

SESA1015 Astronautics

Environment



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Natural and Induced Environments

Possible Environments Include:

Fabrication, Assembly, Test, Terrestrial transportation, Launch. Space Environment

Induced Natural

Space Environment:

... hostile or friendly?



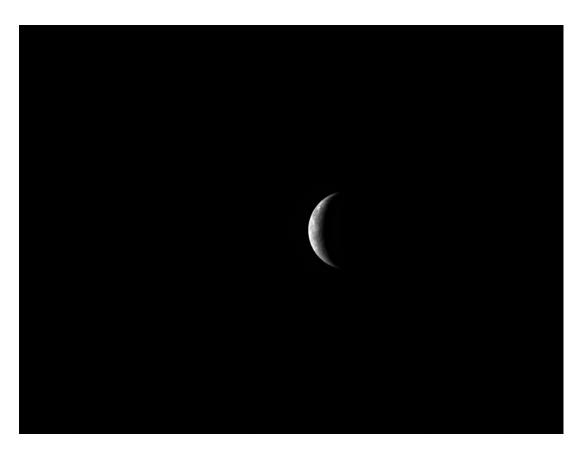


What is the Space Environment?





What is the Space Environment?



Temperature: 2.7° K

Vacuum

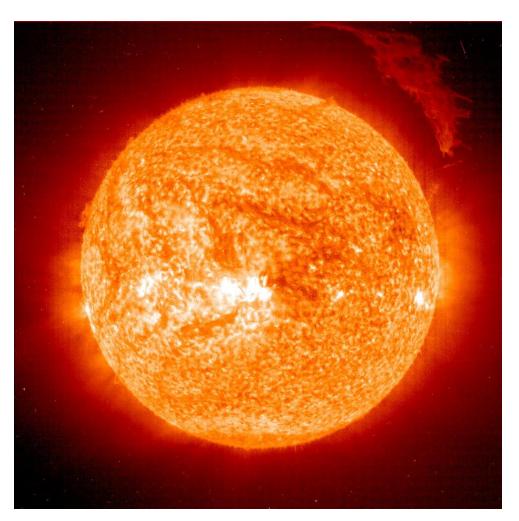
Microgravity

In our solar system, the main input to the space environment is... the sun.

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Our Solar System

The Sun

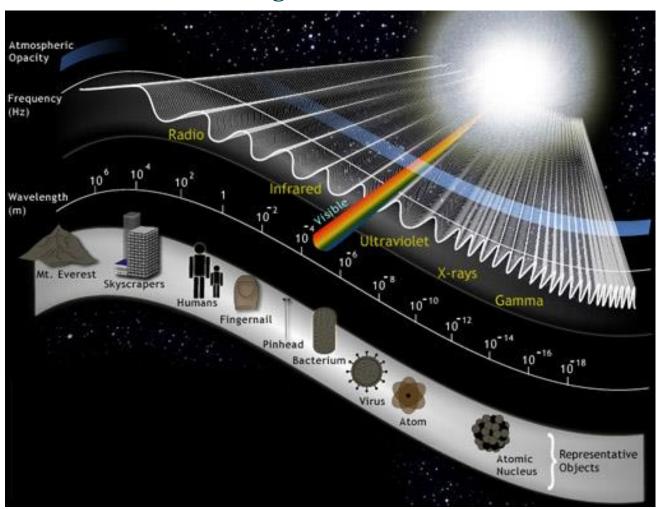


Diameter: 1,400,000 km Source of energy: Nuclear fusion

- Electromagnetic Radiation
- Particle Radiation



The Sun – Electromagnetic Radiation – The Electromagnetic Spectrum

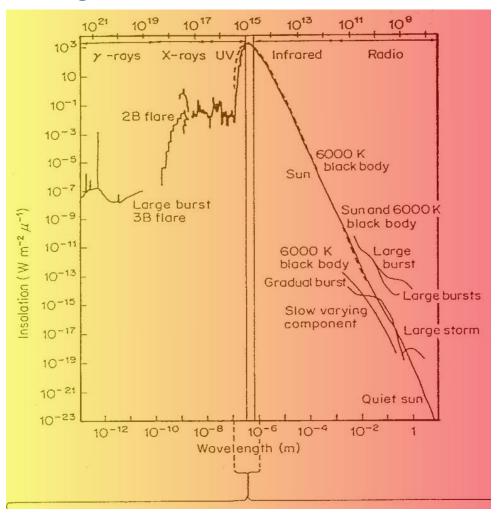


Average Energy output at 1 AU: 1370 W/m²

Solar Cycle – 11 year cycle

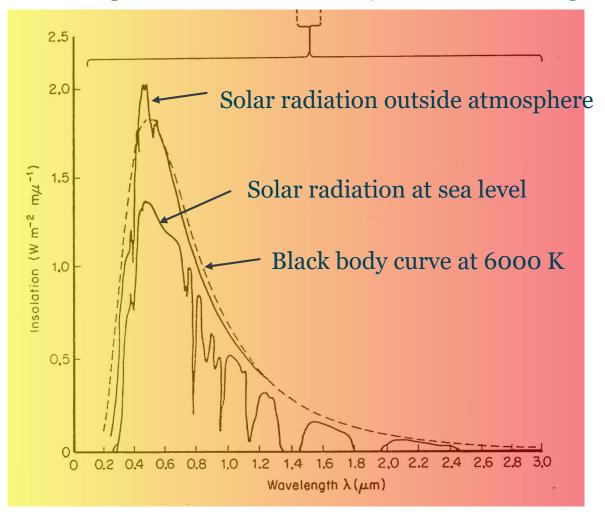


The Sun – Electromagnetic Radiation





The Sun – Electromagnetic Radiation – UV/visable wavelengths





The Sun – Electromagnetic Radiation

- Solar spectrum approximates to a black body at ~6000K in UV, visible and IR.
- Peak output in visible at ~0.5 μm wavelength
- Solar Variability:

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Radio \times 100;
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IR small;

Visible <1%;

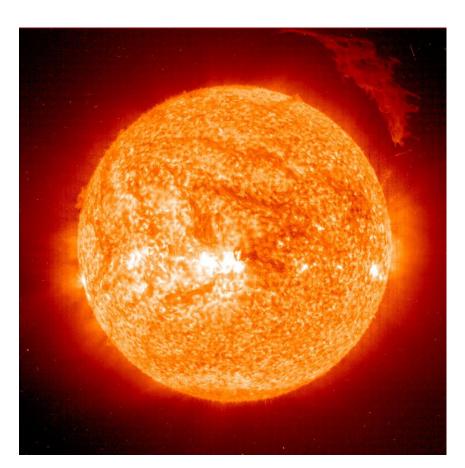
 $UV \times few;$

X-ray \times 100.

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Our Solar System

The Sun – Particle Radiation



'stream of energetic (high speed) subatomic particles called the *solar wind*'

Source: violent eruptions that take place on the Sun's surface and in its atmosphere

Material is thrown into space as a result, consisting of *protons*, *electrons* and *ions* (the nuclei of atoms stripped of their electrons)

At Earth (depending on solar activity) $n \sim 5 \times 10^6 \text{ protons/m}^3$, $v \sim 200 - 800 \text{ km/s}$

Effects on spacecraft materials + systems



The Sun – Solar Proton Events (SPE), 'Solar Storm'

At times of a solar maximum the frequency and violence of the outburst's on the suns surface ('solar flares') increase. These events can discharge a cloud of energetic charged particles towards the Earth.

Increase in high energy protons (~ 1 MeV to 1 GeV) for a few hours to a few days.

Frequency – a few / year, peaking at solar max.

Other Particle Radiation – Galactic Cosmic Rays (GCR)

Extremely high energy particles emanating from outside the solar system ($10^9 - 10^{12}$ eV).



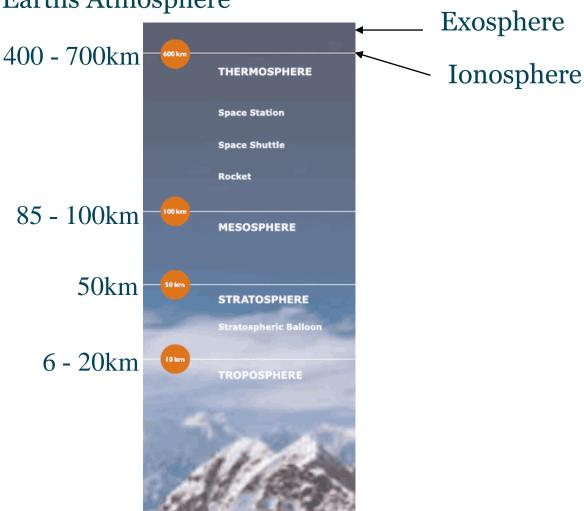
The Earth has some degree of natural protection from the Sun's output...

• The atmosphere

• The Earths Magnetic field

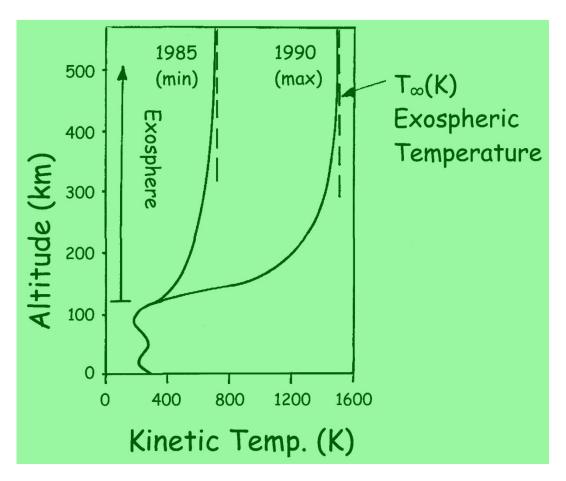


Earths Atmosphere



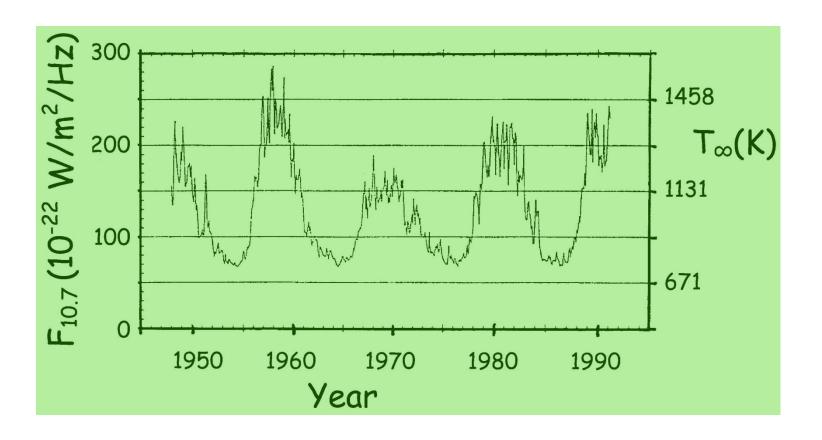


Earths Atmosphere – the atmosphere and the electromagnetic radiation



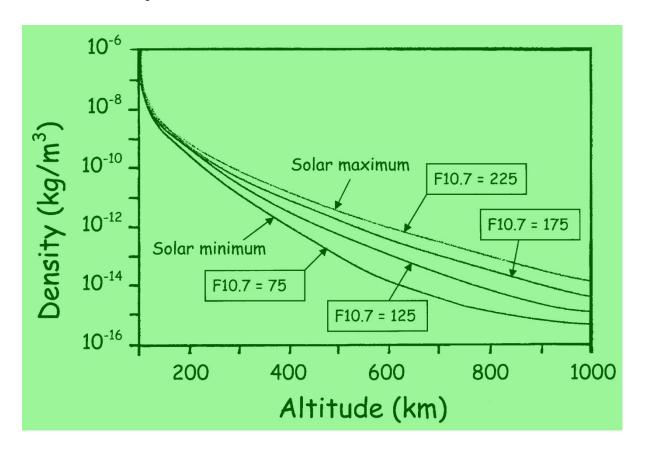


Earths Atmosphere – the atmosphere and the electromagnetic radiation Exospheric Temperature with Solar Cycle





Earths Atmosphere – the atmosphere and the electromagnetic radiation Atmospheric Density





Atmosphere – Density

Density above some reference altitude h_o is give by:

$$\rho_i = \rho_{i0} \exp\left\{\frac{-gM_i}{RT}(h - h_0)\right\},\tag{2.1}$$

where

 ρ_i = density of species *i*

 ρ_{io} = density of species *i* at height h_o

g = acceleration due to gravity

 M_i = molecular weight of species i

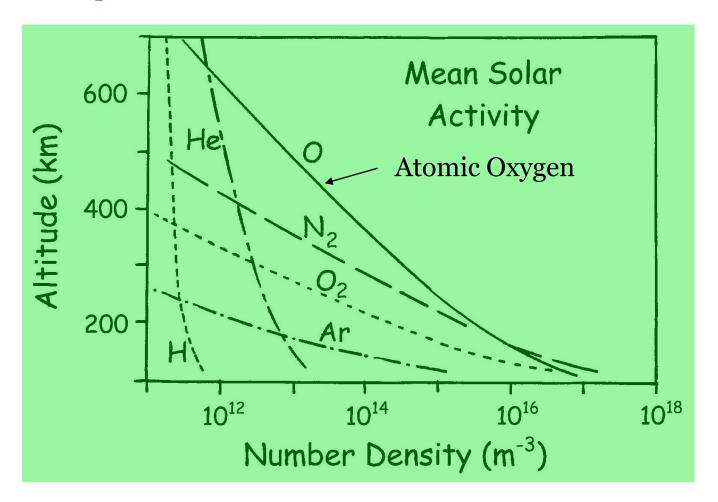
R =Universal Gas Constant

T =atmospheric temperature

Total density is
$$\rho = \rho_0 \exp\left\{-\frac{(h - h_0)}{H}\right\},$$
 where $H = \text{scale height } (\infty T/M_i)$

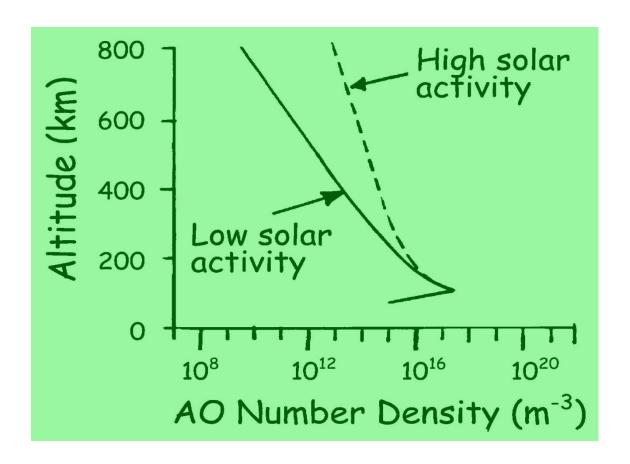


Earths Atmosphere – Constituents



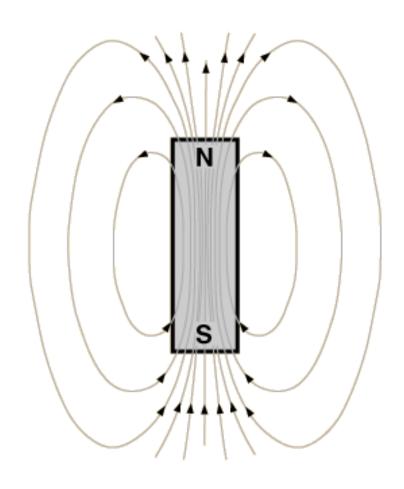


Earths Atmosphere – Atomic Oxygen (AO)



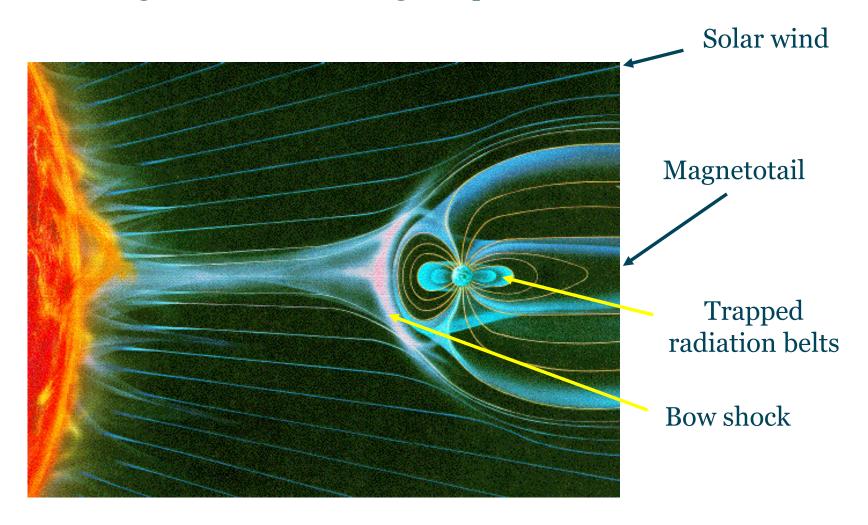


Earths Magnetic Field





Earths Magnetic Field – The Magnetosphere



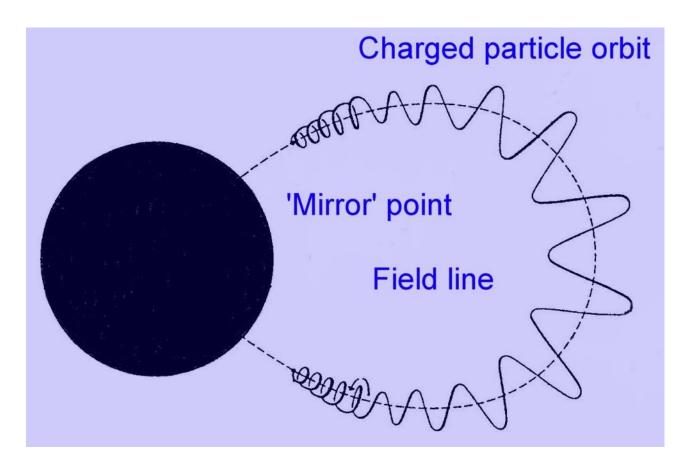


Earths Magnetosphere and Ionizing Particle Radiation

- Structure of magnetosphere formed by interaction of solar wind and Earth's (approximately) dipole field
- Particles penetrate magnetosphere at North and South magnetic poles, and become trapped (Van Allen trapped radiation belts)
- Magnetosphere boundary is dynamic and variable, depending on intensity of the solar wind ('magnetic storms')



Ionizing Particle Radiation – Trapped Radiation

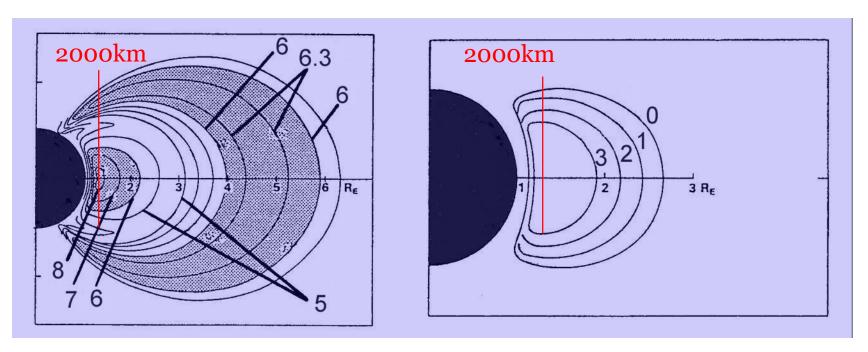




Ionizing Particle Radiation – Trapped Radiation

Structure of the Van Allen radiation belts.

Numerical values are x, where flux is 10^x cm⁻² s⁻¹.



Electrons, E > 0.5 MeV

High energy protons, E ≥100 MeV



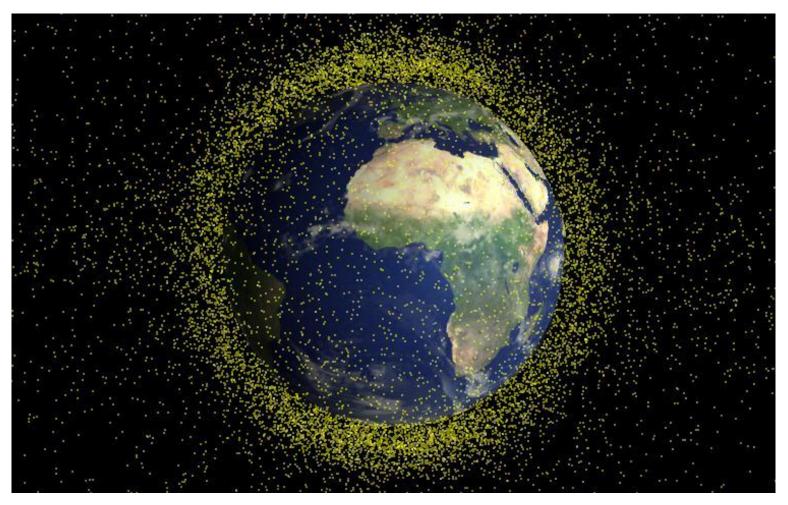
Debris

Divided into:

- *Natural Space Debris* meteroids that the earth encounters through its orbit (estimated that ~100 metric tonnes of this material rains down into the Earth atmosphere each day)
- Artificial Debris man made objects (in 2009 Earths orbital environment contained 3,371 operational satellites and 11,492 debris objects ≥ 10cm, NASA)



Debris





Summary of Space Environment

In the Solar System:

- Low Temperatures*
- Vacuum
- Microgravity
- Sun's Electromagnetic Radiation*
- Sun's Particle Radiation
- External Particle Radiation (Galactic Cosmic Rays)

Environment Around the Earth:

- Atmosphere density (orbital drag), composition (atomic Oxygen)
- Magnetic Field Trapped radiation
- Debris



Is the environment hostile or friendly?

Hostile

- High vacuum
- Particle radiation
- High energy Electromagnetic radiation
- Atomic oxygen (Earths Atmosphere)
- Debris

- Friendly
- No water vapour/rain
- No wind
- Clean environment
- Zero effective gravity



Vacuum

Some materials evaporate in hard vacuum (known as 'outgassing')

- resins
- adhesives
- lubricants
- some metals

Temperature for given outgassing rate

| Material | 0.1 μm/yr | 1 mm/yr |
|-----------|-----------|---------|
| Cadmium | 38°C | 122°C |
| Magnesium | 110°C | 233°C |
| Gold | 660°C | 950°C |
| Titanium | 920°C | 1250°C |

Generally <u>not</u> an issue of structural integrity, but one of contamination.



Micro gravity

- Structural design
- Difficulty in mechanism deployment testing on the ground
- No convective heat transport
- Propellant management problems
- (Physiological effects on humans)



Solar Radiation – Electromagnetic Output

Effects of solar e.m. radiation (UV):

- Changes in optical and electrical properties
- Embrittlement in polymers
- Darkening of solar array cover-glass and adhesives



Ionizing Particle Radiation – Effects on materials and systems

- Degradation of semi-conductor devices (solar arrays)
- On-board software errors by single particle event (Single Event Upsets, SEU)
- Spacecraft charging
- (Physiological effects on humans)



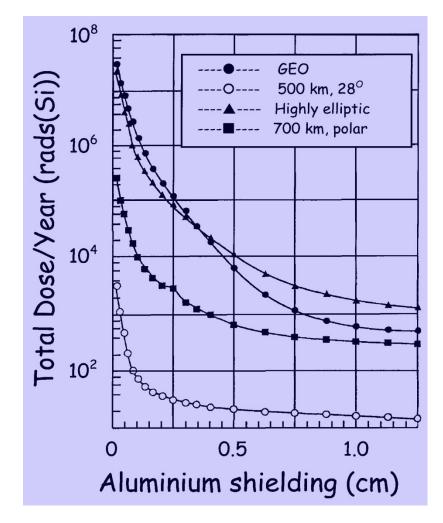
Ionizing Particle Radiation – Total Dose as a Function of Orbit

Standard Unit:

A rad (Si) (Radiation Absorbed Dose) is that amount of radiation which deposits 100 ergs of energy per gram of silicon semiconducting material.

$$(1 \text{ erg} = 10^{-7} \text{ J},$$

 \rightarrow 100 rads = 1 J/kg)





Earth Atmosphere - Effects

- aerodynamic drag, lift and heating (re-entry)
- erosion (AO)

Atomic Oxygen (AO) Erosion Example

Estimate the depth of erosion of kapton for a typical space shuttle mission of 10 days duration during solar maximum.

Kapton:

- material commonly used in spacecraft insulation blankets.
- erosion rate $\sim 2.8 \, \mu m$ for every 10^{24} atoms/m² of AO fluence



Earth Atmosphere – Atomic Oxygen (AO) Erosion Example

AO fluence over time *t* is

$$|F_0 = n_{AO}Vt \text{ (m}^{-2})|$$
 (2.3)

```
where n_{AO} = number density of AO (m<sup>-3</sup>)

V = spacecraft velocity (m s<sup>-1</sup>)
```

Typical shuttle altitude ~300 km, so

$$F_o = (10^{15} / \text{ m}^3) (7.7 \times 10^3 \text{ m/s}) (8.64 \times 10^5 \text{ s})$$

 $\sim 6.7 \times 10^{24} \text{ atoms } / \text{ m}^2$

∴ Depth of erosion
$$\sim 2.8 \ \mu m \ (6.7 \times 10^{24}) \ / \ 10^{24}$$

$$\sim 19 \ \mu m$$



Environmental Effects on Design/Man

Earth Atmosphere – Atomic Oxygen (AO) Erosion rates

| Material | Surface Recession* (µm) |
|-----------|----------------------------|
| Kapton | 2.8 |
| Teflon | 0.029 |
| Aluminium | 0.00035 |
| Silver | 9.8 |

* Surface recession for every 10²⁴ atoms / m² of AO fluence.



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- Debris

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Launch Environment

- Most Severe Environment (usually):
 - main driver for structural design of spacecraft
- 'Static' Acceleration:
 - launch vehicle dependent Ariane $5 \le 4g_0$ Shuttle $\le 3g_0$

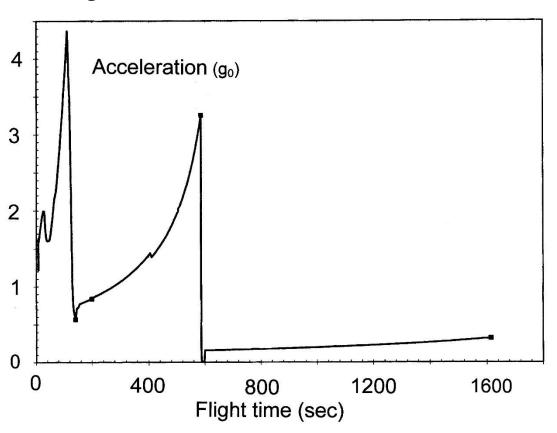
 $Saturn \le 8g_0$

- peak occurs at stage burnout



Static Acceleration

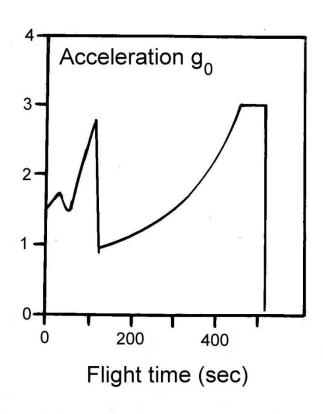
Ariane 5 static acceleration on launch







Static Acceleration



Space shuttle static acceleration

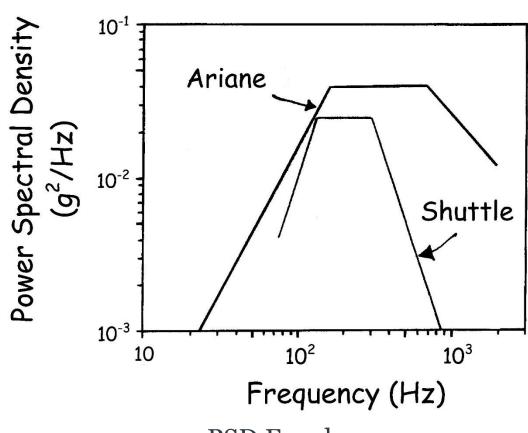


Dynamic Acceleration

(or "oscillatory acceln." or "random vibration")

Generated by:

- Functioning machinery
- Combustion phenomena
- Structural excitation in acoustic field



PSD Envelopes



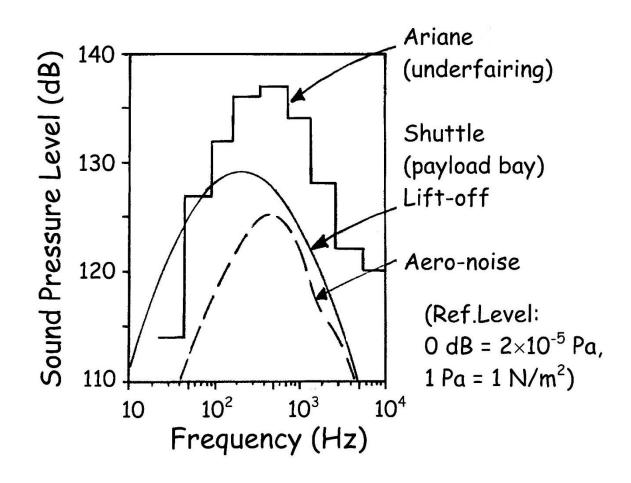
Acoustic Environment

Intense acoustic field generated by:

- engine noise
- aerodynamic noise

Greatest intensity at:

- launcher release
- transonic speed



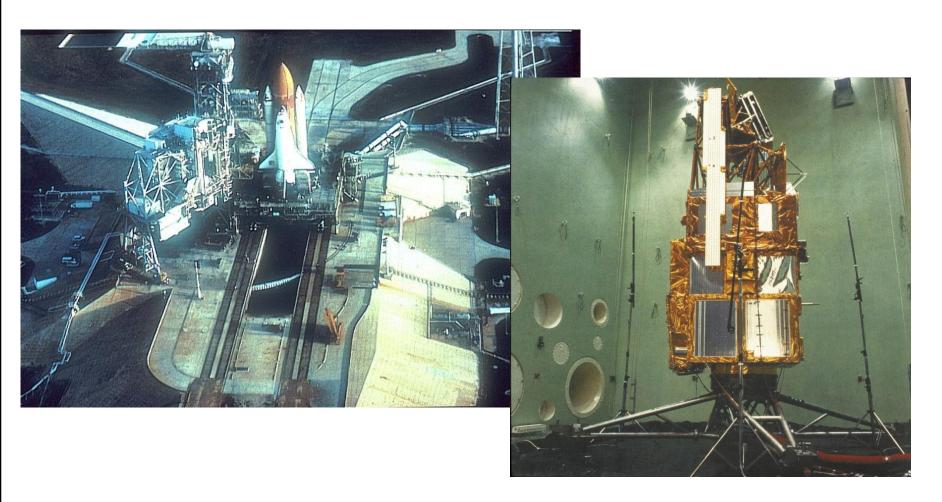
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Launch Environment





Acoustic Environment





Effects on Spacecraft

- Main input to structural design
- Ground testing to ensure spacecraft survival on launch



Environment Summary

Natural and Induced **Environments**

Our Solar System

Earth and the Earth Orbit **Environment**

Environmental Effects on Design/Man

Launch Environment

Key points:

• Categories and types of environments that the satellite will have to survive

• The important environmental inputs to our Solar system and their basic properties

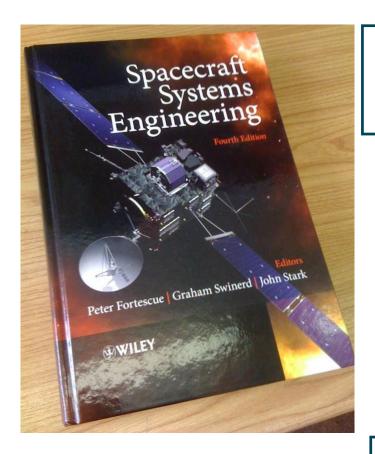
- The important properties of the space environment around the Earth
- How these properties interact with the inputs from the Solar system and
- How these properties generally vary proportional to orbital altitude/location

• How the space environmental properties, both in the Solar system and around the Earth, most significantly affect the design of spacecraft

- The important environmental inputs from launch
- How these environments are quantified
 How they affect the design of spacecraft

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Environment Summary



Read Chapter 2 of Fortescue, Stark & Swinerd

Read Chapter 6 of 'How S/C Fly'

