fmincon handle

By 6110, main function to run fmincon

```
function [sprMss,sprGeo,exp,sigmaVar_final,std_final,wts,pts,exitFlag,
 fmincOutput ] = fmincon handle(wing)
% note return wts, pts for debugging
[pts, wts] = get_pts_wts(wing);
% define anon funcs for fmincon
obj Fun = @(desVar) getMass(desVar, wing);
nlcon = @(desVar) StressConst(desVar,pts,wts,wing);
optns = optimoptions(@fmincon,...
    'Display', 'iter-detailed',...
    'Algorithm', 'sqp',...
    'GradObj', 'on',...
    'SpecifyObjectiveGradient', true, ...
    'SpecifyConstraintGradient', true, ...
    'OptimalityTolerance', 1e-10, ...
    'StepTolerance', 1e-10, ...
    'ConstraintTolerance', 1e-10,....%);
    'FiniteDifferenceType', 'central', ...
    'FiniteDifferenceStepSize', 5e-4,...
    'CheckGradient', false);
[ sprGeo,sprMss,exitFlag, fmincOutput ] =
 fmincon(obj_Fun,wing.x0,wing.A,wing.b,[],
[],wing.lb,wing.ub,nlcon,optns);
[exp,sigmaVar_final,~,std_final,wts,pts] =
 handleMakePlots(sprGeo,pts,wts,wing);
    function [pts, wts] = get_pts_wts(wing)
        % Returns Gauss quadrature points and weights. Note that it is
 done
        % before fmincon, not during to speed up run time.
        % Inputs:
        % wing - spar object that has Spar Length and number of
 collection points
        % Outputs:
        % pts - GQ points
        % wts - GQ weights at points
        fNom0 = 2*wing.w/wing.L;
        mu = zeros(1,4);%
        sigma(1:4) = fNom0.*1./(10*(1:4));
        [xi, w] =GaussHermite(wing.NcolPts); % get points and weights
        pts = zeros((wing.NcolPts^4),4); % prealloc
```

```
wts = zeros((wing.NcolPts^4),1); % prealloc
       i = 1;
       for i1=1:wing.NcolPts
           xi1 = sqrt(2)*sigma(1)*xi(i1) + mu(1); % first layer
           for i2=1:wing.NcolPts
               xi2 = sqrt(2)*sigma(2)*xi(i2) + mu(2); % second layer
               for i3=1:wing.NcolPts
                   xi3 = sqrt(2)*sigma(3)*xi(i3) + mu(3); % third
layer
                   for i4=1:wing.NcolPts
                       xi4 = sqrt(2)*sigma(4)*xi(i4) + mu(4);
fourth layer
                       pts(i,:) = [xi1, xi2, xi3, xi4]; % get points
                       wts(i) = w(i1)*w(i2)*w(i3)*w(i4); % qet
weights
                       i = i+1; % increment i
                   end
               end
           end
       end
       fprintf('pts and wts created\n');
   end
   function [sigmaExp, sigmaVar] = getStress(pts,wts,wing,desVar)
       % Caculates expected (mean) and variance of stress.
       % Inputs:
          wing - spar object that has values for Spar
       % pts - GQ points
         wts - GQ weights at points
          desVar - current design variable
       % Outputs:
           sigmaExp - Expectation of stress
           sigmaVar - Variance of stress
       [r_out, ~, Iyy] = get_RO_RI_Iyy(desVar); % get r_out, Iyy
       sigmaExp = 0; % prealloc
       sigmaVar = 0; % prealloc
       for n=1:(wing.NcolPts^4)
           forceBeam = getForce(pts(n,:),wing); % get force on bream
           dispBeam =
CalcBeamDisplacement(wing.L,wing.E,Iyy,forceBeam,wing.Nx-1); % get
beam displacment
           sigmaBeam =
CalcBeamStress(wing.L,wing.E,r out,dispBeam,wing.Nx-1); % get stress
on beam
           sigmaExp = sigmaExp + wts(n)*sigmaBeam; % get mean
           sigmaVar = sigmaVar + wts(n)*(sigmaBeam).^2;% get var
       end
   end
   function [mass,Jacob] = getMass(desVar,wing)
```

```
% Function to calculate mass of spar, objective function of
fmincon
       % Inputs:
       % wing - spar object that has values for Spar
          desVar - design variable
       % Outputs:
       % Jacob - grad for complex step
         mass - mass of spar
       mass = massFun(desVar, wing); % get mass from mass function
       Jacob = zeros(2*wing.Nx,1); % prealloc for speed
       for i=1:2*wing.Nx % complex step function
           ej = zeros(size(desVar));
           ej(i) = ej(i) + 1j*wing.step;
           x_plus=desVar+ej;
           f plus=massFun(x plus,wing);
           Jacob(i) = imag(f_plus)/wing.step;
       end
       function mass = massFun(desVar, wing) % mass function
           % Nested Function for spar mass, used in complex step
           % Inputs:
           % desVar - design variable
              wing - spar object that has values for Spar
           % Outputs:
             mass - mass of spar
           [r_in,r_out] = getRoutRinFunc(desVar); % get r_in, r_out
           secArea = r_out.^2 - r_in.^2; % get section area
          mass = trapz(secArea)*pi*wing.rho*wing.L/(wing.Nx-1); %
integrate volumed with trapz
       end
   end
   function [c,emptArr,Jacob,emptArr2] =
StressConst(desVar,pts,wts,wing)
       % Function for stress inequality constraint, uses comples step
with gradient
       % Inputs:
       % desVar - design variable
          wing - spar object that has values for Spar
         pts - GQ points
       % wts - GQ weights at points
       % Outputs:
         c - non linear constraint for stress
       % emptArr - empty array to return for fmincon formating
       % Jacob - Gradiant of constraint
           emptArr - empty array to return for fmincon formating
```

```
emptArr=[]; % because you cannot have [] in an output
       emptArr2=emptArr; % and you cannot repete an output
       c = ineqFun(desVar, wing, pts, wts); % get ineq const
       grad = zeros(wing.Nx,2*wing.Nx); % prealloc
       for i=1:2*wing.Nx % stress constraing gradiant
           ej = zeros(size(desVar));
           ej(i) = ej(i) + 1j*wing.step;
           x_plus=desVar+ej;
           f_plus=ineqFun(x_plus,wing,pts,wts);
           grad(:,i) = (imag(f plus)/wing.step);
       end
       Jacob = grad'; % return gradiant
       function c = ineqFun(desVar, wing, pts, wts)
           % Nested Function for inequality constraint, used in
complex step
           % Inputs:
           % desVar - design variable
           % wing - spar object that has values for Spar
           % pts - GQ points
           % wts - GQ weights at points
           % Outputs:
              c - non linear constraint for stress
           [sigmaExp, sigmaVar] = getStress(pts,wts,wing,desVar); %
run calcStress
           sigma = getStandardDev(sigmaExp,sigmaVar); % get standard
deveiation
          c = (sigmaExp + 6*sigma)./wing.Y - 1; % compute c
       end
   end
   function standardDev = getStandardDev(sigmaExp,sigmaVar)
       % Function to compute get standard deviation
       % Inputs:
         sigmaExp - expected value
         sigmaVar - variance
       % Outputs:
         standardDev - standard deviation
       standardDev = sqrt(sigmaVar - sigmaExp.*sigmaExp); % get stdv
   end
   function [force] = getForce(pts,wing)
       % Function to compute force at points
       % Inputs:
```

```
% wing - spar object that has values for Spar
       % pts - GQ points
       % Outputs:
         force - force at points on spar
       delta_F = 0; % prealloc
       x = linspace(0,wing.L,wing.Nx); % create linspace
       forceNomX = (2*wing.w/wing.L)*(1-x./wing.L); % nominal force
       for n=1:wing.NpertPts % for this code it is 4, but for other
problems you might want more than 4 pts
          delta_F = delta_F + pts(n)*cos(((2*n-1).*pi.*x)/
(2*wing.L)); % from project discription
       end
       force = forceNomX + delta F; % return force
   end
   function [sigmaExp,sigmaVar,desVar,sigma,wts,pts] =
handleMakePlots(desVar,pts,wts,wing) % function gives everything
needed for plots
       % Function to get variables used in plots
       % Inputs:
       % desVar - design variable
         wing - spar object that has values for Spar
         pts - GQ points
       % wts - GQ weights at points
       % Outputs:
         sigmaExp - expected value of stress
       % sigmaVar - variance of stress
       % sigma- standard deviation of stress
       % desVar - design variable
         pts - GQ points
       % wts - GQ weights at points
 %_____
       [sigmaExp,sigmaVar] = getStress(pts,wts,wing,desVar); % get
expected value and variance
       sigma = getStandardDev(sigmaExp,sigmaVar); % get standard
deviation
   end
   function [r_in,r_out] = getRoutRinFunc(desVar)
       % Function to get r_out, r_in from design variable
       % Inputs:
       % desVar - design variable
       % Outputs:
       % r_in - inner radius
       % r out - outer radius
       lnDV=length(desVar); % get length of designVariable
       r_in = desVar(1:lnDV/2); % get r_inner
       r_out = r_in+ desVar(lnDV/2+1:end); % get r_outer
   end
```

```
function [r out, r in, Iyy] = get RO RI Iyy(desVar)
       % function to get r_out, r_in, and Iyy
       % Inputs:
       응
          desVar - design variable
       % Outputs:
       % r_out - outer radius
       % r in - inner radius
       % Iyy - Second moment of inertia
  [r_in,r_out] = getRoutRinFunc(desVar); % run function to get
r out, r in
       Iyy = (pi/4).*(r_out.^4-r_in.^4); % calculate Iyy
   end
   function [x, w] = GaussHermite(n)
       % Function to determines the abscisas (x) and weights (w) for
the
       % Gauss-Hermite quadrature of order n>1, on the interval [-
INF, +INF]. Credit to Geert Van Damme (geert@vandamme-iliano.be),
please see referances section
       % Inputs:
       % n - Gauss-Hermite quadrature of order, must be >1
       % Outputs:
       % x - location of GQ points
       % w - weight of GQ points
       i = 1:n-1;
       sqrtNodesHalf = sqrt(i/2); % (nuber of nodes /2)^(1/2)
       CMat = diag(sqrtNodesHalf,1) + diag(sqrtNodesHalf,-1); % CMat
created so that \det(xI-CM)=L_n(x), with L_n the Hermite polynomial
       % Also, CM will be symmetrical.
       [V, L] = eig(CMat); % get L, a diagonal matrix
       [~, indx] = sort(diag(L)); % sort and get indexes
              = V(:,indx)'; w = sqrt(pi) * V(:,1).^2; % get weights
       w=w./sqrt(pi); % adjust weight
   end
```

end

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