

Problem 1

$$\vec{q}(t) = \begin{bmatrix} x(t) \\ y(t) \end{bmatrix} \quad \vec{q}(1) = \begin{bmatrix} x_E \\ y_E \end{bmatrix} \quad \vec{F} = e^{x^2 - y^2} \begin{bmatrix} -ax \\ by \end{bmatrix}$$

$$a) \quad W = \int_{\vec{q}(0)}^{\vec{q}(1)} \vec{F} d\vec{q} = \int_{t=0}^1 \vec{F}(t) \dot{\vec{q}}(t) dt = \int_0^1 e^{t^2(x_E^2 - y_E^2)} \begin{bmatrix} -atx_E \\ bty_E \end{bmatrix} dt$$

$$\vec{q}(t) = \begin{bmatrix} tx_E \\ ty_E \end{bmatrix} \quad \dot{\vec{q}}(t) = \begin{bmatrix} x_E \\ y_E \end{bmatrix} \quad \begin{bmatrix} x_E \\ y_E \end{bmatrix} dt =$$

$$= \int_0^1 e^{t^2(x_E^2 - y_E^2)} [-atx_E^2 + bty_E^2] dt =$$

$$= \int_0^1 [by_E^2 - ax_E^2] t e^{t^2(x_E^2 - y_E^2)} dt = \left[\begin{array}{l} u = t^2(x_E^2 - y_E^2) \\ du = 2t(x_E^2 - y_E^2) dt \\ u(0) = 0 \quad u(1) = x_E^2 - y_E^2 \end{array} \right] =$$

$$= \int_0^{x_E^2 - y_E^2} \frac{(by_E^2 - ax_E^2)}{2(x_E^2 - y_E^2)} e^u du = \frac{by_E^2 - ax_E^2}{2(x_E^2 - y_E^2)} [e^{x_E^2 - y_E^2} - 1]$$

$$b) \quad \gamma_2 = \left\{ \vec{q}(t) = \begin{bmatrix} tx_E \\ 0 \end{bmatrix} \right\} \cup \left\{ \vec{q}(t) = \begin{bmatrix} x_E \\ ty_E \end{bmatrix} \right\}$$

$$W = \int_{t=0}^1 e^{t^2x_E^2} \begin{bmatrix} -atx_E \\ 0 \end{bmatrix} \begin{bmatrix} x_E \\ 0 \end{bmatrix} dt + \int_{t=0}^1 e^{x_E^2 - t^2y_E^2} \begin{bmatrix} -ax_E \\ bty_E \end{bmatrix} \begin{bmatrix} 0 \\ y_E \end{bmatrix} dt =$$

$$= \int_{t=0}^1 e^{t^2x_E^2} (-atx_E^2) dt + \int_{t=0}^1 e^{x_E^2 - t^2y_E^2} (bty_E^2) dt = W_1$$

$$= \frac{-a}{2} e^{t^2x_E^2} \Big|_0^1 + \frac{b}{2} e^{x_E^2 - t^2y_E^2} \Big|_0^1 = \underbrace{-\frac{a}{2} [e^{x_E^2} - 1]}_{W_1} - \underbrace{\frac{b}{2} [e^{x_E^2 - y_E^2} - e^{x_E^2}]}_{W_2}$$

$$*c) \quad W_a \text{ must (at least)} = [W_b = W_1 + W_2]$$

$$\text{So } \frac{b y e^{x^2 - a x^2}}{2(x e^2 - y e^2)} \left[e^{\frac{x e^2 - y e^2}{A}} - 1 \right] = -\frac{a}{2} \left[e^{x e^2} - 1 \right] - \frac{b}{2} \left[e^{x e^2 - y e^2} - e^{x e^2} \right]$$

For same endpoint $[x_e, y_e]$.

$$\text{If } \underline{a=b}, \quad -\frac{a}{2} [e^A - 1] = -\frac{a}{2} [e^A - 1].$$

Necessary (not sufficient): $a=b$.

*d) but when $a=b$ \Rightarrow

$$\begin{aligned} -\nabla \Phi(x, y) &= \begin{bmatrix} -\frac{\partial}{\partial x} \Phi \\ -\frac{\partial}{\partial y} \Phi \end{bmatrix} = \begin{bmatrix} -2xc e^{x^2 - y^2} \\ 2yc e^{x^2 - y^2} \end{bmatrix} = \\ &= e^{x^2 - y^2} \begin{bmatrix} -2cx \\ 2cy \end{bmatrix}, \quad \boxed{c = \frac{1}{2}a} \end{aligned}$$

$$\text{e) } \Phi(x, y(x)) = C = c e^{x^2 - y^2}$$

$$A = e^{x^2 - y^2}$$

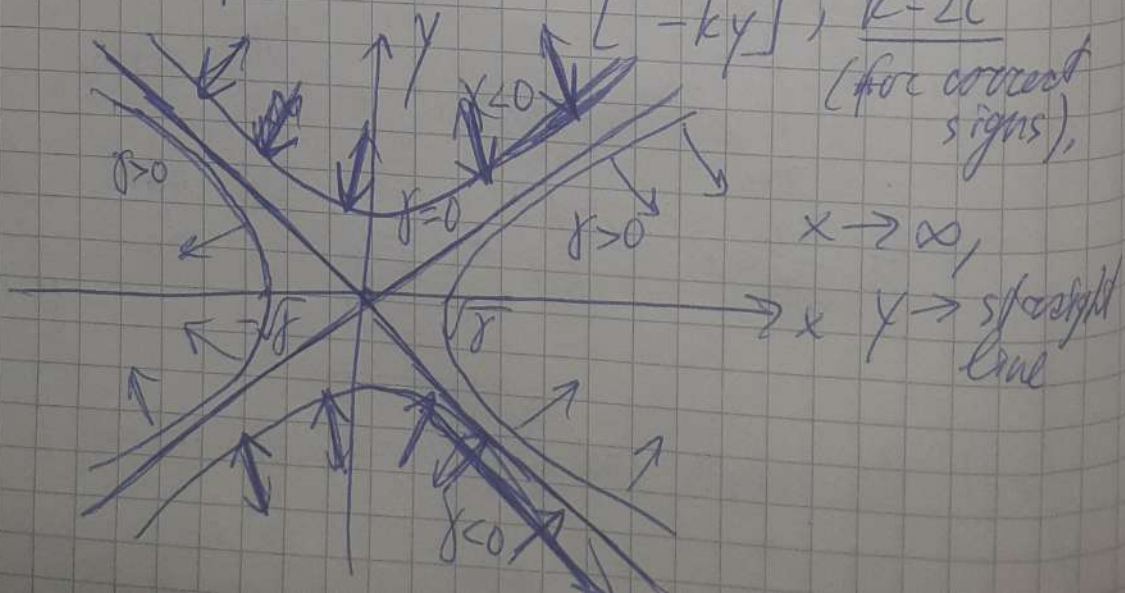
$$e^{y^2} = e^{x^2} \quad \text{or}$$

$$y^2 = x^2 + r, \quad y = \pm \sqrt{x^2 + r}$$

If

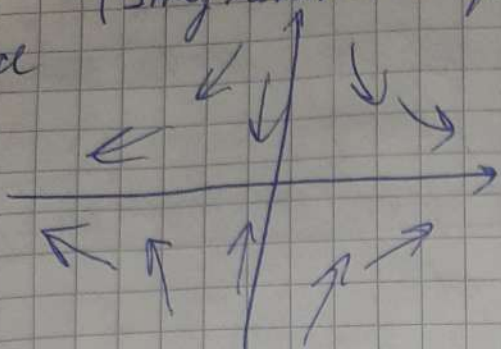
$$\nabla \Phi(x, y) = -F = e^{x^2 - y^2} \begin{bmatrix} -kx \\ -ky \end{bmatrix}, \quad \underline{k=2c}$$

(for correct signs),



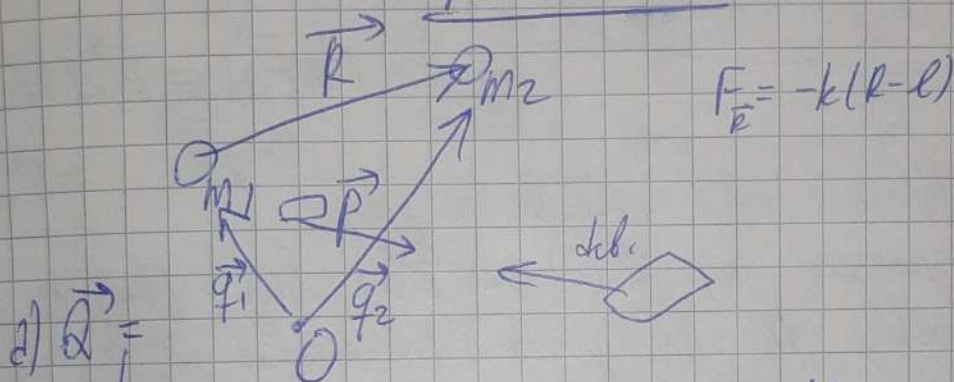
(Imag. continuous potential & gradient)

Since



(for c.c.o. situation directions are inverted).

Problem 2



a) $\vec{Q} =$

$$\frac{m_1 \vec{q}_1(t) + m_2 \vec{q}_2(t)}{\mu}, \text{ but } \ddot{\vec{Q}}(t) = \frac{m_1 \ddot{\vec{q}}_1 + m_2 \ddot{\vec{q}}_2}{\mu} =$$

$$\text{So } \vec{Q}(t) = \vec{Q}_0 + \int_0^t \ddot{\vec{Q}} dt = \frac{\vec{F} - \vec{F}}{\mu} = \vec{0}.$$

$$= \vec{Q}_0 + \int_0^t \vec{Q}_0 dt = \vec{Q}_0 + \vec{Q}_0 t = \vec{Q}_0$$

CM does not move.

EOM $\vec{Q} = \vec{0}$

— enough to describe only relative motion

b) $\vec{R}(t) = \vec{q}_2(t) - \vec{q}_1(t)$

(after describe motion)
Moreover:

1) $m_1 \vec{q}_1 + m_2 \vec{q}_2 = \vec{0}$

2) $\vec{R} = \vec{q}_2 - \vec{q}_1 \Rightarrow$

knowing $\vec{R}(t)$ describes everything.

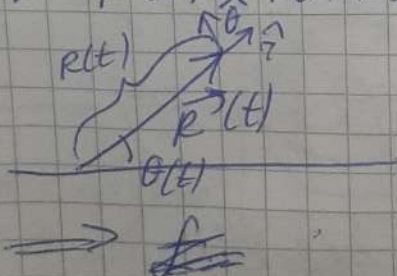
Our further task is just to describe $\vec{R}(t)$ — not even $\vec{R}(t)$, because motion will be in the

$\vec{L} = \vec{0} \Rightarrow$ it will be in the same plane (we choose that plane).
 Full conclusion — 1DOF problem for $R(t)$
 (after showing below) (due to $\vec{L} = \vec{0}$, see below).

b) $\vec{L} = \vec{R}(t) \times \mu \vec{R}(t)$, $\left[\frac{d\vec{L}}{dt} = \underbrace{\vec{R} \times \mu \dot{\vec{R}}}_{\vec{0}} + \underbrace{\vec{R} \times \mu \ddot{\vec{R}}}_{\vec{F}_1 (= -\vec{F}_2)} = \vec{R} \times (K\vec{R}) = \vec{0} \right]$
 [easily shown $\mu \ddot{\vec{R}} = \frac{m_1 m_2}{m_1 + m_2} (\ddot{\vec{q}}_2 - \ddot{\vec{q}}_1) \equiv \ddot{\vec{R}}$ because $(\vec{Q}_{cm} = \vec{0})$,
 $\frac{m_1 m_2 \ddot{\vec{q}}_2}{m_1 + m_2} - \frac{m_1 m_2 \ddot{\vec{q}}_1}{m_1 + m_2} = \frac{m_1 m_2 \ddot{\vec{q}}_2 + m_2^2 \ddot{\vec{q}}_2}{m_1 + m_2} = \frac{m_2 \ddot{\vec{q}}_2}{1} = \ddot{\vec{R}} = -k(R-l)\hat{R}$

* c) ① \vec{q}_1 and \vec{q}_2 (since $\vec{Q}_{cm} = \vec{0}$)
 lie in a plane, and ② \vec{q}_1 and \vec{p} lie in plane (by definition momentum conserv.) and
 ③ \vec{p} and debris lie in plane (by definition of task) and ④ \vec{q}_1 and \vec{q}_2 determine $\vec{R} \Rightarrow$
 motion must be in the same plane with m_1, m_2 , debris
initially and ⑤ it will stay there because
 $\vec{L} = \vec{0}$.

d) $\vec{R}(t) = R(t) \hat{R}(t)$, $\dot{\vec{R}}(t) = \dot{R} \hat{R} + R \dot{\theta} \hat{\theta}$.



$T = \frac{m(\dot{\vec{R}})^2}{2} = \frac{m}{2} [\dot{R}^2 + R^2 \dot{\theta}^2]$

$V = \frac{k}{2} (R-l)^2$

$$\Rightarrow \mathcal{L}(R, \theta) = \frac{m}{2} [\dot{R}^2 + R^2 \dot{\theta}^2] - \frac{k}{2} (R-l)^2$$

(xx) for θ : (θ is cyclic)

$$\frac{\partial \mathcal{L}}{\partial \theta} = 0 = \frac{d}{dt} \left(\frac{\partial \mathcal{L}}{\partial \dot{\theta}} \right) = m R^2 \dot{\theta}$$

for R :

$$\frac{\partial \mathcal{L}}{\partial R} = m R \dot{\theta}^2 - k(R-l) = 0$$

$$\vec{R} = R(t) \hat{e}(t) = \vec{R}(R, \theta)$$

$$\vec{L} = 0$$

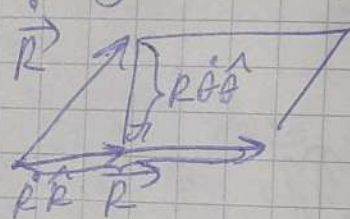
$$= \frac{d}{dt} \left(\frac{\partial \mathcal{L}}{\partial \dot{R}} \right) = \frac{d}{dt} [m \dot{R}] = m \ddot{R} \quad (*)$$

~~Or alternatively for $\vec{R}(t)$:~~ Lagrange is not used in this task anymore.

$$T = \frac{m}{2} \dot{\vec{R}}^2 \quad V = \frac{k}{2} (R-l)^2$$

e) But $\vec{L} = 0, \dot{E} = 0 \Rightarrow$

$$L = |\vec{L}| = \mu \cdot R \cdot R \dot{\theta} = \mu R^2 \dot{\theta}$$



$$\Rightarrow \dot{\theta} = \frac{L}{\mu R^2}, \quad (\text{same information as in (xx)})$$

~~Now from (*)~~

$$m R \dot{\theta}^2 - k(R-l) = m \ddot{R} \Rightarrow$$

$$\frac{m R \frac{L^2}{\mu^2 R^4}}{m R^4} =$$

Now from $\dot{E} = 0$,

$$\frac{d}{dt} E = \frac{d}{dt} \left(\frac{m}{2} [\dot{R}^2 + R^2 \dot{\theta}^2] + \frac{k}{2} (R-l)^2 \right) =$$

$$= \frac{d}{dt} \left(\frac{m}{2} [\dot{R}^2 + R^2 \frac{L^2}{\mu^2 R^4}] + \frac{k}{2} (R-l)^2 \right) =$$

$$= \frac{d}{dt} \left(\frac{m}{2} [\dot{R}^2 + \frac{L^2}{\mu^2 R^2}] + \frac{k}{2} (R-l)^2 \right)$$

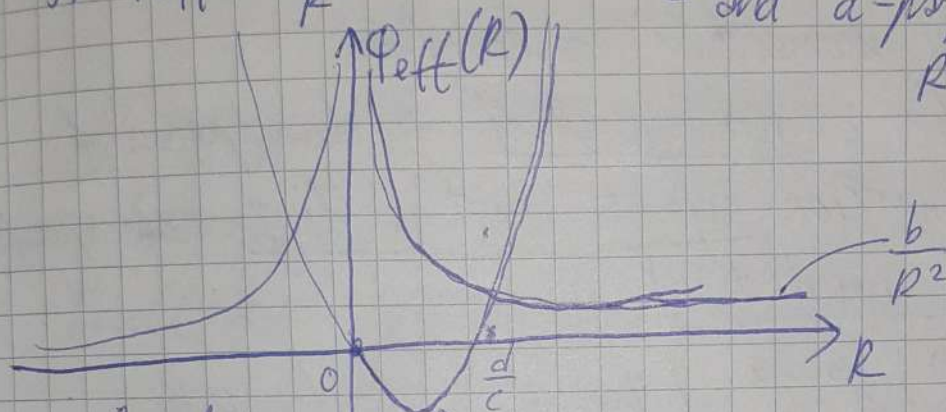
E

$$\text{So } E = \frac{M}{2} \dot{R}^2 + \frac{L^2}{2\mu R^2} + \frac{k}{2} R^2 - k l R + \frac{k l^2}{2}$$

$$\text{So } a = \frac{M}{2} \quad c = \frac{k}{2} \quad e = \frac{k l^2}{2}$$

$$b = \frac{L^2}{2\mu} \quad d = k l, \text{ but easier to look at.}$$

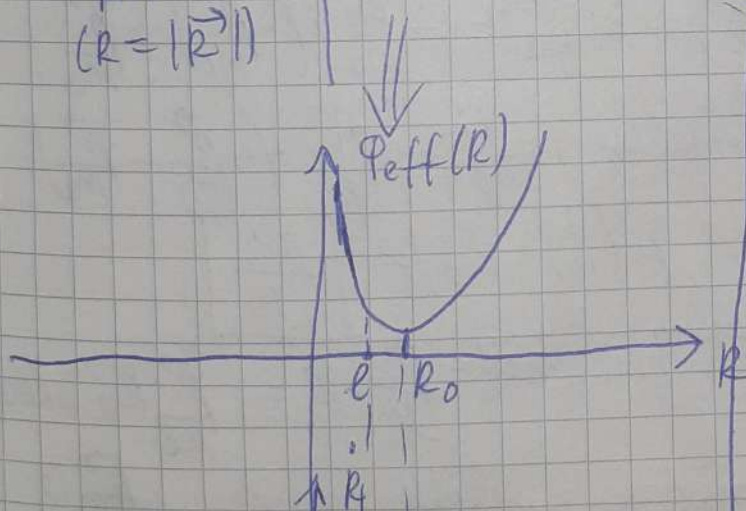
f) $\Phi_{\text{eff}} = \frac{b}{R^2} + c R^2 - d R$ [e is just constant shift, and a-priori depends on \dot{R} and not on R]



Consider just $R > 0$
($R = |\vec{R}|$)

$$c R^2 - d R$$

$$c R^2 - d R = R(c R - d) = 0$$



Check minima:

$$\Phi' = -\frac{2b}{R^3} + 2cR - d \stackrel{!}{=} 0$$

$$-2b + 2cR^4 - dR^3 = 0$$

How related to $R = l$?

$$\Phi'(l) = -2b + 2cl^4 - dl^3 =$$

$$= -\frac{L^2}{\mu} + kl^4 - kl^3 < 0$$

\Rightarrow therefore $R(\Phi_{\text{min}}) > l$, R_0

due to rotational energy impact,

Stable minimum is not at l , but at $R_0 > l$.

*g) Initial speed determined from \vec{p} , let it R_0
In phase space it is point (\vec{p}, R_0)

part * h) $\frac{b}{R^2}$ is energy to stop rotation.

So $E_{\text{stop}} = \frac{L^2}{2\mu R^2} = \frac{L_0^2}{2\mu R^2}$ (due to $\dot{L}=0, L=L_0$)

but $L_0 = \mu R_0^2 \dot{\theta}_0 = \mu l^2 \dot{\theta}_0$, So $E_{\text{stop rotation}} =$
 $= \frac{\mu^2 l^4 \dot{\theta}_0^2}{2\mu R^2} = \frac{\mu l^4}{2R^2} \dot{\theta}_0^2$.

At any given R , $E(R) = \frac{\mu l^4}{2R^2} \dot{\theta}_0^2$, where $\dot{\theta}_0$ stores dependency on \vec{p} .

Recall that $\vec{R}(0) = \frac{\vec{p}}{\mu} = \dot{R}(0)\hat{r} + R\dot{\theta}(0)\hat{\theta} =$

$= \dot{R}(0)\hat{r} + l\dot{\theta}(0)\hat{\theta}$

set up by task situation

set up by coordinate system

See above

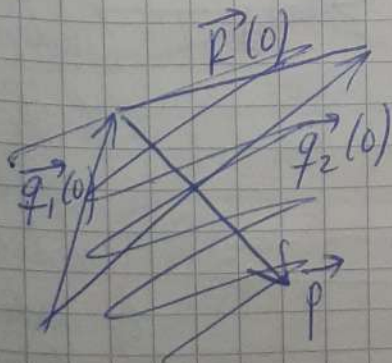
in phase space

$\frac{p^2}{\mu^2} = \dot{R}_0^2 + l^2 \dot{\theta}_0^2, \quad \dot{\theta}_0^2 = \frac{(\frac{p}{\mu})^2 - \dot{R}_0^2}{l^2}$

$\Rightarrow E(R) = \frac{\mu l^4}{2R^2} \left[\frac{(\frac{p}{\mu})^2 - \dot{R}_0^2}{l^2} \right] = \frac{\mu l^2}{2R^2} \left[\left(\frac{p}{\mu}\right)^2 - \dot{R}_0^2 \right]$

$= \frac{\mu l^2}{2}$

Now,



For given R E is minima when \dot{R}_0^2 initially was the maxima,

so $\frac{\vec{p} \cdot \hat{r}}{\mu}$ is maxima,

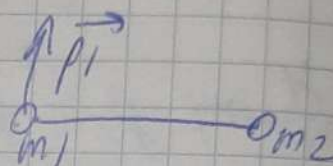
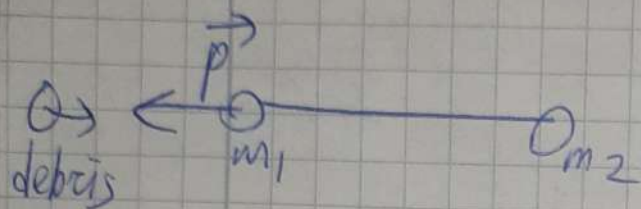
So \vec{p} is along $\vec{R}(0)$.

Which makes sense, because
then there is no rotation;

and

maximum rotation

when
↓ debris



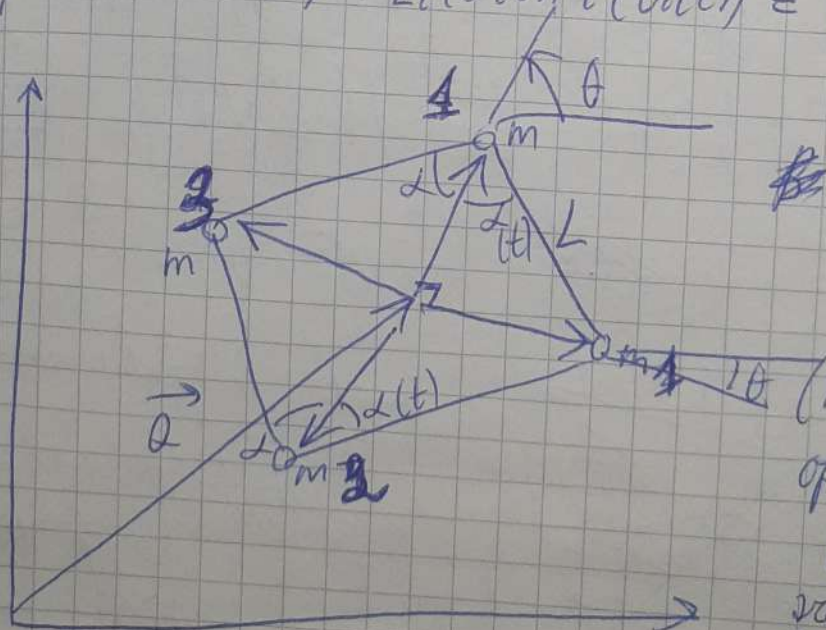
Problem 3

$$a) \quad \vec{Q}(t) = \frac{\sum_{i=1}^4 m_i \vec{q}_i}{4m} \Rightarrow 4m\vec{Q} = \sum_{i=1}^4 m_i \vec{q}_i$$

$$\Rightarrow \vec{q}_i(t) =$$

From geometry it is clear that

$$\vec{q}_i(t) = \vec{Q}(t) + L_i(t) \hat{r}(\theta_i(t)) = \vec{Q}(t) + L_i(t) \cos \theta_i(t) \hat{r}(\theta_i(t))$$



(independently
of L

$\theta_i(t \neq 0)$
are determined
by initial orienta-
tion θ)

$$\vec{q}_i = \begin{cases} \vec{Q}(t) + L \sin \theta \hat{e}(\theta_1) & (m_1) \\ \vec{Q}(t) + L \sin \theta \hat{e}(\theta_3) & (m_3) \\ \vec{Q}(t) + L \cos \theta \hat{e}(\theta_2) & (m_2) \\ \vec{Q}(t) + L \cos \theta \hat{e}(\theta_4) & (m_4) \end{cases}$$

Can be further simplified,
but that is unnecessary
for \mathcal{L}

b) $V = -Mg \vec{Q}$

c) $\vec{q}_i = \begin{cases} \vec{Q} + L \cos \theta \dot{\theta} \hat{e}(\theta_1) + L \sin \theta \dot{\theta} \hat{e}(\theta_2) \\ \vec{Q} + L \cos \theta \dot{\theta} \hat{e}(\theta_3) + L \sin \theta \dot{\theta} \hat{e}(\theta_4) \\ \vec{Q} + L \sin \theta \dot{\theta} \hat{e}(\theta_2) + L \cos \theta \dot{\theta} \hat{e}(\theta_4) \\ \vec{Q} - L \sin \theta \dot{\theta} \hat{e}(\theta_1) + L \cos \theta \dot{\theta} \hat{e}(\theta_3) \end{cases}$ because all masses have constant shifts from $\theta(t)$

$$T = \underbrace{\frac{4m\dot{\vec{Q}}^2}{2}}_{CM} + \frac{m}{2} \left[\underbrace{L^2 \cos^2 \theta \dot{\theta}^2 + L^2 \sin^2 \theta \dot{\theta}^2}_{1,3} \cdot 2 + \underbrace{L^2 \sin^2 \theta \dot{\theta}^2 + L^2 \cos^2 \theta \dot{\theta}^2}_{2,4} \cdot 2 \right]$$

$$= 2m\dot{\vec{Q}}^2 + m[L^2\dot{\theta}^2 + L^2\dot{\theta}^2] = 2m\dot{\vec{Q}}^2 + mL^2[\dot{\theta}^2 + \dot{\theta}^2]$$

d) $\mathcal{L} = T - V = \underbrace{2m\dot{\vec{Q}}^2}_A + \underbrace{mL^2\dot{\theta}^2}_B + \underbrace{mL^2\dot{\theta}^2}_C + \underbrace{Mg\vec{Q}}_D$

e) For $\theta(t)$: $\frac{\partial \mathcal{L}}{\partial \theta} = 0 = \frac{d}{dt} [2c\dot{\theta}] = 2c\ddot{\theta}$
const, $\dot{\theta} = \text{const}$

Similarly

for $L(t)$: $\frac{\partial \mathcal{L}}{\partial L} = 0 = \frac{d}{dt} [2b\dot{L}]$
 $\downarrow \dot{L} \text{ const}$

$$\begin{cases} \theta(t) = \theta_0 + \dot{\theta}t \\ L(t) = L_0 + \dot{L}t \end{cases}$$

f) Full description then:

Initial Conditions: $\theta(t_0) = 0, L(t_0) = \frac{\pi}{4}$
 $\vec{Q}(t_0) = \vec{0}, \dot{\theta}(t_0) = \Omega, \dot{L}(t_0) = \omega.$

$$L(t) = \frac{\pi}{4} + \omega(t-t_0)$$

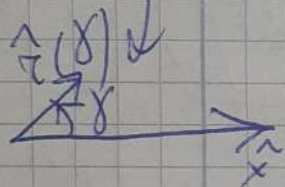
$$\theta(t) = \Omega(t-t_0)$$

CM motion:

$$\vec{Q}(t) = \vec{Q}(t_0) + \vec{Q}'(t_0)(t-t_0) + \frac{\vec{g}(t-t_0)^2}{2}$$

$$\Rightarrow \vec{Q}_i(t) = \underbrace{\vec{Q}(t_0) + \vec{V}(t-t_0) + \frac{\vec{g}(t-t_0)^2}{2}}_{\vec{Q}'(t)} + L \sin\left(\frac{\pi}{4} + \omega(t-t_0)\right)$$

using convention like



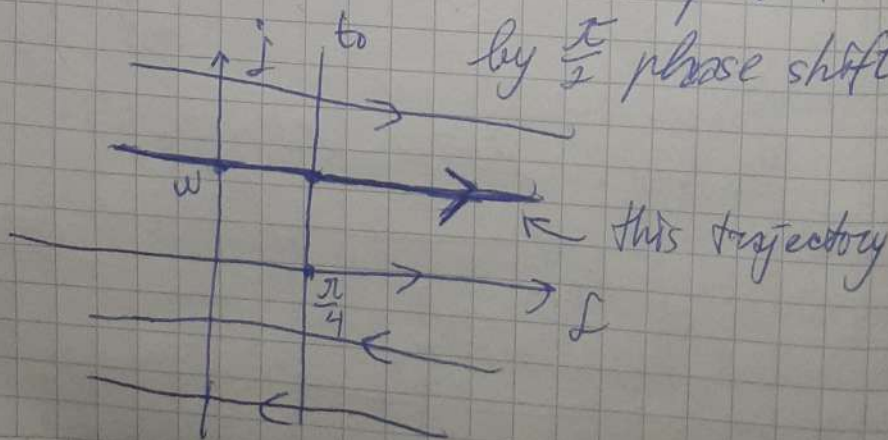
$$\vec{Q}_2(t) = \vec{Q}'(t) + L \cos\left(\frac{\pi}{4} + \omega(t-t_0)\right) \cdot \hat{e}_r$$

$$\cdot \hat{e}_r \left(\frac{\pi}{2} + \Omega(t-t_0) \right)$$

$$\vec{Q}_3(t) = \vec{Q}'(t) + L \sin\left(\frac{\pi}{4} + \omega(t-t_0)\right) \cdot \hat{e}_r \left(\frac{\pi}{2} + \Omega(t-t_0) \right)$$

$$\vec{Q}_4(t) = \vec{Q}'(t) + L \cos\left(\frac{\pi}{4} + \omega(t-t_0)\right) \cdot \hat{e}_r \left(-\Omega(t-t_0) \right)$$

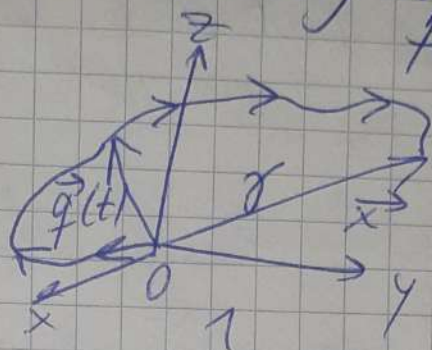
*g) CM moves in free fall,
 and corners rotate around it separated
 by $\frac{\pi}{2}$ phase shift.



$L(t)$ in this description changes linearly as in rigid body, without any additional potential.

Problem 7 - Velocity-dependent force

$$\vec{F} = \vec{A} \times \dot{\vec{q}} + \vec{B}$$



a) $\gamma = \{ \vec{q}(t) = t\vec{x} \mid 0 \leq t \leq 1 \}$

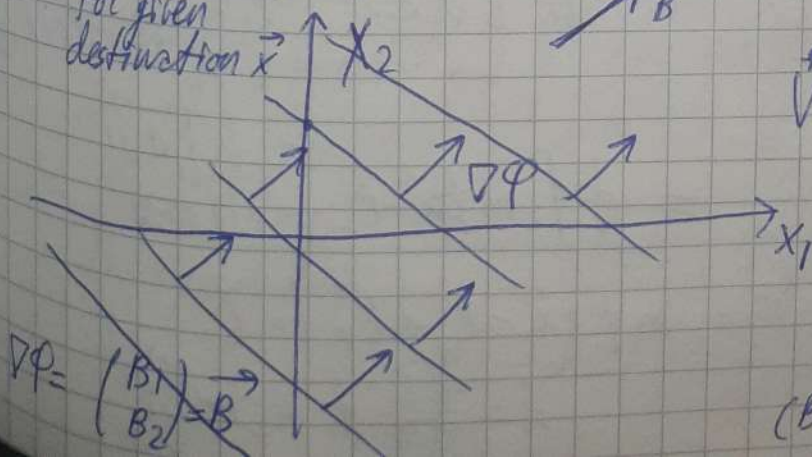
$$W = \int_{\gamma} [\vec{A} \times \dot{\vec{q}} + \vec{B}] \cdot d\vec{q} = \int_0^1 [\vec{A} \times \vec{x} + \vec{B}] \cdot \vec{x} dt =$$

$$\begin{aligned} \dot{\vec{q}}(t) &= \vec{x} \\ &= \int_0^1 \vec{x} \cdot (\vec{A} \times \vec{x}) + \vec{B} \cdot \vec{x} dt = \\ &= \int_0^1 \underbrace{\vec{A} \cdot (\vec{x} \times \vec{x})}_0 + \vec{B} \cdot \vec{x} dt = \vec{B} \cdot \vec{x} \cdot 1. \end{aligned}$$

*b) $W(\vec{x}) = -\frac{1}{2} \vec{B} \cdot \vec{x}$

$$\begin{aligned} W(\vec{x}) &= \int_{t(\vec{0})}^{t(\vec{x})} [\vec{A} \times \dot{\vec{q}} + \vec{B}] \cdot \dot{\vec{q}} dt = \int_{t(\vec{0})}^{t(\vec{x})} \vec{B} \cdot \dot{\vec{q}} dt = \\ &= \vec{B} \int_{\vec{0}}^{\vec{x}} d\vec{q} = \vec{B} \cdot \vec{x}, \text{ always} \end{aligned}$$

c) But then ~~for~~
for given destination \vec{x}



$$W(\vec{x}) = \vec{B} \cdot \vec{x} = \text{const}$$

$$B_1 x_1 + B_2 x_2 = C$$

straight lines
(\vec{B} is constant)

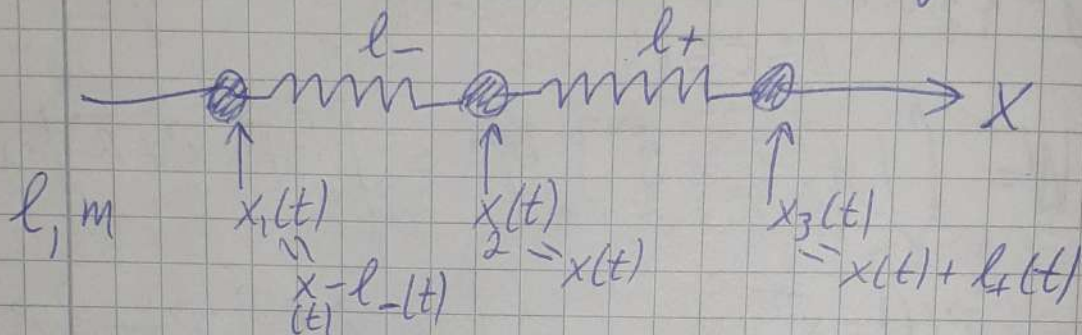
$$x_2 = \frac{C}{B_2} - \frac{B_1}{B_2} x_1$$

($B_2, B_1 > 0$)

*d) It is strictly speaking not conservative because $\nabla\phi = \vec{B} \neq \vec{F}$.

But it is conservative when $\vec{A} = \vec{0}$ and there is no velocity dependence.

Problem 6 Springs



$$\begin{aligned}
 a) \quad T &= \frac{m}{2} (\dot{x}_1^2 + \dot{x}_2^2 + \dot{x}_3^2) = \frac{m}{2} (\dot{x} - \dot{l}_-)^2 + \dot{x}^2 + (\dot{x} + \dot{l}_+)^2 \\
 &= \frac{m}{2} (\dot{x}^2 + \dot{l}_-^2 - 2\dot{x}\dot{l}_- + \dot{x}^2 + \dot{x}^2 + \dot{l}_+^2 + 2\dot{x}\dot{l}_+) \\
 &= \frac{m}{2} (3\dot{x}^2 - 2\dot{x}\dot{l}_- + 2\dot{x}\dot{l}_+ + \dot{l}_-^2 + \dot{l}_+^2) \\
 V &= \frac{k}{2} [(l - l_-)^2 + (l - l_+)^2]
 \end{aligned}$$

b) $\mathcal{L} = T - V$,
 x is cyclic.

Then $\frac{\partial \mathcal{L}}{\partial x} = 0 = \frac{d}{dt} \left[\frac{\partial \mathcal{L}}{\partial \dot{x}} \right] = \frac{d}{dt} \left[\frac{m}{2} [2(\dot{x} - \dot{l}_-) + 2\dot{x} + 2(\dot{x} + \dot{l}_+)] \right]$

$$= \frac{d}{dt} [m(\dot{x} - \dot{l}_- + \dot{l}_+)] = \frac{d}{dt} [0] = 0$$

So

$$m[3\dot{x} - \dot{l}_- + \dot{l}_+] = C$$

$$\dot{x} = \frac{1}{3} \left(\frac{C}{m} + \dot{l}_- + \dot{l}_+ \right)$$

So $\ddot{x}' = -\frac{1}{3}(\ddot{l}_+'' + \ddot{l}_-'')$

$\downarrow c$

Center of mass moves with constant velocity.

Enough to describe 2DOF - l_+ and l_- .

c) From \mathcal{L} :

1) for l_-

$$\frac{\partial \mathcal{L}}{\partial l_-} = \frac{m}{2} [-2(\dot{x}' - \dot{l}_-')] + \frac{k}{2} \cdot 2[l - l_-] =$$

$$= -\frac{k}{2} \cdot 2(l - l_-)(-1) = k(l - l_-) =$$

$$= \frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{l}_-'} = \frac{d}{dt} \left[\frac{m}{2} \cdot 2(\dot{x}' - \dot{l}_-')(-1) \right] =$$

$$= \frac{d}{dt} [-m(\dot{x}' - \dot{l}_-')] =$$

$$= m\ddot{l}_-'' - m\ddot{x}'$$

$$\text{So } k(l - l_-) = m\ddot{l}_-'' - m\ddot{x}'$$

$$\Rightarrow \frac{k(l - l_-)}{m} = \ddot{l}_-'' - \underbrace{\ddot{x}'}_{c(\ddot{l}_+'' - \ddot{l}_-'')} \Rightarrow$$

$$\ddot{l}_-'' [c+1] - c\ddot{l}_+'' = -\frac{k}{m}(l_- + l_+)$$

$$\Rightarrow \left[\ddot{l}_-'' - c(\ddot{l}_+'' + \ddot{l}_-'') = -\omega^2(l_- - l_+), \right.$$

$$\left. \begin{array}{l} c = \frac{1}{3}, \\ \omega = \sqrt{\frac{k}{m}} \end{array} \right]$$

By symmetry ...

$$\ddot{l}_+'' + c(\ddot{l}_+'' - \ddot{l}_-'') = -\omega^2(l_+ - l_-)$$

□

$$d) \Delta = \frac{(l_+ - l_-)}{l} \quad \Sigma = \frac{(l_+ + l_- - 2l)}{l}$$

$$\tau = \omega(t - t_0)$$

$$\Delta' = \frac{l_+' - l_-'}{l} \quad \Sigma' = \frac{l_+' + l_-'}{l}$$



for $\Delta(\tau)$:

$$\begin{cases} l_+' + cl\Delta' = -\omega^2(l_+ - l) & (*) \\ l_-' - cl\Delta' = -\omega^2(l_- - l) & (**) \end{cases}$$

$$2\Delta' + 2cl\Delta' = -\omega^2 \Delta$$

$$\Delta'(1 + 2c) = -\omega^2 \Delta$$

In new units

 ~~Δ'~~

$$\Delta' = -\frac{1}{1+2c} \Delta = -\frac{3}{5} \Delta$$

then $c = -\frac{1}{3}$ not $\frac{1}{3}$

(changed before)

(not -3 - then due to arithmetic errors I should got $c = -\frac{1}{3}$ before)

$$\Delta' \cdot \frac{1}{3} = -\Delta \Rightarrow \Delta' = -3\Delta$$

(2) (2)

for $\Sigma(\tau)$

$$\Sigma' = \frac{1}{l} [-\omega^2(l_+ - l) - \omega^2(l_- - l)] =$$

$$= \frac{1}{l} [\omega^2] [2l - (l_+ + l_-)] =$$

$$= -\omega^2 \Sigma$$

(now adding (*) and (**))

$\Sigma' + 2c\Sigma$

in τ units $\Sigma_i(\tau) = -\Sigma_i(\tau)$

e) But then $\Delta(\tau) = A \cos(\sqrt{3}\tau + \varphi_0)$
 f) $\Sigma_i(\tau) = B \cos(\tau + \varphi_1)$ } EOM $\Delta(0) = A \cos \varphi_0$
 $\Delta'(0) = -A\sqrt{3} \sin \varphi_0$
 $\Sigma(0) = B \cos \varphi_1$
 $\Sigma'(0) = -B \sin \varphi_1$ } Case f)

$$\Delta(0) = \frac{l_+(0) - l_-(0)}{L}$$

$$\Delta'(0) = \frac{\dot{l}_+(0) - \dot{l}_-(0)}{L}$$

$$\Sigma(0) = \frac{l_+(0) + l_-(0) - 2l}{L}$$

$$\Sigma'(0) = \frac{\dot{l}_+(0) + \dot{l}_-(0)}{L}$$

Case e)

$$0$$

$$0$$

$$\frac{2(l+L) - 2l}{L} = \frac{2L}{L}$$

$$0$$

$$\Delta(\tau) = A \cos(\sqrt{3}\tau + \varphi_0)$$

$$\dot{\Delta}(\tau) = -As$$

$$\Sigma_i(\tau) = B \cos(\tau + \varphi_1)$$

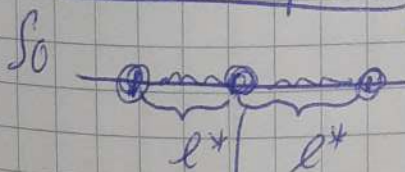
$$\Delta(\tau) = 0$$

$$B \cos \varphi_1 = \frac{2L}{L}$$

$$-B \sin \varphi_1 = 0, \varphi_1 = 0$$

$$\Sigma(\tau) = \frac{2L}{L} \cos(\tau)$$

Conclusion for e)



System CM moves uniformly
 and ~~edge~~ left/right masses
 oscillate with same displace-
 ments.

With different
 frequencies / periods

$$\frac{l+L-l}{L} = \frac{L}{L}$$

$$0$$

$$\frac{l+L+l-2l}{L} = \frac{L}{L}$$

$$0$$

$$\Delta(\tau) = A \cos \varphi_0 = \frac{L}{L}$$

$$-A\sqrt{3} \sin \varphi_0 = 0$$

$$\varphi_0 = 0$$

$$\Delta(\tau) = \frac{L}{L} \cos(\tau)$$

$$B \cos \varphi_1 = \frac{L}{L}$$

$$\Sigma_i(\tau) = \frac{L}{L} \cos(\tau)$$

Conclusion for f)

$$\Delta(\tau) = \Sigma(\tau)$$

System CM moves uniformly
 differences &
 sums of lengths
 oscillate.

Check if
 possible yes!