

MEASURING REGENER-PFOTZER MAXIMUM USING DIFFERENT TYPES OF IONIZING RADIATION DETECTORS AND A NEW TELEMETRY SYSTEM

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Regener-Pfotzer maximum is located at the altitude where the intensity of cosmic radiation is greatest, usually around 20 km. However, its exact location depends on many parameters such as atmospheric conditions, geographical locations, solar activity and also the type of detected particles. During experimental flights of stratospheric balloons we use different types of ionizing radiation detectors: Geiger-Muller tubes, SPACEDOS silicon detectors and AIRDOS-C scintillation detectors with NaI(Tl) crystal.

Due to necessary data processing of data measured by instruments, the balloon gondola radiation measurements carry a number of additional supplementary sensors measuring humidity, temperature, location and orientation, altitude, atmospheric pressure, acceleration, etc. All of these sensors have to be monitored during the launch of the stratospheric balloon in order to verify their correct function and furthermore, it is necessary to record their data during the flight for future processing. This was the reason why a new universal system TF-ATMON was gradually developed during the flights of FIK-5 to FIK-6 that is able to solve the above mentioned problems. The system is based on using already existing tools of the PX4 open-source project that makes it possible, apart from data recording and monitoring, to solve other related issues - the possibility to trace the balloon gondola after the flight. For this reason, the originally used telemetry system was made significantly more robust and an IoT LoRa transmitter was added to the system, making it possible to transmit the data necessary for tracing the gondola to the TheThingsNetwork. In this way, a high reliability of finding the gondola and recording the data is ensured. In case of active detectors, it is also necessary that the flight trajectory is recorded synchronously with the supplementary quantities. The proposed TF-ATMON system is very variable and thus it can be used in other applications than just balloon measurements. However, its application will be demonstrated on stratospheric balloon flights measuring parameters of Regener-Pfotzer maximum.

INTRODUCTION

Particles of primary cosmic radiation with sufficient energy interact in the upper part of the atmosphere and generate showers of secondary cosmic radiation. With increasing depth of the atmosphere the intensity of primary radiation decreases whereas the secondary component increases. At an altitude of about 20 km the intensity of secondary cosmic radiation reaches its maximum, called the Pfotzer maximum [10], [1]. The Pfotzer maximum varies with geomagnetic vertical cutoff rigidity and with solar cycle and it is generally located at 15–27 km [3]. In the past, several experiments were done with the aim to measure vertical profile of ionization in the atmosphere at various locations in the world, mainly using radiosondes consisting of Geiger tubes [3], [8], [9], [6], [7], or to characterize instruments' response used for space-based missions [2] [4] [5]. To investigate cosmic radiation at high altitudes (around and above Regener-Pfotzer maximum region) and to test new detectors for cosmic radiation measurements, stratospheric balloons are very useful. However, the radiation measurement instruments need to be supplemented also by other sensors measuring temperature, pressure, humidity, altitude, acceleration, etc. All these sensors should be continuously monitored

during the launch and the flight of the balloon to verify their proper function and their values have to be recorded for further processing of all obtained data. Platform developed for experiments at different altitudes consisting of a balloon tracking system and different measuring instruments will be presented. The universal system TF-ATMON, based on the use of existing tools of the open-source project PX4 and supplemented with an IoT LoRa transmitter, enables it to monitor and record data during the flight and to trace the balloon. The application is demonstrated on stratospheric balloon flight FIK-6. Various radiation detectors - Geiger-Mueller tubes, Si-diode based detector SPACEDOS, and scintillation detectors AIRDOS-C with inorganic scintillation crystals - were used to measure the vertical profile of cosmic radiation in the atmosphere.

MATERIALS AND METHODS

During the first measurements, the construction of balloon gondola was always designed for the use of specific detectors. Balloon avionics was therefore built around the chosen detectors. This concept led to a situation when every new flight meant a significant

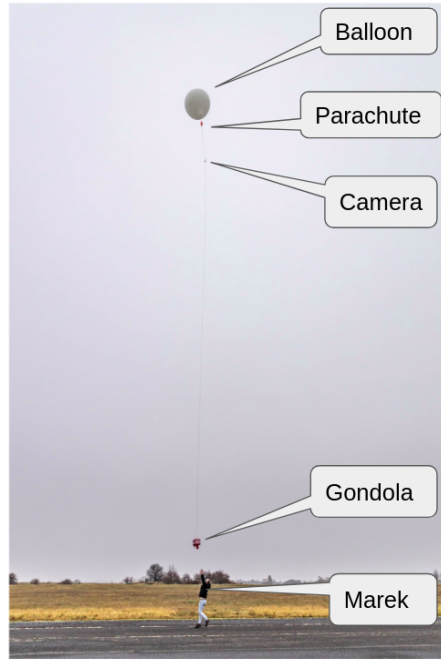


Figure 1. FIK-6 experiment setup

amount of work despite the fact that a number of components was recycled every year and used the next one. The reason was the need to adapt the avionics to the innovated version of detectors.

Due to a relatively high value of payload it was necessary to ensure the return of the gondola every flight. The main construction criteria were as follows:

- Reliable transmission of information about the geographical location of the gondola
- Good resistance to impact
- Ensuring the function in temperatures far below zero

Due to the fact that since the first flight there were various partial shortcomings and despite the generally successful nature of all flights and the fact that the gondola was always found, we tried different technologies to eliminate the complications, which have emerged. During the last experimental flights these included the way the data is recorded. An overall overview of the success of the used technologies is summarized in the table 1.

Apart from technologies used in gondolas a number of supplementary tools have undergone intensive development. For example, in order to find the balloon it was necessary to have an accurate real-time map of its position together with a prediction of the subsequent flight and the site of impact. In the case of the last two flights, FIK-5 and FIK-6, the problem is solved by [15]. As can be seen from the table, during the

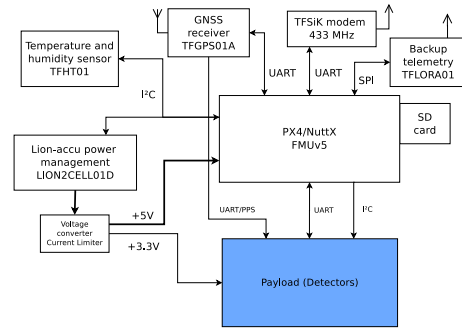


Figure 2. The schematic diagram of the new avionics

last flights the avionics was implemented using an UAV technology. It uses the Pixhawk autopilot with PX4 firmware. Telemetry is implemented by a very reliable combination of LoRa modem and SiK modem. Power is provided by Li-ion 18650 batteries that also have a reliable flight history.

Universal avionics

Based on our experience we have used a universal concept of avionics called TF-ATMON that makes it possible to connect different types of payloads and carry out various atmospheric measurements. Furthermore, it provides the detectors with basic services such as power supply, time and position information and provides a possibility to record data in a common log. Therefore it is extremely useful for testing high-altitude cosmic radiation detectors and dosimeters.

The schematic diagram of the new avionics is summarized in the figure 2.

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Table 1. Table summarising success of used technologies, with colours representing a degree of reliability:

	Failure	Partial failure	Perfect function			
Flight	FIK-1	FIK-2	FIK-3	FIK-4	FIK-5	FIK-6
Location	CZ (2015)	CZ (2017)	CZ (2018)	SE (2019)	CZ (2019)	CZ (2020)
Payload	Candy detector, Web camera	Candy detector, Web camera	AIRDOS, 360 deg camera	AIRDOS-C CRY19, SPACEDOS, G-M, Socrat-R	AIRDOS-C CRY19.SPACEDOS, G-M, 360 deg camera	AIRDOS-C NaI(Tl), SPACEDOS, G-M, Ionmeter, 360 deg camera
Landing site	vineyard Austria	rapeseed field	Poland	swamp (Finland)	forest	railway corridor
Power source	Li-ion 18650 accu	Li-ion 18650 and li-pol accu	Lithium primary cells and li-pol accu	Lithium primary cells	Lithium primary cells	Li-ion 18650 accu
Telemetry system	GSM	GSM, 868 MHz Proprietary Modem	SigFox, 868 MHz Proprietary Modem	Outsourced	LoRa, SigFox, SiK 433 MHz	2x LoRa, SiK 433 MHz
Rescue beacon	433 MHz CW	433 MHz CW	433 MHz CW	Outsourced	433 MHz CW	433 MHz CW
Flight control computer	Odroid-U2	Odroid-U2	Not used	Not used/outsourced	PX4, FMU v5	PX4, FMU v5

Transition to the concept, where the balloon specific parts of avionics are completely separate from the system of detectors, simplified the realization of next balloon flights. It reduced the complexity of connecting different types of detectors and at the same time it improved the integrity of supplementary data measurements. Overall, the new features can be summarized as follows:

- Easy implementation of different payloads
- Redundant telemetry links
- Gondola orientation and spatial position tracking and logging
- Reliable IMU sensor processing and calibration
- Possible use of relative high-power consumption payloads
- Pre-flight continuous charging as an option
- Power monitoring and maximum uptime calculation relevant to actual temperature
- Real-time pre-flight payload diagnostic

The documentation of used blocks could be found in the following sources TFGPS01 [16], TFSIK01 [17], TFHT01 [18], TFLORA01 [19], [20].

Further improvements

As can be seen from the table, the last unsolved problem with the balloon flights is the landing site. Therefore, in the future flights we are planning to use the autopilot

for a controlled descent as well. The descent will be carried out using an unmanned autogyro carrying a payload. The purpose of this solution is in using the autogyro's rotor instead of the original parachute as it has the advantage of good controllability. Thus it would be possible to choose the landing site and reduce the possible risk of creating dangerous situations. And at the same time it is able to control the descent rate.

Payload

In the case of FIK-5 and FIK-6 flights that served as the test flights for our novel approach the payload was not fully optimized for this use. All the detectors thus have their own SD card for data recording and some even have their own power supply.

Detectors

The payload of FIK-5 and FIK-6 flights contains three different types of ionizing radiation detectors. There were deployed SPACEDOS with silicon PIN diode sensor (namely SPACEDOS02A), AIRDOS-C with scintillation crystal and silicon photomultiplier and a G-M tube. The total payload mass was 2 kg.

SPACEDOS is a lightweight dosimeter intended for space applications for measurements on board spacecraft. The detector has been described in [12]. The detector is a silicon PIN diode with a volume $300 \mu\text{m} \times 10 \text{ mm} \times 20 \text{ mm} = 0.06 \text{ cm}^3$. The energy range of this detector is from 0.2 MeV to 9 MeV.

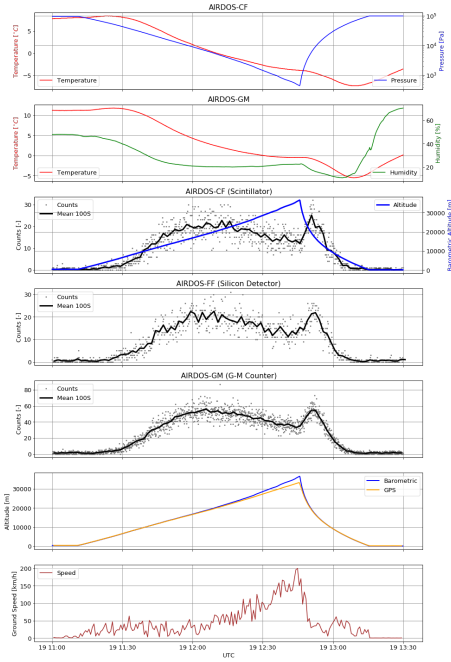


Figure 3. Raw data measured during the whole flight. From top to down Temperature near scintillation crystal, Air Pressure inside the box of crystal, Temperature inside the gondola, Relative Humidity inside the gondola, Radiation Events per 10 seconds counted by scintillator, Radiation Events per 10 seconds counted by silicon detector, Radiation Events per 10 seconds counted by G-M counter, Barometric altitude, altitude from GNSS, Horizontal Speed relative to ground.

AIRDOS-C is a scintillation detector with a small crystal. The detector has been described in [13]. This detector enhances NaI(Tl) crystal with a volume $\varnothing 10\text{mm} \times 20\text{mm} = 1.6 \text{ cm}^3$ and SiPM (Silicon Photomultiplier) with a sensitive surface $6 \text{ mm} \times 6 \text{ mm}$. The energy range of this detector is from 0.2 MeV to 18 MeV.

The G-M tube STS-5 was used in the Geiger–Müller counter. The volume of the tube is $\varnothing 10 \text{ mm} \times 76 \text{ mm} = 6 \text{ cm}^3$. This detector is capable of registering only the flux.

RESULTS

During 1 hour and 40 minutes of flight Dec 18th, 2020 FIK-6 recorded temperature, air pressure, humidity and radiation characteristics as histograms of deposited energy of radiation events. Recorded data are shown in Figure 3. The barometric altitude is calculated by the International Standard Atmosphere model 1976 [14].



Figure 4. Chasing cars with telemetry receivers at the landing site short after touchdown.

The graphs show that the maximum reached altitude was approximately 33km above sea level. The length of the flight trajectory in this case was approximately 100km. During the flight the balloon passed the Regener-Pfotzer maximum twice. The rescue team follows the balloon along the whole flight trajectory. The precision of tracking allows some participants of the rescue team to actually see the gondola touchdown visually. Therefore the gondola was successfully rescued within few minutes after touchdown.

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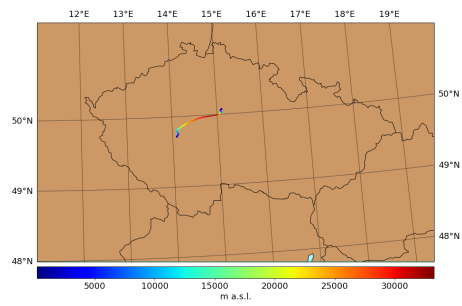


Figure 5. FIK-6 flight path

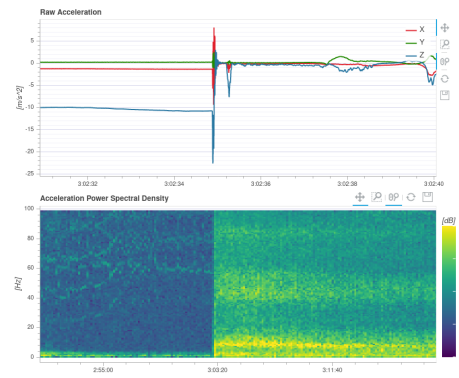


Figure 6. FIK-6 balloon burst

By processing the data a graph of detected altitude fig. X can be obtained. It shows that the measured altitude of the Regener-Pfotzer maximum was for all detector types around 19km above sea level.

DISCUSSION

Z naměřených dat je patrné, že během letu je velmi znatelný rozdíl mezi počtem bodů kvalitou měřených dat ve směru vzhůru a při klesání. Tento rozdíl je dán kromě odlišných teplotních podmínek hlavně rozdílnou hodnotou vertikální rychlosti. Tuto záležitost se v následujících balonových letech chystáme vyřešit kontrolovaným sestupem, kde by bylo možné rychlost klesání v některých fázích letu snížit. Tak aby byla více srovnatelná s rychlostí výstupu.

Zaroven je videt, že během klesání dochází ke značným vibracím, skokovým zmenám zrychlení a také dochází k rotaci gondoly. Všechny tyto parametry mohou mít vliv na měření atmosférických veličin a pro některé druhy přístroje musí být kompenzovány. Zaroven je videt při klesání narůst vlhkost, která může v určitých částech letu i namrznat na přístroje.

CONCLUSION

A telemetric system TF-ATMON has been developed. It enables data recording, pre-flight instruments control and their monitoring during flight. Thanks to the availability of different communication interfaces on the basic avionics, the use of various alternative detectors is simplified. The technology at the same time improves the possibilities of a quick location of a balloon gondola after its landing. It is therefore possible to carry out even experiments requiring a very short period till recovering the balloon after the flight. The system was successfully tested during FIK-6 flight.

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