptive Mesh Refinement Versatile Advection Code) ([8]), a multi-dimensional, adaptive mesh (magneto-)hydrodynamics code. In the code, several explicit solvers can be used to solve the conservative hydrodynamical equations (??) with source terms defined by the user. An advantage of working with this particular code is that it is developed in house at the centre for mathematical plasma astrophysics (cmpa), which made a close collaboration with the developers possible.

To run the code, several scripts are provided:

- **Source code:** The source code consists of all the algorithms to solve the equations, apply boundary conditions, refine the mesh, read input and write output, etc... Inside the source code reside several physics modules, e.g. a module for gravity, dust, radiative cooling and viscosity.
- **user module:** In the user module, the user defines the initial conditions. Also, this is the place to add additional subroutines for user defined source terms, boundary conditions, time-step calculations and extra output variables among others. The subroutines in this file are automatically called by the source code.
- **parameter file:** The parameter file is where the computational parameters are defined: grid size and resolution, simulation time, which numerical schemes to use, what type of boundary conditions and how many output files the user wants are

only a few to be named. This file is basically a bunch of knobs for the user.

In the following sections the integration of CAK-theory and FLD in `mpi-AMRVAC` will be explained, but first an introduction to numerical HD.

2.2 HD in mpi-AMRVAC

`MPI-AMRVAC` Solves systems of hyperbolic PDE's, for example the HD-equations (1.2), (1.3) and (1.4). To solve them, the code makes use of a finite volume approach. In this section, the basic ideas of this approach are explained. Remember the shape of the HD equations in their conservative form, in one dimension they look like:

$$\partial_t u + \partial_x f_u = S_u \quad (2.1)$$

where u is the conserved quantity and F_u is its relevant flux. S_u is the sourceterm, which is put to zero in this explanation. In earlier more primitive methods such as the finite difference methods, this equation would be discretised by writing down the derivatives in a discrete way at each cell center. A finite volume method on the other hand treats the value of u as an average over a small volume instead as a value at a single point. The flux F_u is a quantity existing at the cell faces. Discretization for a finite volume method thus begins by taking the average of u in a finite volume cell by means of integration:

$$\int_{t^n}^{t^{n+1}} \int_{x_{i-\frac{1}{2}}}^{x_{i+\frac{1}{2}}} (\partial_t u + \partial_x f_u) dt dx = 0 \quad (2.2)$$

Define U as the space averaged value for u and F as the time averaged value for f_u , the above equation 2.2 can be written as:

$$U_i^{n+1} = U_i^n - \frac{\Delta t}{\Delta x} \left(F_{i+\frac{1}{2}}^n - F_{i-\frac{1}{2}}^n \right) \quad (2.3)$$

A good solver will give an accurate yet computationally cheap expression for the time integrated fluxes $F_{i+\frac{1}{2}}^n$ and $F_{i-\frac{1}{2}}^n$, which are not known analytically. Estimating the values of F is done by a *Riemann solver*. `MPI-AMRVAC` has a few of these Riemann solvers, including hll, hllc, roe, tvdlf, etc.. [11]. The approximations for $F_{i+\frac{1}{2}}^n$ will depend on u and f_u .

2.3 CAK-theory

CAK- theory describes a line driving force and thus an acceleration g_{CAK} . This acceleration, together with electron scattering acceleration g_e enter the HD momentum equation as source terms. The momentum source terms are added in the user module which is called from the source code. The simulations are done in a 1D numerical domain and the

radiative force are pointed outward. All parameters are defined, and dv/dr is calculated using a centred finite difference.

$$S_{\rho\vec{v},i,j} = \rho_{i,j} \frac{1}{1-\alpha} \frac{\kappa_e L_* \bar{Q}}{4\pi r_{i,j}^2 c} \left(\frac{1}{\rho_{i,j} c \bar{Q} \kappa_e} \right)^\alpha \frac{v_{i+1} - v_{i-1}}{r_{i+1} - r_{i-1}} \quad (2.4)$$

Because of the extra source terms, it is critical to adjust the numerical time step. One doesn't want matter to be accelerated over the length of multiple cells in one time step. In the gravity module for **MPI-AMRVAC**, the timestep is taken to be the minimum of the hydrodynamical time step calculated on the basis of the characteristic speeds (see section 2.4) and the time it takes for an gas element to be accelerated the distance of one numerical cell Δx .

$$dt = \min \left(dt_{HD}, \sqrt{\frac{\Delta x}{g_{grav}}} \right) \quad (2.5)$$

A reasoning is used in the CAK-routine, only now not only g_{grav} but also g_e and g_{CAK} are taken into account.

$$dt = \min \left(dt_{HD}, \sqrt{\frac{\Delta x}{g_{grav}}}, \sqrt{\frac{\Delta x}{g_e}}, \sqrt{\frac{\Delta x}{g_{CAK}}} \right) \quad (2.6)$$

When modelling stellar winds, it is important to also take into account a gravitational acceleration g_{grav} in the opposite direction. This term too is calculated in the user module.

In the modelling of stellar winds done in this thesis, the CAK-winds where assumed to be isothermal gas and adding the g_{cak} and g_{grav} source terms to the momentum equation. The isothermal gas assumption is made by switching off the solving for the energy equation in the parameter file. The code now neglects equations (1.4) and (??), and replaces them with a new closure relation:

$$p = a_{adiabatic} p^\gamma \quad (2.7)$$

The two free parameters a_{adiab} and γ are set to and to match the conditions in a typical CAK wind for hot massive stars. CITE HERE

2.4 Flux Limited Diffusion

Solving the Radiation hydrodynamics equations with an FLD approximation is more complicated than only adding the radiation force as in CAK-theory. For CAK, only the continuity and momentum equations where solved in only one dimension and a sourceterm was added to the momentum equation. For FLD, an entirely new equation has to be solved, source terms need to be added to this new equation, source terms need to be added to the momentum and gas energy equations and a bunch of new variables, parameters and subroutines have to be defined. Given the complexity of the problem it is solved in a new physics module in the source code. The module will first be worked out for a Cartesian,

2D-case. Lets have a closer look at the additions to the code

Continuity equation (??) is as it is and doesn't need any source terms.

$$S_\rho = 0 \quad (2.8)$$

Momentum equation (1.19) needs a source term equivalent to the radiation force, which can be computed with the help of the FLD closure relation (1.34).

$$S_{\rho\vec{v}} = \frac{\rho\kappa F}{c} \quad (2.9)$$

The radiative energy is added to the code as a variable and evolved through time. Every time step, the radiative flux is computed from this radiative energy density. $S_{\rho\vec{v}}$ is added to the momentum field every time step.

The gas energy equation needs two source terms, the radiation heating and the radiation cooling.

$$S_e = -4\pi\kappa\rho B + c\kappa\rho E \quad (2.10)$$

The source terms in the gas energy equation (1.4) depend on E and evolve on a radiation timescale instead of a hydrodynamical timescale, for this reason they will be implemented with an implicit scheme, see section 2.4.4. To staff this argument, let's give some order of magnitude estimate of the two timescales on which the gas dynamics and the radiation dynamics change. The hydrodynamical timescale can be given by the time it takes for a sound wave, travelling at the speed of sound a_s , to cross one scale height H_{eff} . On the other hand, the radiation timescale is the time it takes a photon to travel the photon mean free path l_γ at lightspeed c .

$$\tau_{HD} = \frac{H_{eff}}{a_s} = \frac{a_s}{(1-\Gamma)g} \quad (2.11)$$

$$\tau_{rad} = \frac{l_\gamma}{c} = \frac{1}{\kappa\rho c} \quad (2.12)$$

After plugging in the typical values for a stellar atmosphere surrounding a $150M_\odot$ star: $a_s \sim 10^6 \frac{m}{s}$, $\Gamma \sim 0.5$, $g \sim 10^3 \frac{m}{s^2}$, $\kappa \sim 0.34 \frac{cm^2}{g}$ and $\rho \sim 10^{-8}$ we get the following orders of magnitude for the timescales: $\tau_{HD} \sim 10^3$ and $\tau_{rad} \sim 10^{-2}$, which is a 5 magnitude difference.

$$\tau_{rad} \ll \tau_{HD} \quad (2.13)$$

The radiation energy equation (2.14) is entirely new to the code, and the same timescale argument can be applied here. A partially implicit method will be used for solving this equation, in an operator split manner.

$$\partial_t(E) + \underbrace{\vec{\nabla} \cdot (\vec{v}E)}_1 = - \underbrace{\vec{\nabla} \cdot \vec{F}}_2 - \underbrace{\vec{\nabla} \cdot \vec{v}P + 4\pi\kappa\rho B - c\kappa\rho E}_3 \quad (2.14)$$

$$(2.15)$$

- **1: The advection term**

$$\partial_t E + \vec{\nabla} \cdot (\vec{v} E) = 0 \quad (2.16)$$

can be handled using already existing `mpi-AMRVAC` routines, this part has a similar shape to the conservative form of the hd equations (2.1). This advection is the movement of the radiation field with the gas, it's a process evolving on a similar timescale as the advection of other quantities. Implementation is as simple as defining the fluxes f and calling the necessary, already existing, subroutines. The code will use the same Riemann solver as used for the other HD variables.

- **2: The diffusion term**

$$\partial_t E = -\vec{\nabla} \cdot \vec{F} \quad (2.17)$$

can, by using the fld closing relation equation 1.34, be written as:

$$\partial_t E = -\vec{\nabla} \cdot \left(-\frac{\lambda c}{\kappa \rho} \nabla E \right) \quad (2.18)$$

Trying to solve this equation together with the other hydro equations gives rise to two problems. First: the timescale on which the radiation energy E evolves is much shorter than the hydrodynamical timescale on which ρ , \vec{v} and e evolve, and second: this is a hyperbolic equation, which makes it hard to solve with the explicit schemes which are used by `MPI-AMRVAC` to solve the other equations. A way to overcome these problems is solving for the diffusion implicitly. The technique used in developing the new physics module is an Alternative Direction Implicit scheme (ADI), and is based on [12].

- **3: The photon tiring and the radiation heating and cooling sourceterms**

$$\partial_t E = -\vec{\nabla} \cdot \vec{v} P + 4\pi\kappa\rho B - c\kappa\rho E \quad (2.19)$$

are also evolving on the faster timescale, hence they must be solved implicitly. This is done with an implicit bisection scheme similar to the technique found in [12], which will be explained below.

In `MPI-AMRVAC`, the time step at which to evolve the primitive variables is computed based on the characteristic velocities in the system. These are obtained by computing the eigenvalues of the Jacobian flux matrix $A_{i,j} = \frac{\partial F_i}{\partial u_j}$, where F_i and u_j are the fluxes and conserved quantities in the conservative equations (1.5). For HD, these eigenvalues are $c_{adiab} + v$, c_{adiab} and $c_{adiab} - v$, where $c_{adiab} = \sqrt{\gamma \frac{p_{gas}}{\rho}}$ is the adiabatic soundspeed. During computations, waves must not travel across cells, thus the time step should be smaller than the time it takes for an acoustic wave with velocity $c_{adiab} + v$ to travel a distance Δx which is the width of one numerical cell.

In RHD environments there is an extra type of waves, namely radiation acoustic waves [6] which can, depending on the ratio between the gas and radiation pressure, travel faster

than the sound speed. The time step in the RHD calculations will be calculated as a function of $\sqrt{\gamma \frac{p_{gas} + P_{rad}}{\rho}}$ instead of $\sqrt{\gamma \frac{p_{gas}}{\rho}}$.

$$dt = \min \left(\frac{\Delta x}{v + \sqrt{\gamma \frac{p_{gas} + P_{rad}}{\rho}}} \right) \quad (2.20)$$

This is the maximum time step at which the advection should be calculated to be numerically viable. The diffusion of the radiation energy field on the other hand will generally happen much quicker, depending on the gradient of the E -field and diffusion coefficient. This problem is circumvented by using an implicit scheme, described in section 2.4.2

2.4.1 Elliptic vs Hyperbolic

2.4.2 ADI

The diffusion part of the radiative energy equation will be solved using the Alternating Direction Implicit (ADI) scheme. This is a numerical scheme which solves the equation implicitly in the first spatial direction and explicitly in the second for half a time step, and then implicitly in the second spatial direction and explicitly in the first for another half time step. ADI is computationally cheaper than a completely implicit scheme, yet it is possible to converge with very big time steps compared to explicit schemes. Lets first simplify the equation by defining the diffusion coefficient $D = \frac{\lambda c}{\kappa \rho}$.

$$\partial_t E = \vec{\nabla} \cdot (D \nabla E) \quad (2.21)$$

Lets now construct a numerical scheme. First, the left hand side of (2.21) is written in a discrete form. Let $E_{i,j}$ be the radiative energy in cell (i, j) on a grid with Δx grid spacing in the x -direction and Δy grid spacing in the y -direction. $D_{i,j}$ is the cell centred diffusion coefficient, whilst $D1_{i,j} = \frac{1}{2}(D_{i+1,j} + D_{i,j})$ and $D2_{i,j} = \frac{1}{2}(D_{i,j+1} + D_{i,j})$ are the cell faced values. To transform from cell center to cell face it is simplest to take the mean of the two surrounding cells, taking the mean of the six surrounding cells is also an option.

$$\left(\vec{\nabla} \cdot (D \nabla E) \right)_{i,j} = \frac{D1_{i+1,j}(\nabla E)_x - D1_{i,j}(\nabla E)_x}{\Delta x} \quad (2.22)$$

$$+ \frac{D2_{i,j+1}(\nabla E)_y - D2_{i,j}(\nabla E)_y}{\Delta y} \quad (2.23)$$

$$= \frac{D1_{i+1,j}(E_{i+1,j} - E_{i,j}) - D1_{i,j}(E_{i,j} - E_{i-1,j})}{\Delta x^2} \quad (2.24)$$

$$+ \frac{D2_{i,j+1}(E_{i,j+1} - E_{i,j}) - D2_{i,j}(E_{i,j} - E_{i,j-1})}{\Delta y^2} \quad (2.25)$$

$$(2.26)$$

For the implicit scheme the diffusion coefficients are chosen at the previous time step, as they are to be computed from a known radiation energy. Evolving (2.21) over a time step Δt from time n to time $n + 1$ means solving the following equations:

$$\frac{E^{n+1} - E^n}{\Delta t} = \vec{\nabla} \cdot (D^n \nabla E^{n+1}) \quad (2.27)$$

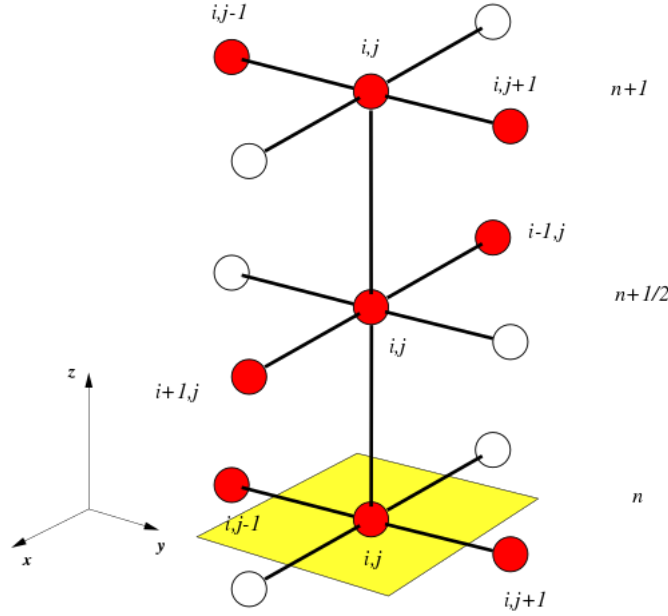


Figure 2.1: Courtesy: wikipedia. This stencil portraits the working of the ADI scheme, the first half time step from $t = n$ to $t = n + \frac{1}{2}$ depends implicitly on the cells with coordinates $(i + 1, j)$, (i, j) and $(i - 1, j)$. The second half time step to $t = n + 1$ depends implicitly on cells $(i, j + 1)$, (i, j) and $(i, j - 1)$.

As mentioned before, the ADI scheme first solves half a time step implicitly in for example the x -direction and explicitly in the y -direction:

$$\frac{E_{i,j}^{n+\frac{1}{2}} - E_{i,j}^n}{\Delta t} = \frac{D1_{i+1,j}^n}{\Delta x^2} (E_{i+1,j}^{n+\frac{1}{2}} - E_{i,j}^{n+\frac{1}{2}}) \quad (2.28)$$

$$- \frac{D1_{i,j}^n}{\Delta x^2} (E_{i,j}^{n+\frac{1}{2}} - E_{i-1,j}^{n+\frac{1}{2}}) \quad (2.29)$$

$$+ \frac{D2_{i,j+1}^n}{\Delta y^2} (E_{i,j+1}^n - E_{i,j}^n) \quad (2.30)$$

$$- \frac{D2_{i,j}^n}{\Delta y^2} (E_{i,j}^n - E_{i,j-1}^n) \quad (2.31)$$

Here it is clear how this step is implicit in the x -direction, this expression features the terms $E_{i+1,j}^{n+\frac{1}{2}}$ and $E_{i-1,j}^{n+\frac{1}{2}}$. Yet it is implicit in y , there are no $E_{i,j+1}^{n+\frac{1}{2}}$ or $E_{i,j-1}^{n+\frac{1}{2}}$ terms. This is portrayed in figure 2.4.2.

In a more simple form, this can be written as:

$$\left(1 + \frac{\Delta t D1_{i+1,j}^n}{\Delta x^2} + \frac{\Delta t D1_{i,j}^n}{\Delta x^2}\right) E_{i,j}^{n+\frac{1}{2}} - \frac{\Delta t D1_{i+1,j}^n}{\Delta x^2} E_{i+1,j}^{n+\frac{1}{2}} - \frac{\Delta t D1_{i,j}^n}{\Delta x^2} E_{i-1,j}^{n+\frac{1}{2}} = b_{i,j} \quad (2.32)$$

where $b_{i,j} = \left(1 + \frac{\Delta t D2_{i,j+1}^n}{\Delta y^2} + \frac{\Delta t D2_{i,j}^n}{\Delta y^2}\right) E_{i,j}^n + \frac{\Delta t D2_{i,j+1}^n}{\Delta y^2} E_{i,j+1}^n + \frac{\Delta t D2_{i,j}^n}{\Delta y^2} E_{i,j-1}^n$ Or in matrix

form:

$$\begin{bmatrix} \ddots & & & & \\ & \ddots & & & \\ & & \left(1 + \frac{\Delta t D1_{i,j}^n}{\Delta x^2} + \frac{\Delta t D1_{i-1,j}^n}{\Delta x^2}\right) & -\frac{\Delta t D1_{i,j}^n}{\Delta x^2} & \\ & & -\frac{\Delta t D1_{i,j}^n}{\Delta x^2} & \left(1 + \frac{\Delta t D1_{i+1,j}^n}{\Delta x^2} + \frac{\Delta t D1_{i,j}^n}{\Delta x^2}\right) & -\frac{\Delta t D1_{i,j}^n}{\Delta x^2} \\ & & & -\frac{\Delta t D1_{i+1,j}^n}{\Delta x^2} & \left(1 + \frac{\Delta t D1_{i+2,j}^n}{\Delta x^2} + \frac{\Delta t D1_{i+1,j}^n}{\Delta x^2}\right) & \ddots \\ & & & & \ddots & \ddots \end{bmatrix} \quad (2.33)$$

$$\times \begin{bmatrix} \vdots \\ E_{i-1,j}^{n+\frac{1}{2}} \\ E_{i,j}^{n+\frac{1}{2}} \\ E_{i+1,j}^{n+\frac{1}{2}} \\ \vdots \end{bmatrix} = \begin{bmatrix} \cdots & b_{i,j-1} & b_{i,j} & b_{i,j+1} & \cdots \end{bmatrix} \quad (2.34)$$

This is, depending on boundary conditions, a tridiagonal matrix. In **MPI-AMRVAC**, this matrix system is solved on the entire computational domain and the innermost layer of ghost cells. The boundary conditions on the radiation energy field are applied after the halved time step. This way, the matrix can be kept tridiagonal independent of those boundary conditions. The tridiagonal system is then solved using Thomas' algorithm, which is a simplified form of Gaussian elimination. If the computational grid is N cells by M cells this is an $(N+2) \times (N+2)$ system, and there are $M+2$ such systems to be solved to evolve half a time step.

The second half of the time step will be solved implicit in the y -direction and explicit in x -direction. Notice again the diffusion coefficient can only be updated at the end of the cycle:

$$\frac{E_{i,j}^{n+1} - E_{i,j}^{n+\frac{1}{2}}}{\Delta t} = \frac{D1_{i+1,j}^n}{\Delta x^2} (E_{i+1,j}^{n+\frac{1}{2}} - E_{i,j}^{n+\frac{1}{2}}) \quad (2.35)$$

$$- \frac{D1_{i,j}^n}{\Delta x^2} (E_{i,j}^{n+\frac{1}{2}} - E_{i-1,j}^{n+\frac{1}{2}}) \quad (2.36)$$

$$+ \frac{D2_{i,j+1}^n}{\Delta y^2} (E_{i,j+1}^{n+1} - E_{i,j}^{n+1}) \quad (2.37)$$

$$- \frac{D2_{i,j}^n}{\Delta y^2} (E_{i,j}^{n+1} - E_{i,j-1}^{n+1}) \quad (2.38)$$

And this system, of course, has a matrix notation completely similar to the one in the first half of the time step, which can be solved in a completely similar way. This time there are $N+2$ matrices of size $(M+2) \times (M+2)$ to be solved.

The accuracy of this scheme can be checked by computing and comparing both the left hand side and right hand side of equation (2.27) after calculating E^{n+1} . The error is defined by taking the maximum value of the difference of the LHS and RHS weighted

with the ratio of the radiative energy and the hydrodynamical time step [12]:

$$Error = \max_{i,j} \left(\frac{\frac{E^{n+1} - E^n}{\Delta t} - \vec{\nabla} \cdot (D \nabla E)}{E^n / dt} \right) \quad (2.39)$$

In a completely correct computation, the left hand side would correspond exactly to the right hand side, however due to numerical computations and assumptions this will almost never truly hold.

Test calculations show that this error is often rather large, depending on the distribution of the radiation and density field in the simulation space. A solution is given in [12] by means of pseudo-time stepping. This method will be explained in the next section.

2.4.3 pseudo-timestepping

Instead of solving equation (2.21) a new variable is introduced, the *pseudotimestep* w . The ADI-method is used to solve

$$\partial_w E = \partial_t E - \vec{\nabla} \cdot (D \nabla E) \quad (2.40)$$

which reduces to (2.21) when $\partial_w E = 0$. The pseudo-time step Δw can be adjusted independently from the hydrodynamical time step and is increasing in size to converge toward a w -stationary state where $\partial_w E = 0$ [12].

The implicit scheme to evolve half a pseudo-time step from m to $m + \frac{1}{2}$, implicit in the x -direction, looks like:

$$\frac{E_{i,j}^{m+\frac{1}{2}} - E_{i,j}^m}{\Delta w} = \frac{E_{i,j}^{m+\frac{1}{2}} - E_{i,j}^n}{\Delta t} \quad (2.41)$$

$$- \frac{D1_{i+1,j}^n}{\Delta x^2} (E_{i+1,j}^{m+\frac{1}{2}} - E_{i,j}^{m+\frac{1}{2}}) \quad (2.42)$$

$$+ \frac{D1_{i,j}^n}{\Delta x^2} (E_{i,j}^{m+\frac{1}{2}} - E_{i-1,j}^{m+\frac{1}{2}}) \quad (2.43)$$

$$- \frac{D2_{i,j+1}^n}{\Delta y^2} (E_{i,j+1}^m - E_{i,j}^m) \quad (2.44)$$

$$+ \frac{D2_{i,j}^n}{\Delta y^2} (E_{i,j}^m - E_{i,j-1}^m) \quad (2.45)$$

Where, again, the Diffusion coefficient is considered constant throughout the time step. This corresponds to a similar tridiagonal matrix equation which has to be solved for every pseudo-time step. The size of the pseudo-time step Δw is exponentially increasing in size per iteration $m = 1 \dots W$ and is given by:

$$\Delta w = \Delta w_0 \left(\frac{\Delta w_1}{\Delta w_0} \right)^{\frac{m-1}{W-1}} \quad (2.46)$$

Δw_0 And Δw_1 are one quarter of the size of a grid cell and one quarter of the size of the numerical domain, respectively. The error of this combined ADI-pseudo-time step

method is measured after W pseudo-time steps using (2.39), if it is too large W and Δw_1 are increased whilst Δw_0 is decreased. If this still doesn't suffice, the above scheme can be applied twice to half a hydrodynamical time step, four times to a quarter hydrodynamical time step, ...

2.4.4 Bisection Implicit scheme

Solving for the source terms in the gas and radiation energy equations happens with another implicit scheme.

$$e^{n+1} - e^n = \Delta t \left(-4\kappa\sigma \left(\frac{(\gamma - 1)e^{n+1}}{\rho} \right)^4 + c\kappa E^{n+1} \right) \quad (2.47)$$

$$E^{n+1} - E^n = \Delta t \left(+4\kappa\sigma \left(\frac{(\gamma - 1)e^{n+1}}{\rho} \right)^4 - c\kappa E^{n+1} - \nabla \vec{v} P^{n+1} \right) \quad (2.48)$$

This can be rewritten as:

$$e^{n+1} - e^n = -a_1 (e^{n+1})^4 + a_2 E^{n+1} \quad (2.49)$$

$$E^{n+1} - E^n = a_1 (e^{n+1})^4 - a_2 E^{n+1} - a_3 E^{n+1} \quad (2.50)$$

where $a_1 = 4\kappa\sigma \left(\frac{(\gamma-1)}{\rho^{n+1}} \right)^4 \Delta t$, $a_2 = c\kappa\Delta t$ and $a_3 = \frac{\nabla \vec{v} P^{n+1}}{E^{n+1}} \Delta t$. Manipulation of the equations returns:

$$E^{n+1} = \frac{a_1 (e^{n+1})^4 + E^n}{1 + a_2 + a_3} \quad (2.51)$$

And

$$(e^{n+1})^4 + \frac{1 + a_2 + a_3}{a_1 + a_3} e^{n+1} - \frac{(1 + a_2 + a_3)e^n + a_2 E^n}{a_1 + a_3} = 0 \quad (2.52)$$

Equation (2.52) is a 4th degree polynomial in e^{n+1} , with a single root between 0 and $\frac{1+a_2+a_3}{a_1+a_3}$. This is solved for e^{n+1} using the bisection method to calculate the contribution of the radiative heating and cooling to the gas energy. The newly calculated e^{n+1} is then plugged in equation (2.51) to find the contribution of the radiative heating and cooling, and the photon tiring to the radiation energy.

2.5 Dimensionless problem

Computationally, floating point values are the most precise around unity. For this reason, it is preferable to rescale physical quantities such as ρ , \vec{v} , e , ... with values typical to the simulation domain. `mpi-AMRVAC` already does part of this for us, in the user module, one can define either the units for either number density, temperature and length (N_0 , T_0 , l_0) or number density, velocity and length (N_0 , v_0 , l_0). The other units will be then

computed from these quantities using the following relations:

$$\rho_0 = \mu m_p N_0 \quad (2.53)$$

$$p_0 = \gamma N_0 k_b T_0 \quad (2.54)$$

$$v_0 = \sqrt{\frac{p_0}{\rho_0}} \quad (2.55)$$

$$t_0 = \frac{l_0}{v_0} \quad (2.56)$$

Calculations in amrvac are always done with unit less quantities, in HD these are: $\tilde{\rho} = \frac{\rho}{\rho_0}$, $\tilde{\vec{v}} = \frac{\vec{v}}{v_0}$, $\tilde{p} = \frac{p}{p_0}$ and $\tilde{e} = \frac{e}{p_0}$. Using this, we can transform the continuity equation (1.2) to dimensionless units:

$$\frac{t_0}{\rho_0} \frac{\partial}{\partial t} \rho + \frac{t_0}{\rho_0} \vec{\nabla} \cdot (\rho \vec{v}) = \frac{t_0}{\rho_0} S_\rho \quad (2.57)$$

$$\frac{\partial}{\partial \tilde{t}} \tilde{\rho} + \vec{\nabla} \cdot (\tilde{\vec{v}} \tilde{\rho}) = \tilde{S}_\rho \quad (2.58)$$

Where ∇ has the units of reciprocal length and $\tilde{S}_\rho \rho_0 S_\rho$. Equivalently, for the momentum equation (1.3) and the gas energy equation (1.4):

$$\frac{\partial}{\partial \tilde{t}} (\tilde{\rho} \tilde{\vec{v}}) + \vec{\nabla} \cdot (\tilde{\vec{v}} \tilde{\rho} \tilde{\vec{v}} + \tilde{p}) = \tilde{S}_{\rho \vec{v}} \quad (2.59)$$

$$\frac{\partial}{\partial \tilde{t}} \tilde{e} + \vec{\nabla} \cdot (\tilde{\vec{v}} \tilde{e} + \tilde{\vec{v}} \tilde{p}) = \tilde{S}_e \quad (2.60)$$

$$(2.61)$$

$\tilde{S}_{\rho \vec{v}} = S_{\rho \vec{v}} \frac{t_0}{\rho_0 v_0}$ and $\tilde{S}_e = S_e \frac{t_0}{e_0}$ are the momentum and energy equation source terms. For the RHD equations this will rescale the physical constants:

$$\tilde{S}_{\rho \vec{v}} = \frac{\kappa \rho}{c} \vec{F} \frac{t_0}{\rho_0 v_0} \quad (2.62)$$

$$= \frac{\tilde{\kappa} \tilde{\rho}}{\tilde{c}} \tilde{\vec{F}} \quad (2.63)$$

$$\tilde{S}_e = -4\pi \kappa \rho B \frac{t_0}{e_0} + c \kappa \rho E \frac{t_0}{e_0} \quad (2.64)$$

$$= -4\tilde{\sigma} \tilde{\kappa} \tilde{T}^4 + \tilde{c} \tilde{\rho} \tilde{E} \quad (2.65)$$

The rescaled Stefan-Boltzmann constant $\tilde{\sigma}$, the rescaled speed of light \tilde{c} and the rescaled opacity $\tilde{\kappa}$ are given by $\tilde{\sigma} = \frac{\sigma T_0^4}{e_0 v_0}$, $\tilde{c} = \frac{c}{v_0}$ and $\tilde{\kappa} = \kappa v_0 t_0$. The unit of the radiation energy density and radiation flux can also be defined: $E_0 = e_0$ and $F_0 = \rho_0 v_0^3$. These are used to transform the RHD equation for radiation energy (1.21) and the FLD closure relation (1.34).

2.6 visualisation

mpi-AMRVAC output is given in binary .vtu files. Special software such as visit [1] and paraview are used to open and analyse the simulation results. The standard output consists of the primitive variables at every simulation cell for a specified number of time steps. Extra output variables can be defined in the user module.

Chapter 3

Results

The first result is of course a working CAK subroutine and FLD module in mpi-AMRVAC. This is something that didn't exist before and it will open the way toward simulations of new physical regimes where radiation plays a role in the dynamics of the system. Other than writing the software, there are also some scientific results. These will be described in this chapter.

3.1 CAK-Theory

3.1.1 Massive star stellar wind

The CAK momentum equation for a point like star, so when disregarding the finite disk correction factor, has an analytic solution. This solution will be derived below, based on CITE OWOCKI NOTES [], for comparison with the numerical models. Begin by writing down the steady state momentum equation in 1D spherical coordinates, disregarding the pressure term. The pressure gradient will be orders of magnitude smaller than the gravitational and radiative accelerations due to the low density environment.

$$\nabla_r (v_r \rho v_r) = \rho g_{CAK} + \rho g_e - \rho g_{grav} \quad (3.1)$$

Using the steady state continuity equation and the electron scattering Eddington factor Γ_e this can be written as:

$$v \frac{dv}{dr} = g_{CAK} + (\Gamma_e - 1) \frac{GM_*}{r^2} \quad (3.2)$$

Lets introduce two new variables: $x = 1 - \frac{R_*}{r}$ is a dimensionless inverse radius coordinate, and $w = \frac{v^2}{v_{esc}^2}$ is the kinetic energy as ratio of the kinetic energy at effective escape velocity. The effective escape velocity is similar to the general expression for escape velocity, but corrected for electron scattering force: $v_{esc} = \sqrt{\frac{GM_*(1-\Gamma_e)}{2R_*}}$. Using the notation $w' = \frac{dw}{dx} = \frac{dw}{dr} \frac{dr}{dx}$ and $v' = \frac{dv}{dx} = \frac{dv}{dr} \frac{dr}{dx}$, w' can be written as:

$$w' = vv' \frac{r^2}{GM_*(1-\Gamma_e)} \quad (3.3)$$

Fill in $v \frac{dv}{dr}$ from equation (3.2), where $vv' = v \frac{dv}{dr} \frac{dr}{dx} = v \frac{dv}{dr} \frac{r^2}{R_*^2}$:

$$w' = v \frac{dv}{dr} \frac{r^2}{R_*^2} \frac{r^2}{GM_*(1 - \Gamma_e)} \quad (3.4)$$

$$= \left(g_{CAK} + (\Gamma_e - 1) \frac{GM_*}{r^2} \right) \frac{r^2}{R_*^2} \frac{r^2}{GM_*(1 - \Gamma_e)} \quad (3.5)$$

$$= Cw'^\alpha - 1 \quad (3.6)$$

SOMETHING FISHY WITH $\frac{r^2}{R_*^2}$ HERE

Where, if we define the constant mass loss rate $\dot{M} = 4\pi r^2 \rho(r) v(r) = c^{ste}$, C is given by:

$$C = \frac{1}{1 - \alpha} \left(\frac{\bar{Q}\Gamma_e}{1 - \Gamma_e} \right)^{1-\alpha} \left(\frac{L_*}{\dot{M}c^2} \right)^\alpha \quad (3.7)$$

C Depends inversely on the mass loss rate \dot{M} , depending on the value of C , equation 3.6 has either zero ($C < C_{crit}$), one ($C = C_{crit}$) or two ($C > C_{crit}$) solutions. A low value for C leads to a high \dot{M} , so the solution with the maximal mass loss rate is the single solution at $C = C_{crit}$. The critical value for C occurs when the function Cw'^α intersects function $1 + w'$ in a tangent point. This occurs at critical argument $w'_{crit} = \frac{\alpha}{1-\alpha}$ for critical C -value $C = \frac{\alpha^{-\alpha}}{(1-\alpha)^{1-\alpha}}$. One can now integrate over w'_{crit} to obtain a velocity profile.

$$w'_{crit} = \frac{\alpha}{1 - \alpha} \quad (3.8)$$

$$\frac{d}{dx} \frac{v^2}{v_{esc}^2} = \quad (3.9)$$

$$\frac{dr}{dx} \frac{d}{dr} \frac{v^2}{v_{esc}^2} = \quad (3.10)$$

$$v^2 = v_{esc}^2 \frac{\alpha}{1 - \alpha} \left(1 - \frac{R_*}{r} \right) \quad (3.11)$$

$$v(r) = v_{esc} \sqrt{\frac{\alpha}{1 - \alpha}} \left(1 - \frac{R_*}{r} \right)^{\frac{1}{2}} \quad (3.12)$$

Where, when $r \rightarrow \infty$, $v(\infty) \rightarrow v_{esc} \sqrt{\frac{\alpha}{1 - \alpha}} = v_\infty$. The maximal CAK mass loss rate connected to the critical value C_{crit} can also be analytically determined:

$$\dot{M}_{CAK} = \frac{L_*}{c^2} \frac{\alpha}{1 - \alpha} \left(\frac{\bar{Q}\Gamma_e}{1 - \Gamma_e} \right)^{\frac{1-\alpha}{\alpha}} \quad (3.13)$$

Now that we have two testable observables (the steady state CAK mass loss rate \dot{M}_{CAK} and the steady state velocity profile $v(r)$), it's time to compare some simulations.

The CAK source term is used to model the wind of a massive O-type star. The parameters used are $L = \dots L_\odot$, $M = \dots M_\odot$, $c_{adiab} = \dots$

The tricky bit in the setup of this simulation is choosing correct lower boundary conditions. The stellar wind is launched from the stellar surface and begins with a subsonic velocity. Computationally this is controlled by setting the lower boundary density ρ_0 . If the density on the lower bound is too high, the simulation domain begins "too deep" in the stellar atmosphere, there is too much mass to be lifted away. If ρ_0 is too low, the simulation domain begins in the supersonic region.

The lower boundary for the radial velocity is set according to the continuity equation $\nabla(\rho v) = 0$, meaning $v_0 = \frac{\rho_1 v_1}{\rho_0}$. Initial conditions are evolved towards a stable, steady state, results of the simulations and their analytical counterparts are plotted in figures ?? and ??.

When accounting for the finite disk correction....

3.2 Flux Limited Diffusion

The diffusion model is first tested against some analytical results, these tests will give us knowledge on how accurately we can interpret the simulations of physical phenomena. The diffusion term and advection term are both tested separately by comparison to an exact solution and the photon tiring, radiative cooling and radiative heating source terms are tested together versus a Runge-Kutta solver of a simplified problem.

3.2.1 Testcase 1: Advection and Diffusion

The advection problem and the diffusion problem, solved by the already existing Riemann solver and the new ADI solver respectively can be tested by comparing them to simplified situation where an exact analytical solution can be found. The Riemann solver is nothing new, it's just a matter of checking the correct implementation in **MPI-AMRVAC**. The problem is tested on a numerical domain with constant density, a constant velocity, a constant gas energy density and an initial radiation energy density $E_0(x, y, t)$ given by:

$$E_0(x, y, t) = 2 + \sin(2\pi x) \sin(2\pi y) \quad (3.14)$$

If diffusion and other source terms are ignored, the radiation field will evolve as

$$E^{adv}(x, y, t) = 2 + \sin(2\pi(x - v_x t)) \sin(2\pi(y - v_y t)) \quad (3.15)$$

If diffusion is switched on but the advection is ignored by fixing the velocity field to $\vec{0}$ every iteration, and the diffusion coefficient is chosen constant at $D = 1$, the field evolves as

$$E^{diff}(x, y, t) = 2 + \exp(-8\pi^2 t) \sin(2\pi x) \sin(2\pi y) \quad (3.16)$$

Function $E_0(x, y, t)$ describes a series of dots of more and less radiative energy, see figure ??. The numerical domain is chosen in such a way that there is one region of lower and one region of higher energy in each direction, $-0.5 \leq x \leq 0.5$, $0 \leq y \leq 1$. In time, the diffusion test runs until the amplitude of the dots diminish by two orders of magnitude $\exp(-8\pi^2 t) = 10^{-1}$ and the advection test runs until a point has passed the computational

domain twice $\min(v_x, v_y)t = 2$. Boundary conditions are set periodical. Computational result \tilde{E} can be compared with the analytical results to define the residuals:

$$RES^{adv} = \left| \frac{\tilde{E}^{adv} - E^{adv}}{E^{adv}} \right| \quad (3.17)$$

$$RES^{diff} = \left| \frac{\tilde{E}^{diff} - E^{diff}}{E^{diff}} \right| \quad (3.18)$$

Which are plotted for different timesteps in figure 3.2.1 and 3.2.1 together with the numerical solutions \tilde{E}^{adv} and \tilde{E}^{diff}

3.2.2 Testcase 2: Photon Tiring, Heating and Cooling

To test the implicit bisection scheme used for adding the photon tiring, radiative heating and radiative cooling source terms, we make comparisons with an explicit Runge-Kutta solver. Of course this Runge-Kutta solver is not to be used in the actual code, because the time step chosen in the Runge-Kutta solver will be orders of magnitude smaller. The equation at hand is:

$$\frac{de}{dt} = c\rho\kappa E - 4\rho\kappa\sigma T^4 \quad (3.19)$$

Equivalently the test can be ran on the source terms for the radiative energy equation $\frac{dE}{dt} = -\vec{\nabla} \cdot \vec{v}P + 4\pi\kappa\rho B - c\kappa\rho E$, where the photon tiring term would drop out due to the velocity field being zero. Except for the gas energy density, all primitive variables ($\rho = \rho_0$, $\vec{v} = 0$ and $E = E_0$) are kept constant. The computational domain is taken as small as possible and radiative diffusion is switched off.

The system would be in radiative equilibrium when $c\rho\kappa E = 4\rho\kappa\sigma T^4$. Using $T = \frac{p}{\rho} \frac{m_p\mu}{k_b} = \frac{(\gamma-1)e}{\rho} \frac{m_p\mu}{k_b}$, on can compute the equilibrium gas energy.

$$e_{eq.} = \frac{\rho}{\gamma - 1} \frac{k_b}{m_p\mu} \left(\frac{cE}{4\sigma} \right)^{\frac{1}{4}} \quad (3.20)$$

Different initial conditions are chosen for e_0 ranging from $10^2 e_{eq.}$ to $10^{-10} e_{eq.}$. Comparisons between the bisection method and a simple Runge-Kutta solver are plotted in figure 3.2.2. Remember that the implicit bisection method uses a time step which is several orders of magnitude larger than the explicit Runge-Kutta method.

3.2.3 Isothermal Thomson Atmosphere

A first practical use for the FLD module is modelling a stellar atmosphere surrounding a massive star. For convenience, the atmosphere will be considered isothermal and plane parallel, so the flux in the initial condition and the gravitational acceleration are constant. Let's also assume the only absorption an emission is done by means of electron scattering (Thomson atmosphere), with a constant opacity κ . The model is done in 2D.

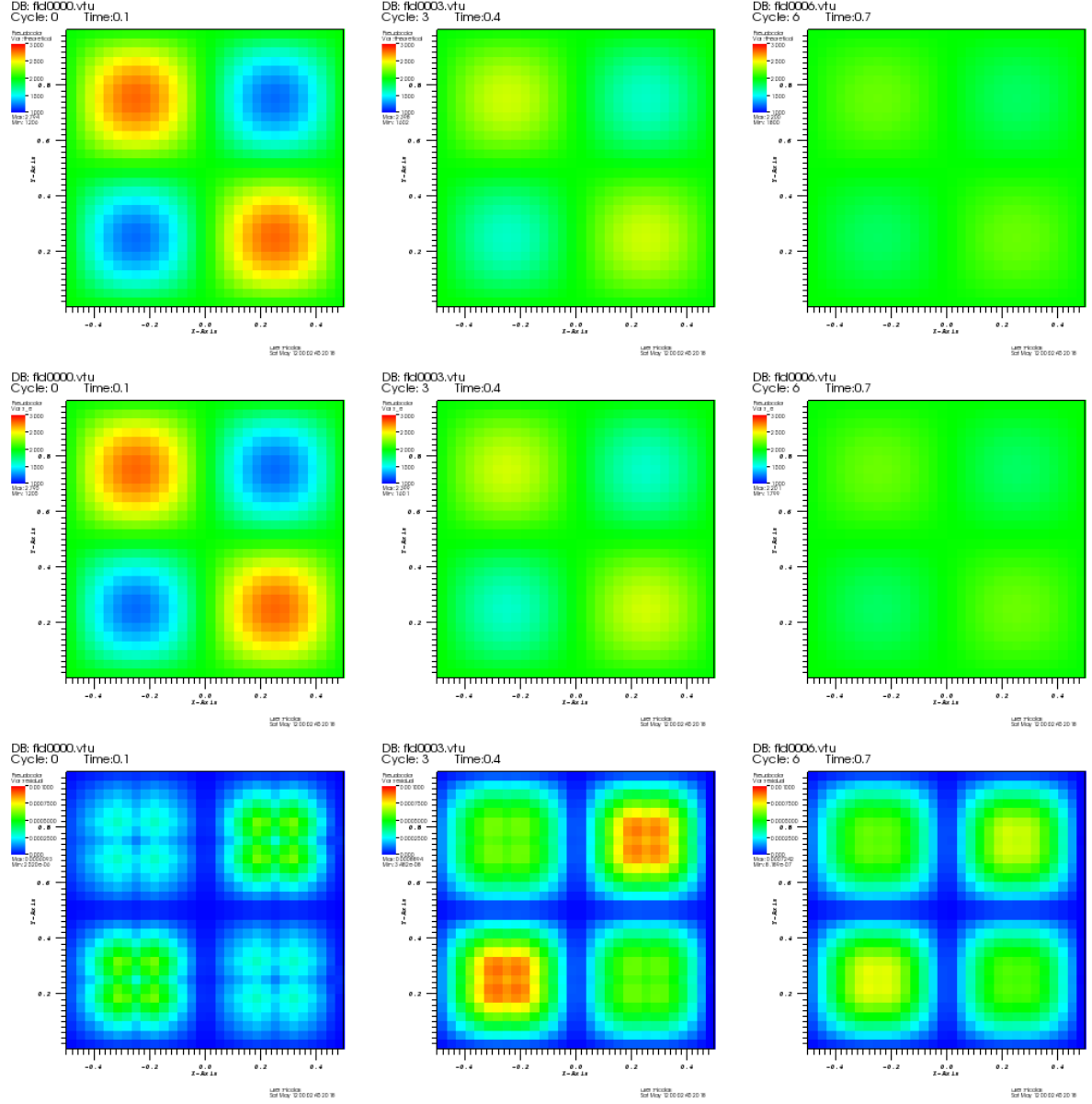


Figure 3.1: top: theoretical solution E^{diff} , middle: computed result \tilde{E}^{diff} and bottom: residual RES^{diff} . Left: 0.1 t_0 , middle 0.4 t_0 and right 0.7 t_0 . The scale goes from 1 to 2 for the upper two rows and from 0 to 0.01 for the bottom row.

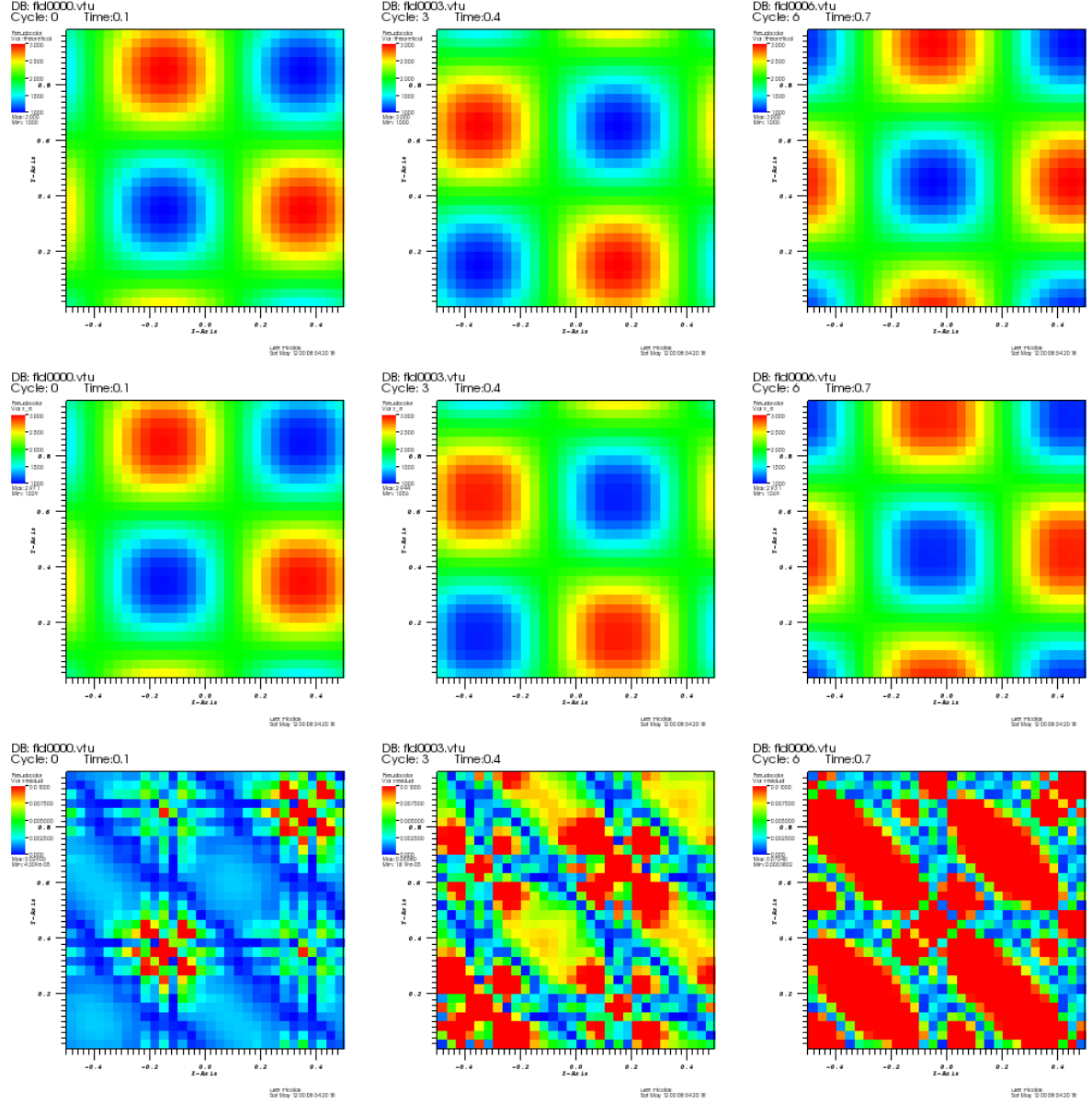


Figure 3.2: top: theoretical solution E^{adv} , middle: computed result \tilde{E}^{adv} and bottom: residual RES^{adv} . Left: $0.1t_0$, middle $0.4t_0$ and right $0.7t_0$. The scale goes from 1 to 2 for the upper two rows and from 0 to 0.001 for the bottom row.

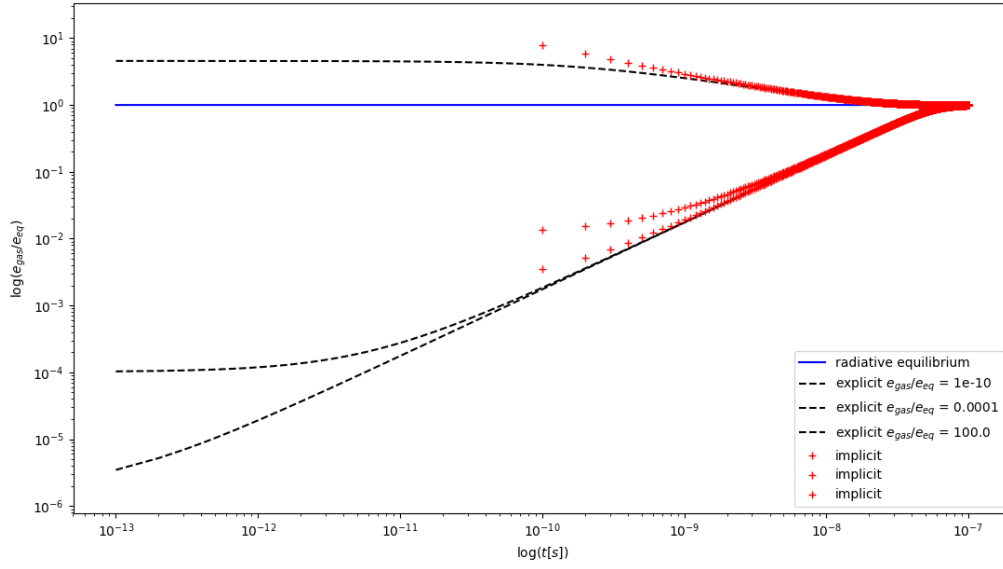


Figure 3.3: Comparison between the implicit method (red) and explicit method (black) for computing the radiation heating and cooling source terms. The gas energy evolves toward radiative equilibrium (blue) from different initial values. The time step for the implicit method is 3 orders of magnitude larger than the time step for the explicit method.

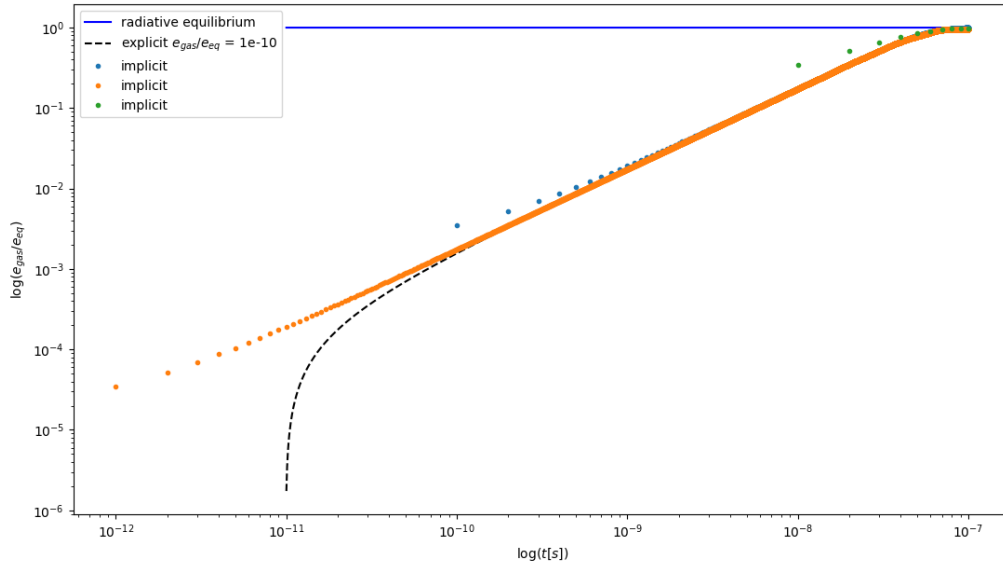


Figure 3.4: Comparison between the implicit method (orange, blue, green) and explicit method (black) for computing the radiation heating and cooling source terms. The gas energy evolves toward radiative equilibrium (blue) with different time steps for the implicit method.

The initial conditions are crucial in stabilizing simulations. In an isothermal atmosphere, the density and gas pressure decay exponentially on the length of a scale height $H_{eff} = \frac{c_{sound}^2}{g_{eff}}$. Where $g_{eff} = g_{grav}(\Gamma - 1)$ is the sum of radiation and gravitational accelerations.

$$\rho(y) = \rho_0 \exp\left(-\frac{y}{H_{eff}}\right) \quad (3.21)$$

$$p(y) = p_0 \exp\left(-\frac{y}{H_{eff}}\right) \quad (3.22)$$

The velocity field is $\vec{0}$ everywhere. Boundary conditions ρ_0 and p_0 can be computed by defining the optical depth τ .

$$\tau(y) = \int_{\infty}^y \kappa \rho dy \quad (3.23)$$

Let $dm = \rho dy$, if this is substituted in (3.23) we get an expression for the column mass in terms of opacity and optical depth:

$$\frac{\tau}{\kappa} = \int_0^{m'} \rho dm \quad (3.24)$$

In a static medium, the gravitational and radiative acceleration of the gas is countered by the gas pressure gradient. Concerning the radiation field, one can make a similar statement. The radiative acceleration is countered by the radiation pressure gradient:

$$\frac{dp}{dy} \frac{1}{\rho} = \frac{dp}{dm} = -g_{grav}(1 - \Gamma) \quad (3.25)$$

$$\frac{dP}{dy} \frac{1}{\rho} = \frac{dP}{dm} = -g_{grav}\Gamma \quad (3.26)$$

Manipulating equation and substituting the expression for column mass leads to the following relation between optical depth and gas pressure:

$$\int_0^{p_0} dp = -g_{grav}(1 - \Gamma) \int_0^{m_0} dm \quad (3.27)$$

This expression can be integrated towards a value for p_0 at a given optical depth τ_0 . (3.26) And (3.23) can be divided by one another to get a relation between dp and dP .

$$p_0 = g_{eff} \frac{\tau_0}{\kappa} \quad (3.28)$$

$$dP = \frac{\Gamma}{\Gamma - 1} dp \quad (3.29)$$

The gas energy density can be set from the gas pressure profile. Because of the plane parallel approximation, Γ is constant as well and in the Eddington limit, $E = 3P$. Now we have a boundary condition for the radiation energy field as well. The initial conditions

are given by:

$$\rho(y) = g_{eff} \frac{\tau_0}{\kappa C_{sound}^2} \exp\left(-\frac{y}{H_{eff}}\right) \quad (3.30)$$

$$\rho \vec{v} = \vec{0} \quad (3.31)$$

$$e = \frac{\tau_0}{\kappa(\gamma - 1)} \exp\left(-\frac{y}{H_{eff}}\right) \quad (3.32)$$

$$E = \frac{3\Gamma}{1 - \Gamma} \frac{\tau_0}{\kappa} \exp\left(-\frac{y}{H_{eff}}\right) \quad (3.33)$$

$$(3.34)$$

ρ_0 And p_0 are also used as lower boundary conditions for density and gas energy during the simulations. The boundary condition for the x -component of the velocity is 0 and the one for the y -component is set with the help of the steady state continuity equation $\partial_y(v_y \rho) = 0$. In a discrete form this can be written as:

$$v_{y,0} = \frac{v_{y,1} \rho_1}{\rho_0} \quad (3.35)$$

Where the value in the first ghostcell is indicated with index 0.

For the radiation energy it is not only important that E has the correct value, even more important is the value for F_y . Lets begin by writing down a discrete form of the initial conditions:

$$E_{i+1} = \frac{3\Gamma_{i+1}}{1 - \Gamma_{i+1}} \quad (3.36)$$

$$\Gamma_{i+1} = \frac{1}{1 + \frac{3p_{i+1}}{i+1}} \quad (3.37)$$

Using the definition of the Eddington parameter $\Gamma_{i+1} = \frac{\kappa F_{y,i+1}}{cg_{grav}}$ and a discrete form of the fld closure relation equation (1.34) $F_{i+1} = -\frac{c\lambda_{i+1}}{\kappa\rho_{i+1}} \frac{E_{i+2} + E_i}{2\Delta x}$ one can write down another expression for Γ :

$$\Gamma_{i+1} = \frac{\lambda_{i+1}}{g_{grav}\rho_{i+1}} \frac{E_i + E_{i+2}}{2\Delta x} \quad (3.38)$$

Eliminating the Eddington parameter in equations 3.37 and (3.38) makes it possible to give an elegant expression for E_i :

$$E_i = E_{i+2} + \frac{2\Delta x}{1 + \frac{3p_{i+1}}{i+1}} \frac{g_{grav}\rho_{i+1}}{\lambda_{i+1}} \quad (3.39)$$

This expression gives a radiation energy field which leads to a physical flux when calculating it from the closure relation. On the lower boundary, a "no inflow" condition is used. This means that the gradient of the densities is preserved as long as it is smaller than 0. Periodic boundary conditions are used on the sides.
STILL HAVE TO CHECK LAST STATEMENT!!!

After every time step, the gas energy density is set as function of the gas density and momentum to match the constant temperature structure to keep the atmosphere isothermal. This is done using the inner boundary condition module in **MPI-AMRVAC**.

The parameters defining the physics of the system are the Eddington parameter Γ , the optical depth at the lower boundary τ_0 and the sound speed c_{sound} . The Eddington parameter contains information about the mass, luminosity and radius of the star, it determines whether the atmosphere will be blown away, collapsing in on itself or relax in a steady state. τ_0 Sets the physical lower boundary of the numerical domain. A high value means the model starts at in a high density environment near the core, a low value means the model simulates the outermost boundaries of the star. The sound speed determines the temperature of the atmosphere and the velocity at which waves can travel. Together with the Mass which sets the gravitational field, the sound speed also determines the scale height of the gas.

Calculations are done on a Cartesian grid, with the y-direction parallel to the radius of the star. The mass of the star is chosen at $M_* = 50M_\odot$. Using Leavitt's law, we get a luminosity for a typical $50M_\odot$ star of $L_* = ...L_\odot$ and an Eddington parameter $\Gamma = ...$. The calculation domain begins at $\tau_0 = ...$ and the grid resolves about ... scale heights in the y-direction and ... in the x-direction. The resolution of the grid is chosen such that there are ... cells per scale height. The simulation stops after ... times the hydrodynamical timescale, this is the time needed for a sound wave to travel across the computational domain.

constant flux discrepancy

3.2.4 Strange mode instabilities

3.2.5 Non-isothermal evolution?

This section describes the result of evolving the initial conditions described in section 3.2.3 without the isothermal conditions. So now, also the radiation heating and cooling source terms are taken into account for the gas energy equation. Snapshots in the simulation are compared visually to snapshot from the isothermal atmosphere.

comparison to mesa structure

Chapter 4

Conclusions

qsdfqhofjqmoj

Chapter 5

Future Work

5.1 Alternative Implicit Schemes

5.2 MPI

5.3 AMR

5.4 Non-Isothermal atmospheres

5.5 Super Eddington Limit

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