1 Poisson equation in one dimension

In this section, we will solve the one-dimensional Poisson equation

$$u_{xx} = f(x) \qquad (0 < x < 1) \tag{1}$$

subject to a source term f(x) and different boundary conditions at x = 0 and x = 1. First, we will solve it with finite difference methods of first and second order on a uniform grid. Finally, we solve it on a non-uniform grid and investigate how adaptive mesh refinement (AMR) can be used to obtain accurate solutions by distributing fewer points more cleverly along the grid.

1.1 Analytical solution

One way to express the analytical solution is to simply integrate equation (1) twice to get

$$u(x) = C_1 + \int^x dx' u_x(x')$$

$$= C_1 + \int^x dx' \left(C_2 + \int^{x'} dx'' u_{xx}(x'') \right)$$

$$= C_1 + C_2 x + \int^x dx' \int^{x'} dx'' f(x''),$$
(2)

where the constants C_1 and C_2 are determined from two boundary conditions and the integrals can be done from any lower limit. Note that this is equivalent to saying that the solution is a sum of the solution to the homogenuous equation $u_{xx} = 0$ and a solution to the inhomogenuous equation $u_{xx} = f(x)$.

Note that if two Neumann boundary conditions $u_x(0) = a$ and $u_x(1) = b$ are imposed, then the solution u(x) is unique only up to a constant. If u(x) is a solution to the boundary value problem defined by equation (1) and $u_x(0) = a$ subject to $u_x(1) = b$, then also $(u + C)_{xx} = u_{xx}$ in the interior and $(u + C)_x(0) = u_x(0) = a$ on the left boundary, and similarly on the right boundary x = 1. It can also be seen by observing that C_1 is undetermined when 2 is differentiated.

1.2 Numerical solution on a uniform grid

First, consider the boundary value problem defined by equation (1), subject to the boundary conditions

$$u(0) = a$$
 or $u_x(0) = a$ and $u(0) = b$ or $u_x(1) = b$.

To solve the equation numerically, we divide the interval [0,1] into the uniform grid

$$x_0 = 0$$
 x_1 x_2 x_m x_{M-1} x_M $x_{M+1} = 1$

of M+2 points and step length h. We approximate the second derivative at interior points with the central difference

$$\frac{\partial u}{\partial x}(x_m) = \frac{u_{m-1} - 2u_m + u_{m+1}}{h^2} + \mathcal{O}(h^2) \qquad (1 \le m \le M - 1).$$

To handle the Dirichlet boundary condition u(0) = a at the left edge or u(1) = b at the right edge, we insert the trivial equation

$$1 \cdot u_0 = a \qquad \text{or} \qquad 1 \cdot u_{M+1} = b.$$

To handle the Neumann boundary condition $u_x(0) = a$ at the left edge or $u_x(1) = b$ at the right edge to second order, we approximate the first derivative to second order with forward or backward differences to

get

$$u_x(0) = \frac{-\frac{3}{2}u_0 - 2u_1 - \frac{1}{2}u_2}{h} + \mathcal{O}(h^2) = b \qquad \text{or} \qquad u_x(1) = \frac{\frac{1}{2}u_{M-1} - 2u_M + \frac{3}{2}u_{M+1}}{h} + \mathcal{O}(h^2) = b.$$

Writing all these equations in $(M+2) \times (M+2)$ -matrix form AU = b, we obtain for example with u(0) = a and $u_x(1) = b$

$$\begin{bmatrix} 1 & & & & & \\ +1/h^2 & -2/h^2 & +1/h^2 & & & \\ & \ddots & \ddots & \ddots & & \\ & & +1/h^2 & -2/h^2 & +1/h^2 \\ & & & +1/2h & -2/h & +3/2h \end{bmatrix} \begin{bmatrix} U_0 \\ U_1 \\ \vdots \\ U_M \\ U_{M+1} \end{bmatrix} = \begin{bmatrix} a \\ f(x_1) \\ \vdots \\ f(x_M) \\ b \end{bmatrix},$$
(3)

where the first and last rows of the matrix generally vary depending on the boundary conditions.

Note that if the numerical solution is subject to two Neumann boundary conditions, the matrix becomes singular and the solution non-unique. In this case, we impose the additional constraint $U_0 = 0$ by setting all entries in the first column of A to zero. To handle the singular matrix, we solve the system with a least-squares method instead of a GESV LU factorization method.

We now apply our method to the boundary value problem with the source function

$$f(x) = x + \cos(2\pi x).$$

Inserting it into equation (2) and doing the integrals, we get the exact solution

$$u(x) = C_1 + C_2 x + \frac{1}{3!} x^3 - \frac{1}{4\pi^2} \cos(2\pi x)$$
.

In figure 1, we present numerical solutions for three different combinations of boundary conditions.

Our approach to handling the boundary conditions is not the only possible approach. The system of equations is equivalent if we remove the first row and column of A and the first entries in U and b, but simultaneously modify the entry $f(x_1) \to f(x_1) - a/h^2$. This approach is more consistent with treating U_0 as a known variable, since its precise value is defined by the Dirichlet boundary condition. However, our approach of inserting a trivial equation $1 \cdot U_0 = a$ keeps the matrix dimensions independent of boundary conditions and makes it easier to reason with how the discretized differential operator represented by A operates on the grid point U_0 in the same way it operates on all other grid points.

Neumann boundary conditions can also be handled differently. Instead of approximating the second derivative only on actual grid points, we could approximate it with a fictuous point x_{-1} and a central difference $u_x(0) \approx (U_1 - U_{-1})/(2h)$. Then we could use this together with the central difference $(U_{-1} - 2U_0 + U_1)/h^2 = f(x_0)$ to eliminate U_{-1} . Eliminating U_{-1} , the first equation becomes $(U_1 - U_0)/h = a + hf(x_0)/2$, so the boundary condition could be handled by setting the first row to $[-1/h, +1/h, 0, \dots]$ and modifying the first entry in b to $a \to a + hf(x_0)/2$. This would also be second order, and would allow us to use the same stencil also at x_0 , but we would then have to pay the price of modifying the right side of the matrix equation in an unnatural way.

1.3 Adaptive numerical solution on a non-uniform grid

We will now demonstrate how the numerical solution can be generalized to a non-uniform grid with $x_i-x_{i-1} \neq$ const. Then we will attempt to make the numerical solution as good as possible using as few grid points as possible, by placing points tighter where the solution varies rapidly.

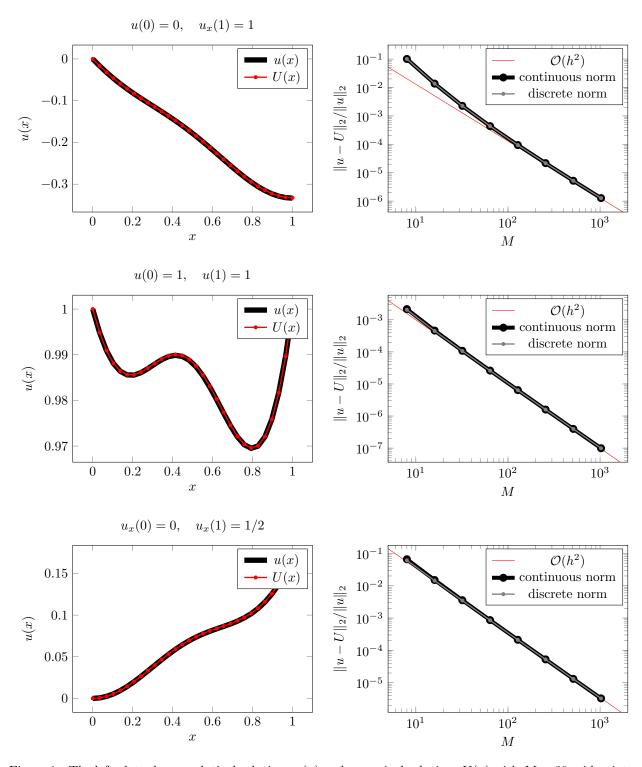


Figure 1: The left plots show analytical solutions u(x) and numerical solutions U(x) with M=30 grid points for $u_{xx}=x+\cos 2\pi x$ subject to three different boundary conditions. The right plots show convergence plots corresponding to the same boundary conditions, where the error is measured with both the a continuous and discrete L_2 -norm.

To derive a nonuniform stencil for the second derivative at x_m , we proceed similarly to the uniform stencil. First approximate one derivative by stepping halfway left and right, landing at $x_{m-1/2}$ and $x_{m+1/2}$. Then we approximate another derivative by stepping halfway to the sides again, landing at x_{m-1} , x_m and x_{m+1} . This yields

$$u_m'' \approx \frac{u_{m+1/2}' - u_{m-1/2}'}{x_{m+1/2} - x_{m-1/2}} \approx \frac{2}{x_{m+1} - x_{m-1}} \left(\frac{u_{m+1} - u_m}{x_{m+1} - x_m} - \frac{u_m - u_{m-1}}{x_m - x_{m-1}} \right).$$

Assuming Dirichlet boundary conditions, the nonzero entries of A (indexed from zero) becomes

$$A_{0,0} = A_{M+1,M+1} = 1$$

$$A_{m,m-1} = \frac{2}{x_{m+1} - x_{m-1}} \frac{1}{x_m - x_{m-1}}$$

$$A_{m,m} = \frac{-2}{x_{m+1} - x_{m-1}} \left(\frac{1}{x_m - x_{m-1}} + \frac{1}{x_{m+1} - x_m}\right)$$

$$A_{m,m+1} = \frac{2}{x_{m+1} - x_{m-1}} \frac{1}{x_{m+1} - x_m}$$

The job is then once again to solve the system AU = b. Note that the stencil reduces to the one in equation (3) when $x_m - x_{m-1} = x_{m+1} - x_m = h$, as it should.

To do adaptive mesh refinement, we will

- 1. Start with a coarse uniform grid, such as $[x_0, x_1] = [0, 1]$.
- 2. Wisely choose one grid interval $[x_m, x_{m+1}]$ based on some strategy.
- 3. Split the interval in half by inserting a new point at $(x_m + x_{m+1})/2$.
- 4. Repeat step 2 and 3 until the grid has the desired resolution.

We will compare three different strategies for selecting the grid interval:

1. Error strategy: Select the interval $[x_m, x_{m+1}]$ with the largest error

$$\int_{x_m}^{x_{m+1}} dx |u(x) - U(x)|, \quad \text{where } U(x) = U_m + \frac{x - x_m}{x_{m+1} - x_m} (U_{m+1} - U_m)$$

is a linearly interpolated numerical solution on the *current* grid and u(x) is the exact solution. This strategy requires knowledge of the exact solution u(x) and solving the system numerically before each splitting.

2. Truncation error strategy: Select the interval $[x_m, x_{m+1}]$ with the largest absolute truncation error

$$\left| \frac{2}{x_{m+1} - x_m} \left(\frac{u_{m+1} - u_{m+1/2}}{x_{m+1} - x_{m+1/2}} - \frac{u_{m+1/2} - u_m}{x_{m+1/2} - x_m} \right) - f(x_m) \right|,$$

upon insertion of a middle point $x_{m+1/2} = (x_m + x_{m+1})/2$, where u(x) is the exact solution. This strategy also requires knowledge of the exact solution u(x), but does not rely on intermediate computations of the numerical solution.

3. Source strategy: Select the interval $[x_m, x_{m+1}]$ with the largest "absolute source"

$$\int_{x_m}^{x_{m+1}} \mathrm{d}x |f(x)|.$$

In physical applications, f(x) is typically mass density or charge density. The idea is to refine intervals on which there is much mass or charge, as the solution is expected to vary faster there. This splitting strategy requires neither knowlege of the exact solution or the numerical solution, only on the source function f(x), as is typically the case in practice.

In figure 2, we demonstrate how the initial grid [0,0.5,1] and the numerical solution U(x) evolves through adaptive refinement with the error strategy. Observe how the refinement concentrates on resolving critical areas of the solution near the peak and the inflection points.

In figure 3, we compare the convergence of the three adaptive refinement strategies to the $\mathcal{O}(h^2)$ -convergence of uniform refinement on the same problem. The error strategy requires knowledge of the exact solution and intermediate computations, but in return it is the most effective strategy. The source strategy requires neither, but is also the least effective strategy. We can say that the more knowledge of the exact solution and intermediate computations, the greater the accuracy.

Note that the errors are not strictly decreasing with each refinement $M \to M+2$. In particular, the error from the error strategy exhibits an oscillating pattern for $M \geq 32$. This is a weakness of refining only exactly two symmetric intervals for each refinement. An alternative method is to refine multiple intervals at every refinement step using a criterion that splits not only the interval with the largest error, but all intervals with error above some reference error. This procedure removes our control over the exact number of intervals, but in return gives us control over the maximal acceptable error on any interval. In section 6.4.2, we will improve our AMR strategy exactly in this way. The effect is that the oscillating pattern is eliminated and that the error decreases strictly with each refinement step. This will be equivalent to jumping directly from one local minimum in the oscillation to the next.



Figure 2: During adaptive mesh refinement (AMR) with the error strategy (strategy number 1), the interval with the largest error $\int \mathrm{d}x |u(x) - U(x)|$ is split in half. Here, $u(x) = \exp(-(x-1/2)^2/0.1)$ is the symmetric solution to whichever Poisson equation has $f(x) = u_{xx}$ on $x \in [0, 1]$. Symmetry is imposed numerically by also adding the point 1-x to the grid whenever a point $x \neq 1/2$ is added.

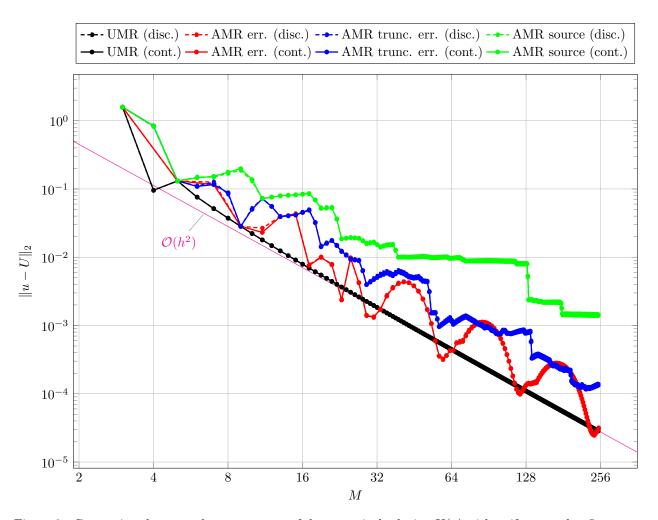


Figure 3: Comparison between the convergence of the numerical solution U(x) with uniform mesh refinement (UMR) and adaptive mesh refinement (AMR) on the problem $u_{xx} = f(x)$ on $x \in [0, 1]$ with analytical solution $u(x) = \exp(-(x-1/2)^2/0.1)$. The adaptive refinement is done using three different strategies that subdivide the interval with the largest absolute error |u-U|, largest truncation error Lu - f(x) (where $L \approx \partial^2/\partial x^2$ is the discretized differentiation operator) or largest amount of source $\int \mathrm{d}x |f(x)|$. Errors $||u-U||_2$ are measured with the continuous and discrete L_2 -norm.

2 Heat equation in one dimension

In this section, we consider the one-dimensional heat equation for u = u(x, t),

$$u_t = u_{xx}, \quad u(x,0) = f(x), \quad x \in [0,1] := \Omega,$$

with either Neumann or Dirchlet boundary conditions, and solve it numerically using both the Backward Euler method and Crank-Nicolson method. These are $\mathcal{O}(k+h^2)$ and $\mathcal{O}(k^2+h^2)$ methods respectively, and we will analyze and compare their convergence using mesh refinement as we did in section 1. However we will here restrict our attention to uniform grids only.

2.1 Numerical solution method

To solve the heat equation numerically we first perform semi-discretization, i.e. we do spatial discretization and keep the time continuous. As in section 1, we divide the interval Ω into M+2 equidistant nodes with separation h=1/(M+1), so that we get a uniform grid with M internal nodes and two boundary nodes. We then express the spatial derivative using the central finite difference to get

$$u_t(x_m, t) = \frac{1}{h^2} \delta_x^2 u(x_m, t) + \mathcal{O}(h^2), \quad m = 0, ..., M + 1.$$

We now introduce the single variate functions $v_m(t)$ as the approximation to $u(x_m, t)$, at each node x_m , turning the PDE into a set of ODEs

$$\frac{dv_m(t)}{dt} = \frac{1}{h^2} \delta_x^2 v_m(t), \quad v_m(0) = f(x_m).$$

The problem is then generally solved by imposing the boundary conditions, and numerically integrating the equations in time. The integration can be done by using one of the many off-shelf routines for ODEs, e.g. Euler's method, and for convenience we employ the θ -method, which for general ODEs y' = g(y, t) is given as

$$y^{n+1} = y^n + \Delta t((1-\theta)g(y^n, t_n) + \theta g(y^{n+1}, t_{n+1})).$$

Using a constant time step $\Delta t = k$ leads to uniform discretization of the time interval, which we divide into N equidistant points, and the final discrete grid for space and time is illustrated in figure 4. This gives the approximate solution of $v_m(t)$ at N finite times $t_n = nk$, n = 0, ..., N-1, k = 1/(N-1), and we denote the fully discretized approximation of $u(x_m, t_n)$ as U_m^n . After organizing the terms, the θ -method for the 1D heat equation is then written out as

$$(1 - \theta r \delta_x^2) U_m^{n+1} = (1 + (1 - \theta) r \delta_x^2) U_m^n, \tag{4}$$

where we have defined $r = k/h^2$. Setting the value of θ in (4) determines the specific numerical scheme. We have

Forward Euler
$$\theta = 0$$

Backward Euler $\theta = 1$
Crank-Nicolson $\theta = \frac{1}{2}$,

and we will as mentioned consider the Bacward-Euler and Crank-Nicolson methods.



Figure 4: Discrete uniform grid

2.1.1 Boundary conditions

For Dirchlet boundary conditions $u(0,t)=\sigma$, $u(1,t)=\beta$, we substitute $U_0^{n+1}=\sigma$ and $U_M^{n+1}=\beta$ in (4) for m=0 and m=M+1. In matrix form, the equation for the remaining inner nodes is then written as

For Neumann boundary conditions, $u_x(0,t) = \sigma$, $u_x(1,t) = \beta$, we introduce fictitious nodes to the left and right of the spatial grid, and approximate the first derivatives at the boundaries by

$$\frac{U_1 - U_{-1}}{2h} = \sigma \quad and \quad \frac{U_{M+2} - U_M}{2h} = \beta.$$

Inserting this into (4) for m=0 and m=M+1, and eliminating the fictitious nodes we get

$$(1+r\theta)U_0^{n+1} - r\theta U_1^{n+1} = (1-2r(1-\theta))U_0^n + r(1-\theta)U_1^n - 2hr\sigma \quad (for m = 0)$$

$$(1+r\theta)U_{M+1}^{n+1} - r\theta U_M^{n+1} = (1-2r(1-\theta))U_{M+1}^n + r(1-\theta)U_M^n - 2hr\beta \quad (for m = M+1),$$

which we combine with (4) for the inner nodes, $1 \le m \le M$, to get the equation as a $(M+2) \times (M+2)$ linear system

$$\begin{bmatrix} 1 + \theta r & -\theta r \\ -\theta r & 1 + 2\theta r & -\theta r \\ & \ddots & \ddots & \ddots \\ & & -\theta r & 1 + 2\theta r & -\theta r \\ & & -\theta r & 1 + \theta r \end{bmatrix} \begin{bmatrix} U_0^{n+1} \\ U_1^{n+1} \\ \vdots \\ U_{M+1}^{n+1} \\ U_{M+1}^{n+1} \end{bmatrix}$$

$$= \begin{bmatrix} 1 - 2r(1-\theta) & r(1-\theta) \\ r(1-\theta) & 1 - 2r(1-\theta) & r(1-\theta) \\ & \ddots & \ddots & \ddots \\ & & & r(1-\theta) & 1 - 2r(1-\theta) & r(1-\theta) \\ & & & & & r(1-\theta) & 1 - 2r(1-\theta) \end{bmatrix} \begin{bmatrix} U_0^n \\ U_1^n \\ \vdots \\ U_M^n \\ U_{M+1}^n \end{bmatrix} + \begin{bmatrix} -2hr\sigma \\ 0 \\ \vdots \\ 0 \\ -2hr\beta \end{bmatrix}. (6)$$

All quantities on the right hand sides in (5) and (6) are known, i.e. the equations are on the form $A\vec{x} = \vec{b}$, and the known \vec{b} is just written via a matrix-vector product for notational convenience. The system of equation is then solved at each time step using "scipy.sparse.linalg.spsolve", and the left hand side matrix is stored as a compressed sparse row/column matrix.

With the numerical schemes in hand we now solve the heat equation with the following Neumann boundary conditions and initial condition,

$$u_x(0,t) = u_x(1,t) = 0, \quad u(x,0) = 2\pi x - \sin(2\pi x).$$
 (7)

The computed solutions for $t \in [0, 0.5]$ is plotted in figure 5, and qualitatively the solution behaves in accordance to what one should expect for the heat equation. To quantify and compare the accuracy of the numerical schemes we will now proceed to analyze convergence using mesh refinement, similar to what we did in section 1.

2.2 Convergence and mesh refinement

As in section 1 we now analyze the convergence of the numerical solution methods, by doing mesh refinement of the spatial grid x_m . For (7), the the analytical solution is not available in closed form, so in order to analyze convergence we compute a reference solution using a sufficiently high M, which we use in place of the analytical solution when computing the error.

In order to analyze the convergence further, we also consider the heat equation with a set of boundary and initial conditions for which the analytical solution is known. Specifically we consider

$$u(0,t) = u(1,t) = 0, \quad u(x,0) = \sin(\pi x),$$
 (8)

on the same domain $x \in [0,1] := \Omega$ and t > 0. Note that we now have Dirchlet boundary conditions, and the analytical solution is readily available as ¹

$$u(x,t) = \sin(\pi x)e^{-\pi^2 t}.$$
(9)

¹The analytical solution can be computed using e.g. separation of variables.

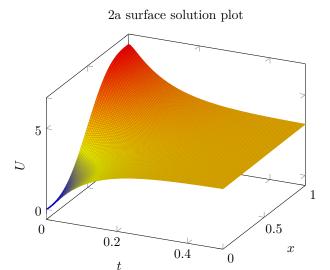


Figure 5: I am a surface plot, Hooray:)

When doing mesh refinement of the spatial grid, we vary the number of spatial grid points M, and compute the numerical solution at the same point in time $t = t_{end}$. We keep the number of time steps N fixed, so that the error from the time discretization does not vary, and compute the L_2 discrete relative error with respect to the reference solution. For equation (8) we also compute the l_2 continuous relative error, and compute errors with respect to the analytical solution. The resulting convergence plots are shown in figure 6 and 7.

Both methods are second order in the spatial step h, and are unconditionally stable, unlike e.g. the explicit Forward Euler method ($\theta = 0$). Despite this, the solution with the Neumann boundary conditions 7 seems to break down for sufficiently low M using Backward Euler, and the error blows up as seen in figure 6. Comment from KA: I don't really know why.

Crank-Nicolson is however one order higher in the time step k, so that the total error is lower for Crank-Nicolson than Backward-Euler when M and N are the same. This is seen when refining the spatial step h; at large h (low M), the error is dominated by the spatial error, and as we decrease h (increase M) we should see the expected second order convergence in the spatial step. At some point, when the spatial error has become sufficiently small, the total error will be dominated by the error in the time step k, and we expect this to happen earlier for the Backward-Euler method. In figure 7 we see exactly this. The error of the Crank-Nicolson solution with N=10000 exhibits second order convergence throughout the entire refinement, but when lowering the number of time steps to N=1000, the time step error starts to dominate towards the end of the refinement and the error curve flattens. For the solution computed with the Backward-Euler method we see the flattening happening much earlier, which is due to this method being less accurate in time.

3 Invicid Burgers' equation

In this section we turn to solve the inviscid Burgers' equation with given Dirchlet boundary conditions and initial condition

$$u_t = -uu_x$$
, $u(0,t) = u(1,t) = 0$, $u(x,0) = exp(-400(x-1/2)^2)$. (10)

2a Convergence plot L2 discrete rel. error

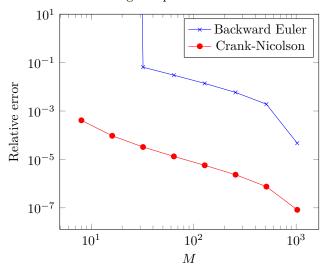


Figure 6: Convergence plot, h refinement (2a)

This equation exhibits breaking; after some point in time t_b the solution breaks, and the unique solution does not exist, leading to the formation of a *shock wave*.[?] The time t_b before this can happen is given by

$$t_b = \frac{-1}{minf'(x)},\tag{11}$$

where f(x) is the given initial condition u(x,0) = f(x).[?]

Numerical solution method

To solve (10) numerically we perform semidiscretization in the same way as we did for the heat equation in section 2, also on a uniform spatial grid as described in section 1. The resulting system of ODEs is

$$\frac{\partial v_m}{\partial t} = -v_m \frac{1}{2h} (v_{m+1} - v_{m-1}).$$

We impose the Dirchlet boundary conditions and integrate the ODEs using *solve_ivp* from the SciPy library, with the default explicit Runge-Kutta method of order 4(5)[?].

Comment/question: Is it fine to use scipy.integrate.solve_ivp, or should it be all home cooking?

3.1 Time of breaking

Insertion of u(x,0) for f in (11), gives $t_b \approx 0.058$. To get a criterion for when the numerical solution has broken down, we use that the stable solution should be strictly increasing from x=0 to towards the apex, and then strictly decreasing from the apex towards the right boundary at x=1. When this is no longer the case we say that the solution has broken, and the time for which this happened for our solution was at $t^* \approx 0.055$. Figure 8 shows the solution sampled around the time of breaking.



Figure 7: Convergence plot, h refinement (2b)



Figure 8: Numerically computed solution to the Invicid Burgers' equation around the time of breaking.

4 Laplace equation in two dimensions

In this section, we will solve the two-dimensional Laplace equation on a quadratic domain

$$u_{xx} + u_{yy} = 0, (x, y) \in \Omega := [0, 1]^2,$$
 (12)

with boundary conditions on the edges of Ω

$$u(0, y) = 0,$$

 $u(1, y) = 0,$
 $u(x, 0) = 0,$
 $u(x, 1) = \sin(2\pi x).$

We will solve this equation numerically using a five point stencil, but first, we solve it analytically to provide a reference solution which can be compared with the numerical one.

4.1 Analytical solution

The solution of equation 12 can be found by separation of variables. First, assume that we can write

$$u(x, y) = \alpha(x)\beta(y),$$

which implies that

$$u_{xx} + u_{yy} = \alpha''(x)\beta(y) + \alpha(x)\beta''(y) = 0,$$

where the prime markers ' denote differentiation of the single variable functions $\alpha(x)$ and $\beta(y)$. Rearranging, we get that

$$\frac{\alpha''(x)}{\alpha(x)} = \frac{\beta''(y)}{\beta(y)} = c$$

must be constant, since α and β are functions of independent variables. Thus, we have two second order differential equations

$$\alpha''(x) - c\alpha(x) = 0,$$

$$\beta''(x) - c\beta(x) = 0,$$

with boundary conditions

$$\alpha(0) = \alpha(1) = \beta(0) = 0,$$

$$\alpha(x)\beta(1) = \sin(2\pi x).$$

Setting $\beta(1)$ to 1 yields $\alpha(x) = \sin(2\pi x)$, so that $\alpha''(x) = -4\pi^2 \alpha(x)$ where y = 1, we find that $c = -4\pi^2$. Solving the equation for $\beta(y)$, we find that

$$\beta(y) = b_1 e^{\sqrt{c}y} + b_2 e^{-\sqrt{c}y}.$$

Inserting $c = 4\pi$ and the boundary conditions $\beta(0) = 0$ and $\beta(1) = 1$, we get

$$\beta(y) = \frac{\sinh(2\pi y)}{\sinh(2\pi)},$$

and finally

$$u(x,y) = \frac{\sin(2\pi x) \cdot \sinh(2\pi y)}{\sinh(2\pi)}.$$

4.2 Numerical solution

We solve the equation numerically by discretizing the domain $\Omega = [0,1]^2$, approximate the equation on that domain using a five point stencil, and solving the approximated system. The domain is discretized with M+2 and N+2 points in the x and y direction, so that there are M and N internal points in each direction. The total system to be solved is thus $M \times N$ points, as the boundaries are known.

Rewriting Laplace's equation using central differences, we get

$$\begin{split} \partial_x^2 u(x_m, y_n) &= \frac{1}{h^2} [u(x_{m-1}, y_n) + 2u(x_m, y_n) + u(x_{m+1}, y_n)] + \mathcal{O}(h^2) \\ &= \frac{1}{h^2} \delta_x^2 u(x_m, y_n) + \mathcal{O}(h^2), \\ \partial_y^2 u(x_m, y_n) &= \frac{1}{k^2} [u(x_m, y_{n-1}) + 2u(x_m, y_n) + u(x_m, y_{n+1})] + \mathcal{O}(k^2) \\ &= \frac{1}{k^2} \delta_y^2 u(x_m, y_n) + \mathcal{O}(k^2), \end{split}$$

where (x_m, y_n) denote the point (m, n) in the grid. Adding these expressions, and naming our approximated solution $U_{m,n} := u(x_m, y_n)$, we find that the Laplace equation can be approximated

$$0 = \partial_x^2 u(x_m, y_n) + \partial_y^2 u(x_m, y_n) \approx \frac{1}{h^2} \delta_x^2 U_{m,n} + \frac{1}{k^2} \delta_y^2 U_{m,n},$$

or, simplifying the notation with the notation visualized in figure 9,

$$\frac{1}{k^2}(U_{\text{above}} + U_{\text{below}} - 2U_{\text{center}}) + \frac{1}{h^2}(U_{\text{left}} + U_{\text{right}} - 2U_{\text{center}}) = 0.$$

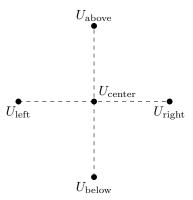


Figure 9: The five-point stencil corresponding to central difference differentiation in both the x- and y-direction.

We will now construct the matrix A such that we can write our equation as the matrix equation AU = 0, where U is the flattened solution. Ignoring firstly the above and below nodes of the stencil, we can easily set up a matrix A' in the same way as in Section 1. Note that this is done only in order to clarify the derivation – the matrix A' is merely a "stepping stone" – not a useful result.

$$A'U = \frac{1}{h^2} \begin{bmatrix} -2 & 1 & & & \\ 1 & -2 & 1 & & \\ & \ddots & \ddots & \ddots & \\ & & 1 & -2 & 1 \\ & & & 1 & -2 \end{bmatrix} \begin{bmatrix} U_{1n} \\ U_{2n} \\ \vdots \\ U_{M-1,n} \\ U_{M,n} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 0 \end{bmatrix},$$

Note also that this equation only considers one particular value of y, corresponding to n. In order to actually solve our entire system, we must include the nodes above and below as well. This can be done by considering a much larger matrix A and a much longer vector U. The latter being a stacked vector containing all M elements U_{11}, \ldots, U_{M1} , followed by $U_{12}, \ldots U_{M2}$ and so on. In this formulation of the problem, the values U_{right} and U_{left} correspond to the neighbouring points in U. The above and below nodes – instead of being above and below U_{center} – are now to the sides, M nodes away. See figure 10.

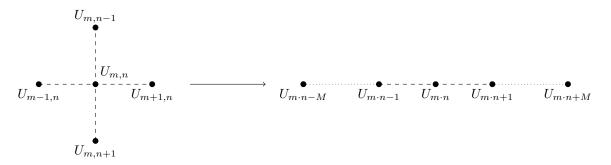


Figure 10: By flattening the five-point stencil, we can write the system of equations, which is now on the form AU = 0.

We thus write

which can be solved. It is important to note the zeros on the upper and lower diagonal, which correspond to the nodes that have less than four neighbours, ie. the nodes on the border. Gien initial conditions on the boundary $\partial\Omega$, these may easily be incorporated by adding them as an inhomogenity. Thus, the final equation will be on the form AU = b, where b contains the boundary conditions.

The matrix A is also showed in a more managable way in figure 11, where it is plotted as a heatmap. Noticing its recursive structure one may realize that the matrix can be constructed by a Kronecker sum. This procedure is discussed in more depth in 7 where the Biharmonic equation is solved using the fast poisson solver. We will here however only take the observation about the Kronecker sum as a convenient way to implement the construction of our matrix. Let K_n be the system matrix of the one dimensional finite central

difference scheme of size n introduced earlier. That is the $N \times N$ matrix

$$K_n = \begin{bmatrix} -2 & 1 & & & \\ 1 & -2 & 1 & & & \\ & \ddots & \ddots & \ddots & \\ & & 1 & -2 & 1 \\ & & & 1 & -2 \end{bmatrix}.$$

The matrix A is then compactly written as

$$A = \frac{1}{h^2} K_N \oplus \frac{1}{k^2} K_M = \frac{1}{h^2} K_N \otimes I_M + I_N \otimes \frac{1}{k^2} K_M.$$
 (13)

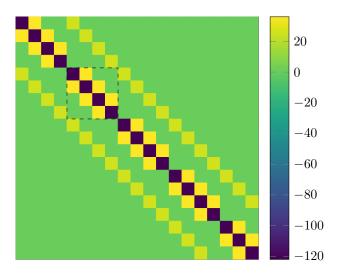
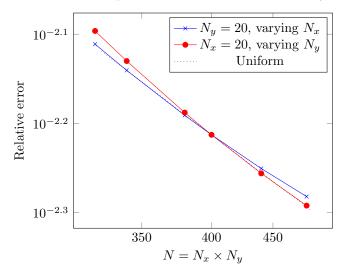


Figure 11: The five point stencil matrix for the case $M=4,\,N=5$. Notice the recursive structure.

Error plot for varios values of Nx and Ny



5 Linearized Korteweg-De Vries equation in one dimension

In this section, we will study the one-dimensional linearized Korteweg-De Vries equation

$$\frac{\partial u}{\partial t} + \left(1 + \pi^2\right) \frac{\partial u}{\partial x} + \frac{\partial^3 u}{\partial x^3} = 0 \qquad (t \ge 0) \quad (-L/2 \le x \le +L/2),\tag{14}$$

where the solution u = u(x,t) is subject to periodic boundary conditions

$$u(x+L,t) = u(x,t).$$

5.1 Analytical solution

As the solution is periodic in space at every time t, it can be expressed as a Fourier series

$$u(x,t) = \sum_{n=-\infty}^{+\infty} c_n(t) \exp(iq_n x)$$
(15)

with wavenumbers $q_n = 2\pi n/L$ and time-dependent coefficients $c_n(t)$ that ensure spatial periodicity at all times. This can be derived formally by separation of variables. Inserting the Fourier series into equation (14) gives the condition

$$\sum_{n} \left(\dot{c}_n(t) + i \left(\left(1 + \pi^2 \right) q_n - q_n^3 \right) c_n(t) \right) \exp\left(i q_n x \right) = 0$$

on the coefficients. Due to orthogonality of the Fourier basis functions $\exp(iq_nx)$, the sum can vanish only if all prefactors vanish separately. This gives a first-order differential equation for each coefficient with the solution

$$c_n(t) = c_n(0) \exp\left(-i\left(\left(1 + \pi^2\right)q_n - q_n^3\right)t\right). \tag{16}$$

5.2 Numerical solution method

To find a numerical solution $U_m^n = U(x_m, t_n) \approx u(x_m, t_n) = u_m^n$ of the Korteweg-De Vries equation, we will discretize it with central differences in space and integrate over time with the Euler method and the Crank-Nicholson method. For the first order spatial derivative, we use the central difference

$$\frac{\partial u_m^n}{\partial x} \approx \frac{\delta u_m^n}{2h} = \frac{u_{m+1}^n - u_{m-1}^n}{2h}.$$

We repeat the same finite difference three times to approximate the third order spatial derivative as

$$\frac{\partial^3 u_m^n}{\partial x^3} \approx \frac{\delta^3 u_m^n}{(2h)^3} = \frac{u_{m+3}^n - 3u_{m+1}^n + 3u_{m-1}^n - u_{m-3}^n}{8h^3}.$$

Inserting these approximations into equation (14), we get the intermediate result

$$\frac{\partial u_m^n}{\partial t} \approx F(u^n) = -\left(1 + \pi^2\right) \frac{u_{m+1}^n - u_{m-1}^n}{2h} - \frac{u_{m+3}^n - 3u_{m+1}^n + 3u_{m-1}^n - u_{m-3}^n}{8h^3}.$$

For later convenience, we write the Euler method and Crank-Nicholson method collectively with the θ method. This gives the final system of difference equations for the numerical solution

$$\frac{U_m^{n+1} - U_m^n}{k} = (1 - \theta)F(U^n) + \theta F(U^{n+1}),\tag{17}$$

where the Euler method or the Crank-Nicholson method is obtained by inserting $\theta = 0$ or $\theta = 1/2$, respectively. In matrix form, the system can be written

$$(I - \theta kA) U^{n+1} = (I - (1 - \theta) kA) U^n, \tag{18}$$

where $U^n = \begin{bmatrix} U_0^n & \dots & U_{M-1}^n \end{bmatrix}^T$ and A =

where we have imposed periodic boundary conditions $U_m^n = U_{m+M}^n$ by simply wrapping the spatial derivative stencils around the matrix.

We then solve the system by preparing U^0 from the initial condition u(x,0) and solve equation (18) repeatedly to step forward in time. Note that with the constant time step k, all matrices in equation (18) are constant in time, and the process of solving the system many times can be accelerated by for example LU-factorizing the matrix on the left side.

Next, we test our numerical solution on the problem defined by the initial condition $u(x,0) = \sin(\pi x)$ on $x \in [-1,+1]$ with L=2. The Fourier series of the analytical solution then has nonzero coefficients $c_{\pm 1}(0) = \pm 1/2i$ and wavenumbers $q_{\pm 1} = \pm \pi$, which give rise to the analytical solution $u(x,t) = \sin(\pi(x-t))$ when inserted into equation (15). As shown in figure 12, the solution represents a sine wave traveling with velocity 1 to the right.

In figure 13, we compare snapshots of the numerical solution at t = 1 from the Euler method and the Crank-Nicholson method. Note that the Crank-Nicholson method approaches the exact solution with as little as N = 10 time steps and a few hundred spatial grid points M, while the Euler method produces garbage with

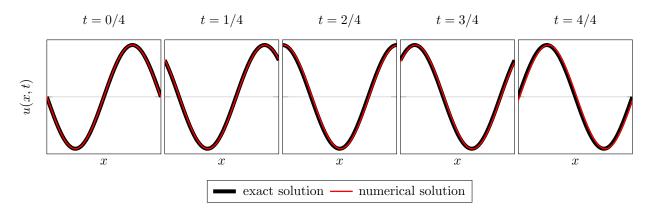


Figure 12: Comparison between the time evolution of the exact solution $u(x,t) = \sin(\pi(x-t))$ and the numerical solution from the Crank-Nicholson method with h = 1/799 and k = 1/99.

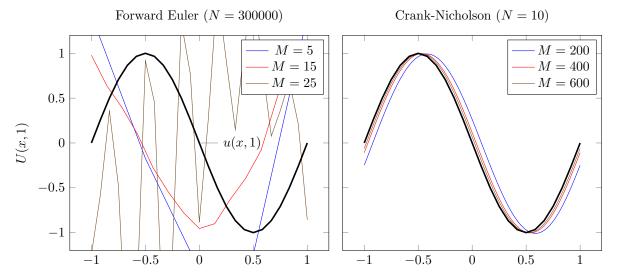


Figure 13: Snapshots of the numerical solution U(x,1) and the exact solution u(x,1) for a constant number of time steps N, but varying number of grid points M with the Forward Euler and Crank-Nicholson method. The left plot is meant to demonstrate the downfall of the Euler method and is not supposed to look pretty.

as many as N = 300000 time steps and becomes less stable as the number of spatial grid points increases. The convergence plot of the error at t = 1 in figure 14 supports our suspicions, showing the stable second order (TODO: why is it first order? is it supposed to be?) nature of the Crank-Nicholson method and the unstability of the Euler method.

5.3 Stability analysis

Motivated by the examples of the Euler method and the Crank-Nicholson method, we perform a Von Neumann analysis of their stability. Just like the exact solution, the numerical solution is subject to periodic boundary conditions in space and can therefore be expanded in a Fourier series

$$U_m^n = U(x_m, t_n) = \sum_l C_l^n \exp(iq_l x_m).$$
 (19)

Consider now a single Fourier mode $C_l^n \exp(iq_lx_m)$ in this series. Inserting it into equation (17), dividing by $\exp(iq_lx_m)$ and expanding exponentials using Euler's identity gives

$$\frac{C_l^{n+1} - C_l^n}{k} = i\left((1 - \theta)C_l^n + \theta C_l^{n+1}\right)f(q_l), \text{ where } f(q_l) = \left(-\left(1 + \pi^2\right)\frac{\sin(q_l h)}{h} - \frac{\sin^3(q_l h)}{h^3}\right).$$

Now look at the amplification factor $G_l = C_l^{n+1}/C_l^n$ of Fourier mode l over one time step. With $\theta = 1/2$, the Crank-Nicholson method gives

$$G_l = \frac{1 + ikf(x)/2}{1 - ikf(x)/2} \quad \Longrightarrow \quad |G_l| = 1. \tag{20}$$

The amplitude of all Fourier modes is thus preserved over time independently of k and h, and we say the Crank-Nicholson method is **unconditionally stable**.

The Euler method has $\theta = 0$ and gives

$$G_l = 1 + ikf(x) \implies |G_l| = \sqrt{1 + k^2 f(q_l)^2}.$$
 (21)

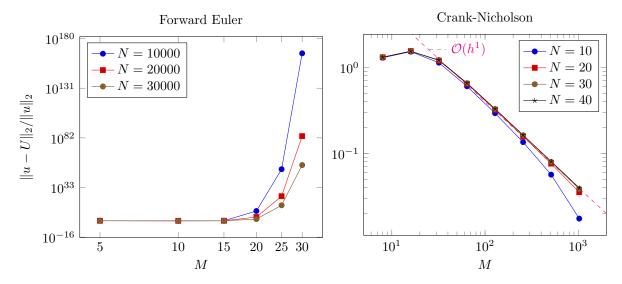


Figure 14: Convergence plots of the numerical solution U(x,t) for the Forward Euler and Crank-Nicholson method on the problem defined by the exact solution $u(x,t) = \sin(\pi(x-t))$.

Since $|\sin(q_l h)| \leq 1$ for all q_l , we can bound $f(q_l)$ by

$$|f(q_l)| \le \frac{(1+\pi^2)}{h} + \frac{1}{h^3} = \frac{1}{h^3} \left((1+\pi^2)h^2 + 1 \right) \le \frac{1}{h^3} \left((1+\pi^2)L^2 + 1 \right).$$

Then $|G_l| = \sqrt{1 + O(k^2/h^6)}$. The Von Neumann stability criterion $|G_l| \le 1 + O(k)$ [1] is attained only with $k \le O(h^6)$, corresponding to extremely small time steps. Thus, while the Euler method in theory is **conditionally stable**, it is unstable for all practical time steps. The Crank-Nicholson method is far superior, as it remains stable while allowing much coarser spatial resolution and larger time steps.

5.4 Time evolution of norm

The stability of the finite difference methods can be even better illustrated by investigating the time evolution of the L_2 -norm of the solution. To this end, we will first show that the L_2 -norm of the analytical solution is preserved over time. Then we will show the time evolution of the L_2 -norm of numerical solutions.

The L_2 -norm of the analytical solution is defined as

$$||u(x,t)||_2 = \left(\frac{1}{2} \int_{-L/2}^{+L/2} |u(x,t)|^2 dx\right)^{1/2}.$$

To understand why it is constant in time, insert the Fourier expansion equation (15) for u(x,t) to get

$$\int_{-1}^{+1} dx \left| u(x,t) \right|^2 = \sum_{m,n} c_m(t) c_n^*(t) \underbrace{\int_{-L/2}^{+L/2} \exp\left(i(q_m - q_n)x\right)}_{L\delta_{m,n}} dx = L \sum_n \left| c_n(t) \right|^2.$$

Since the Fourier coefficients (16) have constant magnitude $|c_n(t)| = |c_n(0)|$, the L_2 -norm is the same at any time t. Note that it is the *odd* number of spatial derivatives in equation (14) that gives the exponential in equation (16) a purely imaginary argument, and preserves the amplitude of each Fourier mode.

We now investigate the norm of the numerical solution with the initial gaussian $u(x, 0) = \exp(-x^2/0.1)$. The time evolution illustrated in figure 15 shows how multiple Fourier modes are activated. In figure 16, we show

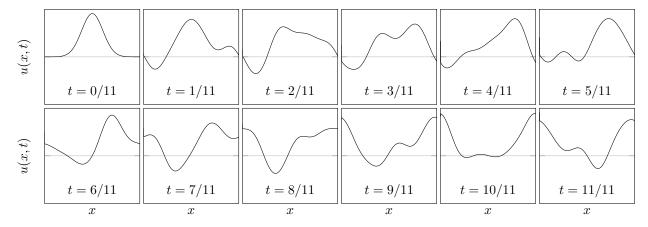


Figure 15: Time evolution of a initial gaussian $u(x,0) = \exp(-x^2/0.1)$ computed from the Crank-Nicholson method on a grid with M = 800 points in space and N = 100 points in time.

how the norm of the numerical solution evolves over time. The Euler method diverges even with tiny time steps, reflecting the amplification factor $G_l > 1$ found in equation (21). In contrast, the Crank-Nicholson metod is always stable and preserves the norm of the solution, reflecting the amplification factor $G_l = 1$ found in equation (20).

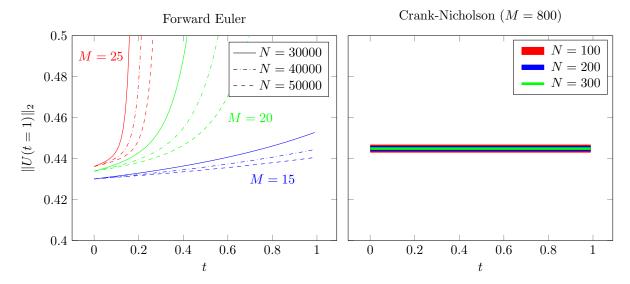


Figure 16: Time evolution of the discrete L_2 -norm of the initial gaussian $u(x,0) = \exp(-x^2/0.1)$ computed from the Euler and the Crank-Nicholson method, on different grids.

The stability and norm preservation of the Crank-Nicholson method makes it the method of choice for problems like this, where the analytical solution is known to have the same property.

6 Poisson equation in one dimension finite element method

In this section, we will again solve the Poisson equation

$$-\frac{\partial^2 u}{\partial x^2} = f(x), \quad u(a) = \alpha, \quad u(b) = \beta, \quad (a \le x \le b)$$
 (22)

subject to Dirichlet conditions, but this time using finite elements instead of finite differences.

6.1 Analytical solution

The solution to the Poisson equation is the same as in 2, but with $f(x) \to -f(x)$, so that

$$u(x) = C_1 + C_2 x - \int_0^x dx' \int_0^{x'} dx'' f(x'').$$
 (23)

6.2 Weak formulation for the exact solution

To derive a finite element method, we first split the solution into two terms

$$u(x) = \hat{u}(x) + r(x)$$
, with $\hat{u}(a) = \hat{u}(b) = 0$ and $r(x) = \alpha \frac{x-b}{a-b} + \beta \frac{x-a}{b-a}$. (24)

Note that $u''(x) = \hat{u}''(x)$ and $r(a) = \alpha$ and $r(b) = \beta$. The purpose of this splitting is that \hat{u} solves 22 with homogenuous Dirichlet boundary conditions, while r(x) lifts the values at the boundaries to satisfy the inhomogenuous boundary conditions.

Now insert 24 into equation (22), multiply it by a **trial function** v(x) and integrate both sides from a to b. We let v(a) = v(b) = 0 and use integration by parts on the left, dropping the boundary term term $-[u'(x)v(x)]_a^b$. This gives the **weak formulation** of the problem:

Find
$$\hat{u}(x)$$
 such that
$$\int_a^b dx \, \hat{u}'(x)v'(x) = \int_a^b dx \, f(x)v(x) - \int_a^b dx \, r'(x)v'(x) \quad \text{for all } v(x). \tag{25}$$

The weak formulation 25 is equivalent to the original boundary value problem 22. Any u(x) that solves 22 also solves 25, and it is possible to show that the converse is also true.

6.3 Weak formulation for the approximate solution

We have not made any approximations yet. The approximation lies in seeking a solution $U(x) \approx u(x)$ that belongs to a function space different from the one in which the exact solution u(x) belongs. Here, we suppose U(x) lies in the space of piecewise linear functions. We will then repeat the process above to derive a weak formulation for U(x), similarly as for the exact solution.

To see how this works, we first divide the interval [a, b] into the grid

$$a = x_0 < x_1 < \dots < x_M < x_{M+1} = b \tag{26}$$

and let U(x) be piecewise linear in each **finite element** $[x_i, x_{i+1}]$. Similarly to u(x), we split the approximate solution into

$$U(x) = \hat{U}(x) + R(x), \quad \text{with} \quad \hat{U}(a) = \hat{U}(b) = 0 \quad \text{and} \quad R(x) = \begin{cases} \alpha \frac{x_1 - x}{x_1 - a} & (a \le x \le x_1) \\ 0 & (x_1 \le x \le x_M) \\ \beta \frac{x - x_M}{b - x_M} & (x_M \le x \le b) \end{cases}$$
 (27)

Now again insert 28 into 22, multiply by a trial function V(x) that vanishes at a and b, integrate from a to b and drop a boundary term. This leads to the weak formulation for the approximate solution:

Find
$$\hat{U}(x)$$
 such that
$$\int_a^b dx \, \hat{U}'(x) V'(x) = \int_a^b dx \, f(x) V(x) - \int_a^b dx \, R'(x) V'(x) \quad \text{for all } V(x). \tag{28}$$

6.4 Numerical solution

To obtain a matrix equation for approximate solution U(x), the next step is to expand

$$\hat{U}(x) = \sum_{i=0}^{M+1} \hat{U}_i \varphi_i(x)$$
 and $V(x) = \sum_{i=0}^{M+1} V_i \varphi_i(x)$ (29)

in a basis for the approximate solution function space, namely the piecewise linear functions on the grid 26. The most natural basis for this space are the functions

$$\varphi_{i}(x) = \begin{cases} (x - x_{i-1})/(x_{i} - x_{i-1}) & \text{if } x_{i-1} \leq x \leq x_{i} \\ (x_{i+1} - x)/(x_{i+1} - x_{i}) & \text{if } x_{i} \leq x \leq x_{i+1} = \\ 0 & \text{otherwise} \end{cases}$$

$$(30)$$

With this basis, the coefficients $\hat{U}_i = \hat{U}(x_i)$ are simply the values at the grid points, making it straightforward to plot the solution. Inserting this expansion into 28 gives

$$\sum_{i,j} \hat{U}_i V_j \int_a^b \mathrm{d}x \, \varphi_i'(x) \varphi_j'(x) = \sum_j V_j \int_a^b \mathrm{d}x \, \varphi_j(x) f(x)$$
$$-a \sum_j V_j \int_a^b \mathrm{d}x \, \varphi_0'(x) \varphi_j'(x) - b \sum_j V_j \int_a^b \mathrm{d}x \, \varphi_{M+1}'(x) \varphi_j'(x).$$

We can write this as the neat matrix equation $V^T A \hat{U} = V^T F$ by introducing

$$\hat{U} = [\hat{U}_1, \dots, \hat{U}_M]^T, \quad V = [V_1, \dots, V_M]^T, \quad A_{ij} = \int_a^b \mathrm{d}x \, \varphi_i'(x) \varphi_j'(x) \quad \text{and} \quad F_j = \int_a^b \mathrm{d}x \, \varphi_j(x) f(x).$$

for $0 \le i, j \le M+1$. Since this must hold for any V(x) and thus V, we must have

$$A\hat{U} = F. \tag{31}$$

This is the matrix equation we will solve to find U(x). After finding \hat{U} , we simply sum 29 and add R(x) to find U(x). Note that our particular choice of basis 30 gives the convenient property $U(x_i) = U_i$, so summing is not necessary in practice.

To solve 31, we must first calculate the so-called **stiffness matrix** A and the **load vector** F. The former involves only the known basis functions and gives nonzero entries

$$A_{00} = \frac{1}{x_1 - x_0} \qquad A_{M+1M+1} = \frac{1}{x_{M+1} - x_M}$$

$$A_{ii} = \frac{1}{x_i - x_{i-1}} + \frac{1}{x_{i+1} - x_i} \qquad A_{ii+1} = A_{i+1i} = \frac{1}{x_{i+1} - x_i}.$$

The latter involves integrals over an arbitrary source function f(x) times the basis functions $\varphi_j(x)$. This integral must be approximated numerically and should be split from x_{i-1} to x_i and x_i to x_{i+1} to properly handle the spike in $\varphi_j(x)$ at x_j . (TODO: comment on numerical integration procedure, Gaussian integration)

We now impose $\hat{U}(a) = \hat{U}(b) = 0$ by removing the first and last entries in the matrix equation after calculating the entire $(M+2) \times (M+2)$ system described above. This gives an $M \times M$ equation. Then we construct U by appending α and β at the beginning and end of the M-vector \hat{U} .

6.4.1 Uniform refinement

We test our method on four problems, shown in figure 16, with uniform elements $x_i - x_{i-1} = (b-a)/(M+1)$.

The approximate solutions resembles the exact solution with few points. For the symmetric Gaussian problems, it is vital to choose an odd number of grid points to capture the spike in the center. In the final problem, the source function diverges at the left boundary, but the numerical integration is still able to find a good numerical solution.

In all but the first problem, errors distribute non-evenly across the elements. Computational resources are wasted by using many points in areas where the solution varies slowly. These resources would be better spent by increasing the grid resolution in the areas where the error is large. This is the motivation for turning to adaptive refinement and non-uniform grids.

6.4.2 Adaptive refinement

Motivated by the uneven error distribution from using uniform elements, we will now do adaptive refinement, similarly to what we did in section 1.3. We start with a uniform grid and successively split those elements on which the error is largest. Contrary to what we did in section 1.3, we will not split only *one* element between each iteration of the numerical solution, but split *all* elements on which the error is greater than some reference error. This leaves us with less control over the number of elements, but in return we will see that the error strictly decreases in each iteration, eliminating the oscillating error in figure 3.

This time, we use two strategies that both involve the exact error:

1. Average error strategy: Split the interval $[x_m, x_{m+1}]$ with error

$$||u(x) - U(x)||_2 > 0.99 \frac{||u(x) - U(x)||_2}{N},$$

where N is the number of intervals. The safety factor $0.99 \approx 1$ ensures that intervals are split also when all errors are equal (up to machine precision), so the procedure does not halt unexpectedly.

2. Maximum error strategy: Split the interval $[x_m, x_{m+1}]$ with error

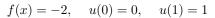
$$||u(x) - U(x)||_2 > 0.70 \max ||u(x) - U(x)||_2$$

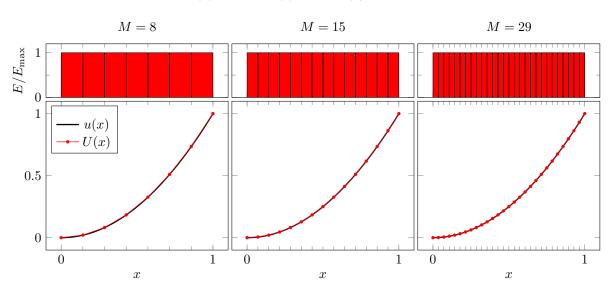
where N is the number of intervals.

In figure 16, we show how the errors distribute on the same four problems as in figure 16 using the average error strategy.

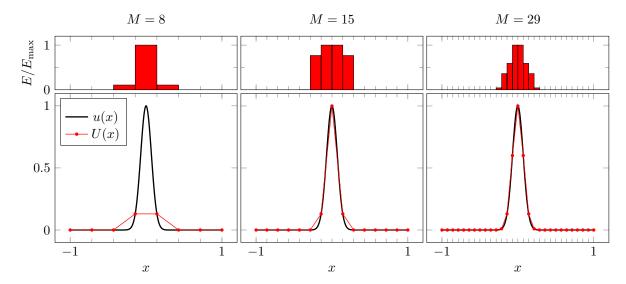
Observe how only elements with large error are refined, while others are left untouched. In the symmetric Gaussian problems, the refinement ensure that the middle element is split immediately if we do not start with a grid point at the peak. In the final problem, we see that it is almost only elements close to the left boundary where the source diverges that needs to be refined.

As discussed above, we see that the errors in the first problem distribute evenly on the initial uniform grid. This shows the importance of the safety factor 0.99 in the average error strategy. Without it, precision issues would make some elements skip the refinement criterion.





$$f(x) = -(40000x^2 - 200) \exp(-100x^2), \quad u(-1) = e^{-100}, \quad u(1) = e^{-100}$$



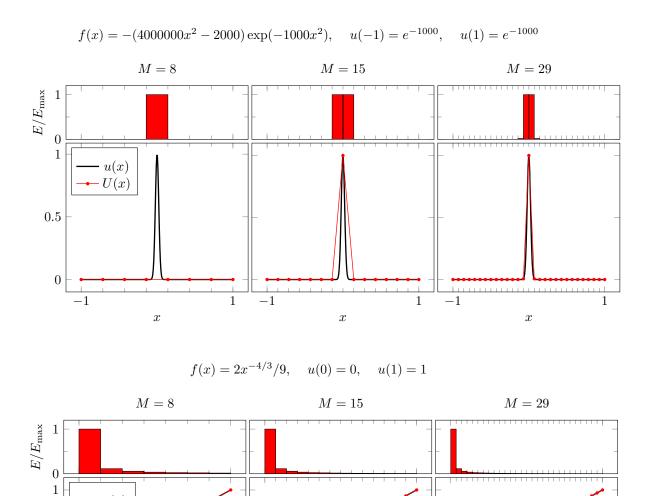


Figure 16: Uniform mesh refinement on four problems with the average error strategy, showing the evolution of an initial mesh whose elements are split in half over three iterations, along with the distribution of L_2 -errors across the elements.

u(x) U(x)

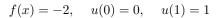
0.5

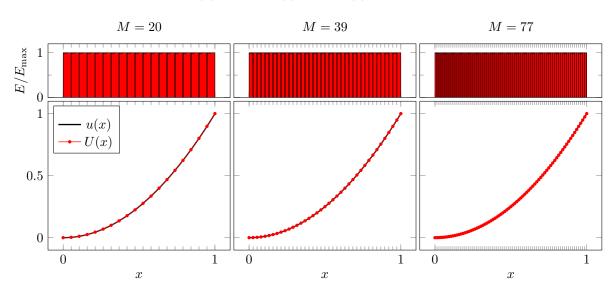
6.4.3 Comparison of convergence

Finally, in figure 17, we compare the convergence of uniform and adaptive refinement strategies.

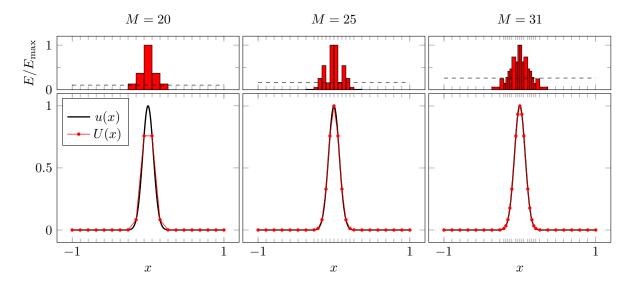
In all problems, adaptive refinement yields greater or equal accuracy for a given number of elements compared to uniform refinement. This is in contrast to what was the case for the finite difference method in figure 3, where adaptive refinement gave errors only comparable and usually larger than those from uniform refinement. It is only in the first problem that all strategies behave identically, as the errors here distribute evenly across the elements.

By splitting multiple intervals between each iteration of the numerical solution, we have eliminated the oscillating error pattern in figure 3. Now the error strictly decreases between each refinement of the grid. This suggests that the oscillating pattern is due to refinements where intervals with large error are present even after refining the element with greatest error. For example, it would be a bad idea to refine only *one* element in the first problem in figure 16, where errors are even across the elements.





$$f(x) = -(40000x^2 - 200) \exp(-100x^2), \quad u(-1) = e^{-100}, \quad u(1) = e^{-100}$$



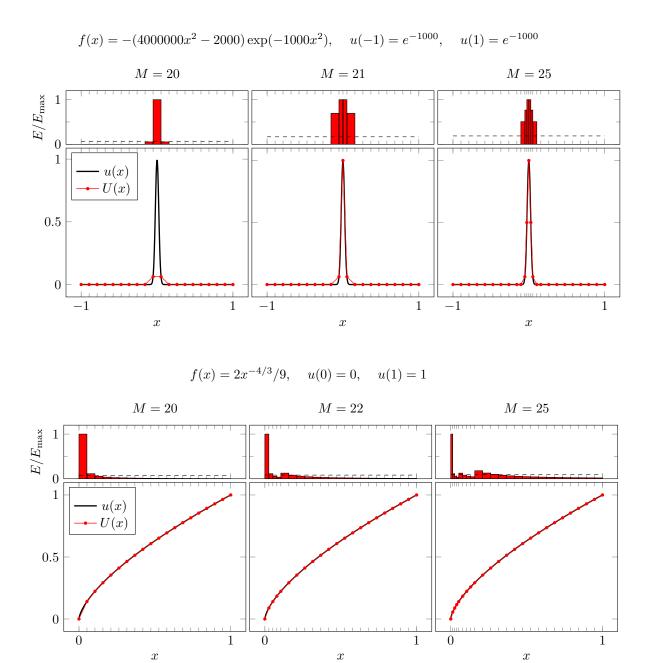


Figure 16: Adaptive mesh refinement on four problems with the average strategy, showing the evolution of an initial uniform mesh as it is refined over three iterations. Elements whose L_2 -error lie above the reference error - - - are split in half.

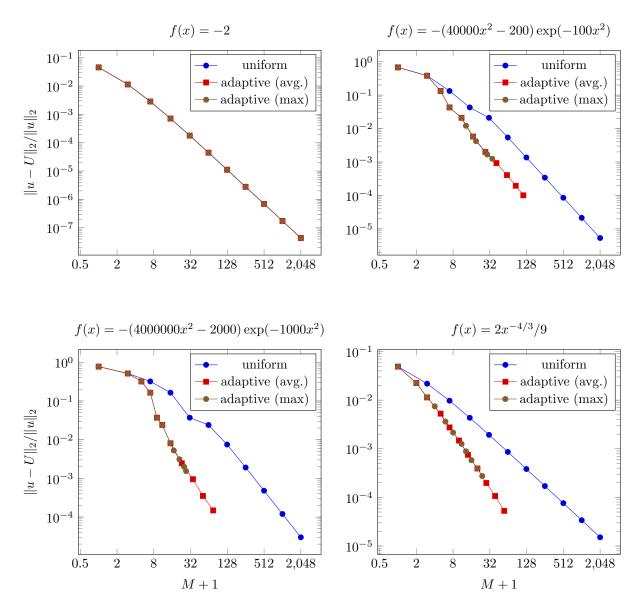


Figure 17: Convergence plots for four problems that compare uniform mesh refinement with the two adaptive mesh refinement strategies. (TODO: show order?)

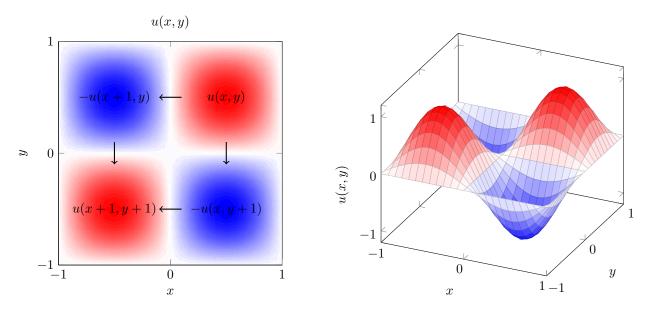


Figure 18: The function u(x,y) is originally defined on $[0,1] \times [0,1]$, but we extend the definition to the full xy-plane with the rules u(x+1,y) = u(x,y+1) = -u(x,y). This makes u(x,y) periodic and permits Fourier analysis. Here is the continuation on $[-1,1] \times [-1,1]$.

7 Biharmonic equation

Consider the inhomogeneous Biharmonic equation with clamped boundary conditions on the unit square $\Omega = [0, 1]^2$:

$$\nabla^4 u = f \quad , (x, y) \in \Omega, \tag{32a}$$

$$u = 0, \nabla^2 u = 0 \quad , (x, y) \in \partial\Omega.$$
 (32b)

7.1 Analytical solution

To use Fourier analysis, let us extend the definition of u(x,y) on $[0,1] \times [0,1]$ to the full xy-plane $[-\infty, +\infty] \times [-\infty, +\infty]$ as the antisymmetric continuation with the rules u(x,y+1) = u(x+1,y) = -u(x,y). The procedure is illustrated in figure 18. Using the antisymmetry, the conditions u = -u = 0 and $\nabla^2 u = -\nabla^2 u = 0$ are automatically satisfied at the boundaries. This can also be seen for the simple trial function in figure 18, which vanishes and inflects at the boundaries. Now u(x,y) = u(x+2,y) = u(x,y+2) is periodic in both directions with period 2 and can therefore be written as a Fourier series

$$u(x,y) = \sum_{m=-\infty}^{+\infty} \sum_{n=-\infty}^{+\infty} \hat{u}_{mn} e^{i2\pi mx/2} e^{i2\pi ny/2}$$

with coefficients (TODO: note on orthogonality or take this as known?)

$$\hat{u}_{mn} = \frac{1}{4} \int_{-1}^{+1} dx \int_{-1}^{+1} dy \, u(x, y) e^{-i2\pi mx/2} e^{-i2\pi ny/2}.$$

By the antisymmetry u(x+1,y) = u(x,y+1) = -u(x,y) and the symmetry u(x+1,y+1) = u(x,y), the Fourier coefficients satisfy

$$u_{m,n} = -u_{-m,n} = -u_{m,-n} = u_{-m,-n},$$

so $\hat{u}_{00} = -\hat{u}_{00} = 0$ and we can write the Fourier series as

$$u(x,y) = \sum_{m=1}^{+\infty} \sum_{n=1}^{+\infty} \hat{u}_{mn} \left(e^{+i\pi mx} e^{+i\pi ny} - e^{-i\pi mx} e^{+i\pi ny} - e^{+i\pi mx} e^{-i\pi ny} + e^{-i\pi mx} e^{-i\pi ny} \right)$$
$$= -4 \sum_{m=1}^{+\infty} \sum_{n=1}^{+\infty} \hat{u}_{mn} \sin(m\pi x) \sin(n\pi y)$$

after simplifying all complex expontentials using Euler's identity $e^{ix} = \cos x + i \sin x$. Rescaling $\hat{u}_{mn} \rightarrow -\hat{u}_{mn}/4$, we then begin by expressing our analytical solution as the double sine series

$$u(x,y) = \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} \hat{u}_{mn} \sin(m\pi x) \sin(n\pi y).$$
 (33)

Plug this Fourier series into equation (32) and act with the biharmonic operator to get

$$\nabla^4 u(x,y) = \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} \left((m\pi)^2 + (n\pi)^2 \right)^2 \hat{u}_{mn} \sin(m\pi x) \sin(n\pi y) = f(x,y).$$

For simplicity, we restrict ourselves to sources that can also be written (TODO: what restrictions does this impose on f?)

$$f(x,y) = \sum_{m=1}^{\infty} \sum_{n=1}^{\infty} \hat{f}_{mn} \sin(m\pi x) \sin(n\pi y). \tag{34}$$

The Fourier series for $\nabla^4 u(x,y)$ and f(x,y) can be equal only if their coefficients are equal. This can be seen formally by multiplying both by $\sin(2m'\pi x)\sin(2n'\pi y)$, integrating over x and y and using orthogonality of the sine functions,

$$\int_0^1 \mathrm{d}x \sin(m\pi x) \sin(m'\pi x) = \frac{1}{2} \delta_{mm'}.$$

Therefore, the coefficients of the solution are

$$\hat{u}_{mn} = \frac{\hat{f}_{mn}}{\left((m\pi)^2 + (n\pi)^2 \right)^2},\tag{35}$$

and the solution u(x,y) is available by summing its Fourier series 33.

If we know \hat{f}_{mn} , it is straightforward to compute \hat{u}_{mn} and thus the solution u(x,y) itself from its Fourier series. If we only know f(x,y), we can find the coefficients by using the orthogonality of the sine functions again. Multiply the Fourier series by $\sin(m'\pi x)\sin(n'\pi y)$ and integrate over x and y to get

$$\int_0^1 dx \int_0^1 dy f(x, y) \sin(m'\pi x) \sin(n'\pi y)$$

$$= \sum_{m=1}^\infty \sum_{n=1}^\infty \hat{f}_{mn} \underbrace{\int_0^1 dx \sin(m\pi x) \sin(m'\pi x)}_{\delta_{mm'}/2} \underbrace{\int_0^1 dy \sin(\pi y) \sin(n'\pi y)}_{\delta_{nn'}/2}$$

$$= \hat{f}_{m'n'}/4.$$

Read from bottom to top,

$$\hat{f}_{mn} = 4 \int_0^1 dx \int_0^1 dy \, f(x, y) \sin(m\pi x) \sin(n\pi y).$$
 (36)

Note that a general source f(x,y) may only be represented exactly by an infinite Fourier series. To make the Fourier series solution viable, we must cut it off to include only a finite number of terms. In this case, we should analyze f(x,y) to make sure that we exclude only Fourier modes that contribute insignificantly to the solution. However, we can also construct problems with a finite number of terms in the *exact* solution by simply defining the source f(x,y) in terms of a finite number of nonzero Fourier coefficients.

7.2 Numerical solution method

We may transform (32) into a system of Poisson equation by introducing $g = \nabla^2 u$,

$$\nabla^2 g = f, (37)$$

$$\nabla^2 u = q. \tag{38}$$

$$g = u = 0, \quad \text{on } \partial\Omega.$$
 (39)

In solving the poisson equation $\nabla^2 u = f$ will consider here two discretization schemes, the five point stencil and nine point stencil, which we will denote ∇_5^2 and ∇_9^2 .

Written in stencil diagrams, the five point stencil is given as

$$u\begin{pmatrix} 1 \\ 1 \\ 1 \end{pmatrix} = h^2 f$$

while the nine point stencil is given by

$$u\left(\begin{array}{c|c} \frac{1}{6} & \frac{2}{3} \\ \frac{1}{6} & \frac{1}{6} \\ \frac{1}{6} & \frac{2}{3} \end{array}\right) = h^2 f\left(\begin{array}{c|c} \frac{1}{12} \\ \frac{2}{3} & \frac{1}{12} \\ \frac{1}{12} & \frac{2}{3} & \frac{1}{12} \\ \end{array}\right).$$

We will now describe the structure of the matricies involved in the problem, starting with the simpler five point stencil. We will use the same flattening for the 2D discrete function u as described in section 4.2. We recognize that in U neighbouring elements correspond to grid points that are left and right of each other, while the grid point up and down is the element N places before and after. Thus, writing out the five point stencil K2D, we get a toeplitz and symmetric matrix, where the main diagonal has -4 and the ± 1 and $\pm N$ off diagonals have 1. We can compactley write this as

$$K2D = \begin{bmatrix} K & 0 & & & \\ 0 & K & 0 & & \\ 0 & 0 & K & & \\ & & & \ddots \end{bmatrix} + \begin{bmatrix} -2I & I & 0 & & \\ I & -2I & I & 0 & \\ 0 & I & -2I & I & 0 \dots \end{bmatrix}.$$

where I is the identity matrix and K is the one dimensional central finite difference matrix of order 2,

$$K = \begin{bmatrix} -2 & 1 & & & \\ 1 & -2 & 1 & & \\ & 1 & -2 & 1 & \\ & & & \ddots & \\ & & 1 & -2 & 1 \\ & & & 1 & -2 \end{bmatrix}.$$

Using the Kronecker product, this may be written as

$$K2D = \mathtt{kron}(I, K) + \mathtt{kron}(K, I)$$

which is useful when working with sparse matricies.

We will split the nine point stencil matrix into two, according to the stencil diagram above. One matrix represents a five point stencil with weights altered to $\frac{-10}{3}$, $\frac{2}{3}$ instead of -4, 1, we denote this matrix by $K2D^{(9)}$. The other is the "X"-stencil taking care of the NE, NW, SE, and SW points, each with a weight of $\frac{1}{6}$, as shown in the diagram. We will denote this matrix by $\Sigma 2D$.

Let

$$\Sigma = \frac{1}{\sqrt{6}} \begin{bmatrix} 0 & 1 & & & \\ 1 & 0 & 1 & & & \\ & 1 & 0 & 1 & & \\ & & 1 & 0 & \ddots & \\ & & & \ddots & \ddots & \\ & & & \ddots & \ddots & \\ \end{bmatrix}$$
(40)

be the TST matrix with ones on its off-diagonals and zero on the diagonal. Then we have $\Sigma 2D = \Sigma \otimes \Sigma$. The nine point stencil is then

$$somename9stencil = K2D^{(9)} + \Sigma 2D$$
 (41)

The eigenvalues of the Kronecker product $A \otimes B$ is the product of the eigenvalues of A and B. The eigenvalue of the Kronecker sum $A \oplus B = A \otimes I + I \otimes B$ is the sum of eigenvalues of A and B. (TODO: these are non-obvious properties, need a reference, e.g. http://www.siam.org/books/textbooks/0T91sample.pdf theorem 13.12 and 13.16) In general the eigenvalues of a TST matrix are

$$\lambda_k = a + 2b\cos\left(\frac{k\pi}{N+1}\right)$$

where a is the main diagonal and b is the off diagonal. Thus, the eigenvalues of K2D are

$$\left(-2+2\cos\left(\frac{k\pi}{N+1}\right)\right)+\left(-2+2\cos\left(\frac{l\pi}{N+1}\right)\right).$$

The eigenvalues of $K2D^{(9)}$ are

$$\frac{1}{3}\left(-10+4\cos\left(\frac{k\pi}{N+1}\right)\right)+\frac{1}{3}\left(-10+4\cos\left(\frac{l\pi}{N+1}\right)\right)$$

and the eigenvalues of $\Sigma \otimes \Sigma$ are

$$\frac{4}{6}\cos\left(\frac{k\pi}{N+1}\right)\cos\left(\frac{l\pi}{N+1}\right).$$

The eigenvalues of the nine point stencil are thus

$$-\frac{10}{3} + \frac{4}{3} \left(\cos(\frac{k\pi}{N+1}) + \cos(\frac{l\pi}{N+1}) \right) + \frac{4}{6} \cos(\frac{k\pi}{N+1}) \cos(\frac{l\pi}{N+1}). \tag{42}$$

7.3 Stability and order of the five and nine point stencils

Definition 1. The *n*-point stencil $\nabla_n^2 f(x_0) = \sum_{i=0}^{n-1} a_i f(x_i)$, where x_i are the neighbouring points of x_0 .

Definition 2. We denote by a proper stencil a stencil where $a_i > 0$ for i > 0 and $\sum_{i=1}^n a_i = -a_0$.

Lemma 1. If $\nabla_n^2 f \geq 0$ on Ω , and ∇_n^2 is a proper stencil, the maximum value of f is attained on $\partial \Omega$, that is

$$\max(f)_{\Omega} \leq \max(f)_{\partial\Omega}$$
.

Proof. Suppose the opposite is true,

$$\max(f)_{\Omega} > \max(f)_{\partial\Omega}.$$

Then, there is an internal grid point x_0 on which f attains it maximal value. Then

$$-a_0 f(x_0) = \sum_{i=1}^{n-1} a_i f(x_i) - \nabla_n^2 f(x_0) \le \sum_{i=1}^{n-1} a_i f(x_i) \le \sum_{i=1}^{n-1} a_i f(x_0) = -a_0 f(x_0).$$
 (43)

As the RHS is equal to the LHS, the equality must hold throughout. Since $f(x_i) \leq f(x_0)$ all points of the stencil must be equal, $f(x_1) = f(x_2) = \cdots = f(x_n) = f(x_0)$. Applying this same argument to each of the neighbours, and then to their neighbours and so on, we ultimately reach the conclusion that the same value is also attained on $\partial\Omega$, and we thus have a contradiction.

Lemma 2. If $\exists \phi_n > 0$ such that the proper stencil $\nabla_n^2 \phi_n = 1$ on Ω , and v is a descrete function that equals zero on $\partial \Omega$, then

$$||v||_{\infty} \le \max_{\partial \Omega} (|\phi_n|) ||\nabla_n^2 v||_{\infty}. \tag{44}$$

Proof. Let $\|\nabla_n^2 v\|_{\infty} = M$. Then

$$\nabla_n^2(v + \phi_n M) = \nabla_n^2 v + M \ge 0.$$

 ∇_n^2 is a proper stencil, so by lemma 1 $\nabla_n^2(v+\phi_n M)$ attains its maximum value on $\partial\Omega$. Thus

$$||v||_{\infty} \le ||v + \phi_n M||_{\infty} \le \max_{\partial \Omega} (|v + \phi_n M|) = \max_{\partial \Omega} (|\phi_n|) M = \max_{\partial \Omega} (|\phi_n|) ||\nabla_n^2 v||_{\infty}.$$

Remark. For the five and nine point stencil, such functions exist.

$$\phi_5(x,y) = \frac{1}{4} \left(\left(x - \frac{1}{2} \right)^2 + \left(y - \frac{1}{2} \right)^2 \right) \tag{45}$$

has the property $\nabla_5^2 \phi_5(x, y) = 1$, takes the values $0 \le \phi_5(x, y) \le 1/8$ on Ω and attains the maximum on $\partial\Omega$. Similarly, the function

$$\phi_9(x,y) = \frac{1}{5} \left(\left(x - \frac{1}{2} \right)^4 + \left(y - \frac{1}{2} \right)^4 \right) - \frac{6}{5} \left(x - \frac{1}{2} \right)^2 \left(y - \frac{1}{2} \right)^2 + \frac{1}{4} \left(\left(x - \frac{1}{2} \right)^2 + \left(y - \frac{1}{2} \right)^2 \right)$$
(46)

is such that $\nabla_9^2 \phi_9(x,y) = 1$, takes the values $0 \le \phi_9(x,y) \le 3/40$ on Ω and attains the maximum on $\partial\Omega$. Both are shown in figure 19.

Theorem 1 (Five point stencil stability). If $\nabla^2 u = f$ and $\nabla_5^2 v = f$, then

$$||u - v||_{\infty} \le Ch^4 |D^4 v|_{\infty}.$$

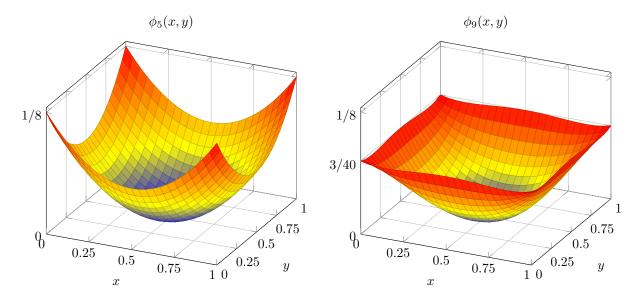


Figure 19: The functions $\phi_5(x, y)$ and $\phi_9(x, y)$ range from $\phi_5(\frac{1}{2}, \frac{1}{2}) = \phi_9(\frac{1}{2}, \frac{1}{2}) = 0$ at the center to $\phi_5(0, 0) = 1/8$ and $\phi_9(0, 0) = 3/40$ at the boundaries of $[0, 1] \times [0, 1]$.

Proof. By lemma 2

$$||u-v||_{\infty} \le \frac{1}{8} ||\nabla_5^2 (u-v)||_{\infty}.$$

By Taylor expanding u(x+h,y) and u(x-h,y) we get

$$\frac{1}{h^2}\delta_x^2 u = \partial_x^2 u + \frac{h^2}{12}\partial_x^4 + \mathcal{O}(h^4),$$

and similarly for y. Therefore we get, to order h^2

$$\begin{split} \nabla_5^2 u &= \frac{1}{h^2} (\delta_x^2 + \delta_y^2) u \\ &= (\partial_x^2 + \partial_y^2 + \frac{h^2}{12} \partial_x^4 + \frac{h^2}{12} \partial_y^4) u \\ &\leq \nabla^2 u + \frac{h^2}{12} \max(\partial_x^4 u, \partial_y^4 u) \\ &\leq \nabla^2 u + C h^2 |D^4 u|_{\infty}. \end{split}$$

Here $|D^n u|_{\infty}$ is to be understood as the maximal value of all *n*-index derivatives of u. For example for D^2 this would be u_{xx}, u_{xy}, u_{yx} , and u_{yy} .

Theorem 2 (Nine point stencil stability). TODO verify it should be Ff not f. If $\nabla^2 u = f$ and $\nabla_9^2 v = Ff = f + \frac{h^2}{12} \nabla_5^2 f$, then

$$||u - v||_{\infty} \le Ch^2 |D^6 v|_{\infty}.$$

Proof. By lemma 2

$$||u-v||_{\infty} \le \frac{3}{40} ||\nabla_9^2(u-v)||_{\infty}.$$

Taylor expanding $\nabla_0^2 u$, similarly to what was done in the five point stencil case, we get

$$\nabla_9^2 u = (\nabla^2 + \frac{h^2}{12} \partial_x^4 + \frac{h^2}{12} \partial_y^4 + \frac{h^2}{6} \partial_x^2 \partial_y^2) u + h^4 C \mathcal{O}(D^6 u)$$
$$= \nabla^2 u + \frac{h^2}{12} \nabla^2 f + C h^4 \mathcal{O}(D^6 u).$$

Here $\mathcal{O}(D^n u)$ is to be understood as terms involving n-index derivatives of u. Notice that

$$\nabla^2 f = \nabla_5^2 f + Ch^2 \mathcal{O}(D^4 f),$$

and that $\mathcal{O}(D^4 f) = \mathcal{O}(D^4 \nabla^2 u) = \mathcal{O}(D^6 u)$.

Thus

$$\|\nabla_9^2(u-v)\|_{\infty} = \|Ch^4\mathcal{O}(D^6u)\|_{\infty} \tag{47}$$

$$\leq Ch^4 \|D^6 u\|_{\infty} \tag{48}$$

7.4 The Fast Poisson Solver

We are to solve the equation

$$AU = F$$
.

Assuming that the eigenvectors of A are a complete set, we may write

$$F = a_1y_1 + a_2y_2 + ...,$$

where $y_1, y_2, ...$ are the eigenvectors of A. (TODO: U is also expanded in the eigenvectors?) The solution U is the of course

$$U = \frac{a_1}{\lambda_1} y_1 + \frac{a_2}{\lambda_2} y_2 + \dots,$$

where λ_i is the eigenvalue corresponding to y_i . In general, however, this is not a viable way to solve the problem, as it requires knowing all the eigenvalues and eigenvectors, as well as finding the coefficients a_i . However, for the set of eigenvectors in this problem, which are sines, we have a very efficient algorithm for computing the coefficients, the discrete fast sine transform. We also have simple analytical expressions for the eigenvalues.

Written as matrix expressions

$$AU = F \tag{49}$$

$$S\Lambda SU = F \tag{50}$$

$$\Rightarrow$$
 (51)

$$U = S\Lambda^{-1}SF, (52)$$

where we used the fact that $S^{-1} = S$. Moreover, formulated more directly with regards to implementing the solver, we have that the solution is

$$U = IFST(FST(F)/\Lambda),$$

where IFST, FST are the inverse and normal fast sine tranform.

Demonstration of order of five and nine point stencil

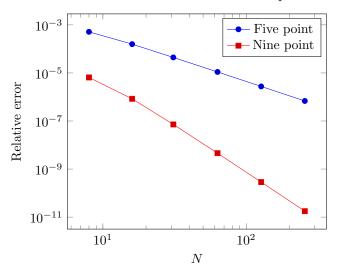


Figure 20: Error as a function discretization points N TODO: figure is to be updated with prettier figure TODO: Most appropriate with function of h or N?

7.5 Demonstration of order

7.6 Solving the Biharmonic equation

Consider the equation

$$\nabla^4 u = f(x, y) \qquad (x, y) \in \Omega$$
$$u = \nabla^2 u = 0 \qquad (x, y) \in \partial\Omega$$

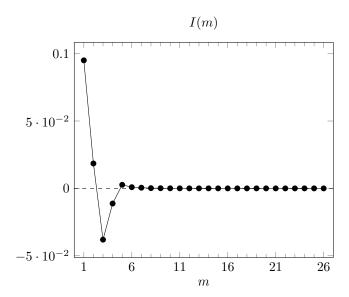
where $f(x,y) = (\sin \pi x \sin \pi y)^4 e^{-(x-0.5)^2 - (y-0.5)^2}$.

Then

$$\hat{f}_{mn} = \left[2 \int_0^1 dx \sin^4(\pi x) \sin(m\pi x) e^{-(x-1/2)^2} \right] \left[2 \int_0^1 dy \sin^4(\pi y) \sin(n\pi y) e^{-(y-1/2)^2} \right].$$

The integrals do have analytical solutions, but they are not easy to compute. We have computed them with the SAGE CAS suite. Below is an easy-to-copy expression for the first bracket = I(m) in the equation above. Then $\hat{f}_{mn} = I(m) I(n)$, and the analytical solution can be computed (but it has infinitely many terms)

$$\begin{split} & \text{I}(\text{m}) = 2 * \text{integral} \big(\sin \big(\text{pi} * \text{x} \big) \, ^{4} * \sin \big(\text{m} * \text{pi} * \text{x} \big) * \exp \big(- \big(\text{x} - 1/2 \big) \, ^{2} \big) \,, \ \text{x}, \ 0, \ 1 \big) \\ >>> & \text{I}(\text{m}) = 48 * \big(\text{pi} \, ^{9} * \text{m} \, ^{5} * \text{e} \, ^{2} \big) - 20 * \big(\text{pi} \, ^{9} * \text{e} \, ^{2} \big) + 2 * \text{pi} \, ^{7} * \text{e} \, ^{2} \big) * \text{m} \, ^{3} - \big(\text{pi} \, ^{9} * \text{m} \, ^{5} - 20 * \big(\text{pi} \, ^{9} + 2 * \text{pi} \, ^{7} \big) * \text{m} \, ^{3} + 16 * \big(4 * \text{pi} \, ^{9} + 15 * \text{pi} \, ^{7} \big) + 5 * \text{pi} \, ^{5} \big) * \text{m} \big) * \big(-1 \big) \, ^{\text{m}} + 16 * \big(4 * \text{pi} \, ^{9} * \text{e} \, ^{2} \big) + 15 * \text{pi} \, ^{7} \times \text{e} \, ^{2} \big) * \text{m} \big) / \big(\text{pi} \, ^{1} 0 * \text{m} \, ^{1} 0 * \text{e} \, ^{3} \big) - 20 * \big(2 * \text{pi} \, ^{1} 10 * \text{e} \, ^{3} \big) - 20 * \big(2 * \text{pi} \, ^{1} 10 * \text{e} \, ^{3} \big) - 20 * \big(2 * \text{pi} \, ^{1} 10 * \text{e} \, ^{3} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) * \text{m} \, ^{6} \big) + 40960 * \big(2 * \text{pi} \, ^{3} \big) * \text{m} \, ^{2} \big) + 10 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) * \text{m} \, ^{4} \big) + 33792 * \big(2 * \text{pi} \, ^{3} \big) * \text{m} \, ^{2} \big) + 256 * \big(16 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi} \, ^{3} \big) \times \text{m} \, ^{2} \big) + 20 * \big(2 * \text{pi}$$



References

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