

Potentials and challenges in the application of peridynamics for the determination of virtual allowables

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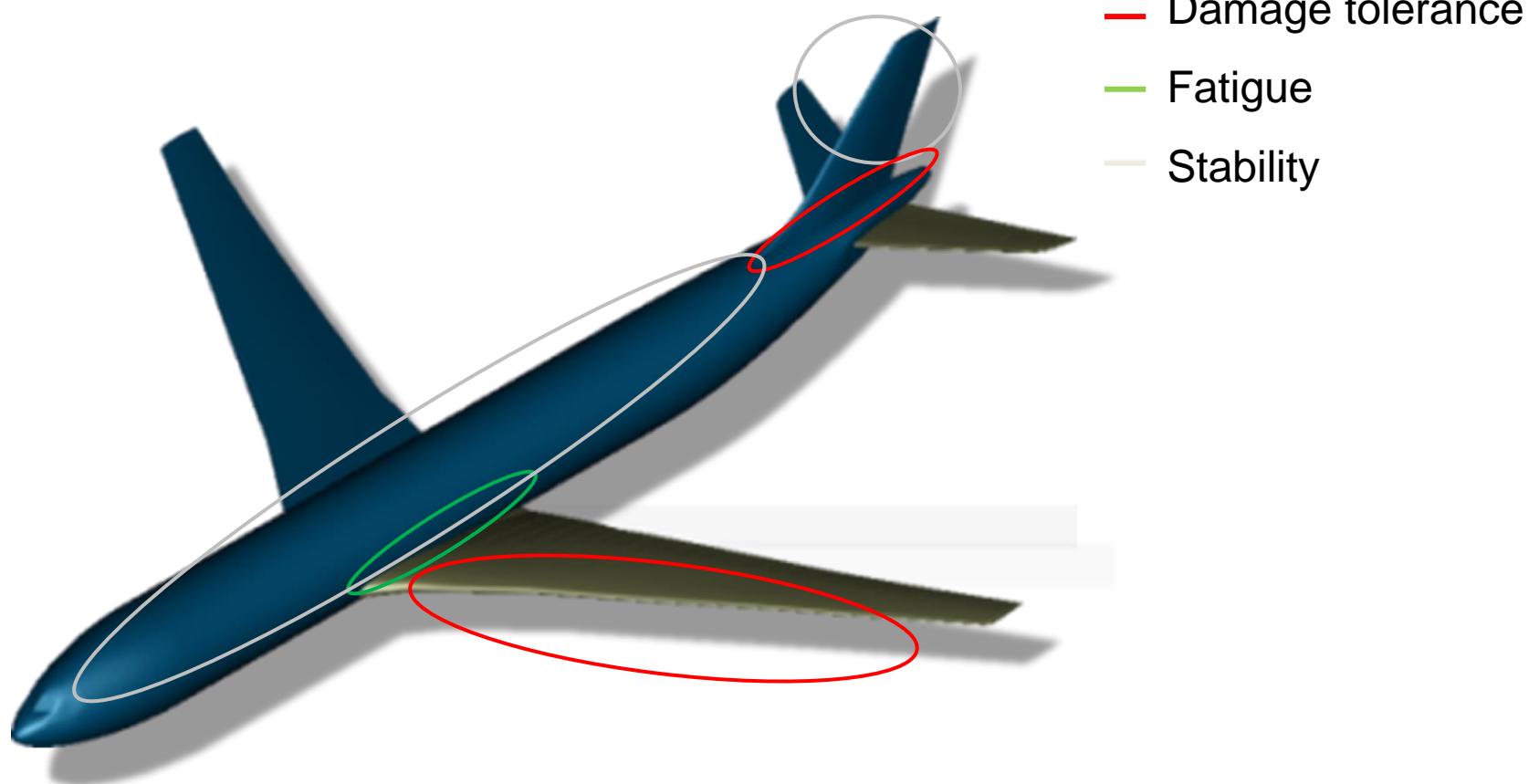
Knowledge for Tomorrow

Motivation – State-of-the-art design process in aeronautics

- Design criteria
 - Fatigue
 - Stability
 - ***Damage tolerance***
 - Plain and bearing strength
 - ...
- Example: Impact Analysis
 - Development of an experimental database of the most frequently used laminates
 - Not tested laminates are charged with safety factors
 - **Model-based characteristic values are only used to a limited extent, since trust in the simulations is lacking.**
 - fail safe (alternative load paths) vs. **safe life (no damage)**



Motivation - Main Design Driver

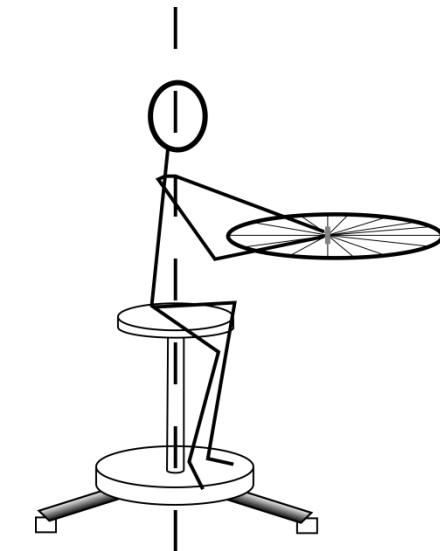
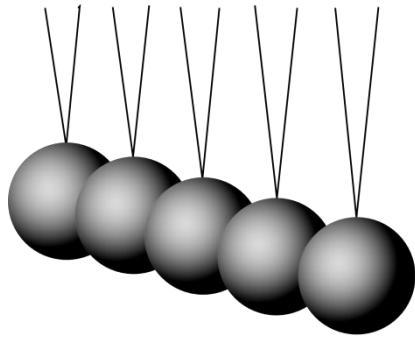


Motivation - Summary

- Micromechanical or damage models are not directly used in the design process
 - These models can be used to verify simplified criteria
 - Robustness of damaged structures can be evaluated
 - Reduction of cost-intensive experiments
-
- **A better understanding of damage initiation can be used to improve criteria and avoid expensive experiments.**



Physically motivated material modeling

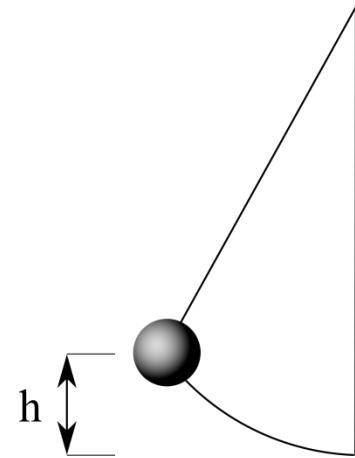


Conservation of momentum,

angular momentum

and

energy



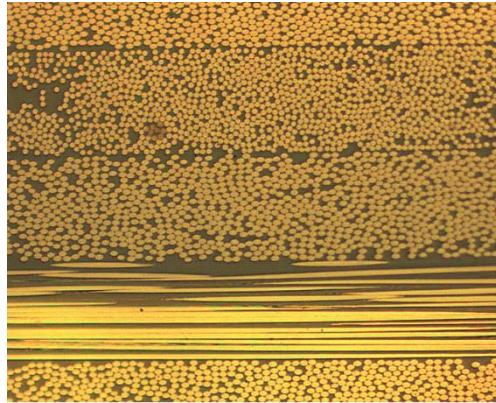
- If the conservation equations are fulfilled + if the material behaviour is described, it is a physically motivated modelling



Assumptions in classical continuum mechanics¹

1. The medium is continuous
2. Internal forces are contact forces (interaction only with the neighbourhood)
3. Deformations are twofold continuously derivable (in the weak formulation only simple)
4. The conservation equations are fulfilled

Points 1 and 3 are not fulfilled for heterogeneous materials and in case of damage



¹Bobaru, F.; Foster, J. T.; Geubelle, P. H. & Silling, S. A. „Handbook of Peridynamic Modeling“ CRC Press, 2016

Conservation equation

- Momentum conservation in continuum mechanics

$$\operatorname{div}(\boldsymbol{\sigma}) + \mathbf{b} = \rho \ddot{\mathbf{u}}$$

$$\operatorname{div}(\boldsymbol{\sigma}) = \left(\frac{\partial \sigma_{xx}}{\partial x} + \frac{\partial \sigma_{yx}}{\partial y} + \frac{\partial \sigma_{zx}}{\partial z}, \frac{\partial \sigma_{xy}}{\partial x} + \frac{\partial \sigma_{yy}}{\partial y} + \frac{\partial \sigma_{zy}}{\partial z}, \frac{\partial \sigma_{xz}}{\partial x} + \frac{\partial \sigma_{yz}}{\partial y} + \frac{\partial \sigma_{zz}}{\partial z} \right)$$

$$\boldsymbol{\sigma} = \mathbf{C} : \mathbf{E}$$

$$\mathbf{E} = 0.5(\operatorname{Grad} \mathbf{u} + \operatorname{Grad}^T \mathbf{u})$$

If points 1 and 3 are violated, the conservation of impulses is no longer strictly fulfilled!

Bertram, A. & Glüge, R. „Festkörpermechanik“ Otto-von-Guericke-Universität Magdeburg, ISBN 978-3-940961-88-4, 2013



Implications

- The continuum mechanics is not able to model damages
- FEM is a method for the numerical solution of a differential equation
- If the model based on a differential equation loses its validity, the FEM is also not able to model damage.

$$\mathbf{K} = \int_V \mathbf{B}^T \mathbf{C}_{Voigt} \mathbf{B} dV$$

$$\mathbf{B} = \mathbf{D}\mathbf{N}$$

D - Differential operator, which requires continuously differentiable ansatz functions **N**

Derivation only required in the element, i.e. discontinuities can be represented at the element boundaries. Continuum mechanics is not strictly fulfilled.



Damages

- Strictly speaking, they cannot be represented in FEM.
 - Damages are implemented via additional elements based on fracture mechanics.
- Problems with consistency!
- Additional assumptions to enable the jump between the theories (continuum mechanics and fracture mechanics).



Peridynamics

1. The medium is continuous
2. Internal forces are contact forces (interaction only with the neighbourhood)
3. Deformations are twofold continuously derivable (in the weak formulation only simple)
4. **The conservation equations are fulfilled**

$$\operatorname{div}(\boldsymbol{\sigma}) + \mathbf{b} = \rho \ddot{\mathbf{u}}$$

$$\int_H (\underline{\mathbf{T}}(\mathbf{x}, t) \langle \mathbf{q} - \mathbf{x} \rangle - \underline{\mathbf{T}}(\mathbf{q}, t) \langle \mathbf{x} - \mathbf{q} \rangle) dV + \mathbf{b} = \rho \ddot{\mathbf{u}}$$

$$\lim_{H \rightarrow 0} \int_H (\underline{\mathbf{T}}(\mathbf{x}, t) \langle \mathbf{q} - \mathbf{x} \rangle - \underline{\mathbf{T}}(\mathbf{q}, t) \langle \mathbf{x} - \mathbf{q} \rangle) dV = \operatorname{div}(\boldsymbol{\sigma})$$

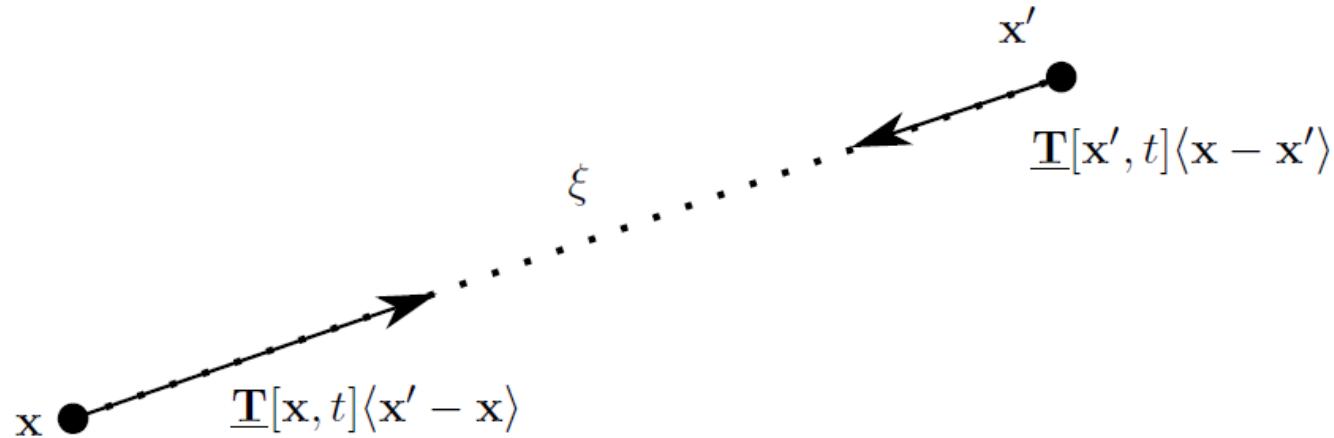


Peridynamic formulations

- bond based (Poisson's ratio of 0.25 for 3D & 2D plane strain)
 - too easy
 - a non-local spring formulation
 - should be avoided
- Extension to States
 - States are no longer feathers and should not be interpreted as such!
 - ordinary state based
 - Balance of forces is fulfilled at integral, however not at every bond potential.
 - non-ordinary state based
 - Force and moment equilibrium is fulfilled in the integral, but not in every bond potential.



Peridynamics – ordinary state based formulation



$$\rho(\mathbf{x}) \ddot{\mathbf{u}}(\mathbf{x}, t)$$

$$= \int_{\mathcal{H}} (\underline{\mathbf{T}}[\mathbf{x}, t] \langle \mathbf{x}' - \mathbf{x} \rangle - \underline{\mathbf{T}}[\mathbf{x}', t] \langle \mathbf{x} - \mathbf{x}' \rangle) dV + \mathbf{b}(\mathbf{x}, t)$$



Peridynamics – ordinary state based formulation

$$W_{CM} = \frac{1}{2}K [\epsilon_{kk}]^2 \delta_{ij} + 2G [\epsilon_{ij}^d]^2 \stackrel{!}{=} W_{PD}$$

$$\underline{\mathbf{Y}}\langle\xi\rangle = \mathbf{F}\xi = \mathbf{F}\langle\mathbf{x}' - \mathbf{x}\rangle \quad \forall \xi \in \mathcal{H}$$

- For small deformations and isotropic material

$$\underline{x} = |\underline{\mathbf{X}}\langle\xi\rangle| \qquad \underline{y} = |\underline{\mathbf{Y}}\langle\xi\rangle| \qquad \underline{e}\langle\xi\rangle = \underline{y} - \underline{x}$$

$$\underline{e}\langle\xi\rangle = |\mathbf{F}\xi| - |\xi| = \epsilon_{ij}\xi_i \frac{\xi_j}{|\xi|}$$

$$\underline{e}^d\langle\xi\rangle = \epsilon_{ij}^d\xi_i \frac{\xi_j}{|\xi|} \qquad \underline{e}^i\langle\xi\rangle = \epsilon_{ii}\xi_i \frac{\xi_i}{|\xi|}$$

$$W_{PD} = \frac{A}{2} \int_{\mathcal{H}} \underline{\omega}\langle\xi\rangle \left[\epsilon_{ij}^d \xi_i \frac{\xi_j}{|\xi|} \right]^2 dV_{\xi} + \frac{B}{2} \int_{\mathcal{H}} \underline{\omega}\langle\xi\rangle \left[\epsilon_{ii}\xi_i \frac{\xi_i}{|\xi|} \right]^2 dV_{\xi}$$



Peridynamics – ordinary state based formulation

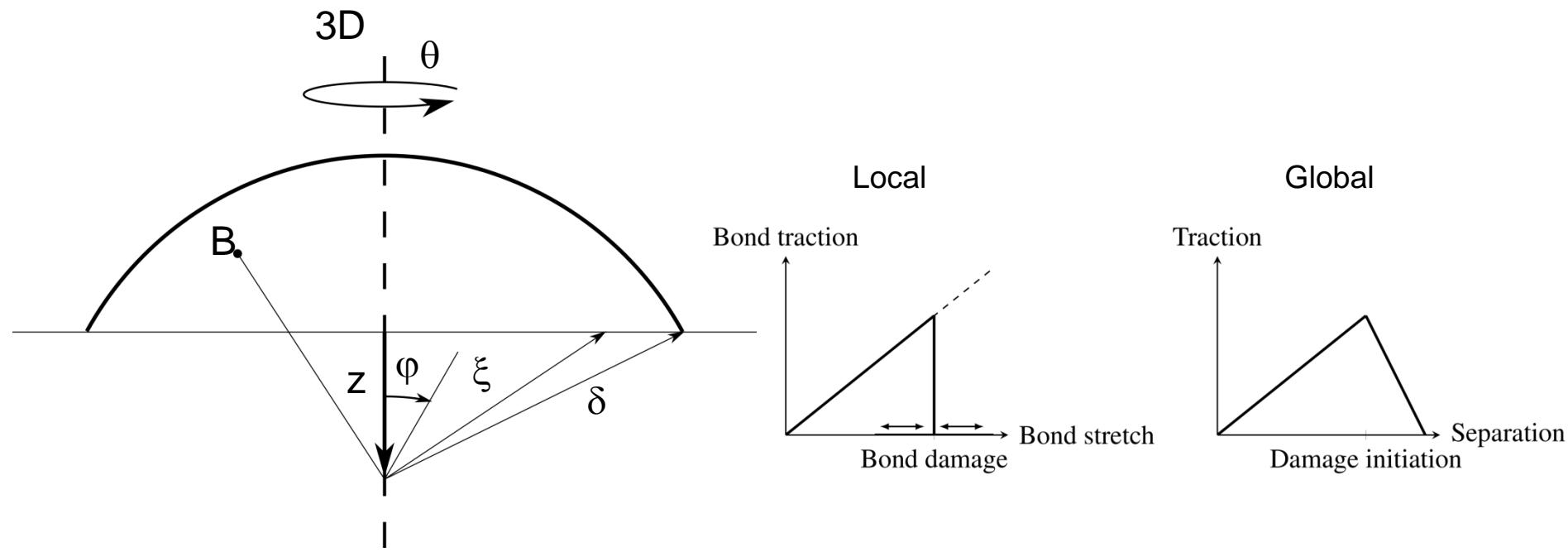
$$A = \frac{3K}{m_V} \quad \text{and} \quad B = \frac{15G}{m_V}$$

$$m_V = \int_{\mathcal{H}(x)} \underline{\omega} \langle \xi \rangle \underline{x} \underline{x} \, dV_\xi \quad \theta = \frac{3}{m_V} \int_{\mathcal{H}(x)} \underline{\omega} \langle \xi \rangle \underline{x} e \langle \xi \rangle \, dV_\xi$$

$$\underline{t} \langle \xi, t \rangle = \frac{\underline{\omega} \langle \xi \rangle}{m_V} [3K\theta \underline{x} + 15G \underline{e}^d]$$

$$\underline{\mathbf{T}} = \underline{t} \frac{\underline{\mathbf{Y}}}{|\underline{\mathbf{Y}}|}$$

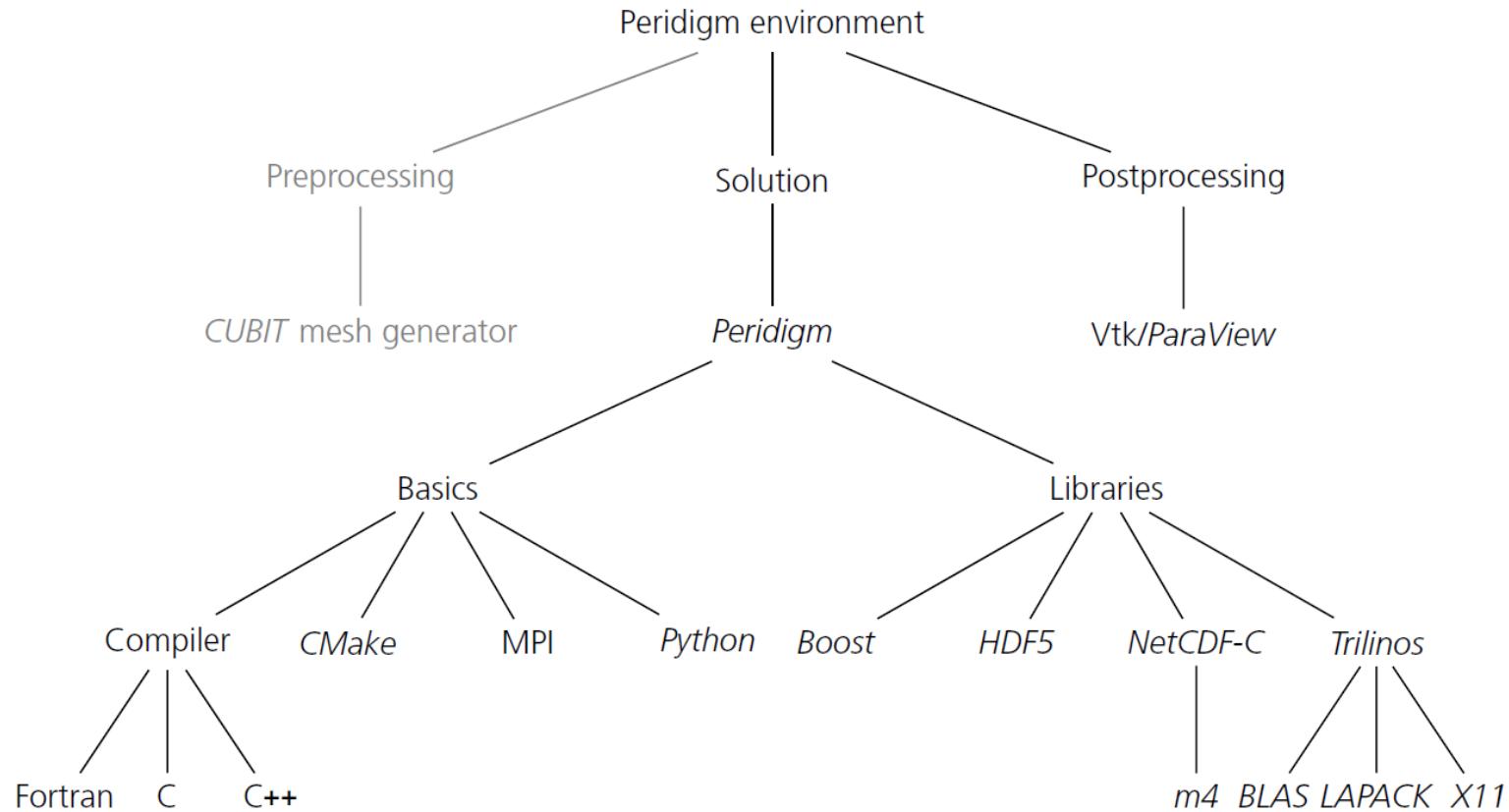




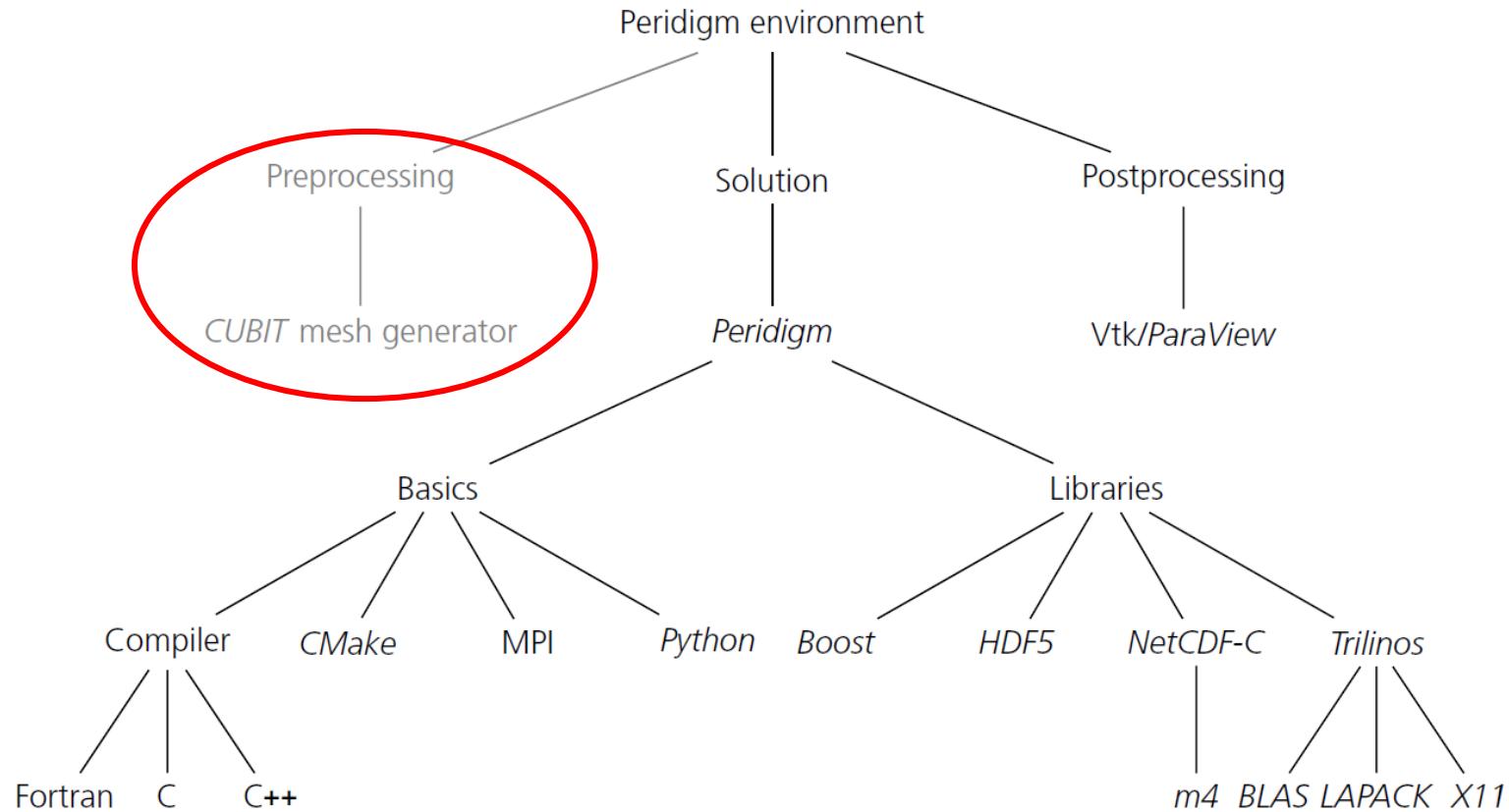
$$G = 2 \int_0^\delta \int_{H_r} w_c dV_\xi \quad G = \int_0^\delta \int_0^{2\pi} \int_z^\frac{\delta}{\cos^{-1} z/\xi} w_c \xi^2 \sin \varphi d\varphi d\xi d\theta dz$$

$$w_c = \frac{4G}{\pi \delta^4}$$

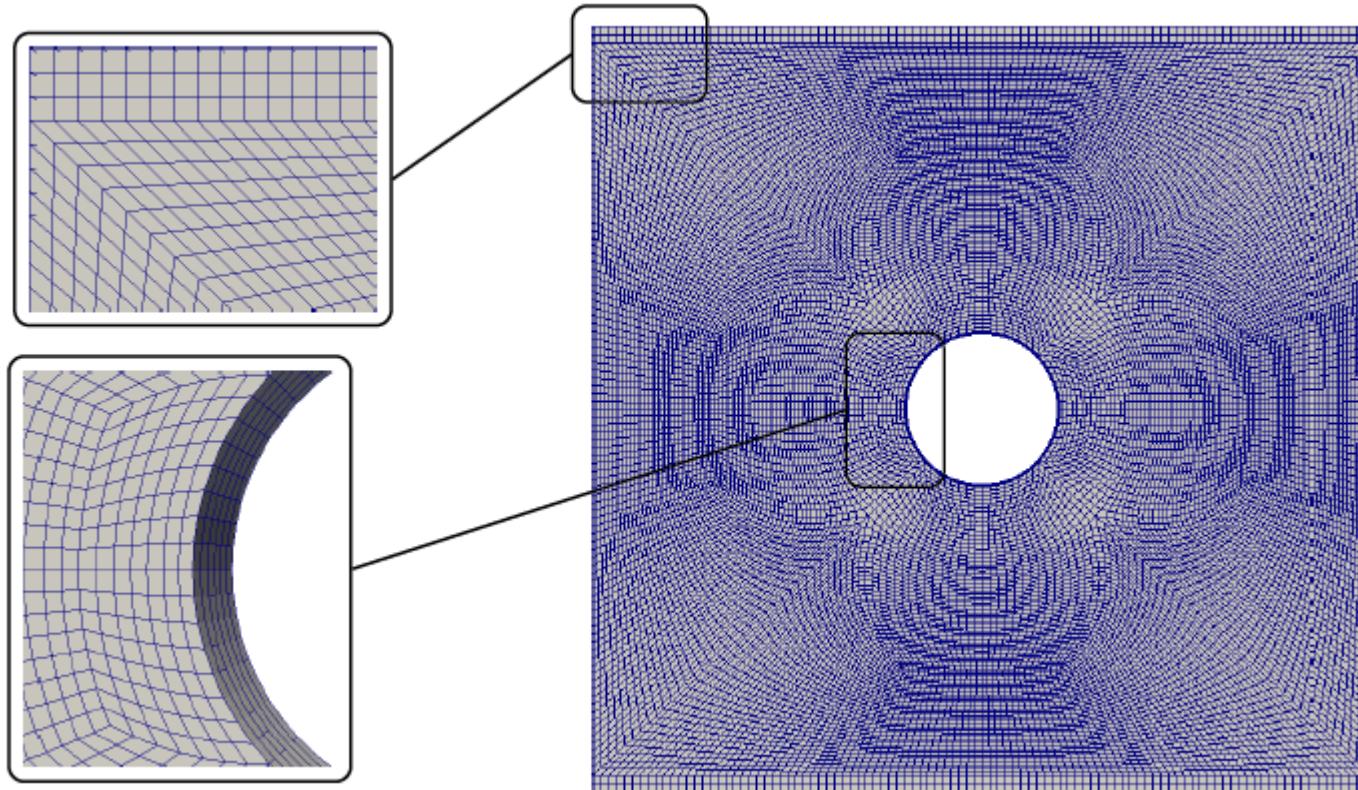
Framework



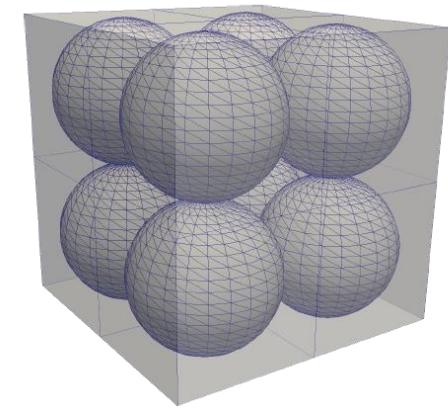
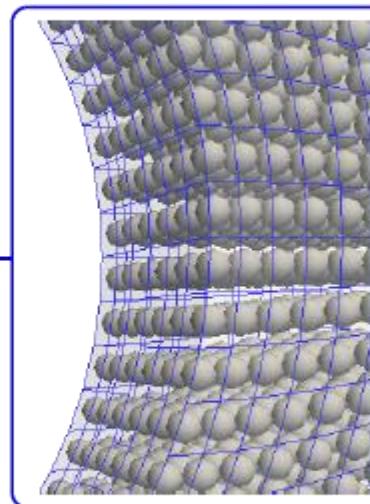
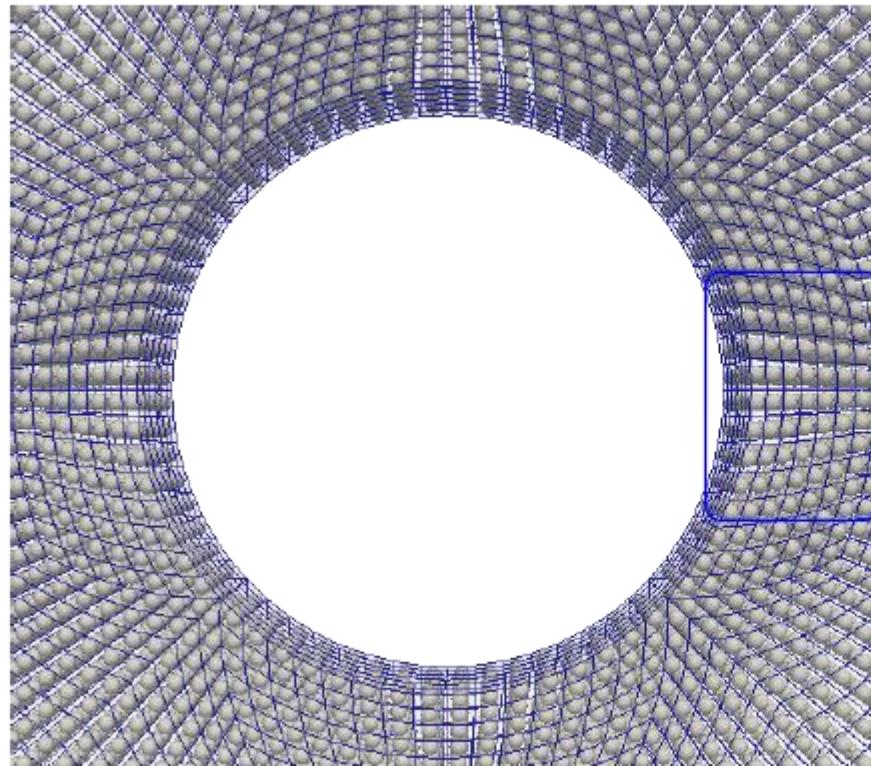
Framework



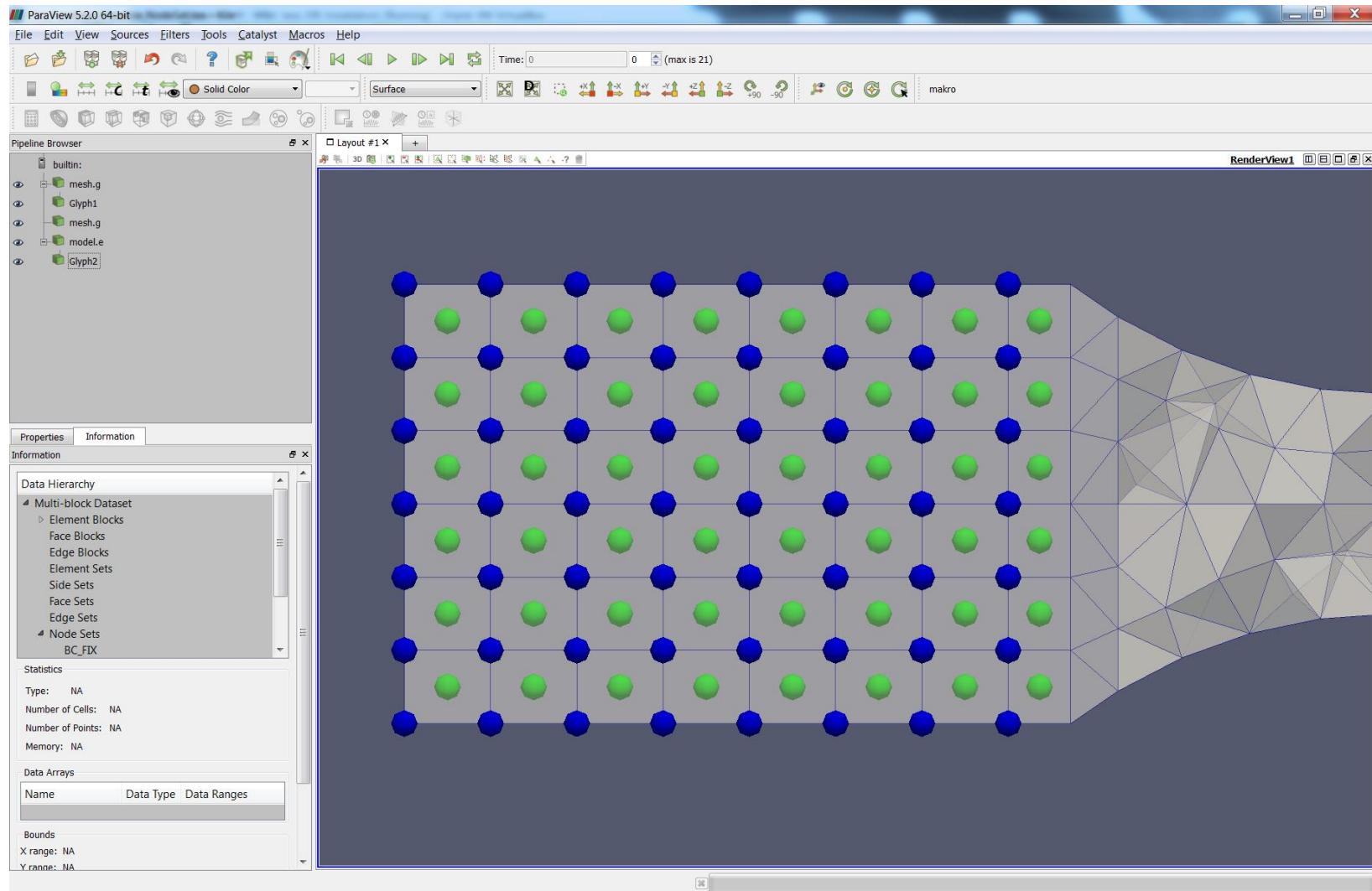
Preprocessor



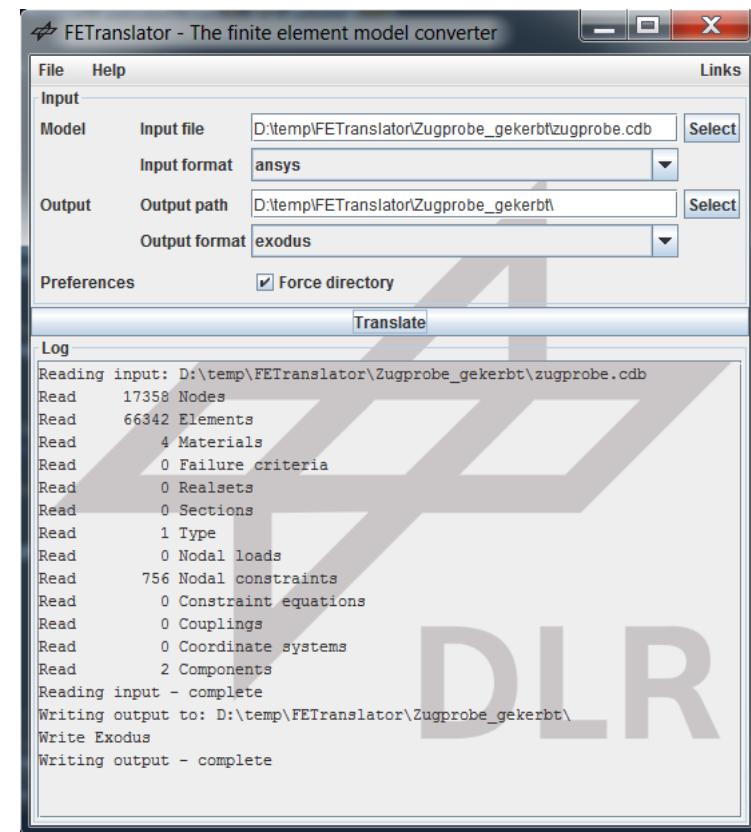
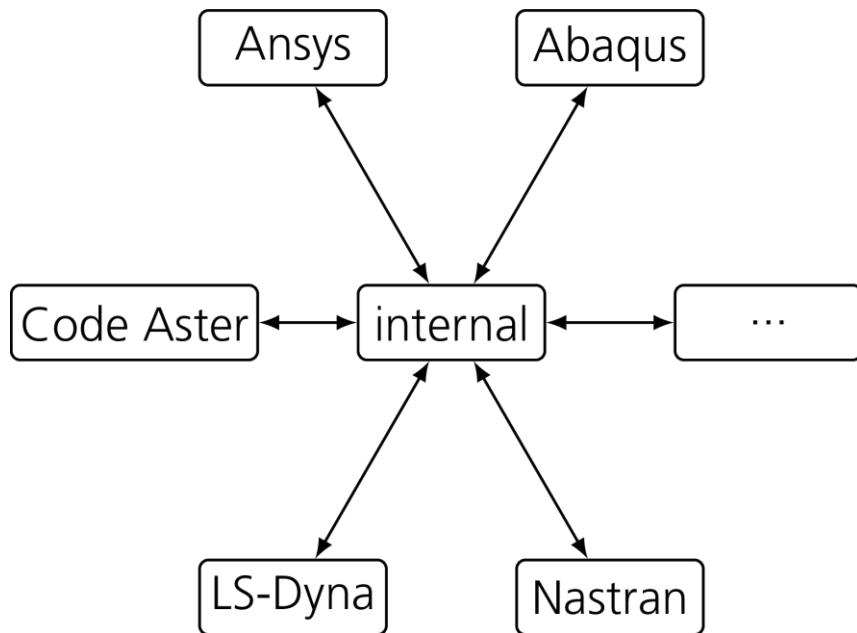
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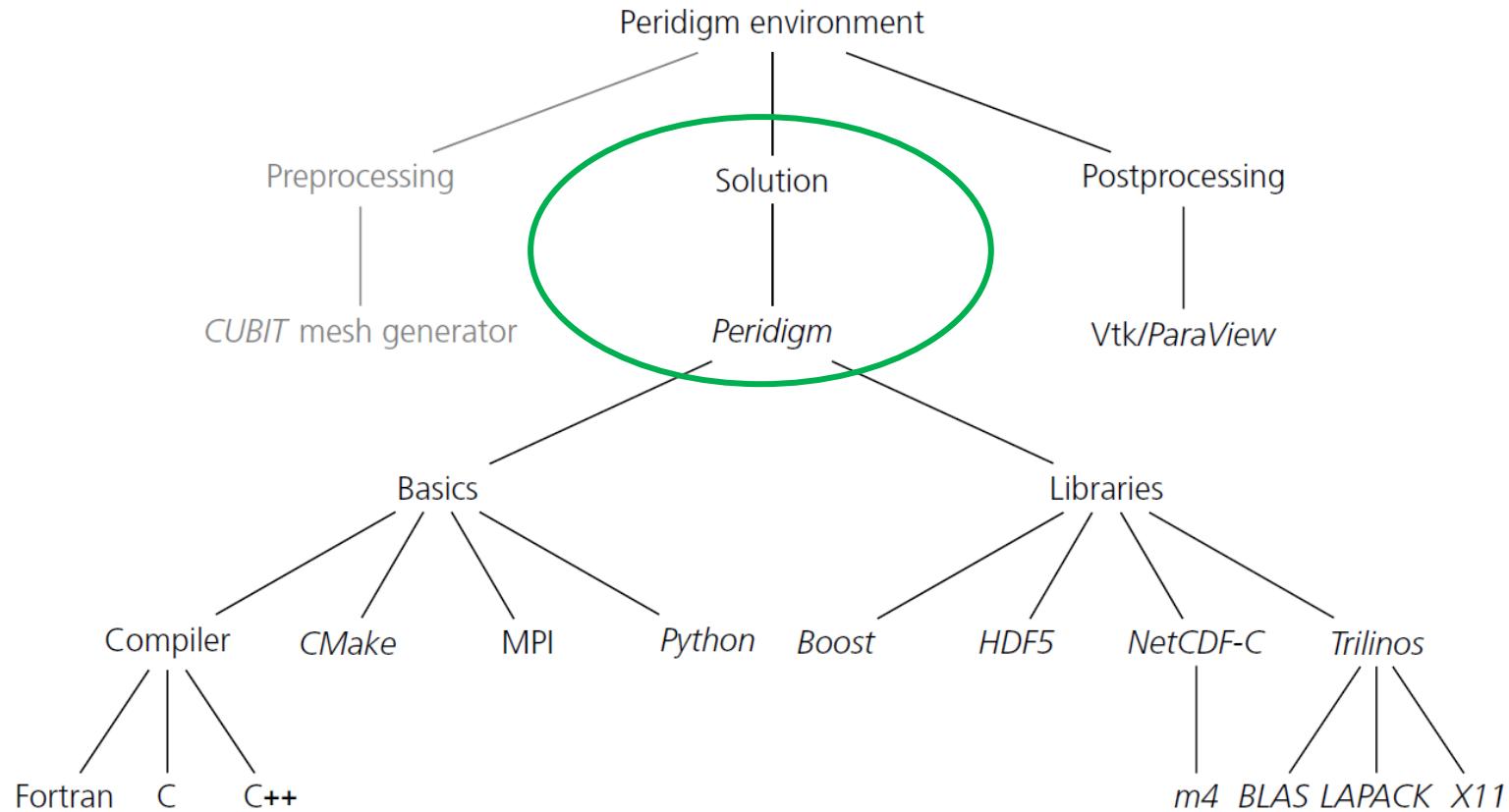
Preprocessor



Preprocessor



Framework



Damage models

$$\chi(\xi, t) = \begin{cases} 1 & \text{no failure} \\ 0 & \text{failure} \end{cases}$$

- Critical energy model by Foster et al.

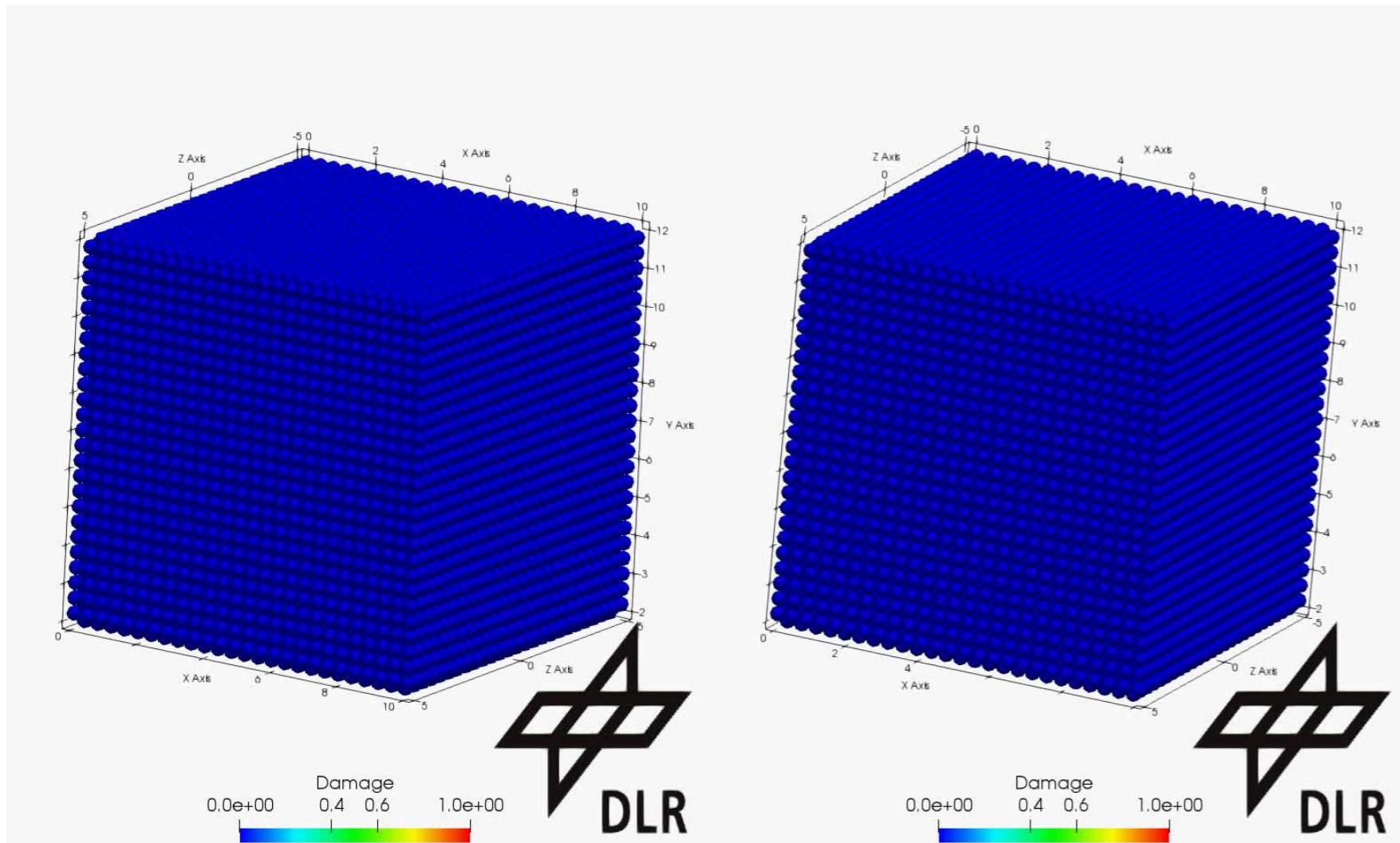
$$W_C = \frac{4G_{0C}}{\pi\delta^4}$$

$$W_{\text{bond}} = 0.25\chi(\underline{e}\langle\xi\rangle, t) \{\underline{t}[\mathbf{x}, t] - \underline{t}[\mathbf{x}', t]\} \underline{e} > W_C$$

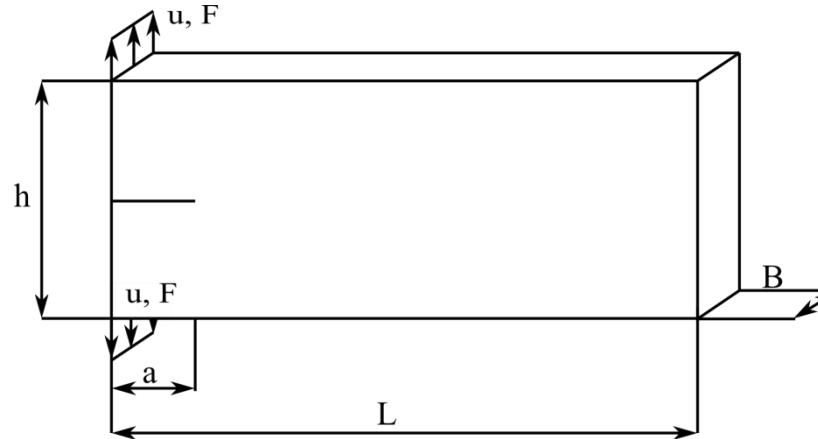
- Critical stretch model

$$s_C = \sqrt{\frac{G_{0C}}{\left[3G + \left(\frac{3}{4}\right)^4 \left(K - \frac{5G}{3}\right)\right] \delta}}$$





Verification

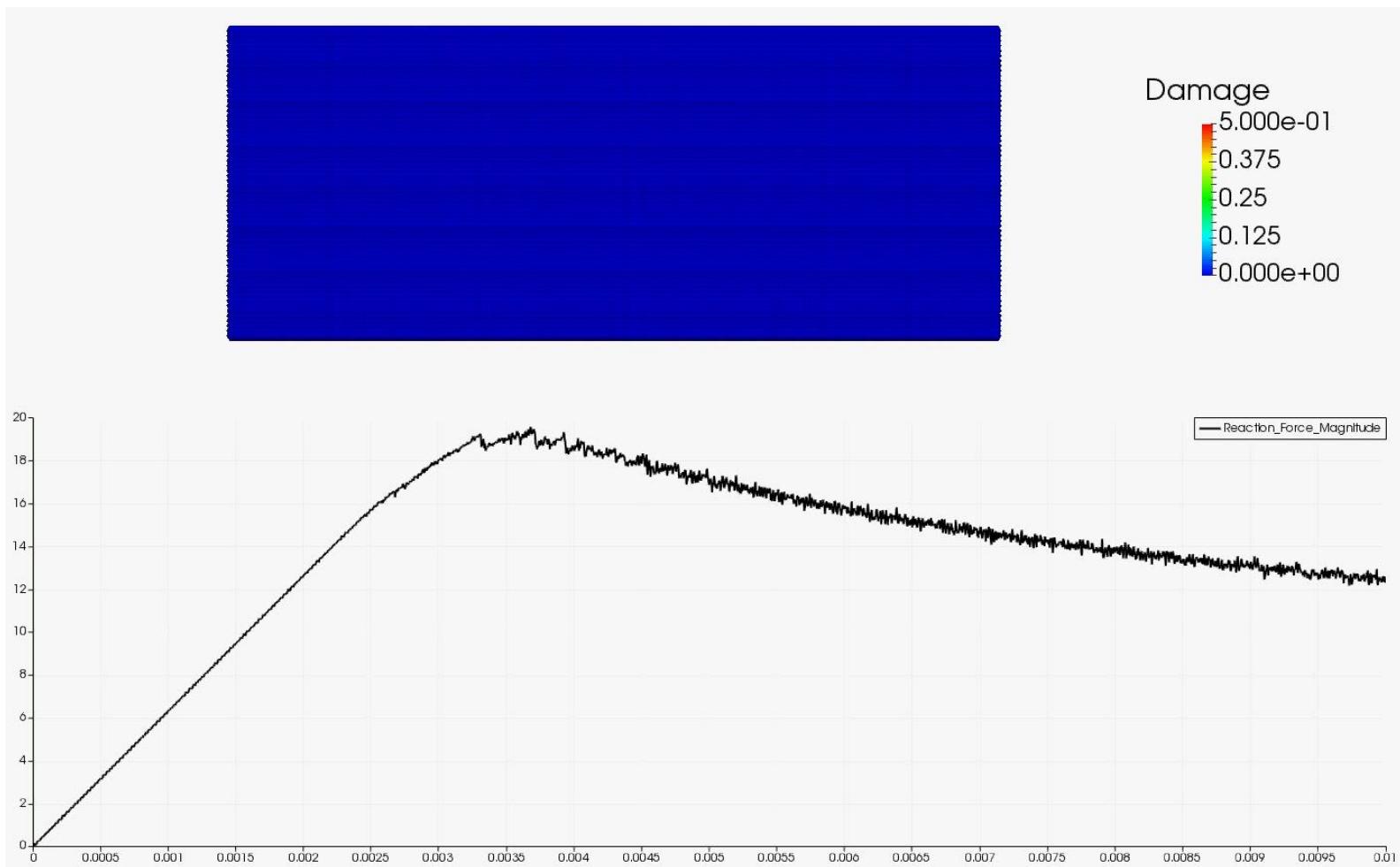


Geometry	a	h	L	B
	0.005m	0.02m	0.05m	0.003m

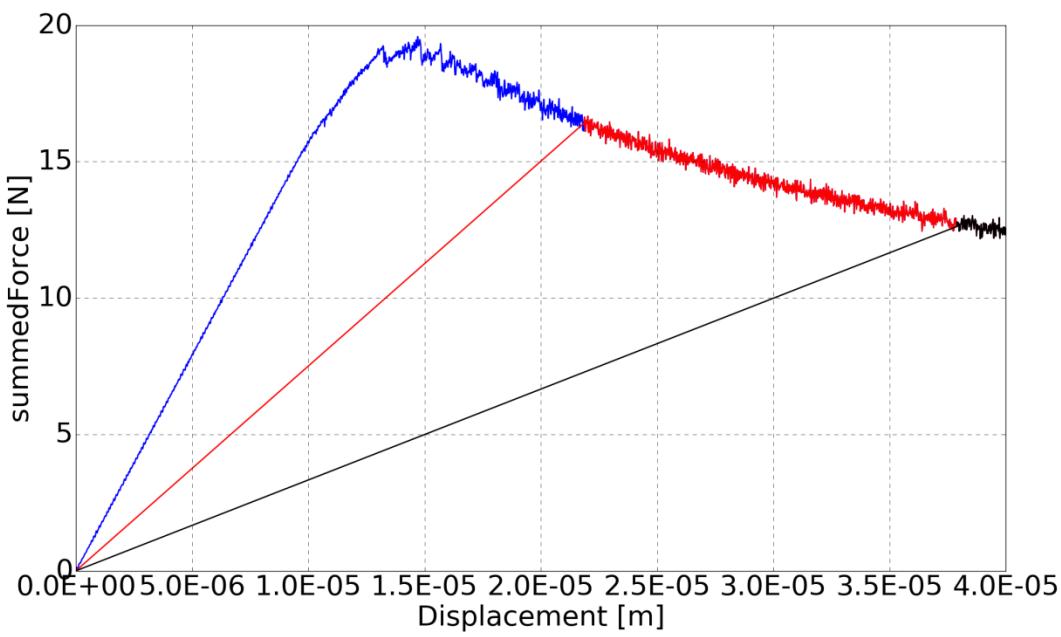
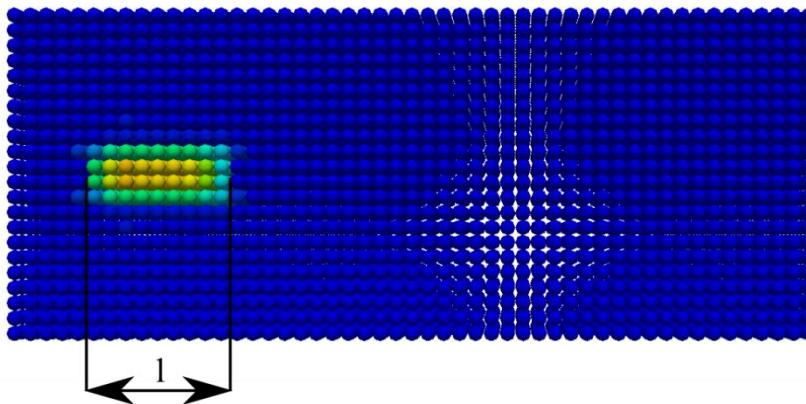
Material	Bulk Modulus	Shear Modulus	Density	G_0
	1.75E+09 Nm ⁻²	8.08E+08 Nm ⁻²	2000 kgm ⁻³	12 Nm ⁻¹

Mesh	2.01dx	3.01dx	4.01dx	5.01dx
0.0005	0.001005	0.001505	0.002005	0.002505
0.00033	0.000663	0.000993	0.001323	0.001653
0.00025	0.000503	0.000753	0.001003	0.001253
0.000125	0.000251		0.000501	

Verification: Double Cantilever Beam (DCB)

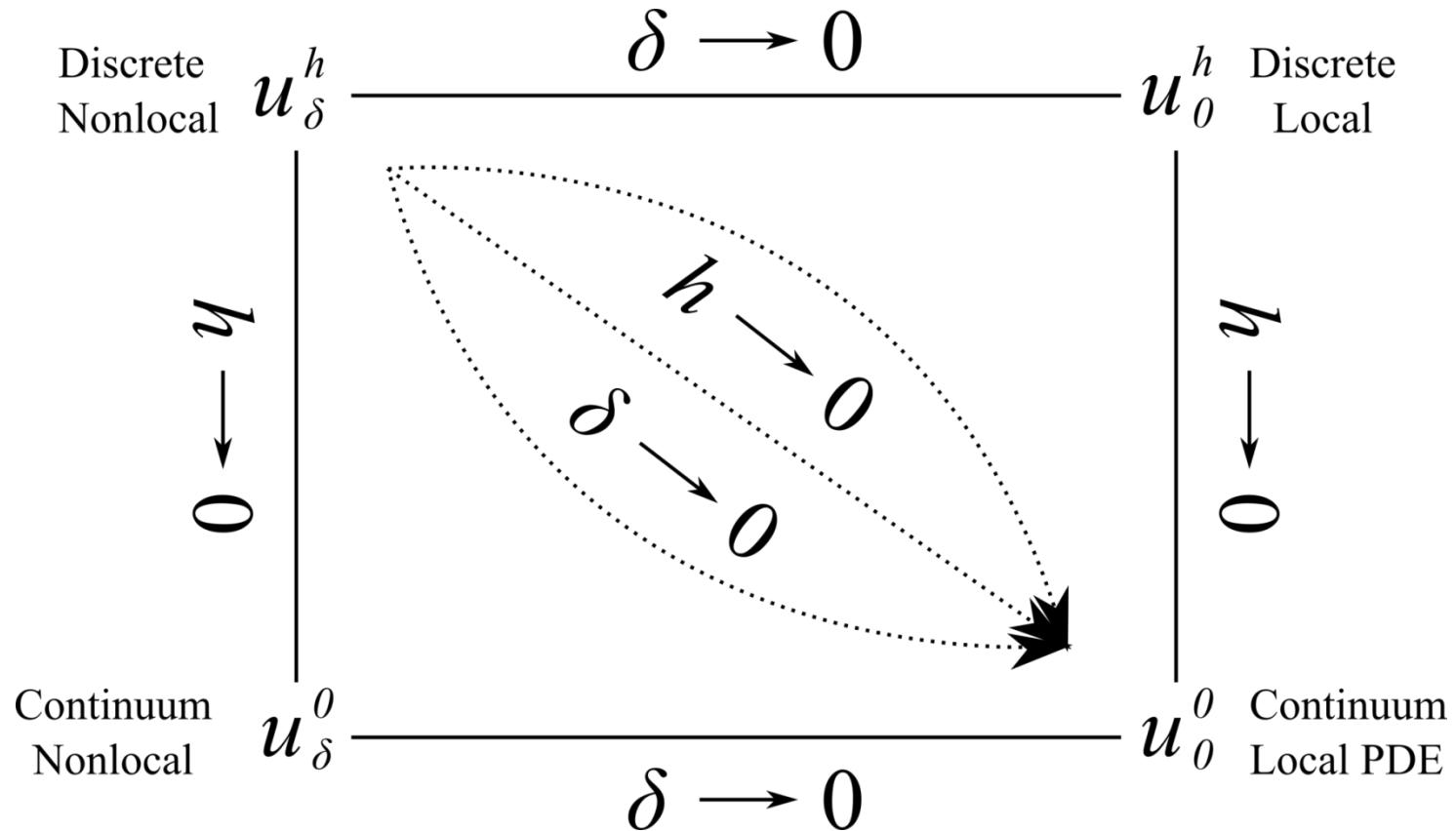


Verification

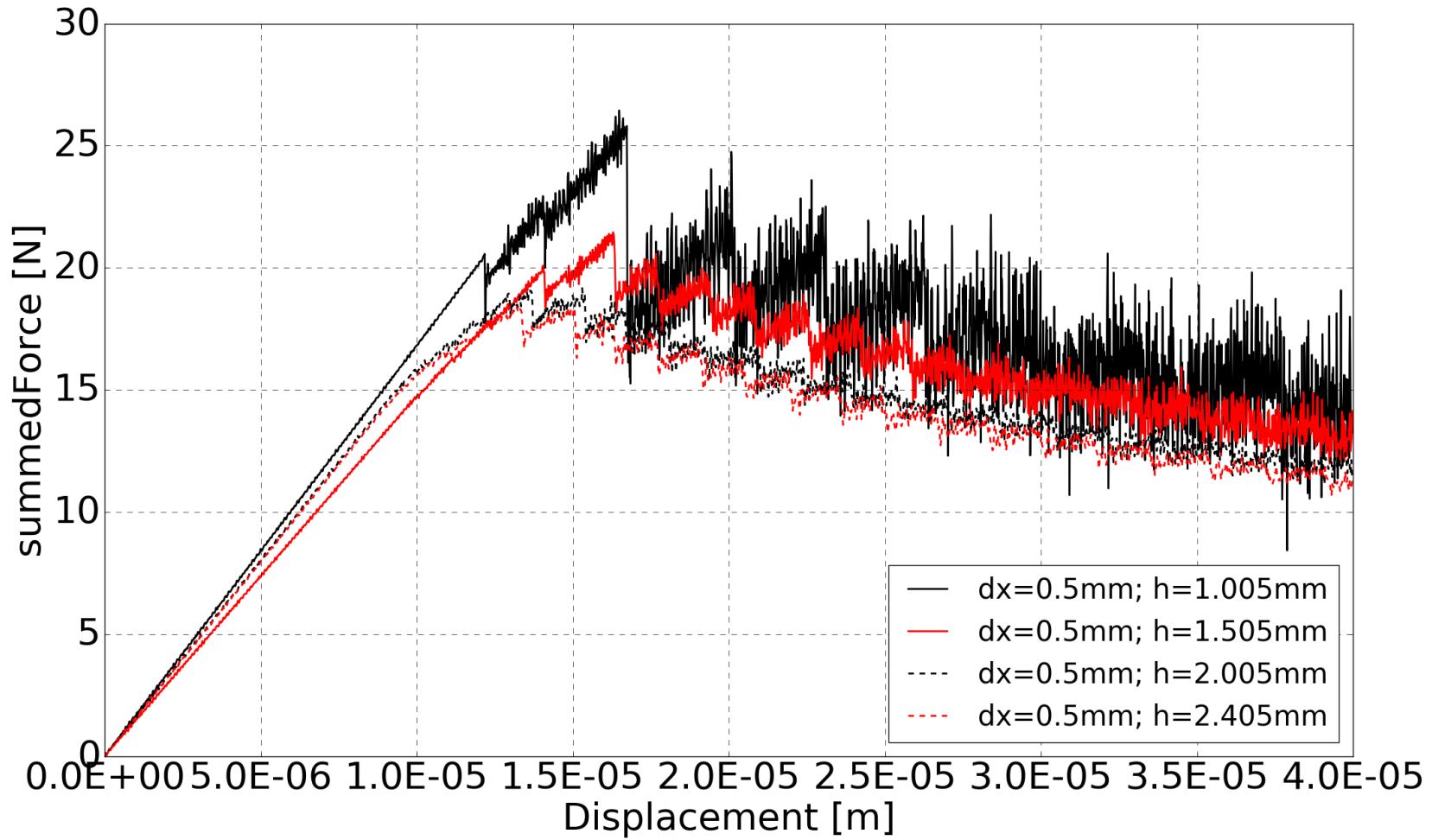


δ [m]	G_0 [N/m]	G_0 [N/m]
$2.015 \cdot 10^{-3}$	12.8	11.4
$3.015 \cdot 10^{-3}$	13.1	12.9
$4.015 \cdot 10^{-3}$	11.1	11.3
$5.015 \cdot 10^{-3}$	11.2	11.9

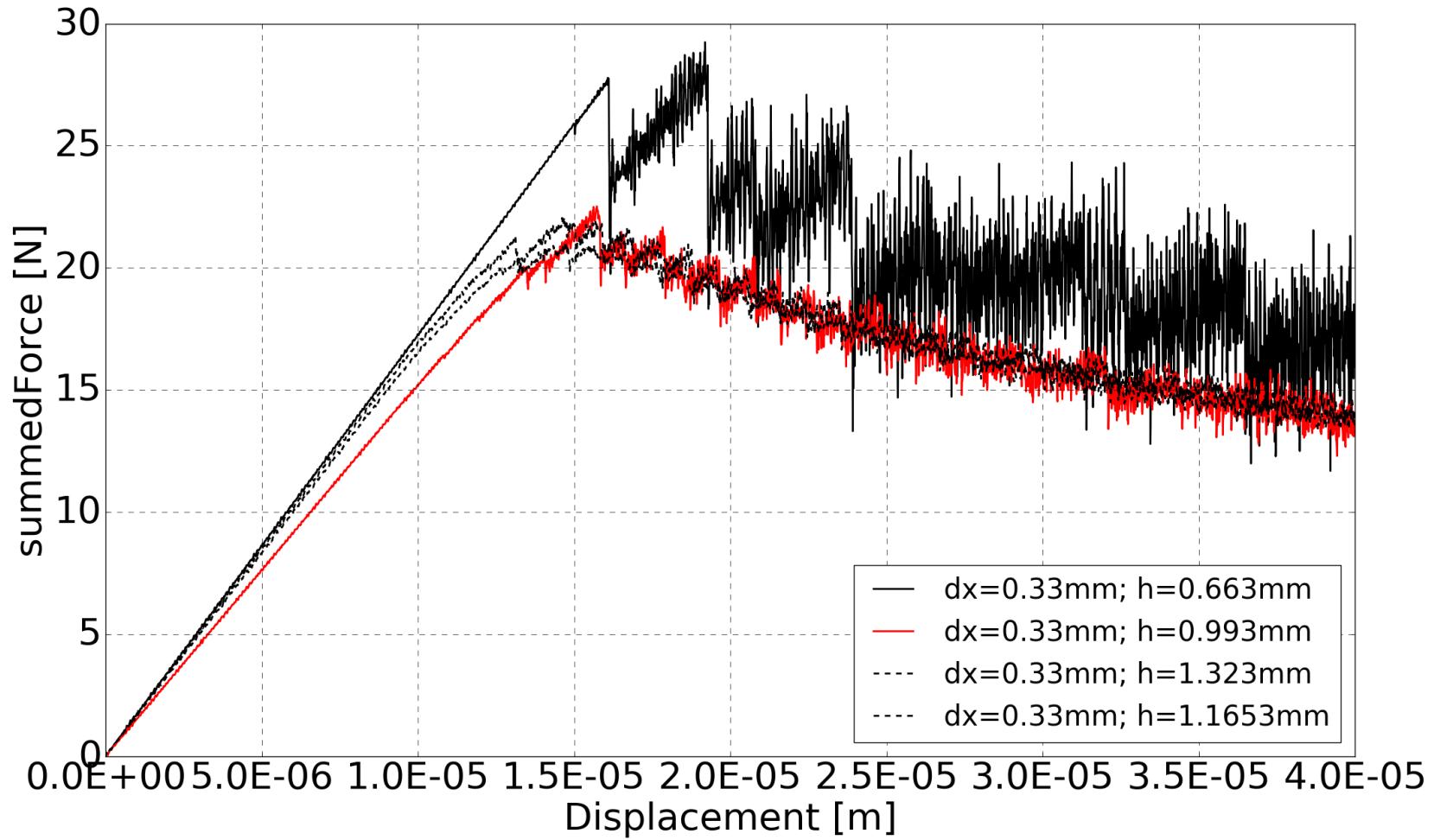
Verification: Convergence



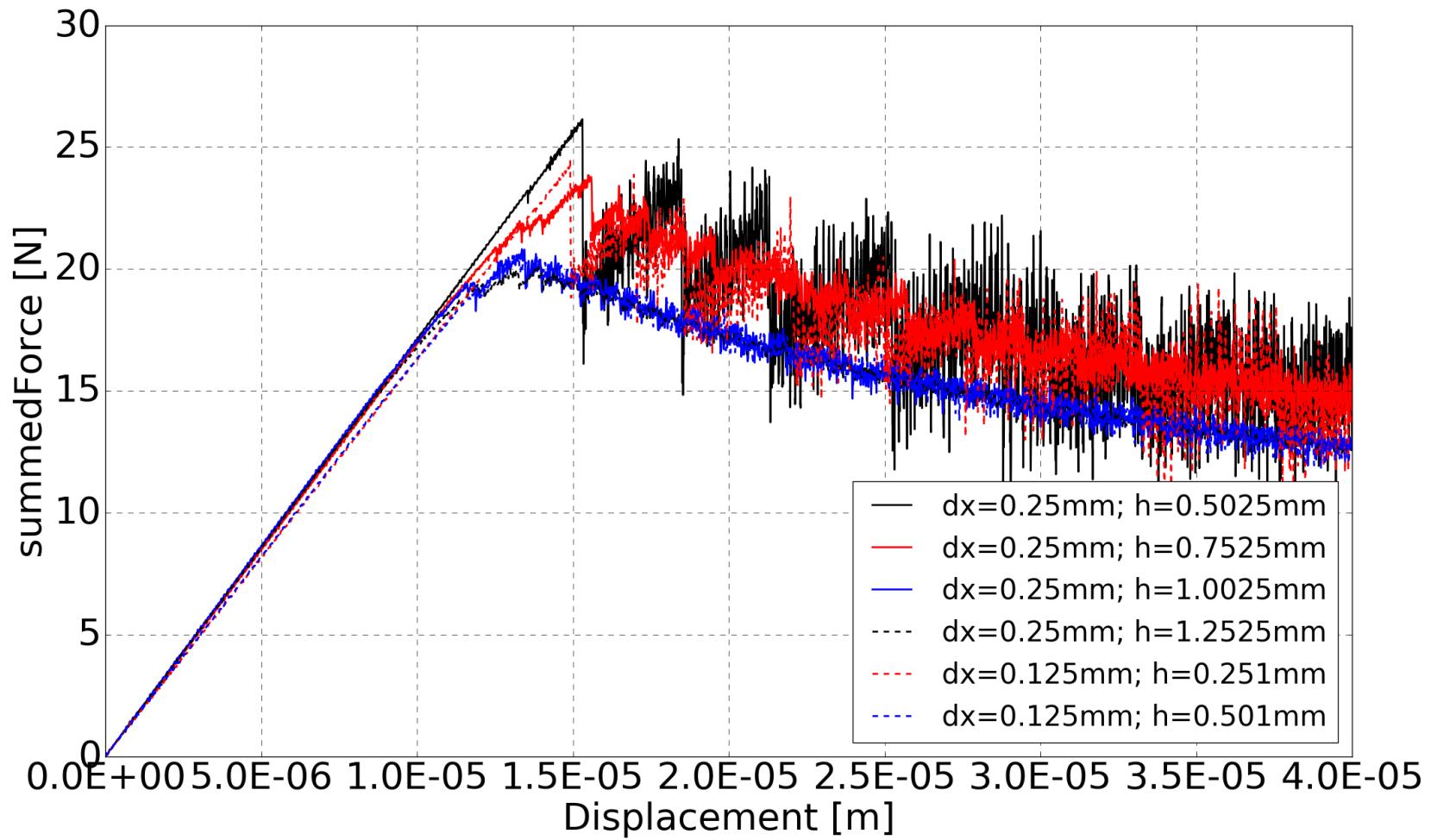
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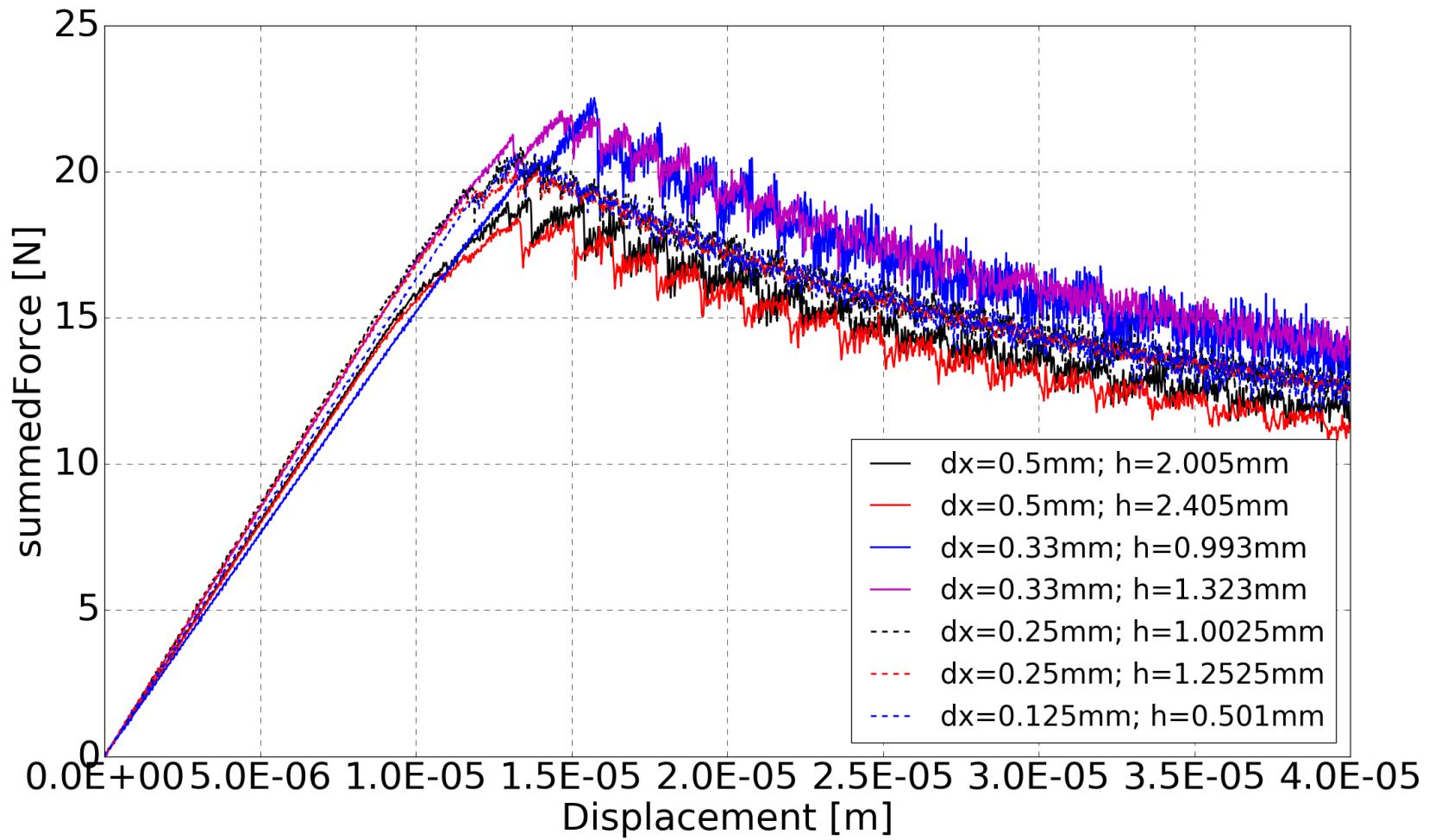
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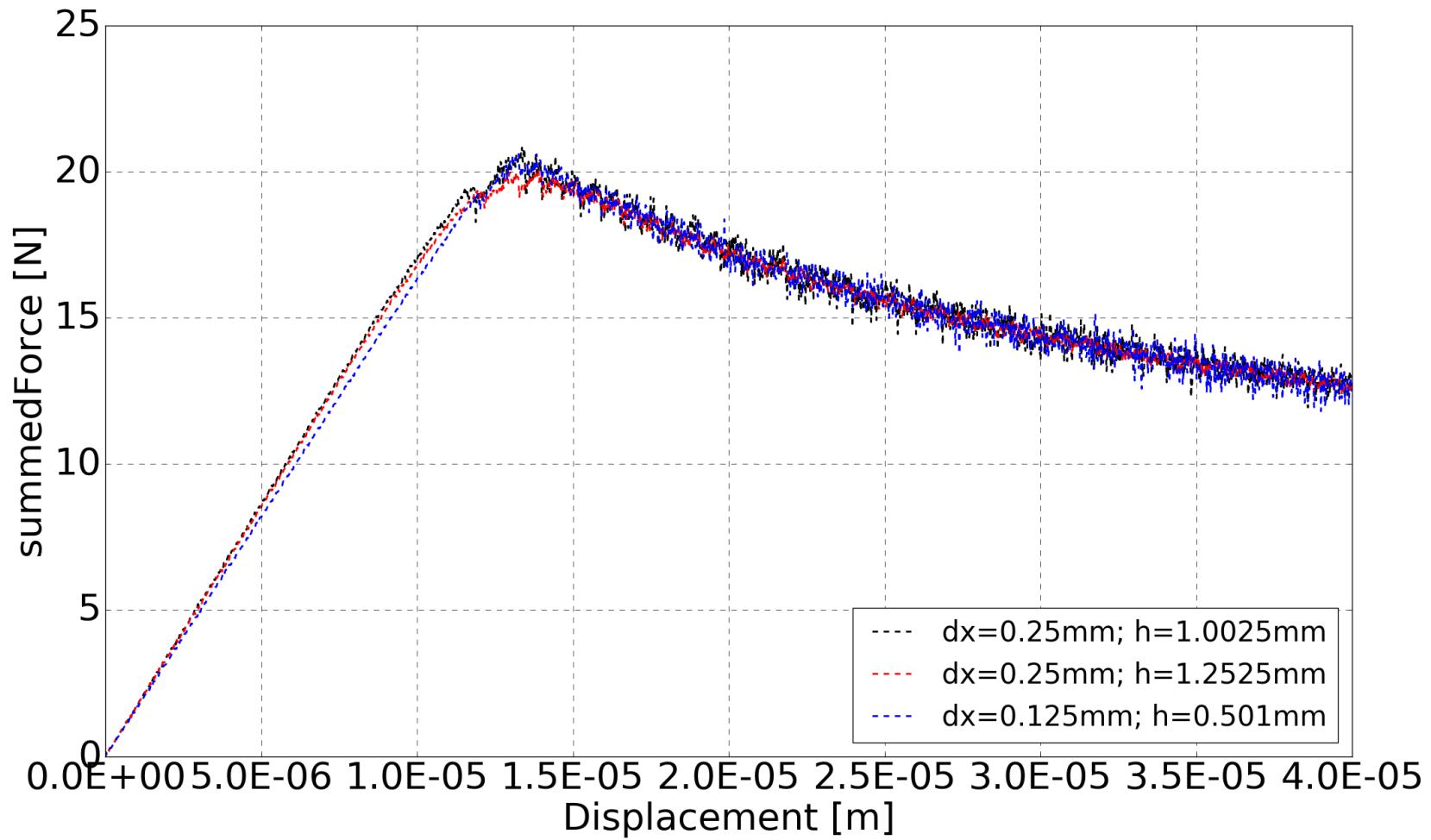
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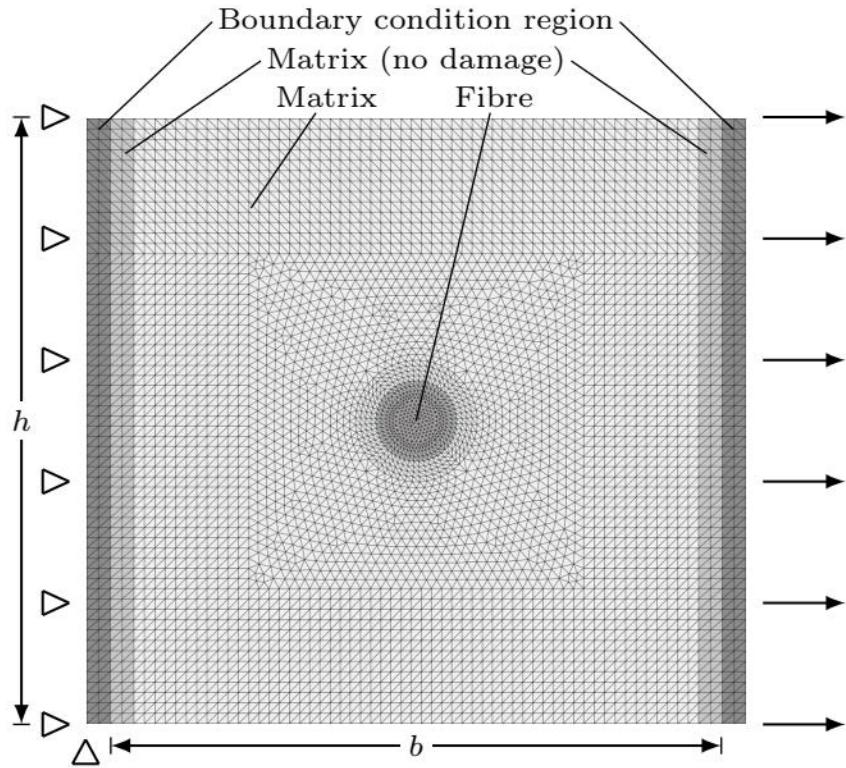
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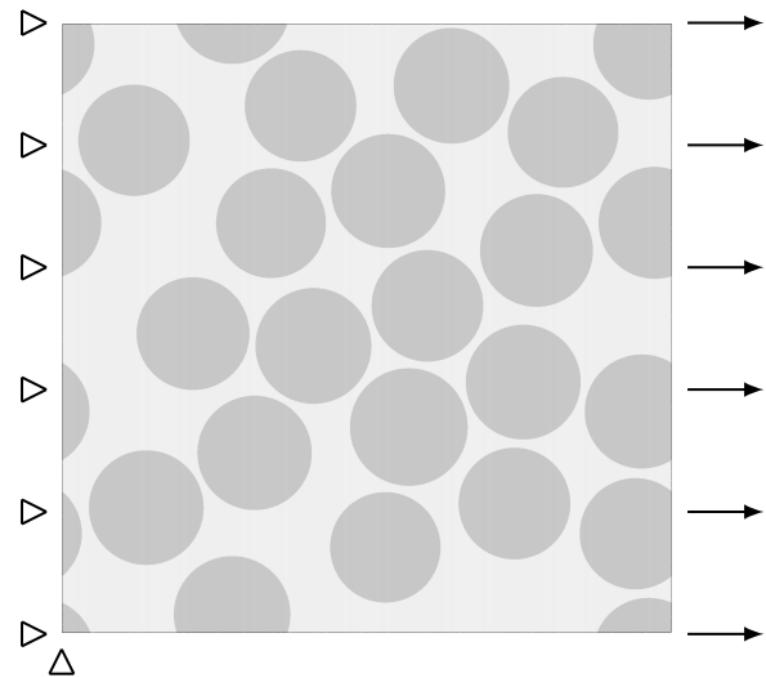
Verification: Convergence



Comparison: Fibre-matrix models

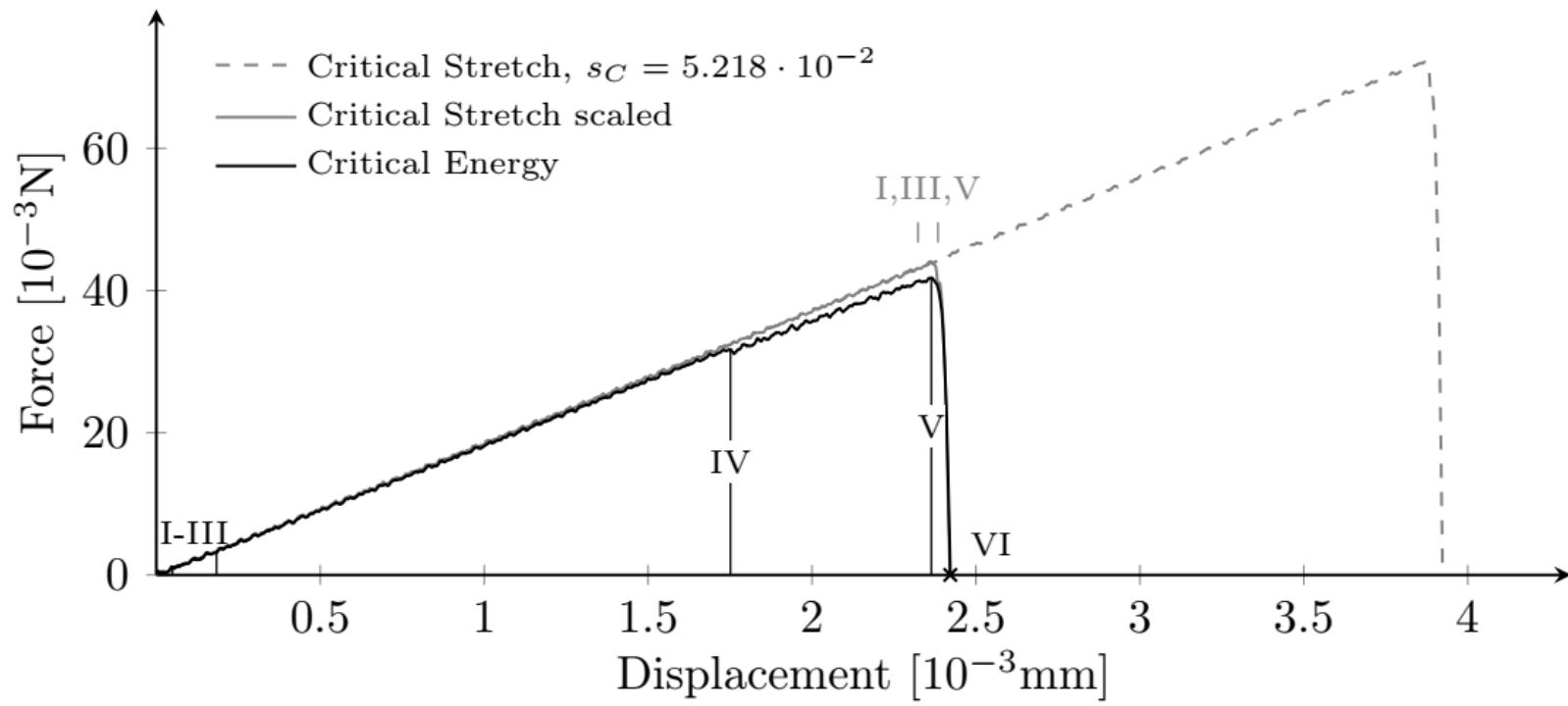


Single fibre

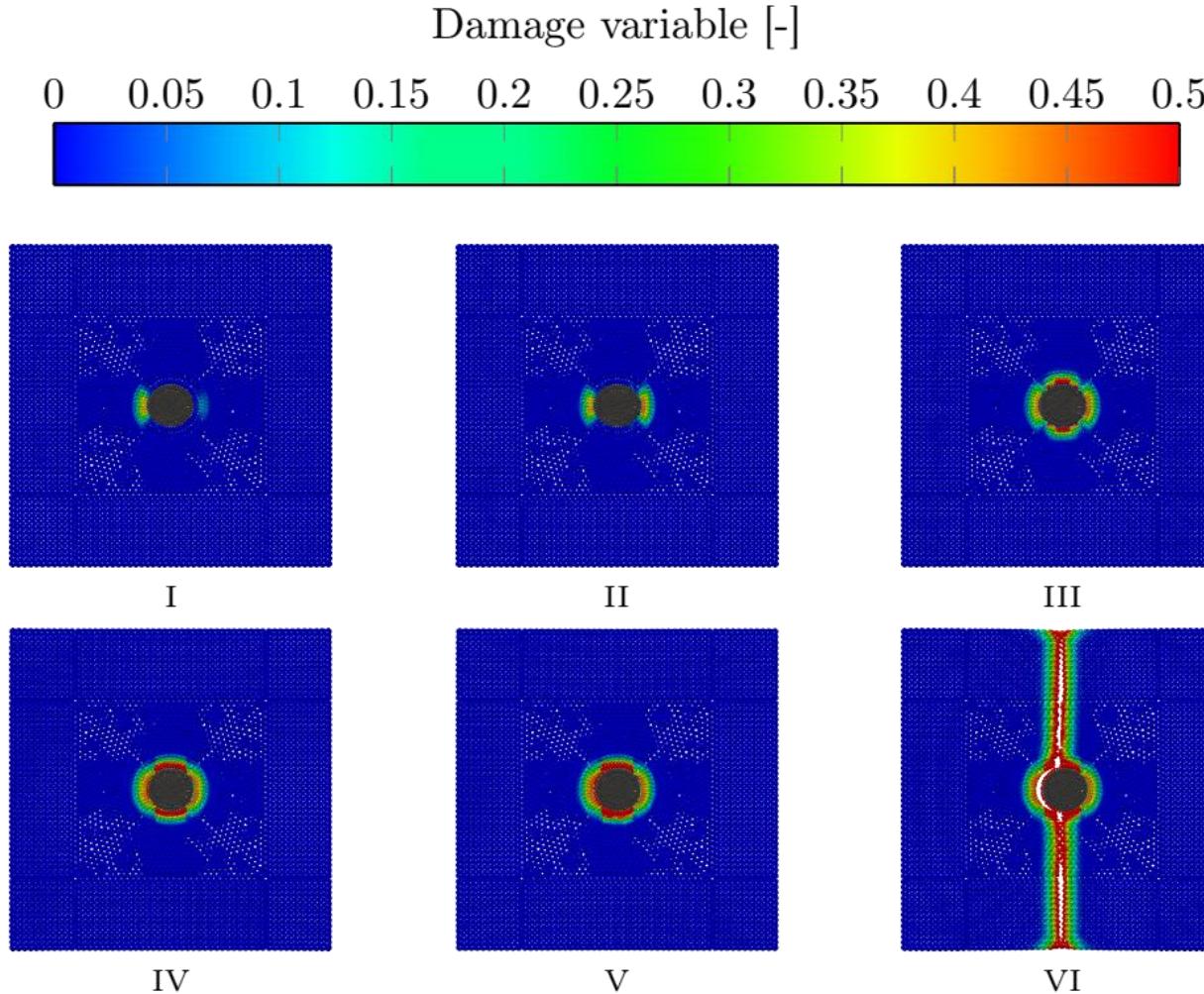


RVE

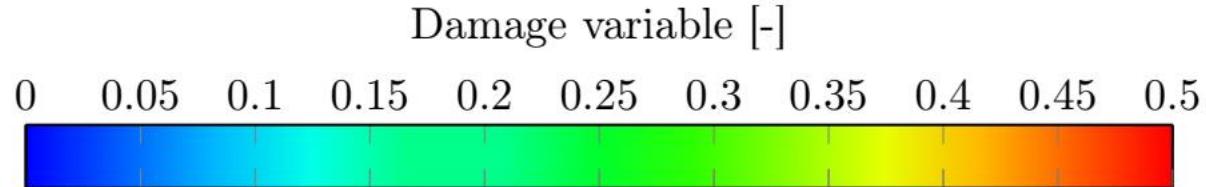
Comparison: Single fibre – Force-Displacement



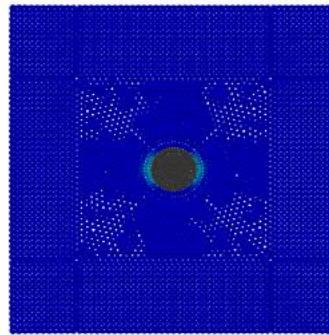
Comparison: Single Fibre – Energy Criterion



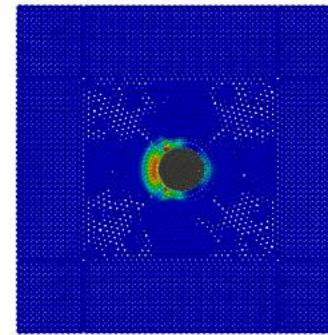
Comparison: Single fibre



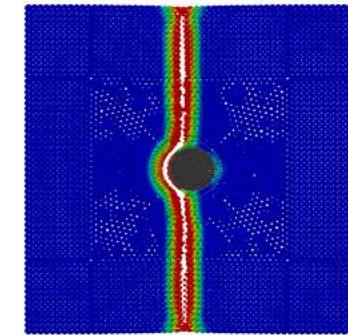
Critical Stretch



I

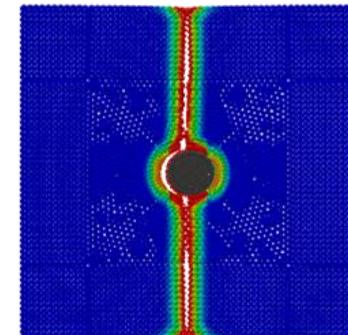
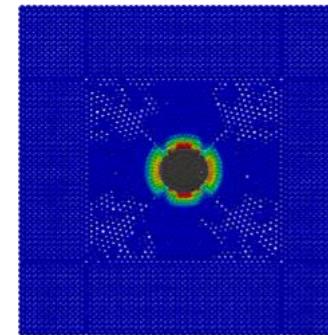
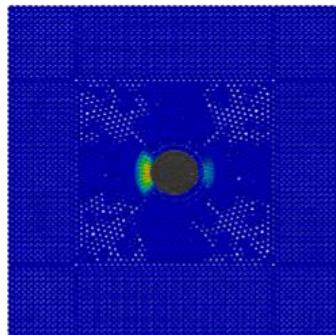


III

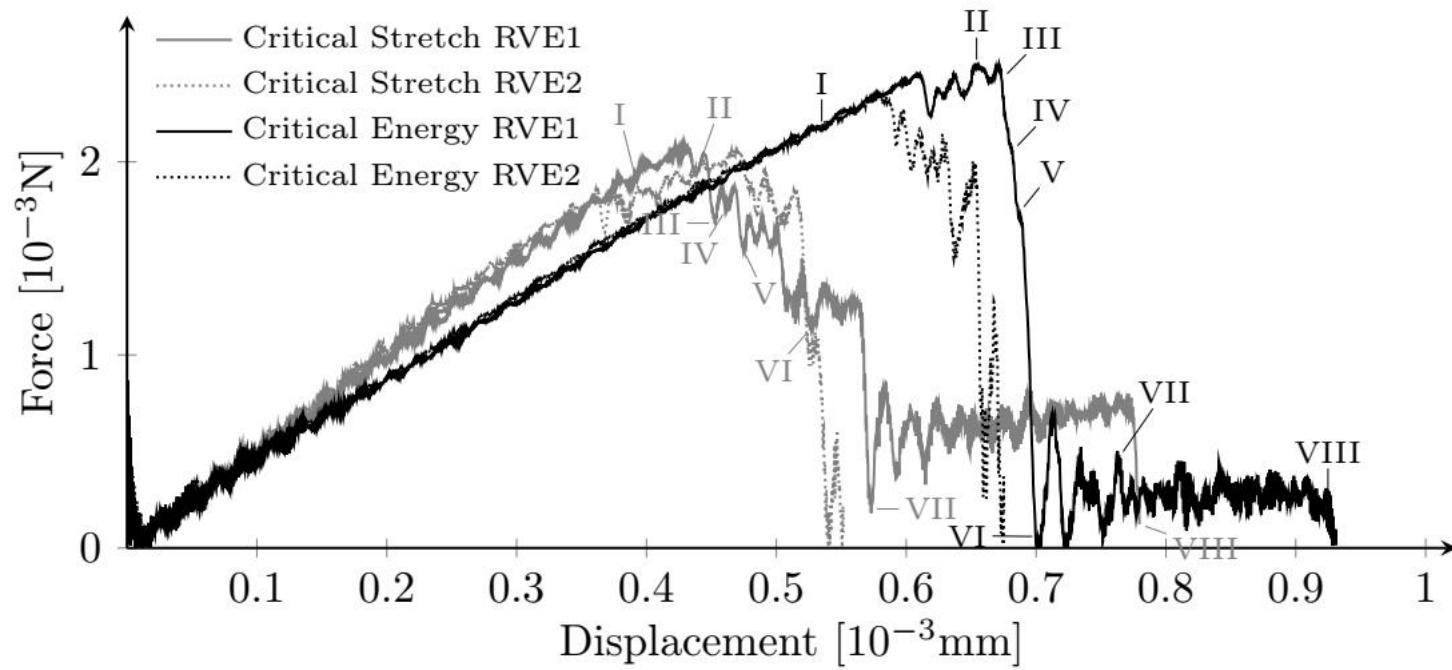


VI

Critical Energy

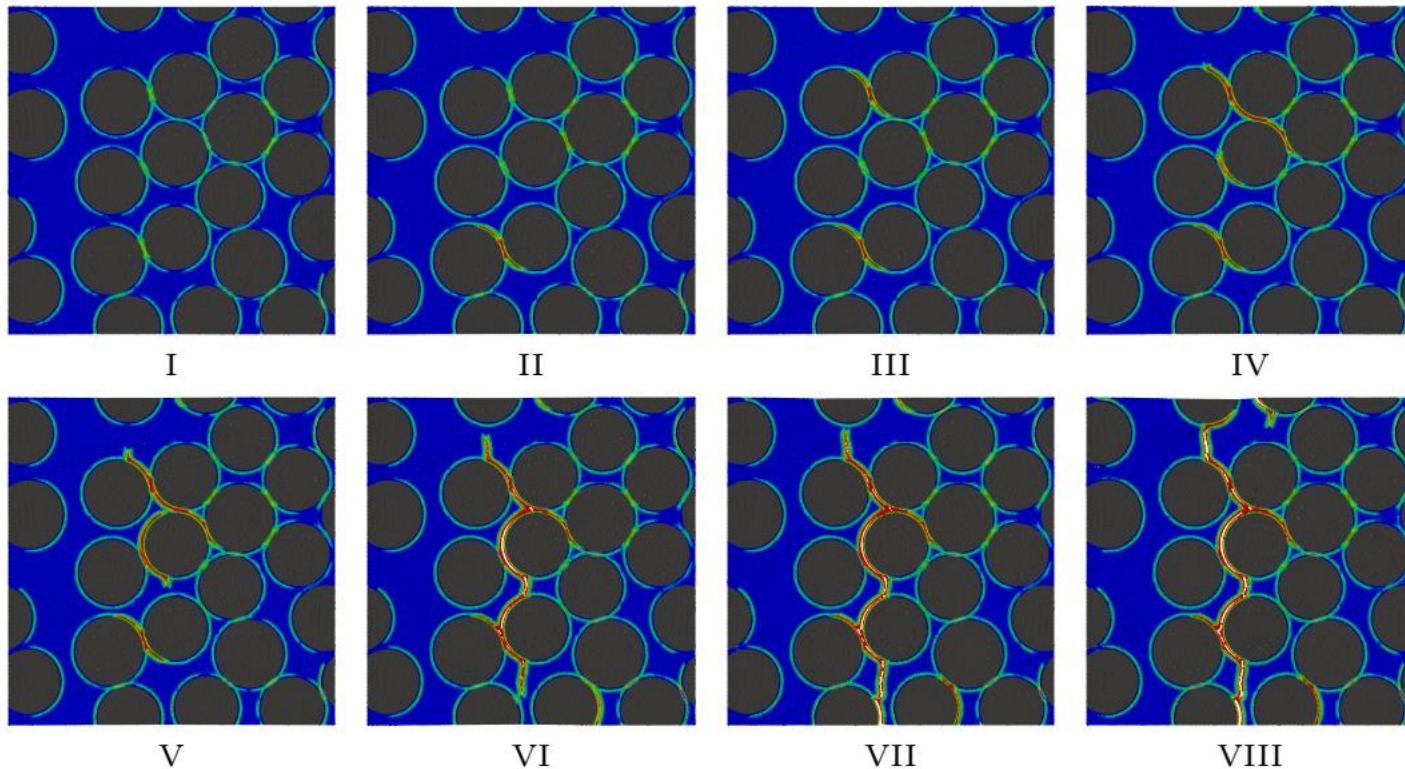
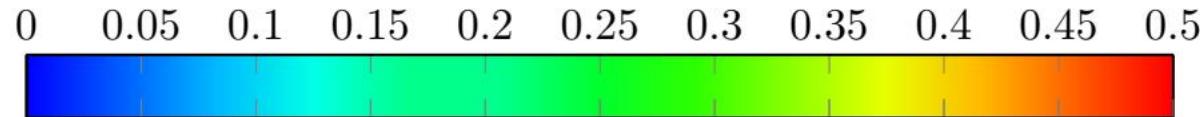


Comparison: RVE-Force-Displacement

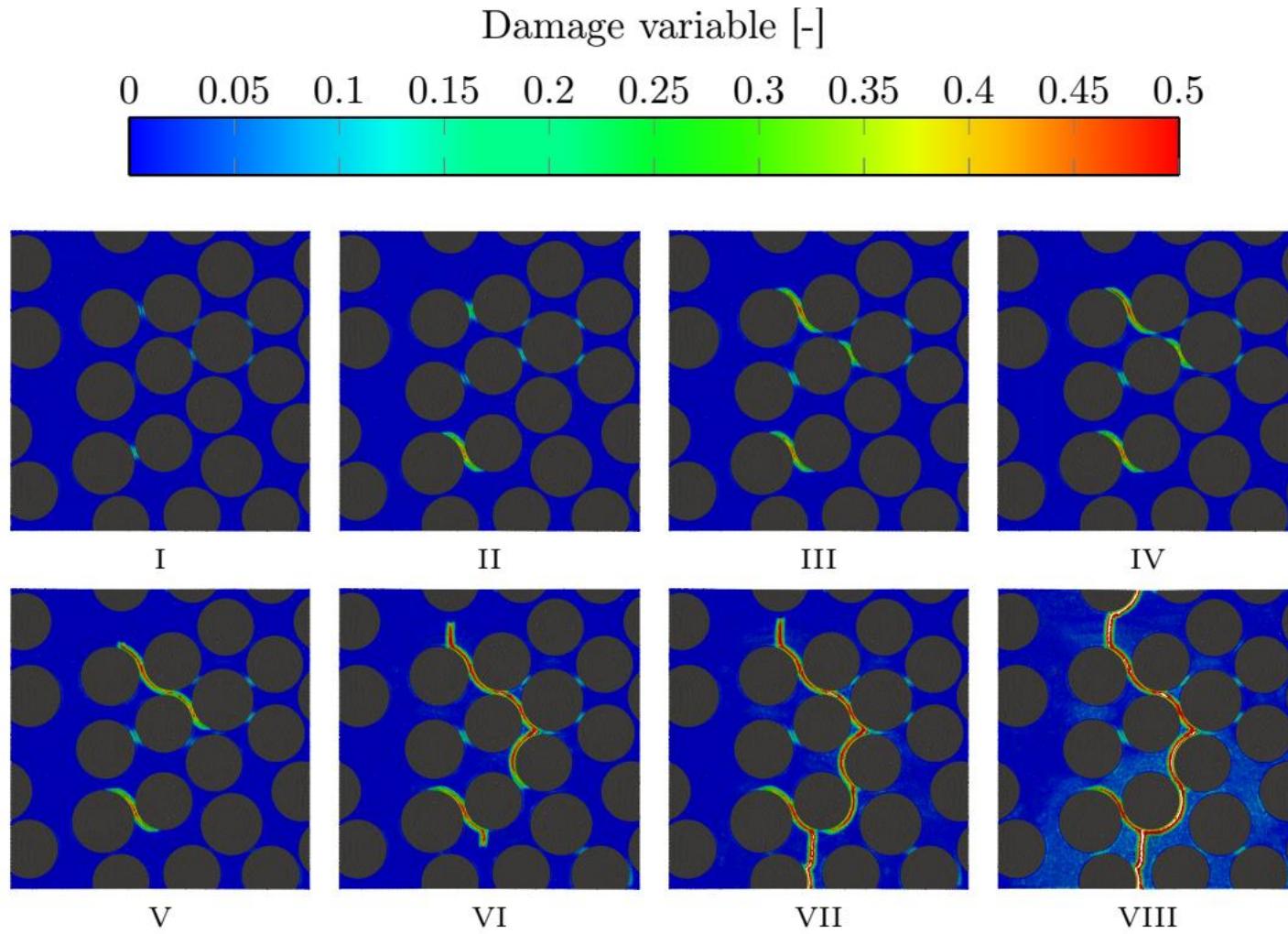


Comparison: RVE – Energy Criterion

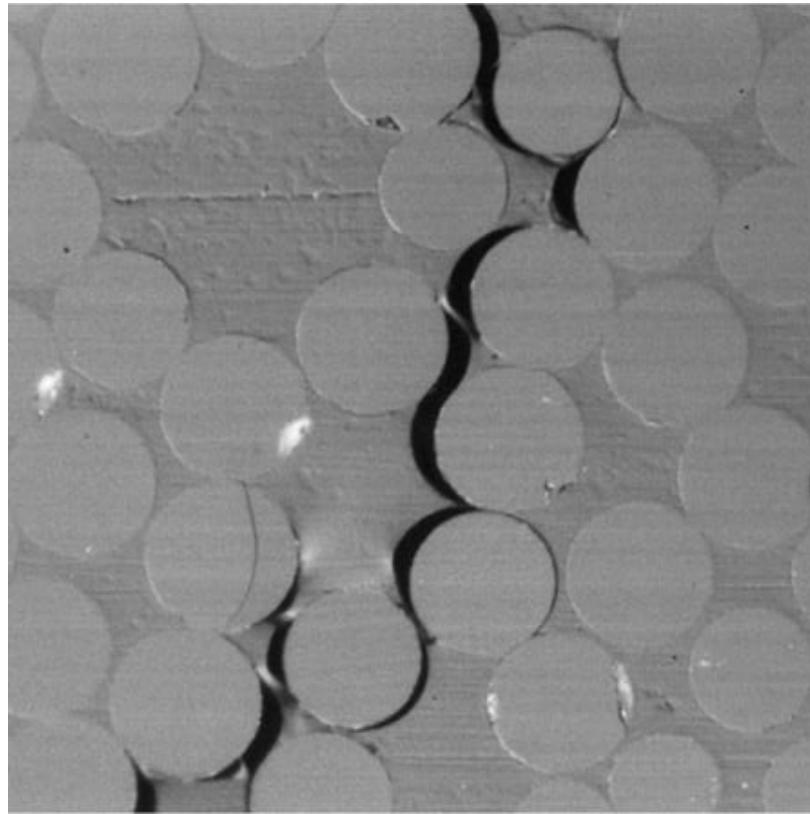
Damage variable [-]



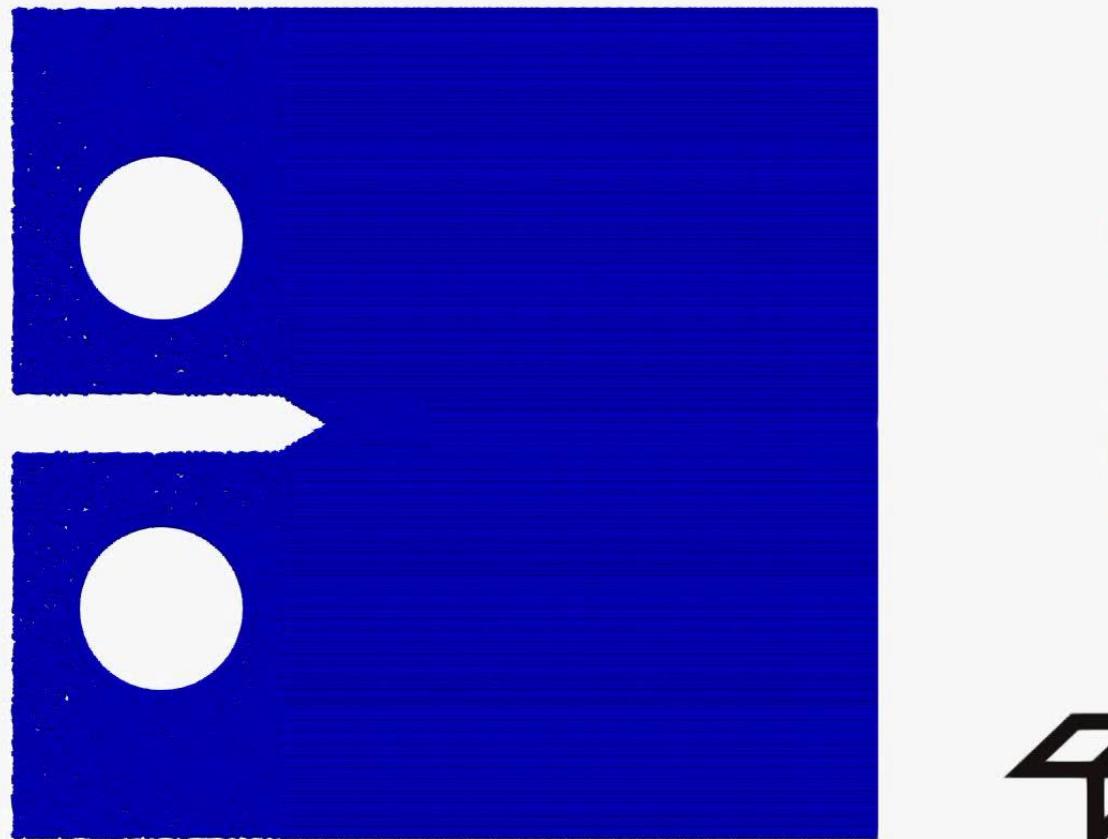
Comparison: RVE - Critical Stretch



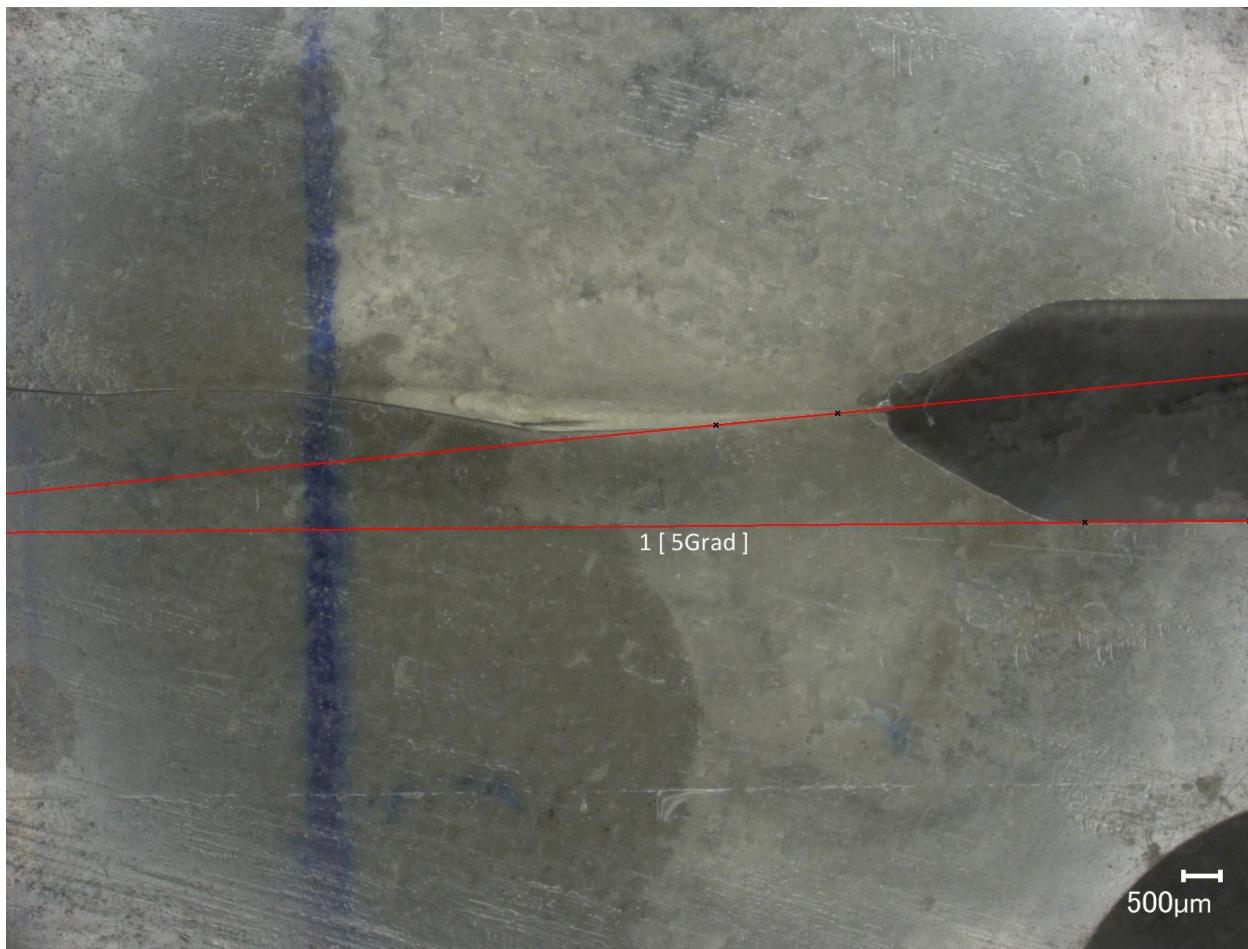
Experimental comparison



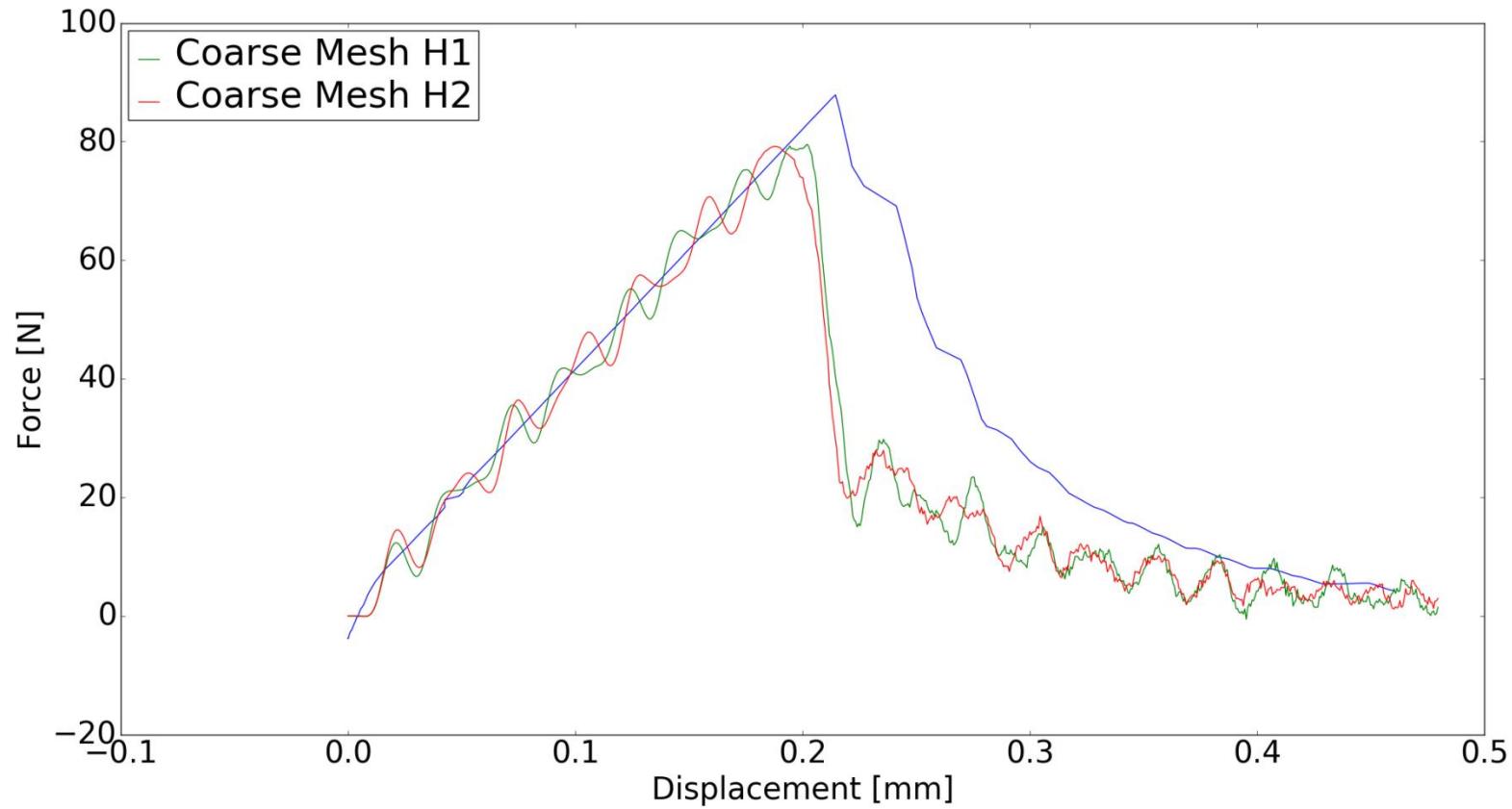
Validation



Validation



Validation



Conclusion

**Peridynamics is an interesting approach, but there is still a lot of work to do...
Especially in terms of validation...**



Thank you!

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All presented models and source code can be found here

Rädel, M. & Willberg, C. PeriDoX Repository
<https://github.com/PeriDoX/PeriDoX>

A partial view of a globe showing the Earth's surface with blue oceans and green continents. The text 'Knowledge for Tomorrow' is overlaid on the globe.

Knowledge for Tomorrow