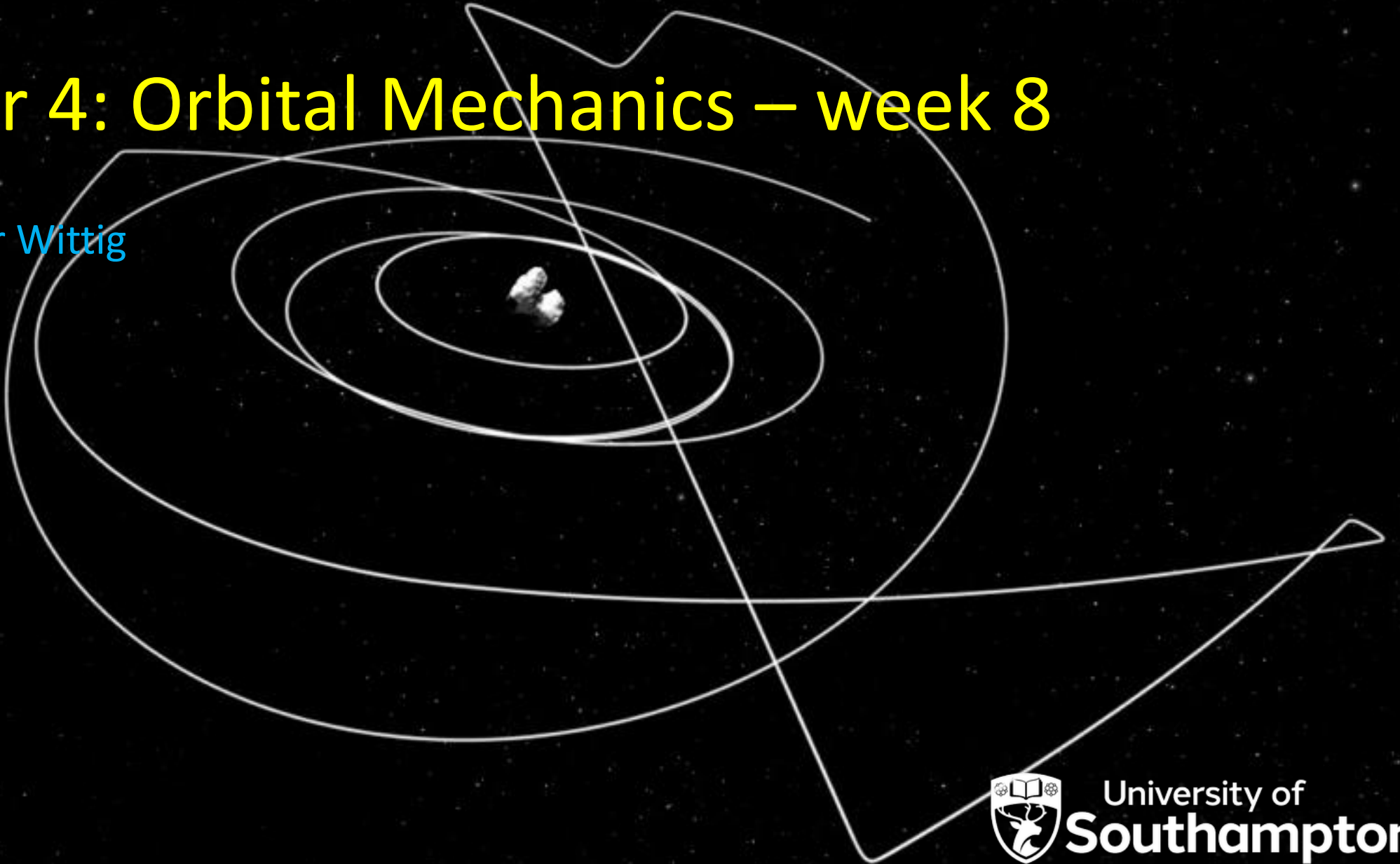


Advanced Astronautics (SESA3039)

Chapter 4: Orbital Mechanics – week 8

Dr. Alexander Wittig



- F2F lecture: Content of the week and “big picture”

- Monday 13:00 in 35/1001
Tuesday 15:00 in 35/1001
- recorded, uploaded to Panopto
- Slides available on BB
- Questions welcome
 - You may want to look at the in-depth videos beforehand

- Tutorial sessions

- Tuesdays 16:00 in 35/1001
- Q&A on Problem Sheets
- Tutors on hand for questions

- Pre-recorded Material

- Filling in details from lectures
- More in depth coverage of topics
- **fully examinable material**
- Videos and lecture notes on BB

- Quiz: 16/12

- similar to tutorial sheet questions
- numerical answers
- same rules as other quizzes

Problem Sheet

- Covers full 4 weeks of material
- Attempt questions on material of each week
- Additional questions available in textbooks

Tutorial

- Tuesdays 16:00 in 35/1001
- Suggested in groups, self-organized
- **Active learning** session: prepare by attempting problem sheet
- Demonstrators available to help with questions
- 1 problem solved together (let me know preference 1 day before)

■ Week 8: Orbital Motion

- Math Basics
- Spherical Trigonometry
- Keplerian Motion from First Principles

■ Week 10: Orbit Representation

- Coordinates
- Dates & Times
- Orbital Elements

■ Week 9: Orbit Properties

- Constants of Motion
- Eccentricity Vector
- Conic Sections

■ Week 11: Time Dependence

- Eccentric Anomaly
- Hyperbolic Anomaly
- Kepler's Equations

Orbital Mechanics for Engineering Students

Curtis, H.

Butterworth-Heinemann, 2013.
ISBN-13 978-0123747785

Orbital Mechanics

Chobotov, V.

AIAA Education Series, 2002
ISBN-13 978-1563475375

Fundamentals of Astrodynamics and Applications

Vallado, D. A.

Springer, 2007
ISBN-13 978-0387718316

An Introduction to the Mathematics and Methods of Astrodynamics

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Satellite Orbits: Models, Methods and Applications

Montenbruck, O. and Gill, E.

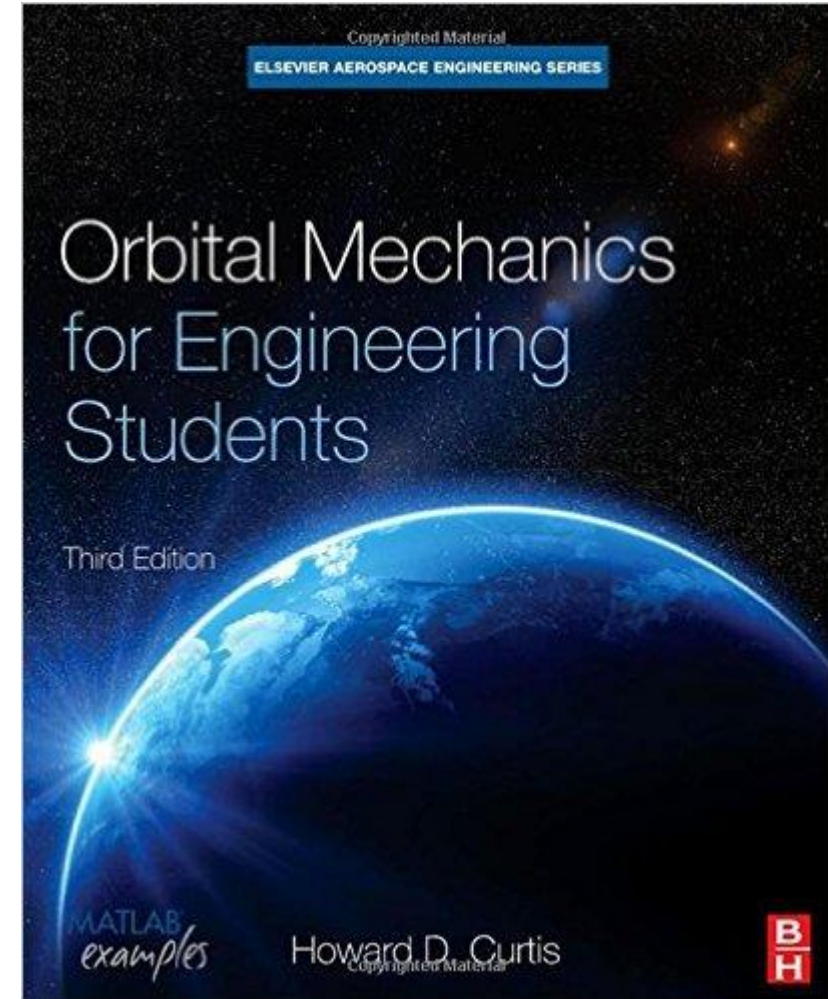
Springer, 2011
ISBN-13 978-3540672807

Subjective observations:

- good introductory textbook
- very explicit mathematical derivations
- copious amount of worked examples
- not as complete as others
- not as useful as a reference

Chapter 4 will follow this textbook closely.

**Available as free full text with
Soton Login from publisher!**



<https://www.sciencedirect.com/book/9780080977478/orbital-mechanics-for-engineering-students>

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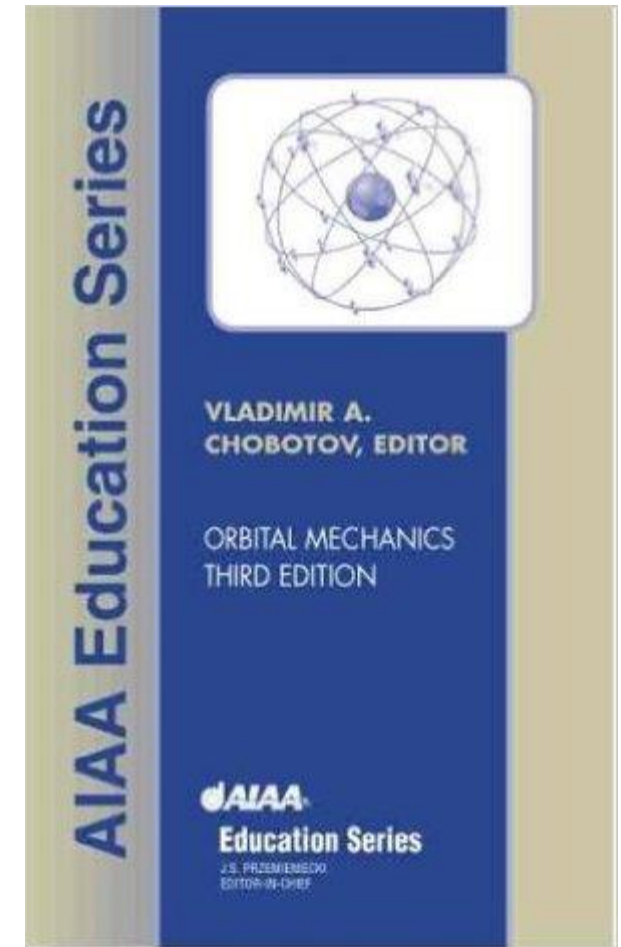
Springer, 2011
ISBN-13 978-3540672807

Subjective observations:

- wide range of topics covered
- mathematical derivations often challenging
- written as an expert text, not an introductory text
- includes references to research papers

eBook available at:

<https://ebookcentral.proquest.com/lib/soton-ebooks/detail.action?docID=3111521>



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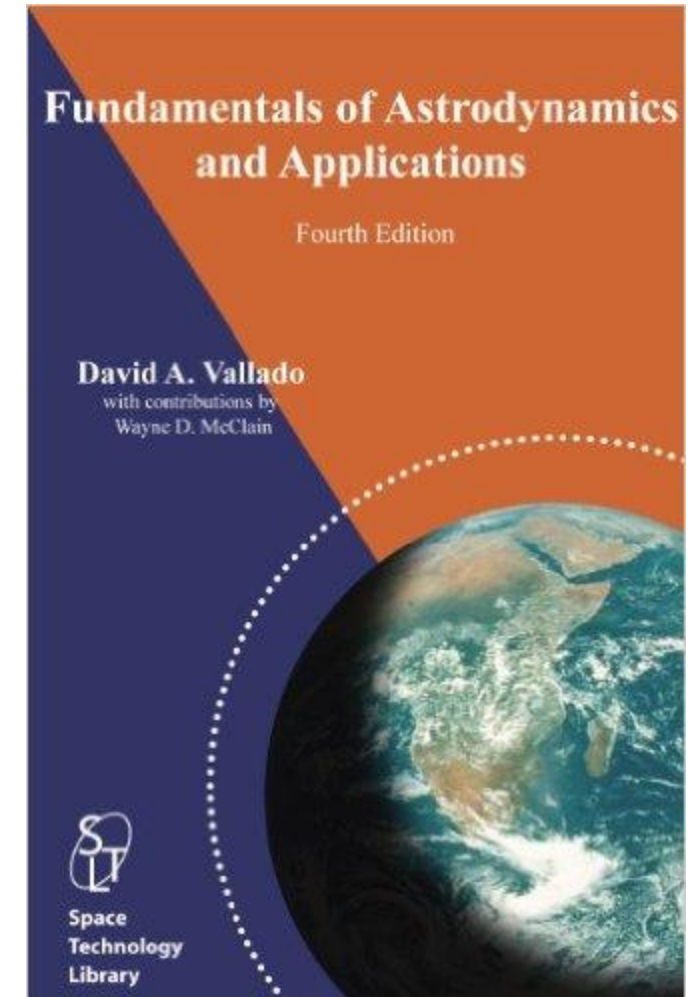
Montenbruck, O. and Gill, E.

Springer, 2011
ISBN-13 978-3540672807

Subjective observations:

- good and quite readable, complete textbook
- understandable mathematical derivations
- focus on numerical implementation of algorithms (including code in various languages)
- also useful as a reference

Author worked on SGP4, the US airforce standard orbit propagator for NORAD TLEs.



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Springer, 2011
ISBN-13 978-3540672807

Subjective observations:

- most complete reference on the subject
- sometimes challenging mathematical derivations
- written by a mathematician!
- includes many exercises
- extremely useful as a reference

Author wrote the Apollo mission guidance computer code.

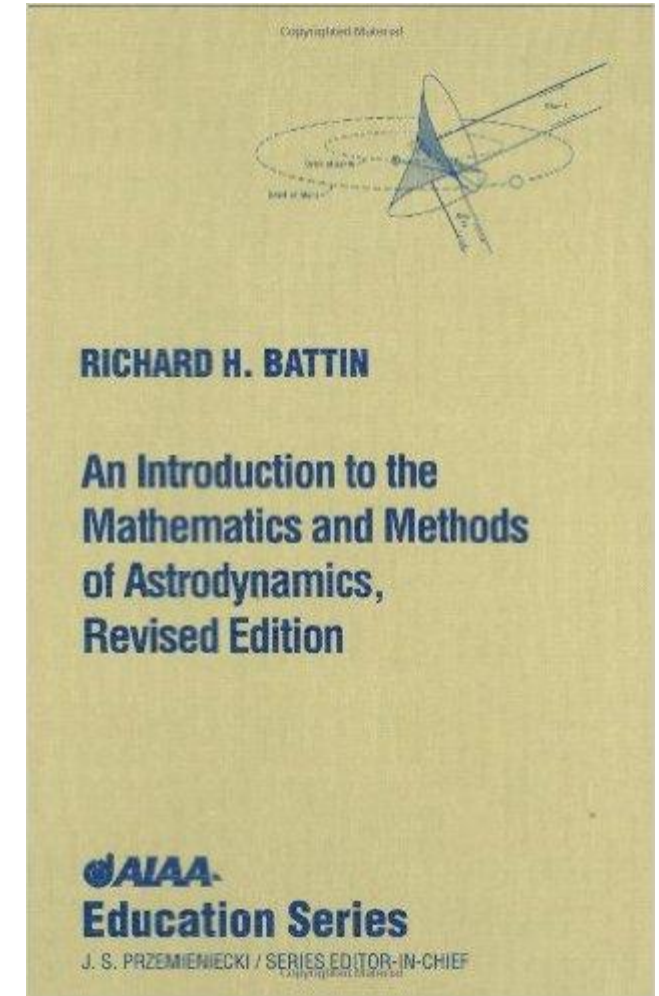
eBook available at:

<https://ebookcentral.proquest.com/lib/soton-ebooks/detail.action?docID=3111476>

Check out his MIT lecture

“A Funny Thing Happened on the Way to the Moon” for some history of the early NASA program on YT:

<http://y2u.be/ieiEoTo8-XY>



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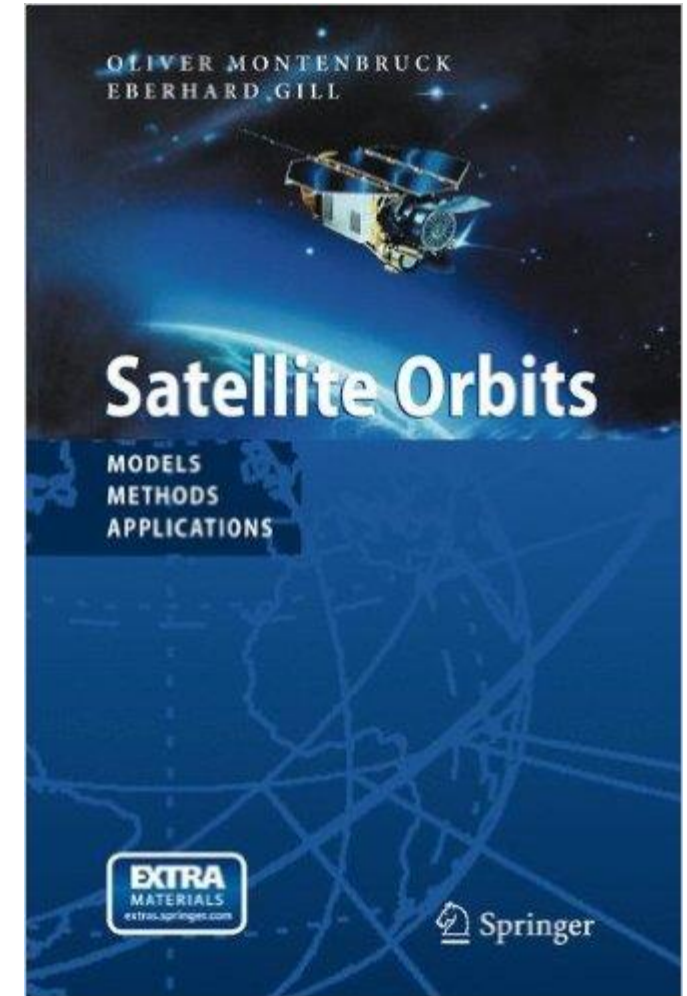
Satellite Orbits: Models, Methods and Applications

Montenbruck, O. and Gill, E.

Springer, 2011
ISBN-13 978-3540672807

Subjective observations:

- focused specifically on satellites
- quite readable, few long derivations
- includes numerical and practical engineering considerations
- good textbook and reference on its limited subset of the subject



Mathematical Concepts

- Recall basic concepts of **vector algebra**

- Vector identities
- Coordinate representation

→ ensure you understand
vector manipulation on
your model of calculator

- Relate **Kinematics** and **Dynamics** in different frames

- Accelerating reference frames
- Equations of motion in accelerating reference frames

- Apply trigonometry to problems on the sphere

- Trigonometric identities on the sphere

(watch recorded
material)

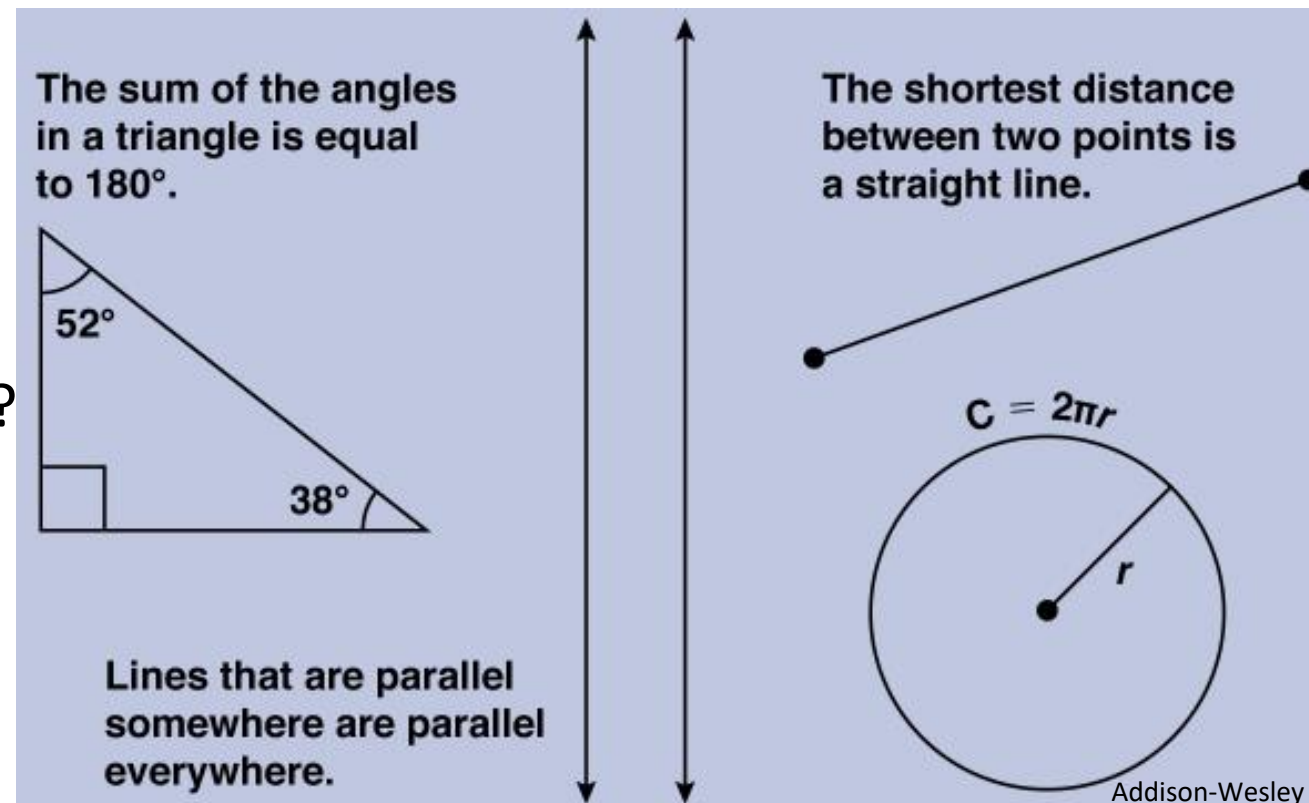
Vector Algebra & Mechanics

see Chapter 4 Video Lecture 1

Spherical Trigonometry

see also Chapter 4 Video Lecture 2

- Euclidean geometry is in flat space
- In curved space Euclidean properties generally **do not** hold
- Sphere is a simple curved space relevant in orbital mechanics and astronomy
- Line segment: shortest path between two points
 - Is it unique?
 - How often can different lines intersect?
 - How often do parallel lines intersect
 - How many lines are needed for a polygon?



Geometry in curved space

here we define "lines" as great circle arcs

- Euclidean geometry is in flat space
- In curved space Euclidean properties generally **do not** hold
- Sphere is a simple curved space relevant in orbital mechanics and astronomy

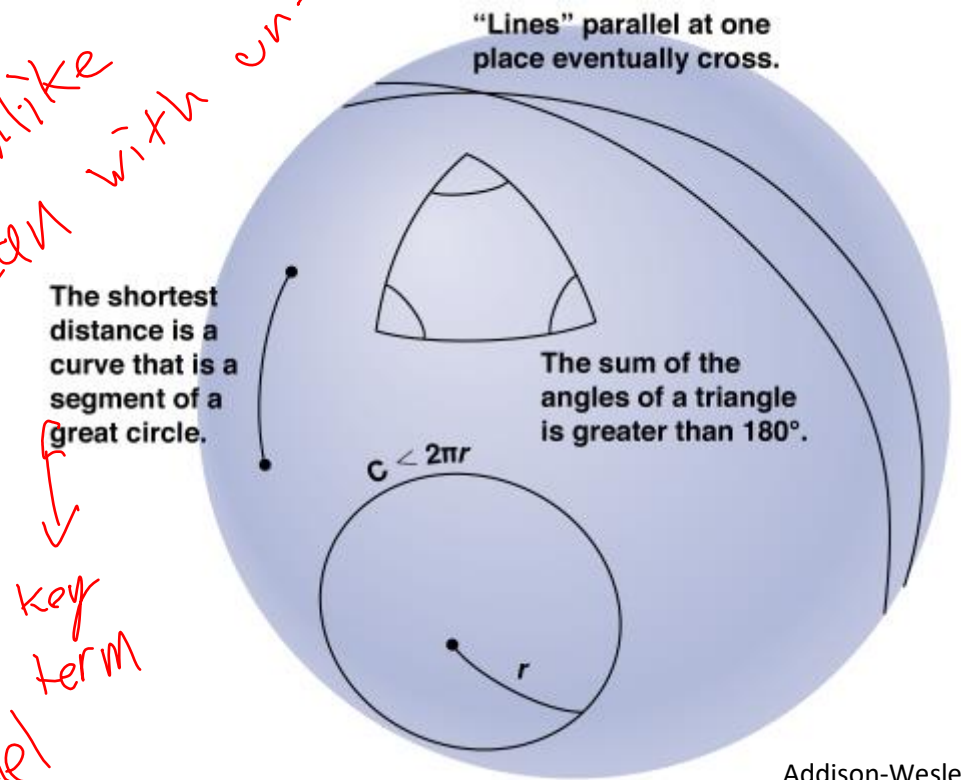
■ Line segment: shortest path between two points

- Is it unique?
- How often can different lines intersect?
- How often do parallel lines intersect
- How many lines are needed for a polygon?

↓
2

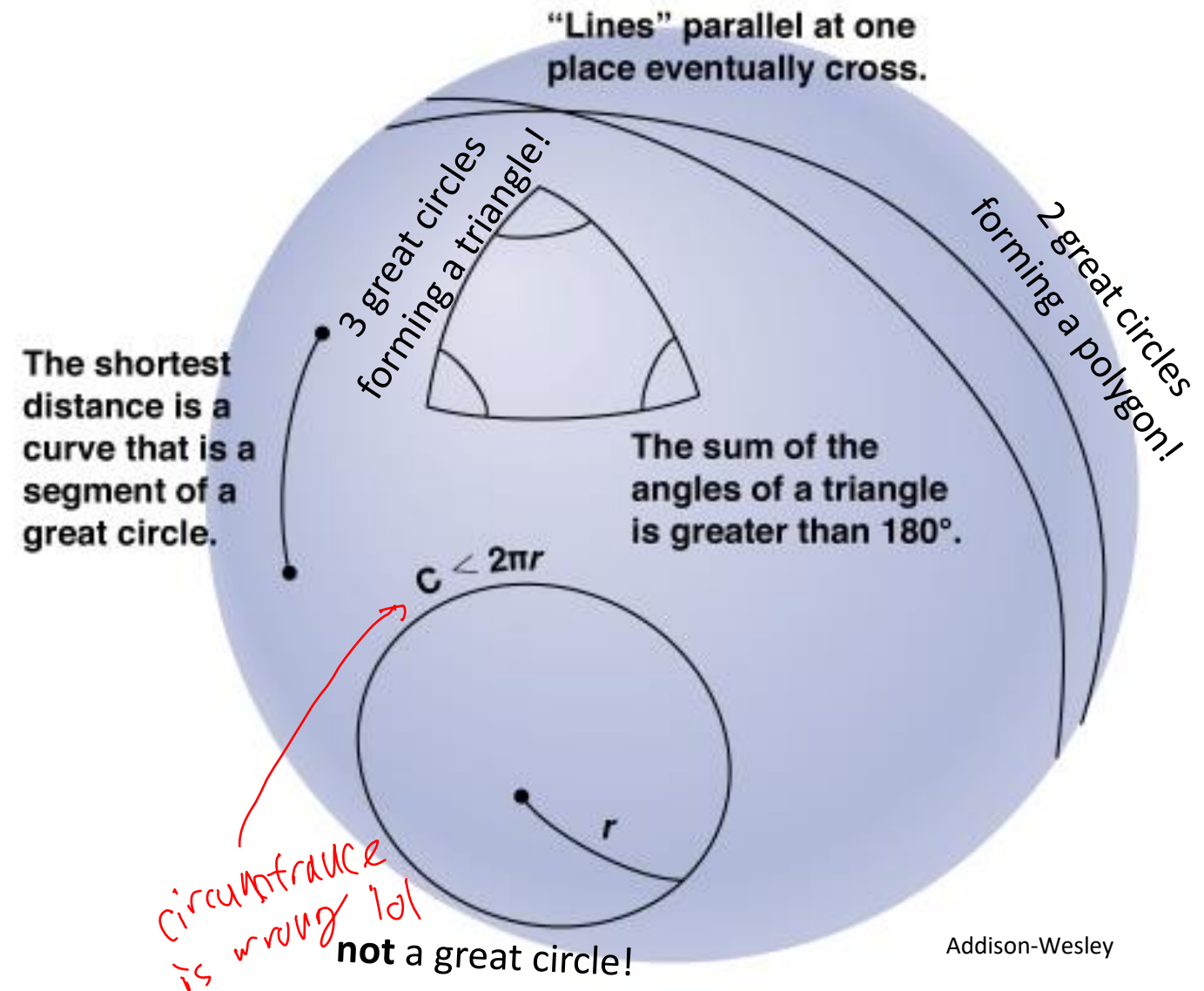
lines can be locally parallel

twice, unlike Euclidean with only

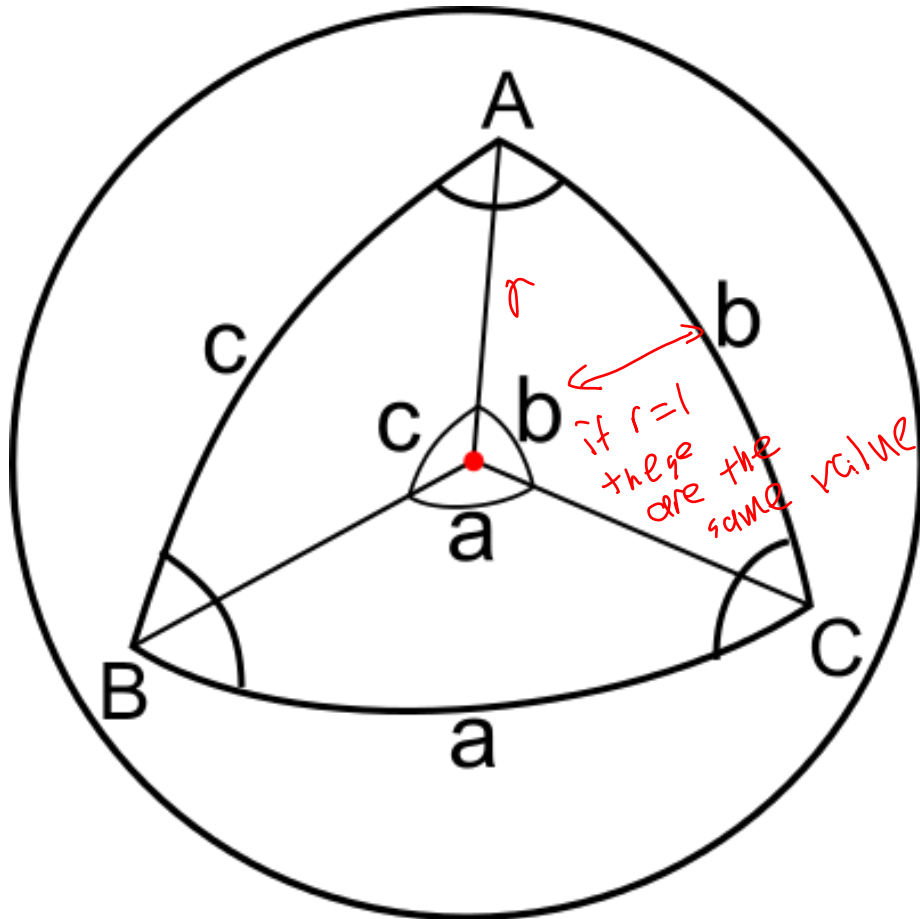


Addison-Wesley

- **Great circle:** circle on a plane including the center point of the sphere
- **Line:** segment of a **great circle**
- **Polygon:** closed loop of line segments on sphere surface



Addison-Wesley

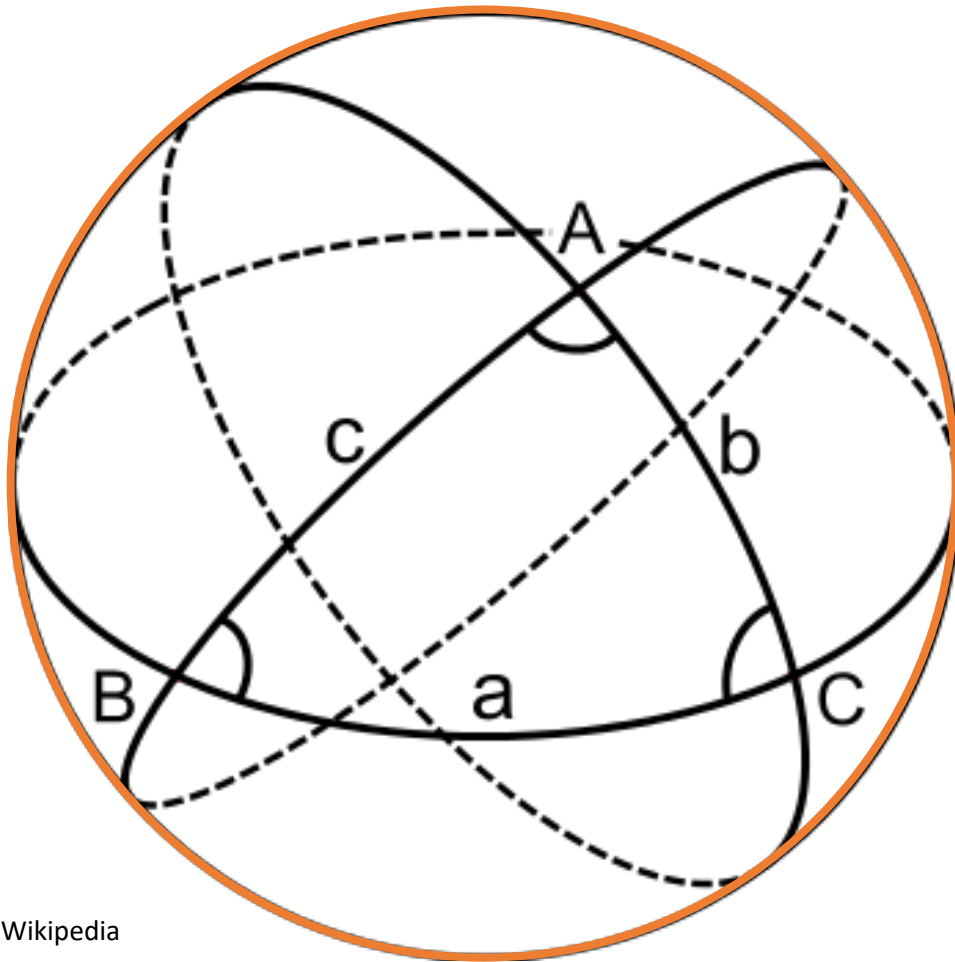


Wikipedia

Notation:

- Vertices and angles (in rad) at vertices denoted by capitals: A, B, C
- Angles are between tangent vectors to great circles, or equivalently between planes containing the great circles
- Sides and the angles (in rad) at sphere center denoted by lower case a, b, c
- Assuming a unit sphere (!) a, b, c are also lengths of sides (chords)

Three distinct great circles form a triangle



Wikipedia

Given angle A and side lengths b, c what is the missing length a?

- Planar case: law of cosines:

$$c^2 = a^2 + b^2 - 2ab \cos(\gamma)$$

- But: planar trigonometry not true on sphere!

Question

How many spherical triangles are there in this picture?

Cosine Rules

$$\cos a = \cos b \cos c + \sin b \sin c \cos A$$

$$\cos b = \cos c \cos a + \sin c \sin a \cos B$$

$$\cos c = \cos a \cos b + \sin a \sin b \cos C$$

Complementary Cosine Rules

$$\cos A = -\cos B \cos C + \sin B \sin C \cos a,$$

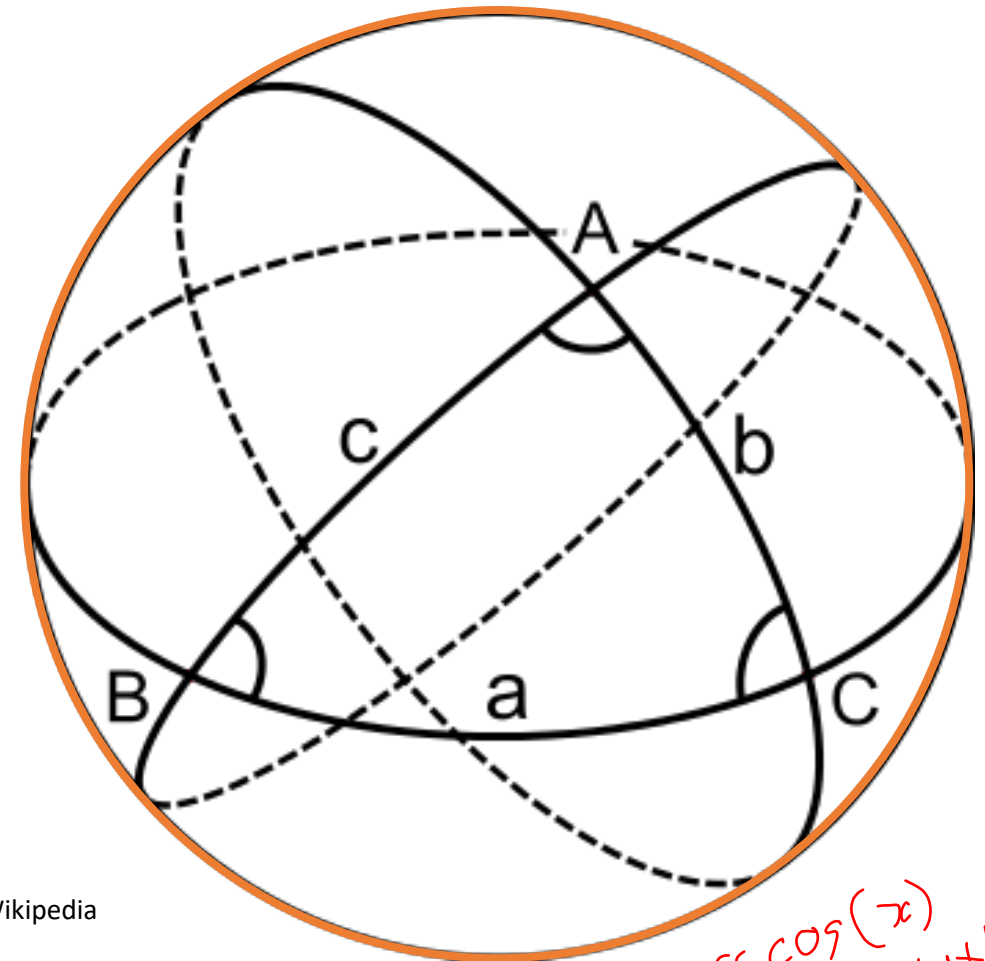
$$\cos B = -\cos C \cos A + \sin C \sin A \cos b,$$

$$\cos C = -\cos A \cos B + \sin A \sin B \cos c.$$

Sine Rules

$$\frac{\sin A}{\sin a} = \frac{\sin B}{\sin b} = \frac{\sin C}{\sin c}$$

CAREFUL WITH QUADRANT DISAMBIGUATION!



Wikipedia

*arc cos(x)
has 2 solutions*

Distances on a sphere

One application: distances on a sphere (e.g. Earth):

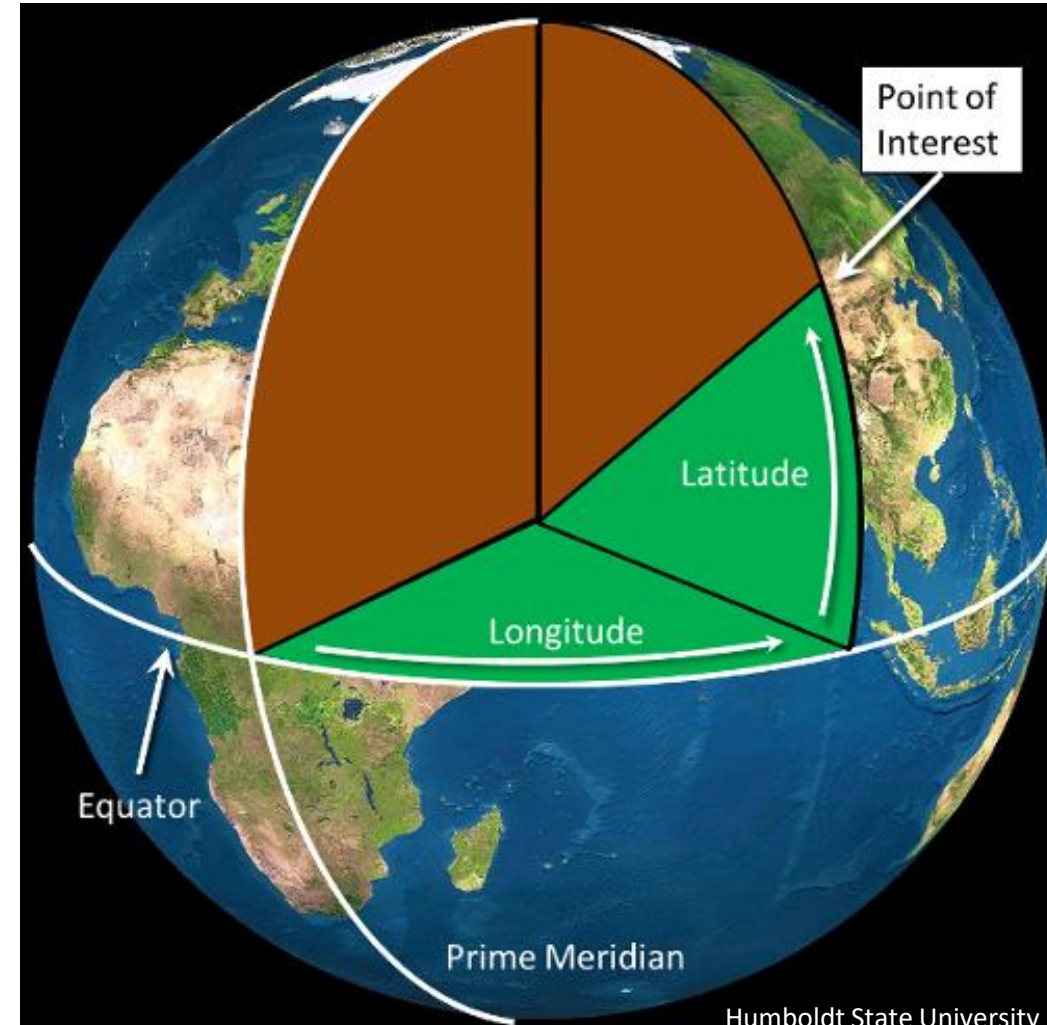
- Points located by latitude and longitude
- Prime meridian = 0 longitude
- Equator = 0 latitude

Question: given two points on Earth, what's the shortest distance between them?

$$\Phi_1, \lambda_1$$

$$\Phi_2, \lambda_2$$

$$d = ???$$



Distances on a sphere

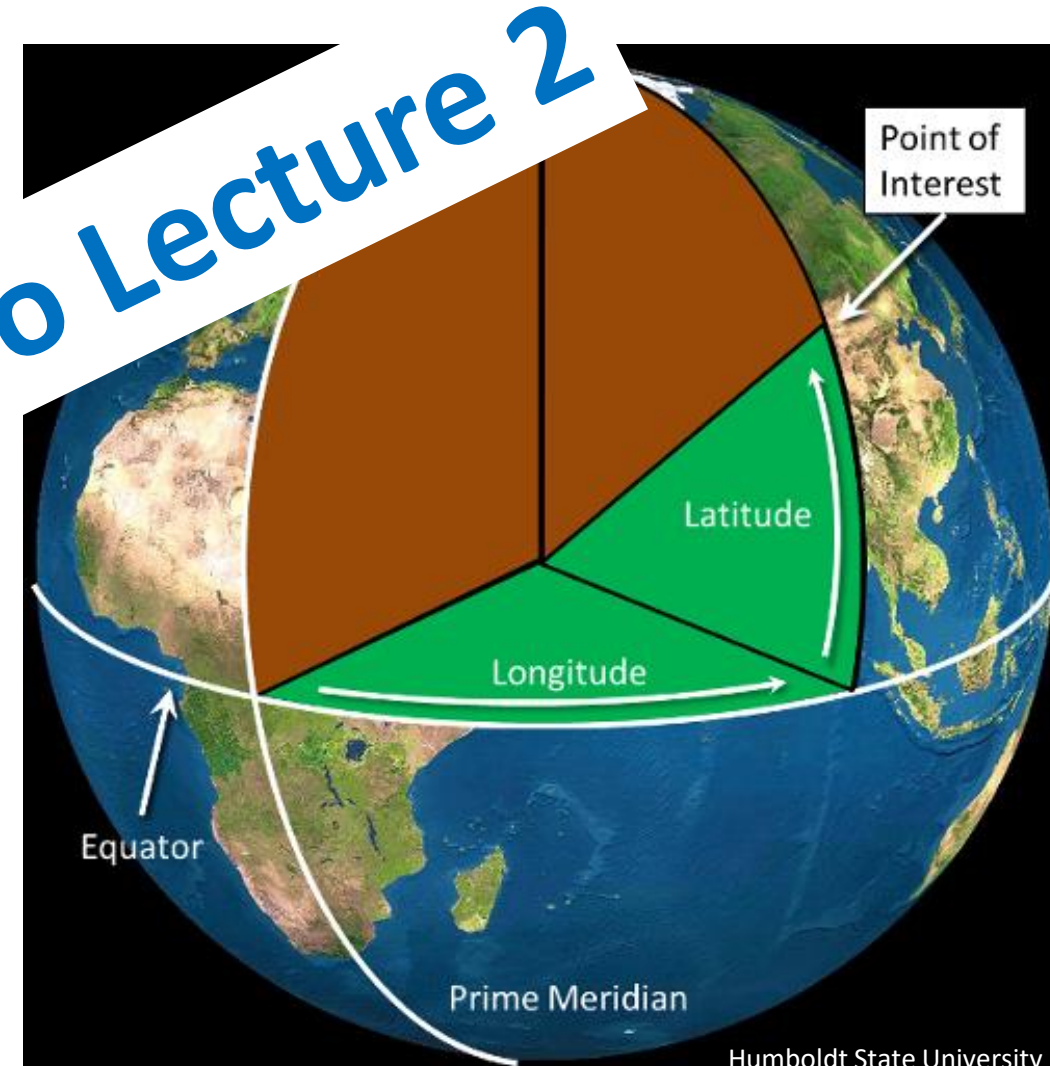
One application: distances on a sphere (e.g. Earth):

- Points located by latitude and longitude
- Prime meridian = 0 longitude
- Equator = 0 latitude

Question: given two points ϕ_1, λ_1 and ϕ_2, λ_2
the shortest distance d

see Chapter 4 Video Lecture 2

$$d = ???$$



- Cross products, vector identities, triple product
- Dynamics, Newton's law, inertial frame, accelerating frame, rotating frame, fictitious forces
- Ordinary Differential Equations, initial value problem, initial conditions
- trigonometry, law of sines, law of cosines, spherical trigonometry, distance on a sphere, latitude & longitude

Further reading

- Curtis, Chapter 1

Also:

- Chobotov, Chapter 1
- Batin, Chapter 2

Orbits from First Principles

- State solution of bound orbital motion of light satellite around a heavy primary
- Derive the correct solution for the unperturbed two body problem in inertial and non-inertial reference frames
- Quantify the accuracy of restricted 2BP
- Discuss the notion that the Earth orbits around the Sun

Previous lectures (Astronautics, Physics)

- Equation of motion:

$$\ddot{\vec{x}} = -\frac{GM}{|\vec{x}|^3}\vec{x} = -\frac{\mu}{|\vec{x}|^3}\vec{x}$$

\vec{x} = vector to centre of body

central force problem

vector form of $y = \frac{\mu}{R^2}$

- Shape of solution:

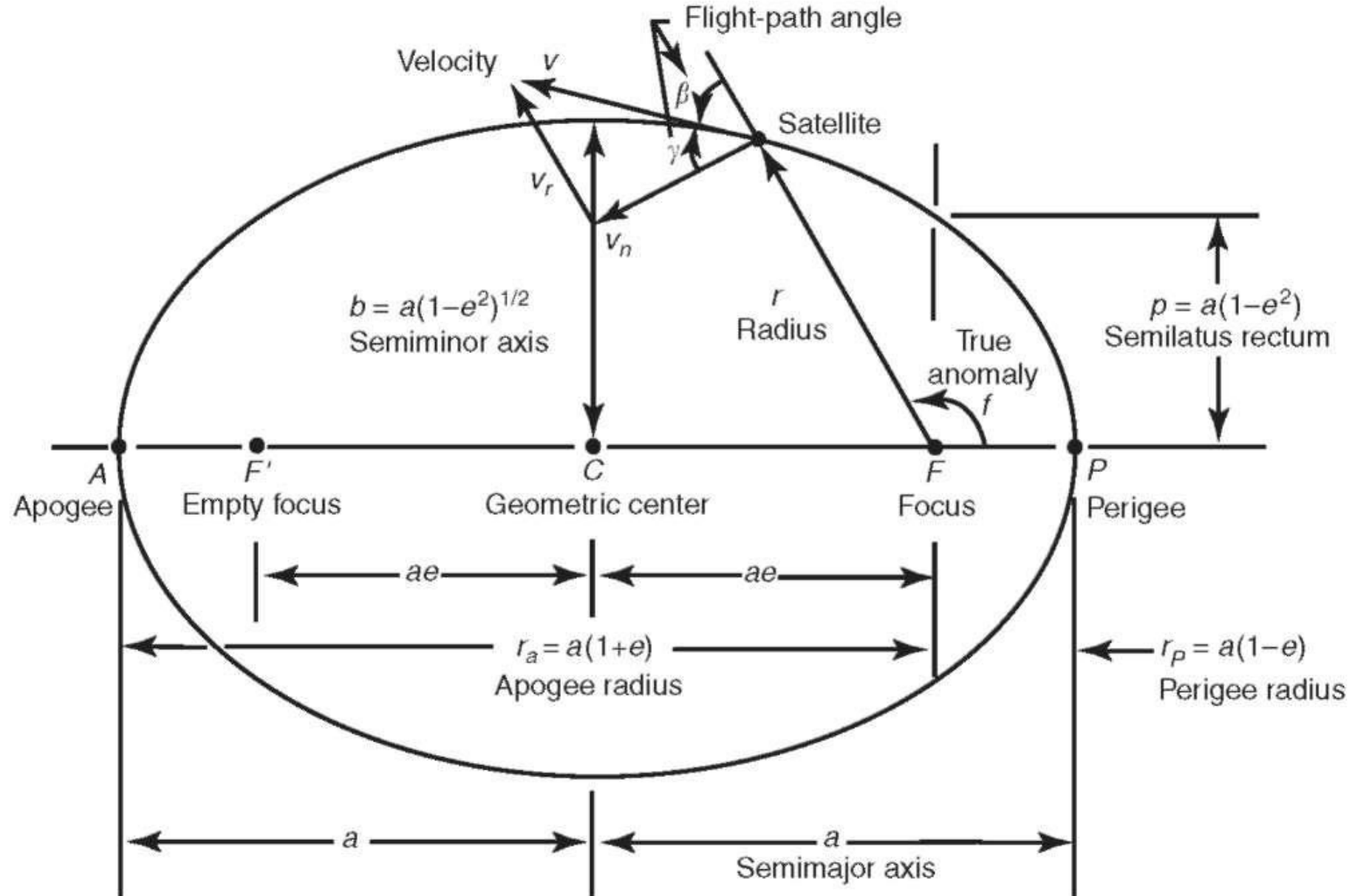
$$r(\theta) = \frac{p}{1 + e \cos(\theta)}$$

conic section



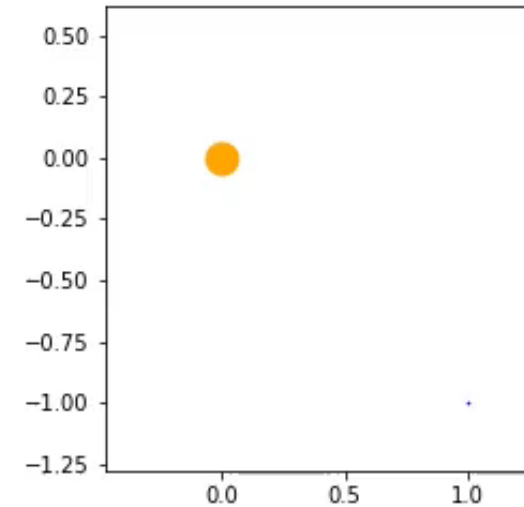
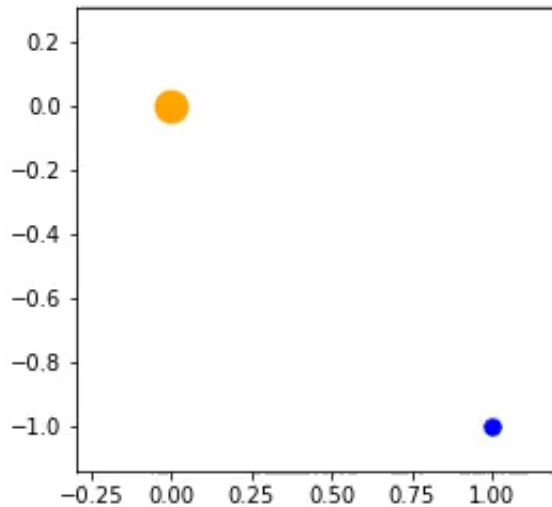
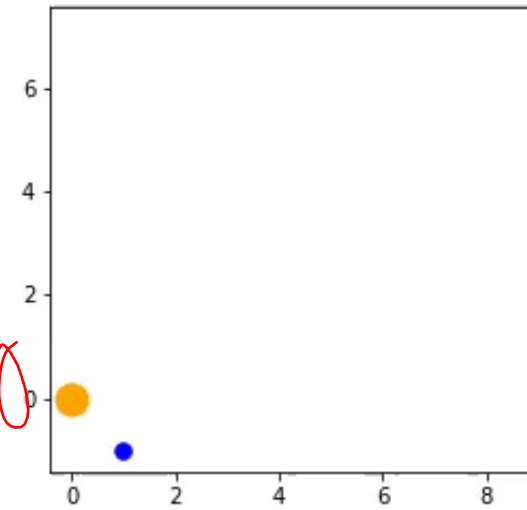
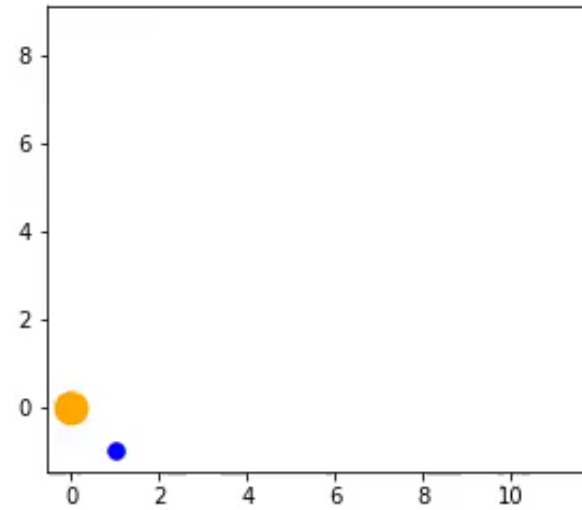
Unknown artist (around 1610)

Geometry of bounded 2BP motion



Which one is 2BP motion?

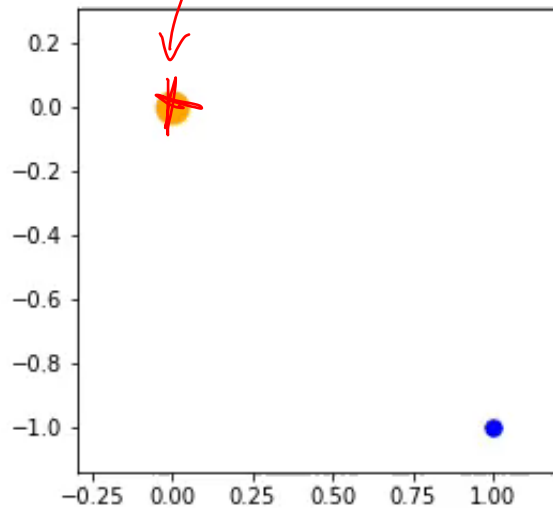
Radius of objects
proportional to mass



*(these are 9th
showing 2 body
problems)*

Reference frames

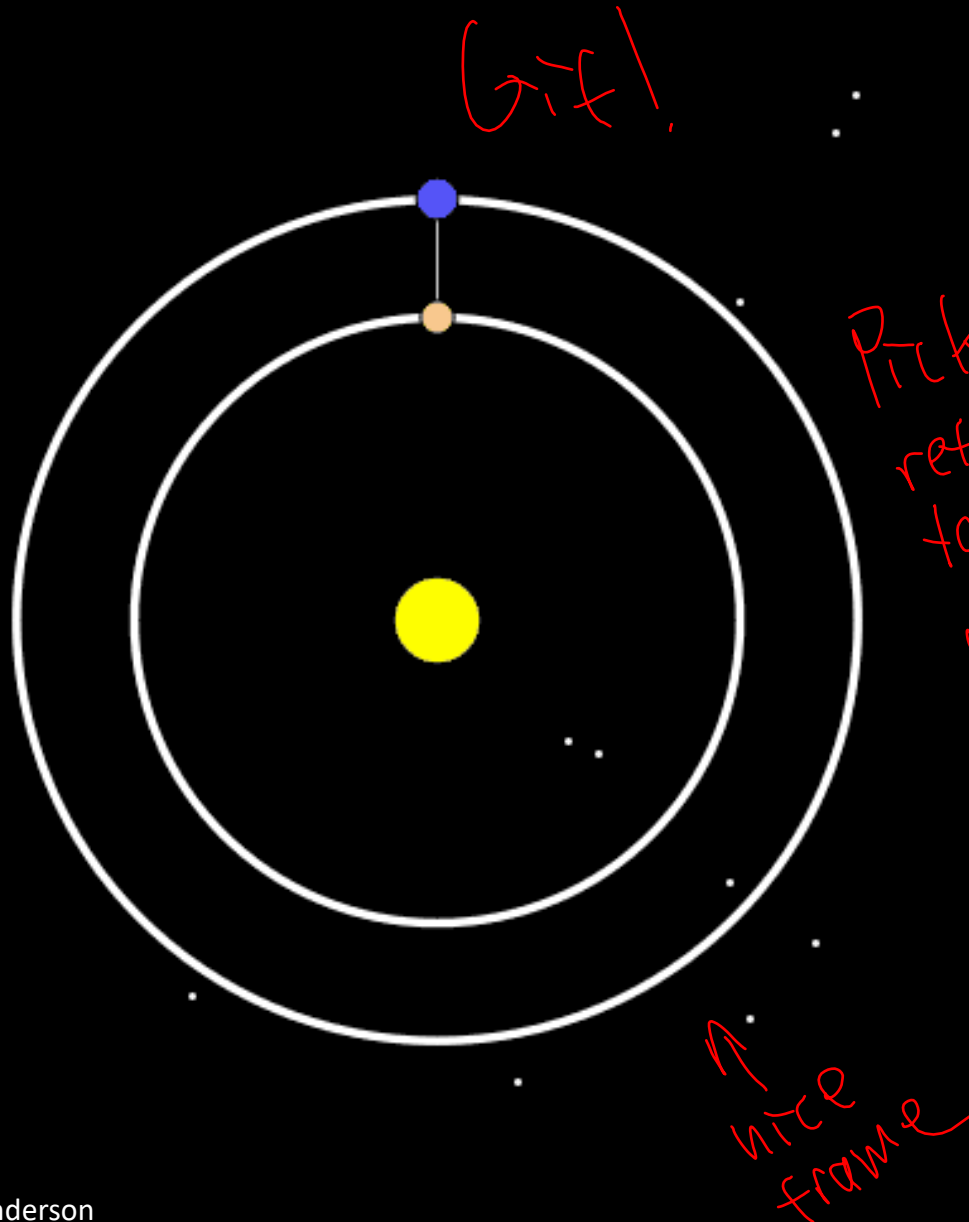
- All of these were 2BP motion!
- Motion can be viewed from different reference frames
- **Bonus question:** which ones were inertial frames?



The only
non-inertial frame!



Reference frames

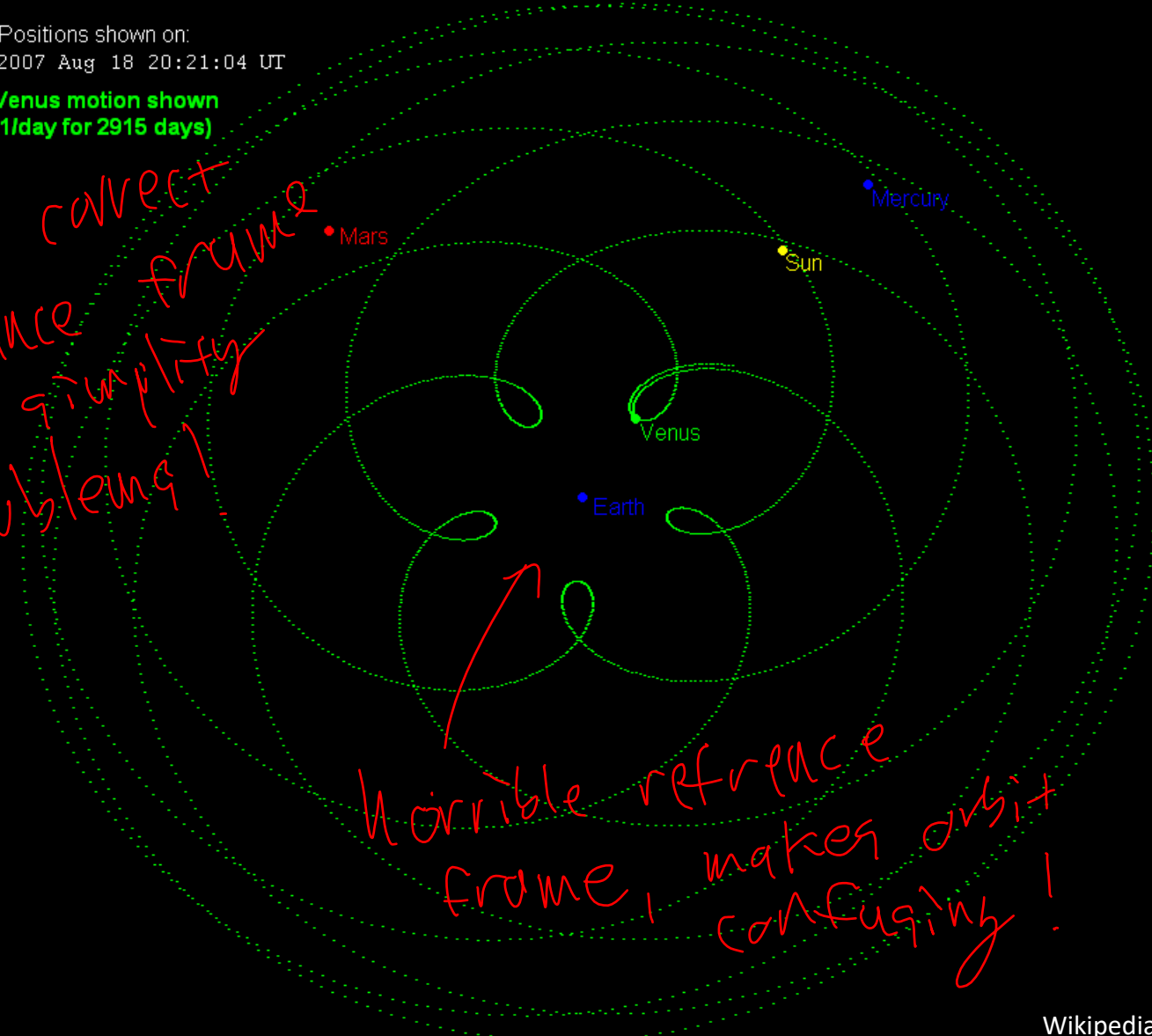


Five inferior conjunctions of Venus repeat in a precessing pentagram

Positions shown on:
2007 Aug 18 20:21:04 UT

Venus motion shown
(1/day for 2915 days)

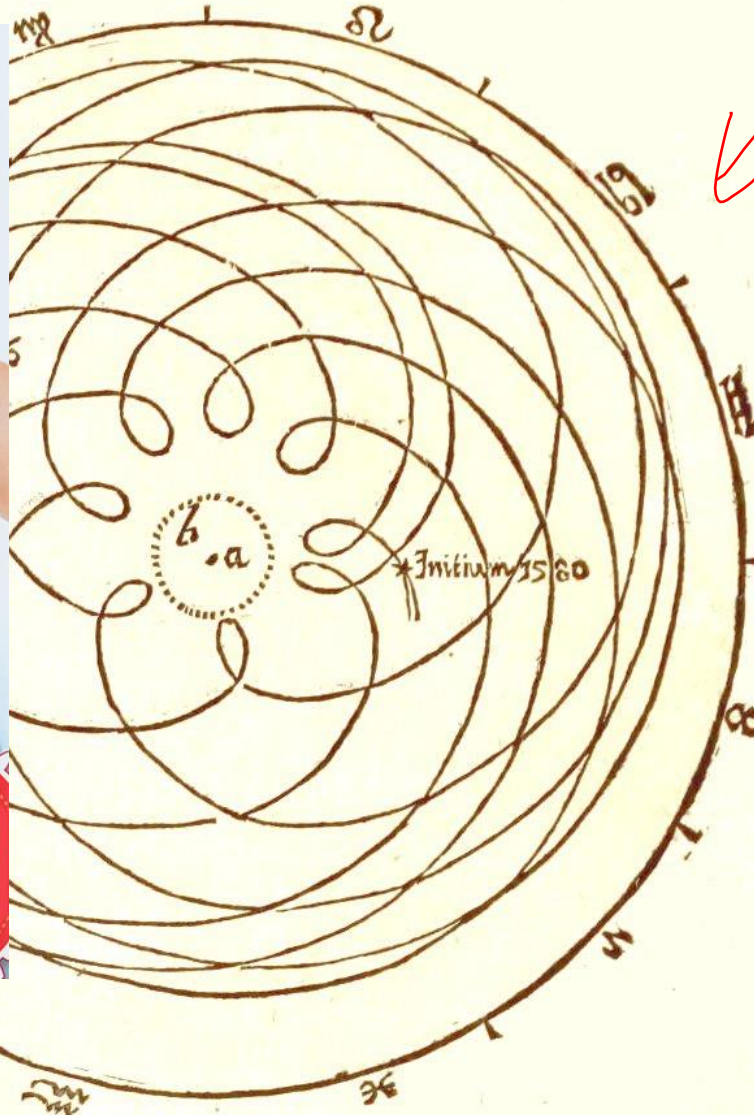
Pick correct
reference frame
to simplify
problems!



Reference frames

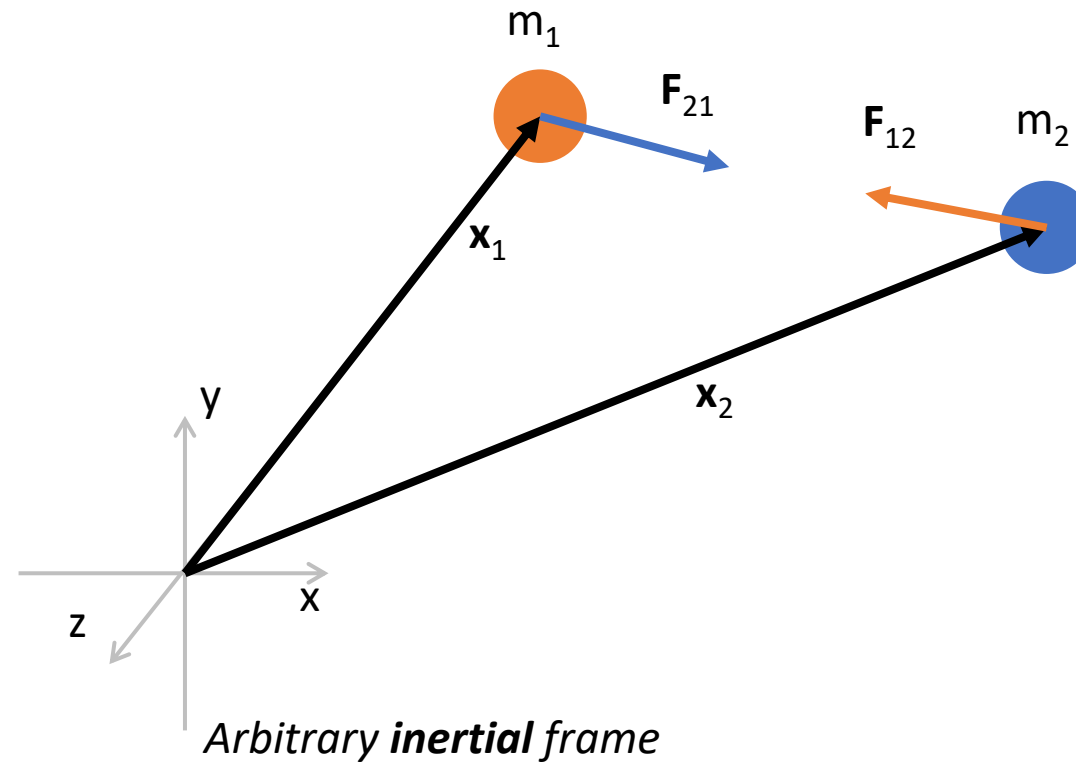


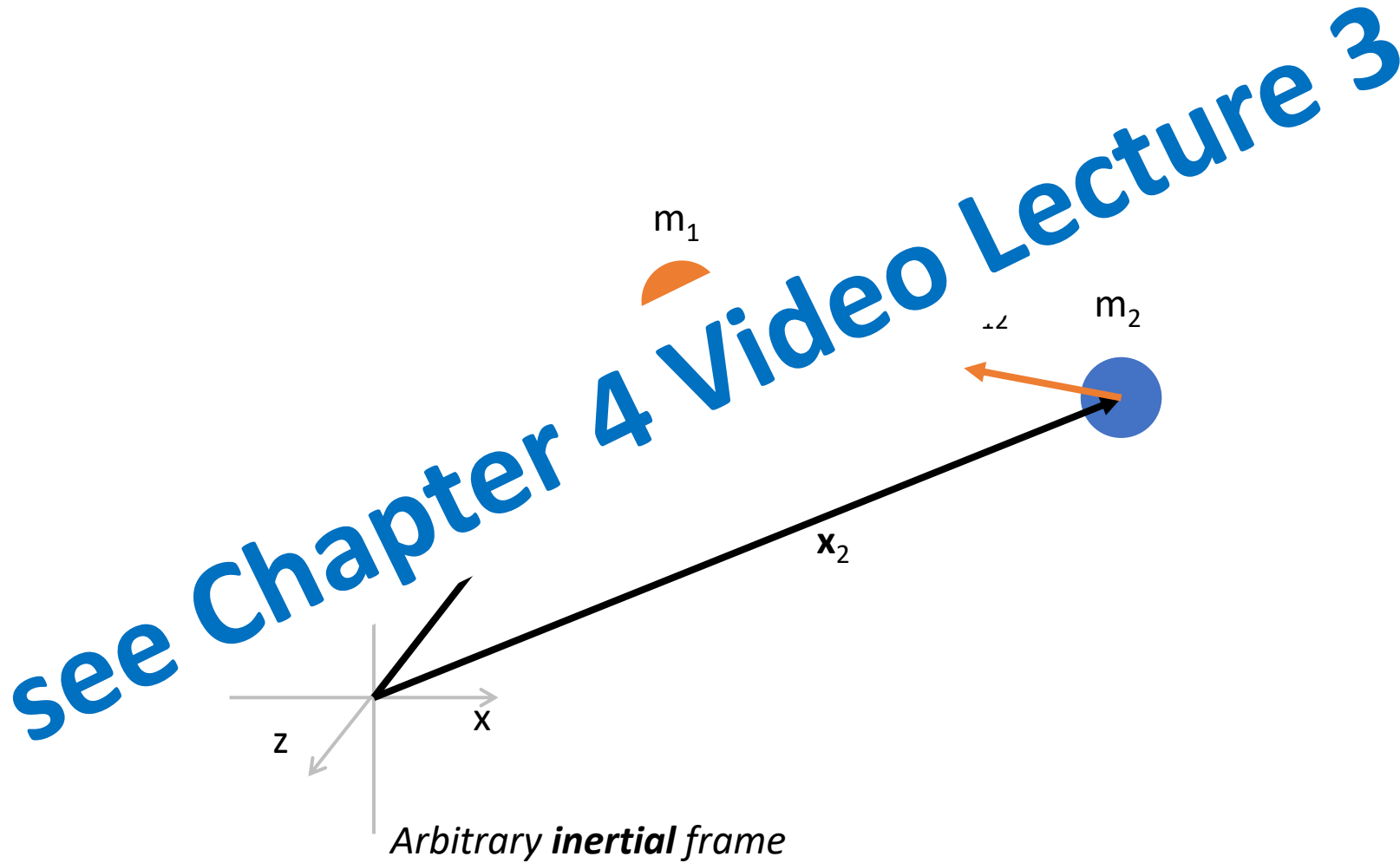
DE MOTIB. STELLÆ MARTIS



✓ Kepler's observation
of Mars' position

Johannes Kepler/Astronomia Nova





After some math: Motion splits in two

- Center of mass: moves uniformly

of "combined"
+ body

$$\vec{x}_{cm} = \frac{m_1 \vec{x}_1 + m_2 \vec{x}_2}{m_1 + m_2}$$

$$\ddot{\vec{x}}_{cm} = \vec{0}$$

$\vec{v}_{cm} = \text{constant}$

uniform motion!
(no acceleration)

- Relative distance: moves on conic section

$$\vec{d} = \vec{x}_2 - \vec{x}_1$$

$$\ddot{\vec{d}} = -\frac{\mu}{|\vec{d}|^3} \vec{d}$$

Restricted 2BP

Observations:

- Both bodies on ellipses around **barycenter**
- Heavy mass moves closer to **barycenter** than light mass

If $m_1 \gg m_2$:

- $\mu \approx Gm_1$
- $\vec{x}_{cm} \approx \vec{x}_1$

Restricted Two-Body Problem

eg: satellite and Earth

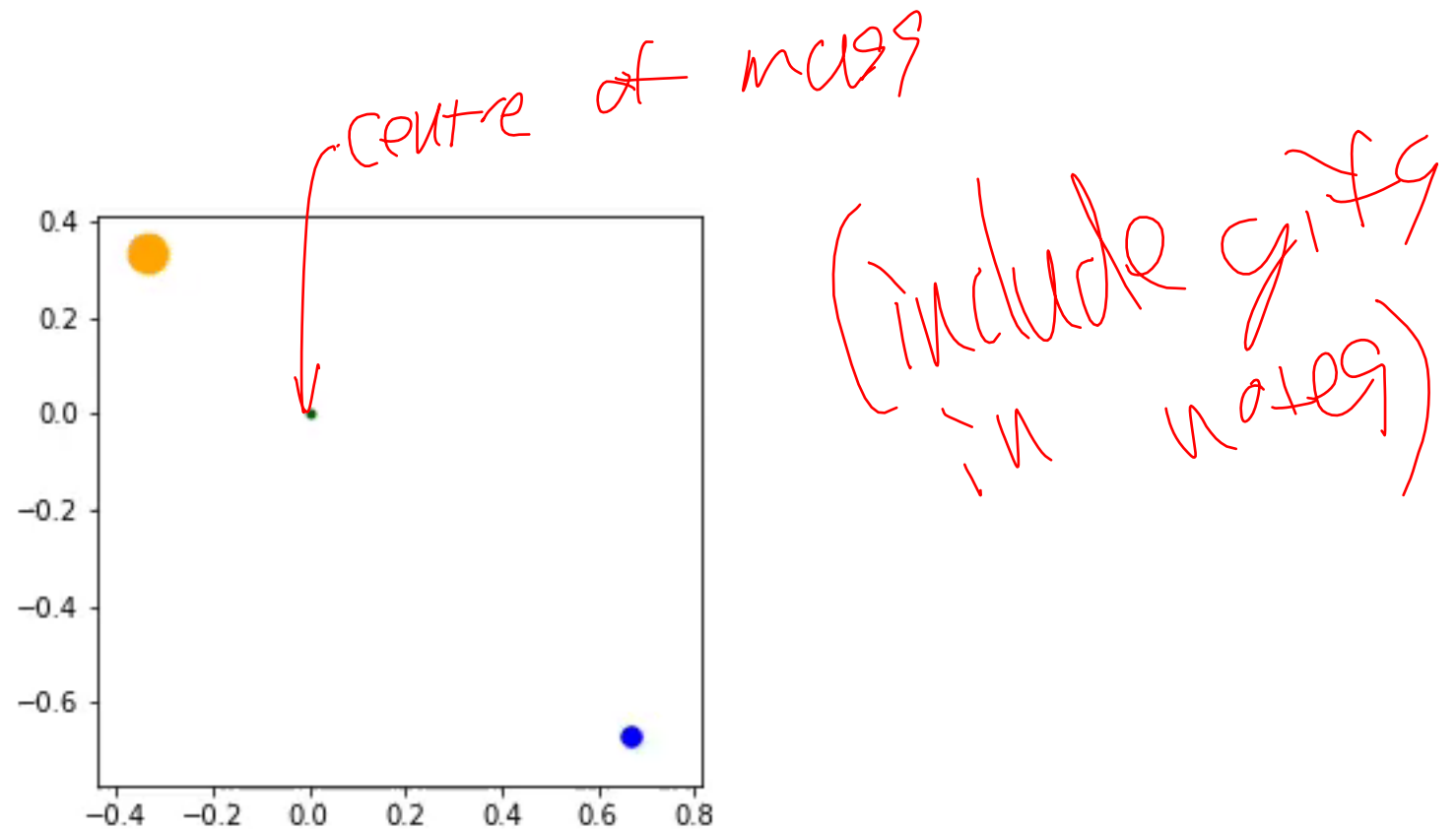
This is an
inertial frame
since it's not
accelerating!

key word!

$$\vec{x}_1 = \vec{x}_{cm} - \frac{m_2}{m_1 + m_2} \vec{d}$$

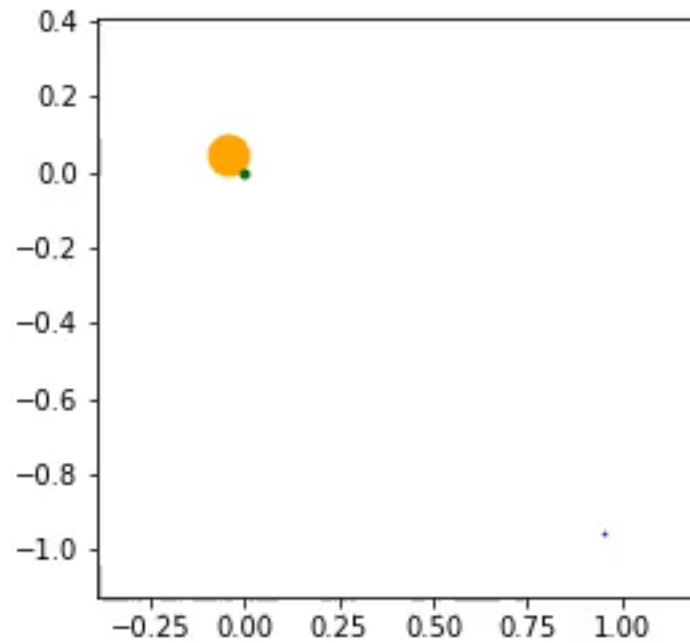
$$\vec{x}_2 = \vec{x}_{cm} + \frac{m_1}{m_1 + m_2} \vec{d}$$

$$\mu = G(m_1 + m_2)$$



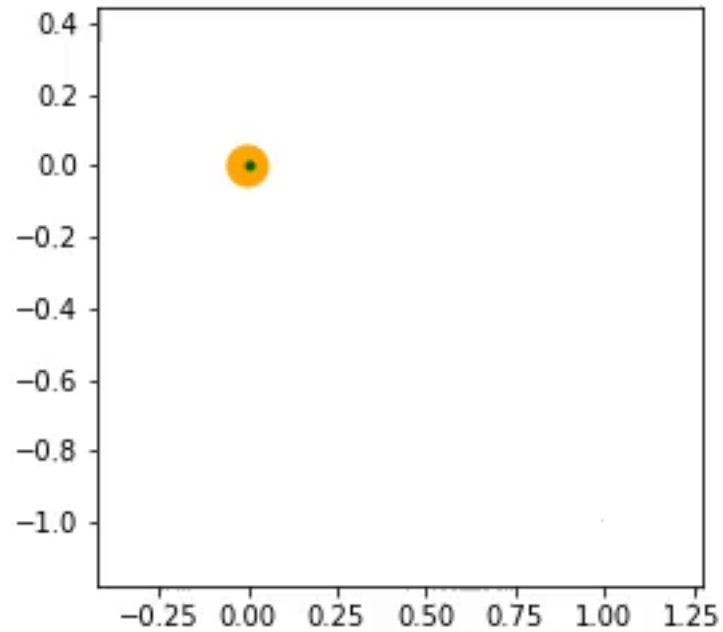
mass ratio: 1/2

mass ratio 1:20



mass ratio: 1/20

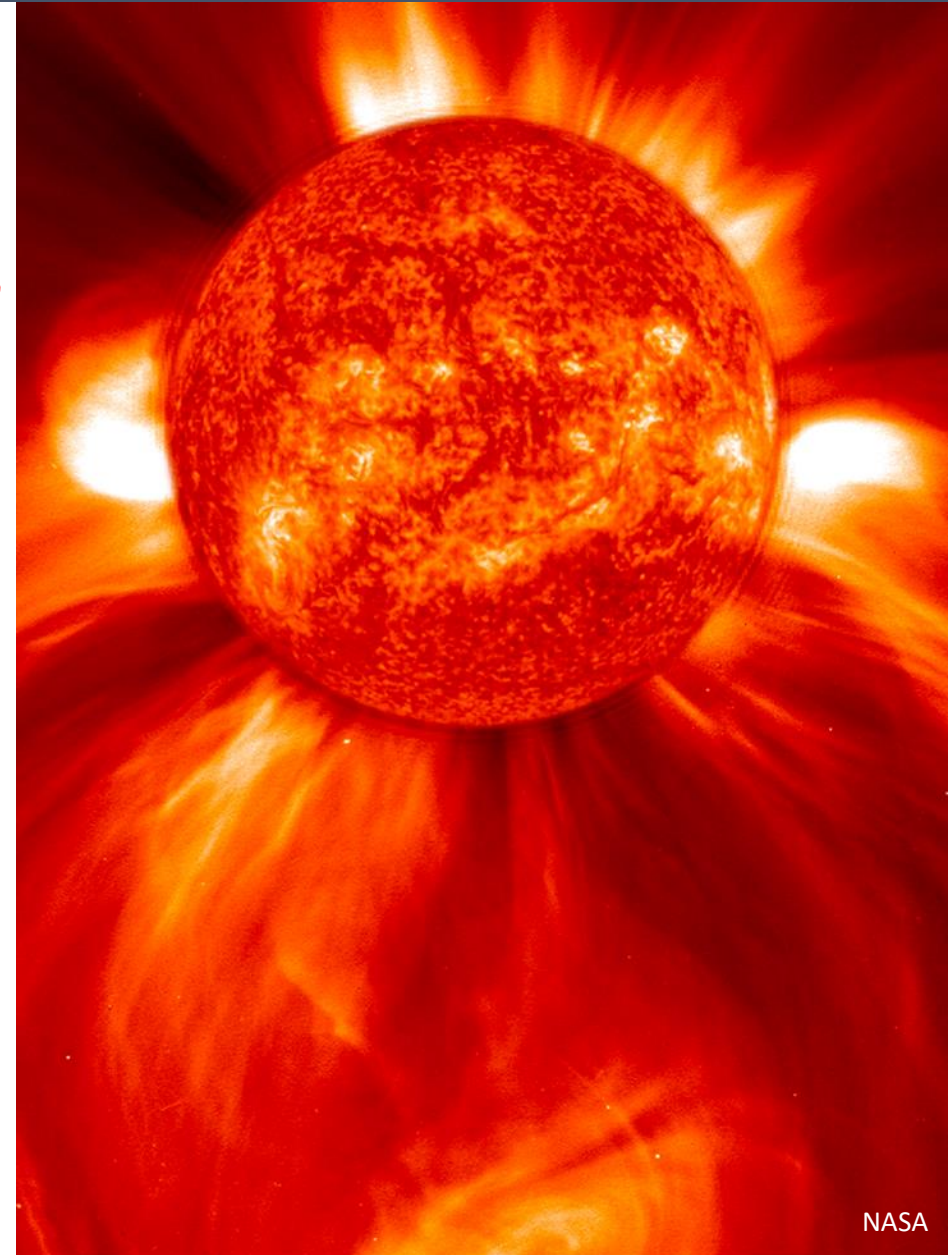
mass ratio 1:200



mass ratio: 1/200

- Full 2BP separates
 - Center of mass (barycenter) moves along line at constant velocity
 - Relative motion: moves on conic section
- Barycentric frame is inertial
- Both bodies orbit barycenter
- Large mass ratio: heavy body \approx barycenter (restricted 2BP)

✓ This holds for 2, 3, 4, 5, ... bodies
→ paggers



Advanced Astronautics (SESA3039)

Chapter 4: Orbital Mechanics – week 8

Dr. Alexander Wittig

