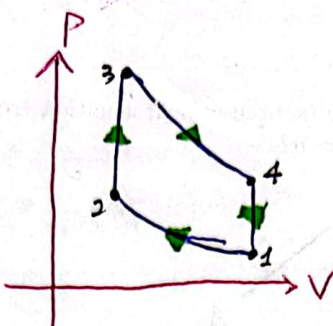
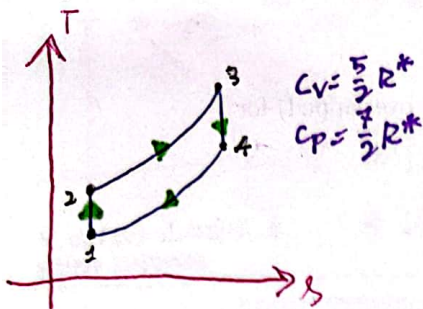


(E.3)  $\Delta V = 1800 \mu = \frac{1800 \cdot 10^{-6} \text{ m}^3}{4} = V_1 - V_2 = 4,5 \cdot 10^{-4} \text{ m}^3$   
 $r = \frac{V_1}{V_2} = 9$   $\leadsto V_1 = 5,06 \cdot 10^{-4} \text{ m}^3$   
 $V_2 = 5,63 \cdot 10^{-5} \text{ m}^3$

$P_1 = 0,05 \cdot 10^5 \text{ Pa}$   $T_1 = 323,15 \text{ K}$   $T_3 = 2773,15 \text{ K}$



	P [Pa]	V [m³]	T [K]
1	$0,05 \cdot 10^5$	$5,06 \cdot 10^{-4}$	323,15
2	$2,04 \cdot 10^6$	$5,63 \cdot 10^{-5}$	778,34
3	$7,25 \cdot 10^6$	$5,63 \cdot 10^{-5}$	2773,15
4	$3,35 \cdot 10^6$	$5,06 \cdot 10^{-4}$	1152,17

$R^* = \frac{R}{M_m} = \frac{8,314 \text{ J/mol K}}{28,97 \text{ kg/mol}} = 288,7 \frac{\text{J}}{\text{mol K}}$

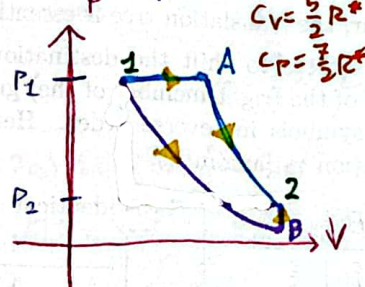
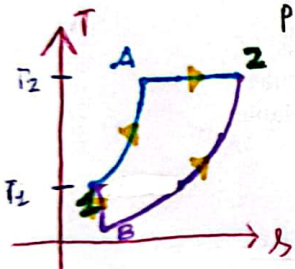
$PV = mRT$   $m = \frac{PV_1}{R^* T_1} = 5,1 \cdot 10^{-4} \text{ kg}$

1-2)  $\Delta S = 0 = m(C_p \ln(\frac{T_2}{T_1}) + R^* \ln(\frac{V_2}{V_1})) \rightarrow T_2 = 778,34 \text{ K} \rightarrow P_2 = 2,04 \cdot 10^6 \text{ Pa}$

2-3)  $\Delta V = 0$   $V_3 = V_2 \rightarrow P_3 = 7,25 \cdot 10^6 \text{ Pa}$

4-1)  $\Delta V = 0$   $V_4 = V_1$  3-4)  $\Delta S = 0 = m(C_p \ln(\frac{T_4}{T_3}) + R^* \ln(\frac{V_4}{V_3})) \rightarrow T_4 = 1152,17 \text{ K} \rightarrow P_4 = 3,35 \cdot 10^6 \text{ Pa}$

$M = 2 \text{ kg}$   $M_{N_2} = 28,97 \text{ mol} \rightarrow R^* = 296,95 \frac{\text{J}}{\text{kg K}}$   $P_1 = 6 \cdot 10^5 \text{ Pa}$   $T_1 = 293,15 \text{ K}$   $P_2 = 2 \cdot 10^5 \text{ Pa}$   $T_2 = 343,15 \text{ K}$



	P [Pa]	V [m³]	T [K]
1	$6 \cdot 10^5$	0,29	293,15
2	$2 \cdot 10^5$	1,02	343,15
3	$1,03 \cdot 10^5$	0,34	177,26
4	$6 \cdot 10^5$	0,29	343,15

$\Delta U = m C_v (T_2 - T_1)$   $\Delta H = m C_p (T_2 - T_1)$   $\Delta S = m(C_p \ln(\frac{T_2}{T_1}) - R^* \ln(\frac{P_2}{P_1}))$   $\Delta V = V_2 - V_1$

$V_1 = 0,29 \text{ m}^3$   $V_2 = 1,02 \text{ m}^3$

•  $V_3 = 0,34 \text{ m}^3$  •  $\Delta S = 0 = m(C_p \ln(\frac{T_3}{T_2}) + R^* \ln(\frac{V_3}{V_2})) \rightarrow T_3 = 177,26 \text{ K}$   $P_3 = 1,03 \cdot 10^5 \text{ Pa}$

•  $Q_{3A} + Q_{42} = P_3(V_3 - V_1) + m C_p (T_4 - T_1) + m R^* T_2 \ln(\frac{V_2}{V_1}) = 253570 \text{ J}$

$L_{1A} + L_{42} = m P_1 (V_2 - V_1) - m R^* T_2 \ln(\frac{V_2}{V_1}) = 253570 \text{ J}$

•  $Q_{1B} + Q_{23} = 0 + m C_v (T_2 - T_3) = 363150 \text{ J}$

$Q_{1B} + Q_{23} = -m C_v (T_3 - T_2) + 0 = 253570 \text{ J}$

$M = 0,3 \text{ kg}$   $C_{p, \text{BIAIUMICO}} = 28,97 \text{ J/mol K}$   $T_1 = 293,15 \text{ K}$   $P_1 = 10^5 \text{ Pa}$   $\beta = \frac{V_1}{V_2} = 1$   $P_2 = 10^6 \text{ Pa}$   $M_{N_2} = 28,97 \text{ kg/mol}$

$P_1 T_1 = m R^* T_2$   $R^* = 296,95 \frac{\text{J}}{\text{kg K}}$   $(V_1 = 0,265 \text{ m}^3)$   $(V_2 = 0,066 \text{ m}^3) \rightarrow T_2 = 740 \text{ K}$   $L = -\Delta U = -m C_v (T_2 - T_1) = -9,84 \cdot 10^4 \text{ J}$

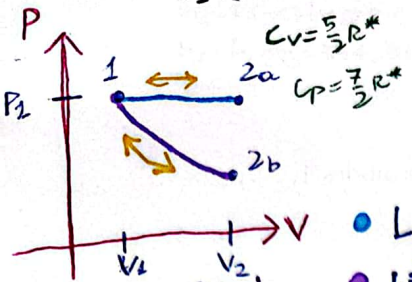
$\Delta S = m(C_p \ln(\frac{T_2}{T_1}) - R^* \ln(\frac{P_2}{P_1})) = 78,3 \frac{\text{J}}{\text{K}} > 0 \Rightarrow \text{IRREV.}$

•  $\Delta S_{\text{tot}} = 0$   $m(C_p \ln(\frac{T_2}{T_1}) - R^* \ln(\frac{P_2}{P_1})) = 0 \rightarrow P_{2, \text{rev}} = 2,12 \cdot 10^6 \text{ Pa}$   
 $(L_{1B} = L_R)$



$T_1 = 293,15^\circ\text{K}$  (Es. 4)  $P_1 = 58 \cdot 10^5 \text{ Pa}$   $V_1 = 0,1 \text{ m}^3$   $V_2 = 0,3 \text{ m}^3$   $R^* = 287 \frac{\text{J}}{\text{kg}^\circ\text{K}}$  BIATOMICO

$P_1 V_1 = m R^* T_1 \rightarrow m = 6,89 \text{ kg}$



	P [Pa]	V [m³]	T [°K]
1	$58 \cdot 10^5$	0,1	293,15
2a	$58 \cdot 10^5$	0,3	879,93
2b	$19,3 \cdot 10^5$	0,3	293,15

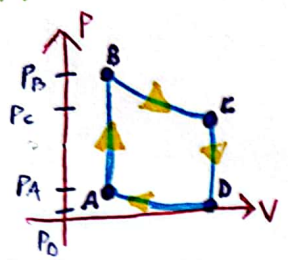
$T_{20c} = \frac{P_{2a} V_{2a}}{m R^*} = 879,93^\circ\text{K}$

$P_{2b} = \frac{m R^* T_{2b}}{V_{2b}} = 19,3 \cdot 10^5 \text{ Pa}$

$L = P \cdot (V_{2a} - V_1) = 1,16 \cdot 10^6 \text{ J}$   $\Delta U = m C_V (T_{2a} - T_1) = 2,4 \cdot 10^6 \text{ J}$   $Q = m C_P (T_{2a} - T_1) = 4,06 \cdot 10^6 \text{ J}$   
 $L = R^* T_1 \ln(\frac{V_{2b}}{V_1}) = 6,4 \cdot 10^5 \text{ J}$   $\Delta U = 0$   $\Delta H = 0$   $Q = L = 6,4 \cdot 10^5 \text{ J}$

Si noti che  $\Delta U = Q - L$  è SEMPRE RISPONDA

$n = 2 \text{ mol}$   $PV = nRT$  (Es. 5)



	P [Pa]	V [m³]	T [°K]
A	$2 \cdot 10^5$	0,01	122
B	$8 \cdot 10^5$	0,01	488
C	$2,285 \cdot 10^5$	0,04	488
D	$0,571 \cdot 10^5$	0,04	122

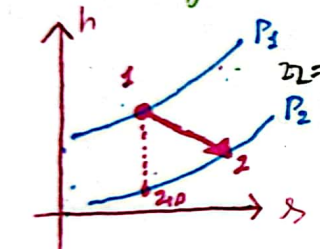
$\Delta U_{\text{tot}} = 0$

$\rightarrow Q_{\text{tot}} = L_{\text{tot}}$

$L_{\text{tot}} = L_{AB} + L_{BC} + L_{CD} + L_{DA}$

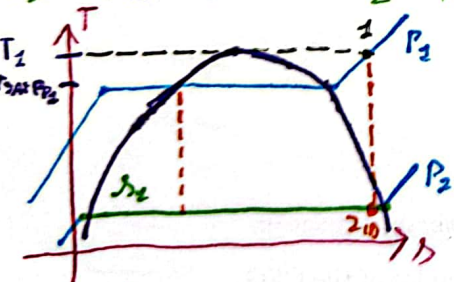
$= 0 + n R T_B \ln(V_C/V_B) + 0 + n R T_D \ln(V_D/V_D) = 8,4 \cdot 10^3 \text{ J}$

$Q_R = 2000 \frac{\text{kJ}}{\text{kg}}$  (Es. 6)  $M_{H_2} = 4 \cdot 10^{-3} \frac{\text{kg}}{\text{mol}}$  MONOATOMIC



$P_1 = 8 \text{ Bar}$   $T_1 = 1073,15^\circ\text{K}$   $P_2 = 2 \cdot 10^5 \text{ Pa}$   $T_2 = ?$   
 $C_P = \frac{5}{2} R^*$   $C_V = \frac{3}{2} R^*$   $R^* = \frac{R}{M_{H_2}} = 2078,62 \frac{\text{J}}{\text{kg}^\circ\text{K}}$   
 $1-2_{10} \Delta h = 0 = m (C_P \ln(\frac{T_{210}}{T_1}) - R^* \ln(\frac{P_2}{P_1}))$   
 $\Rightarrow \eta = \frac{Q_R}{C_P (T_{210} - T_1)} = 84,3\%$

$P_1 = 20 \text{ MPa}$   $T_1 = 500^\circ\text{C}$   $P_2 = 0,17 \text{ MPa}$   $L_R = 650 \frac{\text{kJ}}{\text{kg}}$  (Es. 7)



$\eta = \frac{Q_R}{L_{10}} = \frac{L_{10}}{h_1 - h_{210}}$   $T_2 > T_{\text{SAT}} @ P_2 \rightarrow \text{ACQUA SURRISCALDATA}$   
 $h_2 = h(20 \text{ MPa}, 500^\circ\text{C}) = 3393,5 \frac{\text{kJ}}{\text{kg}}$   
 $h_1 = h_{210} = h(20 \text{ MPa}, 500^\circ\text{C}) = 6,3348 \frac{\text{kJ}}{\text{kg}^\circ\text{K}}$   
 $h_{10} @ P_2 = 617080 \frac{\text{kJ}}{\text{kg}^\circ\text{K}}$   $h_{10} @ P_2 = 1,9922 \frac{\text{kJ}}{\text{kg}^\circ\text{K}}$   $\Delta h = L_{210} - L_{10}$

$x = \frac{h_{10} - h_1}{h_{210} - h_1} = 0,92$   $h_2 = (1-x) h_1 @ P_2 + x h_{10} @ P_2 = 2598,19 \frac{\text{kJ}}{\text{kg}} \rightarrow \eta = 81,7\%$

$H = 100 \text{ m}$   $d = 100 \text{ m}$   $T_1 = 293,15^\circ\text{K}$   $P_1 = 10^5 \text{ Pa}$   $w_1 = 12 \text{ m/s}$   $w_0 = 3 \text{ m/s}$   $\eta = 60\%$  (Es. 8)

BILANCO DI ENERGIA  $\dot{m} (h_2 - h_0 + g(z_2 - z_0) + \frac{w_2^2 - w_0^2}{2}) + \dot{Q} - \dot{L} = 0$  SOPPONIAMO ADIABATICA  $M_{\text{ACQUA}} = 28,8 \frac{\text{kg}}{\text{kmol}}$   
 $\eta = 60\% \cdot (\dot{m} \frac{w_2^2 - w_0^2}{2})$

EQUAZIONE DI CONTINUITA  $\dot{m} = P w_1 \Omega = P_1 w_1 (\pi \frac{D^2}{4})$   $P_2 = 1/w_2$   
 $P_1 w_1 = R^* T_1$   $w_1 = \frac{R/M \cdot T_1}{P_1} = 0,82 \text{ m}^3/\text{kg}$   $\rightarrow \dot{m} = 11,3 \cdot 10^4 \text{ kg/s}$

$L = 4,7 \text{ MW}$



(E5.9)

$P = 2,7 \cdot 10^5 \text{ Pa}$   $M_1 = 4 \text{ kg}$   $X_1 = 0,2$   $M_2 = 2 \text{ kg}$   $X_2 = 1$   $T_2 = 353,15^\circ \text{K}$   $T_{eq}?$

$\Delta H = 0$   $M_1 h_{eq} = M_1 h_1 + M_2 h_2$   $M_{tot} = 6 \text{ kg}$

$h_1 = (1-x) h_{1p} + x h_{1v} = 0,2 \cdot 546,31 + 0,8 \cdot 2720,5 = 984,15 \frac{\text{kJ}}{\text{kg}}$

$h_2 = h_{2p} = 334,91 \frac{\text{kJ}}{\text{kg}}$   $h_{eq} = \frac{M_1 h_1 + M_2 h_2}{M_{tot}} = 765,77 \frac{\text{kJ}}{\text{kg}}$

$h_{eq} > h_1 \Rightarrow \text{SISTEMA BIFASE}$   $T_{eq} = T_{sat}(0,27 \text{ MPa}) = 130^\circ \text{C}$

(E5.10)

CILINDRATA =  $0,0018 \text{ m}^3$   $P = \frac{V_1}{V_2} T_2 = 2773,15^\circ \text{K}$   $P_1 = 0,95 \cdot 10^5 \text{ Pa}$   $T_1 = 323,15^\circ \text{K}$   $M_m = 288 \cdot 10^{-3} \frac{\text{kg}}{\text{mol}}$   $\eta?$

$\Delta V = V_2 - V_1 = \frac{V_1}{P_1} (P_1 - P_2) = 4,5 \cdot 10^{-4} \text{ m}^3$

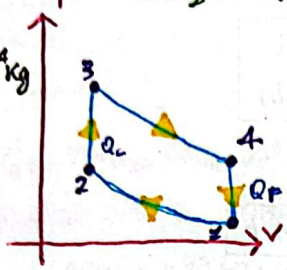
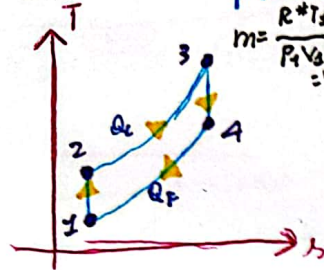
$\rightarrow V_3 = 5,0625 \cdot 10^{-4} \text{ m}^3$   $V_2 = 0,5625 \cdot 10^{-4} \text{ m}^3$

$R^* = \frac{R}{M_m} = 288,7 \frac{\text{J}}{\text{kg} \cdot \text{K}}$

$PV = mR^*T$

$m = \frac{R^* T_1}{P_1 V_1} = 5,7 \cdot 10^{-4} \text{ kg}$

$C_v = \frac{5}{2} R^*$   $C_p = \frac{7}{2} R^*$



	$P [\text{Pa}]$	$V [\text{m}^3]$	$T [^\circ \text{K}]$
1	$0,95 \cdot 10^5$	$5,0625 \cdot 10^{-4}$	323,15
2	$2,04 \cdot 10^6$	$0,5625 \cdot 10^{-4}$	778,34
3	$7,25 \cdot 10^6$	$0,5625 \cdot 10^{-4}$	2773,15
4	$3,35 \cdot 10^6$	$5,0625 \cdot 10^{-4}$	152,57

1-2)  $\Delta h = 0 = m [C_p \ln(T_2/T_1) + R^* \ln(V_2/V_1)] \rightarrow T_2 = 778,34^\circ \text{K}$   $P_2 = 2,04 \cdot 10^6 \text{ Pa}$

2-3)  $V_3 = V_2 \rightarrow P_3 = 7,25 \cdot 10^6$  3-4)  $V_4 = V_1$  3-4)  $\Delta h = 0 = \frac{5}{2} R^* \ln(T_4/T_3) + R^* \ln(V_4/V_3) \rightarrow T_4 = 152,57^\circ \text{K}$   $P_4 = 3,35 \cdot 10^6 \text{ Pa}$

$\eta = 1 - \frac{Q_F}{Q_C} = 1 - \frac{m C_p (T_4 - T_1)}{m C_p (T_2 - T_1)} = 58,4\%$

(E5.11)

$T_C = 673,15^\circ \text{K}$   $T_F = 288,15^\circ \text{K}$   $T_{FP} = 312,15^\circ \text{K}$   $\eta_{II}?$

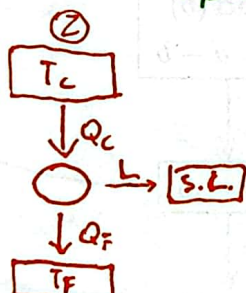
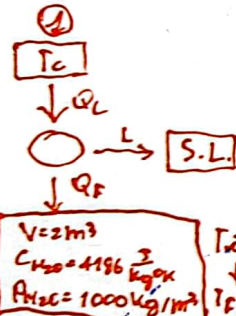
$T_F = 288,15^\circ \text{K}$

$\eta = 1 - \frac{Q_F}{Q_C}$

①  $Q_C = m C_{p,b} T_C \ln(T_F/T_{FP})$   $Q_F = m C_{H_2O} (T_{FP} - T_F)$   $\rightarrow \eta = 55\%$

②  $\eta = 1 - \frac{T_F}{T_C} = 57,2\%$

$\eta_{II} = \frac{\eta_1}{\eta_2} = 96,2\%$



$\dot{V}_1 = 700 \text{ m}^3/\text{h}$   $T_1 = 273,15^\circ \text{K}$   $P_1 = 10^5 \text{ Pa}$   $T_2 = 313,15^\circ \text{K}$   $T_F = 273,15^\circ \text{K}$   $T_C = 363,15^\circ \text{K}$   $Q_A? Q_C? Q_F? \epsilon?$

$m(h_2 - h_1) + \dot{Q}_A - \dot{Q}_F = 0$   $\dot{m} = \frac{\dot{V}_1}{v_1}$   $P v_1 = R^* T_1$   $R^* = 288,7 \frac{\text{J}}{\text{kg} \cdot \text{K}}$

$\dot{v}_1 = 0,789 \text{ m}^3/\text{s}$   $\rightarrow \dot{m} = 0,873 \text{ kg/s}$

$h_1 - h_2 = C_p (T_1 - T_2)$

$\dot{Q}_A = \dot{m} \cdot \frac{7}{2} R^* (T_2 - T_1) = 9,94 \cdot 10^3 \text{ W}$

BILANCIO ENERGIA  $\dot{Q}_A \neq \dot{Q}_C + \dot{Q}_F$   
BILANCIO ENTROPIA  $\frac{\dot{Q}_C}{T_C} + \frac{\dot{Q}_F}{T_F} - \dot{m} (C_p \ln(T_2/T_1) - R^* \ln(P_2/P_1)) = 0$

$\rightarrow \dot{Q}_C = 2,67 \cdot 10^3 \text{ W}$

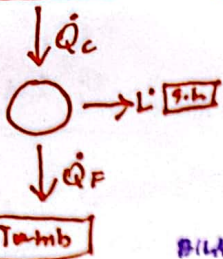
$\dot{Q}_F = 7,27 \cdot 10^3 \text{ W}$

$\epsilon = \frac{\dot{Q}_A}{\dot{Q}_C} = 3,72$



ES.33  $\dot{m} = 1 \text{ kg/s}$   $T_1 = 473,15^\circ\text{K}$   $x_1 = 1$   $T_2 = 298,15^\circ\text{K}$   $P_2 = 0,1 \text{ MPa}$   $T_{\text{amb}} = 298,15^\circ\text{K}$   $L?$

SISTEMA APERTO



①  $\dot{m} (h_2 - h_1 + g(z_2 - z_1) + \frac{w_2^2 - w_1^2}{2}) + \dot{Q}_c - \dot{L} = 0$

$\rightarrow \dot{Q}_c = \dot{m} (h_2 - h_1)$

$h_1 = h_f(200^\circ\text{C}) = 2793,15 \frac{\text{kJ}}{\text{kg}}$

$h_2 = h_g(200^\circ\text{C}) = 6,4323 \frac{\text{kJ}}{\text{kg}}$

$T_2 < T_{\text{sat}}(0,1 \text{ MPa})$   $h_2 = h_l(25^\circ\text{C}) = 104,89 \frac{\text{kJ}}{\text{kg}}$

$h_2 = h_l(25^\circ\text{C}) = 0,3674 \frac{\text{kJ}}{\text{kg}}$

$\Rightarrow \dot{Q}_c = 217,10^3 \text{ kW}$

BILANCIO ENTALPIA

$\dot{m} (h_2 - h_1) + \frac{\dot{Q}_f}{T_{\text{amb}}} = 0 \rightarrow \dot{Q}_f = 1,75 \cdot 10^3 \text{ kW}$   $L = 9,5 \cdot 10^3 \text{ kW}$

ES.34  $\dot{Q}_c = 20 \text{ kW}$   $T_{\text{H}_2\text{O}} = 300,15^\circ\text{K}$   $T_{\text{H}_2\text{O}} = 310,15^\circ\text{K}$   $T_{\text{ARIA}} = 279,15^\circ\text{K}$   $T_{\text{ARIA}} = 275,15^\circ\text{K}$   $\dot{m}_{\text{H}_2\text{O}}, \dot{m}_{\text{ARIA}}, L, E?$

$C_{P\text{H}_2\text{O}} = 4186 \frac{\text{J}}{\text{kg}^\circ\text{K}}$   $C_{P\text{ARIA}} = \frac{7}{2} \cdot \frac{R}{M_{\text{ARIA}}} = 1020,45 \frac{\text{J}}{\text{kg}^\circ\text{K}}$

$\dot{Q}_c = \dot{m}_{\text{H}_2\text{O}} C_{P\text{H}_2\text{O}} \Delta T_{\text{H}_2\text{O}} \rightarrow \dot{m}_{\text{H}_2\text{O}} = 1,49 \text{ kg/s}$

BILANCIO ENTALPIA  $\dot{m}_{\text{H}_2\text{O}} \Delta h_{\text{H}_2\text{O}} - \dot{m}_{\text{ARIA}} \Delta h_{\text{ARIA}} = 0$

$\Delta h_{\text{H}_2\text{O}} = \dot{m}_{\text{H}_2\text{O}} C_{P\text{H}_2\text{O}} \ln\left(\frac{T_{\text{H}_2\text{O}}}{T_{\text{H}_2\text{O}}}\right)$   $\Delta h_{\text{ARIA}} = C_{P\text{ARIA}} \ln\left(\frac{T_{\text{ARIA}}}{T_{\text{ARIA}}}\right)$

$\rightarrow \dot{m}_{\text{ARIA}} = 5,95 \text{ kg/s}$

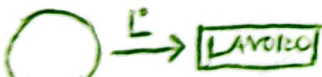
$\dot{Q}_f = \dot{m}_{\text{ARIA}} C_{P\text{ARIA}} \Delta T_{\text{ARIA}} = 17,95 \text{ kW}$

$L = \dot{Q}_f - \dot{Q}_c = 2,05 \text{ kW}$

$E_{\text{PDL}} = \frac{\dot{Q}_c}{L} = 9,76$

ES.35  $T_c = 1473,15^\circ\text{K}$

$\dot{Q}_c = 2000 \text{ kW}$



$\eta_I = \frac{L}{\dot{Q}_c}$   $\eta_{II} = \frac{\eta_I}{\eta_{ID}}$

$\eta_{ID} = 1 - \frac{T_F}{T_C} = 80,1\% \rightarrow \eta_I = 40\%$

$L = \dot{Q}_c \cdot \eta_I = 40 \text{ kW}$

$T_F = 243,15^\circ\text{K}$

ES.36  $T_F = 273,15^\circ\text{K}$   $T_{1A} = 473,15^\circ\text{K}$   $P_{1A} = 2 \text{ MPa}$   $T_{1B} = 473,15^\circ\text{K}$   $\text{VAPOR}$   $T_2 = 293,15^\circ\text{K}$   $P_2 = 0,1 \text{ MPa}$   $L? \dot{Q}_{cA} = \dot{Q}_{cB}$



①  $T_{1A} < T_{\text{sat}} @ 2 \text{ MPa} \rightarrow h_1 = h_l @ 200^\circ\text{C} = 852,45 \frac{\text{kJ}}{\text{kg}}$   $h_2 = h_l @ 200^\circ\text{C} = 7,3709 \frac{\text{kJ}}{\text{kg}}$

②  $h_2 = h_v @ 200^\circ\text{C} = 2793,2 \frac{\text{kJ}}{\text{kg}}$   $h_2 = h_v @ 200^\circ\text{C} = 6,4323 \frac{\text{kJ}}{\text{kg}}$

$h_f = h_l @ 200^\circ\text{C} = 83,96 \frac{\text{kJ}}{\text{kg}}$   $h_g = h_l @ 200^\circ\text{C} = 0,2966 \frac{\text{kJ}}{\text{kg}}$

$L = \dot{Q}_c - \dot{Q}_f$   $\dot{Q}_c = \dot{m}_A \Delta h_A = \dot{m}_B \Delta h_B$   $\dot{m}_A (h_f - h_1) = \dot{m}_B (h_g - h_2)$

$\rightarrow \dot{m}_A = 3,53 \text{ kg/s}$

$\dot{m}_A \Delta h_A + \frac{\dot{Q}_f}{T_F} = 0$  (o  $\dot{m}_B \Delta h_B$ )

$\rightarrow \dot{Q}_c = 217,10^3 \text{ kW}$

$\dot{Q}_f = 1,96 \cdot 10^3 \text{ kW}$

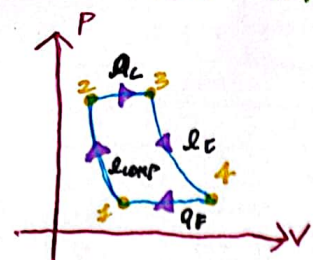
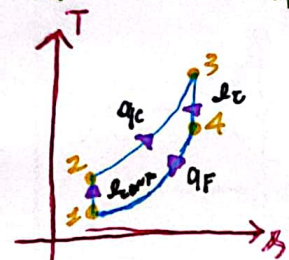
$L = 0,75 \cdot 10^3 \text{ kW}$



ES. 17

$$m = 28,8 \cdot 10^{-3} \text{ kg/m}^3$$

$$T=20 \quad q_c? \quad T_4? \quad \eta? \quad L=250 \cdot 10^6 \text{ W m}^3 \quad v_3=100 \text{ m/s} \quad \eta_3?$$



	P [Pa]	T [°K]
1	$10^5$	300,15
2	$20 \cdot 10^5$	706,42
3	$20 \cdot 10^5$	1700,17
4	$10^5$	722,38

$$R^* = 289,69 \frac{\text{J}}{\text{mol} \cdot \text{K}} \quad C_v = \frac{5}{2} R^* \quad C_p = \frac{7}{2} R^*$$

$$T = P_2/P_1 \rightarrow P_2 = 20 \cdot 10^5 \text{ Pa}$$

$$12) \Delta h = 0 = m C_p \ln(T_2/T_1) - R^* \ln(P_2/P_1) \rightarrow T_2 = 706,42^\circ \text{K}$$

$$23) \Delta P = 0 \quad 41) \Delta P = 0$$

$$34) \Delta h = 0 = C_p \ln(T_4/T_3) - R^* \ln(P_4/P_3) \rightarrow T_4 = 722,38^\circ \text{K}$$

$$q_c = C_p (T_3 - T_2) = 1004,08 \cdot 10^3 \frac{\text{J}}{\text{kg}}$$

$$q_F = C_p (T_4 - T_1) = 426,63 \cdot 10^3 \frac{\text{J}}{\text{kg}}$$

$$\eta = 1 - \frac{q_c}{q_F} = 57,5\%$$

$$L = \dot{m} (L_{\text{TURBINA}} - L_{\text{COMPRESSORE}}) = \dot{m} C_v ((T_3 - T_4) - (T_2 - T_1)) \rightarrow \dot{m} = \frac{L}{C_v ((T_3 - T_4) - (T_2 - T_1))} = 283,58 \frac{\text{kg}}{\text{s}}$$

$$\text{EQUAZIONE DI CONTINUITÀ} \quad \dot{m} = \rho W \Omega$$

$$P_3 = \frac{1}{\eta_3}$$

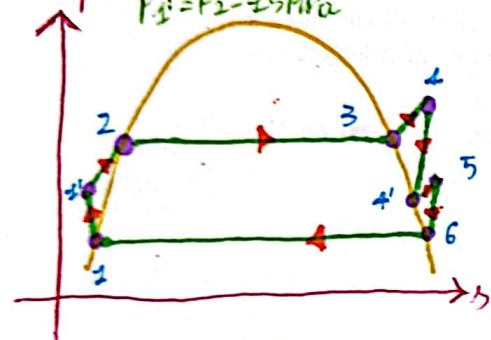
$$P v = R^* T \rightarrow v_3 = \frac{R^* T_3}{P_3} = 4,91 \text{ m}^3/\text{kg} \quad \Omega_3 = 13,92 \text{ m}^2$$

$$Q_c = \dot{m} q_c = 284,7 \text{ MW} \quad Q_F = \dot{m} q_F = 120,98 \text{ MW}$$

$$\dot{Q}_{cR} = 90\% \cdot Q_F = 108,89 \text{ MW} \quad T_1 = 33^\circ \text{C} \quad P_1 = P_2 = 15 \text{ MPa}$$

$$\text{TURBINA ALTA} \quad 4 = 600^\circ \text{C} \quad 15 \text{ MPa} \quad 4' = 1 \text{ MPa}$$

$$\text{TURBINA BASSA} \quad 600^\circ \text{C}, 1 \text{ MPa} \quad \text{DRAINARE, L, DILUORIBIMATO?}$$



$$32,88^\circ \text{C} \quad \text{DA TURBINA IN P}$$

	$h [\frac{\text{kJ}}{\text{kg}}]$	$s [\frac{\text{kJ}}{\text{kg} \cdot \text{K}}]$
1	137,82	
1'	152,82	
2		
3		
4	3582,3	6,6776
4'	2806,98	6,6776
5	3697,9	8,0290
6	2440,32	8,0290

NON RILEVANTI

$$1) h_2 = h_1 @ 33^\circ \text{C} = 137,82 \frac{\text{kJ}}{\text{kg}}$$

$$2) h_2' - h_2 = v (P_2 - P_1) \quad \text{ASSUMENDO } v = 0,001 \text{ m}^3/\text{kg} \quad P_1 = P @ 32,88^\circ \text{C} = 5 \text{ kPa} \rightarrow h_2' = 152,82 \frac{\text{kJ}}{\text{kg}}$$

$$3)$$

$$4)$$

$$4) h_4 = h @ (600^\circ \text{C}, 15 \text{ MPa}) = 3582,3 \frac{\text{kJ}}{\text{kg}} \quad s_4 = s @ (600^\circ \text{C}, 15 \text{ MPa}) = 6,6776 \frac{\text{kJ}}{\text{kg} \cdot \text{K}}$$

$$4') s_{4'} = s_4 \quad \text{DA VALVOLA SU } 1 \text{ MPa} \quad P_{ES0} = \frac{s_4 - s @ 547}{s @ 2000 - s @ 547} = 0,85 \quad h_{4'} = (1 - P_{ES0}) h @ 547 + P_{ES0} \cdot h @ 2000 = 2806,98 \frac{\text{kJ}}{\text{kg}}$$

$$5) h_5 = h @ (600^\circ \text{C}, 1 \text{ MPa}) = 3697,9 \frac{\text{kJ}}{\text{kg}} \quad s_5 = s @ (600^\circ \text{C}, 1 \text{ MPa}) = 8,0290 \frac{\text{kJ}}{\text{kg} \cdot \text{K}}$$

$$6) s_6 = s_5 \quad \text{DA VALVOLA SU } P_2 = 5 \text{ kPa} \quad s @ 5 \text{ kPa} = 8,3953 \frac{\text{kJ}}{\text{kg} \cdot \text{K}} \quad x = \frac{s_6 - s @ 5 \text{ kPa}}{s @ 5 \text{ kPa} - s @ 547} = 0,95 \rightarrow h_6 = (1 - x) h @ 547 + x h @ 5 \text{ kPa} = 2440,32 \frac{\text{kJ}}{\text{kg}}$$

$$\dot{m}_R = \frac{Q_{cR}}{q_{cR}} \quad q_{cR} = (h_4 - h_2) + (h_5 - h_{4'}) = 4335,4 \frac{\text{kJ}}{\text{kg}} \quad \dot{m}_R = 25,11 \frac{\text{kg}}{\text{s}}$$

$$L_R = \dot{m}_R (L_{\text{TURBINA1}} + L_{\text{TURBINA2}} - L_{\text{COMP}}) = \dot{m} ((h_4 - h_{4'}) + (h_5 - h_6) - (h_2' - h_2)) = 50669,47 \text{ kW}$$

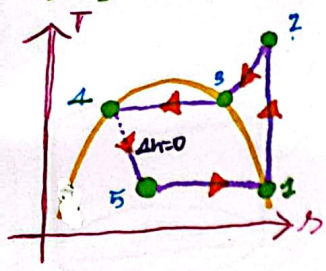
$$\eta_R = \frac{L_R}{Q_{cR}} = 46,54\%$$

$$\eta_{\text{CICLO COMBINATO}} = \frac{L_{SB} + L_R}{Q_{cR}} = 100\%$$



ES-18) R134a

$T_5 = T_1 = -28^\circ\text{C}$   $P_2 = 0,8 \text{ MPa}$   $\dot{Q}_F = 200 \cdot 10^3 \text{ W}$   $\epsilon_F?$   $\dot{m}?$   $\dot{Q}_C?$   $\Delta S_{4-5}?$



	$h \left[ \frac{\text{kJ}}{\text{kg}} \right]$
1	230,38
2	274,84
3	NON RIEMPIRE
4	93,42
5	93,42

$h_1 = h_{v@-28^\circ\text{C}} = 230,38 \frac{\text{kJ}}{\text{kg}}$   $h_2 = h_{v@-28^\circ\text{C}} = 230,38 \frac{\text{kJ}}{\text{kg}}$   
 $h_2 = h_1$   
 $h_2 = h(P_2, P_2)$   
 $\text{PES0} = \frac{h_2 - h_{v@-28^\circ\text{C}}}{h_{v@-28^\circ\text{C}} - h_{v@-28^\circ\text{C}}} = 0,11$   
 $h_2 = (1 - \text{PES0}) h_{v@-28^\circ\text{C}} + \text{PES0} h_{v@-28^\circ\text{C}} = 274,84 \frac{\text{kJ}}{\text{kg}}$   
 $h_4 = h_{v@0,8 \text{ MPa}} = 93,42 \frac{\text{kJ}}{\text{kg}}$   $h_4 = h_{v@0,8 \text{ MPa}} = 93,42 \frac{\text{kJ}}{\text{kg}}$   
 $h_5 = h_4$   
 $L = \dot{m} (h_2 - h_1)$   
 $h_5 = (1 - X) h_{v@-28^\circ\text{C}} + X \cdot h_{v@-28^\circ\text{C}} \rightarrow X_5 = 0,37$   
 $\rightarrow h_5 = (1 - X) h_{v@-28^\circ\text{C}} + X h_{v@-28^\circ\text{C}} = 0,382 \frac{\text{kJ}}{\text{kg}}$

$\epsilon_F = \frac{Q_F}{L}$

$\dot{Q}_C = \dot{m} (h_2 - h_4)$

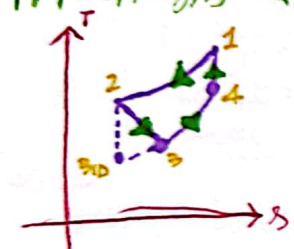
$\dot{Q}_F = \dot{m} (h_2 - h_5)$

$\Delta S_{4-5} = \dot{m} (s_5 - s_4)$

$\dot{Q}_F = \dot{m} = \frac{\dot{Q}_F}{h_2 - h_5} = 1,46 \text{ kg/s}$   $L = 64,91 \text{ kW}$   $\epsilon_F = 3,08$   $\dot{Q}_C = 264,87 \text{ kW}$   $\Delta S_{4-5} = 0,053 \frac{\text{kJ}}{\text{kg}^\circ\text{K}}$

ES-33) R12

$\dot{m} = 0,7 \text{ kg/s}$   $\dot{Q}_C?$   $\dot{Q}_F?$   $L?$   $\epsilon?$   $\text{ARIA}$

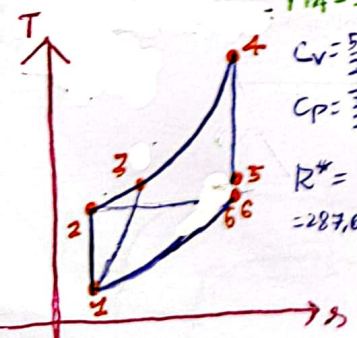


	$T [^\circ\text{C}]$	$P [\text{Pa}]$
1	370	$3 \cdot 10^5$
2	310	$3 \cdot 10^5$
3	217,75	$10^5$
3 ID	226,48	$10^5$
4	270	$10^5$

$M_m = 28,9 \cdot 10^{-3} \frac{\text{kg}}{\text{mol}}$   $\text{BIATONCO} \rightarrow C_p = \frac{7}{2} R^*$   $C_v = \frac{5}{2} R^*$   
 $1-2) \Delta P = 0$   
 $3-4) \Delta P = 0$   
 $2-3) \Delta h = 0 = c_p \ln \left( \frac{T_3}{T_2} \right) - R^* \ln \left( \frac{P_3}{P_2} \right)$   
 $\rightarrow T_{3ID} = 226,48^\circ\text{C}$

$\eta_c = \frac{T_{3ID} - T_2}{T_3 - T_2} \rightarrow T_3 = 217,75^\circ\text{C}$   
 $\dot{Q}_C = \dot{m} c_p (T_2 - T_3) = 4,24 \cdot 10^4 \text{ W}$   $\dot{Q}_F = \dot{m} c_p (T_4 - T_3) = 4,82 \cdot 10^4 \text{ W}$   $L = 6,95 \cdot 10^4 \text{ W}$   $\epsilon = \frac{\dot{Q}_C}{\dot{Q}_F - \dot{Q}_C} = 6,24$

$M_A = 28,9 \cdot 10^{-3} \frac{\text{kg}}{\text{mol}}$   $r = \frac{P_2}{P_1} = 6$   $w_{v2} = 150 \cdot 10^{-3} \text{ kg}$   $P_5?$   $L?$   $\dot{m}?$   $T_3?$   $\eta?$



	$P [\text{Pa}]$	$T [^\circ\text{C}]$
1	$100 \cdot 10^3$	300
2	$600 \cdot 10^3$	500,55
3	$600 \cdot 10^3$	958,94
4	$600 \cdot 10^3$	1600
5	$375,18 \cdot 10^3$	2399,45
6	$100 \cdot 10^3$	958,94

$P_2 = 6 \cdot P_1 = 600 \text{ kPa}$   $2-3-4$  ISOBARA  $5-6$  ISOBARA  
 $1-2) \Delta h = 0 = c_p \ln \left( \frac{T_2}{T_1} \right) - R^* \ln \left( \frac{P_2}{P_1} \right) \rightarrow T_2 = 500,55^\circ\text{C}$   
 $\dot{Q}_{\text{COMPRESSORE}} = L_{\text{URBANA-1}}$   
 $C_p (T_2 - T_1) = C_p (T_4 - T_5) \rightarrow T_5 = 1399,45^\circ\text{C}$   
 $4-5) \Delta h = 0 = \frac{7}{2} R^* \ln \left( \frac{T_4}{T_5} \right) - R^* \ln \left( \frac{P_4}{P_5} \right) \rightarrow P_5 = 375,48 \text{ kPa}$   
 $5-6) \Delta h = 0 = \frac{7}{2} R^* \ln \left( \frac{T_5}{T_6} \right) - R^* \ln \left( \frac{P_5}{P_6} \right) \rightarrow T_6 = 958,94^\circ\text{C}$

$\text{COMPONENTI IDEALI} \rightarrow \epsilon_r = 1 = \frac{T_3 - T_2}{T_6 - T_2} \rightarrow T_3 = T_6$   
 $L = L_{v2} + L_{v2} = C_p (T_5 - T_6) = 445,11 \frac{\text{kJ}}{\text{kg}}$

$w_{v2} = L_{v2} \cdot \dot{m} \rightarrow \dot{m} = \frac{w_{v2}}{L_{v2}} = 0,34 \text{ kg/s}$

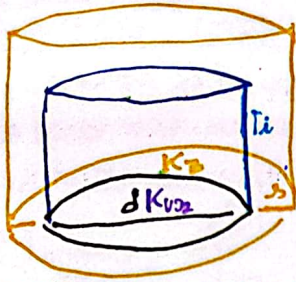
$\eta_{v2} =$

$= \frac{L'}{Q_C} = \frac{L'}{C_p (T_4 - T_3)} = \frac{T_5 - T_6}{T_4 - T_3} = 68,72\%$



ES.21

$d = 0,012m$   $\lambda = 0,001m$   $\alpha = 2 \cdot 10^8 \frac{W}{m^2K}$   $T_3 = 830,15^\circ K$   $K_{UO_2} = 2 \frac{W}{m^2K}$   $K_2 = 25 \frac{W}{m^2K}$   $T_{MAX}?$



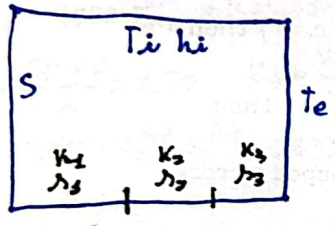
$$T_{UO_2} = \frac{\alpha}{4K_{UO_2}} \left( \frac{d}{2} - r \right) + T_i$$

$\dot{Q}$  COSTANTE  $\dot{Q}_{UO_2} = \dot{Q}_2$

$$\pi \left( \frac{d}{2} \right)^2 \alpha = \frac{K_2}{\lambda} \cdot \frac{2\pi r}{2} (T_i - T_3) \rightarrow T_i = 852^\circ K$$
$$T_{UO_2 (MAX)} = T_{UO_2} (r=0) = 1752^\circ K$$

ES.22

$S = 0,5m^2$   $\lambda_1 = 0,08m$   $K_1 = 3 \frac{W}{m^2K}$   $\lambda_2 = 0,03m$   $K_2 = 0,1 \frac{W}{m^2K}$   $\lambda_3 = 0,005m$   $K_3 = 20 \frac{W}{m^2K}$   
 $T_i = 850^\circ C = 1123,15^\circ K$   $T_e = 0^\circ C = 273,15^\circ K$   $h_i = 100 \frac{W}{m^2K}$   $h_e = 30 \frac{W}{m^2K}$   $R_{TOT}?$   $T_1?$   $T_2?$   $T_3?$   $\dot{Q}?$



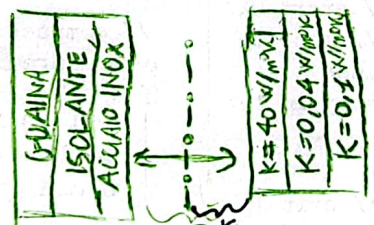
$$R_{TOT} = R_{conv,i} + R_{cond,1} + R_{cond,2} + R_{cond,3} + R_{conv,e}$$
$$R_{conv,i} = \frac{1}{h_i S}$$
  $R_{cond,1} = \frac{\lambda_1}{K_1 S}$   $R_{cond,2} = \frac{\lambda_2}{K_2 S}$   $R_{cond,3} = \frac{\lambda_3}{K_3 S}$   $R_{conv,e} = \frac{1}{h_e S}$ 
$$R_{TOT} = 0,740^\circ K/W$$
  $\dot{Q} = \frac{\Delta T_{TOT}}{R_{TOT}} = \frac{T_i - T_e}{R_{TOT}} = 1148,6W$   $\dot{Q}$  COSTANTE

$$\dot{Q} = \frac{T_i - T_1}{R_{cond,1}} \rightarrow T_1 = 827,03^\circ C$$
$$\dot{Q} = \frac{T_2 - T_3}{R_{cond,2}} \rightarrow T_2 = 76,62^\circ C$$
$$\dot{Q} = \frac{T_1 - T_2}{R_{cond,3}} \rightarrow T_3 = 76,62^\circ C$$

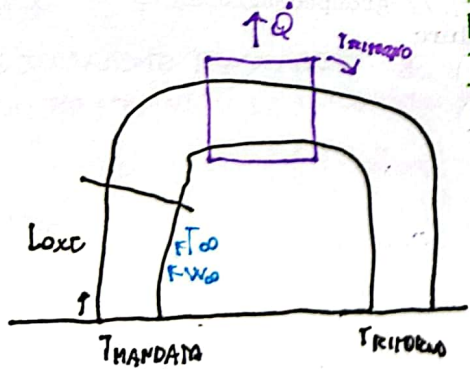
ES.23

$L = 1200m$   $L_{exc} = 100m$   $T_\infty = 0^\circ C = 273,15^\circ K$   $w_\infty = 10m/s$   $\dot{Q} = 20774W$   $T_{MANDATA} = 100^\circ C = 373,15^\circ K$   
 $T_{RITORNO} = 57^\circ C = 330,15^\circ K$

IPOTESI:  $\dot{V}_{H_2O} = 2m^3/s$   
• POTENZA DISPERSA NEL TRATTO ESTERNO TRASCURABILE



D?  
P (POTENZA DI RISCALDAMENTO)?



$$\dot{m} = \rho w_\infty A = \rho w_\infty \pi \left( \frac{D_o}{2} \right)^2 \rightarrow D_o = \sqrt{\frac{2 \dot{m}}{\rho w_\infty \pi}}$$
$$\dot{Q} = \dot{m} c_p \Delta T_{H_2O}$$
  $\dot{m} = \frac{\dot{Q}}{c_p \Delta T_{H_2O}}$ 
$$c_p = \frac{c_{p@330K} + c_{p@373K}}{2} = 4,2 \frac{kJ}{kg^\circ K}$$
  $\dot{m} = 115 kg/s$ 
$$\rho = \frac{\rho@330K + \rho@373K}{2} = 971 kg/m^3$$
  $D_o = \sqrt{\frac{4 \dot{m}}{\pi \rho w_\infty}} = 0,27m$

NEL MONDO REALE SI AVRA'  $D = 0,3m \rightarrow$  RICALCOLO  $w \Rightarrow w_{H_2O} = 4,67m/s$  ACCETTABILE

- $r_i = 0,15m$
- $r_{all} = 0,155m$
- $r_{0,3} = 0,255m$
- $r_{exc} = 0,275m$

PERDITA DI CARICO  $\Delta P = f \frac{L}{2D} \rho \cdot \dot{V} \frac{4V}{2D} \rightarrow P = f \frac{L}{2D} \dot{V} \frac{4V}{2D}$

COEFFICIENTE DI ATRITO

SI CERCA I PARAMETRI TERMOFISICI INTERMEDI A QUEL DATI, QUINDI 350°C CIRCA

$P = 0,184 Re^{-0,2}$  (MOTO TURBOLENTO)

$Re = \frac{wD}{\nu}$

$\nu = \frac{\gamma_{340K} + \gamma_{360K}}{2} = 0,138 \cdot 10^{-6} m^2/s$

$Re = 1,32 \cdot 10^6 > Re_{critico} = 2300$  (CONFERMA DI MOTO TURBOLENTO)  $\Rightarrow f = 0,011$   $P = 7056 W$

PER FINIRE, VERIFICARE L'IPOTESI CHE SIA TRASCURABILE RISPETTO A  $Q \approx 21 MW$

! VALORE TRA INGRESSO ED USCITA DEI 100m

$\dot{Q}_{DISPERSA_{100m}} = \frac{\Delta T}{R_{TOT}}$

$R_{TOT} = \frac{1}{h_i A_i} + \frac{\ln(\frac{r_{ext}}{r_i})}{2\pi L K_{acc}} + \frac{\ln(\frac{r_{ext}}{r_i})}{2\pi L K_{is}} + \frac{\ln(\frac{r_{ext}}{r_{ext}})}{2\pi L K_g} + \frac{1}{h_o A_o}$

ACQUA FLUSSO INTERNO AI TUBI

PARAMETRI TERMOFISICI A  $100^\circ C = 373,15^\circ K$

$C_p = 4220 \frac{J}{kg \cdot K}$   $\mu = 0,681 \frac{W}{m \cdot K}$   $\nu = 0,129 \cdot 10^{-6} m^2/s$   $\rho = 958 kg/m^3$

$Pr = \frac{C_p \gamma \rho}{\mu} = 1,72$   $Re = \frac{wD}{\nu} = 1,73 \cdot 10^6 > Re_{critico} \rightarrow$  MOTO TURBOLENTO

$Nu = 0,023 \cdot Re^{0,8} \cdot Pr^{0,4} = 2692$   $Nu = \frac{hD}{k} \rightarrow h_i = 6111 \frac{W}{m^2 \cdot K}$

ARIA FLUSSO ESTERNO. AL CILINDRO

IPOTESI: PARAMETRI TERMOFISICI CALCOLATI ALLA TEMPERATURA DI FILM DI  $280^\circ K$  ( $7^\circ C$ )

$k = 0,0246 \frac{W}{m \cdot K}$   $Pr = 0,717$   $\nu = 1,4 \cdot 10^{-5} m^2/s$

$Re = \frac{w_{ext} D_{ext}}{\nu}$   $D_{ext} = D + 2\delta_i + 2\delta_{acc} + 2\delta_{is} + 2\delta_{ext} = 0,55m \rightarrow Re = 3,9 \cdot 10^5 > Re_{critico} \rightarrow$  MOTO TURBOLENTO

$Nu = C \cdot Re^m \cdot Pr^{1/3}$  PER  $40.000 < Re < 400.000 \Rightarrow C = 0,027; m = 0,805$

$\rightarrow Nu = 765,6 \rightarrow h_e = \frac{Nu k}{D_{ext}} = 34,24 \frac{W}{m^2 \cdot K}$

$\Rightarrow R_{TOT} = 211 \cdot 10^{-4} \frac{K}{W}$

$\Delta T = 100^\circ K = T_{MANDATA} - T_{\infty}$

$\rightarrow \dot{Q} = 4739 W$

DECI SAMENTE TRASCURABILE

ESIN SUPERFICIE

ULTIMO STEP: VALORE LA TEMPERATURA DISPERSA

$\dot{Q}_{DISPERSA_{100m}} = h_e A_e (T_s - T_{\infty}) \rightarrow T_s = T_{\infty} + \frac{\dot{Q}}{h_e \pi D_{ext} L} = 0,8^\circ C \neq 7^\circ C$  IPOTIZZATO

(IN PRATICA, A QUESTO PUNTO RIADATTARE I PARAMETRI DELL'ARIA A  $7^\circ C$ )



$p = f \cdot (Lw2) / (2D)$

$P = v_p = (1/f) \cdot ((Lw2) / (2D))$

$\rightarrow P = f \cdot (Lw2) / (2D)$