

# SESA6085 – Advanced Aerospace Engineering Management

Lecture 10

2024-2025





- In some cases, it is not possible to use a single failure distribution to model failure data
- This can occur if a component fails through a number of different mechanisms
- Semi-conductors, for example, are affected by humidity which can cause failures due to:
  - Large increases in current
  - Mechanical stresses due to volume expansion



- In such cases competing risk models can be used to model failure data
  - Also known as a compound model, series system model or multi-risk model
- Such models can only be used if:
  - The failure modes are completely independent
  - The component fails when the first mode is encountered
  - Each mode has it's own distribution



- Let's construct such a model
- We have a component with n independent failure modes
- For each mode we have a distribution function,  $F_i(t)$
- Failure of the component occurs when the component reaches a failed state for any mode
  - Does this sound familiar?
- Recall that this is identical to our definition of a series system



For a series system our probability of failure is given by:

$$P[X = 0] = 1 - P[X_1 = 1]P[X_2 = 1] \cdots P[X_n = 1]$$

 Hence the failure distribution of our competing risk model is given by:

$$F(t) = 1 - [1 - F_1(t)][1 - F_2(t)] \cdots [1 - F_n(t)]$$

$$F(t) = 1 - \prod_{i=1}^{n} R_i$$

Hence the reliability of our model is:

$$R(t) = \prod_{i=1}^{n} R_i$$



Take a model defined by two different failure distributions

$$R(t) = R_1(t)R_2(t)$$

The definition of a density function is:

$$f(t) = F'(t) = (1 - R(t))'$$

• Differentiating 1 - R(t) with respect to time, t gives:

$$f(t) = f_1(t)R_2(t) + f_2(t)R_1(t)$$



Expanding this out to n models we see:

$$f(t) = f_1(t)R_2(t)\cdots R_n(t) + \cdots + f_n(t)R_1(t)\cdots R_{n-1}(t)$$

Or alternatively:

$$f(t) = R(t) \left[ \sum_{i=1}^{n} \frac{f_i(t)}{R_i(t)} \right]$$

Which we can then use to find the parameters



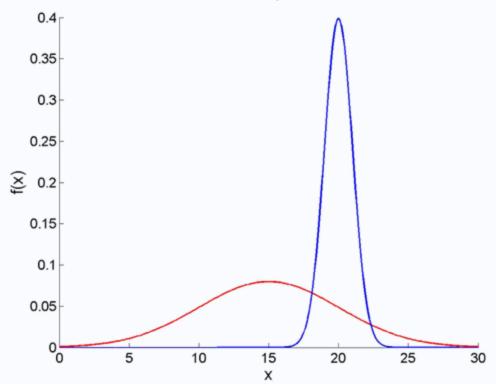
### Competing Risk – An Example

Consider the case where we have a two component system

$$-\mu_1 = 15.0 \& \sigma_1 = 5$$
  $\mu_2 = 20.0 \& \sigma_2 = 1$ 

$$\mu_2 = 20.0 \& \sigma_2 = 1$$

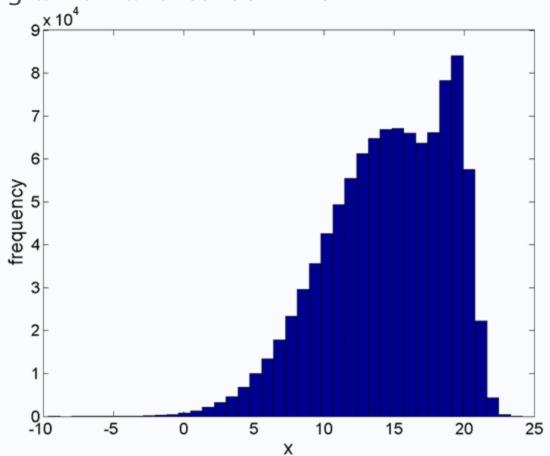
• The PDFs for the individual components look like this...





## Competing Risk - An Example

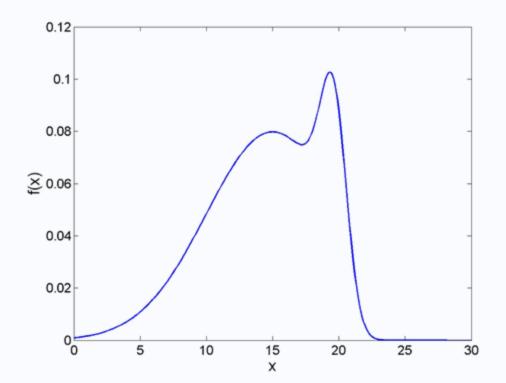
 If we run a Monte Carlo Analysis on the system what would the histogram of failures look like?





### Competing Risk – An Example

- We can define an analytical function of the PDF for our competing risk model using our known component data
- This could be derived using MLE but requires us to know the number of risks and nature of their distributions





## **Fault Tree Analysis**



### Fault Tree Analysis (FTA)

- FTA is a reliability (as well as safety) design analysis technique.
- It starts from consideration of system failure effects.
- The analysis proceeds by determining how these 'top events' can be caused by individual or combined lower level failures or events.



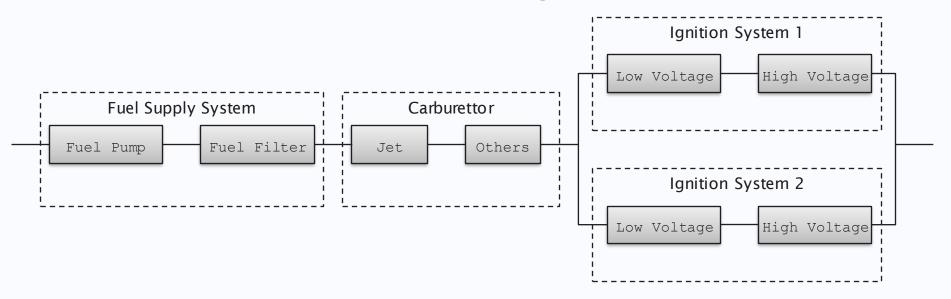
## Some FTA Symbols

Symbol	Description
	Basic Event: A basic fault event that requires no further development. It is s-independent of other events
111	And Gate: Failure (next higher event) will occur if all inputs fail (parallel redundancy)
<b>***</b>	Or Gate: Failure (next higher level) will occur if any input fails (series reliability)
	Combination Event: An event that results from the combination of basic events through the input logic gates
	Undeveloped Event: It is s-dependent upon lower events, but not developed downwards



#### RBD vs FTA

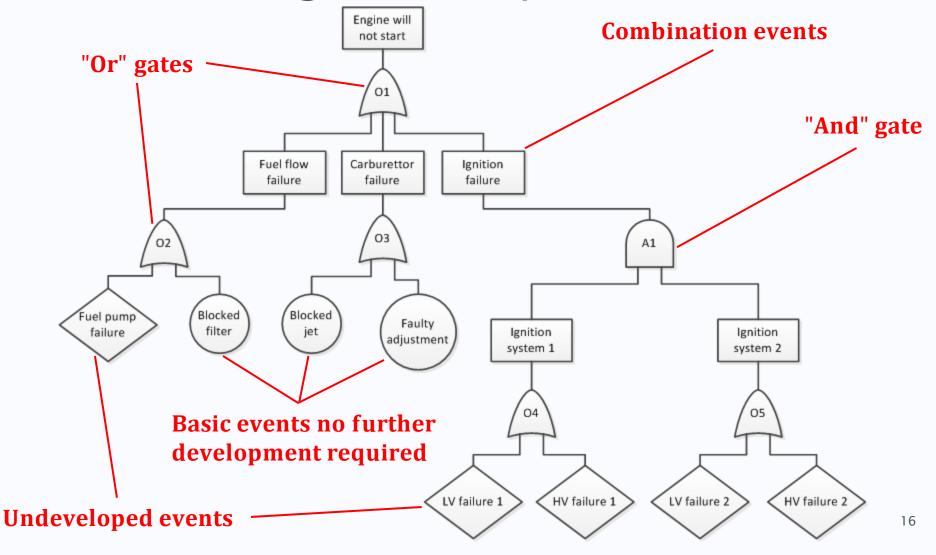
• Consider the RBD of an aircraft engine...



How can we build an FTA for engine failing to start?



### FTA for an Engine (Incomplete)



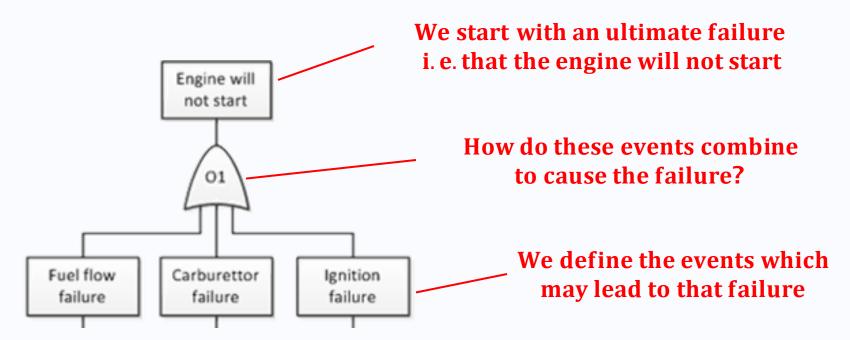


#### What is the Difference?

- RBD: We build the model by concentrating on the success of the system
- FTA: We build the hierarchy in terms of failures
- While an RBD considers the entire system the FTA may disregard parts of the system not contributing to a fault
  - RBD of an aircraft includes engine, wing, gear etc.
  - FTA for an ignition failure may not include anything outside of the engine sub-system and perhaps only some engine components



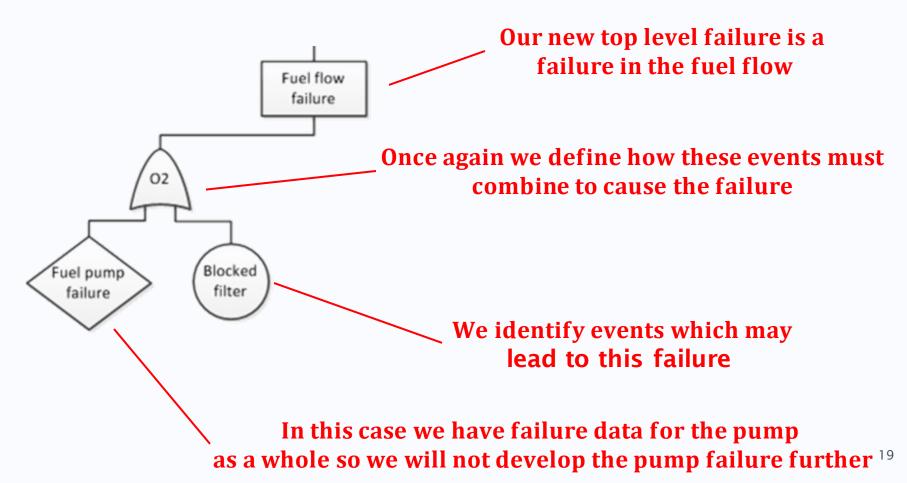
### Building a FTA for an Engine





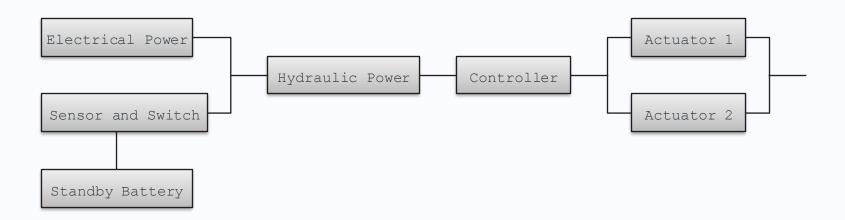
### Building a FTA for an Engine

The process is then repeated for each sub-failure...



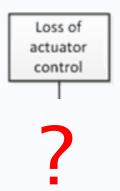


- A control system consists of an electrical power supply, a standby battery supply which is activated by a sensor and switch if the main supply fails, a hydraulic power pack, a controller, and two actuators acting in parallel
- Lets consider the RBD for this system:





- Lets construct a FTA diagram for this system
- Consider the ultimate failure mode as the "loss of actuator control"

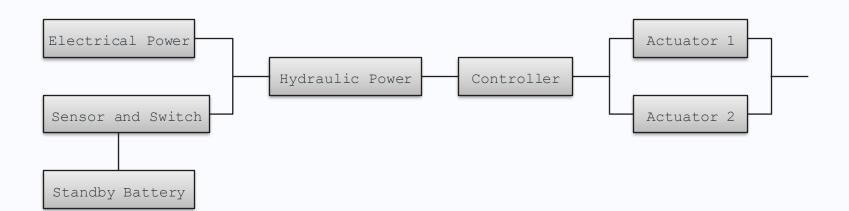




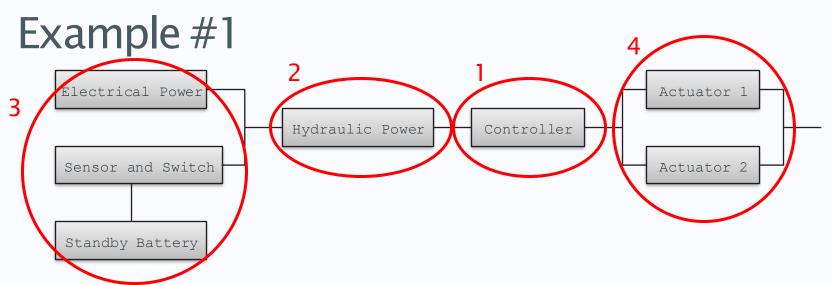
What comes next?



- What events will cause the actuator to fail?
- Note: Actuator control loss is not the same as failure of only an actuator



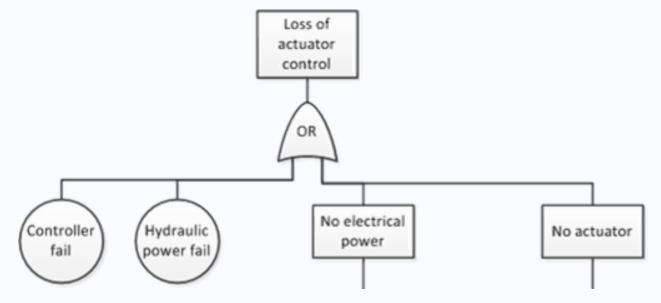




- Actuator control could be lost due to:
  - 1. Controller failure
  - 2. Hydraulic power failure
  - 3. Electrical power failure
  - 4. Actuator failure
- How should these events be combined to cause a failure?



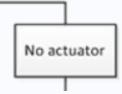
Which gives use the following FTA diagram



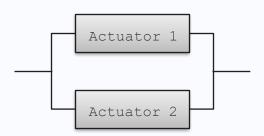
What's next?



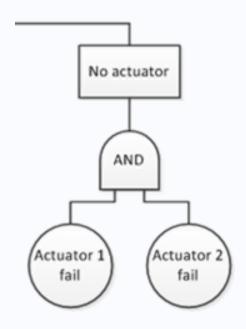
Let's consider the actuator failure...



- What events will cause this failure?
  - Failure of actuator #1
  - 2. Failure of actuator #2
- How do these events combine?

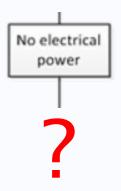


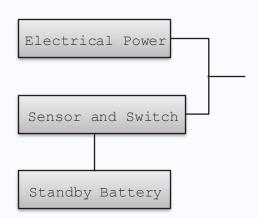




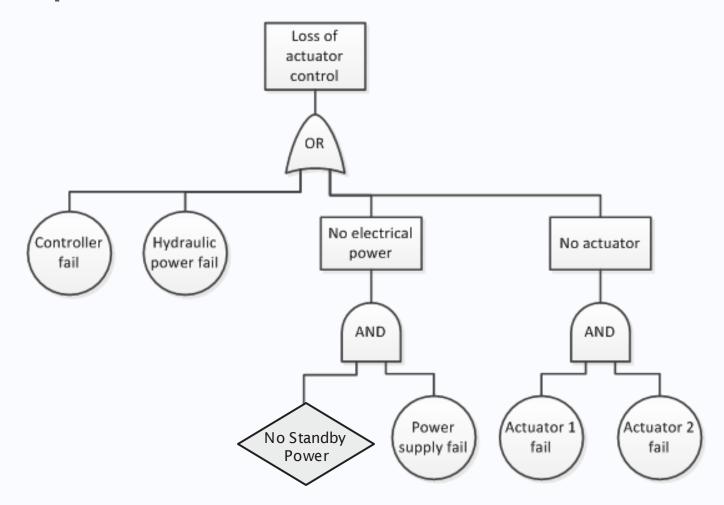


 Anythoughts on the expansion of the failure of the electrical power?









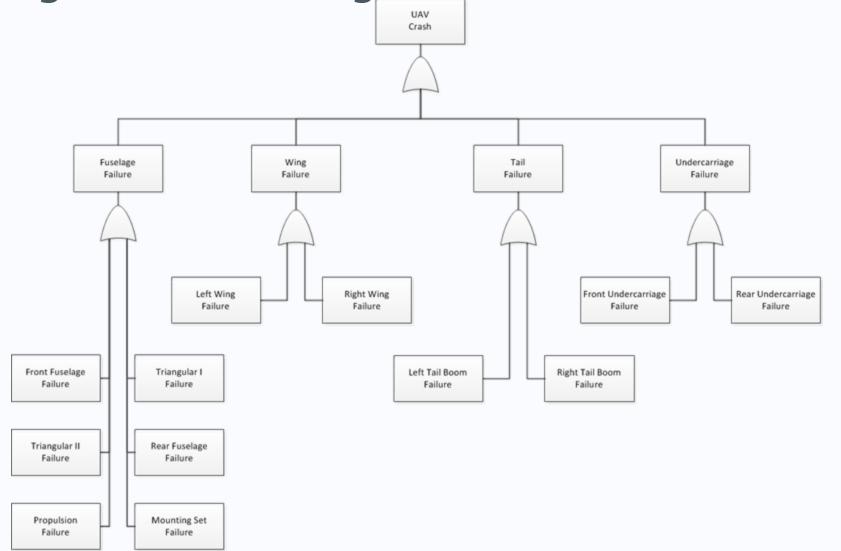


### What About for a UAV?



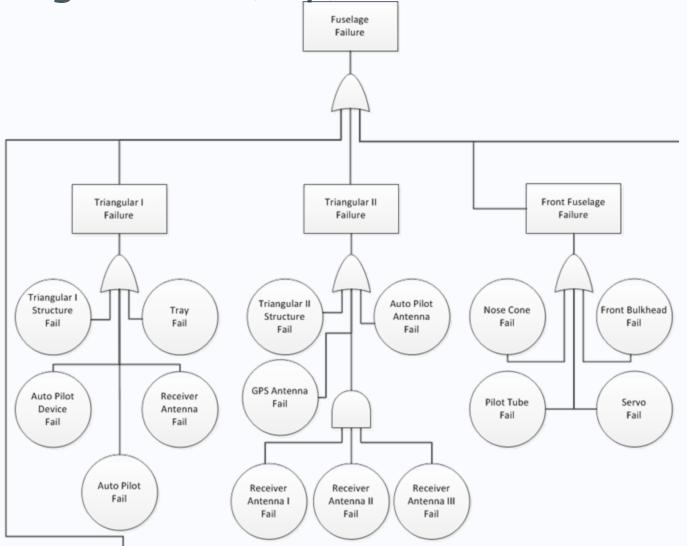


High Level FTA Diagram of a UAV





Fuselage Failure (Top)





Fuselage Failure (Bottom) Rear Fuselage Failure Mounting Set Failure Fuel Filler Fuel Filter Fuel Tube Fail Fail Fail Rear Fuselage Structure Fail Inner Spar Fail Propulsion **Power Switch** Failure Spectrum Servo Clunk Fail Fail Fail Carbon Stringer Carbon Stringer Carbon Stringer Fail Fail Fail

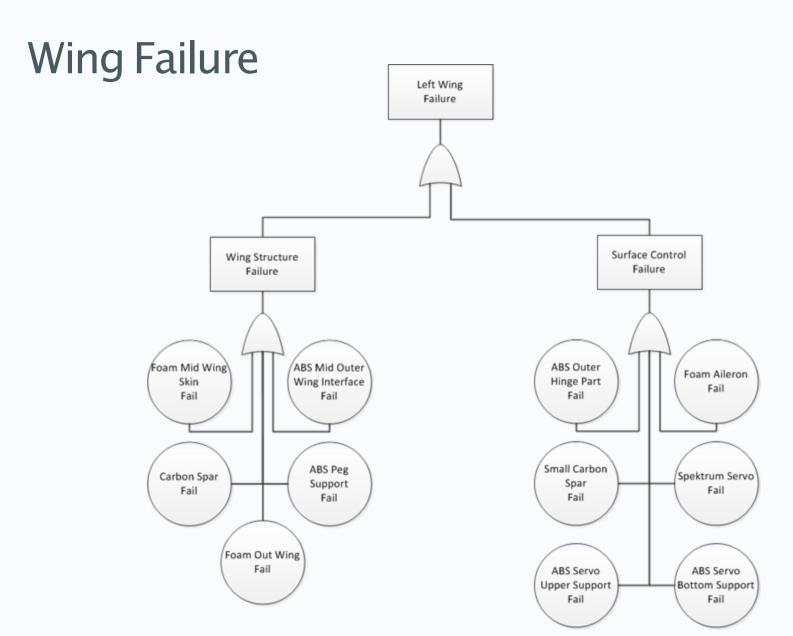
Propeller Blade

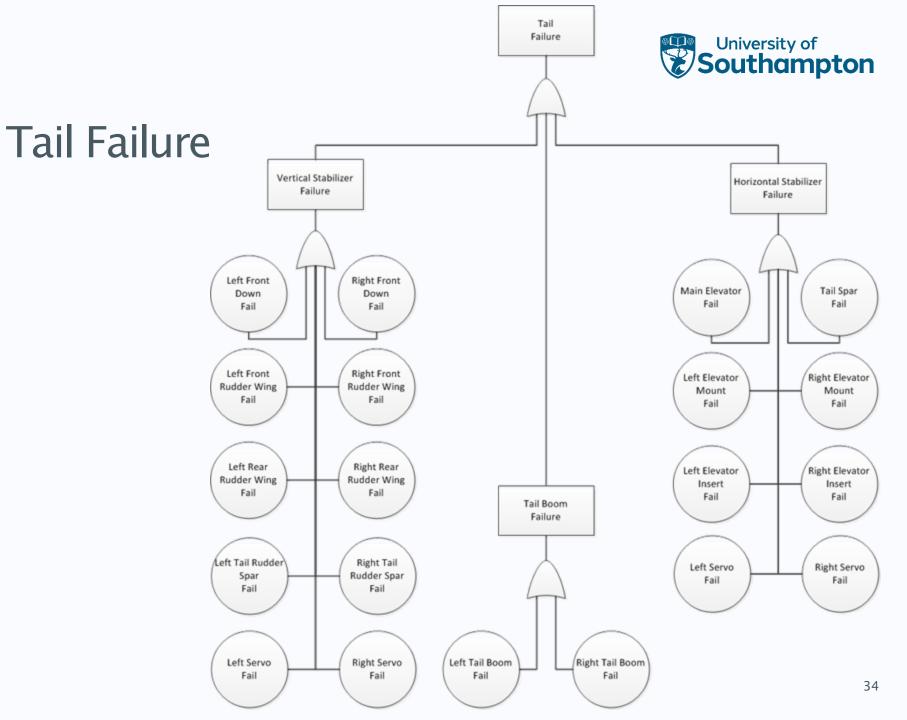
Fail

IC Engine

Fail









## Performing a Fault Tree Analysis

- As well as illustrating logical connections between failure events FTA can be used to quantify event probabilities
- Failure probabilities can be assigned to the failure events and used to calculate top level failure probabilities

$$- P(A \text{ and } B) = P(A)P(B)$$



- P(A or B) = P(A) + P(B) - P(A)P(B)





### Performing a Fault Tree Analysis

- A different FTA may need to be constructed for different top level events as the logical connections may change
- Although FTA's from sub-failures may be repeatedly used in multiple analyses e.g.
  - The FTA for our "no electrical power branch" may be used in multiple FTAs if the same power supply provides power to multiple components
- FTAs can be created and solved using specialist software e.g.
  Relex

