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▼ Исходные данные
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Коэф. запаса:
                   safety = 1.3
                                                                        turbine = "ТВД"
Горючее:
             Fuel = "Керосин"
Высота движения (м): H_{11} = 0
                                                                  32.30
Массовый расход перед Т (кг/с):
                                                                                                                 32.30
                                                                               if turbine = "ТВД"
Массовый расход утечек Т (кг/с):
                                                              106.96.10
                                                                                                                  0.11
                                          G<sub>cooling</sub>
Массовый расход на охл Т (кг/с):
                                                                              if turbine = "ТНД"
                       N_T = 10^6 \cdot 14.893 if turbine = "TBД" = 14.893 \cdot 10^6
Мощность Т (Вт):
                                      15.181 if turbine = "ТНД"
                                        P_{\Gamma}^* = 10^3 \cdot | 2731.8 \text{ if turbine} = "ТВД" = 2731.8 \cdot 10^3
Полное давление перед Т (Па):
                                                       927.5 if turbine = "ТНД"
                                         T^*_{\Gamma} = \begin{vmatrix} 1773 & \text{if turbine} = "ТВД" = 1773.0 \\ 1368.9 & \text{if turbine} = "ТНД" \end{vmatrix}
Полная температура перед Т (К):
                                \alpha_{\rm ox} = 2.267 if turbine = "ТВД" = 2.267
Коэф. избытка воздуха в Т:
                                           2.493 if turbine = "ТНД"
                                                                  P^*_{\text{cooling}} = 10^3 \cdot | 2845.6 \text{ if turbine} = "ТВД" = 2845.6 \cdot 10^3
Полное давление отбора охлаждающего воздуха (К):
                                                                                        319.4 if turbine = "ТНД"
                                                                                   806.9 if turbine = "ТВД" = 806.9
Полная температура отбора охлаждающего воздуха (К):
                                                                   T*cooling =
                                                                                    418.2 if turbine = "ТНД"
                                                          \sigma_{\text{cooling}} = 0.97
Коэф. сохранения полного давления охлаждения:
Подогрев охл. от КС [К]:
                               \Delta T_{\text{охл.подогрев}} = 40
Газовая постоянная (Дж/кг/К): R_{\Gamma a3}(\alpha_{ox}, Fuel) = 288.5
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 $T_{\Pi, \text{ДО}\Pi} = 1373$ 

Допустимая температура Л (К):

Абс. скорость перед Т (м/с):

Абс. скорость после Т (м/с):

[1, c.15]

$$80 \le c_T \le 400 = 1$$

Лопаточный КПДТ:  $\eta_{\Pi} = 88\%$ 

$$\eta_{\rm JI} = 88\%$$

# $88\% \le \eta_{\rm JI} \le 95\% = 1$

Угол входа в Т:  $\alpha_{\Gamma} = 90$ .  $^{\circ}$ 

$$\alpha_{\Gamma} = 90^{\circ}$$

Окр. скорость Л последней ступени на ср. диаметре Т (м/с):

$$\begin{pmatrix} c_{\Gamma} \\ c_{T} \end{pmatrix} = \begin{pmatrix} 100 \\ 180 \end{pmatrix} \text{ if turbine} = "ТВД" = \begin{bmatrix} 1 \\ 1 \\ 100.0 \\ 2 \\ 180.0 \end{bmatrix}$$

$$\begin{pmatrix} 180 \\ 260 \end{pmatrix} \text{ if turbine} = "ТНД"$$

$$u_{\rm T} = \begin{bmatrix} 520 & \text{if turbine} = "ТВД" = 520.0 \\ 260 & \text{if turbine} = "ТНД" \end{bmatrix}$$



$$z = ORIGIN...N_r$$

 $P^*_{\text{cooling}} = P^*_{\text{cooling}} \cdot \sigma_{\text{cooling}} = 2760.2 \cdot 10^3$ Полное давление отбора охлаждающего воздуха (К):

 $T^*_{\text{cooling}} = T^*_{\text{cooling}} + \Delta T_{\text{охл.подогрев}} = 846.9$ Полная температура отбора охлаждающего воздуха (К):

 $G_{\Gamma} = G_{\Gamma} - G_{leak} = 32.2$ Массовый расход перед Т (кг/с):

 $G_{T} = G_{\Gamma} + G_{cooling} = 35.4$ Массовый расход после Т (кг/с):

Удельная работа T (Дж/кг):  $L^*_T = \frac{N_T}{\text{mean} \left(G_\Gamma, G_T\right)} = 440.4 \cdot 10^3$   $L^*_T \leq 550 \cdot 10^3 = 1$  Располагаемый теплоперепад в T (Дж/кг):  $H_T = \frac{L^*_T + 0.5c_T^{-2}}{\eta_{_{I\!I}}} = 518.9 \cdot 10^3$ 

$$\begin{vmatrix} \text{iteration} \\ k_{\Gamma} \\ P_{\Gamma} \\ T_{\Gamma} \end{vmatrix} = \begin{vmatrix} \text{iteration} = 0 \\ k_{\Gamma} = k_{a,\Pi} \left( \text{Cp}_{\Gamma a3} \left( \text{P*}_{\Gamma}, \text{T*}_{\Gamma}, \alpha_{\text{OX}}, \text{Fuel} \right), \text{R}_{\Gamma a3} \left( \alpha_{\text{OX}}, \text{Fuel} \right) \right) \\ \text{while } 1 > 0 \end{vmatrix} = \begin{vmatrix} \frac{1}{1} & \frac{1.0}{2} & \frac{1.3}{3} & \frac{3}{2705198.4} \\ \text{while } 1 > 0 \end{vmatrix} \\ \begin{vmatrix} \text{Iteration} = \text{iteration} + 1 \\ \text{Cp}_{\Gamma} = \frac{k_{\Gamma}}{k_{\Gamma} - 1} \cdot \text{R}_{\Gamma a3} \left( \alpha_{\text{OX}}, \text{Fuel} \right) \end{vmatrix} \\ T_{\Gamma} = T^*_{\Gamma} - \frac{c_{\Gamma}^2}{2 \cdot \text{Cp}_{\Gamma}} \\ k^*_{\Gamma} = k_{a,\Pi} \left( \text{Cp}_{\Gamma a3} \left( \text{P}_{\Gamma}, \text{T}_{\Gamma}, \alpha_{\text{OX}}, \text{Fuel} \right), \text{R}_{\Gamma a3} \left( \alpha_{\text{OX}}, \text{Fuel} \right) \right) \\ \text{if } \left| \text{eps} \left( \text{"rel"}, k_{\Gamma}, k_{\Gamma} \right) \right| \leq \text{epsilon} \\ k_{\Gamma} = k'_{\Gamma} \\ \text{break} \\ k_{\Gamma} = k'_{\Gamma} \\ \text{(iteration } k_{\Gamma} \mid \text{P}_{\Gamma} \mid \text{T}_{\Gamma} \right)^{T} \end{vmatrix}$$

Количество итераций: iteration = 1

Показатель адиабаты перед Т:  $\mathbf{k}_{\Gamma} = 1.283$ 

Статическое давление перед Т (Па):  $P_{\Gamma} = 2705.2 \cdot 10^3$ 

Статическая температура перед T(K):  $T_{\Gamma} = 1769.2$ 

Теплоем кость перед Т (Дж/кг/К):  $Cp_{\Gamma} = Cp_{\Gamma a3}(P_{\Gamma}, T_{\Gamma}, \alpha_{oX}, Fuel) = 1309$ 

$$\begin{vmatrix} \text{iteration} \\ k_T \\ P_T \\ T_T \end{vmatrix} = \begin{vmatrix} \text{iteration} = 0 \\ k_T = k_\Gamma \\ \text{while } 1 > 0 \\ \\ \text{iteration} = \text{iteration} + 1 \\ k_{Cp} = \text{mean}(k_\Gamma, k_T) \\ \\ Cp = \frac{k_{Cp}}{k_{Cp} - 1} \cdot R_{\text{ra3}}(\alpha_{\text{ox}}, \text{Fuel}) \\ \\ P_T = P^*_{\Gamma} \cdot \left(1 - \frac{H_T}{C_P \cdot T^*_{\Gamma}}\right)^{\frac{k_{Cp}}{k_{Cp} - 1}} \\ \\ T_T = T^*_{\Gamma} - \frac{H_T \cdot \eta_{\Pi}}{C_P} \\ k^*_T = k_{a,\Pi} \left(C_{P_{\text{ra3}}}(P_T, T_T, \alpha_{\text{ox}}, \text{Fuel}), R_{\text{ra3}}(\alpha_{\text{ox}}, \text{Fuel})\right) \\ \text{if } |\exp(\text{"rel"}, k_T, k_T'_T)| \leq \text{epsilon} \\ |k_T = k^*_T| \\ \text{break} \\ k_T = k^*_T \\ \text{(iteration } k_T \cdot P_T \cdot T_T)^T \\ \end{vmatrix}$$

Количество итераций: iteration = 1

Показатель адиабаты после Т:  $k_{\rm T} = 1.293$ 

Статическое давление после T (Па):  $P_{T} = 866.5 \cdot 10^{3}$   $P_{T} \geq P_{ATM}(H_{U}) = 1$ 

Статическая температура после T (K):  $T_T = 1424.1$ 

Теплоем кость после Т (Дж/кг/К):  $Cp_T = Cp_{\Gamma a3}(P_T, T_T, \alpha_{oX}, Fuel) = 1271.6$ 

Ср. показатель адиабаты Т: 
$$k = mean(k_T, k_T) = 1.288$$

Ср. теплоемкость Т (Дж/кг/К): Ср 
$$= \frac{k}{k-1} \cdot R_{\Gamma a3} (\alpha_{ox}, \text{Fuel}) = 1289.8$$

Степень понижения давления: 
$$\pi_{T} = \frac{P^{*}_{\Gamma}}{P_{T}} = 3.15$$

Удельный объём перед Т (м³/кг): 
$$\begin{pmatrix} v_{\Gamma} \\ v_{T} \end{pmatrix} = R_{\Gamma a 3} \Big( \alpha_{OX}, \text{Fuel} \Big) \cdot \begin{pmatrix} \frac{T_{\Gamma}}{P_{\Gamma}} \\ \frac{T_{T}}{P_{T}} \end{pmatrix} = \frac{1}{1 \quad 0.189}$$
 (С. 11)

Площадь кольцевого сечения перед T (м²): 
$$\begin{pmatrix} F_{\Gamma} \\ F_{T} \end{pmatrix} = \begin{pmatrix} \frac{G_{\Gamma} \cdot v_{\Gamma}}{c_{\Gamma}} \\ \frac{G_{\Gamma} \cdot v_{T}}{c_{T}} \end{pmatrix} = \begin{pmatrix} \frac{1}{1 & 60741} \\ \frac{1}{2 & 93341} \end{pmatrix} \cdot 10^{-6}$$

$$y_0 = 0.55$$

Коэф. использования скорости:

$$\mu_c = \text{mean}(0.7, 1) = 0.9$$

 $0.7 \le \mu_{\rm c} \le 1 = 1$ 

▼ Определение количества ступеней Т

Рекомендуемое количество ступеней:  $Z_{recomend} = 1$ 

Количество ступеней:  $Z = \begin{bmatrix} 1 & \text{if turbine} = \text{"ТВД"} = 1 \\ 4 & \text{if turbine} = \text{"ТНД"} \end{bmatrix}$ 

Дискретизация ступеней: i = 1..Z

Дискретизация сечений:  $ii = 1...2 \cdot Z + 1$ 

Плотность материала 
$$J(\kappa r/m^3)$$
: 
$$\rho\_blade_i = 7938 \quad \text{if material\_blade}_i = "BKHA-1B"$$
 
$$8390 \quad \text{if material\_blade}_i = "BKM7"$$
 
$$8760 \quad \text{if material\_blade}_i = "KC-36"$$
 NaN otherwise

$$\rho_{\text{blade}}^{\text{T}} = \boxed{\begin{array}{c|c} 1 \\ 1 \\ 7938 \end{array}}$$

$$\sigma\_blade\_long^T = \boxed{\begin{array}{c|c} 1 \\ 1 \\ 205 \end{array}} \cdot 10^6$$

Коэф. формы:  $k_n = 6.8$ 

Модуль Юнга I рода материала  $\Pi$  ( $\Pi$ a):  $E_blade = 210 \cdot 10^9$ 

Коэф. Пуассона материала  $\Pi$  ():  $\mu$  steel = 0.3

Мах частота вращения ротора на входе (об/мин):

$$\sqrt{\frac{\sigma_{\text{blade\_long}Z}}{\text{safety} \cdot k_{\text{n}} \cdot F_{\Gamma}}} = 19539$$

Мах частота вращения ротора на выходе (об/мин):

$$n_{\text{max}} = \sqrt{\frac{\sigma_{\text{blade\_long}Z}}{\text{safety} \cdot k_{\text{n}} \cdot F_{\text{T}}}} = 15762$$

Рекомендукмая ном. частота вращения (об/мин):

$$n = n_{\text{max}} \cdot 0.95 = 14974$$

$$_{\text{W}} = \begin{vmatrix} 15000 & \text{if turbine} = \text{"ТВД"} \\ 5300 & \text{if turbine} = \text{"ТНД"} \end{vmatrix} = 15000$$

Ном. частога вращения (рад/с):

$$\omega = \frac{2 \cdot \pi \cdot n}{60} = 1570.8$$

Ср. диаметр перед Т (м): 
$$\begin{pmatrix} D_{\Gamma.cp} \\ D_{T.cp} \end{pmatrix} = \frac{2}{\omega} \cdot \begin{pmatrix} u_T \\ u_T \end{pmatrix} = \begin{bmatrix} 1 \\ 1 & 662.1 \\ 2 & 662.1 \end{bmatrix} \cdot 10^{-3}$$

Длина Л первой ступени Т (м):

Длина Л последней ступени Т (м):

	$\begin{pmatrix} F_{\Gamma} \end{pmatrix}$				
$\begin{pmatrix} 1_{\Gamma} \end{pmatrix}$ 1	$D_{\Gamma,cp}$			1	– 3
$\begin{vmatrix} 1 \end{vmatrix} = \frac{1}{\pi}$	$\mathbf{F}_{-}$	=	1	29.20	.10
$\binom{1}{T}$	T		2	44.88	
	$\left(\mathrm{D}_{\mathrm{T.cp}}\right)$				

Диаметр периферии после 
$$T(M)$$
: 
$$\begin{pmatrix} D_{T.Пер} \\ D_{T.Кор} \end{pmatrix} = \begin{pmatrix} D_{T.cp} + l_T \\ D_{T.cp} - l_T \end{pmatrix} = \begin{bmatrix} 1 \\ 1 \\ 707.0 \\ 2 \\ 617.2 \end{bmatrix} \cdot 10^{-3}$$

$$\frac{l_{\rm T}}{D_{\rm T.cp}} = \frac{1}{14}$$

$$N_{cT_i} = \frac{N_T}{Z}$$

Вид проточной части:

("const", "кор", "cp", "пер", "доля от предыдушего диаметра периферии")

### Определение проточной части ОТ

Линейное распределение кольцевых площадей по сечениям:

$$\begin{array}{ll} F_{w} = & \text{ for } i \in 1...2Z + 1 \\ & F_{i} = \frac{F_{T} - F_{\Gamma}}{\text{st}(Z,3) - 1} \cdot i + \left(F_{\Gamma} - \frac{F_{T} - F_{\Gamma}}{\text{st}(Z,3) - 1}\right) \\ & \text{ for } i \in 1...Z \\ & \text{ for } a \in 2...3 \\ & F_{\text{st}(i,a)} = F_{\text{st}(i,a-1)} \quad \text{if } 3\Pi\Pi \Psi_{i,a-1} = \text{"const"} \\ & F \end{array}$$

$F^{T} =$		1	2	3	4	5	6	7	8	9	$10^{-6}$
	1	60741	60741	93341							

$$\begin{array}{ll} D = & \text{ for } i \in 2Z + 1 \\ & \text{ for } r \in 1..N_r \\ & D_{i,r} = & \left| \begin{array}{l} D_{T,KOP} & \text{ if } r = 1 \\ D_{T,cop} & \text{ if } r = av(N_r) \\ D_{T,ncp} & \text{ if } r = N_r \end{array} \right. \\ & \text{ for } i \in Z..1 \\ & \text{ for } r \in 1..N_r \\ & D_{st(i,a),r} = & \left| \begin{array}{l} \text{ if } 3\Pi\Pi^t I_{i,a} = "const" \\ D_{st(i,a+1),av(N_r)} - \frac{F_{st(i,a)}}{\pi \cdot D_{st(i,a+1),av(N_r)}} & \text{ if } r = 1 \\ D_{st(i,a+1),av(N_r)} + \frac{F_{st(i,a)}}{\pi \cdot D_{st(i,a+1),av(N_r)}} & \text{ if } r = N_r \end{array} \right. \\ & \text{ if } 3\Pi\Pi^t I_{i,a} = "kop" \\ & D_{st(i,a+1),1} & \text{ if } r = 1 \\ & \frac{1}{2} \left[ D_{st(i,a+1),1} + \sqrt{\left(D_{st(i,a+1),1}\right)^2 + \frac{4 \cdot F_{st(i,a)}}{\pi}} & \text{ if } r = N_r \end{array} \right. \\ & \text{ if } 3\Pi\Pi^t I_{i,a} = "cp" \\ & D_{st(i,a+1),av(N_r)} - \frac{F_{st(i,a)}}{\pi \cdot D_{st(i,a+1),av(N_r)}} & \text{ if } r = 1 \\ & D_{st(i,a+1),av(N_r)} - \frac{F_{st(i,a)}}{\pi \cdot D_{st(i,a+1),av(N_r)}} & \text{ if } r = 1 \\ & D_{st(i,a+1),av(N_r)} + \frac{F_{st(i,a)}}{\pi \cdot D_{st(i,a+1),av(N_r)}} & \text{ if } r = N_r \end{array} \\ & \text{ if } 3\Pi\Pi^t I_{i,a} = "cp" \\ & D_{st(i,a+1),av(N_r)} + \frac{F_{st(i,a)}}{\pi \cdot D_{st(i,a+1),av(N_r)}} & \text{ if } r = 1 \\ & D_{st(i,a+1),av(N_r)} + \frac{F_{st(i,a)}}{\pi \cdot D_{st(i,a+1),av(N_r)}} & \text{ if } r = N_r \end{array}$$

$$D^{T} = \begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 & 7 & 8 & 9 \\ 1 & 650.0 & 650.0 & 617.2 & & & & & \\ 2 & 678.5 & 678.5 & 662.1 & & & & & \\ 3 & 707.0 & 707.0 & 707.0 & & & & & & \end{bmatrix} \cdot 10^{-3}$$

$$R = \frac{D}{2}$$

$$\overline{d} = \begin{cases} \text{for } i \in 1..Z \\ \text{for } a \in 1..3 \end{cases}$$

$$\overline{d}_{st(i,a)} = \frac{D_{st(i,a),1}}{D_{st(i,a),N_r}}$$

$$\overline{d}$$

$\overline{d}^T =$		1	2	3	4	5	6	7	8	9
	1	0.9194	0.9194	0.8730						

$$\begin{array}{|c|c|} h = & \text{for } i \in 1..2Z + 1 \\ \\ h_i = \frac{F_i}{\pi \cdot D_{i, av(N_r)}} \\ \\ h \end{array}$$

$$\begin{split} & \left[ \frac{D}{st(i,a+1),N_r} \right]^{D-1-r} r \\ & \text{if } \left( 3\Pi\Pi \Psi_{i,\,a} \neq \text{"const"} \right) \wedge \left( 3\Pi\Pi \Psi_{i,\,a} \neq \text{"kop"} \right) \wedge \left( 3\Pi\Pi \Psi_{i,\,a} \neq \text{"cp"} \right) \wedge \left( 3\Pi\Pi \Psi_{i,\,a} \neq \text{"nep"} \right) \\ & \sqrt{\left( \frac{D}{st(i,a+1),N_r} \right)^2 - \frac{4 \cdot F_{st(i,\,a)}}{\pi}} \quad \text{if } r = 1 \\ & \frac{1}{2} \cdot \left[ \sqrt{\left( \frac{D_{st(i,a+1),N_r}}{str2num \left( 3\Pi\Pi \Psi_{i,\,a} \right)} \right)^2 - \frac{4 \cdot F_{st(i,\,a)}}{\pi}} + \frac{D_{st(i,a+1),N_r}}{str2num \left( 3\Pi\Pi \Psi_{i,\,a} \right)} \right] \quad \text{if } r = av \left( N_r \right) \\ & \frac{D_{st(i,a+1),N_r}}{str2num \left( 3\Pi\Pi \Psi_{i,\,a} \right)} \quad \text{if } r = N_r \end{split}$$

D

for 
$$i \in 1...2 \cdot Z + 1$$
  
for  $r \in 1...N_r$   

$$u_{i,r} = \frac{\pi \cdot D_{i,r} \cdot n}{60}$$

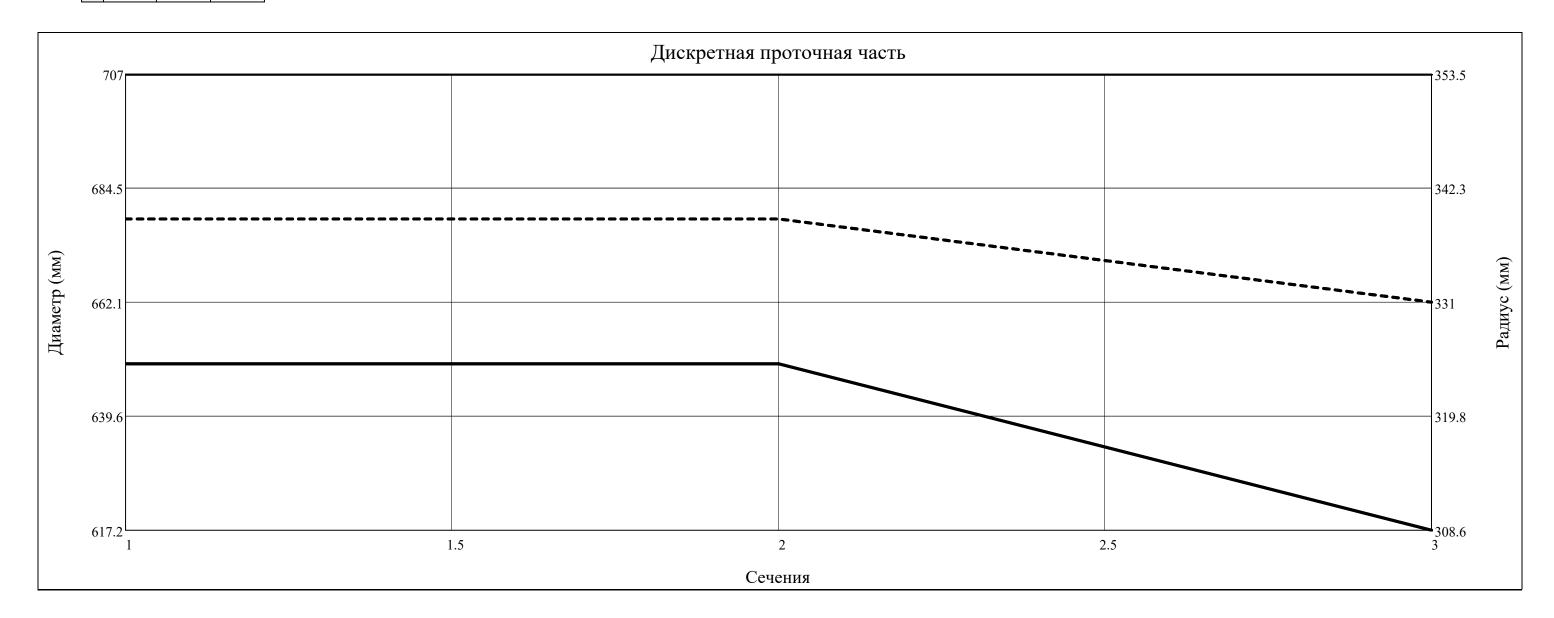
$$u$$

		1	2	3	4	5	6	7	8	9
$u^{T} =$	1	510.5	510.5	484.8						
-	2	532.9	532.9	520.0						
	3	555.2	555.2	555.2						

$$\overline{d}_{1} = 0.9194$$
  $\overline{d}_{1} \le 0.9 = 0$ 

$\overline{d}^T =$		1	2	3	4	5	6	7	8	9
	1	0.9194	0.9194	0.8730						

$$D^{T} = \begin{bmatrix} & 1 & 2 & 3 \\ 1 & 650.0 & 650.0 & 617.2 \\ 2 & 678.5 & 678.5 & 662.1 \\ 3 & 707.0 & 707.0 & 707.0 \end{bmatrix} \cdot 10^{-3}$$



## Осевая ширина Л СА и РК [1, с.183]:

$$\begin{pmatrix} B_{CA} \\ B_{PK} \end{pmatrix} = \begin{vmatrix} \text{for } i \in 1..Z \\ \begin{pmatrix} B_{CA}_i \\ B_{PK_i} \end{pmatrix} = \begin{pmatrix} \frac{D_{st(i,2),av(N_r)} - 0.25 \cdot h_{st(i,2)}}{16.4} \\ \frac{D_{st(i,3),av(N_r)}}{22} \end{vmatrix}$$

$$\begin{pmatrix} B_{CA} \\ B_{PK} \end{pmatrix}$$

$$stack(B_{CA}^{T}, B_{PK}^{T}) = \begin{bmatrix} 1 & 1 \\ 1 & 40.9 \\ 2 & 30.1 \end{bmatrix} \cdot 10^{-3}$$

Радиальный зазор (м):

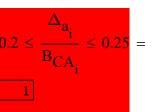
Осевой зазор (м):

(Лучше выбирать большее значение)

$$\begin{pmatrix} \Delta_r \\ \Delta_a \end{pmatrix} = \begin{vmatrix} \text{for } i \in 1...Z \\ \text{for } a \in 1...3 \\ \begin{vmatrix} \Delta_{r_{st(i,a)}} = 0.001 \cdot D_{st(i,a)}, N_r \\ \Delta_{a_i} = 0.25 \cdot B_{CA_i} \\ \begin{pmatrix} \Delta_r \\ \Delta_a \end{pmatrix}$$

$$\Delta_{\mathbf{r}}^{T} = \begin{bmatrix} 1 & 2 & 3 \\ 1 & 0.707 & 0.707 & 0.707 \end{bmatrix} \cdot 10^{-3}$$

$$\Delta_a^T = \boxed{\begin{array}{c|c} 1 \\ \hline 1 & 10.234 \end{array}} \cdot 10^-$$



$$stack\left(\gamma_{\prod \text{$V$}}, \gamma_{\prod \text{$V$}}, \gamma_{\prod \text{$V$}}, \gamma_{\prod \text{$V$}}, \gamma_{\prod \text{$V$}}\right) = \begin{vmatrix} 1 & 2 \\ 1 & 0.00 & -28.56 \\ 2 & -0.00 & 28.56 \\ 3 & 0.00 & 0.00 \end{vmatrix} . \circ$$

$\gamma_{\prod q}^T \le 20 \cdot \circ =$	1	1	2	
$\gamma_{\Pi \Psi}^{T} \leq 25 \cdot \circ = 0$		1	2	
'11'1	1	1	0	

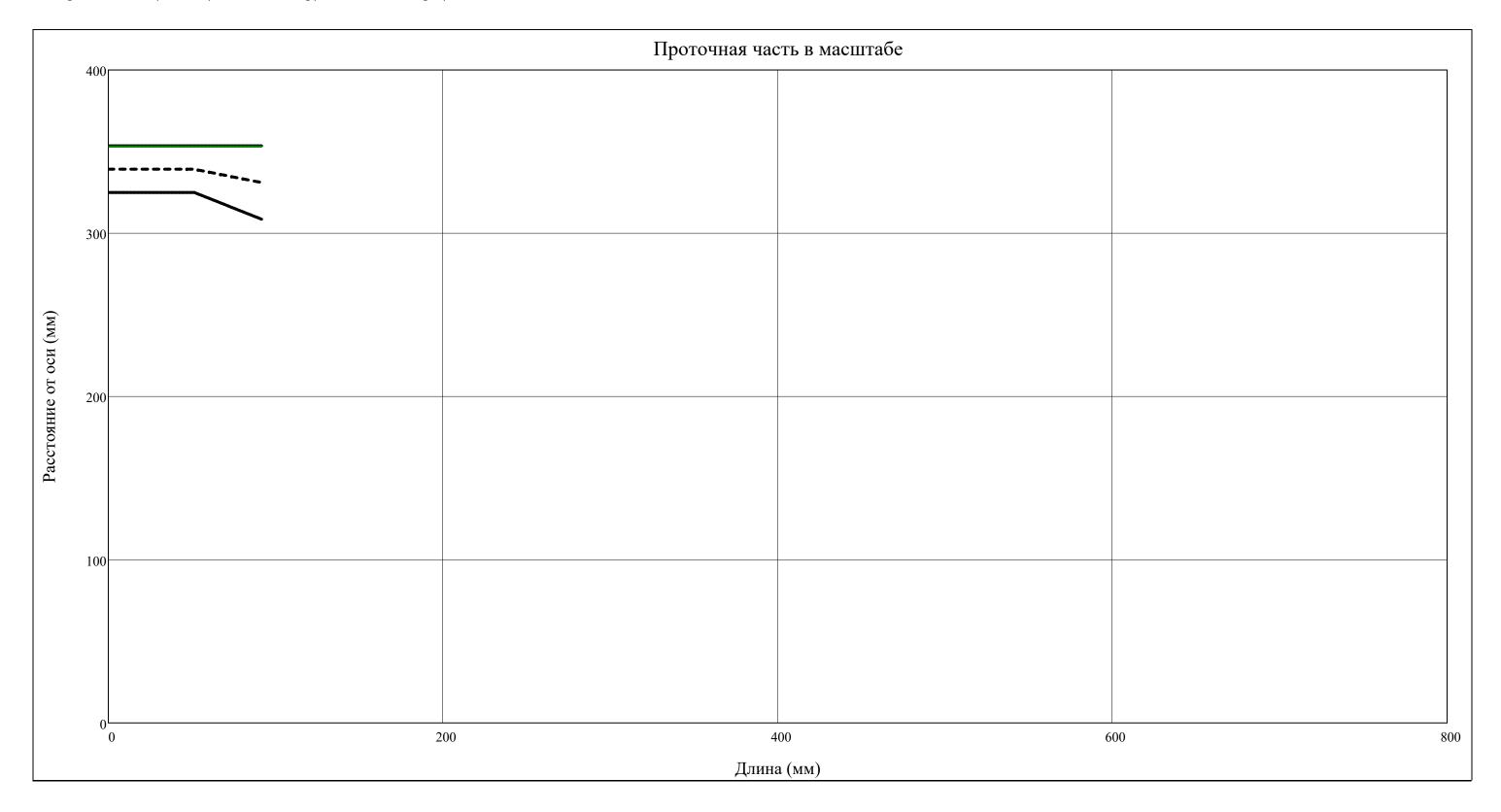
$\gamma_{\text{TIII}} = T > -12.$		1	2	
<sup>ү</sup> ПЧкор > -12. –	1	1	0	
$\gamma_{\text{TIII}} > -15^{\circ} =$		1	2	
<sup>1</sup> ПЧкор > -13. –	1	1	0	

$$\begin{pmatrix} x_{\Pi \mathbf{q}} \\ y_{\Pi \mathbf{q} nep} \\ y_{\Pi \mathbf{q} nep} \\ y_{\Pi \mathbf{q} nep} \\ y_{\Pi \mathbf{q} nep} \end{pmatrix} = \begin{vmatrix} c = 1 \\ x_{\Pi \mathbf{q}_c} = 0 \\ y_{\Pi \mathbf{q} nep_c} = y_{\Pi \mathbf{q} nep_c} - \Delta_{r_c} \\ y_{\Pi \mathbf{q} nep_c} = D_{st(c,1), nep} \\ y_{\Pi \mathbf{q} nep_c} = D_{st(c,1), nep} \\ y_{\Pi \mathbf{q} nep_c} = D_{st(c,1), 1} \\ \text{for } i \in 1 ... \mathbf{Z} \\ \begin{vmatrix} c = c + 1 \\ x_{\Pi \mathbf{q}_c} = x_{\Pi \mathbf{q}_{c-1}} + 0.5 \cdot \Delta_{\mathbf{a}_i} + B_{\mathbf{C}} \mathbf{A}_i + 0.5 \cdot \Delta_{\mathbf{a}_i} \\ y_{\Pi \mathbf{q} nep_c} \\ y_{\Pi \mathbf{q} nep_c} \end{vmatrix} = \begin{pmatrix} D_{st(i,2), N_r} \\ D_{st(i,2), 1} \end{pmatrix} \\ y_{J nep_c} = y_{\Pi \mathbf{q} nep_c} - \Delta_{r_i} \\ c = c + 1 \\ x_{\Pi \mathbf{q}_c} = x_{\Pi \mathbf{q}_{c-1}} + 0.5 \cdot \Delta_{\mathbf{a}_i} + B_{\mathbf{p}} \mathbf{K}_i + 0.5 \cdot \Delta_{\mathbf{a}_i} \\ \begin{pmatrix} y_{\Pi \mathbf{q} nep_c} \\ y_{\Pi \mathbf{q} nep_c} \\ y_{\Pi \mathbf{q} nep_c} \end{pmatrix} = \begin{pmatrix} D_{st(i+1,1), N_r} \\ D_{st(i+1,1), 1} \\ D_{st(i+1,1), 1} \end{pmatrix} \\ y_{J nep_c} = y_{\Pi \mathbf{q} nep_c} - \Delta_{r_i} \\ \begin{pmatrix} x_{\Pi \mathbf{q}} \\ y_{\Pi \mathbf{q} nep_c} \\ y_{\Pi \mathbf{q} nep_c} \\ y_{\Pi \mathbf{q} nep_c} \end{pmatrix} = \begin{pmatrix} T_{st(i+1,1), N_r} \\ T_{st(i+1,1), N_r} \\ T_{st(i+1,1), 1} \end{pmatrix} \\ y_{J nep_c} = y_{J \mathbf{q} nep_c} - \Delta_{r_i} \\ \begin{pmatrix} x_{\Pi \mathbf{q}} \\ y_{\Pi \mathbf{q} nep_c} \\ y_{J \mathbf{q} nep_c} \end{pmatrix} = \begin{pmatrix} T_{st(i+1,1), N_r} \\ T$$

Length = 
$$\sum_{i=1}^{Z} B_{CA_i} + \sum_{i=1}^{Z} \Delta_{a_i} + \sum_{i=1}^{Z} B_{PK_i} = 81.3 \cdot 10^{-3}$$

$$x = \min(x_{\Pi Y}), \min(x_{\Pi Y}) + \frac{\max(x_{\Pi Y}) - \min(x_{\Pi Y})}{N_{dis}} ... \max(x_{\Pi Y})$$

 $\begin{aligned} y_{\Pi \Pi nep}(l) &= linterp\Big(x_{\Pi \Pi}, 0.5 \cdot y_{\Pi \Pi nep}, l\Big) & y_{\Pi \Pi cp}(l) &= linterp\Big(x_{\Pi \Pi}, 0.5 \cdot y_{\Pi \Pi cp}, l\Big) & y_{\Pi \Pi kop}(l) &= linterp\Big(x_{\Pi \Pi}, 0.5 \cdot y_{\Pi \Pi kop}, l\Big) \\ y_{\Pi nep}(l) &= interp\Big(cspline\Big(x_{\Pi \Pi}, 0.5 \cdot y_{\Pi nep}\Big), x_{\Pi \Pi}, 0.5 \cdot y_{\Pi nep}, l\Big) \end{aligned}$ 

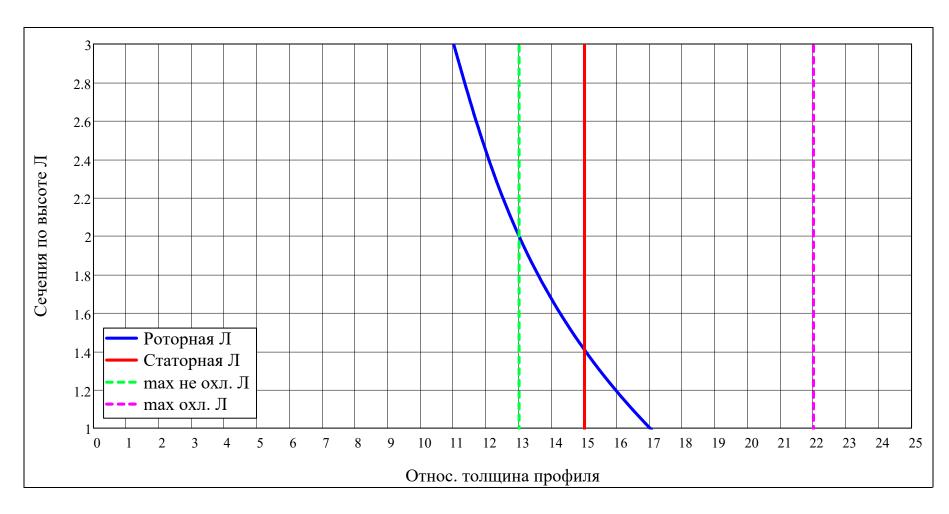


Относ. толщины ЛРК и СА:

$$\overline{c}_{stator.}(r) = \begin{bmatrix} 1 \\ av(N_r) \\ N_r \end{bmatrix}, \begin{bmatrix} 15 \\ 15 \\ 15 \end{bmatrix}, \begin{bmatrix} 1 \\ av(N_r) \\ N_r \end{bmatrix}, \begin{bmatrix} 15 \\ 15 \\ 15 \end{bmatrix}, w, r \end{bmatrix} \text{ if } T_{JI.JQOII} < T^*_{\Gamma}$$

$$\begin{bmatrix} 1 \\ av(N_r) \\ N_r \end{bmatrix}, \begin{bmatrix} 7 \\ 9 \\ N_r \end{bmatrix}, \begin{bmatrix} 7 \\ 9 \\ 11 \end{bmatrix}, \begin{bmatrix} 7 \\ 9 \\ N_r \end{bmatrix}, \begin{bmatrix} 7 \\ 9 \\ 11 \end{bmatrix}, w, r \end{bmatrix} \text{ otherwise}$$

$$\overline{c}_{rotor.}(r) = \begin{bmatrix} 1 \\ av(N_r) \\ N_r \end{bmatrix}, \begin{bmatrix} 17 \\ 13 \\ 11 \end{bmatrix}, \begin{bmatrix} 17 \\ av(N_r) \\ N_r \end{bmatrix}, \begin{bmatrix} 17 \\ 13 \\ 11 \end{bmatrix}$$



$$\begin{bmatrix}
\overline{c}_{stator} \\
\overline{c}_{rotor}
\end{bmatrix} = \begin{bmatrix}
for & i \in 1...Z \\
for & r \in 1...N_r
\end{bmatrix}$$

$$\begin{bmatrix}
\overline{c}_{stator} \\
\overline{c}_{rotor} \\
\overline{c}_{rotor} \\
\overline{c}_{rotor}
\end{bmatrix} = \begin{bmatrix}
\overline{c}_{stator} \\
\overline{c}_{rotor} \\
\overline{c}_{rotor}
\end{bmatrix}$$

$$\begin{bmatrix}
\overline{c}_{stator} \\
\overline{c}_{rotor}
\end{bmatrix}$$

$$\frac{1}{c_{\text{stator}}}^{T} = \begin{vmatrix}
 & 1 \\
1 & 15.00 \\
2 & 15.00 \\
3 & 15.00
\end{vmatrix} .\%$$

$$\frac{T}{c_{rotor}} = \begin{vmatrix}
 & 1 & \\
1 & 17.00 \\
2 & 13.00 \\
3 & 11.00
\end{vmatrix}$$

$$\frac{T}{r} = \begin{bmatrix}
 & 1 & \\
 & 1 & 6.000 \\
 & 2 & 6.000 \\
 & 3 & 6.000
\end{bmatrix}$$

$$\frac{T}{r\_outlet_{stator}}^{T} = \begin{bmatrix} & 1 \\ 1 & 3.000 \\ 2 & 3.000 \\ \hline 3 & 3.000 \end{bmatrix} .\%$$

$$\frac{T}{r\_inlet_{rotor}}^{T} = \begin{bmatrix}
 & 1 \\
 & 1 & 5.950 \\
 & 2 & 4.550 \\
 & 3 & 3.850
\end{bmatrix}$$
.%

$$\frac{T}{r\_outlet_{rotor}}^{T} = \begin{bmatrix}
 & 1 \\
1 & 2.550 \\
2 & 1.950 \\
3 & 1.650
\end{bmatrix} .\%$$

▲ Относ. толщины и радиусы профилей

$$R_{L.ep} = \begin{pmatrix} 0.16 & \text{if turbine} = "ТВД" & 0.15 & 0.18 & 0.185 & 0.5 & 0.5 \\ 0.13 & \text{otherwise} \end{pmatrix}^T$$

## ▼ Поступенчатый расчет ОТ

<i>(:, ::</i>	•• • •
(iteration <sub>CA</sub>	
<u>k</u> .	$R_{\mathrm{L}}$
H* <sub>cT</sub>	$H_{cT}$
H <sub>stator</sub>	$H_{rotor}$
с <sub>ад</sub>	$w_{a\mu}$
P*	P
T*	<u>T</u>
<u>G</u>	v
ρ*	ρ
	$\alpha_{ox}$
α	β
$\epsilon_{ m stator}$	$\varepsilon_{ m rotor}$
$\theta_{\mathrm{CA}}$	$\theta_{ ext{PK}}$
g <sub>охл</sub> СА	g <sub>охл</sub> РК
a* <sub>c</sub>	a* <sub>W</sub>
Тад	т <sub>ад</sub>
Р* <sub>W</sub>	ад Т* <sub>W</sub>
а <sub>зв</sub>	a <sub>3B</sub>
u	u
<u> </u>	c c
c <sub>a</sub>	c <sub>u</sub>
W	W
w <sub>a</sub>	w <sub>u</sub>
$\lambda_{\rm c}$	$M_c$
$\lambda_{ m w}$	$M_{W}$
<sup>v</sup> stator	$v_{ m rotor}$

chordstator	chord <sub>rotor</sub>	$      _{if i = 1} $
<del>t</del> оптCA	<del>-</del> t <sub>опт</sub> РК	
t <sub>stator</sub>	t <sub>rotor</sub>	$\alpha_{\text{oX}_{\text{st}(i,1)}} = \alpha_{\text{oX}}$
Z <sub>stator</sub>	Z <sub>rotor</sub>	$\begin{vmatrix} k_{st(i,1),r} = k_{\Gamma} \\ p_{*,(i,1),r} = p_{*} \end{vmatrix}$
$\overline{v}_{ m stator}$	$\overline{v}_{ m rotor}$	$P^*_{st(i,1),r} = P^*_{\Gamma}$ $P^*_{rt(i,1),r} = P^*_{rt(i,1),r}$
ξ <sub>TpCA</sub>	ξ <sub>τpPK</sub>	$P^*_{W_{St(i,1),r}} = 0$
ξкрСА	ξ <sub>кр</sub> РК	$P_{st(i,1),r} = P_{\Gamma}$
ξ <sub>ReCA</sub>	ξ <sub>RePK</sub>	$T^*_{st(i,1),r} = T^*_{\Gamma}$
ξ <sub>λCA</sub>	$\xi_{\lambda PK}$	$T^*_{W_{\operatorname{st}(i,1),r}} = 0$
$\xi_{\Pi p C A}$	$\xi_{\Pi p P K}$	$T_{st(i,1),r} = T_{\Gamma}$
$\xi_{\mathrm{BTCA}}$	ξ <sub>BT</sub> PK	$v_{st(i,1),r} = \frac{R_{\Gamma a3}(\alpha_{oX_{st(i,1)}}, Fuel) \cdot T_{st(i,1),r}}{P_{st(i,1),r}}$
$\xi_{ m TДCA}$	ξ <sub>тдРК</sub>	1 1 1
ξсмСА	ξсмРК	$G_{st(i,1)} = G_{\Gamma}$
$\xi_{\Delta r}$	$\xi_{ m BMX}$	$c_{st(i,1),r} = c_{\Gamma}$
ξ <sub>тр.в</sub>	ξ <sub>тр.в</sub>	$\alpha_{st(i,1),r} = \alpha_{\Gamma}$
L <sub>ct</sub>	Lu <sub>CT</sub>	$\begin{bmatrix} c_{\mathbf{u}_{st(i,1),r}} \\ c_{\mathbf{a}_{st(i,1),r}} \end{bmatrix} = c_{st(i,1),r} \cdot \begin{pmatrix} \cos(\alpha_{st(i,1),r}) \\ \sin(\alpha_{st(i,1),r}) \end{pmatrix}$
$\eta_{ ext{мощь}}$	$\eta_{ m JO\Pi}$	$\left[ \left( c_{a_{st(i,1),r}} \right)^{-c_{st(i,1),r}} \left( \sin(\alpha_{st(i,1),r}) \right) \right]$
$\eta^*_{ m cT}$	$\eta^*_{\mathrm{cT}}$	$\mathbf{w}_{\mathrm{st}(i,1),r} = 0$
$\eta_{u1}$	$\eta_{u2}$	$\left( \sqrt{k_{st(i,1),r} \cdot R_{\Gamma a3} \left( \alpha_{oX_{st(i,1)}}, Fuel \right) \cdot T_{st(i,1),r}} \right)$
ξCA	ξ <sub>PK</sub>	$ \begin{array}{c c}  & a_{3B} \\ \hline  & 2 \cdot k_{st(i,1),r} \\ \hline  & 7 \cdot k_{st(i,1),r} \end{array} $
Lu <sub>нагрузка</sub>	Lu <sub>нагрузка</sub>	
		$ \begin{pmatrix} a_{3B_{st(i,1),r}} \\ a^*c_{st(i,1),r} \\ a^*w_{st(i,1),r} \end{pmatrix} = \begin{pmatrix} \sqrt{k_{st(i,1),r} \cdot R_{\Gamma a3} \left(\alpha_{ox_{st(i,1)}}, Fuel \right) \cdot T_{st(i,1),r}} \\ \sqrt{\frac{2 \cdot k_{st(i,1),r}}{1 + k_{st(i,1),r}} \cdot R_{\Gamma a3} \left(\alpha_{ox_{st(i,1)}}, Fuel \right) \cdot T^*_{st(i,1),r}} \\ \sqrt{\frac{2 \cdot k_{st(i,1),r}}{1 + k_{st(i,1),r}} \cdot R_{\Gamma a3} \left(\alpha_{ox_{st(i,1)}}, Fuel \right) \cdot T^*_{w_{st(i,1),r}}} \\ \end{pmatrix} $
		$\begin{pmatrix} \lambda_{c_{st(i,1),r}} \\ \lambda_{w_{st(i,1),r}} \end{pmatrix} = \begin{pmatrix} \frac{c_{st(i,1),r}}{a^*c_{st(i,1),r}} \\ 0 \end{pmatrix}$ $\begin{pmatrix} M_{c_{st(i,1),r}} \\ M_{w_{st(i,1),r}} \end{pmatrix} = \frac{1}{a_{3B_{st(i,1),r}}} \begin{pmatrix} c_{st(i,1),r} \\ w_{st(i,1),r} \end{pmatrix}$
		$\begin{bmatrix} & st(1,1),r \end{pmatrix} & 0 & \\ & & $
		$\left  \left  \left  \left  \frac{M_{c}}{st(i,1),r} \right  \right  = \frac{1}{cst(i,1),r} \right $
		$\left[ \begin{array}{c} M_{W_{st(i,1),r}} & a_{3B_{st(i,1),r}} \\ \end{array} \right] \left( \begin{array}{c} W_{st(i,1),r} \end{array} \right)$
		$iteration_{CT_i} = 0$
		while $1 > 0$ $  \text{iteration}_{CT_{i}} = \text{iteration}_{CT_{i}} + 1$
		$iteration_{CT_{i}} = iteration_{CT_{i}} + 1$

$$\begin{aligned} & \operatorname{trace} \left( \operatorname{concat} \left( " & \operatorname{iteration.cr} = " , \operatorname{num2str} \left( \operatorname{iteration}_{\operatorname{CT}_i} \right) \right) \right) \\ & H_{\operatorname{CT}_i} = N_{\operatorname{CT}_i} \cdot \left| \begin{array}{c} \frac{1}{G_{\operatorname{St}(i,1)} \cdot 0.9} & \operatorname{if} \left( \operatorname{iteration}_{\operatorname{CT}_i} = 1 \right) \\ & \frac{1}{\operatorname{mean} \left( G_{\operatorname{St}(i,2)} \cdot , G_{\operatorname{St}(i,3)} \right) \cdot \eta_{\operatorname{Monibs}_i}} & \operatorname{otherwise} \\ & R_{L_{i,r}} = R_{L.\operatorname{cp}_i} \\ & c_{\operatorname{all}_{\operatorname{St}(i,1),r}} = \sqrt{2 \cdot H_{\operatorname{CT}_i}} \\ & H_{\operatorname{Stator}_i} = H_{\operatorname{CT}_i} \cdot \left( 1 - R_{L_{i,r}} \right) \\ & c_{\operatorname{all}_{\operatorname{St}(i,2),r}} = \sqrt{2 \cdot H_{\operatorname{Stator}_i}} \\ & \overline{v}_{\operatorname{Stator}_i} = 1 \\ & \operatorname{iteration}_{\operatorname{CA}_i} = 0 \\ & \operatorname{while} \ 1 > 0 \\ & \operatorname{iteration}_{\operatorname{CA}_i} = \operatorname{iteration}_{\operatorname{CA}_i} + 1 \\ & \operatorname{trace} \left( \operatorname{concat} \left( " & \operatorname{iteration.CA} - " , \operatorname{num2str} \left( \operatorname{iteration}_{\operatorname{CA}_i} \right) \right) \right) \\ & c_{\operatorname{St}(i,2),r} = \overline{v}_{\operatorname{Stator}_i} \cdot \overline{ca}_{\operatorname{Mst}(i,2),r} \\ & \theta_{\operatorname{CA}_i} = \theta_{\operatorname{TMY}} \sigma_{\operatorname{MHA}} \left( T^*_{\operatorname{St}(i,1)} , r, T^*_{\operatorname{cooling}} \cdot T_{\operatorname{II},\operatorname{Toni}} \right) \\ & g_{\operatorname{OX_ICA}_i} = \frac{0.035 \cdot \theta_{\operatorname{CA}_i}}{0 \cdot \operatorname{otherwise}} \cdot \frac{0.035 \cdot \theta_{\operatorname{CA}_i}}{1 - \theta_{\operatorname{CA}_i}} \geq 0 \\ & g_{\operatorname{OX_ICA}_i} = \frac{0}{0 \cdot \operatorname{otherwise}} \\ & G_{\operatorname{St}(i,2)} = G_{\operatorname{St}(i,1)} \cdot \left( 1 + g_{\operatorname{OX_ICA}_i} \right) \\ & \alpha_{\operatorname{OX}_{\operatorname{St}(i,2)}} = \alpha_{\operatorname{OX}_{\operatorname{St}(i,1)}} + g_{\operatorname{OX_ICA}_i} \\ & \alpha_{\operatorname{OKCA}_i} = \operatorname{mean} \left( \alpha_{\operatorname{OX}_{\operatorname{St}(i,1)}} , r, \operatorname{ks_{\operatorname{St}(i,2)}} \right) \\ & k_{\operatorname{St}(i,2), r} = k_{\operatorname{St}(i,1), r} \\ & \text{while} \ 1 > 0 \\ & | k_{\operatorname{CA}_i} = \operatorname{mean} \left( k_{\operatorname{St}(i,1), r}, k_{\operatorname{St}(i,2), r} \right) \\ & H_{\operatorname{Stator}_i} \\ & \frac{k_{\operatorname{CA}_i}}{k_{\operatorname{CA}_i}} \cdot \frac{k_{\operatorname{CA}_i}}{k_{\operatorname{CA}_i}} \cdot \frac{k_{\operatorname{CA}_i}}{k_{\operatorname{CA}_i}} \right) , \\ & f_{\operatorname{CA}_i} = \operatorname{mean} \left( k_{\operatorname{St}(i,1), r}, r, k_{\operatorname{St}(i,2), r} \right) \\ & f_{\operatorname{CA}_i} = \operatorname{mean} \left( k_{\operatorname{St}(i,1), r}, r, k_{\operatorname{St}(i,2), r} \right) \\ & \frac{k_{\operatorname{CA}_i}}{k_{\operatorname{CA}_i}} \cdot \frac{k_{\operatorname{CA}_i}}{k_{\operatorname{CA}_i}} \cdot \frac{k_{\operatorname{CA}_i}}{k_{\operatorname{CA}_i}} \right) , \\ & f_{\operatorname{CA}_i} = \operatorname{mean} \left( k_{\operatorname{CA}_i} \cdot f_{\operatorname{CA}_i} \right) \right) \\ & f_{\operatorname{CA}_i} = \operatorname{mean} \left( k_{\operatorname{CA}_i} \cdot f_{\operatorname{CA}_i} \right) \right)$$

$$\begin{aligned} & P_{\text{SG}(1,2),r} = P^{\text{S}_{\text{G}(1,1),r}} \left( \frac{\Gamma_{\text{Ta}_{\text{GI}(1,1),r}}^{-1}}{\Gamma^{\text{E}_{\text{GI}(1,1),r}}} \right)^{\frac{1}{\text{E}_{\text{CA}_{1}}} - 1} \\ & T_{\text{BG}(2,1,r),r} = P^{\text{E}_{\text{BG}(1,1),r}} \left( \frac{\Gamma_{\text{Ta}_{\text{GI}(1,1),r}}^{-1}}{\frac{k_{\text{CA}_{1}}}{k_{\text{CA}_{1}}} - 1} R_{\text{Ta},\text{Cep}} \left( \alpha_{\text{Na}_{\text{BG}(1,2)}},\text{Fuel} \right) \right) \\ & C_{P_{2}} = C_{P_{\text{TBS}}} \left[ P_{\text{SG}(1,2),r}, T_{\text{AI}_{1,1},r}, \alpha_{\text{ON}_{\text{AI}_{1,2}},r}, \text{Fuel} \right) \\ & k^{-1} + k_{\text{AB}} \left( C_{\text{PA}_{1}}, P_{\text{EB}_{1,1},r}, x_{\text{PB}_{1,1}} \right) \\ & k^{-1} + k_{\text{AB}} \left( C_{\text{PA}_{1}}, P_{\text{EB}_{1,1},r}, x_{\text{PB}_{1,1}} \right) \\ & k^{-1} + k_{\text{AB}_{1}} \left( C_{\text{PA}_{1}}, P_{\text{EB}_{1,1},r}, x_{\text{PB}_{1,1}} \right) \\ & k^{-1} + k_{\text{AB}_{1,1}} \left( C_{\text{PA}_{1,1},r}, x_{\text{PB}_{1,1}} \right) \\ & k^{-1} + k_{\text{AB}_{1,1},2}, r = k^{-1} \\ & \text{Prea}_{\text{AB}_{\text{AB}_{1,1},2}, r} \\ & P^{\text{Ea}_{\text{AB}_{\text{AB}_{1,1},2}, r}} = P_{\text{EB}_{1,1,2}}, r^{-1} \frac{\Gamma_{\text{EB}_{1,1},2}}{\Gamma_{\text{EB}_{1,1},2}, r^{-1}} \frac{k_{\text{EB}_{1,1,2},r}, \alpha_{\text{PA}_{1,1,1},r}}{k_{\text{EB}_{1,1,2},r}, \alpha_{\text{PA}_{1,1,1},r}, \alpha_{\text{PA}_{1,1,1},r}, \alpha_{\text{PA}_{1,1,1},r}} \\ & P^{\text{E}_{\text{AB}_{\text{EB}_{1,1,2},r}, r}} = P_{\text{EB}_{1,1,2}, r}, r^{-1} \frac{\left( c_{\text{EB}_{1,1,2},r}, r^{-1} r_{\text{EB}_{1,1,2},r}, r^{-1} r_{\text{EB}_{1,1,2},r}, r^{-1} r_{\text{EB}_{1,1,2},r}, r^{-1} r_{\text{EB}_{1,1,2},r}, r^{-1}} \frac{\left( c_{\text{EB}_{1,1,2},r}, r^{-1} r_{\text{EB}_{1,1,2},r}, r^{-1}$$

$$\left\{ \begin{array}{l} \left\{ \begin{array}{l} \left\{ ^{c_{1}} c_{1} c_{1} c_{2} \cdot c_{2} c_{1} (z_{2}) \cdot r \\ c_{3} c_{1} (z_{2}) \cdot r \\ c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right\} = c_{3} (i,2) \cdot r \cdot \left\{ \begin{array}{l} c_{3} c_{1} c_{2} c_{3} c_{1} c_{2} c_{3} c_{1} \\ c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right\} = c_{3} (i,2) \cdot r \cdot \left\{ \begin{array}{l} c_{3} c_{1} c_{2} c_{3} c_{1} c_{2} c_{3} c_{1} \\ c_{3} c_{1} c_{2} c_{3} c_{3} c_{3} c_{3} c_{3} c_{3} \\ \end{array} \right\} \\ \left\{ \begin{array}{l} w_{3} c_{1} (z_{2}) \cdot r \\ w_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right\} = \left\{ \begin{array}{l} c_{3} c_{1} (z_{2}) \cdot r \\ c_{3} c_{3} c_{3} (z_{2}) \cdot r \\ \end{array} \right\} \\ \left\{ \begin{array}{l} \left( \begin{array}{l} w_{3} c_{1} (z_{2}) \cdot r \\ w_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \end{array} \right\} \\ \left\{ \begin{array}{l} \left( \begin{array}{l} w_{3} c_{1} (z_{2}) \cdot r \\ w_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ = \left\{ \begin{array}{l} \left( \begin{array}{l} c_{3} c_{1} (z_{2}) \cdot r \\ c_{3} c_{3} c_{3} (z_{2}) \cdot r \\ \end{array} \right) \\ \end{array} \right\} \\ \left\{ \begin{array}{l} \left( \begin{array}{l} c_{3} c_{1} (z_{2}) \cdot r \\ c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{1} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c_{3} c_{3} c_{3} c_{3} (z_{2}) \cdot r \\ \end{array} \right) \\ \left( \begin{array}{l} c_{3} c$$

$$\begin{vmatrix} \operatorname{cril} \left( \frac{\operatorname{common}_{\operatorname{cont}}(x_{1}, x_{1}, x_{1}, x_{2}, x_{1}, x_{1}, x_{2})}{\operatorname{ContCa}_{\operatorname{cont}}(x_{1}, x_{2}, x_{2}, x_{2}, x_{2}, x_{2})} + \operatorname{1 otherwise} \\ \\ \operatorname{for } r = 1. N_{\Gamma} \\ \operatorname{Istanor}_{i, t} = \frac{\pi \cdot \operatorname{mean} \left( \operatorname{D}_{\operatorname{st}(i, 1), t}, \operatorname{D}_{\operatorname{st}(i, 2), t} \right)}{\mathcal{L}_{\operatorname{stator}_{i, t}}} \\ \\ \operatorname{SpCA}_{i} = \operatorname{Specime}^{\left( \operatorname{cont}(i, 1), t, \operatorname{Spec}_{i, 2}, x_{2}, x_{2} \right)} \\ \\ \operatorname{SpCA}_{i} = \operatorname{Specime}^{\left( \operatorname{cont}(i, 1), t, \operatorname{Spec}_{i, 2}, x_{2}, x_{2} \right)} \\ \\ \operatorname{SpCA}_{i} = \operatorname{Specime}^{\left( \operatorname{cont}(i, 2), t, \operatorname{ContCa}_{\operatorname{stor}_{i, t}, t}, \operatorname{ContCa}_{\operatorname{stor}_{\operatorname{stor}_{i, t}, t}, t} \\ \\ \operatorname{SpCA}_{i} = \operatorname{Spec}_{\operatorname{contCantContC}_{\operatorname{stor}_{i, t}, t}, \operatorname{ContCa}_{\operatorname{stor}_{i, t}, t}, \operatorname{ContCa}_{\operatorname{stor}_{i, t}, t}, \operatorname{ContCa}_{\operatorname{stor}_{\operatorname{stor}_{i, t}, t}, t}, \operatorname{ContCa}_{\operatorname{stor}_{\operatorname{stor}_{i, t}, t}, t}, \operatorname{ContCa}_{\operatorname{stor}_{\operatorname{stor}_{i, t}, t}, t}, \operatorname{ContCa}_{\operatorname{stor}_{\operatorname{stor}_{i, t}, t}, t}, \operatorname{ContCa}_{\operatorname{stor$$

$$\begin{split} &\theta_{PK_{i}} = \theta_{PTIYOHHA} \Big( T^{*}w_{st(i,2),r}, T^{*}cooling, T_{JI,JOH} \Big) \\ &g_{OXIPK_{i}} = \begin{cases} &0.035 \cdot \theta_{PK_{i}} \\ &1 - \theta_{PK_{i}} \end{cases} & \text{if} & \frac{0.035 \cdot \theta_{PK_{i}}}{1 - \theta_{PK_{i}}} \geq 0 \\ &0 & \text{otherwise} \end{cases} \\ &G_{st(i,3)} = G_{st(i,2)} \cdot \Big( 1 + g_{OXIPK_{i}} \Big) \\ &\alpha_{OX_{st(i,3)}} = \alpha_{OX_{st(i,2)}} + g_{OXJPK_{i}} \\ &k_{st(i,3),r} = k_{st(i,2),r} \\ &k_{st(i,3),r} = k_{st(i,2),r} \cdot k_{st(i,3),r} \Big) \\ &T_{a\eta_{st(i,3),r}} = T_{st(i,2),r} \cdot \frac{T_{a\eta_{st(i,3),r}}}{k_{PK_{i}} - 1} \cdot R_{ras,cp} \Big( \alpha_{OX_{st(i,2)}}, \alpha_{OX_{st(i,3)}}, rucl \Big) \\ &T_{st(i,3),r} = P_{st(i,2),r} \cdot \frac{T_{a\eta_{st(i,3),r}}}{T_{st(i,2),r}} \Big) \\ &T_{st(i,3),r} = T_{st(i,2),r} - \frac{\left(w_{st(i,3),r}\right)^{2} - \left(w_{st(i,2),r}\right)^{2} - \left(u_{st(i,3),r}\right)^{2} + \left(u_{st(i,2),r}\right)^{2}}{2 \cdot \frac{k_{PK_{i}}}{k_{PK_{i}} - 1}} \\ &Cp_{3} = Cp_{ras} \Big( P_{st(i,3),r}, T_{st(i,3),r}, \alpha_{OX_{st(i,3)}}, Fuel \Big) \\ &k' = k_{aq} \Big( Cp_{3}, R_{ras} \Big( \alpha_{OX_{st(i,3)}}, Fuel \Big) \Big) \\ &k' = k_{aq} \Big( Cp_{3}, R_{ras} \Big( \alpha_{OX_{st(i,3)}}, Fuel \Big) \Big) \\ &k' = k_{st(i,3),r} = k' \\ &break \\ &k_{st(i,3),r} = \frac{R_{ras} \Big( \alpha_{OX_{st(i,3)}}, Fuel \Big) \cdot T_{st(i,3),r}}{P_{st(i,3),r}} \\ &\rho_{st(i,3),r} = asin \Big( \frac{G_{st(i,3),r}, Fus(i,3),r}{W_{st(i,3),r}, Fis(i,3),r} \Big) \\ &c_{u} \\$$

$$\begin{cases} c_{3d(1,3),\tau} = \sqrt{\left(c_{3d(1,3),\tau}\right)^2 + \left(c_{3d(1,3),\tau}\right)^2 + \left(c_{3d(1,3),\tau}\right)^2} \\ c_{3d(1,3),\tau} = \sqrt{\left(w_{3d(1,3),\tau}\right)^2 + \left(c_{3d(1,3),\tau}\right)^2 - \left(c_{3d(1,3),\tau}\right)^2} \\ c_{3d(1,3),\tau} = \sqrt{\left(w_{3d(1,3),\tau}\right)^2 + \left(c_{3d(1,3),\tau}\right)^2 - \left(c_{3d(1,3),\tau}\right)^2} \\ c_{3d(1,3),\tau} = T_{3d(1,3),\tau} + \frac{1}{2 \cdot C_{Pras}\left(P_{3d(1,3),\tau}, T_{3d(1,3),\tau}\right)^2} \\ c_{3d(1,3),\tau} = \frac{1}{2 \cdot C_{3d(1,3),\tau}} \\ c_{3d(1,3),\tau} = \frac{1$$

$$\begin{vmatrix} \sum_{s(i_1,i_2),\tau} \\ \sum_{s'(i_1,i_2),\tau} \\ \\ M_{sd(i_1,i_2),\tau} \\ M_{sd(i_2,i_2),\tau} \end{vmatrix} = \frac{1}{a_{mail_1,3},\tau} \begin{pmatrix} c_{u(i_1,3),\tau} \\ c_{u(i_1,3),\tau} \\ \\ c_{uval_1,3} \\ \\ c_{uval_2,3} \\ \\ c_{uval_2,3}$$

$$\left[ H^*_{cT_{\hat{i}}} = Cp_{\Gamma a3.cp} \left( P_{st(i,1),r}, P_{st(i,3),r}, T_{st(i,1),r}, T_{st(i,3),r}, \alpha_{oX_{st(i,1)}}, \alpha_{oX_{st(i,3)}}, Fuel \right) \cdot T^*_{st(i,1),r} \cdot \left[ 1 - \left( \pi^*_{cT_{\hat{i}}} \right)^{\frac{1}{k_{cp}}} \right] \right]$$

$$\left[ \eta^*_{cT_{\hat{i}}} = \frac{L_{cT_{\hat{i}}}}{H^*_{cT_{\hat{i}}}} \right]$$

for  $i \in 1...Z$ 

for  $j \in 1...3$ 

$$\begin{vmatrix} \rho *_{st(i,j),r} = \frac{P *_{st(i,j),r}}{R_{\Gamma a3} \left(\alpha_{ox_{st(i,j)}}, Fuel\right) \cdot T *_{st(i,j),r}} \\ \rho_{st(i,j),r} = \left(v_{st(i,j),r}\right)^{-1} \\ \left(\varepsilon_{stator_{i,av}\left(N_{r}\right)}\right) = \begin{pmatrix} \alpha_{st(i,2),av\left(N_{r}\right)} - \alpha_{st(i,1),av\left(N_{r}\right)} \\ \beta_{st(i,3),av\left(N_{r}\right)} - \beta_{st(i,2),av\left(N_{r}\right)} \end{pmatrix}$$

(iteration<sub>CA</sub> iteration<sub>PK</sub>)

$$H_{\text{cT}}$$
  $H_{\text{cT}}$ 

11stator 11rotor

P\* P

T\* T

G v

0\*

 $\alpha_{ox}$   $\alpha_{ox}$ 

α β

 $\varepsilon_{
m stator}$   $\varepsilon_{
m rotor}$ 

 $\theta_{\text{CA}}$   $\theta_{\text{PK}}$ 

 $g_{\text{охл}CA}$   $g_{\text{охл}PK}$ 

 $a^*_{\phantom{a}c}$   $a^*_{\phantom{a}w}$ 

 $T_{a \perp}$   $T_{a \perp}$ 

 $P_{W}^{*}$   $T_{W}^{*}$ 

 $a_{3B}$   $a_{3B}$ 

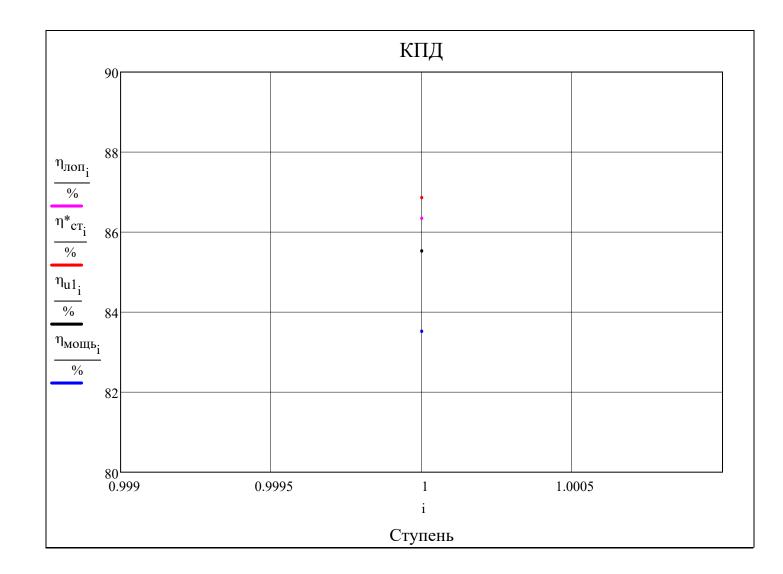
u

c c

c<sub>a</sub> c

ı	I	
	W	W
	w <sub>a</sub>	$w_{u}$
	$\lambda_{\rm c}$	$M_{c}$
	$\lambda_{ m W}$	$M_{W}$
	v <sub>stator</sub>	$v_{ m rotor}$
	chord <sub>stator</sub>	chordrotor
	-t <sub>oπτCA</sub>	т oптРК
	t <sub>stator</sub>	trotor
	Z <sub>stator</sub>	Z <sub>rotor</sub>
	$\frac{Z_{\mathrm{stator}}}{v_{\mathrm{stator}}}$	$\overline{v}_{ m rotor}$
	ξ <sub>TpCA</sub>	$\xi_{\mathrm{TpPK}}$
	ξ <sub>Tp</sub> CA ξ <sub>κp</sub> CA	$\xi_{\text{kpPK}}$
	ξ <sub>ReCA</sub>	ξ <sub>RePK</sub>
	ξ <sub>ReCA</sub> ξ <sub>λCA</sub> ξ <sub>πpCA</sub> ξ <sub>BTCA</sub> ξ <sub>TдCA</sub>	$\xi_{\lambda PK}$
	ξπρСΑ	$\xi_{\Pi p P K}$
	$\xi_{\mathrm{BTCA}}$	$\xi_{BTPK}$
	ξ <sub>тд</sub> СА	$\xi_{TДPK}$
	ξ <sub>cm</sub> CA	$\xi_{\text{CMPK}}$
	$\xi_{\Delta r}$	$\xi_{ m BMX}$
	ξ <sub>Τр.Β</sub>	$\xi_{\mathrm{Tp.B}}$
	L <sub>ct</sub>	$Lu_{CT}$
	η <sub>мощь</sub>	$\eta_{ extit{JOH}}$
	$\eta^*_{ m cT}$	$\eta^*_{ m cT}$
	$\eta_{u1}$	$\eta_{u2}$
	ξ <sub>CA</sub>	$\xi_{ m PK}$
	_ Lи <sub>нагрузка</sub>	

▼ Параметры турбинь



$$\eta_{\Pi O \Pi}^{\quad T} = \boxed{ \begin{array}{c|c} 1 \\ 1 \\ 86.35 \end{array}} \cdot \%$$

$$\operatorname{stack}\!\left(\eta_{u1}^{\phantom{u1}T},\eta_{u2}^{\phantom{u2}T}\right) = \begin{array}{|c|c|c|c|}\hline & 1 \\ \hline 1 & 85.54 \\ \hline 2 & 86.83 \\ \hline \end{array} \cdot \%$$

$$\eta_{\text{MOIЦb}}^{T} = \boxed{\begin{array}{c|c} 1 \\ 1 \\ 83.53 \end{array}} \cdot \%$$

$$\eta_{\text{MOIII}_{i}} \leq \eta_{u1_{i}} \leq \eta^{*}_{cT_{i}} \leq \eta_{\text{JO}\Pi_{i}} = 0$$

Теплоперепад по параметрам торможения (Дж/кг): 
$$\begin{pmatrix} H^*_T \\ H_T \end{pmatrix} = \begin{pmatrix} \sum_{i=1}^Z & H^*_{cT_i} \\ \vdots & \vdots & \vdots \\ \sum_{i=1}^Z & H_{cT_i} \end{pmatrix} = \begin{bmatrix} 1 & 1 & 1 \\ 1 & 516.1 \\ 2 & 536.7 \end{bmatrix} \cdot 10^3$$

Мощность T (Вт): 
$$\sum_{i=1}^{Z} N_{cT_{i}} = 14.89 \cdot 10^{6}$$
 eps ("rel",  $N_{T}$ ,  $\sum_{i=1}^{Z} N_{cT_{i}}$ ) = 0.000 · %

Удельная поступенчатая работ а Т [Дж/кг]: 
$$L_T = \sum_{i=1}^{Z} \frac{N_{cT_i}}{\text{mean}\big(G_{st(i,2)}, G_{st(i,3)}\big)} = 448.6 \cdot 10^3$$

$$k_{T.cp} = k_{ad} \left( Cp_{\Gamma a3.cp} \left( P_{st(1,1),av(N_r)}, P_{st(Z,3),av(N_r)}, T_{st(1,1),av(N_r)}, T_{st(Z,3),av(N_r)}, \alpha_{ox_{st(1,1)}}, \alpha_{ox_{st(Z,3)}}, Fuel \right), P_{ras.cp} \left( \alpha_{ox_{st(1,1)}}, \alpha_{ox_{st(Z,3)}}, Fuel \right) \right) = 1.289$$

Адиабатный КПДТ: 
$$\eta^*_T = \frac{L_T}{H^*_T} = 86.92 \cdot \%$$

Политропический КПД Т: 
$$\eta^*_{T,\Pi} = \eta^*_{n} ($$
"расширение", $\eta^*_{T}, \pi^*_{T}, k_{T,cp} ) = 85.37 \cdot \%$ 

$$\text{Мощностной КПДТ:} \qquad \eta_{\text{Тмощь}} = \frac{\displaystyle\sum_{i=1}^{Z} \frac{N_{\text{CT}_{i}}}{\text{mean}\big(G_{\text{st}(i,2)},G_{\text{st}(i,3)}\big)}}{H_{\text{T}}} = 83.58 \cdot \%$$

$$L_{\text{CT}}^{T} = \boxed{ \begin{array}{c|c} 1 \\ \hline 1 & 448.3 \end{array}} \cdot 10^3$$

$H_{om}^{T} =$		1	$1.10^3$
CT	1	536.7	

$$\operatorname{stack}\left(\mathbf{H}_{\operatorname{stator}}^{\operatorname{T}}, \mathbf{H}_{\operatorname{rotor}}^{\operatorname{T}}\right) = \begin{bmatrix} & 1 \\ 1 & 450.9 \\ 2 & 87.0 \end{bmatrix} \cdot 10^{3}$$

$$submatrix \left( R_L^T, av(N_r), av(N_r), 1, Z \right) = \boxed{\begin{array}{c} 1 \\ 1 \\ \hline \end{array}}$$

$$\operatorname{stack}\!\left(\boldsymbol{\theta}_{CA}^{\phantom{CA}T},\boldsymbol{\theta}_{PK}^{\phantom{PK}T}\right) = \begin{array}{|c|c|c|}\hline & 1 \\ \hline 1 & 0.432 \\ \hline 2 & 0.206 \\ \hline \end{array}$$

stack 
$$\left(g_{\text{OX}}, g_{\text{OX}}, g_{\text{OX}}, g_{\text{OX}}\right) = \begin{bmatrix} 1 & 1 \\ 1 & 26.61 \\ 2 & 9.09 \end{bmatrix} \cdot 10^{-3}$$

$$G_{OXJICA_{i}} = g_{OXJICA_{i}} \cdot G_{st(i,1)}$$

$$G_{\text{OX},PK_i} = g_{\text{OX},PK_i} \cdot G_{\text{st}(i,2)}$$

$$\operatorname{stack}\left(G_{\text{OXJICA}}^{T},G_{\text{OXJIPK}}^{T}\right) = \begin{bmatrix} & 1\\ & 1 & 0.9\\ & 2 & 0.3 \end{bmatrix}$$

$$G_{cooling} = 3.2$$

$$\sum_{i=1}^{Z} G_{\text{oxnCA}_i} + \sum_{i=1}^{Z} G_{\text{oxnCA}_i} \leq G_{\text{cooling}} = 1$$

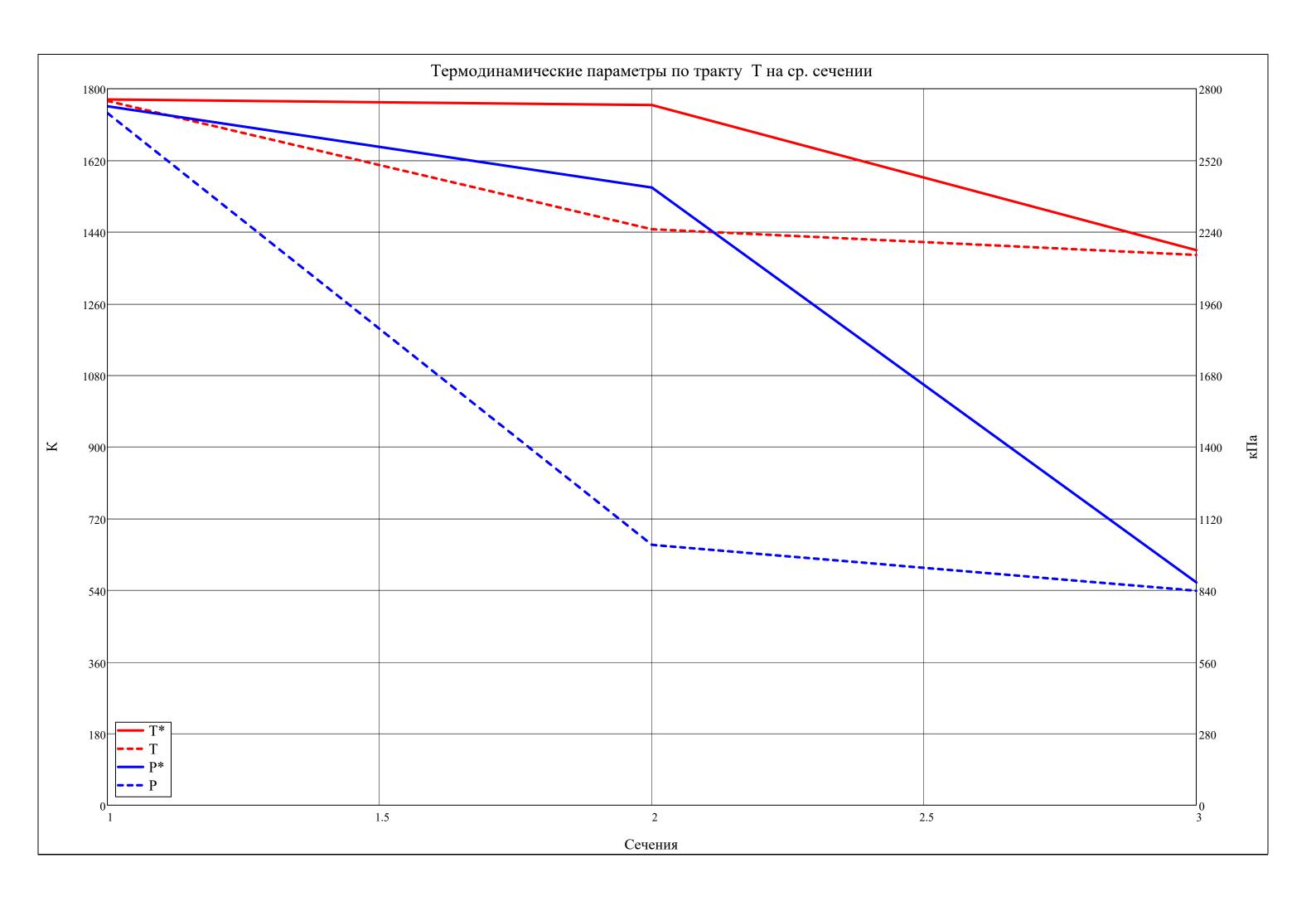
$$stack \left( iteration_{CA}^{T}, iteration_{PK}^{T} \right) = \begin{bmatrix} 1 \\ 1 \\ 2 \\ 2 \end{bmatrix}$$

submatrix 
$$(k^{T}, av(N_{r}), av(N_{r}), 1, 2Z + 1) = \begin{bmatrix} 1 & 2 & 3 \\ 1 & 1.283 & 1.293 & 1.295 \end{bmatrix}$$

submatrix 
$$\left(P^{*T}, av(N_r), av(N_r), 1, 2Z + 1\right) = \begin{bmatrix} 1 & 2 & 3 \\ 1 & 2731.8 & 2413.7 & 870.5 \end{bmatrix} \cdot 10^3$$

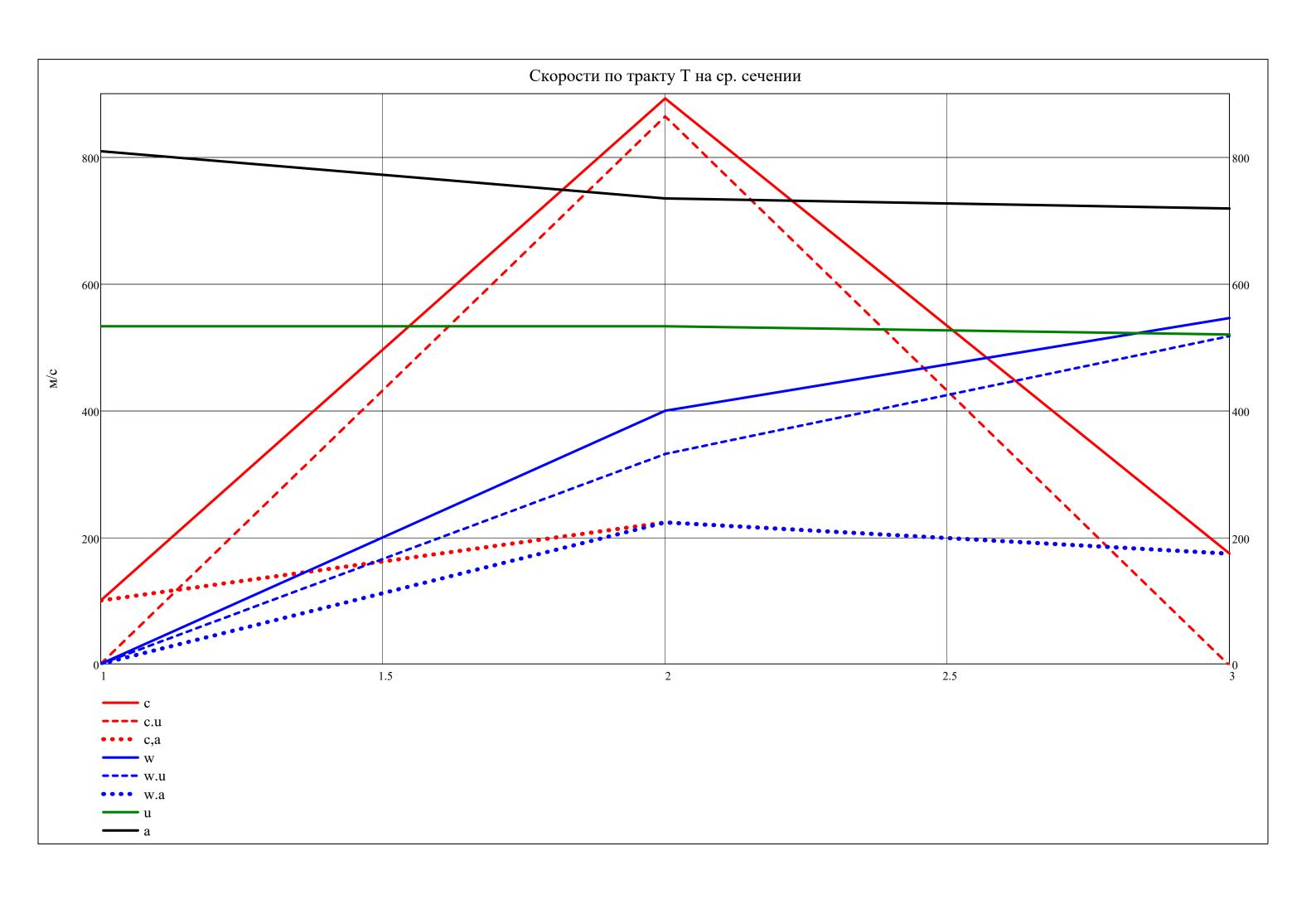
$$\operatorname{submatrix}\left(v^{T},\operatorname{av}\left(N_{r}\right),\operatorname{av}\left(N_{r}\right),1,2Z+1\right) = \begin{array}{|c|c|c|c|c|}\hline 1 & 2 & 3 \\\hline 1 & 0.189 & 0.410 & 0.487 \\\hline \end{array}$$

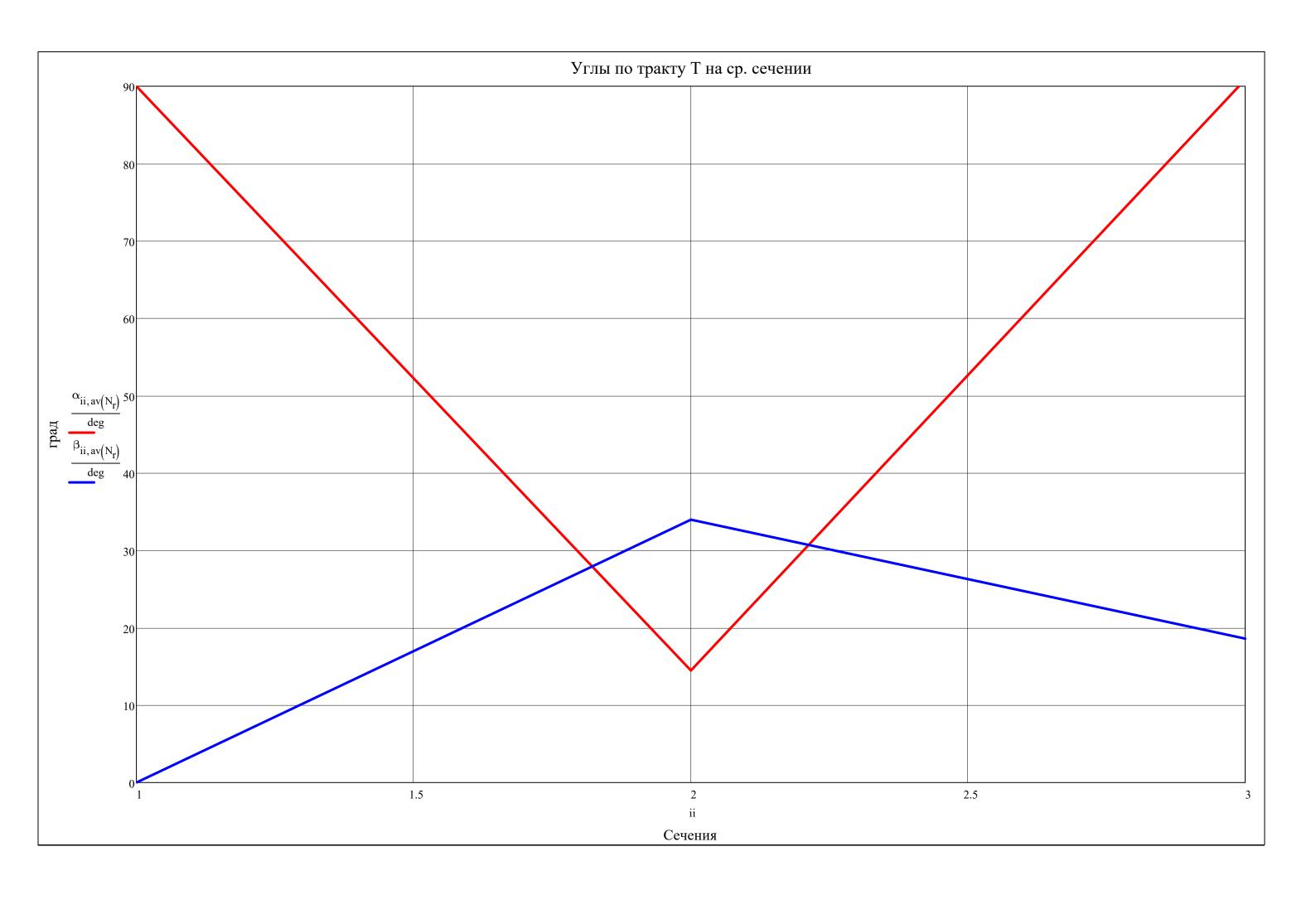
submatrix 
$$\left(\rho^{*^{T}}, av(N_{r}), av(N_{r}), 1, 2Z + 1\right) = \boxed{\begin{array}{c|cccc} 1 & 2 & 3 \\ \hline 1 & 5.341 & 4.756 & 2.164 \end{array}}$$

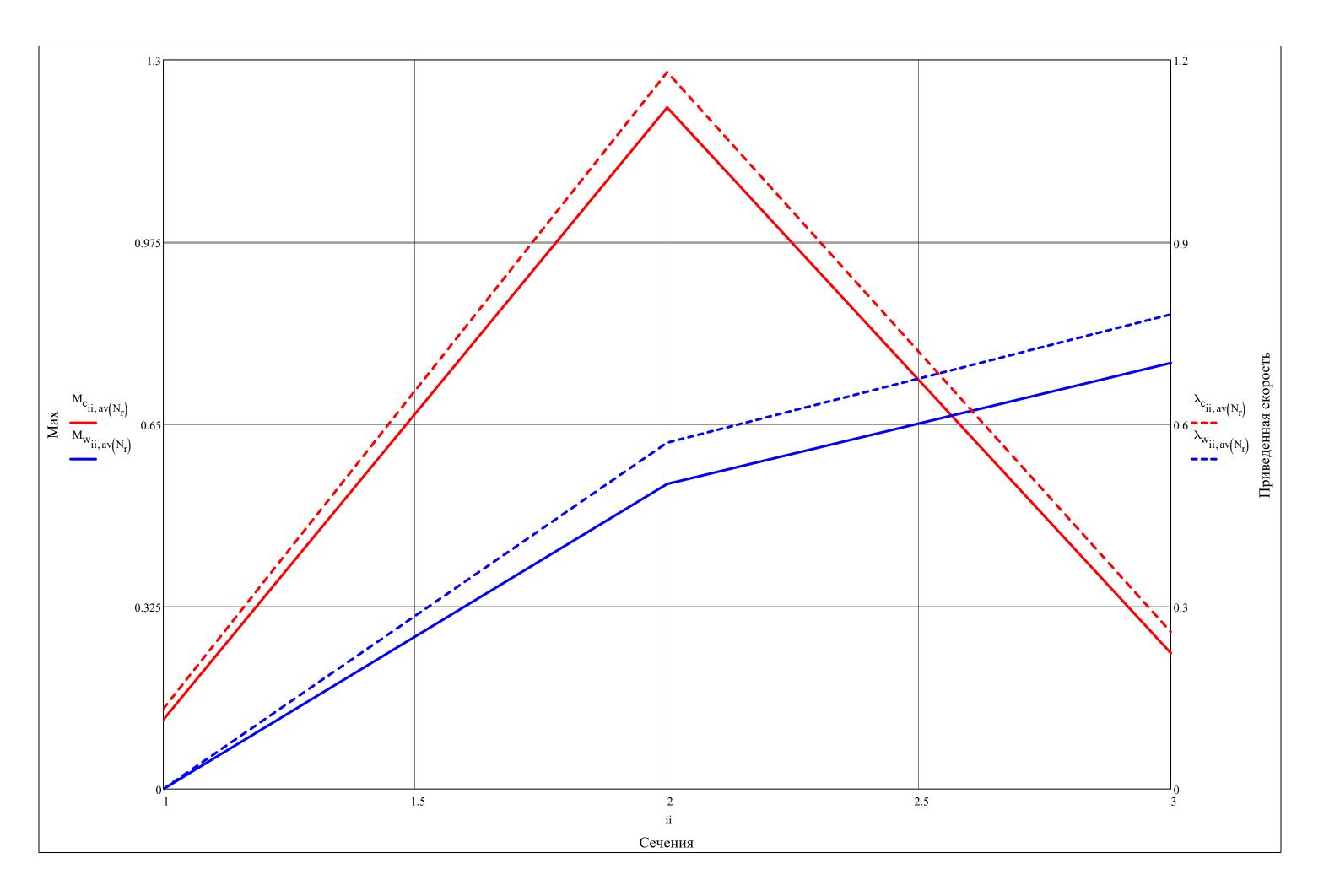


$$\begin{aligned} & \text{submatrix} \bigg( a_{3B}^{-T}, av \Big( N_r \big), av \Big( N_r \big), 1, 2Z+1 \bigg) = \underbrace{ \begin{array}{c} 1 \\ 1 \\ 809.2 \end{array} }_{734.6} \underbrace{ \begin{array}{c} 3 \\ 718.7 \end{array}}_{734.6} \end{aligned} \\ & \text{submatrix} \bigg( a^*_{c}^{-T}, av \Big( N_r \big), av \Big( N_r \big), 1, 2Z+1 \bigg) = \underbrace{ \begin{array}{c} 1 \\ 1 \\ 758.2 \end{array} }_{756.5} \underbrace{ \begin{array}{c} 3 \\ 756.5 \end{array} }_{756.5} \underbrace{ \begin{array}{c} 3 \\ 673.8 \end{array}}_{756.5} \end{aligned} \\ & \text{submatrix} \bigg( a^*_{w}^{-T}, av \Big( N_r \big), av \Big( N_r \big), 1, 2Z+1 \bigg) = \underbrace{ \begin{array}{c} 1 \\ 1 \\ 1 \\ 0.0 \end{array} }_{1} \underbrace{ \begin{array}{c} 2 \\ 3 \\ 1 \\ 10.00 \end{array} }_{802.5} \underbrace{ \begin{array}{c} 3 \\ 174.1 \end{array} }_{174.1} \end{aligned} \\ & \text{submatrix} \bigg( c_{u}^{-T}, av \Big( N_r \big), av \Big( N_r \big), 1, 2Z+1 \bigg) = \underbrace{ \begin{array}{c} 1 \\ 1 \\ 1 \\ 0.0 \end{array} }_{1} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 100.0 \end{array} }_{864.2} \underbrace{ \begin{array}{c} 3 \\ -2.7 \end{array} }_{174.1} \end{aligned} \\ & \text{submatrix} \bigg( c_{u}^{-T}, av \Big( N_r \big), av \Big( N_r \big), 1, 2Z+1 \bigg) = \underbrace{ \begin{array}{c} 1 \\ 1 \\ 1 \\ 0.0 \end{array} }_{1} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 100.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 174.0 \end{array} }_{174.0} \end{aligned} \\ & \text{submatrix} \bigg( w_{u}^{-T}, av \Big( N_r \big), av \Big( N_r \big), 1, 2Z+1 \bigg) = \underbrace{ \begin{array}{c} 1 \\ 1 \\ 1 \\ 0.0 \end{array} }_{239.5} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{331.3} \underbrace{ \begin{array}{c} 3 \\ 517.3 \end{array} }_{517.3} \end{aligned} \\ & \text{submatrix} \bigg( w_{u}^{-T}, av \Big( N_r \big), av \Big( N_r \big), 1, 2Z+1 \bigg) = \underbrace{ \begin{array}{c} 1 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \end{array} }_{223.3} \underbrace{ \begin{array}{c} 3 \\ 1 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c} 3 \\ 2 \\ 3 \\ 1 \\ 0.0 \underbrace{ \begin{array}{c}$$

		1	2	3
$u^T =$	1	510.5	510.5	484.8
•	2	532.9	532.9	520.0
	3	555.2	555.2	555.2







$$t_{\text{stator}}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 56.7 \\ 2 & 59.2 \\ 3 & 61.7 \end{bmatrix} \cdot 10^{-3} \qquad t_{\text{rotor}}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 22.4 \\ 2 & 23.7 \\ 3 & 25.0 \end{bmatrix} \cdot 10^{-3}$$

submatrix 
$$\left( \text{chord}_{\text{stator}}^T, \text{av}(N_r), \text{av}(N_r), 1, Z \right) = \begin{bmatrix} 1 \\ 1 \\ 68.0 \end{bmatrix} \cdot 10^{-3}$$

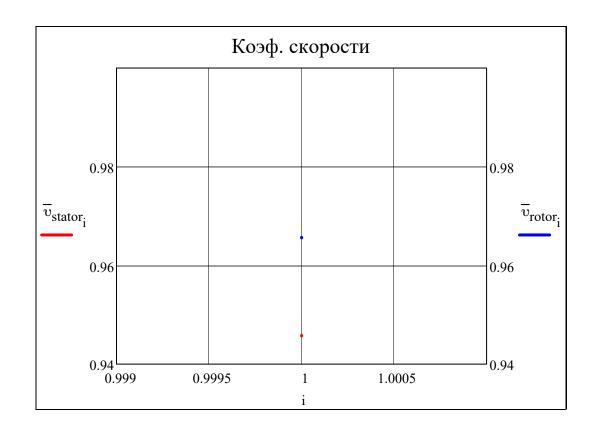
$$submatrix \left( chord_{rotor}^{T}, av(N_r), av(N_r), 1, Z \right) = \boxed{\frac{1}{1 \quad 32.7}} \cdot 10^{-3}$$

$$\operatorname{stack}\left(Z_{\operatorname{stator}}^{\operatorname{T}}, Z_{\operatorname{rotor}}^{\operatorname{T}}\right) = \begin{bmatrix} 1 \\ 1 & 36 \\ 2 & 89 \end{bmatrix}$$

$$stack\left(\overline{t}_{O\Pi TCA}^{T}, \overline{t}_{O\Pi TPK}^{T}\right) = \begin{bmatrix} & 1\\ & 1\\ & 1 & 0.872\\ & 2 & 0.724 \end{bmatrix}$$

$$\frac{t_{stator_{i, av\left(N_{r}\right)}}}{chord_{stator_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{stator_{i, av\left(N_{r}\right)}}}{chord_{stator_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} \le 1 = \frac{t_{rotor_{i, av\left(N_{r}\right)}}}{chord_{rotor_{i, av\left(N_{r}\right)}}} = 0.7 \le \frac{t_{rotor_{i, av\left(N_{r}\right)$$

. (- T - T)		1	
$\operatorname{stack}(v_{\operatorname{stator}}, v_{\operatorname{rotor}}) =$	1	0.9458	
	2	0.9657	



$$stack\left(\xi_{TpCA}^{T},\xi_{TpPK}^{T}\right) = \begin{bmatrix} & 1\\ & 1\\ & 1.398\\ 2 & 2.620 \end{bmatrix}.\%$$

$$stack \left( \xi_{KpCA}^{T}, \xi_{KpPK}^{T} \right) = \begin{bmatrix} 1 \\ 1 \\ 2.753 \\ 2 \\ 1.689 \end{bmatrix} \cdot \%$$

$$stack \left( \xi_{ReCA}^{T}, \xi_{RePK}^{T} \right) = \begin{bmatrix} 1 \\ 1 \\ -0.135 \\ 2 \\ 0.085 \end{bmatrix}$$
 .%

$$stack\left(\xi_{\lambda CA}^{T}, \xi_{\lambda PK}^{T}\right) = \begin{bmatrix} 1\\ 1\\ 2.434\\ 2 & 0.024 \end{bmatrix} \cdot \%$$

$$\operatorname{stack}\left(\xi_{\operatorname{BTCA}}^{\operatorname{T}}, \xi_{\operatorname{BTPK}}^{\operatorname{T}}\right) = \begin{array}{|c|c|c|c|}\hline 1 & 1.453 \\ \hline 1 & 1.453 \\ \hline 2 & 0.881 \\ \hline \end{array}$$

$$stack\left(\xi_{TДCA}^{T},\xi_{TДPK}^{T}\right) = \begin{array}{|c|c|c|c|c|}\hline & 1 & \\\hline 1 & 1.856 \\\hline 2 & 1.200 \\\hline \end{array}$$

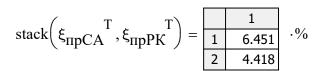
$$stack\left(\xi_{cMCA}^{T}, \xi_{cMPK}^{T}\right) = \begin{bmatrix} 1\\ 1\\ 0.784\\ 2\\ 0.248 \end{bmatrix} \cdot \%$$

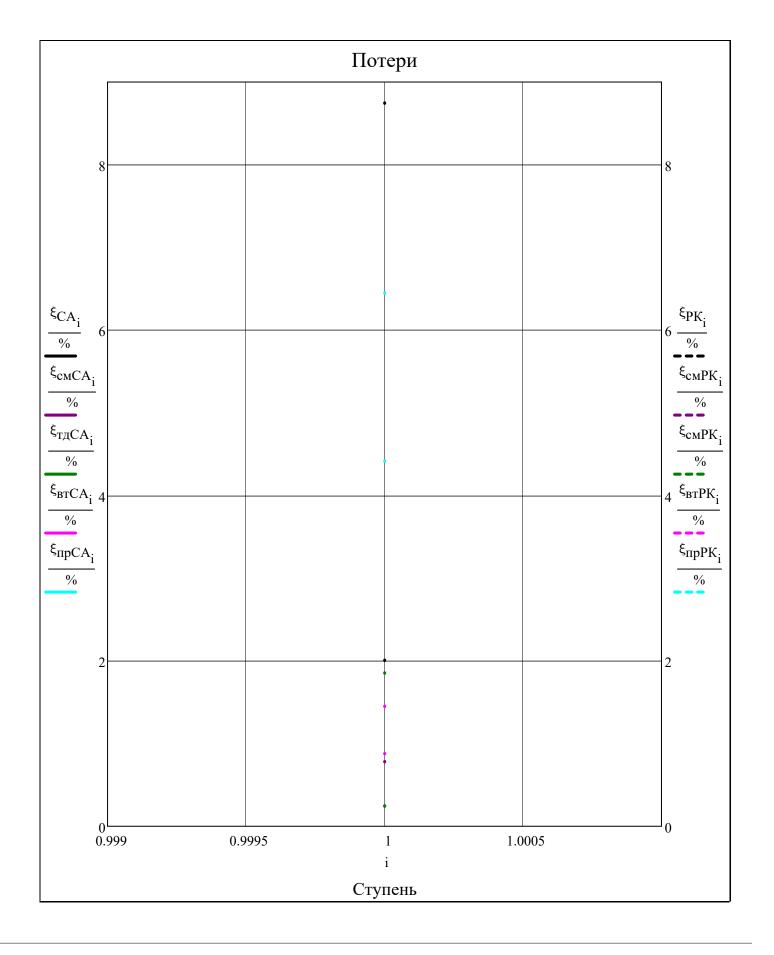
$$\operatorname{stack}\left(\xi_{CA}^{T},\xi_{PK}^{T}\right) = \begin{array}{|c|c|c|c|}\hline & 1 & \\ \hline 1 & 8.748 \\ \hline 2 & 2.008 \\ \hline \end{array}.\%$$

$$\xi_{B b I X}^{\quad T} = \boxed{ \quad \quad 1 \quad \quad } \cdot \%$$

$$\xi_{\Delta r}^{T} = \boxed{\begin{array}{c|c} 1 \\ 1 & 2.476 \end{array}} \cdot \%$$

$$\xi_{Tp.B}^{\quad T} = \boxed{ \begin{array}{c|c} 1 \\ \hline 1 & 0.831 \end{array}} \cdot \%$$





Вывод-результатов поступенчатого расчета-по-ср. сечению-ОТ в EXCEL:

▼ Выбор закона профилирования Л по высоте

$$\mathbf{m} = \begin{pmatrix} \boxed{\overline{v}_{stator_1} \cdot \cos(\alpha_{st(1,2),av(N_r)})^2} & \text{if } Z = 1 \\ -0.5 & \text{otherwise} \end{pmatrix}$$

$$0$$

$$0.25$$

$$1$$

$$1$$

$$\begin{pmatrix} \text{"}\alpha.2 = \text{const"} \\ \text{"}\Gamma = \text{const"} \\ \text{"}m = \text{const"} \\ \text{"}R = \text{const"} \end{pmatrix} = \begin{pmatrix} \cos\left(\alpha_{\text{st(i,2),av(N_r)}}\right)^2 \cdot \overline{\upsilon}_{\text{stator_i}} \\ 1 \cdot \overline{\upsilon}_{\text{stator_i}} \\ 0.2 \\ -1 \cdot \overline{\upsilon}_{\text{stator_i}} \end{pmatrix}$$

$\mathbf{m}^{\mathrm{T}} =$		1	2	3	4	5	6
	1	0.8866	-0.2500	0.0000	0.2500	1.0000	1.0000

▲ Выбор закона профилирования Л по высоте

$$\begin{vmatrix} \mathbf{r}_{i} & \mathbf{m}_{i} & - \mathbf{v}_{stator_{i}}^{-v_{stator_{i}}} \\ \mathbf{c}_{a_{st(i,\,a),\,av}(N_{r})} \\ \mathbf{c}_{a_{st(i,\,a),\,av}(N_{r})} \end{vmatrix} + \frac{\left(1 - \frac{\overline{v}_{stator_{i}}}{R_{st(i,\,a),\,av}(N_{r})}\right)^{2}}{tan\left(\alpha_{st(i,\,2),\,av(N_{r})}\right)^{2}} \quad \text{if } a = 2$$
 
$$\begin{vmatrix} \mathbf{c}_{a_{st(i,\,a),\,av(N_{r})}} \\ \mathbf{c}_{a_{st(i,\,a),\,av(N_{r})}} \end{vmatrix}^{2} \\ + \left[1 - (\overline{v}_{rotor_{i}})^{2}\right] \cdot \left(\mathbf{u}_{st(i,\,a),\,av(N_{r})}\right)^{2} \left[1 - \left(\frac{R_{st(i,\,a),\,r}}{R_{st(i,\,a),\,av(N_{r})}}\right)^{2}\right] - 2 \cdot \mathbf{c}_{\mathbf{u}_{st(i,\,a),\,av(N_{r})}} \\ + \left[1 - (\overline{v}_{rotor_{i}})^{2}\right] \cdot \left(\mathbf{u}_{st(i,\,a),\,av(N_{r})}\right)^{2} \left[1 - \left(\frac{R_{st(i,\,a),\,r}}{R_{st(i,\,a),\,av(N_{r})}}\right)^{2}\right] - 2 \cdot \mathbf{c}_{\mathbf{u}_{st(i,\,a),\,av(N_{r})}} \\ + \left[1 - (\overline{v}_{rotor_{i}})^{2}\right] \cdot \left(\mathbf{u}_{st(i,\,a),\,av(N_{r})}\right) \cdot \left(\mathbf{u}_{st(i,\,a,\,av(N_{r}))}\right)^{2} \cdot \left(\mathbf{u}_{st(i,\,a-1),\,av(N_{r})}\right) + \mathbf{c}_{\mathbf{u}_{st(i,\,a),\,av(N_{r})}} \\ + \left[1 - (\overline{v}_{rotor_{i}})^{2}\right] \cdot \left(\mathbf{u}_{st(i,\,a-1),\,av(N_{r})}\right) \cdot \left(\mathbf{u}_{st(i,\,a-1),\,av(N_{r})}\right)^{2} \cdot \left(\mathbf{u}_{st(i,\,a),\,av(N_{r})}\right)^{2} \cdot \left(\mathbf{u}_{st(i,\,a),\,a$$

for  $i \in 1...2 \cdot Z + 1$ 

for  $r \in 1..N_r$ 

$$\begin{vmatrix} c_{u_{i,r}} \\ c_{a_{i,r}} \end{vmatrix} = c_{i,av(N_r)} \cdot \begin{pmatrix} \cos(\alpha_{i,av(N_r)}) \\ \sin(\alpha_{i,av(N_r)}) \end{pmatrix} \text{ if } (i = 1)$$

$$P^*_{i,r} = P^*_{i,av(N_r)}$$

$$T^*_{i,r} = T^*_{i,av(N_r)}$$

$$\rho^*_{i,r} = \frac{P^*_{i,r}}{R_{ras}(\alpha_{ox_i}, Fuel) \cdot T^*_{i,r}}$$

$$\begin{vmatrix} k_{i,\,r} = k_{aa} \left( Cp_{Bo3,Jyx} \left( P^*_{i,\,r}, T^*_{i,\,r} \right), R_{ra3} \left( \alpha_{oX_i}, Fuel \right) \right) \\ a^*_{c_{i,\,r}} = \sqrt{\frac{2 \cdot k_{i,\,r}}{k_{i,\,r} + 1}} \cdot R_{ra3} \left( \alpha_{oX_i}, Fuel \right) \cdot T^*_{i,\,r} \\ \alpha_{i,\,r} = \operatorname{triangle} \left( c_{a_{i,\,r}}, c_{u_{i,\,r}} \right) \\ c_{i,\,r} = \frac{c_{a_{i,\,r}}}{\sin(\alpha_{i,\,r})} \\ \lambda_{c_{i,\,r}} = \frac{c_{i,\,r}}{a^*_{c_{i,\,r}}} \\ \begin{pmatrix} T_{i,\,r} \\ P_{i,\,r} \end{pmatrix} = \begin{pmatrix} T^*_{i,\,r} \cdot \Gamma \mathcal{H} \Phi \left( ^{"}T^{"}, \lambda_{c_{i,\,r}}, k_{i,\,r} \right) \\ P^*_{i,\,r} \cdot \Gamma \mathcal{H} \Phi \left( ^{"}P^{"}, \lambda_{c_{i,\,r}}, k_{i,\,r} \right) \end{pmatrix} \\ a_{3B_{i,\,r}} = \sqrt{k_{i,\,r}} \cdot R_{ra3} \left( \alpha_{oX_i}, Fuel \right) \cdot T_{i,\,r} \\ M_{c_{i,\,r}} = \frac{c_{i,\,r}}{a_{3B_{i,\,r}}} \\ \beta_{i,\,r} = \operatorname{triangle} \left( c_{a_{i,\,r}}, u_{i,\,r} - c_{u_{i,\,r}} \right) \\ w_{i,\,r} = \frac{c_{a_{i,\,r}}}{\sin(\beta_{i,\,r})} \\ \begin{pmatrix} w_{u_{i,\,r}} \\ w_{a_{i,\,r}} \end{pmatrix} = w_{i,\,r} \cdot \begin{pmatrix} \cos(\beta_{i,\,r}) \\ \sin(\beta_{i,\,r}) \end{pmatrix} \\ T^*_{w_{i,\,r}} = T^*_{i,\,r} - \frac{\left( c_{i,\,r} \right)^2 - \left( w_{i,\,r} \right)^2}{2 \cdot \frac{k_{i,\,r}}{k_{i,\,r} - 1}} \cdot R_{ra3} \left( \alpha_{oX_i}, Fuel \right) \cdot T^*_{w_{i,\,r}} \\ \lambda_{w_{i,\,r}} = \frac{w_{i,\,r}}{a^*_{w_{i,\,r}}} \\ \lambda_{w_{i,\,r}} = \frac{w_{i,\,r}}{a^*_{w_{i,\,r}}} \\ M_{w_{i,\,r}} = \frac{w_{i,\,r}}{a^*_{w_{i,\,r}}} \\ for \ i \in 1 ... Z \\ for \ r \in 1 ... N_r \\ \end{cases}$$

 $\left| \left( \Delta c_a - c_a \right) \right|$ 

$$\begin{bmatrix} \begin{bmatrix} \mathbf{c}^{*}\mathbf{st}(i,1),\mathbf{r} \\ \Delta \mathbf{c}_{\mathbf{a}_{\mathbf{st}}(i,2),\mathbf{r}} \end{bmatrix} = \begin{bmatrix} \mathbf{c}^{*}\mathbf{st}(i,2),\mathbf{r} & \mathbf{c}\mathbf{st}(i,1),\mathbf{r} \\ \mathbf{c}_{\mathbf{a}_{\mathbf{st}}(i,2),\mathbf{r}} - \mathbf{c}_{\mathbf{a}_{\mathbf{st}}(i,2),\mathbf{r}} \end{bmatrix} \\ \mathbf{R}_{\mathbf{L}_{\mathbf{i},\mathbf{r}}} = 1 - \frac{\mathbf{c}_{\mathbf{u}_{\mathbf{st}}(i,2),\mathbf{r}} - \mathbf{c}_{\mathbf{u}_{\mathbf{st}}(i,3),\mathbf{r}}}{\mathbf{u}_{\mathbf{st}(i,2),\mathbf{r}} + \mathbf{u}_{\mathbf{st}(i,3),\mathbf{r}}} \\ \boldsymbol{\varepsilon}_{\mathbf{stator}_{\mathbf{i},\mathbf{r}}} = \begin{bmatrix} \mathbf{c}_{\mathbf{st}(i,2),\mathbf{r}} - \mathbf{c}_{\mathbf{st}}(i,1),\mathbf{r} & \text{if } \mathbf{c}_{\mathbf{st}(i,2),\mathbf{r}} \geq \frac{\pi}{2} \\ \mathbf{c}_{\mathbf{st}(i,1),\mathbf{r}} - \mathbf{c}_{\mathbf{st}(i,2),\mathbf{r}} & \text{otherwise} \end{bmatrix} \\ \boldsymbol{\varepsilon}_{\mathbf{rotor}_{\mathbf{i},\mathbf{r}}} = \begin{bmatrix} \mathbf{c}_{\mathbf{st}(i,3),\mathbf{r}} - \mathbf{c}_{\mathbf{st}(i,2),\mathbf{r}} & \text{if } \mathbf{c}_{\mathbf{st}(i,3),\mathbf{r}} \geq \frac{\pi}{2} \\ \mathbf{c}_{\mathbf{st}(i,1),\mathbf{r}} - \mathbf{c}_{\mathbf{st}(i,2),\mathbf{r}} & \text{otherwise} \end{bmatrix} \\ \begin{bmatrix} \mathbf{P}^{*} & \mathbf{T}^{*} & \mathbf{T} & \mathbf{\rho}^{*} & \mathbf{k} & \mathbf{a}^{*}_{\mathbf{c}} & \mathbf{a}_{3\mathbf{B}} & \mathbf{c} & \mathbf{c}_{\mathbf{u}} & \mathbf{c}_{\mathbf{a}} & \mathbf{\Delta} \mathbf{c}_{\mathbf{a}} & \mathbf{c} & \mathbf{c}_{\mathbf{u}} \\ \mathbf{p} & \mathbf{T}^{*}_{\mathbf{w}} & \mathbf{T} & \mathbf{\rho} & \mathbf{R}_{\mathbf{L}} & \mathbf{a}^{*}_{\mathbf{w}} & \mathbf{a}_{3\mathbf{B}} & \mathbf{w} & \mathbf{w}_{\mathbf{u}} & \mathbf{w}_{\mathbf{u}} & \mathbf{w}_{\mathbf{d}} & \mathbf{c}_{\mathbf{u}} & \mathbf{c}_{\mathbf{u}} & \mathbf{c}_{\mathbf{u}} \end{bmatrix}^{\mathsf{T}} \\ \mathbf{c}_{\mathbf{u},\mathbf{v}} & \mathbf{c}_{\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} \\ \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} & \mathbf{c}_{\mathbf{u},\mathbf{v},\mathbf{v}} \end{bmatrix}$$

▲ Расчет Л по высоте

#### ▼ Результаты расчета Л по высоте

		1	2	3	
$P^{*T} =$	1	2731.8	2413.7	870.5	.10
_	2	2731.8	2413.7	870.5	10
	3	2731.8	2413.7	870.5	

		1	2	3	4	5	6	7	8	9
$T^{*T} =$	1	1773.0	1759.0	1394.2						
-	2	1773.0	1759.0	1394.2						
	3	1773.0	1759.0	1394.2						

		1	2	3	4	5	6	7	8	9
T* T =	1	1878.6	1493.1	1491.4						
1 W -	2	1888.0	1500.7	1508.0						
	3	1897.9	1508.9	1525.6						

$$\rho^{*T} = \begin{array}{|c|c|c|c|c|c|}\hline & 1 & 2 & 3 \\ 1 & 5.341 & 4.756 & 2.164 \\ 2 & 5.341 & 4.756 & 2.164 \\ 3 & 5.341 & 4.756 & 2.164 \\\hline \end{array}$$

$$k^{T} = \begin{bmatrix} & 1 & 2 & 3 \\ 1 & 1.305 & 1.305 & 1.316 \\ 2 & 1.305 & 1.305 & 1.316 \\ 3 & 1.305 & 1.305 & 1.316 \end{bmatrix}$$

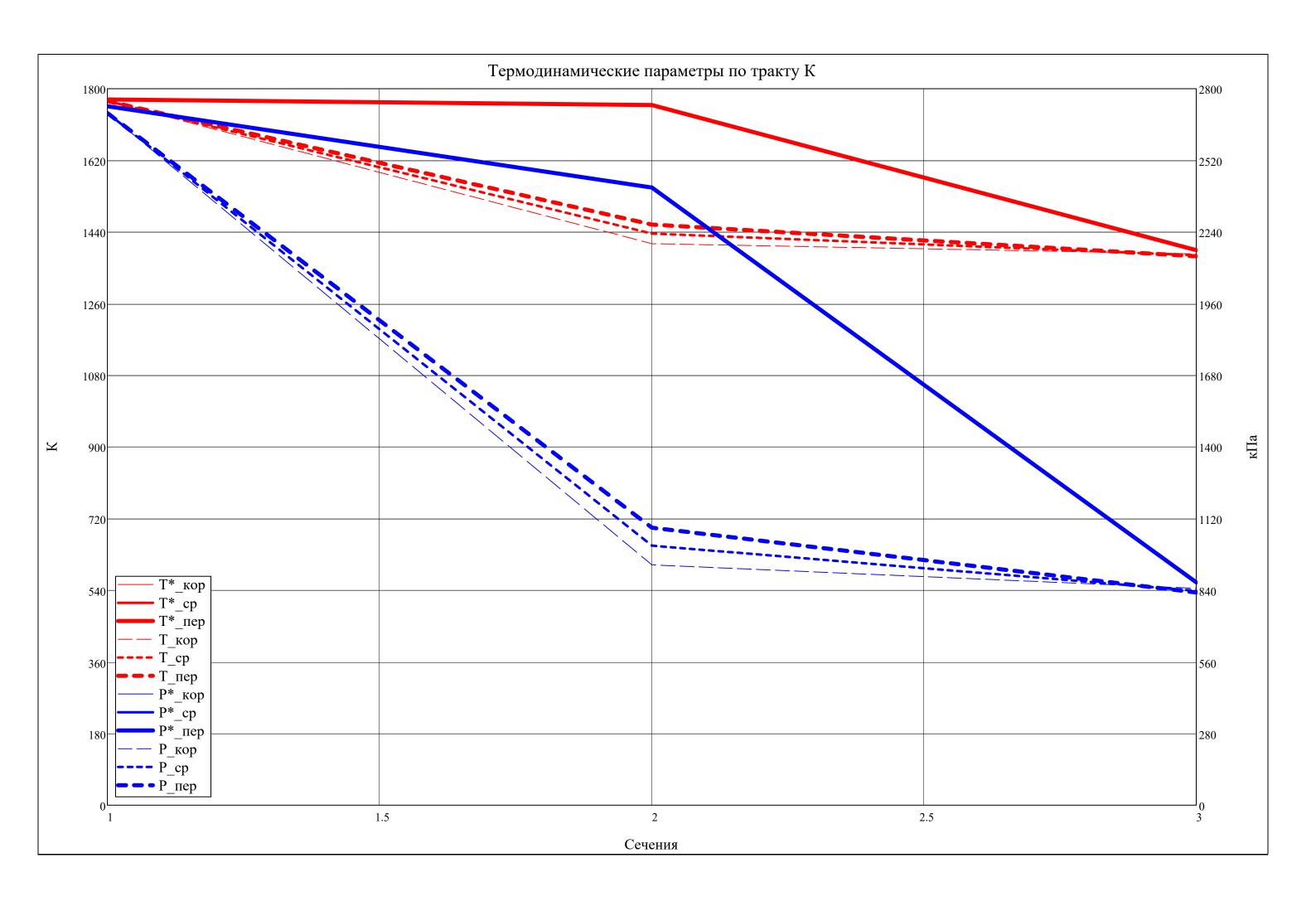
		1	
$R_{\tau}^{T} =$	1	0.0998	
T'L	2	0.1767	
	3	0.2440	

		1	2	3	
$\mathbf{P}^{\mathrm{T}} =$	1	2705.2	939.2	847.2	.10
_	2	2705.2	1014.0	838.1	10
	3	2705.2	1084.4	831.3	

		1	2	3	4	5	6	7	8	9
$T^{T} =$	1	1768.9	1410.5	1385.1						
-	2	1768.9	1436.0	1381.6						
	3	1768.9	1458.8	1378.9						

		1	2	3
$o^{T} =$	1	5.301	2.308	2.120
Р	2	5.301	2.448	2.103
	3	5.301	2.577	2.090

		1
$R_{\rm I}^{T} \ge 0.05 =$	1	1
TL = 0.00	2	1
	3	1



$$a^*c^T = \begin{array}{|c|c|c|c|c|c|}\hline & 1 & 2 & 3 \\ \hline 1 & 761.0 & 758.1 & 676.1 \\ \hline 2 & 761.0 & 758.1 & 676.1 \\ \hline 3 & 761.0 & 758.1 & 676.1 \\ \hline \end{array}$$

$$u^{T} = \begin{array}{|c|c|c|c|c|c|}\hline & 1 & 2 & 3 \\ 1 & 510.5 & 510.5 & 484.8 \\ 2 & 532.9 & 532.9 & 520.0 \\ \hline 3 & 555.2 & 555.2 & 555.2 \\ \hline \end{array}$$

$$c^T = \begin{array}{|c|c|c|c|c|c|}\hline & 1 & 2 & 3 \\ \hline 1 & 100.0 & 927.2 & 147.3 \\ \hline 2 & 100.0 & 892.5 & 174.1 \\ \hline 3 & 100.0 & 860.6 & 191.8 \\ \hline \end{array}$$

$$c_u^T = \begin{array}{|c|c|c|c|c|}\hline & 1 & 2 & 3 \\ \hline 1 & 0.0 & 897.7 & 1.8 \\ \hline 2 & 0.0 & 864.2 & -2.7 \\ \hline 3 & 0.0 & 833.2 & -6.4 \\ \hline \end{array}$$

$$c_{a}^{T} = \begin{array}{|c|c|c|c|c|c|}\hline & 1 & 2 & 3 \\\hline 1 & 100.0 & 231.9 & 147.3 \\\hline 2 & 100.0 & 223.3 & 174.0 \\\hline 3 & 100.0 & 215.3 & 191.7 \\\hline \end{array}$$

$$\Delta c_a^T = \begin{bmatrix} & 1 & 2 \\ 1 & 131.9 & -84.7 \\ 2 & 123.3 & -49.3 \\ 3 & 115.3 & -23.6 \end{bmatrix}$$

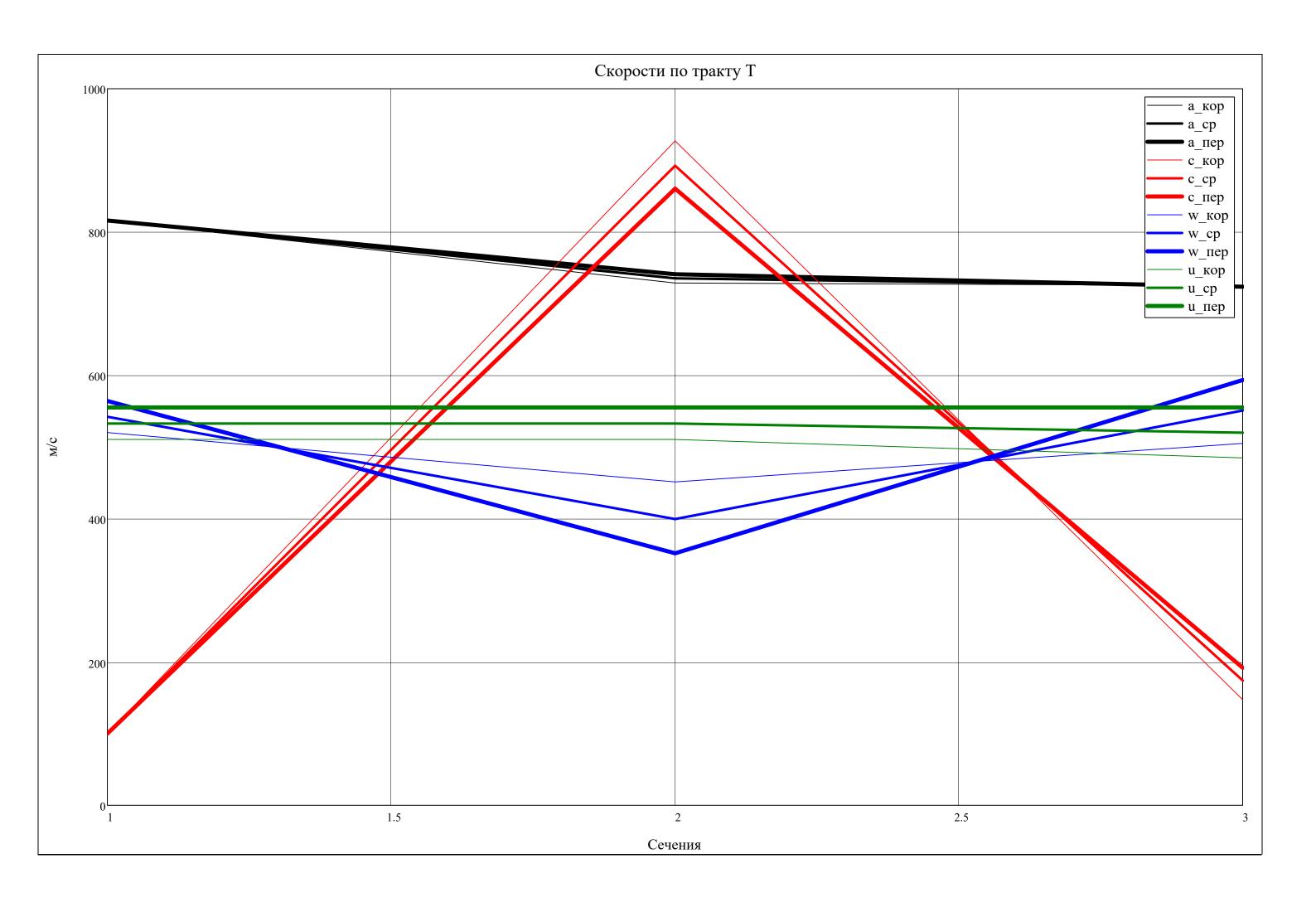
$a^*_W^T =$		1	2	3
	1	783.4	698.4	699.3
	2	785.3	700.2	703.2
	3	787.4	702.1	707.3

$a_{3B}^{T} =$		1	2	3
	1	816.1	728.8	725.3
	2	816.1	735.4	724.4
	3	816.1	741.2	723.7

		1	2	3
$\mathbf{w}^{\mathrm{T}} =$	1	520.2	451.4	504.9
••	2	542.2	399.5	550.9
	3	564.2	351.6	593.4

$$w_{u}^{T} = \begin{vmatrix} 1 & 2 & 3 \\ 1 & 510.5 & -387.2 & 483.0 \\ 2 & 532.9 & -331.3 & 522.7 \\ 3 & 555.2 & -278.0 & 561.6 \end{vmatrix}$$

$\mathbf{w}_{a}^{T} =$		1	2	3
	1	100.0	231.9	147.3
	2	100.0	223.3	174.0
	3	100.0	215.3	191.7



		1	2	3	
$\alpha^{T} =$	1	90.00	14.49	89.31	. 0
33	2	90.00	14.49	90.87	
	3	90.00	14.49	91.90	

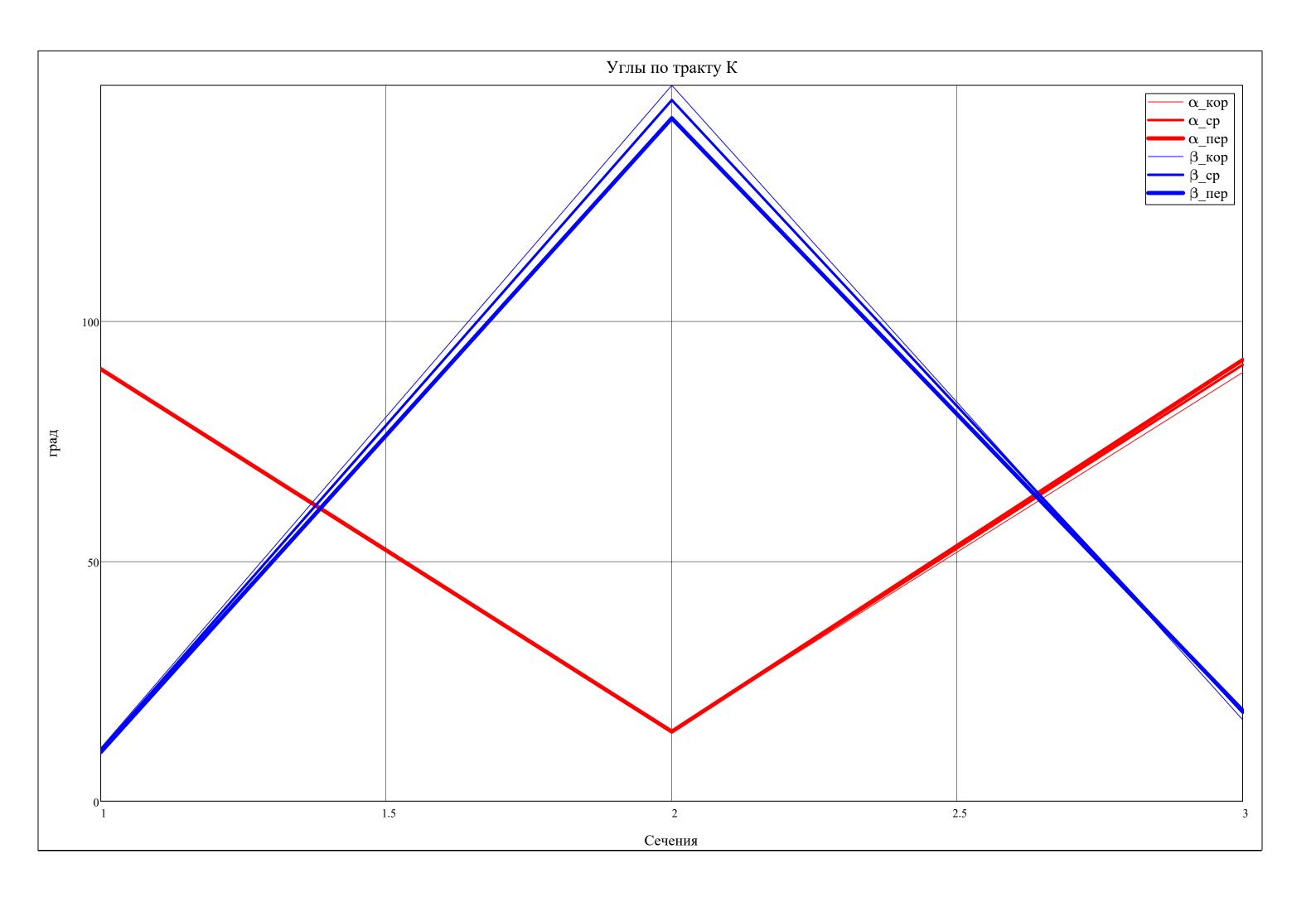
		1	2	3	
$80^{\circ} \leq \alpha^{T} =$	1	1	0	1	
	2	1	0	1	
	3	1	0	1	

		1	
$\varepsilon_{ m stator}^{ m T} =$	1	75.51	.0
stator	2	75.51	
	3	75.51	

[1, c.78]

Угол поворота потока:

$$\varepsilon_{\text{rotor}}^{\text{T}} = \begin{vmatrix}
 & 1 \\
1 & 132.12 \\
2 & 127.61 \\
3 & 123.40
\end{vmatrix}$$

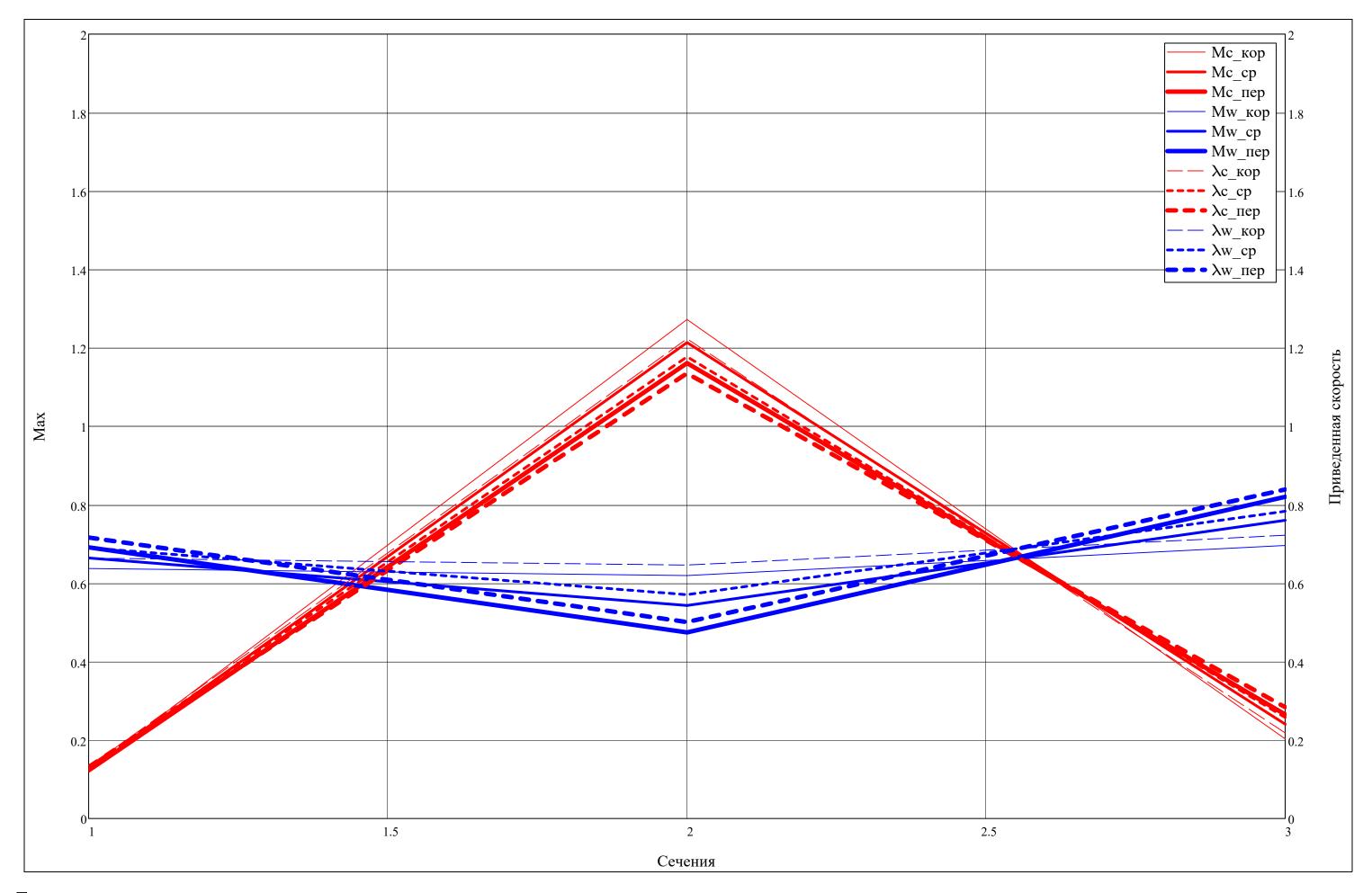


$$\lambda_{c}^{T} = \begin{array}{|c|c|c|c|c|c|}\hline & 1 & 2 & 3 \\ 1 & 0.131 & 1.223 & 0.218 \\ \hline 2 & 0.131 & 1.177 & 0.257 \\ \hline 3 & 0.131 & 1.135 & 0.284 \\\hline \end{array}$$

$$\mathbf{M_c}^T = \begin{array}{|c|c|c|c|c|c|}\hline & 1 & 2 & 3 \\ \hline 1 & 0.123 & 1.272 & 0.203 \\ \hline 2 & 0.123 & 1.214 & 0.240 \\ \hline 3 & 0.123 & 1.161 & 0.265 \\ \hline \end{array}$$

$\lambda_{W}^{T} =$		1	2	3
	1	0.664	0.646	0.722
	2	0.690	0.571	0.783
	3	0.717	0.501	0.839

$M_{W}^{T} =$		1	2	3
	1	0.637	0.619	0.696
	2	0.664	0.543	0.760
	3	0.691	0.474	0.820



D. SAROO, perynatarce pecuera nagamentpos notoce no secone il:

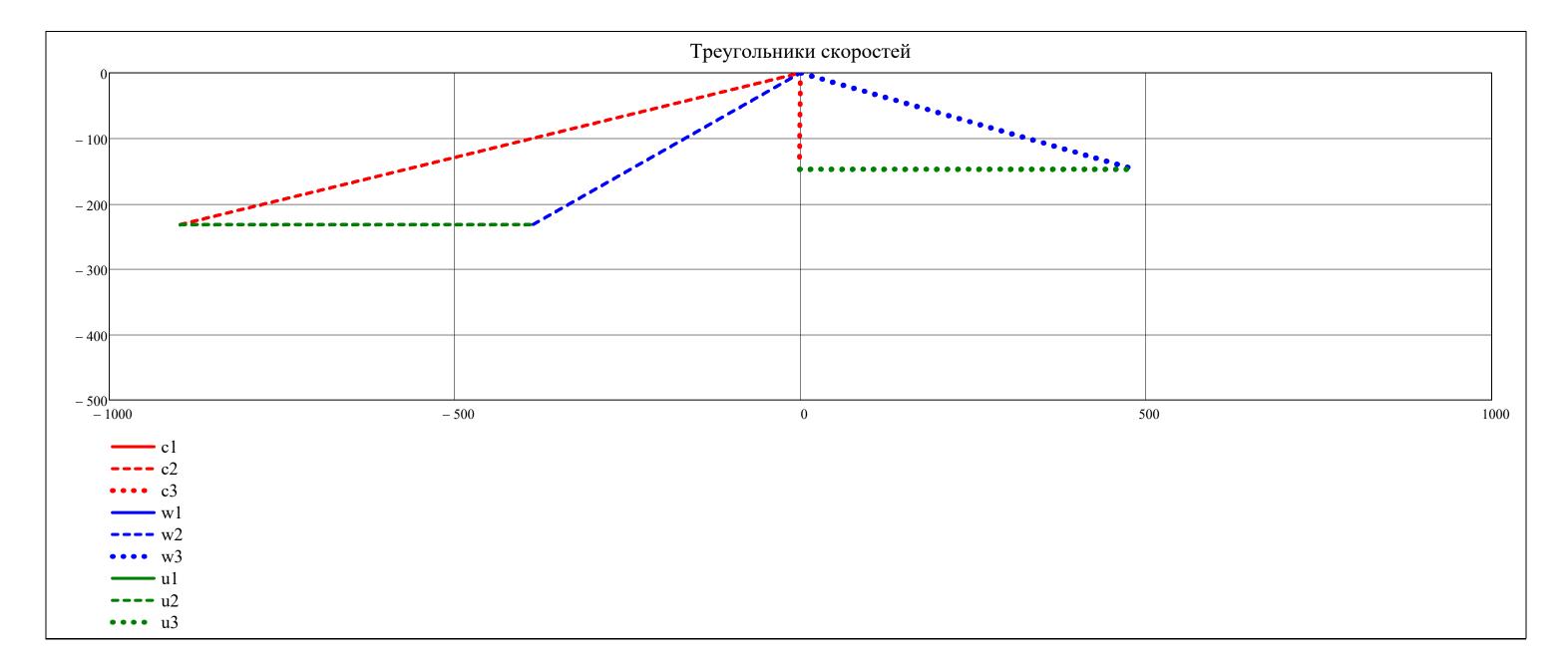
Рассматриваемая ступень: 
$$j = \begin{cases} j = Z \\ j = \end{cases}$$
 "Такой ступени не существует!" if  $(j < 1) \lor (j > Z)$   $j$  otherwise

#### ▼ Построение треугольников скоростей в 3х сечениях

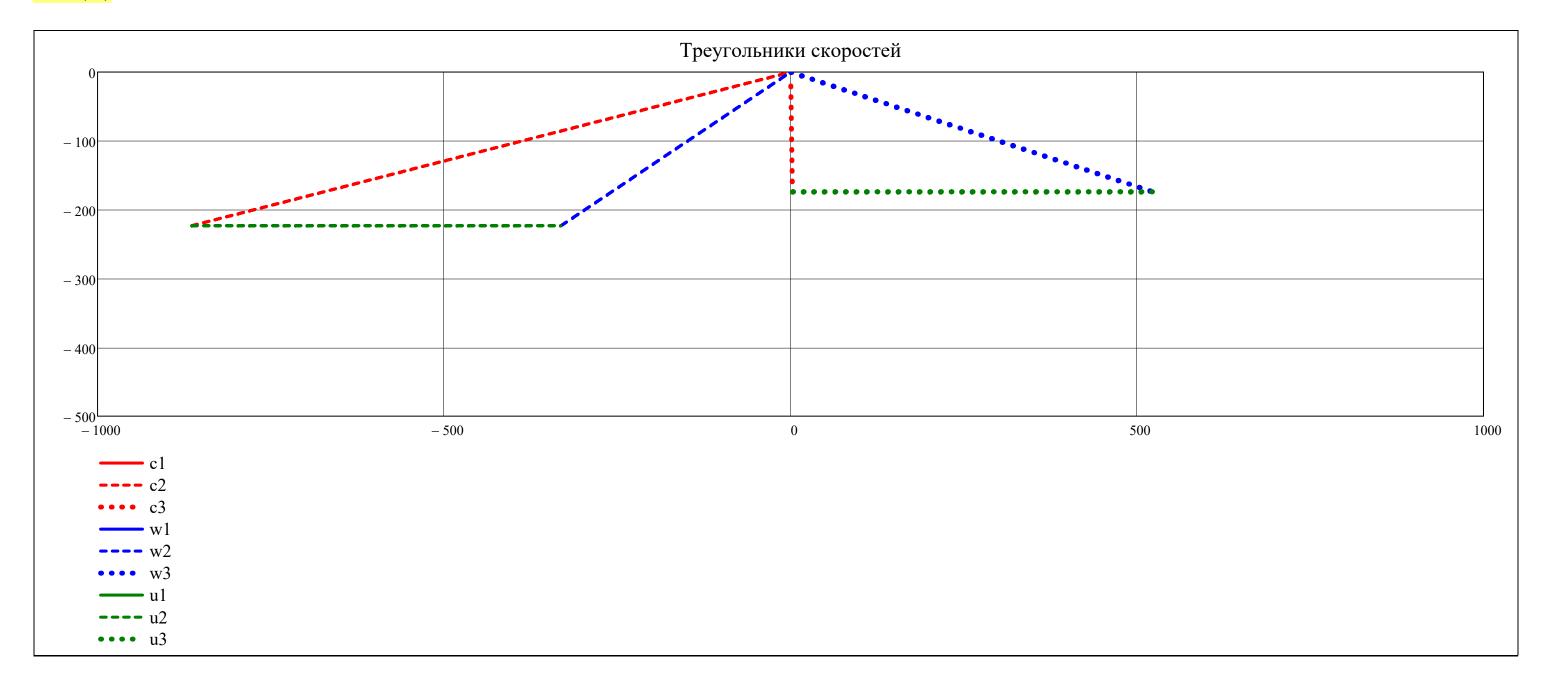
$$\begin{split} \Delta_c(v,i,j,r) &= \left| \begin{array}{l} \tan(\alpha_{st(i,j),r}) \cdot v & \text{if } \left(\tan(\alpha_{st(i,j),r}) \geq 0 \right) \wedge \left(-\left|c_{st(i,j),r} \cdot \cos(\alpha_{st(i,j),r})\right| \leq v \leq 0 \right) \\ & \tan(\alpha_{st(i,j),r}) \cdot v & \text{if } \left(\tan(\alpha_{st(i,j),r}) < 0 \right) \wedge \left(0 \leq v \leq \left|c_{st(i,j),r} \cdot \cos(\alpha_{st(i,j),r})\right| \right) \\ \Delta_W(v,i,j,r) &= \left| -\tan(\beta_{st(i,j),r}) \cdot v & \text{if } \left(-\tan(\beta_{st(i,j),r}) \geq 0 \right) \wedge \left(-\left|w_{st(i,j),r} \cdot \cos(\beta_{st(i,j),r})\right| \leq v \leq 0 \right) \wedge \left(j \neq 1 \right) \\ & -\tan(\beta_{st(i,j),r}) \cdot v & \text{if } \left(-\tan(\beta_{st(i,j),r}) < 0 \right) \wedge \left(0 \leq v \leq \left|w_{st(i,j),r} \cdot \cos(\beta_{st(i,j),r})\right| \right) \wedge \left(j \neq 1 \right) \\ \Delta_U(v,i,j,r) &= \left| -c_{a_{st(i,j),r}} & \text{if } \left(-c_{st(i,j),r} \cdot \cos(\alpha_{st(i,j),r}) \leq v \leq w_{st(i,j),r} \cdot \cos(\beta_{st(i,j),r}) \right) \wedge \left(j \neq 1 \right) \\ & \text{NaN otherwise} \end{split}$$

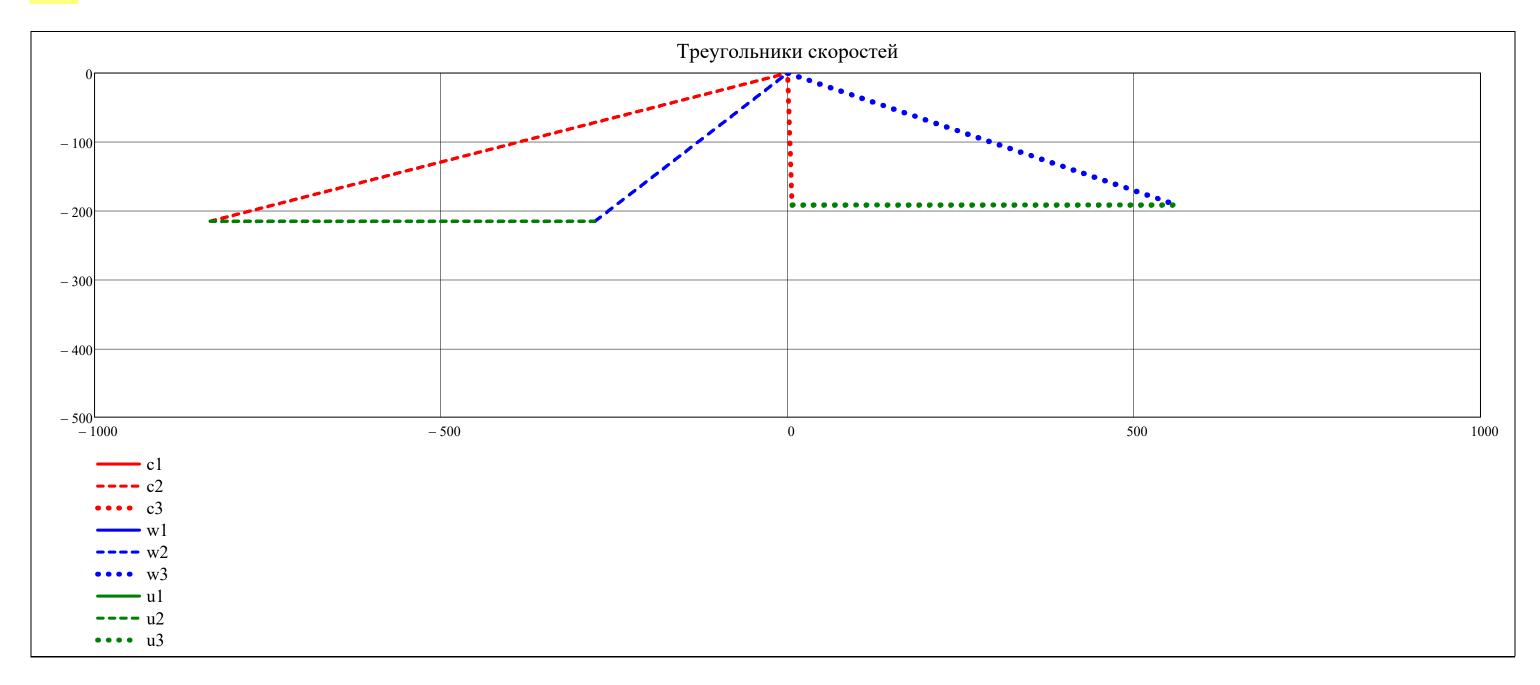
$$v_{lim} = ceil \left( \frac{max(c, w, u)}{10^2} \right) \cdot 10^2 = 1000.0$$

$$v = -max(c, w, u), -max(c, w, u) + \frac{max(c, w, u)}{3000} ... max(c, w, u)$$









■ Построение треугольников скоростей в 3х сечениях

Парусность: 
$$\begin{pmatrix} sail_{stator} \\ sail_{rotor} \end{pmatrix} = \begin{pmatrix} 1 \\ 0.8 \end{pmatrix}$$

▼ Расчет хорд Л по парусности

Длины хорд РК и СА (м):

$$chord_{stator}^{T} = \begin{vmatrix} 1 & 1 \\ 1 & 68.0 \\ 2 & 68.0 \\ 3 & 68.0 \end{vmatrix} \cdot 10^{-3}$$

$$\bar{x}_{f} = 0.45$$

▼ Расчет параметров решетки

$$\begin{bmatrix} s_{tator} & r_{totor} \\ r_{inlet} \\ s_{tator} & r_{inlet} \\ r_{outlet} \\ s_{tator} & r_{outlet} \\ r_{outlet$$

$$\begin{pmatrix} \upsilon_{\text{установки}}(\alpha_{\text{st(i,1),r}},\alpha_{\text{st(i,2),r}}) \\ \upsilon_{\text{установки}}(\beta_{\text{st(i,2),r}},\beta_{\text{st(i,3),r}}) \end{pmatrix}$$

 $\frac{\pi}{2}$  добавляется в виду поворота рисунка на 90 град

### Относительные радиусы профилей ():

$$\frac{T}{\text{r_inlet}_{\text{stator}}} = \begin{vmatrix}
 & 1 & \\
 & 1 & 6.000 \\
 & 2 & 6.000 \\
 & 3 & 6.000
\end{vmatrix} .\%$$

$$\frac{T}{r\_inlet_{rotor}} = \begin{bmatrix}
 & 1 \\
1 & 5.950 \\
2 & 4.550 \\
3 & 3.850
\end{bmatrix} .\%$$

$$\frac{T}{r_{outlet}} = \begin{vmatrix}
 & 1 & \\
 & 1 & 2.550 \\
 & 2 & 1.950 \\
 & 3 & 1.650
\end{vmatrix}$$

### Относительная толщина профиля ():

$$\overline{c}_{stator}^{T} = \begin{bmatrix}
 & 1 \\
1 & 15.00 \\
2 & 15.00 \\
3 & 15.00
\end{bmatrix}$$

$$\overline{c}_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & 17.00\\ 2 & 13.00\\ \hline 3 & 11.00 \end{bmatrix} \cdot \%$$

### Относительный шаг решетки ():

$$\left(\frac{t_{stator}}{chord_{stator}}\right)^{T} = \begin{vmatrix} 1\\ 1 & 0.8345\\ 2 & 0.8711\\ 3 & 0.9076 \end{vmatrix}$$

$$\left(\frac{t_{rotor}}{chord_{rotor}}\right)^{T} = \begin{vmatrix} 1 & 1 \\ 1 & 0.5829 \\ 2 & 0.6918 \\ \hline 3 & 0.7941 \end{vmatrix}$$

# Относительная густота решетки ():

$$\left(\frac{\text{chord}_{\text{stator}}}{t_{\text{stator}}}\right)^{\text{T}} = \begin{vmatrix} 1 & 1.198 \\ 2 & 1.148 \\ 3 & 1.102 \end{vmatrix}$$

$$\left(\frac{\text{chord}_{\text{rotor}}}{t_{\text{rotor}}}\right)^{\text{T}} = \begin{bmatrix} & 1\\ 1 & 1.716\\ 2 & 1.445\\ \hline 3 & 1.259 \end{bmatrix}$$

### Длина хорды профиля [м]:

$$chord_{stator}^{T} = \begin{bmatrix} 1 \\ 1 \\ 68.0 \\ 2 \\ 68.0 \\ 3 \\ 68.0 \end{bmatrix} \cdot 10^{-3}$$

$$chord_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & 38.4\\ 2 & 34.2\\ \hline 3 & 31.4 \end{bmatrix} \cdot 10^{-3}$$

### Радиусы профилей:

$$r_{inlet_{rotor}}^{T} = \begin{bmatrix} & 1 \\ 1 & 2.28 \\ 2 & 1.56 \\ \hline 3 & 1.21 \end{bmatrix} \cdot 10^{-3}$$

$$r\_outlet_{stator}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 2.04 \\ 2 & 2.04 \\ 3 & 2.04 \end{bmatrix} \cdot 10^{-3}$$

$$r\_outlet_{rotor}^{T} = \begin{bmatrix} 1 \\ 1 \\ 2 \\ 0.67 \\ 3 \\ 0.52 \end{bmatrix} \cdot 10^{-3}$$

### Толщина профиля [м]:

$$c_{\text{stator}}^{\text{T}} = \begin{bmatrix} 1 \\ 1 & 10.20 \\ 2 & 10.20 \\ 3 & 10.20 \end{bmatrix} \cdot 10^{-3}$$

$$c_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & 6.52\\ 2 & 4.45\\ \hline 3 & 3.46 \end{bmatrix} \cdot 10^{-3}$$

## Шаг решетки [м]:

$$t_{\text{stator}}^{\text{T}} = \begin{bmatrix} 1\\1&56.7\\2&59.2\\3&61.7 \end{bmatrix} \cdot 10^{-3}$$

$$t_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & 22.4\\ 2 & 23.7\\ \hline 3 & 25.0 \end{bmatrix} \cdot 10^{-3}$$

Угол поворота потока:	$\varepsilon_{\text{stator}}^{\text{T}} = \begin{bmatrix} & & 1 & \\ & 1 & 75.51 \\ & 2 & 75.51 \\ \hline & 3 & 75.51 \end{bmatrix} \cdot \circ$	$\varepsilon_{\text{rotor}}^{\text{T}} = \begin{bmatrix} & & 1 & \\ & 1 & 132.12 \\ & 2 & 127.61 \\ & 3 & 123.40 \end{bmatrix} . \circ$
Угол установки профиля:	$v_{\text{stator}}^{\text{T}} = \begin{bmatrix} 1 \\ 1 \\ 2 \\ 117.3 \\ \hline 3 \\ 117.3 \end{bmatrix} . \circ$	$v_{rotor}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 112.0 \\ 2 & 114.0 \\ 3 & 115.4 \end{bmatrix} \cdot \circ$
Угол изгиба профиля:	$\pi - \varepsilon_{stator}^{T} = \begin{bmatrix} & 1 & \\ 1 & 104.5 \\ \hline 2 & 104.5 \\ \hline 3 & 104.5 \end{bmatrix} \cdot \circ$	$\pi - \varepsilon_{\text{rotor}}^{\text{T}} = \begin{bmatrix} 1 \\ 1 \\ 47.9 \\ 2 \\ 52.4 \\ 3 \\ 56.6 \end{bmatrix}$

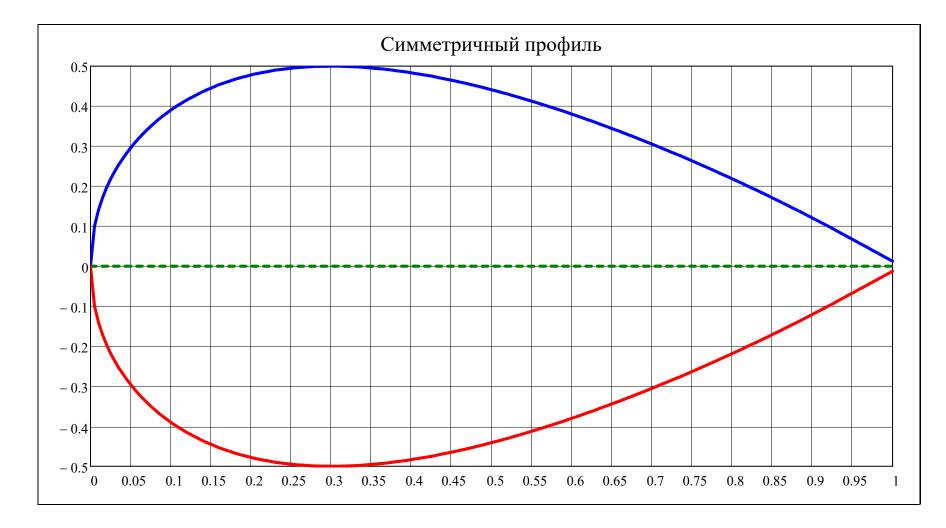
.0

## ▼ Подключение симметричного профиля

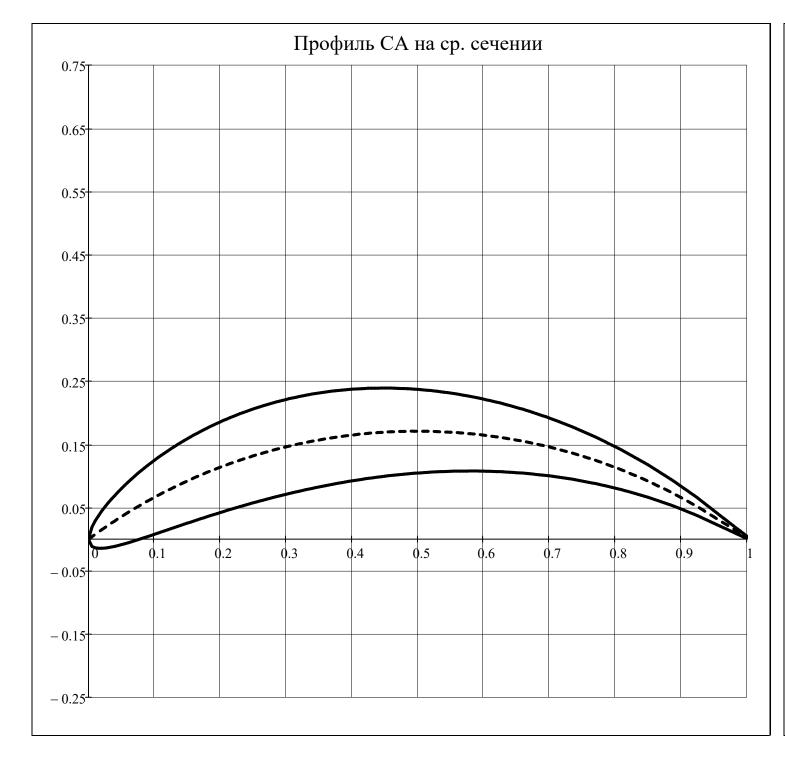
$$\begin{pmatrix} X_{U} & Y_{U} \\ X_{L} & Y_{L} \end{pmatrix} = NACA(0, 0, 100\%, 1)$$

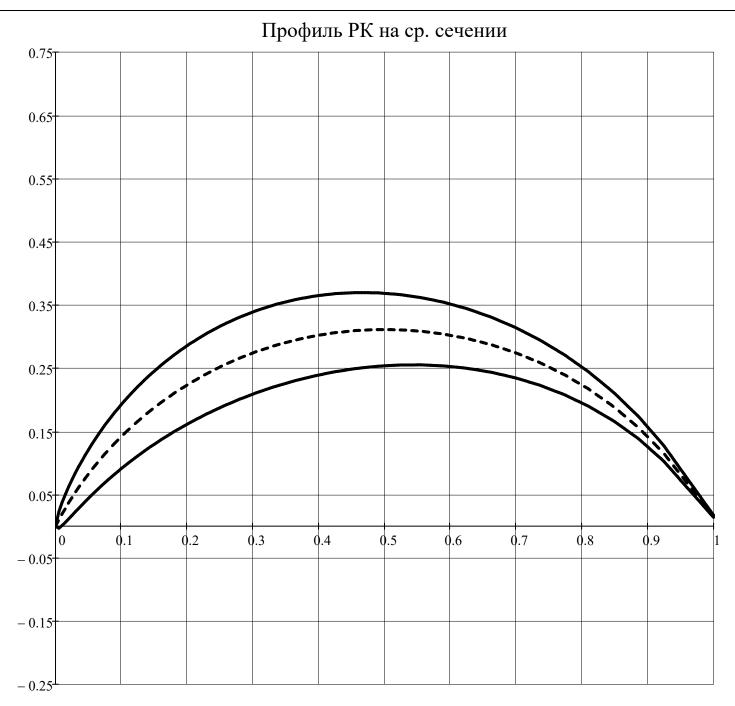
Относ. координаты профиля РК и СА:

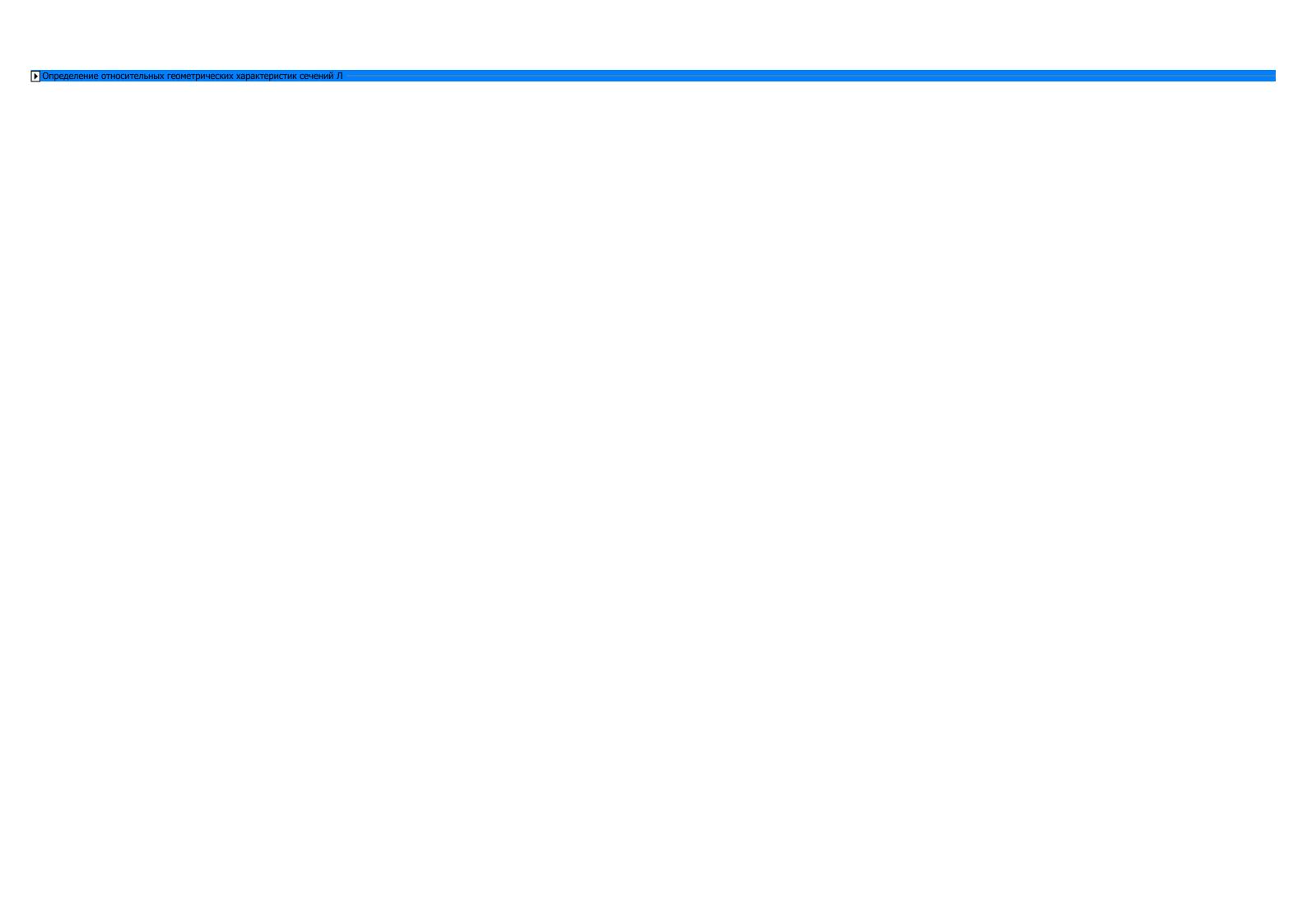
$$\begin{split} \text{AIRFOIL}_0\Big(x, \text{line}\,\overline{,f}\,, \overline{x}_f, \overline{c}\,\Big) &= & \text{if } 0 \leq x \leq 1 \\ & & \text{linterp}\big(X_U, Y_U, x\big) \text{ if line} = "+" \\ & \frac{\text{linterp}\big(X_U, Y_U, x\big) + \text{linterp}\big(X_L, Y_L, x\big)}{2} \text{ if line} = "0" \\ & & \text{linterp}\big(X_L, Y_L, x\big) \text{ if line} = "-" \\ & & \text{NaN otherwise} \end{split}$$



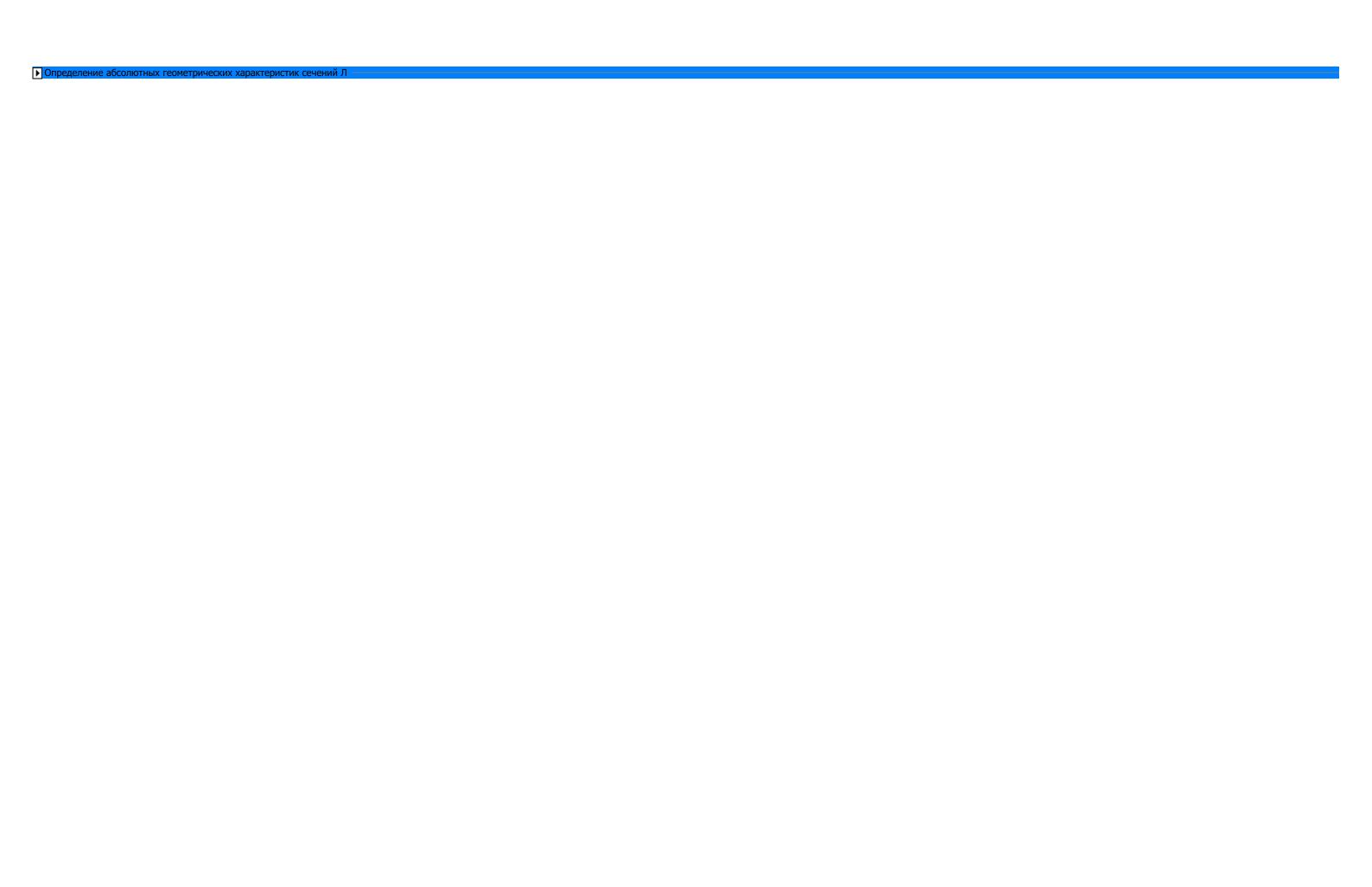
$$\begin{split} \text{AIRFOIL}(x,\text{line},\overline{c}^-,\theta) &= \begin{vmatrix} \text{linterp}\big(X_U,y/b_{cp.\Pi}\big(X_U,\theta\big) + Y_U\cdot\overline{c}^-,x\big) & \text{if line} = "+" \\ \frac{\text{linterp}\big(X_U,y/b_{cp.\Pi}\big(X_U,\theta\big) + Y_U\cdot\overline{c}^-,x\big) + \text{linterp}\big(X_L,y/b_{cp.\Pi}\big(X_L,\theta\big) + Y_L\cdot\overline{c}^-,x\big)}{2} & \text{if line} = "0" \\ \frac{\text{linterp}\big(X_L,y/b_{cp.\Pi}\big(X_L,\theta\big) + Y_L\cdot\overline{c}^-,x\big) + \text{linterp}\big(X_L,y/b_{cp.\Pi}\big(X_L,\theta\big) + Y_L\cdot\overline{c}^-,x\big)}{2} & \text{if line} = "-" \\ \text{NaN otherwise} & \end{aligned}$$







▶ Результат расчета абсолютных геометрических характеристик сечений Л



#### ▼ Результат расчета абсолютных геометрических характеристик сечений Л

$$1\_lower_{stator}^{T} = \begin{bmatrix} & 1 \\ 1 & 70.64 \\ 2 & 70.64 \\ \hline 3 & 70.64 \end{bmatrix} \cdot 10^{-3}$$

$$\operatorname{area}_{\text{stator}}^{\text{T}} = \begin{bmatrix} & 1\\ 1 & 473.87\\ 2 & 473.87\\ 3 & 473.87 \end{bmatrix} \cdot 10^{-6}$$

$$Sx_{stator}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 4232.2 \\ 2 & 4232.2 \\ 3 & 4232.2 \end{bmatrix} \cdot 10^{-9}$$

$$Sy_{\text{stator}}^{\text{T}} = \begin{bmatrix} & 1\\ 1 & 13563.5\\ \hline 2 & 13563.5\\ \hline 3 & 13563.5 \end{bmatrix} \cdot 10^{-9}$$

$$x0_{\text{stator}}^{\text{T}} = \begin{array}{|c|c|c|c|c|}\hline & 1 & \\ 1 & 28.6 \\ \hline 2 & 28.6 \\ \hline 3 & 28.6 \\ \hline \end{array} \cdot 10^{-3}$$

$$y0_{\text{stator}}^{\text{T}} = \frac{\begin{vmatrix} 1 \\ 1 & 8.9 \\ 2 & 8.9 \\ 3 & 8.9 \end{vmatrix}}{2 & 8.9} \cdot 10^{-3}$$

$$1\_upper_{rotor}^{T} = \begin{bmatrix} & & 1 \\ 1 & 52.01 \\ 2 & 44.78 \\ \hline 3 & 40.21 \end{bmatrix} \cdot 10^{-3}$$

$$1\_lower_{rotor}^{T} = \begin{array}{|c|c|c|c|}\hline & 1 \\ \hline 1 & 44.26 \\ \hline 2 & 39.55 \\ \hline 3 & 36.21 \\ \hline \end{array} \cdot 10^{-3}$$

$$area_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & 171.14\\ 2 & 103.96\\ \hline 3 & 74.28 \end{bmatrix} \cdot 10^{-6}$$

$$Sx_{rotor}^{T} = \begin{bmatrix} 1 \\ 1 & 1711.6 \\ 2 & 883.3 \\ 3 & 554.2 \end{bmatrix} \cdot 10^{-9}$$

$$Sy_{rotor}^{T} = \begin{array}{|c|c|c|}\hline & 1 \\ \hline 1 & 2765.3 \\ \hline 2 & 1497.1 \\ \hline 3 & 983.0 \\ \hline \end{array} \cdot 10^{-9}$$

$$x0_{rotor}^{T} = \begin{array}{|c|c|c|}\hline & 1 \\\hline 1 & 16.2 \\\hline 2 & 14.4 \\\hline 3 & 13.2 \\\hline \end{array} \cdot 10^{-3}$$

$$y0_{rotor}^{T} = \begin{array}{|c|c|c|c|}\hline & 1 & \\ \hline 1 & 10.0 \\ \hline 2 & 8.5 \\ \hline 3 & 7.5 \\ \hline \end{array} \cdot 10^{-3}$$

$$Jx_{stator}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 44264 \\ 2 & 44264 \\ \hline 3 & 44264 \end{bmatrix} \cdot 10^{-12}$$

$$Jy_{\text{stator}}^{\text{T}} = \begin{vmatrix} 1 \\ 1 & 508717 \\ 2 & 508717 \\ \hline 3 & 508717 \end{vmatrix} \cdot 10^{-12}$$

$$Jxy_{stator}^{T} = \begin{array}{|c|c|c|}\hline 1 & 1 \\\hline 1 & 128548 \\\hline 2 & 128548 \\\hline 3 & 128548 \\\hline \end{array} \cdot 10^{-12}$$

$$Jx0_{stator}^{T} = \begin{vmatrix} 1 & 1 \\ 1 & 6465 \\ 2 & 6465 \\ \hline 3 & 6465 \end{vmatrix} \cdot 10^{-12}$$

$$Jy0_{\text{stator}}^{\text{T}} = \begin{array}{|c|c|c|c|c|}\hline & 1 & \\ \hline 1 & 120489 \\ \hline 2 & 120489 \\ \hline 3 & 120489 \\ \hline \end{array} \cdot 10^{-12}$$

$$Jxy0_{stator}^{T} = \begin{bmatrix} & 1\\ 1 & 7409\\ 2 & 7409\\ \hline 3 & 7409 \end{bmatrix} \cdot 10^{-12}$$

$$\alpha_{\text{major}_{\text{stator}}}^{\text{T}} = \begin{vmatrix}
 & 1 \\
 & 1 & 3.70 \\
 & 2 & 3.70 \\
 & 3 & 3.70
\end{vmatrix}$$

$$Jx_{rotor}^{T} = \begin{bmatrix} 1\\1\\1\\2\\8184\\3\\4505 \end{bmatrix} \cdot 10^{-12}$$

$$Jy_{rotor}^{T} = \begin{bmatrix} 1\\1\\58548\\2\\28251\\3&17045 \end{bmatrix} \cdot 10^{-12}$$

$$Jxy_{rotor}^{T} = \begin{bmatrix} 1 \\ 1 & 29122 \\ 2 & 13406 \\ \hline 3 & 7736 \end{bmatrix} \cdot 10^{-12}$$

$$Jy0_{rotor}^{T} = \begin{bmatrix} 1 \\ 1 \\ 1 \\ 2 \\ 6691 \\ 3 \\ 4037 \end{bmatrix} \cdot 10^{-12}$$

$$Jxy0_{rotor}^{T} = \begin{bmatrix} 1 \\ 1 & 1466 \\ 2 & 686 \\ \hline 3 & 402 \end{bmatrix} \cdot 10^{-12}$$

$$\alpha_{\text{major}_{\text{rotor}}}^{\text{T}} = \begin{vmatrix}
 & 1 \\
 & 1 & 6.75 \\
 & 2 & 6.43 \\
 & 3 & 6.18
\end{vmatrix}$$
.

$$Ju_{stator}^{T} = \begin{bmatrix} & 1 \\ 1 & 5986 \\ 2 & 5986 \\ \hline 3 & 5986 \end{bmatrix} \cdot 10^{-12}$$

$$Jv_{\text{stator}}^{\text{T}} = \begin{array}{|c|c|c|c|}\hline & 1 & \\ \hline 1 & 120968.8 \\ \hline 2 & 120968.8 \\ \hline 3 & 120968.8 \\ \hline \end{array} \cdot 10^{-12}$$

$$Juv_{stator}^{T} = \begin{vmatrix} & 1 \\ 1 & -0 \\ 2 & -0 \\ \hline 3 & -0 \end{vmatrix} \cdot 10^{-12}$$

$$Jp_{stator}^{T} = \begin{array}{|c|c|c|c|c|}\hline & 1 & \\ 1 & 126955 \\ \hline 2 & 126955 \\ \hline 3 & 126955 \\ \hline \end{array} \cdot 10^{-12}$$

$$Wp_{stator}^{T} = \begin{array}{|c|c|c|}\hline & 1 \\ \hline 1 & 3146.4 \\ \hline 2 & 3146.4 \\ \hline 3 & 3146.4 \\ \hline \end{array} \cdot 10^{-9}$$

$$Ju_{rotor}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 1482 \\ 2 & 602 \\ 3 & 326 \end{bmatrix} \cdot 10^{-12}$$

$$Jv_{rotor}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 14041 \\ 2 & 6769 \\ \hline 3 & 4081 \end{bmatrix} \cdot 10^{-12}$$

$$Juv_{rotor}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 0 \\ 2 & 0 \\ 3 & 0 \end{bmatrix} \cdot 10^{-12}$$

$$Jp_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & 15522\\ 2 & 7371\\ 3 & 4407 \end{bmatrix} \cdot 10^{-12}$$

$$Wp_{rotor}^{T} = \begin{bmatrix} 1\\1&637.2\\2&342.1\\3&224.1 \end{bmatrix} \cdot 10^{-9}$$

stiffness<sub>rotor</sub><sup>T</sup> = 
$$\begin{vmatrix} 1 & 1 \\ 1 & 1676.5 \\ 2 & 473.1 \\ 3 & 204.3 \end{vmatrix} \cdot 10^{-12}$$

		1	
$CPx_{stator}^{T} =$	1	23.790	$\cdot 10^{-3}$
Stator	2	23.790	10
	3	23.790	

$$CPx_{rotor}^{T} = \begin{bmatrix} & 1 & \\ 1 & 13.430 \\ \hline 2 & 11.969 \\ \hline 3 & 10.999 \end{bmatrix} \cdot 10^{-3}$$

$$CPy_{stator}^{T} = \begin{bmatrix} & 1 & \\ 1 & 0.0000 \\ \hline 2 & 0.0000 \\ \hline 3 & 0.0000 \end{bmatrix} \cdot 10^{-3}$$

$$CPy_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & 0.0000\\ \hline 2 & 0.0000\\ \hline 3 & 0.0000 \end{bmatrix} \cdot 10^{-3}$$

Результат расчета абсолютных геометрических характеристик сечений Л

Вывод результатов расчета геометрических хар-к сечений Л

### ▼ Профилирование решеток

Абс. координаты профиля:

▲ Профилирование решеток

Рассматриваемая ступень: j = j = Z

$$j_w = \begin{cases} j = Z \end{cases}$$
 = 1
$$j = \begin{cases} \text{"Такой ступени не существует!"} & \text{if } (j < 1) \lor (j > Z) \end{cases}$$

$$j = \begin{cases} j = Z \end{cases}$$

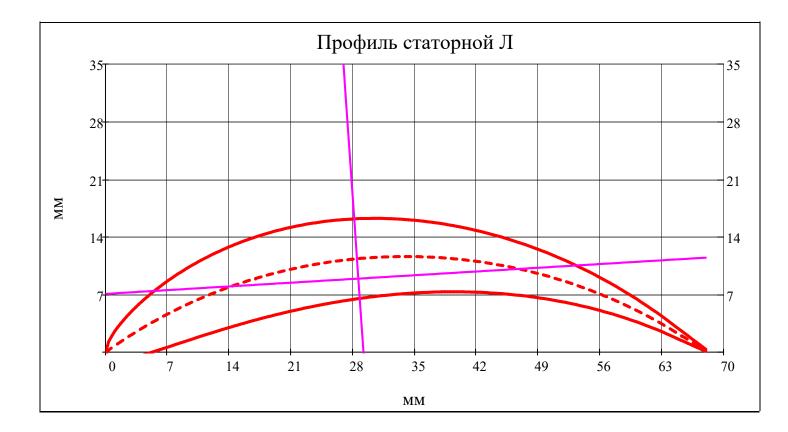
### ▼ Построение профилей Л РК и НА

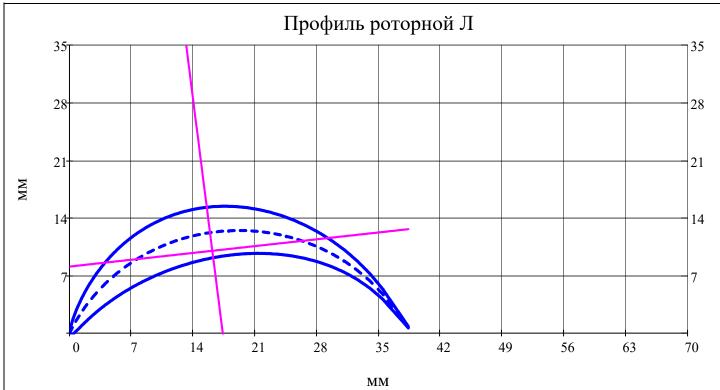
NaN otherwise

$$\begin{aligned} \text{AXLE0(type}, \mathbf{x}, \mathbf{i}, \mathbf{r}) &= \frac{y0_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}}{\text{chord}_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}} + \tan\left(\alpha_{-}\text{major}_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}\right) \cdot \left(\mathbf{x} - \frac{\mathbf{x}0_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}}{\text{chord}_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}}\right) & \text{if type} = \text{"rotor"} \\ \frac{y0_{stator_{\hat{\mathbf{i}}, \mathbf{r}}}}{\text{chord}_{stator_{\hat{\mathbf{i}}, \mathbf{r}}}} + \tan\left(\alpha_{-}\text{major}_{stator_{\hat{\mathbf{i}}, \mathbf{r}}}\right) \cdot \left(\mathbf{x} - \frac{\mathbf{x}0_{stator_{\hat{\mathbf{i}}, \mathbf{r}}}}{\text{chord}_{stator_{\hat{\mathbf{i}}, \mathbf{r}}}}\right) & \text{if type} = \text{"stator"} \\ \text{NaN otherwise} \end{aligned}$$

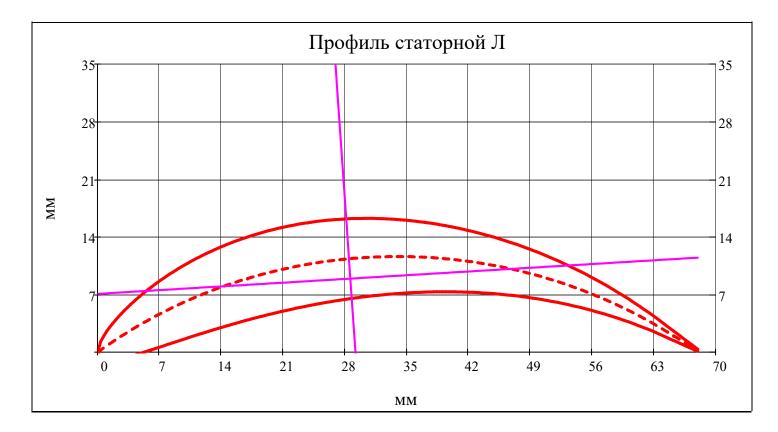
$$\text{AXLE90(type}, \mathbf{x}, \mathbf{i}, \mathbf{r}) &= \frac{y0_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}}{\text{chord}_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}} + \tan\left(\alpha_{-}\text{major}_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}} + \frac{\pi}{2}\right) \cdot \left(\mathbf{x} - \frac{\mathbf{x}0_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}}{\text{chord}_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}}\right) & \text{if (type} = \text{"rotor"}) \land \left|\alpha_{-}\text{major}_{rotor_{\hat{\mathbf{i}}, \mathbf{r}}}\right| \geq 1.^{\circ} \\ y0_{stator} &= \sqrt{(\mathbf{x}0_{stator}, \mathbf{r})} & \sqrt{(\mathbf$$

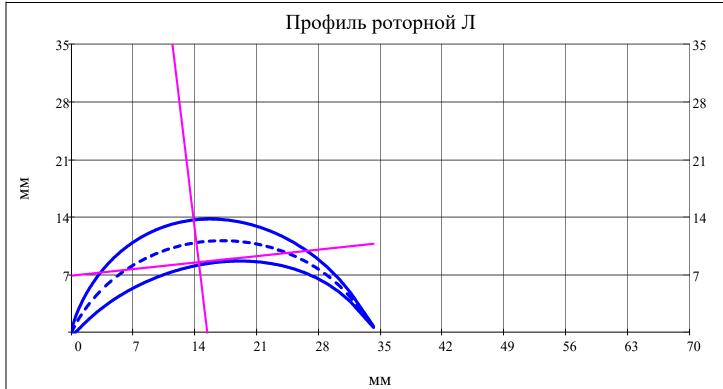
$$b_{lim} = \frac{\text{ceil}\left(\text{max}\left(\text{chord}_{rotor_{j,N_r}}, \text{chord}_{stator_{j,N_r}}\right) \cdot 10^2\right)}{10^2} = 70.0 \cdot 10^{-3}$$



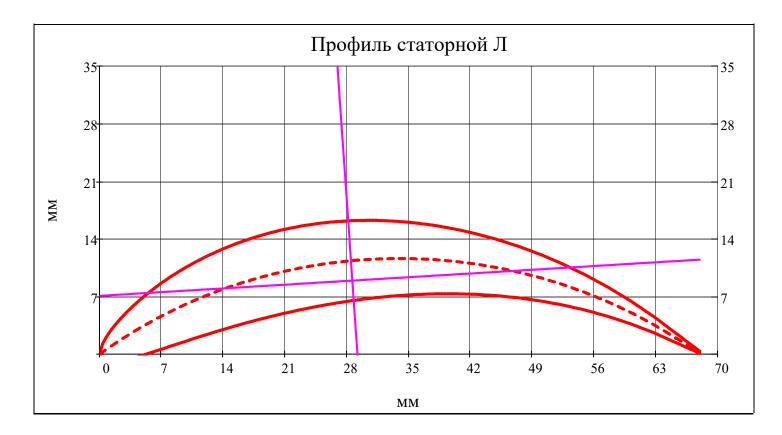


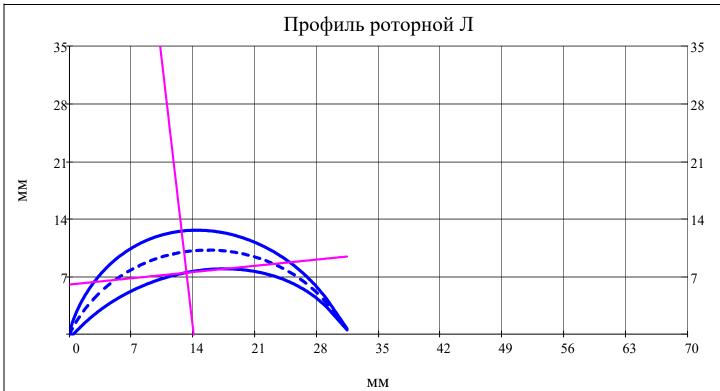
# $rac{r}{m} = av(N_r)$











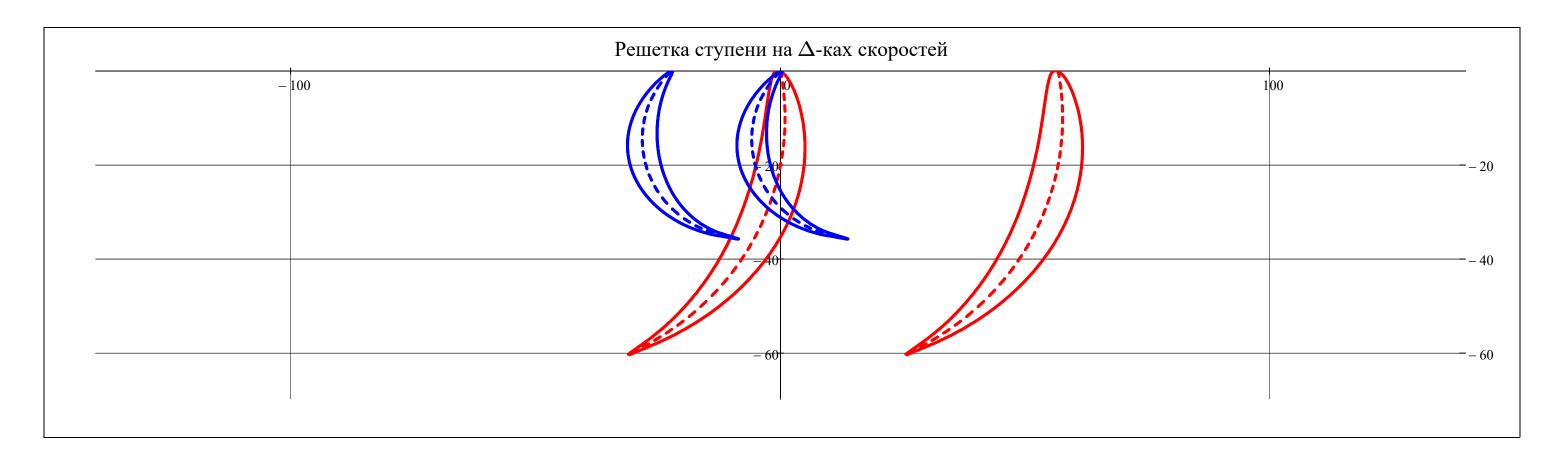
■ Построение профилей Л РК и НА

Рассматриваемая ступень: 
$$j = Z$$
 =  $j = Z$  =  $j = Z$  |  $j = Z$  | "Такой ступени не существует!" if  $(j < 1) \lor (j > Z)$  |  $j$  otherwise

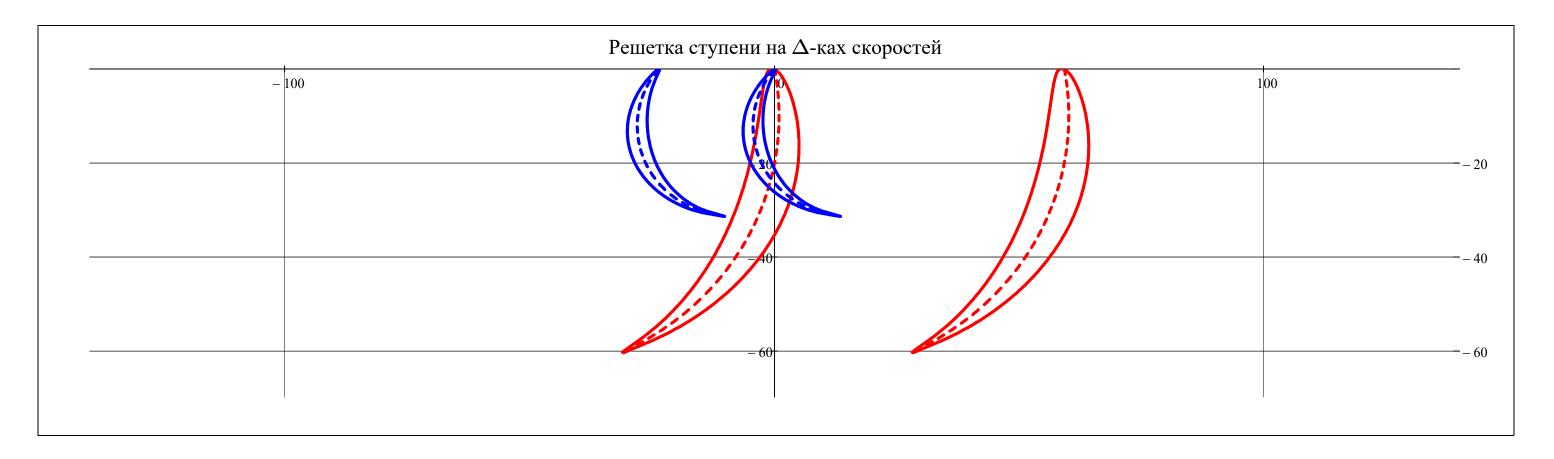
$$b_{\text{Line}} = \frac{\text{ceil}\left(\text{max}\left(\text{chord}_{\text{rotor}_{j,N_r}}, \text{chord}_{\text{stator}_{j,N_r}}\right) \cdot 10^2\right)}{10^2} = 70.0 \cdot 10^{-3}$$

▼ Построение плоских решеток профилей Л на треугольниках скоростей

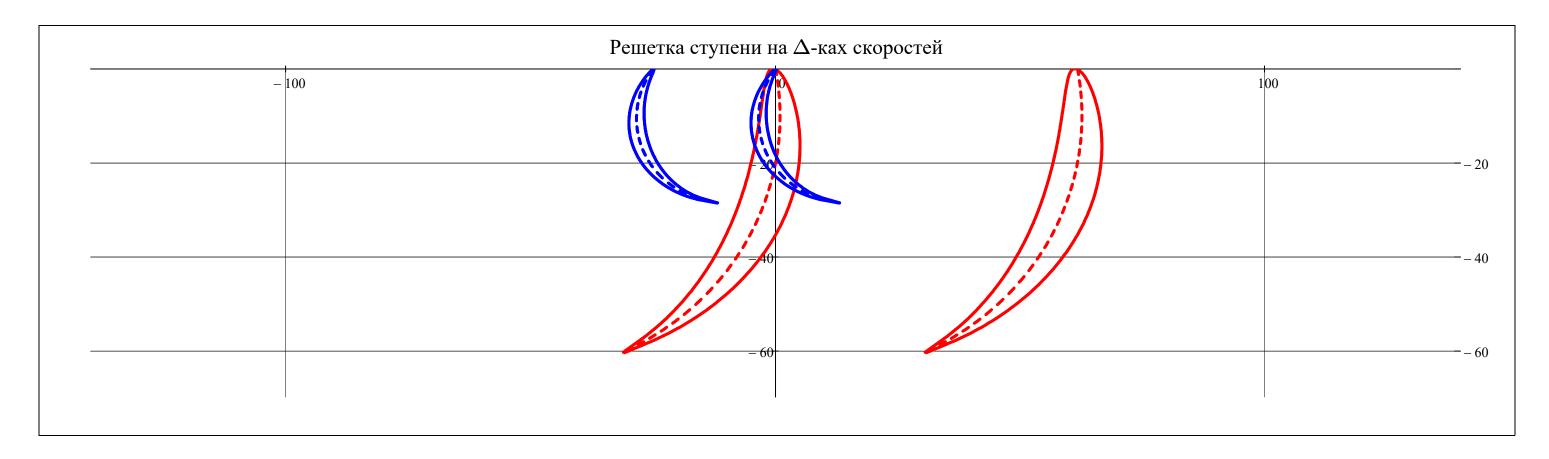




 $r = av(N_r)$ 







▲ Построение плоских решеток профилей Л на треугольниках скоростей

$$j =$$
  $j = Z$   $= 1$   $j =$  "Такой ступени не существует!" if  $(j < 1) \lor (j > Z)$   $j$  otherwise

## ▼ Поперечная часть ступени

$$\mathbf{r} = \min(\mathbf{D}), \min(\mathbf{D}) + \frac{\max(\mathbf{D}) - \min(\mathbf{D})}{N_{dis}} ... \max(\mathbf{D})$$

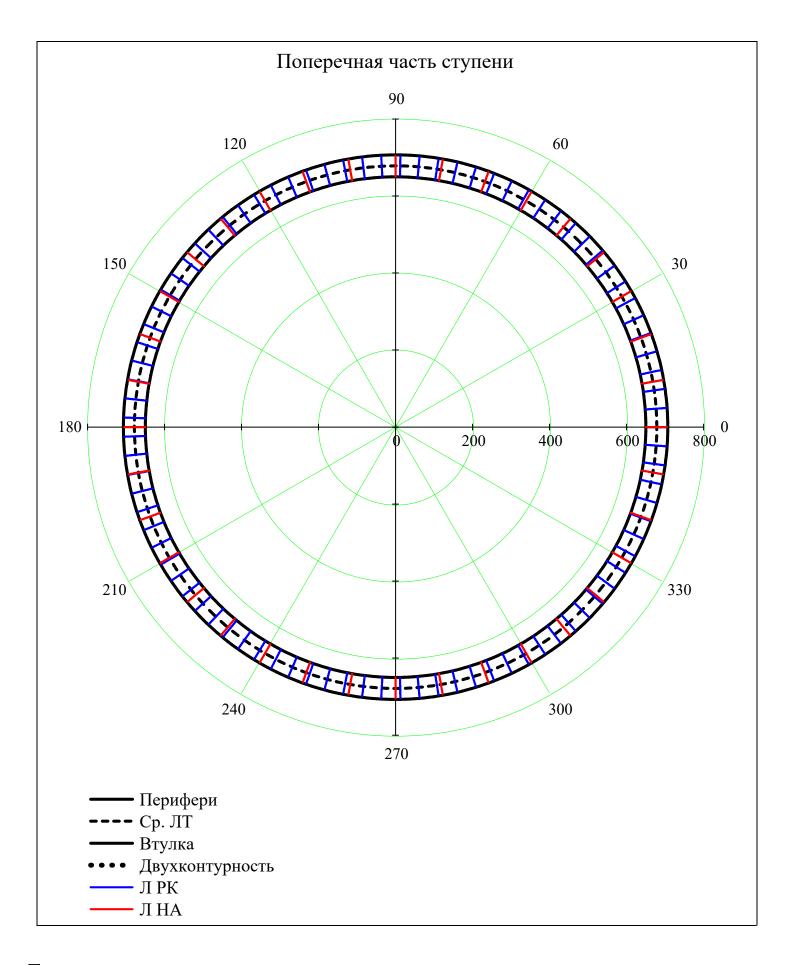
$$\mathbf{i}_{rotor} = 1 ... Z_{rotor_{j}}$$

$$\mathbf{i}_{stator} = 1 ... Z_{stator_{j}}$$

$$\varphi = 0, \frac{2 \cdot \pi}{360} ... 2 \cdot \pi$$

$$\Pi_{PK}(r,j) = \begin{bmatrix} \frac{2 \cdot \pi}{Z_{rotor_{j}}} & \text{if } D_{st(j,1),1} < r < D_{st(j,1),N_{r}} \\ NaN & \text{otherwise} \end{bmatrix}$$

$$\Pi_{\text{HA}}(r,j) = \begin{cases}
\frac{2 \cdot \pi}{Z_{\text{stator}_{j}}} & \text{if } D_{\text{st}(j,2),1} < r < D_{\text{st}(j,2),N_{r}} \\
NaN & \text{otherwise}
\end{cases}$$



```
\nu 0_{\rm M3\Gamma.stator}
                                                                                                                                                                                                                                                                                                                                                                           \nu 0_{\rm M3\Gamma,rotor}
                                                             \nu 0_{
m yr.n.stator}
                                                                                                                                                                                                                                                                                                                                                                           ν0<sub>угл.rotor</sub>
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          for i \in 1...Z
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                for r \in av(N_r)

u^0угл.stator_bondage 
u^0угл.rotor_bondage
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            for mode \in 1...6
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     \nu 0_{\text{M3}\Gamma.\text{stator}_{\hat{1},\,\text{mode}}} = \nu 0_{\text{M3}\Gamma\text{M}\tilde{0}} \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{1},\,1)}\,, h_{\text{st}(\hat{1},\,2)} \Big)\,, \\ E\_\text{blade}\,, \rho\_\text{blade}_{\hat{1}}\,, \text{area}_{\text{stator}_{\hat{1},\,r}}\,, \\ Ju_{\text{stator}_{\hat{1},\,r}} \Big) \Big( h_{\text{st}(\hat{1},\,1)}\,, h_{\text{st}(\hat{1},\,2)} \Big) \Big) \Big( h_{\text{st}(\hat{1},\,2)} \Big) \Big( h_{\text{st}(\hat{1},\,2)} \Big) \Big) \Big( h_{\text{st}(\hat{1},\,2)} \Big) \Big( h_{\text{st}(\hat{1},\,
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  \nu 0_{\text{M3}\Gamma.\text{rotor}_{\hat{i}\,,\,\text{mode}}} = \nu 0_{\text{M3}\Gamma\text{M}\delta} \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big)\,, \\ \text{E\_blade}\,, \rho\_\text{blade}_{\hat{i}}\,, \text{area}_{\text{rotor}_{\hat{i}\,,\,r}}\,, \\ \text{Ju}_{\text{rotor}_{\hat{i}\,,\,r}} \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big)\,, \\ \text{E\_blade}\,, \rho\_\text{blade}_{\hat{i}}\,, \text{area}_{\text{rotor}_{\hat{i}\,,\,r}}\,, \\ \text{Ju}_{\text{rotor}_{\hat{i}\,,\,r}} \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big)\,, \\ \text{E\_blade}\,, \rho\_\text{blade}_{\hat{i}}\,, \text{area}_{\text{rotor}_{\hat{i}\,,\,r}}\,, \\ \text{Ju}_{\text{rotor}_{\hat{i}\,,\,r}} \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big)\,, \\ \text{E\_blade}\,, \rho\_\text{blade}_{\hat{i}}\,, \text{area}_{\text{rotor}_{\hat{i}\,,\,r}}\,, \\ \text{Ju}_{\text{rotor}_{\hat{i}\,,\,r}} \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big)\,, \\ \text{E\_blade}\,, \rho\_\text{blade}_{\hat{i}}\,, \text{area}_{\text{rotor}_{\hat{i}\,,\,r}}\,, \\ \text{Ju}_{\text{rotor}_{\hat{i}\,,\,r}} \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big)\,, \\ \text{E\_blade}\,, \rho\_\text{blade}_{\hat{i}}\,, \text{area}_{\text{rotor}_{\hat{i}\,,\,r}}\,, \\ \text{Ju}_{\text{rotor}_{\hat{i}\,,\,r}} \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,2)}\,, h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big( \text{mode}\,, \text{mean} \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) \Big) \Big( h_{\text{st}(\hat{i}\,,\,3)} \Big) 
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 \nu 0_{\text{yrn.stator}_{i,\,\text{mode}}} = \nu 0_{\text{yrn}} \left( \text{mode}\,, 0\,, \text{mean} \left( h_{\text{st}(i,\,1)}\,, h_{\text{st}(i,\,2)} \right), \text{Jung}(2\,, \mu\_\text{steel}\,, E\_\text{blade})\,, \rho\_\text{blade}_i\,, \text{stiffness}_{\text{stator}_{i,\,r}}\,, \text{Jp}_{\text{stator}_{i,\,r}} \right) \right)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     \nu 0_{y_{\Gamma JI}.rotor_{\hat{1}},\,mode} = \nu 0_{y_{\Gamma JI}} \left(mode,0,mean\left(h_{st(\hat{1},2)},h_{st(\hat{1},3)}\right),Jung(2,\mu\_steel,E\_blade),\rho\_blade_{\hat{1}},stiffness_{rotor_{\hat{1},r}},Jp_{rotor_{\hat{1},r}}\right) \right)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     \nu 0_{\text{YFJI.stator\_bondage}_{i, \, mode}} = \nu 0_{\text{YFJI}} \left( \text{mode} , 1, \text{mean} \left( h_{\text{st}(i, 1)}, h_{\text{st}(i, 2)} \right), \text{Jung}(2, \mu\_\text{steel}, E\_\text{blade}), \rho\_\text{blade}_i, \text{stiffness}_{\text{stator}_{i, r}}, \text{Jp}_{\text{stator}_{i, r}}, \text{Jp}_{\text{stator}_{i,
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     \nu 0_{\text{YFJI.rotor\_bondage}_{\hat{1}, \, mode}} = \nu 0_{\text{YFJ}} \left( \text{mode} \,, 1 \,, \text{mean} \left( h_{st(\hat{1}, 2)} \,, h_{st(\hat{1}, 3)} \right) \,, \\ \text{Jung}(2 \,, \mu\_\text{steel} \,, E\_\text{blade}) \,, \rho\_\text{blade}_{\hat{1}} \,, \\ \text{stiffness}_{rotor_{\hat{1}, \, r}} \,, \\ \text{Jp}_{rotor_{\hat{1}, \, r}} \,, \\ \text{Jp}_{rotor
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       \nu 0_{\text{изг.rotor}}
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   \nu 0_{\rm M3\Gamma.stator}
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               \nu 0_{y_{\Gamma JI}.stator}
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         ν0<sub>угл.rotor</sub>
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    \nu_0^0угл.stator bondage \nu_0^0угл.rotor bondage
```

Частота собственных изгибных колебаний (Гц) [9, с.240]:

$$\operatorname{stack} \left(\nu 0_{\operatorname{M3\Gamma.stator}}, \nu 0_{\operatorname{M3\Gamma.rotor}}\right)^{\mathrm{T}} = \begin{bmatrix} 1 & 2 & 3 & 4 & 5 & 6 & 7 & 8 \\ 1 & 12595 & 5145 & & & & & \\ 2 & 78937 & 32248 & & & & & \\ 3 & 221049 & 90305 & & & & & \\ 4 & 433492 & 177095 & & & & & \\ 5 & 716300 & 292630 & & & & & \\ 6 & 1069752 & 437026 & & & & & & \\ \end{bmatrix}$$

Частота собственных угловых колебаний (Гц) [9, с.243] без и с бандажом:

$$\operatorname{stack} \left(\nu 0_{\text{угл.stator}}, \nu 0_{\text{угл.rotor}}\right)^{\text{T}} = \begin{bmatrix} & 1 & 2 \\ 1 & 8364 & 5507 \\ 2 & 25091 & 16521 \\ 3 & 41819 & 27535 \\ 4 & 58546 & 38548 \\ 5 & 75274 & 49562 \\ 6 & 92001 & 60576 \end{bmatrix}$$

$$\mathrm{stack} \Big(\nu 0_{\mathrm{YLT.stator\_bondage}}, \nu 0_{\mathrm{YLT.rotor\_bondage}}\Big)^{\mathrm{T}} = \begin{bmatrix} & 1 & 2 \\ 1 & 16727 & 11014 \\ 2 & 33455 & 22028 \\ 3 & 50182 & 33041 \\ 4 & 66910 & 44055 \\ 5 & 83637 & 55069 \\ 6 & 100365 & 66083 \\ \end{bmatrix}$$

(D.:: Вывод результатов расчета собственнох частот колебаний 71-

Pасчетный узел: type = "turbine"

Объем бандажной полки (м $^3$ ):  $V_{6\pi} =$ 

Радиус положения ЦМ бандажной полки (м):  $R_{\delta\Pi} = 0$ 

▶ Расчет Л на прочность —

$$\text{neutral\_line(type, x, i, r)} = \begin{vmatrix} y0_{rotor_{i, r}} \\ \frac{y0_{rotor_{i, r}}}{\text{chord}_{rotor_{i, r}}} + \tan\left(\left(\alpha_{major_{rotor_{i, r}}} + \phi_{neutral_{rotor}}(i, R_{st(i, 2), r})\right)\right) \cdot \left(x - \frac{x0_{rotor_{i, r}}}{\text{chord}_{rotor_{i, r}}}\right) \text{ if type} = "rotor"$$

$$\frac{y0_{stator_{i, r}}}{\text{chord}_{stator_{i, r}}} + \tan\left(\left(\alpha_{major_{stator_{i, r}}} + \phi_{neutral_{stator}}(i, R_{st(i, 2), r})\right)\right) \cdot \left(x - \frac{x0_{stator_{i, r}}}{\text{chord}_{stator_{i, r}}}\right) \text{ if type} = "stator"$$

$$\frac{y0_{rotor_{i, r}}}{\text{chord}_{stator_{i, r}}} + \frac{-1}{(x_{major_{stator_{i, r}}})} = \frac{y0_{rotor_{i, r}}}{(x_{major_{stator_{i, r}}})} = \frac{y0_{rotor_{i, r}}}{(x_{ma$$

$$\begin{aligned} & \text{epure(type,x,i,r)} = \boxed{\frac{y0_{rotor_{i,r}}}{\text{chord}_{rotor_{i,r}}} + \frac{-1}{\text{tan}\left(\alpha\_\text{major}_{rotor_{i,r}} + \varphi\_\text{neutral}_{rotor}\left(i,R_{st(i,2),r}\right) - \frac{\pi}{4}\right)} \cdot \left(x - \frac{x0_{rotor_{i,r}}}{\text{chord}_{rotor_{i,r}}}\right) \text{ if type = "rotor"} \\ & \frac{y0_{stator_{i,r}}}{\text{chord}_{stator_{i,r}}} + \frac{-1}{\text{tan}\left(\alpha\_\text{major}_{stator_{i,r}} + \varphi\_\text{neutral}_{stator}\left(i,R_{st(i,2),r}\right) - \frac{\pi}{4}\right)} \cdot \left(x - \frac{x0_{stator_{i,r}}}{\text{chord}_{stator_{i,r}}}\right) \text{ if type = "stator"} \end{aligned}$$

■Определение координат точек профиля Л, наиболее удаленных от НП

Наиболее удаленные точки от НЛ (мм):

$$u_{rotor}^{T} = \begin{bmatrix} & 1\\ 1 & -0.218\\ \hline 2 & 0.395\\ \hline 3 & -5.994 \end{bmatrix} \cdot 10^{-3}$$

$$u_{-l_{rotor}}^{T} = \begin{bmatrix} & 1 & \\ 1 & 21.030 \\ \hline 2 & 18.832 \\ \hline 3 & -8.994 \end{bmatrix} \cdot 10^{-3}$$

$$u_{-}u_{stator}^{T} = \begin{array}{|c|c|c|c|c|}\hline & 1 & & \\ \hline 1 & -7.064 & \\ \hline 2 & 8.165 & \\ \hline 3 & 8.161 & \\ \hline \end{array} \cdot 10^{-3}$$

$$u_{l_{stator}}^{T} = \begin{bmatrix} & 1 & \\ 1 & -11.363 \\ 2 & -25.857 \\ \hline 3 & -25.850 \end{bmatrix} \cdot 10^{-3}$$

$$v_{-}u_{rotor}^{T} = \begin{bmatrix} 1 \\ 1 \\ 2 \\ 4.047 \\ \hline 3 \\ 13.959 \end{bmatrix} \cdot 10^{-3}$$

$$v_{-1}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & -11.789 \\ 2 & -10.076 \\ \hline 3 & -17.325 \end{bmatrix} \cdot 10^{-3}$$

$$v_{u_{stator}}^{T} = \begin{bmatrix} & 1\\ 1 & 29.140\\ \hline 2 & 8.190\\ \hline 3 & 8.194 \end{bmatrix} \cdot 10^{-3}$$

$$v_{l_{stator}}^{T} = \begin{bmatrix} 1 \\ 1 \\ -38.695 \\ 2 \\ -14.465 \\ 3 \\ -14.477 \end{bmatrix} \cdot 10^{-3}$$

$$\begin{pmatrix} \sigma_{-Protor} & \sigma_{-n}rotor \\ \sigma_{-Dstator} & \sigma_{-n}rotor \\ \sigma_{-Dstator}$$

$$\begin{pmatrix} \sigma_{-} P_{rotor.} & \sigma_{-} P_{stator.} \\ \sigma_{-} P_{rotor.} & \sigma_{-} P_{stator.} \end{pmatrix} = \begin{bmatrix} \text{for } i \in 1 ... Z \\ \sigma_{-} P_{rotor.} & \sigma_{-} P_{stator.} \\ \sigma_{-} P_{rotor.} & \sigma_{-} P_{stator.} \end{bmatrix} = \begin{bmatrix} \text{for } i \in 1 ... Z \\ \sigma_{-} P_{rotor.} & \sigma_{-} P_{stator.} \\ \sigma_{-} P_{stator.} & \sigma_{-} P_{stator.} \\ \sigma_{-} P_{stator.} & \sigma_{-} P_{stator.} \\ \sigma_{-} P_{stator.} & \sigma_{-} P_{stator.} \\ \sigma_{-} P_{rotor.} & \sigma_{-} P_{rotor.} & \sigma_{-} P_{rotor.} \\ \sigma_{-} P_{rotor.} & \sigma_{-} P_{rotor.} & \sigma_{-} P_{rotor.} &$$

$$\sigma_{p_{rotor}}^{T} = \begin{array}{|c|c|}\hline & 1 \\\hline 1 & -18.45 \\\hline 2 & -8.87 \\\hline 3 & 0.00 \\\hline \end{array} \cdot 10^{6}$$

$$\sigma_{-n_{rotor}}^{T} = \begin{bmatrix} 1 & 1 \\ 1 & 40.41 \\ 2 & 21.85 \\ \hline 3 & 0.00 \end{bmatrix} \cdot 10^{6}$$

		1	
$\sigma_{\text{ntor}}^{\text{T}} \le 70 \cdot 10^6 =$	1	1	
rotor - / o ro	2	1	
	3	1	

$$\sigma_{p_{stator}}^{T} = \begin{bmatrix} 1 \\ 1 \\ 0.00 \\ 2 \\ 3.47 \\ 3 \end{bmatrix} \cdot 10^{6}$$

		1
$\sigma p_{\text{ototor}} \leq 70 \cdot 10^6 =$	1	1
$\sigma_p_{\text{stator}} \leq /0.10 =$	2	1
	3	1

$$\sigma_{\text{nstator}}^{\text{T}} = \begin{bmatrix} 1 & 1 \\ 1 & 0.00 \\ 2 & -7.07 \\ \hline 3 & -28.26 \end{bmatrix} \cdot 10^{6}$$

		1
$\sigma = 0.410^6 = 0.001$	1	1
$\sigma_{\text{nstator}} \leq /0.10 =$	2	1
	3	1

$$\begin{pmatrix} \sigma_{rotor} \\ \sigma_{stator} \end{pmatrix} = \begin{cases} \text{for } i \in 1...Z \\ \text{for } r \in 1...N_r \end{cases}$$
 
$$\begin{pmatrix} \sigma_{rotor_{i,r}} = \sqrt{\left(\sigma_{rotor}(i,R_{st(i,2),r}) + \max\left(\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}}\right)\right)^2 + \tau_{rotor}(i,R_{st(i,2),r})^2}$$
 
$$\begin{pmatrix} \sigma_{stator_{i,r}} = \sqrt{\left(0 + \max\left(\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}}\right)\right)^2 + \tau_{stator}(i,R_{st(i,2),r})^2} \\ \begin{pmatrix} \sigma_{rotor} \\ \sigma_{stator} \end{pmatrix}$$
 
$$\begin{pmatrix} \sigma_{rotor_{i,r}} = \sqrt{\left(0 + \max\left(\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}}\right)\right)^2 + \tau_{stator_{i,r}}(i,R_{st(i,2),r})^2} \\ \begin{pmatrix} \sigma_{rotor_{i,r}} = \sqrt{\left(0 + \max\left(\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}},\sigma_{rotor_{i,r}}\right)\right)^2 + \tau_{stator_{i,r}}(i,R_{st(i,2),r})^2} \\ \end{pmatrix}$$

$$\begin{pmatrix} \sigma_{rotor.} \\ \sigma_{stator.} \end{pmatrix} = \begin{cases} \text{for } i \in 1..Z \\ \\ \sigma_{rotor.}(i,z) = \text{interp} \Big( \text{lspline} \Big( \text{submatrix} \Big( R, \text{st}(i,2), \text{st}(i,2), 1, N_r \Big)^T, \text{submatrix} \Big( \sigma_{rotor}, i, i, 1, N_r \Big)^T \Big), \text{submatrix} \Big( R, \text{st}(i,2), \text{st}(i,2), 1, N_r \Big)^T, \text{submatrix} \Big( \sigma_{rotor}, i, i, 1, N_r \Big)^T, \text{submatrix} \Big( R, \text{st}(i,2), \text{st}(i,2), 1, N_r \Big)^T, \text{submatrix} \Big( R, \text{st}(i,2), \text{st}(i,2), 1, N_r \Big)^T, \text{submatrix} \Big( \sigma_{stator.}, i, i, 1, N_r \Big)^T, \text{submatrix} \Big( \sigma_{rotor.}, i, i, 1, N_r \Big)^T, \text{submatrix} \Big( \sigma_{stator.}, i, i, 1, N_r$$

$$\sigma_{\text{rotor}}^{\text{T}} = \begin{vmatrix} & 1 \\ 1 & 161.59 \\ 2 & 101.72 \\ \hline 3 & 0.00 \end{vmatrix} \cdot 10$$

$$\sigma_{\text{stator}}^{\text{T}} = \begin{vmatrix} 1 & 1 \\ 1 & 0.00 \\ 2 & 5.61 \\ 3 & 16.44 \end{vmatrix} \cdot 10^{6}$$

$$\begin{vmatrix} safety_{rotor} \\ safety_{stator} \end{vmatrix} = \begin{vmatrix} for \ i \in 1...Z \\ for \ r \in 1...N_r \end{vmatrix}$$
 
$$\begin{vmatrix} safety_{rotor} \\ i,r \end{vmatrix} = \begin{vmatrix} \frac{\sigma\_blade\_long_i}{\sigma_{rotor}} & if \ \sigma_{rotor} \\ \infty & otherwise \end{vmatrix}$$
 
$$safety_{stator} \\ i,r \end{vmatrix} = \begin{vmatrix} \frac{\sigma\_blade\_long_i}{\sigma_{stator}} & if \ \sigma_{stator} \\ \infty & otherwise \end{vmatrix}$$
 
$$\begin{vmatrix} safety_{rotor} \\ safety_{stator} \end{vmatrix}$$
 
$$contact$$
 
$$contact$$

$safety_{stator}^{T} = $		1	
	1	000000000000000000000000000000000000000	
	2	36.53	
		3	12.47

		1	
$safety_{rotor}^{T} \ge safety =$	1	0	
rotor = surety	2	1	
	3	1	

		1	
$safety_{stator}^{T} \ge safety =$	1	1	
salety stator – salety	2	1	
	3	1	

Рассматриваемая ступень: 
$$j=1$$
 if type = "compressor" = 1 Z if type = "turbine"  $j=1$  "Такой ступени не существует!" if  $(j<1)\vee(j>Z)$   $j$  otherwise

$$b_{\text{Line}} = \frac{\text{ceil}\left(\text{max}\left(\text{chord}_{rotor_{j,N_r}}, \text{chord}_{stator_{j,N_r}}\right) \cdot 10^2\right)}{10^2} = 70.0 \cdot 10^{-3}$$

Расстояния от оси ЛМ до рассматриваемой ступени (м):

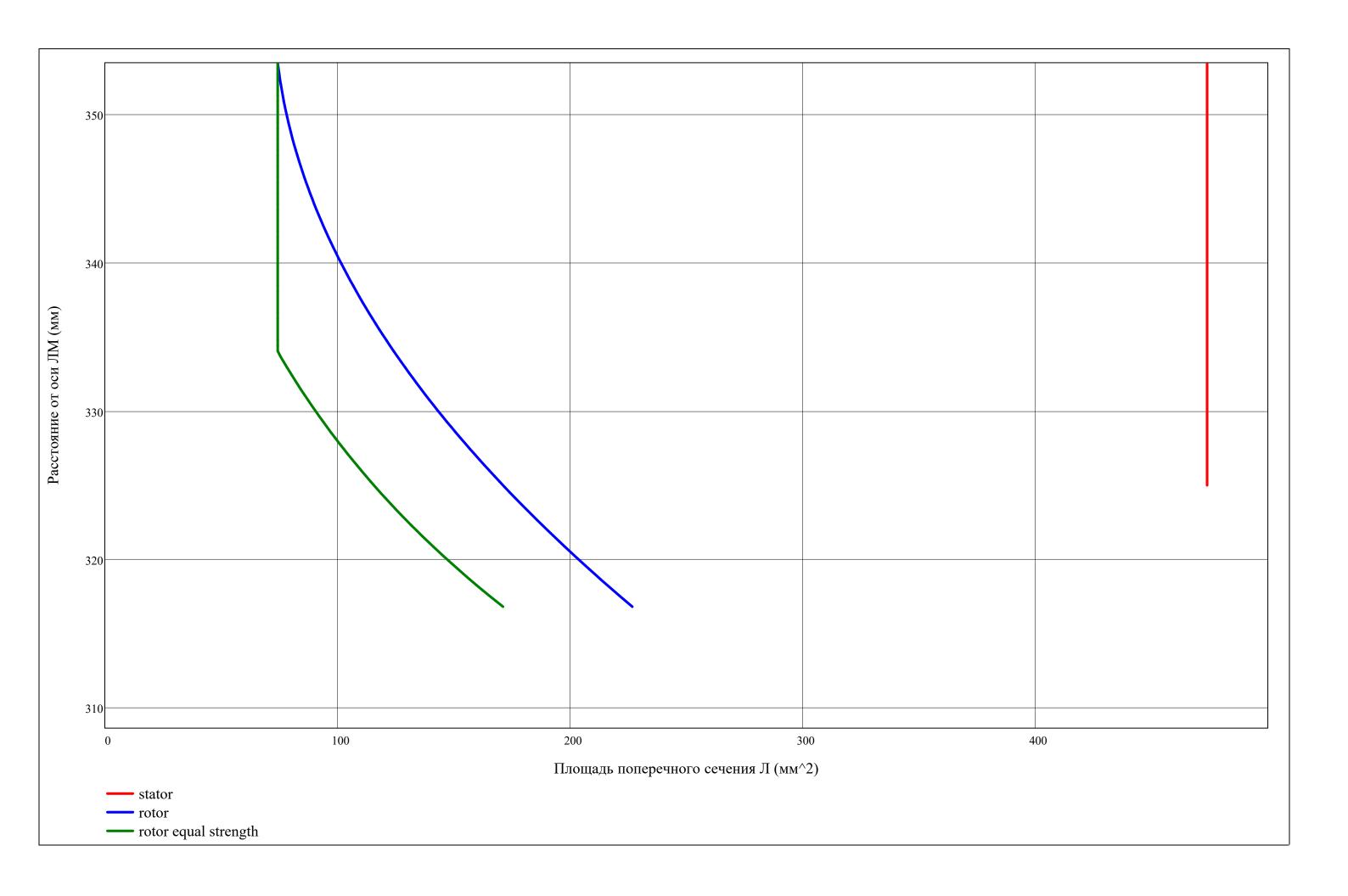
$$Rj = submatrix (R, 2 \cdot j - 1, 2 \cdot j + 1, 1, N_r) = \begin{vmatrix} 1 & 2 & 3 \\ 1 & 325.0 & 339.2 & 353.5 \\ 2 & 325.0 & 339.2 & 353.5 \\ 3 & 308.6 & 331.0 & 353.5 \end{vmatrix} \cdot 10^{-3}$$

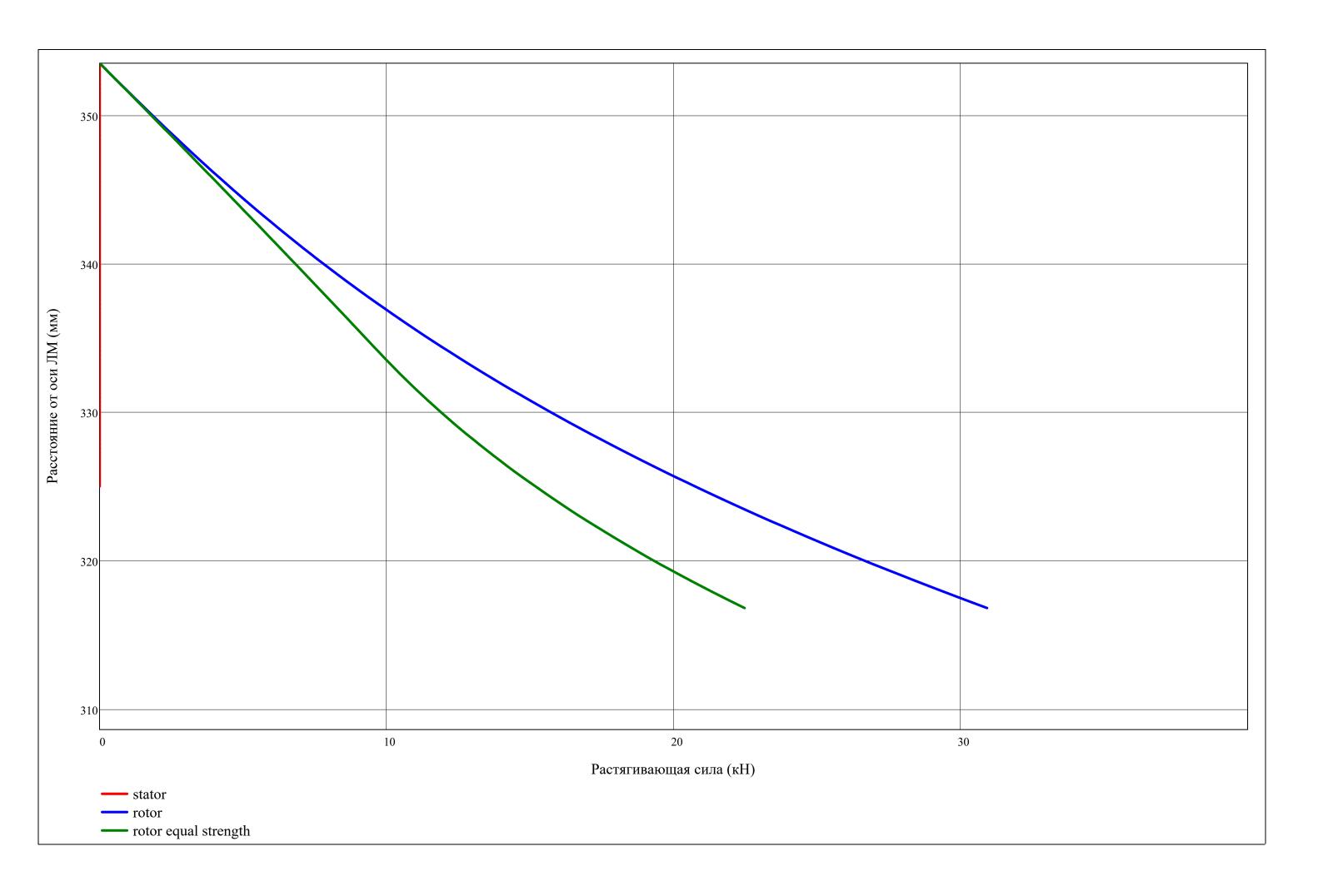
Дискретизация по высоте Л:

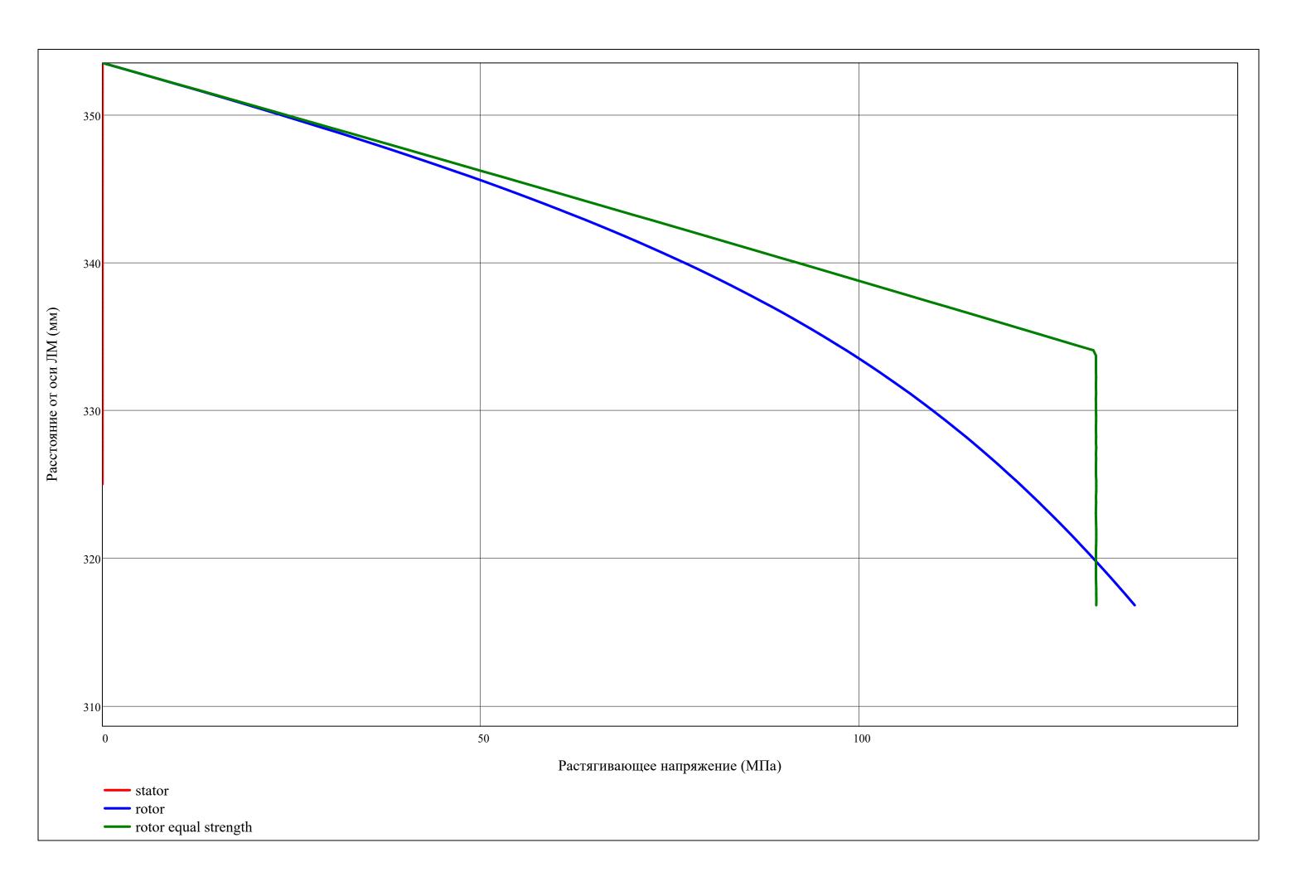
$$z = min(Rj), min(Rj) + \frac{max(Rj) - min(Rj)}{100} ... max(Rj)$$

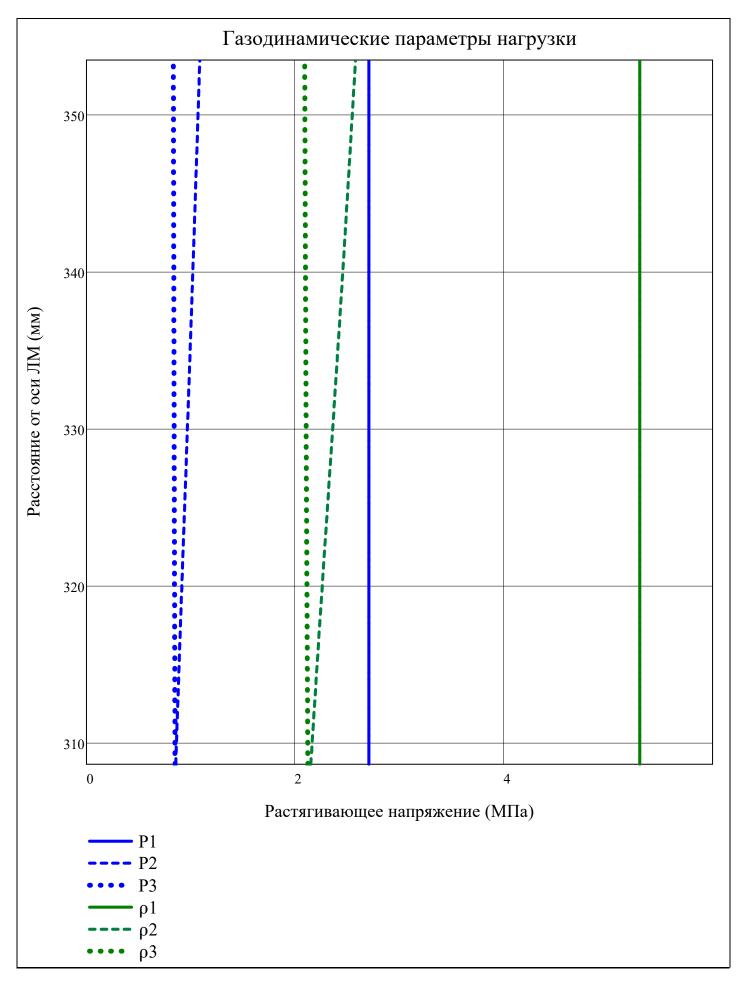
$$z_{rotor} = \begin{vmatrix} mean \left(Rj_{1,1}, Rj_{2,1}\right), mean \left(Rj_{1,1}, Rj_{2,1}\right) + \frac{mean \left(Rj_{1,N_r}, Rj_{2,N_r}\right) - mean \left(Rj_{1,1}, Rj_{2,1}\right)}{100} ... mean \left(Rj_{1,N_r}, Rj_{2,N_r}\right) \text{ if type = "compressor"} \\ mean \left(Rj_{2,1}, Rj_{3,1}\right), mean \left(Rj_{2,1}, Rj_{3,1}\right) + \frac{mean \left(Rj_{2,N_r}, Rj_{3,N_r}\right) - mean \left(Rj_{2,1}, Rj_{3,1}\right)}{100} ... mean \left(Rj_{2,N_r}, Rj_{3,N_r}\right) \text{ if type = "turbine"} \\ 100 \end{aligned}$$

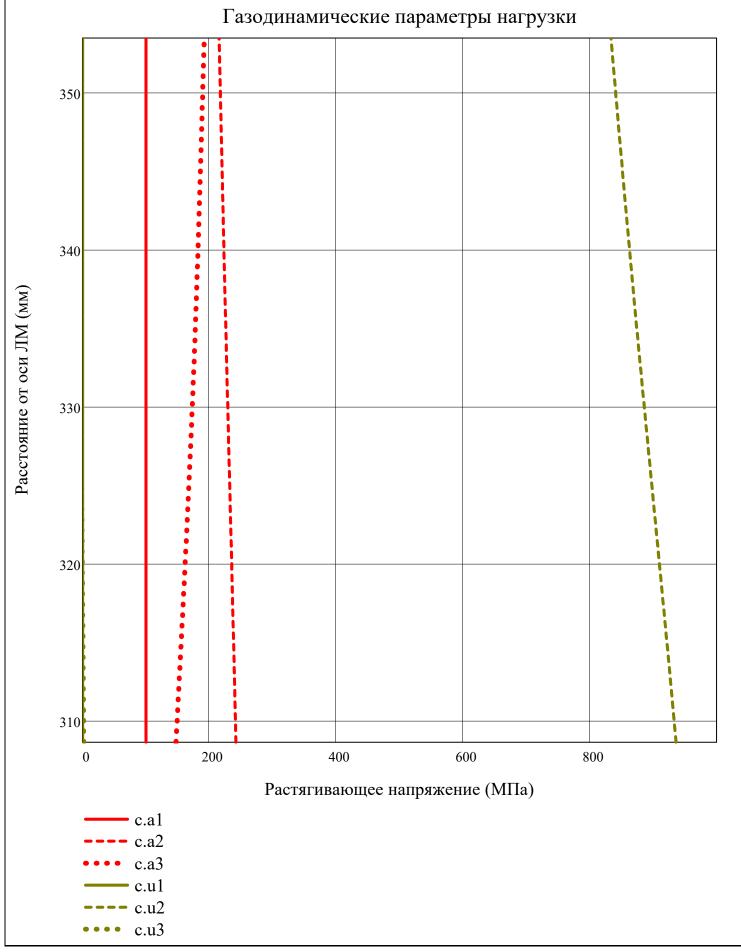
$$z_{stator} = \begin{bmatrix} mean \left(Rj_{2,1}, Rj_{3,1}\right), mean \left(Rj_{2,1}, Rj_{3,1}\right) + \frac{mean \left(Rj_{2,N_r}, Rj_{3,N_r}\right) - mean \left(Rj_{2,1}, Rj_{3,1}\right)}{100} ... mean \left(Rj_{2,N_r}, Rj_{3,N_r}\right) & \text{if type} = "compressor" \\ mean \left(Rj_{1,1}, Rj_{2,1}\right), mean \left(Rj_{1,1}, Rj_{2,1}\right) + \frac{mean \left(Rj_{1,N_r}, Rj_{2,N_r}\right) - mean \left(Rj_{1,1}, Rj_{2,1}\right)}{100} ... mean \left(Rj_{1,N_r}, Rj_{2,N_r}\right) & \text{if type} = "turbine" \\ \end{bmatrix}$$

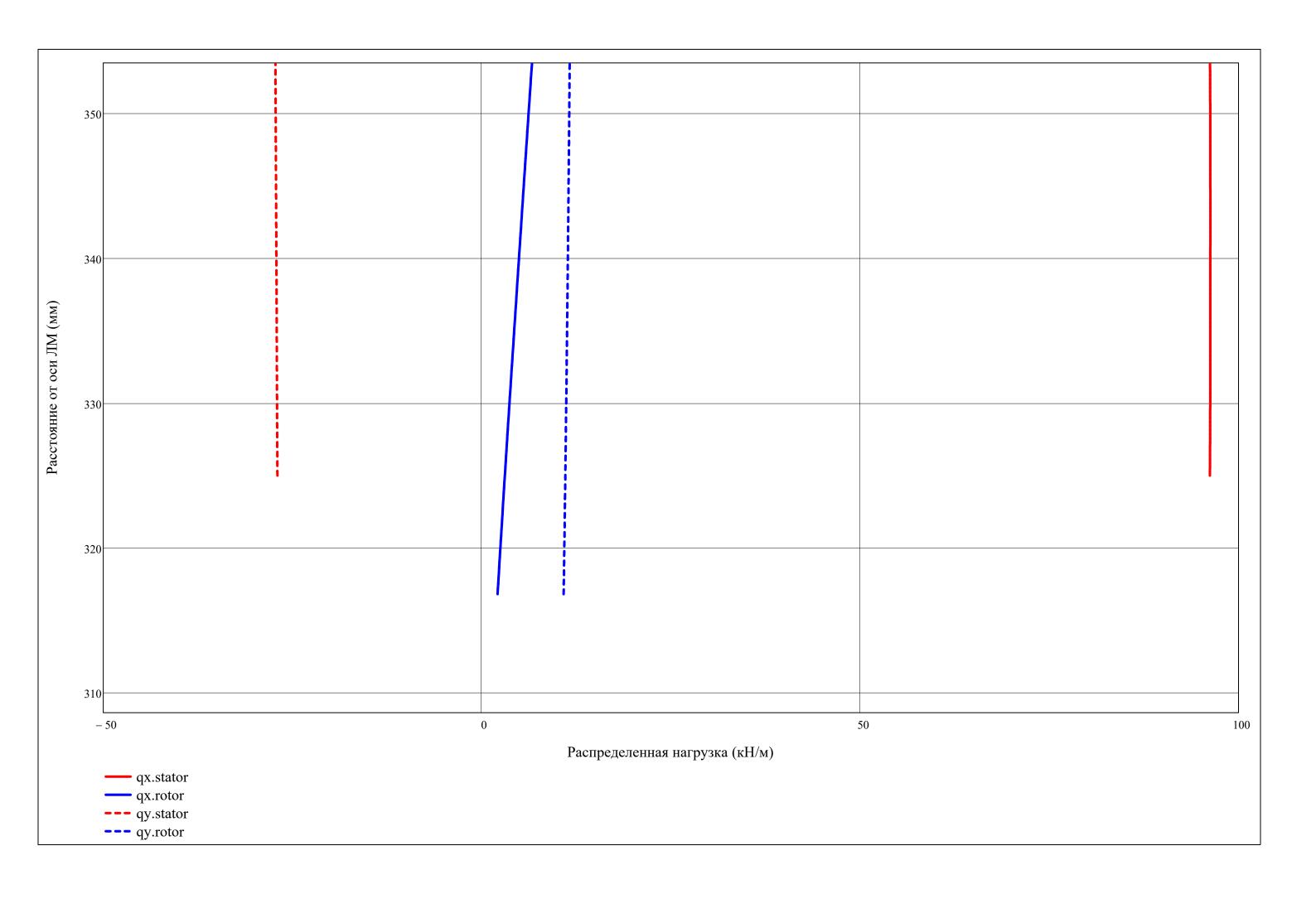


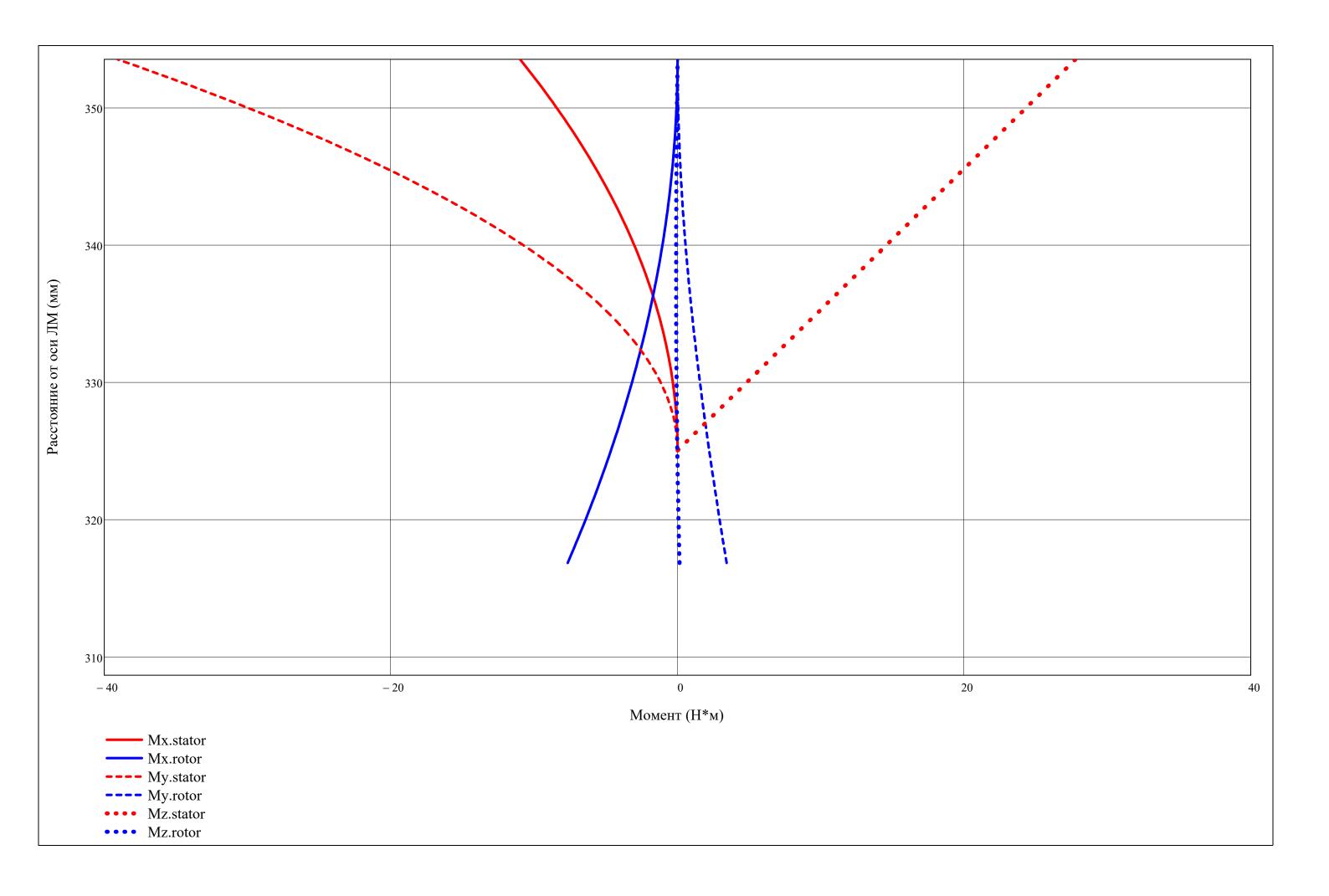


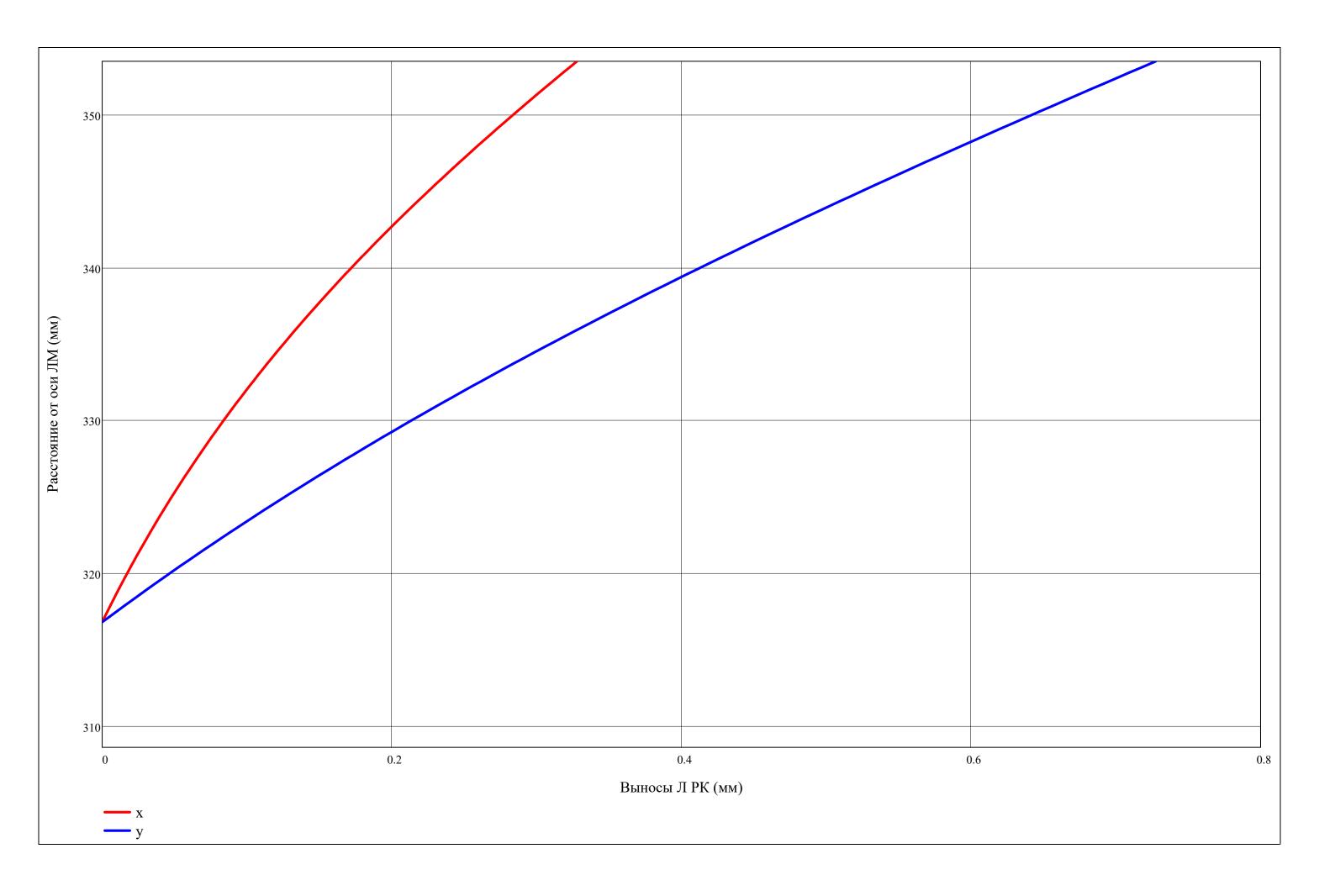


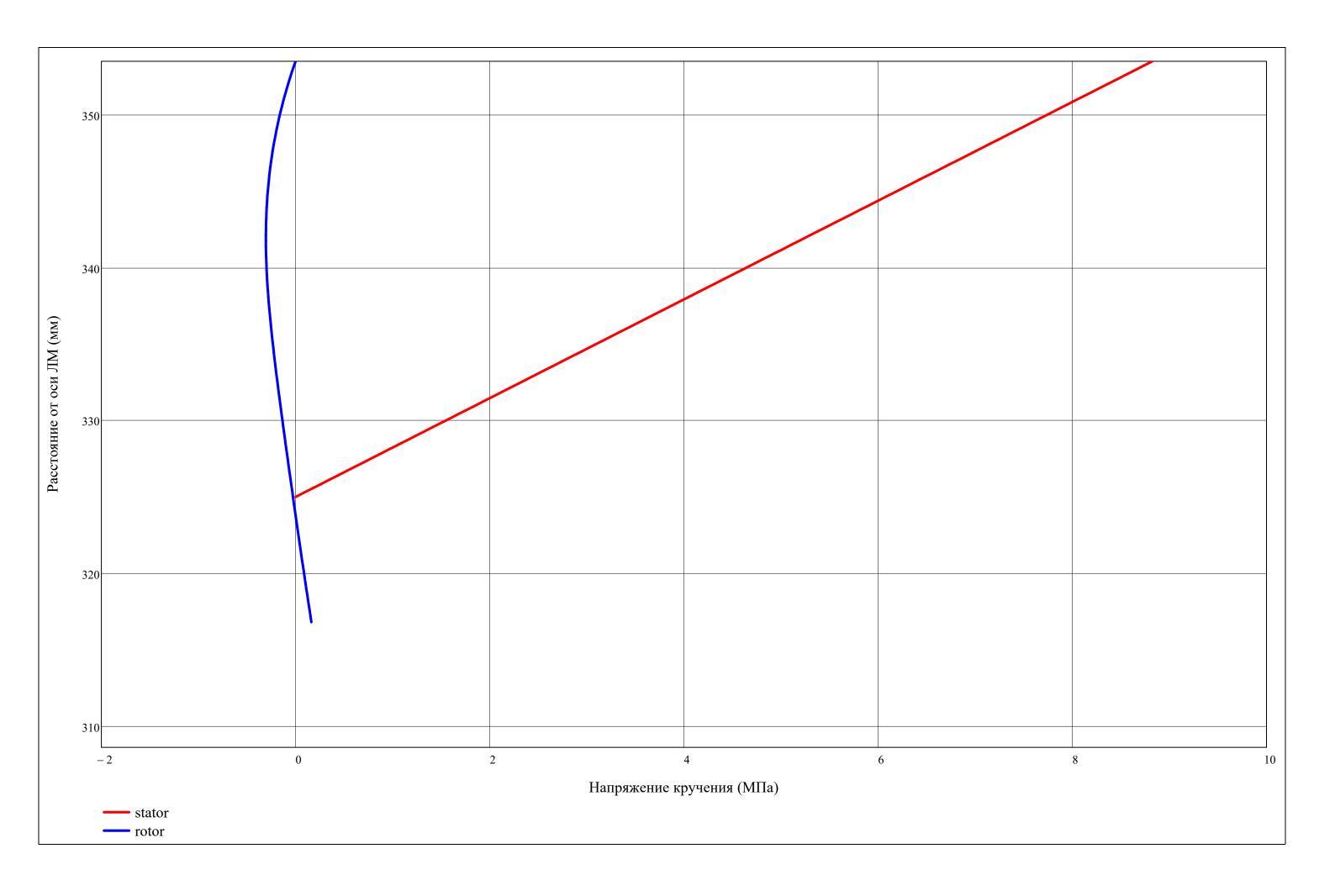


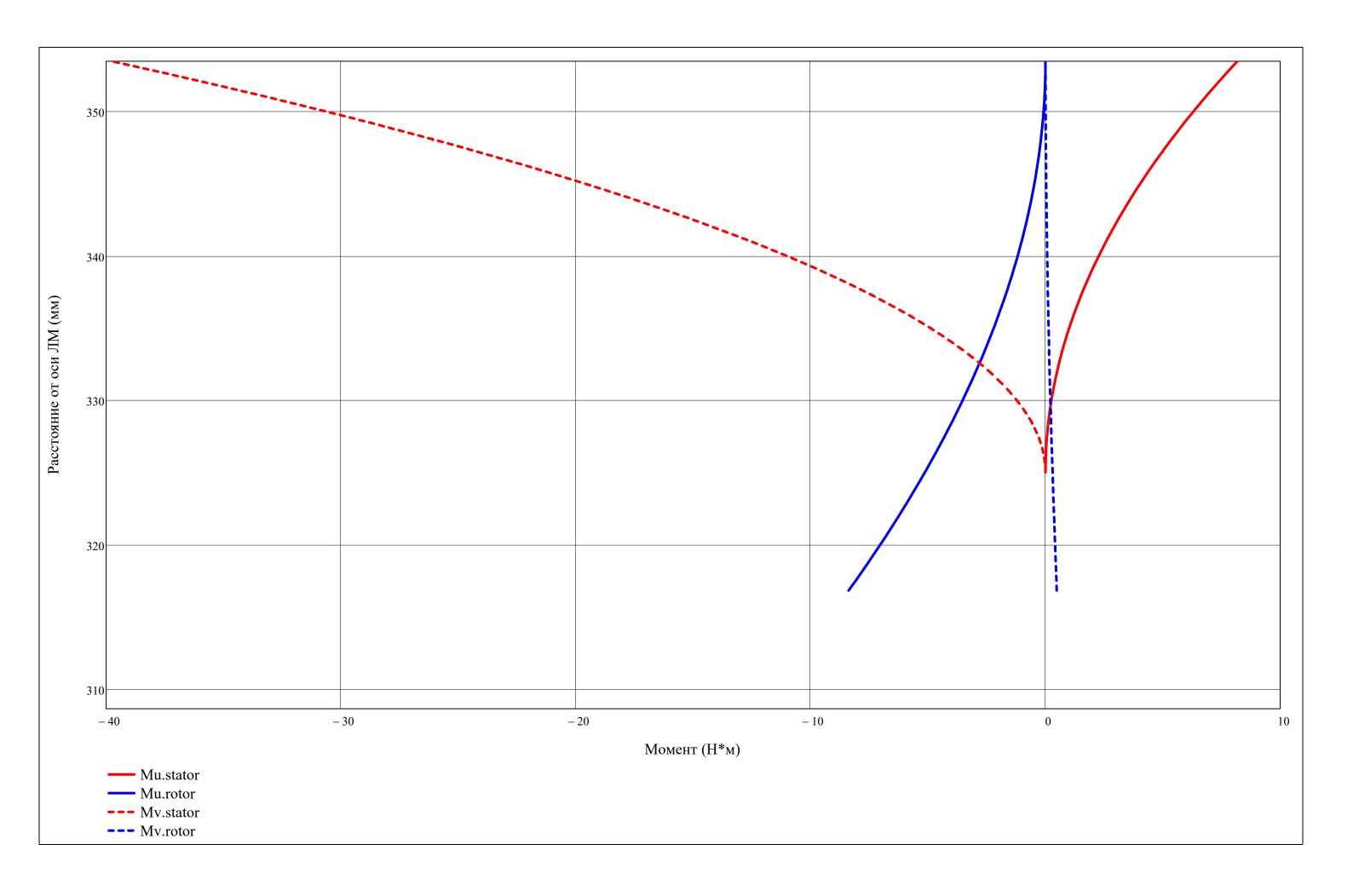


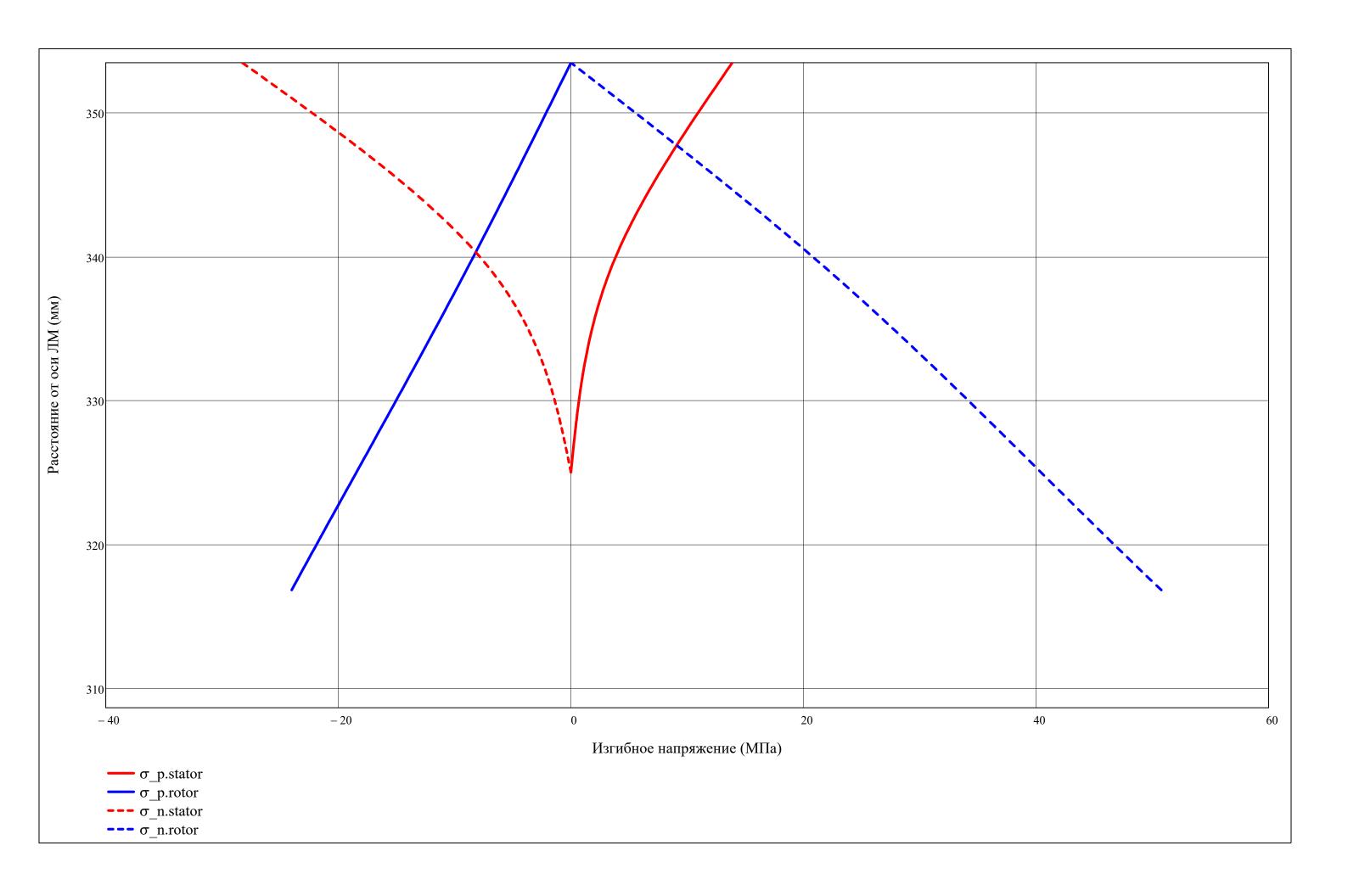


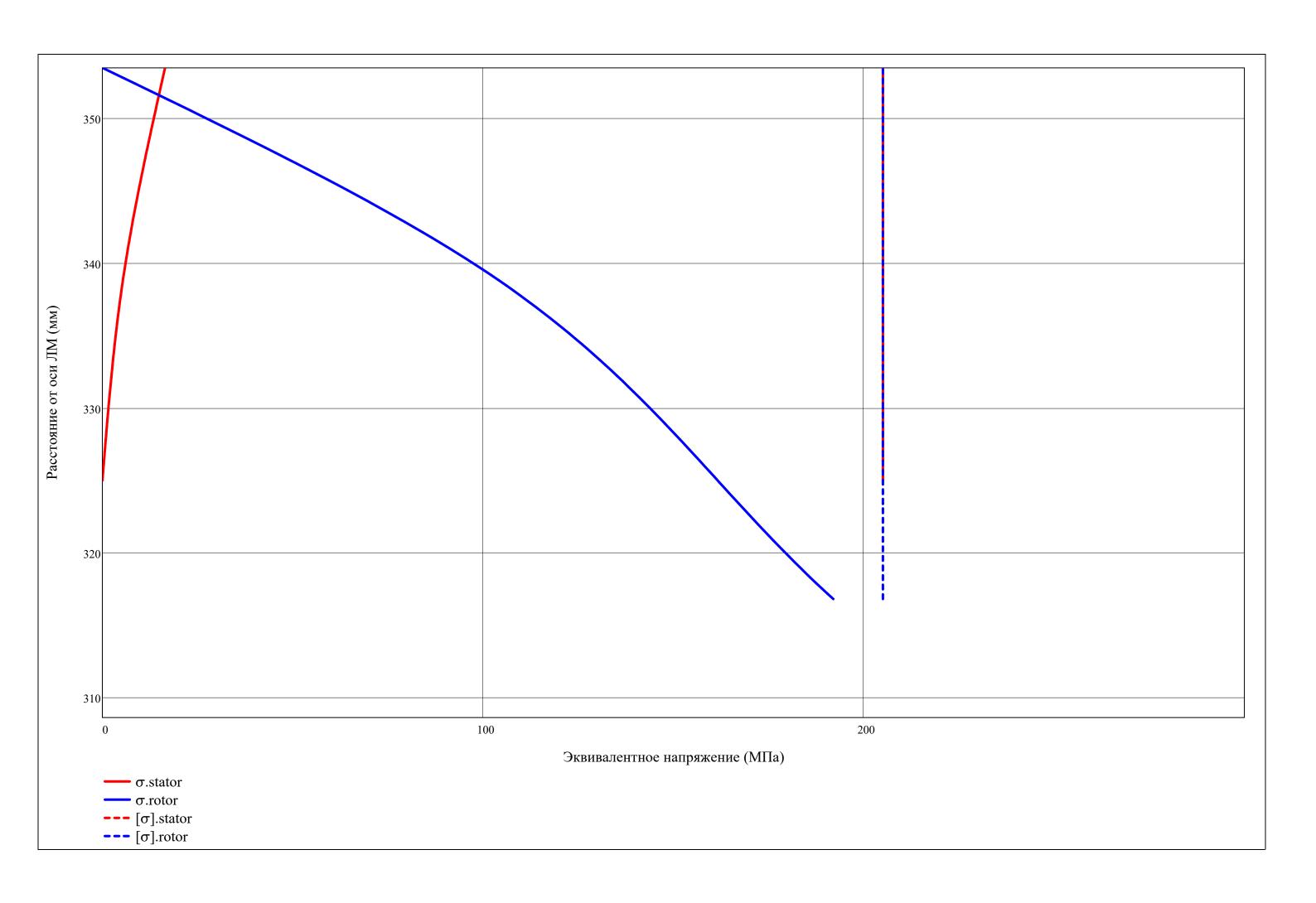












$$\begin{pmatrix} blade \\ r \end{pmatrix} = \begin{pmatrix} "stator" \\ 2 \end{pmatrix}$$

$$\begin{pmatrix} u_{-}u_{rotor_{j},r} & v_{-}u_{rotor_{j},r} \\ u_{-}l_{rotor_{j},r} & v_{-}l_{rotor_{j},r} \\ u_{-}u_{stator_{j},r} & v_{-}u_{stator_{j},r} \\ u_{-}l_{stator_{j},r} & v_{-}l_{stator_{j},r} \end{pmatrix} = \begin{pmatrix} 1 & 2 \\ 1 & 0.40 & 4.05 \\ 2 & 18.83 & -10.08 \\ 3 & 8.16 & 8.19 \\ 4 & -25.86 & -14.47 \end{pmatrix} \cdot 10^{-3}$$

Изгибные напряжения (Па):

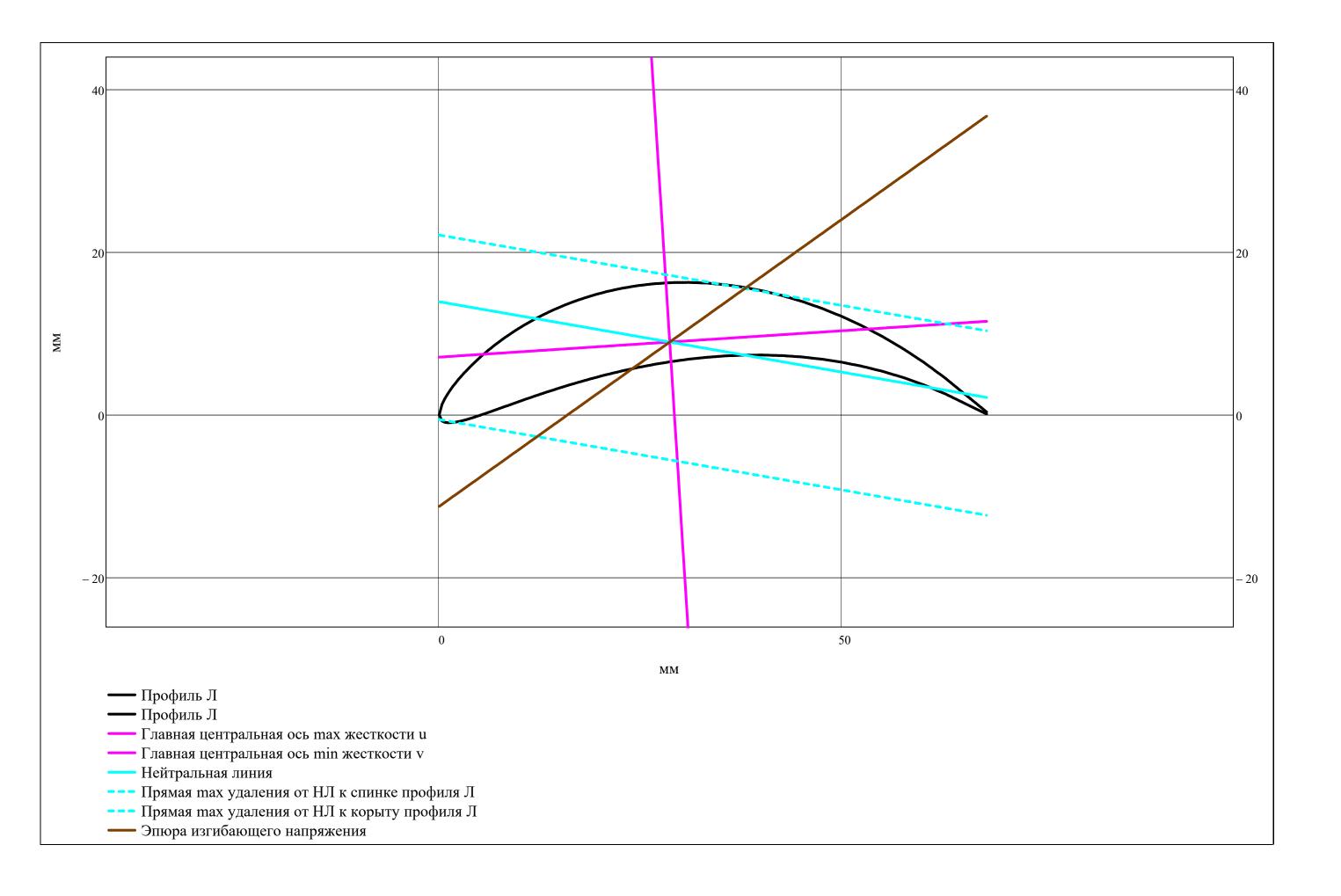
$$\begin{pmatrix} \sigma_{-}p_{rotor_{j}, r} & \sigma_{-}p_{stator_{j}, r} \\ \sigma_{-}n_{rotor_{j}, r} & \sigma_{-}n_{stator_{j}, r} \end{pmatrix} = \begin{pmatrix} 1 & 2 \\ 1 & -8.9 & 3.5 \\ 2 & 21.9 & -7.1 \end{pmatrix} \cdot 10^{6}$$

Эквивалентные напряжения (Па):

$$\begin{pmatrix} \sigma_{\text{stator}_{j,r}} \\ \sigma_{\text{rotor}_{j,r}} \end{pmatrix} = \begin{bmatrix} 1 \\ 1 \\ 2 \end{bmatrix} \cdot 10^{6}$$

Коэф. запаса: 
$$\begin{pmatrix} safety_{stator_{j,r}} \\ safety_{rotor_{j,r}} \end{pmatrix} = \begin{bmatrix} 1 \\ 1 \\ 36.526 \\ 2 \\ 2.015 \end{bmatrix}$$

$$\begin{pmatrix} v_{-}p \\ v_{-}n \end{pmatrix} = \begin{pmatrix} v_{-}u_{rotor_{j},r} \\ v_{-}l_{rotor_{j},r} \end{pmatrix} \text{ if blade = "rotor"} = \begin{pmatrix} x_{0} \\ 1 \\ 2 \\ -14.465 \end{pmatrix} \cdot 10^{-3} \quad \begin{pmatrix} x_{0} \\ y_{0} \end{pmatrix} = \begin{pmatrix} x_{0} \\ y_{0} \end{pmatrix} = \begin{pmatrix} x_{0} \\ y_{0} \\ y_{0} \\ y_{0} \end{pmatrix} = \begin{pmatrix} x_{0} \\ y_{0} \\ y_{0} \\ y_{0} \end{pmatrix} = \begin{pmatrix} x_{0} \\ y_{0} \\ y_{0} \\ y_{0} \end{pmatrix} = \begin{pmatrix} x_{0} \\ y_{0} \\ y_$$



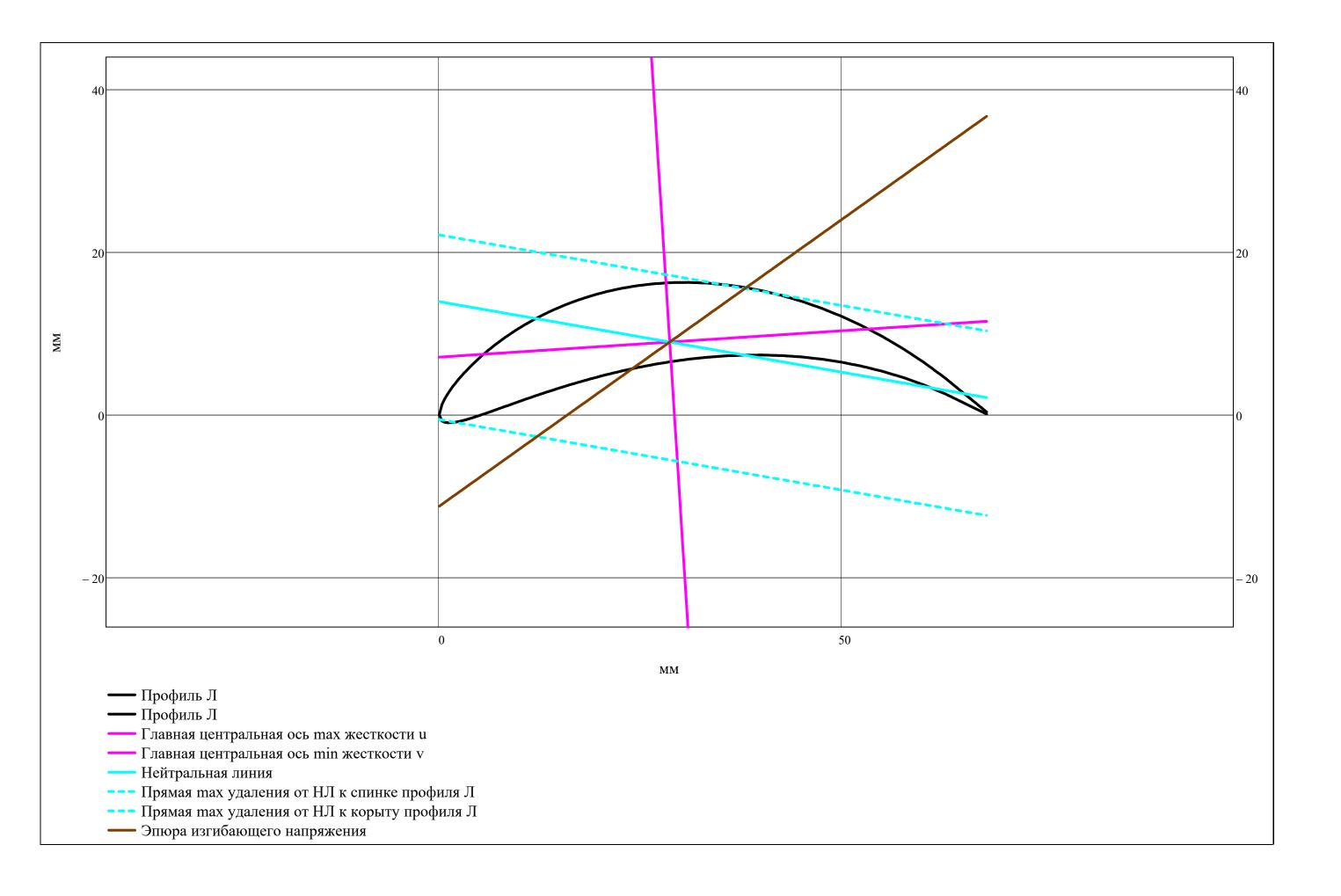
$$\begin{pmatrix} \text{blade} \\ \text{r} \end{pmatrix} = \begin{pmatrix} \text{"stator"} \\ 3 \end{pmatrix}$$

 $\begin{pmatrix} u_{-}u_{rotor_{j,r}} & v_{-}u_{rotor_{j,r}} \\ u_{-}l_{rotor_{j,r}} & v_{-}l_{rotor_{j,r}} \\ u_{-}u_{stator_{j,r}} & v_{-}u_{stator_{j,r}} \end{pmatrix} = \begin{pmatrix} u_{-}u_{rotor_{j,r}} & v_{-}u_{rotor_{j,r}} \\ 1 & 2 & 3 \\ 3 & 4 & 4 \end{pmatrix}$ -5.99 13.96 -8.99 -17.32 8.19 -25.85 -14.48  $u_l = v_j \cdot v_l = v_j \cdot v_j \cdot v_l \cdot v_j \cdot v_l \cdot v_j \cdot v_l \cdot v_j \cdot v_l \cdot v_l$ 

Изгибные напряжения (Па):

 $\begin{pmatrix} \sigma_{-}p_{rotor_{j},r} & \sigma_{-}p_{stator_{j},r} \\ \sigma_{-}n_{rotor_{j},r} & \sigma_{-}n_{stator_{j},r} \end{pmatrix} = \begin{bmatrix} 1 & 2 \\ 1 & 0.0 & 13.9 \\ 2 & 0.0 & -28.3 \end{bmatrix} \cdot 10^{6}$ 

Эквивалентные напряжения (Па):



$$\begin{pmatrix} \text{blade} \\ \text{r} \end{pmatrix} = \begin{pmatrix} \text{"rotor"} \\ 1 \end{pmatrix}$$

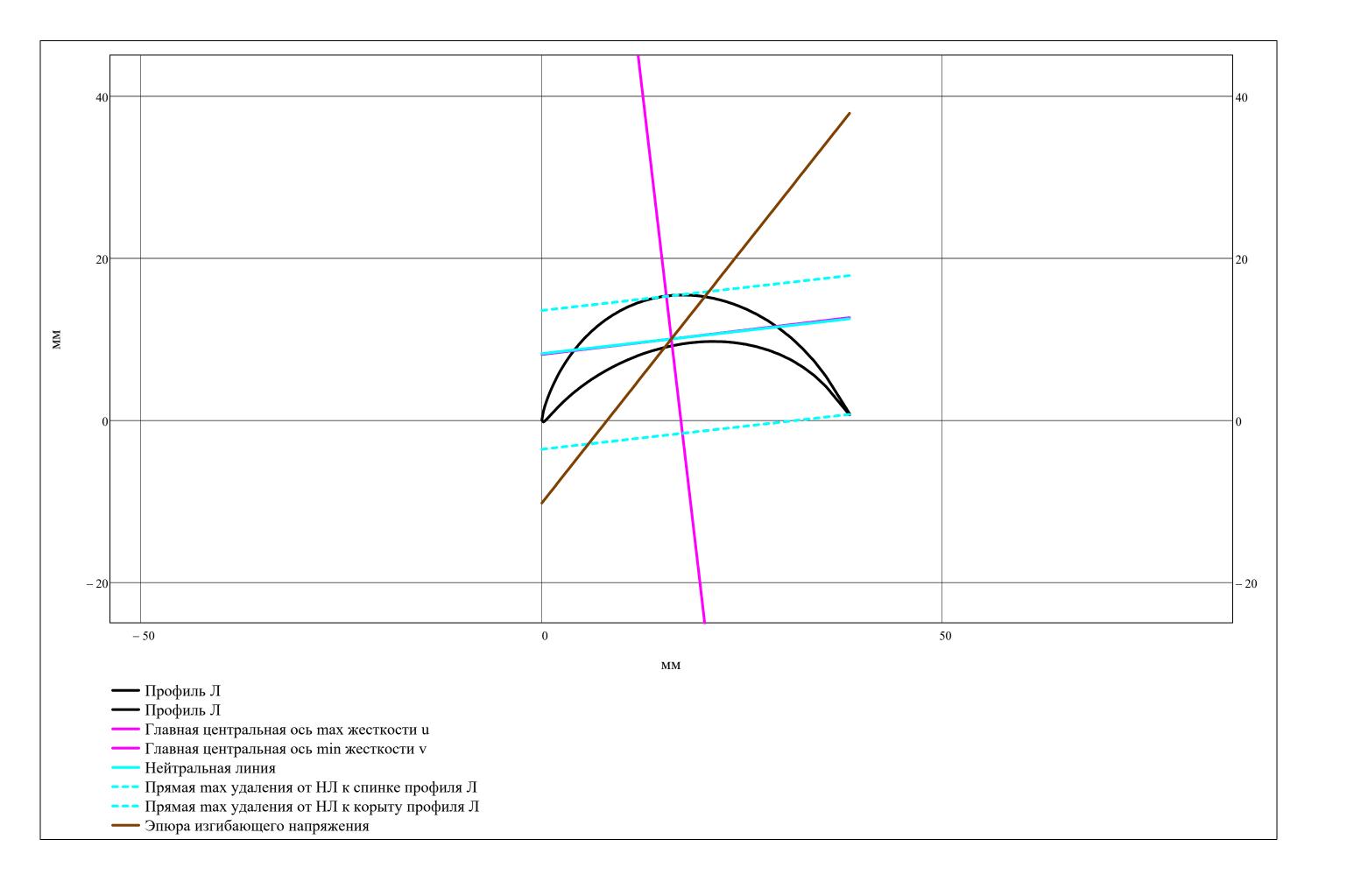
$$\begin{pmatrix} u_{-}u_{rotor_{j},r} & v_{-}u_{rotor_{j},r} \\ u_{-}l_{rotor_{j},r} & v_{-}l_{rotor_{j},r} \\ u_{-}u_{stator_{j},r} & v_{-}u_{stator_{j},r} \\ u_{-}l_{stator_{j},r} & v_{-}l_{stator_{j},r} \end{pmatrix} = \begin{pmatrix} 1 & 2 \\ 1 & -0.22 & 5.32 \\ 2 & 21.03 & -11.79 \\ 3 & -7.06 & 29.14 \\ 4 & -11.36 & -38.70 \end{pmatrix}$$

Изгибные напряжения (Па):

$$\begin{pmatrix} \sigma_{-}p_{rotor_{j}, r} & \sigma_{-}p_{stator_{j}, r} \\ \sigma_{-}n_{rotor_{j}, r} & \sigma_{-}n_{stator_{j}, r} \end{pmatrix} = \begin{bmatrix} 1 & 2 \\ 1 & -18.4 & 0.0 \\ 2 & 40.4 & 0.0 \end{bmatrix} \cdot 10^{6}$$

Эквивалентные напряжения (Па):

$$\begin{pmatrix} \sigma_{\text{stator}_{j,r}} \\ \sigma_{\text{rotor}_{j,r}} \end{pmatrix} = \begin{bmatrix} 1 \\ 1 \\ 0.0 \\ 2 \end{bmatrix} \cdot 10$$



$$\begin{pmatrix} \text{blade} \\ \text{max} \\ \text{c} \end{pmatrix} = \begin{pmatrix} \text{"rotor"} \\ 2 \end{pmatrix}$$

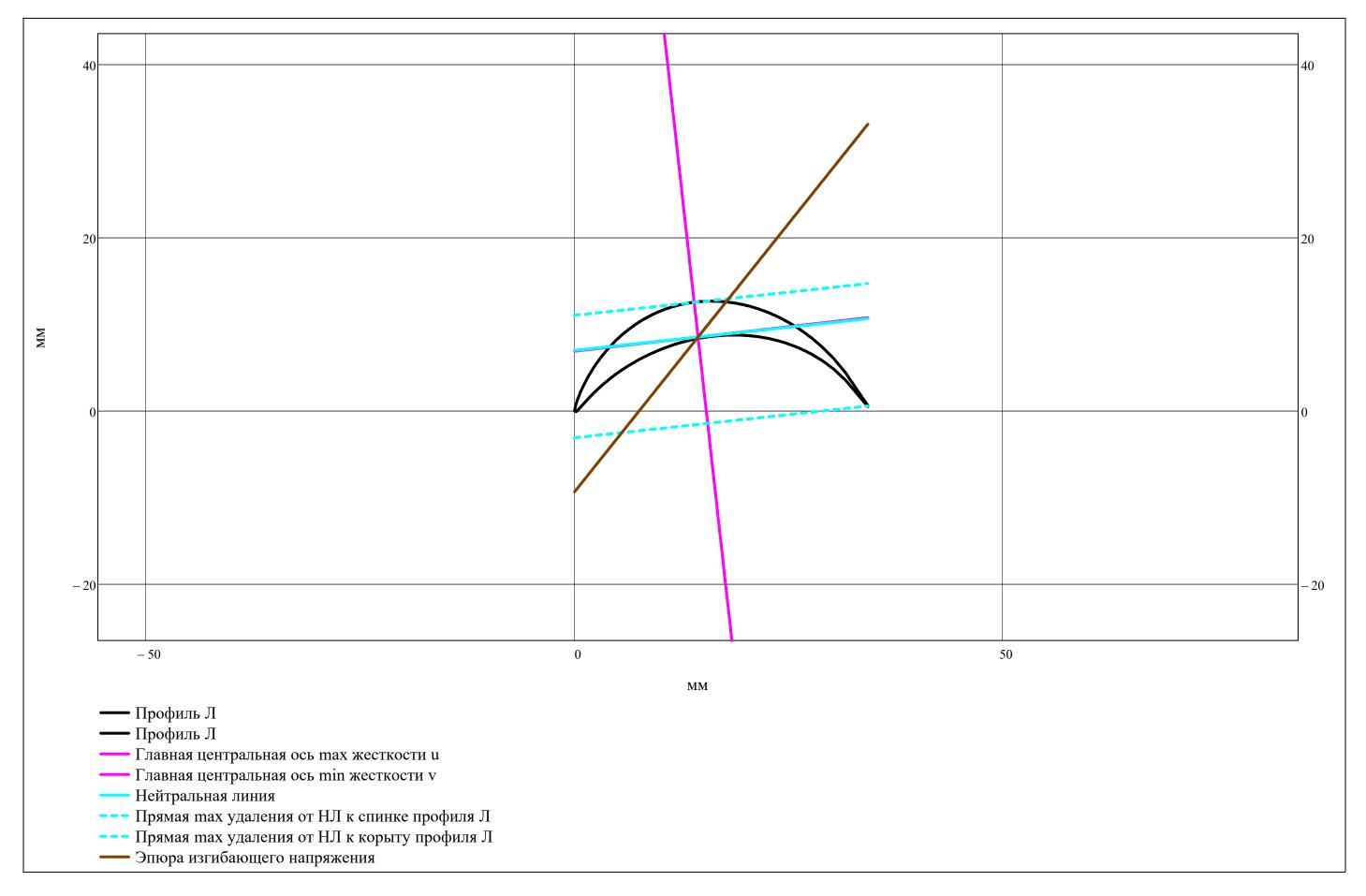
$$\begin{pmatrix} u_{-}u_{rotor_{j},r} & v_{-}u_{rotor_{j},r} \\ u_{-}l_{rotor_{j},r} & v_{-}l_{rotor_{j},r} \\ u_{-}u_{stator_{j},r} & v_{-}u_{stator_{j},r} \\ u_{-}l_{stator_{j},r} & v_{-}l_{stator_{j},r} \end{pmatrix} = \begin{pmatrix} 1 & 2 \\ 1 & 0.40 & 4.05 \\ 2 & 18.83 & -10.08 \\ 3 & 8.16 & 8.19 \\ 4 & -25.86 & -14.47 \end{pmatrix} \cdot 10^{-3}$$

Изгибные напряжения (Па):

$$\begin{pmatrix} \sigma_{p_{rotor_{j},r}} & \sigma_{p_{stator_{j},r}} \\ \sigma_{n_{rotor_{j},r}} & \sigma_{n_{stator_{j},r}} \end{pmatrix} = \begin{bmatrix} 1 & 2 \\ 1 & -8.9 & 3.5 \\ 2 & 21.9 & -7.1 \end{bmatrix} \cdot 10^{6}$$

Эквивалентные напряжения (Па):

$$\begin{pmatrix} \sigma_{\text{stator}_{j,r}} \\ \sigma_{\text{rotor}_{j,r}} \end{pmatrix} = \begin{bmatrix} 1 \\ 1 \\ 2 \end{bmatrix} \cdot 10^{6}$$



Вывод результатов расчета Л на прочность

## ▼ Выбор материала Д

Запас по температуре (K):  $\Delta T_{\text{safety}} = 0$ 

Выбранный материал Д: material\_disk<sub>i</sub> = "ВЖ175" if turbine = "ТВД" "ЭП742" if turbine = "ТНД"

## Плотность материала Д (кг/м^3):

$$\rho\_disk_i = \begin{bmatrix} 8266 & if material\_disk_i = "BЖ175" \\ 8320 & if material\_disk_i = "ЭП742" \\ 8393 & if material\_disk_i = "ЖС-6К" \\ 7900 & if material\_disk_i = "BT41" \\ 4500 & if material\_disk_i = "BT25" \\ 4570 & if material\_disk_i = "BT23" \\ 4510 & if material\_disk_i = "BT9" \\ 4430 & if material\_disk_i = "BT6" \\ NaN & otherwise \\ \end{bmatrix}$$

$$\rho\_{disk}^T = \boxed{\begin{array}{c|c} 1 \\ 1 \\ 8266 \end{array}}$$

$$\sigma\_disk\_long^T = \boxed{\begin{array}{c|c} 1 \\ \hline 1 & 620 \end{array}} \cdot 10^6$$

## Предел длительной прочности Д (Па):

$$\sigma_{disk\_long_i} = 10^6 \cdot \begin{vmatrix} 620 & \text{if material\_disk}_i = "BK175" \\ 680 & \text{if material\_disk}_i = "ЭП742" \\ 125 & \text{if material\_disk}_i = "KC-6K" \\ 123 & \text{if material\_disk}_i = "BT41" \\ 150 & \text{if material\_disk}_i = "BT25" \\ 230 & \text{if material\_disk}_i = "BT23" \\ 200 & \text{if material\_disk}_i = "BT9" \\ 210 & \text{if material\_disk}_i = "BT6" \\ NaN & \text{otherwise} \end{vmatrix}$$

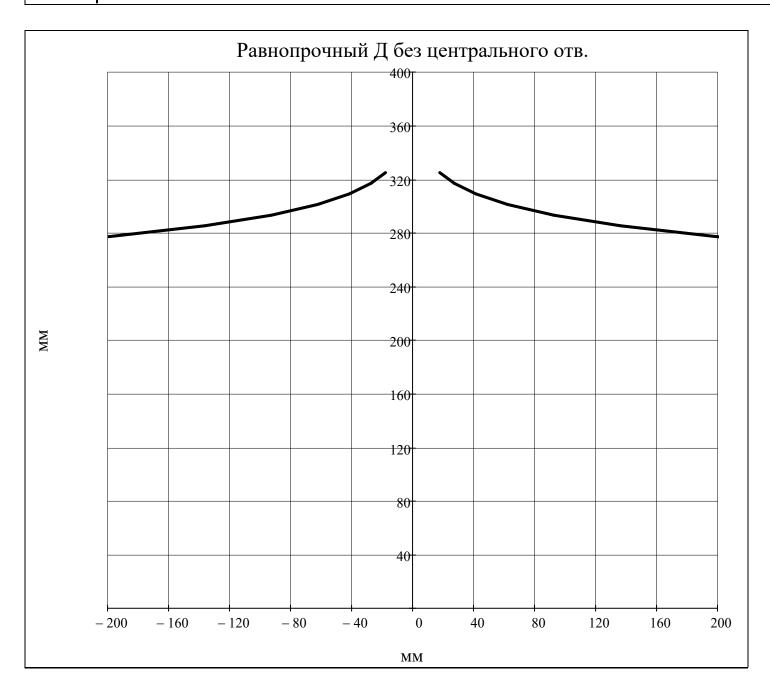
Рассматриваемая ступень: 
$$j_w = \begin{cases} j = Z \\ j = \end{cases}$$
 "Такой ступени не существует!" if  $(j < 1) \lor (j > Z)$ 

▼ Профилирование равнопрочного Д без центрального отв.

$$h(i,z) = \begin{pmatrix} chord_{rotor_{i},ORIGIN} \cdot sin \Big(\upsilon_{rotor_{i},ORIGIN}\Big) \Big) \cdot e^{\frac{\rho_{-}disk_{i} \cdot \omega^{2}}{2} \cdot \frac{1}{\sigma_{-}z_{rotor}(i,R_{st(i,2),ORIGIN})} \cdot \Big[ \big(R_{st(i,2),ORIGIN}\big)^{2} - z^{2} \big]} \\ \text{if } z \leq R_{st(i,2),ORIGIN} \end{pmatrix}$$

$$NaN \quad otherwise$$

$$z = 0, \frac{R_{st(j,2),ORIGIN}}{N_{dis}} .. R_{st(j,2),ORIGIN}$$



Рассматриваемая ступень: j = j = 1

$$j = \begin{vmatrix} j = 1 \\ j = \end{vmatrix}$$
 "Такой ступени не существует!" if  $(j < 1) \lor (j > Z)$   $j$  otherwise

•

Приведенная скорость по профилю Л

D.... Общие параметры охлаждения

— В параметры охлаждения

— В параметры охлаждения

— В параметры охлаждения

Конвективное охлаждение ——



▶ Результаты конвективного охлаждения ————————————————————————————————————		

**Г**Конвективно-пленочное охлаждение —

▶ Результат конвективно-пленочного охлаждения ———

Вывод результатов охлаждения

$$\begin{pmatrix} c_{st(j,1),r} \\ c_{st(j,2),r} \\ c_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 100.00 \\ 927.16 \\ 147.28 \end{pmatrix}$$

$$\begin{pmatrix} c_{a_{st(j,1),r}} \\ c_{a_{st(j,2),r}} \\ c_{a_{st(j,3),r}} \end{pmatrix} = \begin{pmatrix} 100.00 \\ 231.94 \\ 147.27 \end{pmatrix}$$

$$\begin{pmatrix} W_{st(j,1),r} \\ W_{st(j,2),r} \\ W_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 520.18 \\ 451.35 \\ 504.94 \end{pmatrix}$$

$$\begin{pmatrix} \alpha_{st(j,1),r} \\ \alpha_{st(j,2),r} \\ \alpha_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 90.00 \\ 14.49 \\ 89.31 \end{pmatrix} \cdot \circ$$

$$\begin{pmatrix} u_{st(j,1),r} \\ u_{st(j,2),r} \\ u_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 510.48 \\ 510.48 \\ 484.75 \end{pmatrix}$$

$$\begin{pmatrix}
\beta_{st(j,1),r} \\
\beta_{st(j,2),r} \\
\beta_{st(j,3),r}
\end{pmatrix} = \begin{pmatrix}
11.08 \\
149.08 \\
16.96
\end{pmatrix}$$

$$\varepsilon_{\mathrm{stator}_{j,r}} = 75.51^{\circ}$$

$$\varepsilon_{\text{rotor}_{j,r}} = 132.12^{\circ}$$

$$\begin{pmatrix} c_{st(j,1),r} \\ c_{st(j,2),r} \\ c_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 100.00 \\ 892.55 \\ 174.05 \end{pmatrix}$$

$$\begin{pmatrix} c_{a_{st(j,1),r}} \\ c_{a_{st(j,2),r}} \\ c_{a_{st(j,3),r}} \end{pmatrix} = \begin{pmatrix} 100.00 \\ 223.28 \\ 174.03 \end{pmatrix}$$

$$\begin{pmatrix} W_{st(j,1),r} \\ W_{st(j,2),r} \\ W_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 542.17 \\ 399.52 \\ 550.87 \end{pmatrix}$$

$$\begin{pmatrix} \alpha_{st(j,1),r} \\ \alpha_{st(j,2),r} \\ \alpha_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 90.00 \\ 14.49 \\ 90.87 \end{pmatrix}$$

$$\begin{pmatrix} u_{st(j,1),r} \\ u_{st(j,2),r} \\ u_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 532.86 \\ 532.86 \\ 520.00 \end{pmatrix}$$

$$\begin{pmatrix} \beta_{st(j,1),r} \\ \beta_{st(j,2),r} \\ \beta_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 10.63 \\ 146.02 \\ 18.42 \end{pmatrix} \cdot \circ$$

$$\varepsilon_{\mathrm{stator}_{j,r}} = 75.51^{\circ}$$

$$\varepsilon_{\text{rotor}_{j,r}} = 127.61^{\circ}$$

$$\begin{pmatrix} c_{st(j,1),r} \\ c_{st(j,2),r} \\ c_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 100.00 \\ 860.58 \\ 191.81 \end{pmatrix}$$

$$\begin{pmatrix} c_{a_{st(j,1),r}} \\ c_{a_{st(j,2),r}} \\ c_{a_{st(j,3),r}} \end{pmatrix} = \begin{pmatrix} 100.00 \\ 215.28 \\ 191.70 \end{pmatrix}$$

$$\begin{pmatrix} W_{st(j,1),r} \\ W_{st(j,2),r} \\ W_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 564.18 \\ 351.59 \\ 593.43 \end{pmatrix}$$

$$\begin{pmatrix} \alpha_{st(j,1),r} \\ \alpha_{st(j,2),r} \\ \alpha_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 90.00 \\ 14.49 \\ 91.90 \end{pmatrix} \cdot \circ$$

$$\begin{pmatrix} u_{st(j,1),r} \\ u_{st(j,2),r} \\ u_{st(j,3),r} \end{pmatrix} = \begin{pmatrix} 555.25 \\ 555.25 \\ 555.25 \end{pmatrix}$$

$$\begin{pmatrix}
\beta_{st(j,1),r} \\
\beta_{st(j,2),r} \\
\beta_{st(j,3),r}
\end{pmatrix} = \begin{pmatrix}
10.21 \\
142.24 \\
18.85
\end{pmatrix}$$

$$\varepsilon_{\mathrm{stator}_{j,r}} = 75.51^{\circ}$$

$$\varepsilon_{\text{rotor}_{j,r}} = 123.4^{\circ}$$