# **HydDown**

User guide

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# 1 Introduction

HydDown is an open source python tool for calculation of Hydrogen (or other pure gas phase species) vessel/container depressurization and filling. The HydDown logo shown in Fig. 1.1 visualizes the key parameters and transport phenomena during gas vessel filling or discharging. The thermodynamic state inside the vessel changes over time as seen from immediately observable variables temperature (T) and pressure (P). This is caused by change in fluid inventory (density) due to flow of gas either in- or out of the vessel. Further, heat is transfered from or to the surroundings via convective heat transfer on the in- and outside of the vessel - with heat being conducted thorugh the vessel wall.

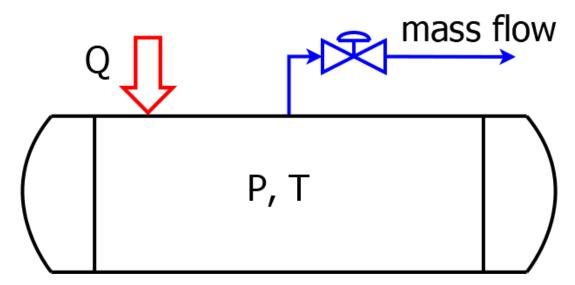


Figure 1.1: HydDown logo

Run the code as simple as:

python main.py input.yml

where main.py is the main script and input.yml is the input file in Yaml syntax.

### 1.1 Background

This is a small spare time project for calculation of vessel depressurisation behaviour. This is mainly to demonstrate, that although perceived as a very tedious/difficult task to write your own code for such an apparent complex problem, actually a fairly limited amount of code is necessary if you have a good thermodynamic backend.

A few choices is made to keep things simple to begin with:

- Coolprop is used as thermodynamic backend
- · Only pure substances are considered
- · Gas phase only
- No temperture stratification in the gas phase
- No temperture gradient through vessel wall
- Heat transfer is modelled as simple as possible

These choices makes the problem a lot more simple to solve, First of all the the pure substance Helmholtz energy based equation of state (HEOS) in coolprop offers a lot of convenience in terms of the property pairs/state variables that can be set independently. Using only a single gas phase species also means that component balances is redundant and 2 or 3-phase flash calculations are not required. That being said the principle used for a single component is more or less the same, even for multicomponent mixtures with potentially more than one phase.

### 1.2 Description

The following methods are implemented:

- Isothermal i.e. constant temperature of the fluid during depressurisation (for a very slow process with a large heat reservoir)
- Isenthalpic/Adiabatic (no heat transfer with surroundings, no work performed by the expanding fluid)
- Isentropic (no heat transfer with surroundings, PV work performed by the expanding fluid)
- Constant internal energy
- Energy balance. This is the most general case and includes both the ability to transfer heat with surroundings as well as accounting for PV work.

A simple (naive) explicit Euler scheme is implemented to integrate the mass balance over time, with the mass rate being calculated from an orifice/valve equation. For each step, the mass relief/ left in the vessel is known. Since the volume is fixed the mass density is directly given. For the calculation

methods (isentropic,isenthalpic,isenergetic etc), Coolprop allows specifying density and either H,S or U directly - this is very handy and normally only TP, PH, TS property pairs are implemented, and you would need to code a second loop to make it into am UV, VH or SV calculation. Coolprop is very convenient for this, however for a cubic EOS and for multicomponent Helmholtz energy EOS coolprop only supports a subset of state variables to be specified directly (T,P,quality). For this reason single component HEOS is the main target of this project.

### 1.3 Basic usage

The Yaml input file is edited to reflect the system of interest. For isothermal/isenthalpic/isentropic/isenergetic calculations the minimal input required are:

- Initial conditions (pressure, temperature)
- vessel dimensions (ID/length)
- valve parameters (Cd, diameter, backpressure)
- Calculation setup (time step, end time)
- Type of gas

If heat transfer is to be considered the calculation type "energybalance" is required. A few options are possible:

- Fixed U (U-value required, and ambient temperature)
- Fixed Q (Q to be applied to the fluid is requried)
- Specified h, the external heat transfer coefficient is provided and either the internal is provided or calculated from assumption of natural convection from a vertical cylinder at high Gr number. Ambient temperature is required.
- Detailed (not yet implemented)
- Fire (Not yet implemented)

### 1.4 Requirements

- Python (3.8 at least python3)
- Numpy
- matplotlib
- Coolprop (6.4.1)
- Yaml

The script is running on Windows 10 x64, with stock python installation from python.org and packages installed using pip. Should run om linux or in any conda environment as well, but I haven't checked.

# 2 Usage

- 2.1 Script
- 2.2 Module import
- 2.3 Input file

# 3 Theory

- 3.1 Thermodynamics
- 3.1.1 Isothermal process
- 3.1.2 Isentropic process
- 3.1.3 Isenthalpic process
- 3.1.4 General energy balance
- 3.2 Flow devices
- 3.2.1 Restriction Orifice
- 3.2.2 Relief valve
- 3.2.3 Control Valve
- 3.3 Heat transfer

#### 3.3.1 Natural convection

To determine the heat transfer for the gas-wall interface, Newton's law of cooling is applied, as given in equation eq. 3.1.

$$\frac{dQ}{dt} = -hA(T_s - T_{gas}) \tag{3.1}$$

- dQ is the change in thermal energy due to convective heat transfer. [J]
- dt is the change in time during the heat transfer. [s]
- h is the convective heat transfer. [W/m<sup>2</sup> · K]

- A is the area normal to the direction of the heat transfer. [m $^2$ ]
- $T_s$  is the surface temperature of the geometry. [K]
- $T_{qas}$  is the temperature of the surrounding gas. [K]

Equation [eq. ??] indicates the heat transfer to the surface of any geometry such as a plate or cylindrical wall, by means of heat convection from its surroundings (Çengel and Ghajar 2011)(Geankoplis 1993).

The convective heat transfer will need to be estimated for the the gas-wall interface, by the use of empirical relations for the Nusselt number. The Nusselt number describes the ratio of convective heat transfer to conductive heat transfer, normal to a surface area, as given in equation eq. 3.2.

$$Nu = \frac{hL}{k} \tag{3.2}$$

- Nu is the Nusselt number. [-]
- h is the convective heat transfer. [W/m<sup>2</sup>·K]
- L is a characteristic length of the geometry. [m]
- k is the thermal conductivity of the gas. [W/m·K]

The characteristic length L used is the height of the gas volume.

The empirical correlations used to calculate the Nusselt number of the gas-wall interface is a function of the Rayleigh number, which can be defined by the Grashof number and Prandtl number, as in equation eq. 3.3.

$$Ra = Gr \cdot Pr \tag{3.3}$$

- Ra is the Rayleigh number. [-]
- Gr is the Grashof number. [-]
- Pr is the Prandtl number. [-]

The Grashof number is a dimensionless number which approximates the ratio of the buoyancy forces to viscous forces, as given in equation [eq. 3.4]. The Prandtl number is a dimensionless number defined as the ratio of the momentum diffusivity to thermal diffusivity, as given in equation eq. 3.5.

$$Gr = \frac{\beta g \rho^2 L^3 \Delta T}{\mu^2} \tag{3.4}$$

$$Pr = \frac{c_p \mu}{k} \tag{3.5}$$

- $\beta$  is the coefficient of volume expansion. [1/K]
- g is the standard acceleration of gravity. [m/s²]

- $\rho$  is the gas density. [kg/m<sup>3</sup>]
- L is the characteristic length. [m]
- $\Delta T$  is the temperature difference of the surface and gas. [K]
- $\mu$  is the dynamic viscosity. [kg/m·s]
- $c_p$  is the heat capacity of gas. [J/kg·K]
- k is the thermal conductivity of gas. [J/m·K]

#### 3.3.2 Mixed convection

#### 3.3.3 Conduction

#### 3.3.4 Fire heat loads

The heat transfer from the flame to the shell is modelled using the recommended approach from Scandpower (Scandpower 2004). The heat transfer from the flame to the vessel shell is divided into radiation, convection and reradiation as seen in equation eq. 3.6.

$$q_f = \underbrace{\alpha_s \cdot \varepsilon_f \cdot \sigma \cdot T_f^4}_{\text{Radiation}} + \underbrace{h_f \cdot (T_f - T_s(t))}_{\text{Convection}} - \underbrace{\varepsilon_s \cdot \sigma \cdot T_s(t)^4}_{\text{Reradiation}}$$
(3.6)

- $q_f$  is the flame heat flux. [W/m<sup>2</sup>]
- $\alpha_s$  is the vessel surface absorptivity. [-]
- $\varepsilon_f$  is the flame emissivity. [-]
- $\sigma$  is the Stefan-Boltzmann constant,  $\sigma$  =  $5.67 \cdot 10^{-8} \ [\mathrm{W/m^2 \cdot K^4}]$
- $T_f$  is the flame temperature. [K]
- $h_f$  is the convection heat transfer coefficient between the flame and the surface. [W/m<sup>2</sup>·K]
- $T_s(t)$  is the time dependent surface temperature. [K]
- $\varepsilon_s$  is the surface emissivity. [-]

This model assumes that the pressure vessel is fully engulfed by the flame. This means that the view factor for the radiation is unity and is therefore not taken into consideration. The convective heat transfer coefficients for a jet fire and a pool fire, and recommended values for the emissivity and absorptivity, are given by Scandpower as (Scandpower 2004)

- $h_{iet\ fire} = 100 \ [W/m^2 \cdot K]$
- $h_{pool\ fire}$  = 30 [W/m $^2$ ·K]
- $\alpha_s = 0.85$
- $\varepsilon_s = 0.85$
- $\varepsilon_f = 1.0$  (optical thick flames, thickness > 1 m)

The flame temperature is found by solving equation eq. 3.7 for the incident heat flux in relation to the ambient conditions. The flame temperature is kept constant throughout the simulation.

$$q_{total} = \sigma \cdot T_f^4 + h_f \cdot (T_f - T_{amb}) \tag{3.7}$$

- $q_{total}$  is the incident flame heat flux as given in table tbl. 3.1. [W/m $^2$ ]
- $T_{amb}$  is the ambient temperature pprox 293 K (20 $^{\circ}$  C)

The heat flux used to calculate the flame temperature is given in table tbl. 3.1.

Table 3.1: Incident heat fluxes for various fire scenarios given by Scandpower (Scandpower 2004)

	Small jet fire [kW/m²]	Large jet fire [kW/m²]	Pool fire [kW/m²]
Peak heat load	250	350	150
Background heat load	0	100	100

### 3.4 Model implementation

# 4 Validation

# 4.1 Nitrogen discharge

The code is mainly for demonstration, and is provided as-is. However, a comparison has been made to experiment I1 from Haque et al. (see full ref below). The results are shown below.

## 4.2 Hydrogen filling

## 4.3 Air discharge/filling

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