



PROJECT PROPOSAL

# Octanis 1: A low-cost autonomous rover for Antarctica permitting real-time sensor data transfer

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[www.octanis.org](http://www.octanis.org)

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# 1 Introduction

»octanis | discovery and exploration« [2] is a student initiative to embark on ambitious and challenging projects together as multi-disciplinary students. Octanis 1 is our first mission to build a low-cost autonomous rover for polar, snow or ice covered regions. In this first iteration we will be specialising the rover for the coastal regions of Antarctica and its specific weather conditions. The rover will transmit sensor data like temperature, air pressure, relative humidity, current position and pose. There will be optical instruments on-board as well to sense obstacles and take pictures of the environment in its close vicinity. Other more specific environmental data collection is possible, but needs to be specially addressed due to the non-negligible power and weight requirements. These could be the concentration of an atmospheric gas,  $\alpha, \beta, \gamma$  radiation levels or snow and ice sampling.

We have selected the periodic sampling of snow and ice to be the most suitable initial science mission for Octanis 1 [1]. The rover will have a small hollow device capable of drilling into the surface ice or snow, melting the sample and analysing the pH.

Energy will be provided by the solar panels on Octanis' surfaces and used throughout the Antarctic summer in combination with a battery pack. Communications will go through the Iridium satellite network [3], providing a low cost, low power and reliable data transmission at any location on Earth.

## 2 Mission Overview

The aim of this mission, Octanis 1, is to provide a low-cost, low environmental impact rover platform for scientific experiments in cold to extremely cold environments. The rover will be small and light-weight enough to be carried on weather balloons. Its design will allow it to traverse icy terrain and be resistant to wind gusts. It will generate its own power with solar panels and regulate internal temperature as it's first priority. The rover will be on a four-wheel drive platform, each wheel being on a controllable strut, allowing it to drive in any orientation and right itself should it flip over.

### 2.1 Objectives

#### Robotics

#### Deployment

**Sensing** WP1 Airborne platform: - Long-range High Altitude Balloon: Reach Antarctica - Transmit telemetry and experiment data in real time - Payload is autonomous for an extended period of time (up to 6 months) and can be tracked

WP2 Chemistry: - Conduct a chemical experiment on-board a high altitude balloon flight to measure the [...] of the atmosphere at different altitudes.

WP3 Robotics: - Rover concept: Balloon is piggybacked to Antarctica by a rover platform. Rover is landed by a parachute if the landing perimeter is allowed to be large. Other landing concepts include a delta glider

## 2.2 Schedule

Dates are for the year 2014 if not otherwise stated. The detailed dates for the science mission is to be determined in coordination with the collaborating Antarctic base. The project is managed transparently and in an agile manner and the development is strictly cyclic as to quickly incorporate urgent features as their need emerges. This method has worked well in the past and has proven to enable fast response to bugs and keep the costs low. Transparency is provided by opening all documentation to the public via GitHub [4] repositories.

Code	Mission Phase	Dates	Weeks
A1	Prototype Design & Development	1.6. - 1.10.	12
A2	Software Development	1.6. - 1.10.	12
B1	Initial Testing	1.10. - 14.10.	2
B2	Drive & Sampling Testing on Swiss Glacier	14.10. - 14.11.	4
D	Mechanical Stress Testing	14.11. - 21.12.	1
E	Energy Management Testing	21.11. - 14.12.	3
F	Parachute Deployment Testing	14.12. - 21.12.	1
0.1	Deployment Opportunity in Antarctica	14.2.2015	1
0.2	Deployment Opportunity in Antarctica	1.12.2016	1

Table 1: Development timeline.

After deployment, the schedule of the science mission will proceed as described in the following table. Note that during the whole mission there will be a constant satellite data connection to the rover. Basic heartbeat information and sensory data will be transmitted multiple times per day. Commands to intercept the mission progress can be sent at any time.

Code	Mission Phase	Weeks
0	Deployment	1
1	All Systems Check	1
2	Drive & Sample	3
3	Camera Test & Image Transmission	3
4 (*)	Testing of Amateur Packet Radio Transmission	3

Table 2: Science Mission timeline.

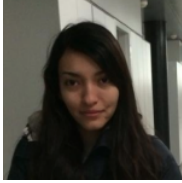
\*Depending on the deployment location and Antarctic base station radio capabilities.

## 2.3 Team

We are a group of students willing to challenge ourselves to the limit. Octanis 1 will not only be built by us, but with many helping professors and advisors that we will meet along the course of the mission. The following presentation is therefore one of the core team:



**Sam Sulaimanov** has seven years of experience working as a programmer and communications network engineer. He has been working with microcontrollers and electronics since he was a little boy and will make sure Octanis' brain functions correctly.



**Ana Roldàn** is a passionate physicist-to-be and involved in every corner of the project. She is not afraid to ask the big questions and inspires everyone with her strong passion for science.



**Raffael Tschui** is an Electrical Engineer, EPFL, B. Sc. He has the vital role of energy generation, control and regulation in the project. He keeps the rovers heart beating. He is currently on a mission in Columbia to help an EPFL professor build a bio-reactor.



**Pamela Canjura** has been interested in chemistry since she was five and competes regularly in the Chemistry Olympics. She is working on the chemical analyses that can be done on-board Octanis.

## 2.4 Budget

The following table describes the planned costs with an accuracy of  $\pm 20\%$  (m.c. = mission critical):

Cost Item	Priority	One-Time Costs
Rover Solar & Heating	m.c.	1000 CHF
Rover Communications	m.c.	500 CHF
Rover Instruments & Sensors	m.c.	500 CHF
Rover Electronics	m.c.	700 CHF
Rover Mechanics & Mobility	m.c.	1000 CHF
Balloon Transportation	opt.	1500 CHF
		mission critical: 3700 CHF
		optional: 1500 CHF

Table 3: General budget.

The only reoccurring costs are caused by the Iridium Satellite Messenger [3]. These range from 0.04-0.12 GBP per message (340 bytes / Message-Out) and a standard line-rental fee of 8 GBP / month. Per-message cost depends on the message volume needed.

## 3 Rover Transportation and Deployment

### 3.1 Environmental Sustainability

It is desirable to not leave any waste behind as well as it is required by the Antarctic Treaty. Depending on the method of deployment, the total mass and types of materials that will be included on a mission to Antarctica vary. A deployment of the rover to a target out of reach of normal Antarctic expeditions is done by using a High Altitude Balloon or a standard 3kg latex weather balloon. In such a long-distance mission, the rover will be separated from the balloon at the target location and the balloon will continue to fly unattended. Therefore the landing location of the balloon can not be known. Due to the nature of such a long-distance mission, typically the rover is unreachable to any expedition or base - the rover is left for the benefit of information on that location.

In a first step, it is therefore proposed to select a landing site that is in reach of normal Antarctic expeditions or bases. It is even possible to release the rover by other means like bringing it to the location manually or via helicopter.

Concluding, the rover is a construction of various polymers and metals, and a small risk exists that it will end up being uncontrollable due to malfunction. The total rover mass however does not exceed 2.5kg and therefore the pollution produced is small. In the unlikely event of failure, the rover can then be retrieved manually thanks to the last known transmitted location.

## 4 Rover Subsystems

### 4.1 Electronics

### 4.2 Power & Thermal

The energy of the sunlight per square meter is  $E_0 * \sin(\alpha)$  with  $E_0 = 1367W/m^2$  and  $\alpha$  the angle between the direct incoming sunrays and the horizontal. For the Antarctic, we can use  $\alpha \approx 23.5 + 90 + \phi$  where  $\phi$  is the latitude, to get an estimate of the maximum energy (i.e. in summer) at a certain geographic point (see figure xxxx).

This would lead to  $1050W/m^2$  for the northmost point of the antarctic (@ 65°S) and  $545W/m^2$  at the south pole (@ 90°S). Bare in mind that, during the six sunny months this value will oscillate between 0 and the maximum value just calculated. Assuming our panels are installed vertically and considering current solar panel efficiency (around 15-20%) this means that per square meter we can convert max. 200 W into electrical energy. If we take a pessimistic efficiency and a “radiation mean value” during the 6 months of the mission (just 1/2 of the maximum value) we would need a panel of the size of  $0.13m^2$  (@ 65°S) to  $0.25m^2$  (@ 90°S) to output a power of 10 W. But, this didn’t consider the daily oscillation of the radiation angle, nor that it does get night during certain time periods for any point north from the south pole. Therefore, it is more reasonable to

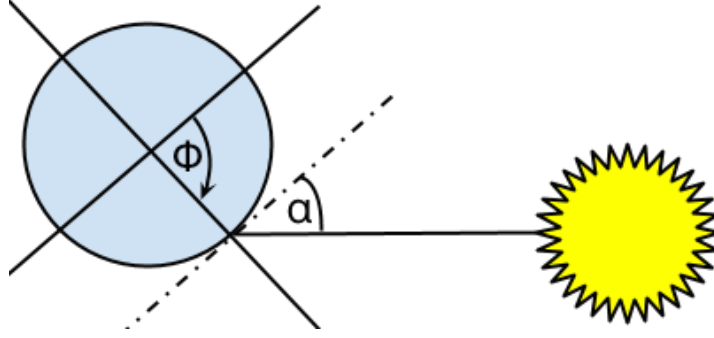


Figure 1: Octanis rover design concept built after MUSES-CN nanorover.

take the pessimistic  $0.25m^2$  value (also because the energy supply will decrease with bad weather).

To summarise, a panel surface of  $0.25m^2$  will provide us with an average (seasonal, daily) power of 10W and these numbers are proportional.

The ultimate priority is to keep the system at  $-20^{\circ}C$  or above, which means that all the energy from the solar cell and, if necessary, from the battery will first be used to maintain this goal. The heating system is therefore directly connected between the solar cell and the battery charge controller. It is controlled with electronic comparators that enable one or another energy source to power the heating pads depending on temperature thresholds listed below. The charge controller itself takes care of the temperature limits for charging the battery and switches itself on and off automatically.

### 4.3 Mechanical

The rover body design (fig. 1) has been substantially influenced by NASA's MUSES-CN nanorover [5]. The nanorovers design is a unique design as it allows almost complete freedom. We have designed Octanis 1 to be lightweight with a large enough surface area to produce enough solar energy for the mission duration. The body dimensions are roughly 30cm x 30cm x 6cm.

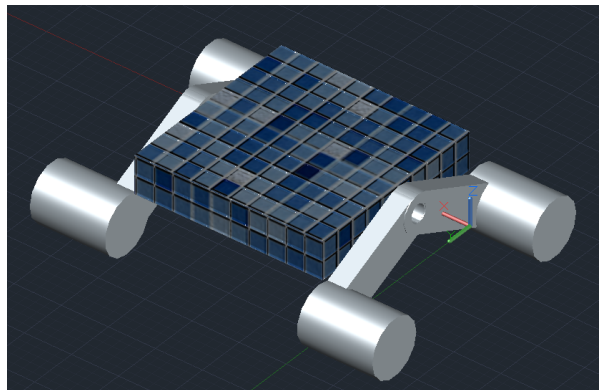


Figure 2: Octanis rover design concept built after MUSES-CN nanorover.

As depicted in figure 2, this rover design allows many orientations for different situ-

ations. Main concerns of the mission are energy supply and weather resistance. As the whole rover body is covered with solar cells, the rover can right itself up according to sun position to expose most of its area to sunlight. In harsher weather conditions it is also crucial to make sure snow doesn't cover up the panels. This can be prevented by periodically flipping the rover. Not only snow, but wind can be an issue and sometimes even flipping over the rover. If this occurs, the rover can return to a driving position by actuating the struts. All wheel struts can be rotated around the axis and are fitted with a separate motor each. They are moved through a worm drive to keep them static when they are not rotating without having to apply constant power.

Another benefit of this design is it allowing us to build a simpler drilling device. When drilling an ice core, the rover can lower itself to the ground exploiting its own weight as pressure onto the drill. The drill bit just needs to spin and doesn't need to retract into the rover body.

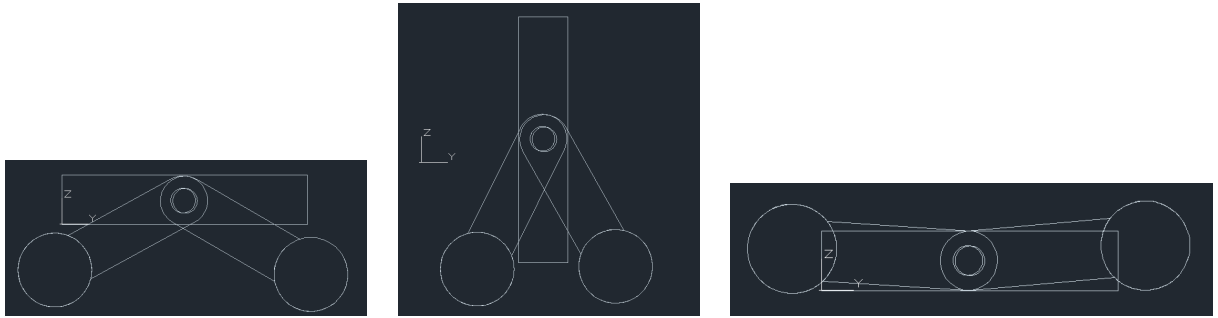


Figure 3: Different wheel strut configurations l.t.r.: (1) drive mode, (2) solar recharge mode, (3) wind protect mode.

## 4.4 Mobility

Snowy and icy terrain will be Octanis' main habitat and so its wheels need to be specifically designed to be wide, lightweight and should provide a good traction going upwards. All four wheels will be individually motorised and provide enough torque to move Octanis on an inclination of 30 degrees. Turning can be achieved by rotating the wheels on each side at different speeds.

An inertial measurement unit (IMU) will be used to know the rovers orientation at any point in time. It consists of a gyroscope, accelerometer, precision barometer and magnetometer. The sensors information is fed back to the control systems that actuate the motors. Should the rover need to go over an obstacle, the wheel struts will be rotated as far as possible so that the rover body is always in a plane parallel to the ground. This minimises the risk of being tipped over and should allow the easy traversal of obstacles higher than the wheel radius and is similar to the rocker-bogie suspension system used on NASAs mars rovers [6]. Due to the high risk of overturn caused by wind gusts the passive rocker-bogie design was not chosen, since there would be no simple way of righting back up.



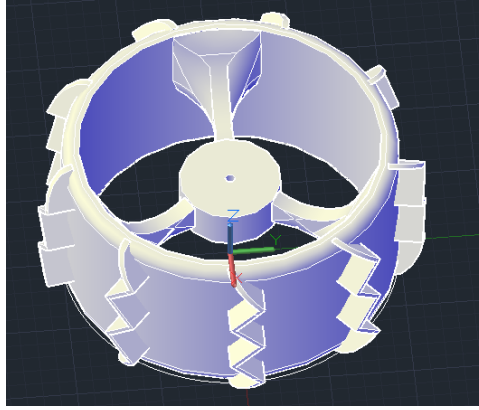


Figure 4: Initial wheel design. Fins allow the wheel to dig into the snow while driving providing better traction.

## 4.5 Communication

Regular data transmission is a mission critical feature of the rover. It is achieved by using the Iridium Short Burst Data (SBD) service with the help of the RockBLOCK Iridium 9602 modem [3]. This modem allows the transmission of data packets of up to 340 bytes outwards and 270 bytes inwards approximately every 20 seconds. Larger data frames can be fragmented and transmitted with multiple packets. The key to this modem and network is that is available everywhere on Earth at any time, something that cannot be said of the GSM network or even amateur radio. That the modem require just as low power as the GPS on board adds to the list of benefits. Information sent through the Iridium network is automatically sent to a server and uploaded to the project website, where the public can view Octanis' state and data.

Octanis 1 will also be equipped with a 5 watt one-way APRS transmitter, transmitting sensor data and location information on the 2-meter band. This can be received by amateur radio operators (or unlicensed individuals with radio scanners) who are in range. The range of this transmitter is typically hundreds of kilometers when travelling in the sky (i.e. attached to a balloon). On the ground, the range is orders of magnitude smaller.

## 4.6 Optical

# 5 Science Mission

## 5.1 Snow and Ice

[1]

## References

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