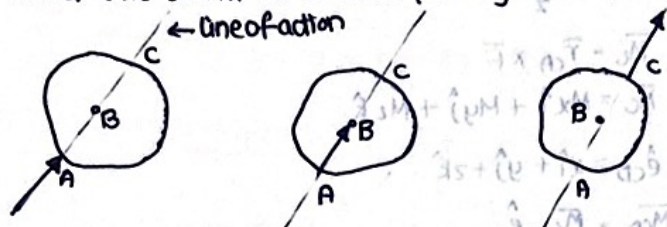


SYSTEM OF FORCES

- coplanar
 - concurrent
 - parallel
 - general (non-concurrent and non-parallel)
- non-coplanar (space forces)

(1) principle of transmissibility of force: a force being a sliding vector will not affect state of a rigid body (whether at rest or in motion) if force acts from a different point along its line of action.
eg. in train, engine can be located at front pulling other cars with it or at back pushing them forward.



(2) sign conventions: $\uparrow (+)$, $\downarrow (-)$

(3)

$$R = \sqrt{(\Sigma F_x)^2 + (\Sigma F_y)^2}$$

$$\tan \alpha = \frac{\Sigma F_y}{\Sigma F_x}$$

(4) Varignon's Theorem:

algebraic sum of moments of a system of coplanar forces about any point in plane is equal to moment of resultant force of system about same point.

(5) moment of a force: $M = F \times d$ unit: N-m

(6) sign conventions: anticlockwise (+)
clockwise (-)

(7)

$$R = \sqrt{P^2 + Q^2 + 2PQ \cos \alpha}$$

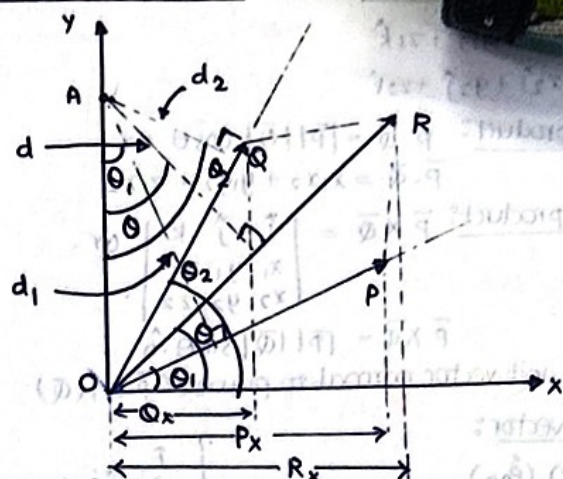
$$\tan \theta = \frac{Q \sin \alpha}{P + Q \cos \alpha}$$

(8) couple: two non-collinear parallel forces of equal magnitude but opp. in directions form couple.
- causes rotation of body.

$$M = F \times d$$

(9) $\Sigma M_O = |R \times d| = |\Sigma F_x \times y| = |\Sigma F_y \times x|$

Varignon's Theorem



Let P and Q be two concurrent forces at O, making angle θ_1 and θ_2 with x-axis, let R be their resultant making angle θ with x-axis.
Let A be a point on y-axis about which we shall find moments of P and Q and also of resultant R.
Let d_1 , d_2 and d be moment arm of P, Q and R from moment centre A.

Let x component of forces P, Q and R be P_x , Q_x and R_x respectively.

moment of P about A = $M_A^P = P \times d_1$ — (1)

" Q " = $M_A^Q = Q \times d_2$ — (2)

" R " = M_A^R

= $R \times d$

= $R \times (OA \cos \theta)$

= $OA (R_x)$

adding (1) and (2),

$$M_A^P + M_A^Q = P d_1 + Q d_2$$

or sum of moments

$$\Sigma M_A^F = + (P \times OA \cos \theta_1) + (Q \times OA \cos \theta_2)$$

$$= OA \cdot P_x + OA \cdot Q_x \quad (P_x = P \cos \theta_1)$$

$$= OA (P_x + Q_x) \quad (Q_x = Q \cos \theta_2)$$

$$\Sigma M_A^F = OA (R_x) \quad \text{--- (4)}$$

($P_x + Q_x = R_x$, since the resultant of forces in x direction, i.e. sum of components of forces in x direction)

Comparing (4) with (3),

$$\Sigma M_A^F = M_A^R$$

Hence proved.

EM

module 1 \rightarrow ch. 2, 7
module 2 \rightarrow ch. 9, 13
module 3 \rightarrow ch. 6
module 4 \rightarrow ch. 3, 4
module 5 \rightarrow ch. 10, 11, 12

Non-coplanar forces:

$$(1) \vec{P} = x_1\hat{i} + y_1\hat{j} + z_1\hat{k}$$

$$\vec{Q} = x_2\hat{i} + y_2\hat{j} + z_2\hat{k}$$

dot product: $\vec{P} \cdot \vec{Q} = |\vec{P}| |\vec{Q}| \cos \theta$ or

$$\vec{P} \cdot \vec{Q} = x_1x_2 + y_1y_2 + z_1z_2$$

cross product: $\vec{P} \times \vec{Q} = \begin{vmatrix} \hat{i} & \hat{j} & \hat{k} \\ x_1 & y_1 & z_1 \\ x_2 & y_2 & z_2 \end{vmatrix}$ or

$$\vec{P} \times \vec{Q} = |\vec{P}| |\vec{Q}| \sin \theta \cdot \hat{n}$$

(\hat{n} → unit vector normal to plane of \vec{P} and \vec{Q})

(2) Force vector:

$$\vec{F} = (F) \cdot (\hat{e}_{AB})$$

$$\vec{F} = Fx\hat{i} + Fy\hat{j} + Fz\hat{k}$$

$$\vec{F} = (F) \left[\frac{(x_2 - x_1)\hat{i} + (y_2 - y_1)\hat{j} + (z_2 - z_1)\hat{k}}{\sqrt{(x_2 - x_1)^2 + (y_2 - y_1)^2 + (z_2 - z_1)^2}} \right]$$

(where \hat{e}_{AB} → unit vector in direction of AB)

(3) magnitude of force and direction angles:

$$|\vec{F}| \text{ or } F = \sqrt{F_x^2 + F_y^2 + F_z^2}$$

$$F_x = F \cos \theta_x, F_y = F \cos \theta_y, F_z = F \cos \theta_z$$

By cosine rule,

$$\cos^2 \theta_x + \cos^2 \theta_y + \cos^2 \theta_z = 1$$

(4) moment vector:

$$\vec{F} = Fx\hat{i} + Fy\hat{j} + Fz\hat{k}$$

$$\vec{r}_{CA} = (x_1 - x_3)\hat{i} + (y_1 - y_3)\hat{j} + (z_1 - z_3)\hat{k} \quad \left\{ \begin{array}{l} \text{Position} \\ \text{vector} \end{array} \right.$$

$$\vec{M}_C = \vec{r}_{CA} \times \vec{F} = \begin{vmatrix} \hat{i} & \hat{j} & \hat{k} \\ x & y & z \\ F_x & F_y & F_z \end{vmatrix} \quad \left\{ \begin{array}{l} \text{moment} \\ \text{vector} \end{array} \right.$$

$$\vec{M}_C = M_x\hat{i} + M_y\hat{j} + M_z\hat{k}$$

(5) Vector component of force along given line:

$$\vec{F} = (F) \cdot (\hat{e}_{AB})$$

unit vector:

$$\hat{e}_{CD} = \frac{(x_4 - x_3)\hat{i} + (y_4 - y_3)\hat{j} + (z_4 - z_3)\hat{k}}{\sqrt{(x_4 - x_3)^2 + (y_4 - y_3)^2 + (z_4 - z_3)^2}}$$

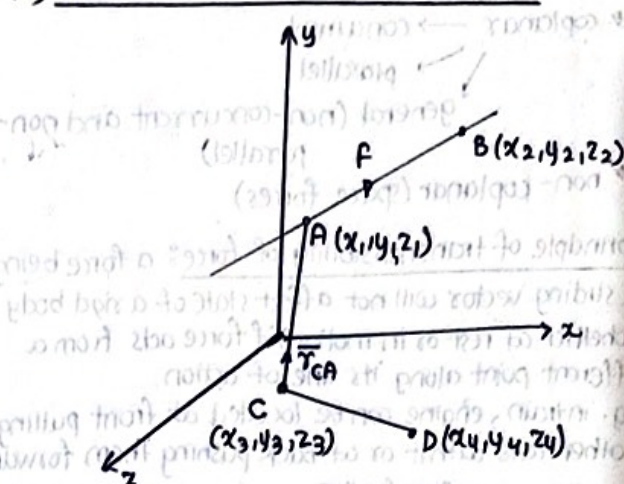
$$\hat{e}_{CD} = x\hat{i} + y\hat{j} + z\hat{k}$$

$$F_{CD} = \vec{F} \cdot \hat{e}_{CD} \rightarrow \text{scalar component}$$

$$F_{CD} = F_x \cdot x + F_y \cdot y + F_z \cdot z$$

$$\vec{F}_{CD} = (F_{CD}) \cdot (\hat{e}_{CD}) \rightarrow \text{vector component}$$

(6) moment of a force about a given line:



$$\vec{M}_C = \vec{r}_{CA} \times \vec{F}$$

$$\vec{M}_C = M_x\hat{i} + M_y\hat{j} + M_z\hat{k}$$

$$\hat{e}_{CD} = x\hat{i} + y\hat{j} + z\hat{k}$$

$$\vec{M}_{CD} = \vec{M}_C \cdot \hat{e}_{CD}$$

$$M_{CD} = M_x \cdot x + M_y \cdot y + M_z \cdot z$$

$$\vec{M}_{CD} = (M_{CD}) \cdot (\hat{e}_{CD})$$

$$(7) \cos^2 \theta_x + \cos^2 \theta_y + \cos^2 \theta_z = 1$$

(8)



$$\vec{b} \times \vec{r} = \vec{M}$$

(+) clockwise direction
(-) anticlockwise



$$R = \sqrt{F_x^2 + F_y^2 + F_z^2}$$

$$\vec{b} \times \vec{r} = \vec{M}$$

couple: two non-collinear parallel forces of equal magnitude but opposite direction forming a couple.

$$\vec{b} \times \vec{r} = \vec{M}$$

Module 1 → 1 solution
Module 2 → 2 solutions
Module 3 → 3 solutions
Module 4 → 4 solutions
Module 5 → 5 solutions

KINEMATICS OF A PARTICLE

(1) equations of motion: (uniform accel'n motion)

$$v = u + at$$

$$v^2 = u^2 + 2as$$

$$s = ut + \frac{1}{2}at^2$$

(2) average velocity: $V_{av} = \frac{\Delta x}{\Delta t}$

$$\text{instantaneous velocity: } v = \lim_{\Delta t \rightarrow 0} \frac{\Delta x}{\Delta t} = \frac{dx}{dt}$$

$$\text{average speed: } s = \frac{\text{distance travelled}}{\text{time interval}}$$

(3) average accel'n: $a_{av} = \frac{\Delta v}{\Delta t}$

$$\text{instantaneous accel'n: } a = \lim_{\Delta t \rightarrow 0} \frac{\Delta v}{\Delta t} = \frac{dv}{dt}$$

(4) (uniform velocity motion) $v = \frac{s}{t}$

(5) (variable accel'n motion) $v = \frac{dx}{dt}, a = \frac{dv}{dt}$

$$\frac{a}{v} = \frac{dv}{dx} \rightarrow a = v \frac{dv}{dx}$$

(6) For a uniformly accelerated rectilinear moving particle, distance covered in nth second is,

$$s^{th} = u + a \times n - \frac{a}{2} = u + a \left(n - \frac{1}{2} \right)$$

1) motion curves:

(i) x-t curve:

$$v = \text{slope of } x\text{-}t \text{ curve}$$

(ii) v-t curve:

$$a = \text{slope of } v\text{-}t \text{ curve}$$

$$x_f = x_i + (\text{area under } v\text{-}t \text{ curve})_{t_i \rightarrow t_f}$$

(iii) a-t curve:

$$v_f = v_i + (\text{area under curve } a\text{-}t)_{t_i \rightarrow t_f}$$

$$x_f = x_i + v_i \times t + (\text{auc } a\text{-}t)_{t_i \rightarrow t_f} \times (t - t_i) \quad [\text{centroid}]$$

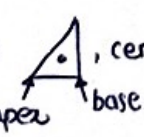
(iv) v-x curve:

$$a = v \times \text{slope of } v\text{-}x \text{ curve}$$

$$\text{* area under concave parabolic curve (up)} = \frac{1}{3} \times b \times h$$

$$\text{* area under concave (down) par. curve} = \frac{2}{3} \times b \times h$$

$$\text{* for a } \Delta, \text{ centroid from base} = \frac{\text{base}}{3}$$



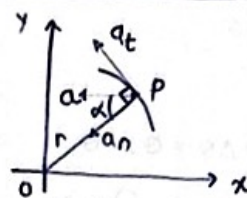
apex base

from apex = $\frac{2 \times \text{base}}{3}$

(consider from left ends)

$$\text{* km/hr} \rightarrow \text{m/s, multiply by } \frac{5}{18}$$

(6) curvilinear motion:



for x-y coor system: $r = x\hat{i} + y\hat{j} + z\hat{k}$

$$v = \frac{dr}{dt}$$

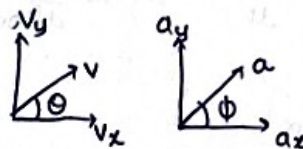
$$v = \left(\frac{dx}{dt} \right) \hat{i} + \left(\frac{dy}{dt} \right) \hat{j} + \left(\frac{dz}{dt} \right) \hat{k}$$

$$v = v_x \hat{i} + v_y \hat{j} + v_z \hat{k}$$

$$a = \frac{dv}{dt} = \frac{d^2 r}{dt^2} = (a_x) \hat{i} + (a_y) \hat{j} + (a_z) \hat{k}$$

$$v = \sqrt{v_x^2 + v_y^2} \rightarrow \theta = \tan^{-1} \left(\frac{v_y}{v_x} \right) \quad \left\{ \begin{array}{l} \text{magnitude} \\ \text{and} \\ \text{angle} \end{array} \right.$$

$$a = \sqrt{a_x^2 + a_y^2} \rightarrow \phi = \tan^{-1} \left(\frac{a_y}{a_x} \right)$$



for 't-r' coor system:

$$a_t = \frac{dv}{dt}; a_n = \frac{v^2}{\rho} \quad \left(v \rightarrow \text{vel. at given time} \right)$$

(ρ \rightarrow rad. of curvature)

$$\rho = \left[1 + \left(\frac{dy}{dx} \right)^2 \right]^{3/2} \quad (\text{if } f(y) = x)$$

at that instant $\frac{d^2 y}{dx^2}$

* constant speed, $a_t = 0$

$$\text{Total accel'n: } a = \sqrt{a_t^2 + a_n^2}$$

$$\alpha = \tan^{-1} \left(\frac{a_t}{a_n} \right)$$

$$\text{also, } \rho = \frac{[v_x^2 + v_y^2]^{3/2}}{v_x a_y - v_y a_x}, \quad \rho = \left[\frac{v^3}{a_x v_y - a_y v_x} \right]$$

($a_t \rightarrow$ tangential component of accel'n)

($a_n \rightarrow$ normal component of accel'n)

* for uniform speed curvilinear motion, $a_t = 0$

* for speed changing at uniform rate,

$$v = u + a_t \cdot t$$

$$s = u \cdot t + \frac{1}{2} a_t \cdot t^2$$

$$v^2 = u^2 + 2 a_t \cdot s$$

$$\vec{r} = x\hat{i} + y\hat{j} + z\hat{k}, \quad r = \sqrt{x^2 + y^2 + z^2}$$

$$\vec{v} = \frac{d\vec{r}}{dt} = v_x \hat{i} + v_y \hat{j} + v_z \hat{k}, \quad v = \sqrt{v_x^2 + v_y^2 + v_z^2}$$

$$\vec{a} = \frac{d\vec{v}}{dt} = a_x \hat{i} + a_y \hat{j} + a_z \hat{k}, \quad a = \sqrt{a_x^2 + a_y^2 + a_z^2}$$

$$\tan \theta_r = \frac{y}{x}, \quad \tan \theta_v = \frac{v_y}{v_x}, \quad \tan \theta_a = \frac{a_y}{a_x}$$

(9) Instantaneous Centre Method (ICR):

* 1 revolution = 2π radians = 360

* $1 \text{ rpm} = \frac{2\pi}{60} \text{ rad/s}$

(i) angular displacement: θ or $\Delta\theta = \theta_2 - \theta_1$

angular velocity: $\omega = \frac{d\theta}{dt}$ \nearrow +ve \searrow -ve

angular accn: $\alpha = \frac{d\omega}{dt}$

uniform angular vel. motion $\rightarrow \omega = \frac{\theta}{t}$

uniform angular accn motion $\rightarrow \omega = \omega_0 + \alpha t$

$$\theta = \omega_0 t + \frac{1}{2} \alpha t^2$$

$$\omega^2 = \omega_0^2 + 2\alpha\theta$$

variable angular accn motion: $\omega = \frac{d\theta}{dt}$

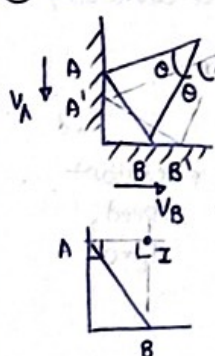
$$\alpha = \frac{d\omega}{dt} \rightarrow \alpha = \omega \cdot \frac{d\omega}{d\theta}$$

(ii) $V = r\omega$

$a_n = r\omega^2$, $a_t = r\alpha$

(iii) ICR: defined as point about which a general plane moving body rotates at any given instant

(A) LADDER:



[Translational motion.]

$$V_A = (IA) \times \omega$$

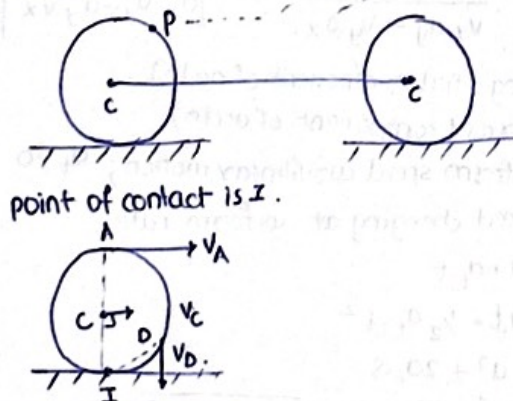
$$V_B = (IB) \times \omega$$

Steps:

(i) locate ICR

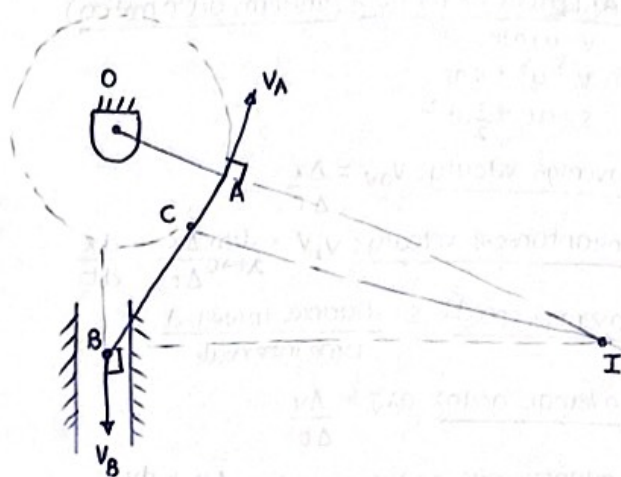
(ii) find ω , V_A , V_B .

(B) ROLLER: [Translational + Rotational]



$$V_D = (ID) \times \omega$$

(C) 2 LINKAGES: [Translational + Rotational]



OA: rotational motion.

$$V_O = 0$$

$$V_A = (OA) \omega$$

AB: rotational + translational.
plane motion.

$$V_A = (IA) \omega$$

$$V_B = (IB) \omega$$

if C is any point on the linkage,

$$V_C = (IC) \omega$$

* vel. of IR (instantaneous rotation) point is zero.

* location of ICR point varies instant to instant

* cosine rule: 2 lengths, 1 angle

(sine rule: 1 length, 2 angles)

$$\sqrt{A^2 + B^2 + 2AB \cos \theta}$$