
fmincon_handle

By 6110, main function to run fmincon

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function [sprMss,sprGeo,exp,sigmaVar_final,std_final,wts,pts,exitFlag,
    fmincOutput ] = fmincon_handle(wing)
% note return wts, pts for debugging
[pts, wts] = get_pts_wts(wing);

% define anon funcs for fmincon
obj_Fun      = @(desVar) CalcMass(desVar,wing);
nlcon       = @(desVar) StressConst(desVar,pts,wts,wing);

optns = optimoptions(@fmincon,...
    'Display','iter-detailed',...
    'Algorithm','sqp',...
    'GradObj','on',...
    'SpecifyObjectiveGradient',true,...
    'SpecifyConstraintGradient',true,...
    'OptimalityTolerance',1e-10,...
    'StepTolerance',1e-10,...
    'ConstraintTolerance',1e-10);
[ sprGeo,sprMss,exitFlag, fmincOutput ] =
    fmincon(obj_Fun,wing.x0,wing.A,wing.b,[],[],wing.lb,[],nlcon,optns);

[exp,sigmaVar_final,~,std_final,wts,pts] =
    handleMakePlots(sprGeo,pts,wts,wing);

function [pts, wts] = get_pts_wts(wing) %% This is important
% This is done at the beginning, as it takes a long time to
run
% DO not use it in fmincon!!
fnom0 = 2*wing.w/wing.L;
mu = zeros(1,4);%
sigma = fnom0*[1/10, 1/20, 1/30, 1/40];

[xi, w] =GaussHermite(wing.NcolPts); % get points and weights
pts = zeros((wing.NcolPts^4),4); % prealloc
wts = zeros((wing.NcolPts^4),1); % prealloc

i = 1;
for i1=1:wing.NcolPts
    xi1 = sqrt(2)*sigma(1)*xi(i1) + mu(1); % first layer
    for i2=1:wing.NcolPts
        xi2 = sqrt(2)*sigma(2)*xi(i2) + mu(2); % second layer
        for i3=1:wing.NcolPts
            xi3 = sqrt(2)*sigma(3)*xi(i3) + mu(3); % third
layer
            for i4=1:wing.NcolPts
                xi4 = sqrt(2)*sigma(4)*xi(i4) + mu(4); %
fourth layer
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        pts(i,:) = [xi1, xi2, xi3, xi4]; % get points
        wts(i) = w(i1)*w(i2)*w(i3)*w(i4); % get
weights
        i = i+1; % increment i
    end
end
end
end
end
fprintf('pts and wts created\n');
end

function [sigmaExp, sigmaVar] = CalcStress(pts,wts,r_out,Iyy,wing)
    sigmaExp = 0; % prealloc
    sigmaVar = 0; % prealloc
    for n=1:(wing.NcolPts^4)
        forceBeam = CalcForce(pts(n,:),wing); % get force on beam
        dispBeam =
CalcBeamDisplacement(wing.L,wing.E,Iyy,forceBeam,wing.Nx-1); % get
beam displacement
        sigmaBeam =
CalcBeamStress(wing.L,wing.E,r_out,dispBeam,wing.Nx-1); % get stress
on beam
        sigmaExp = sigmaExp + wts(n)*sigmaBeam; % get mean
        sigmaVar = sigmaVar + wts(n)*(sigmaBeam).^2;% get var
    end
end

end

function [mass,Jacob] = CalcMass(desVar,wing)

    mass = massFun(desVar,wing); % get mass
    Jacob = zeros(2*wing.Nx,1); % prealloc
    step = 1e-60; % step size

    for i=1:2*wing.Nx % complex step function
        ej = zeros(size(desVar));
        ej(i) = ej(i) + 1j*step;
        x_plus=desVar+ej;
        f_plus=massFun(x_plus,wing);
        Jacob(i) = imag(f_plus)/step;
    end

    function mass = massFun(desVar,wing) % mass function
        [r_in,r_out]= getRoutRinFunc(desVar); % get r_in, r_out
        secArea = r_out.^2 - r_in.^2; % get section area
        mass = trapz(secArea)*pi*wing.rho*wing.L/(wing.Nx-1); %
integrate volumed with trapz
    end

end

function [c,emptArr,Jacob,emptArr2] =
StressConst(desVar,pts,wts,wing) % function for stress constraing
    emptArr=[]; % because you cannot have [] in an output

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emptArr2=emptArr; % and you cannot repete an output
c = ineqFun(desVar,wing,pts,wts); % get ineq const
grad = zeros(wing.Nx,2*wing.Nx); % prealloc
step = 1e-60; % step size

for i=1:2*wing.Nx % stress constraing gradient
    ej = zeros(size(desVar));
    ej(i) = ej(i) + 1j*step;
    x_plus=desVar+ej;
    f_plus=ineqFun(x_plus,wing,pts,wts);
    grad(:,i) = (imag(f_plus)/step);

end
Jacob = grad'; % return gradient

function c = ineqFun(desVar,wing,pts,wts) % function for c
inequality
    [r_out, ~, Iyy] = get_RO_RI_Iyy(desVar); % get r_out, Iyy
    [sigmaExp, sigmaVar] =
CalcStress(pts,wts,r_out,Iyy,wing); % run calcStress
    sigma = sqrt(sigmaVar - sigmaExp.*sigmaExp); % get stdv
    c = (sigmaExp + 6*sigma)./wing.Y - 1; % compute c
end

end

function [force] = CalcForce(pts,wing) % compute force at points
delt = 0; % prealloc
x = linspace(0,wing.L,wing.Nx); % create linspace
forceInit = (2*wing.w/wing.L)*(1-x./wing.L); % inital force
for n=1:obj.NpertPts
    delt = delt + pts(n)*cos(((2*n-1).*pi.*x)/(2*wing.L)); %
from project discription
end
force = forceInit + delt; % return force
end

function [sigmaExp,sigmaVar,desVar,sigma,wts,pts] =
handleMakePlots(desVar,pts,wts,wing) % function gives everything
needed for plots
    [r_out, ~, Iyy] = get_RO_RI_Iyy(desVar); % get r_out, Iyy
    [sigmaExp,sigmaVar] = CalcStress(pts,wts,r_out,Iyy,wing);
    sigma = sqrt(sigmaVar - sigmaExp.*sigmaExp); % get stdv
end

function [r_in,r_out]= getRoutRinFunc(desVar) % function to get
r_out, r_in
    lnDV=length(desVar); % get length of designVariable
    r_in = desVar(1:lnDV/2); % get r_inner
    r_out = r_in+ desVar(lnDV/2+1:end); % get r_outer
end

function [r_out, r_in, Iyy] = get_RO_RI_Iyy(desVar) % function to
get r_out, r_in, and Iyy

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        [r_in,r_out]= getRoutRinFunc(desVar);
        Iyy = (pi/4).*(r_out.^4-r_in.^4);
    end

    function [x, w] = GaussHermite(n)
        % Function to determines the abscisas (x) and weights (w) for
        the
        % Gauss-Hermite quadrature of order n>1, on the interval [-
        INF, +INF].
        % works for n>=2
        % Credit to Geert Van Damme (geert@vandamme-iliano.be)
        % See referances section
        if n<2
            error('Warning Number of Collection points below 2');
        end

        i    = 1:n-1;
        a    = sqrt(i/2);
        CM   = diag(a,1) + diag(a,-1);

        % CM is such that det(xI-CM)=L_n(x), with L_n the Hermite
        polynomial
        % under consideration. Moreover, CM will be constructed in
        such a way
        % that it is symmetrical.
        [V, L] = eig(CM);
        [x, ind] = sort(diag(L));
        V       = V(:,ind)';
        w       = sqrt(pi) * V(:,1).^2;
        w=w./sqrt(pi); % adjust weight
    end

end

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