

# **Multidimensional Modelling of Cross-Beam Energy Transfer for Direct-Drive Inertial Confinement Fusion**

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# List of Acronyms

**Rad-MHD** Radiative-Magnetohydrodynamics

**MHD** Magnetohydrodynamics

**Rad-Hydro** Radiative-Hydrodynamics

**HEDP** High Energy Density Physics

**LPIs** Laser-Plasma Instabilities

**CBET** Cross-Beam Energy Transfer

**ICF** Inertial Confinement Fusion

**PiC** Particle in Cell

**GO** Geometric Optics

**SBS** Stimulated Brillouin Scattering

**SRS** Stimulated Raman Scattering

**TPD** Two Plasmon Decay

**EPW** Electron Plasma Wave

**PCGO** Paraxial Complex Geometric Optics

**IRT** Inverse Ray Tracing

**FRT** Forward Ray Tracing

**LLE** Laboratory for Laser Energetics

**LLNL** Lawrence Livermore National Laboratory

**CEA** Commissariat à l'Énergie Atomique et aux Energies Alternatives

**IAW** Ion Acoustic Wave

**EPW** Electron Plasma Wave

**OpenMP** Open Multi-Processing

**MPI** Message Passing Interface

**Inv-Brem** Inverse-Bremsstrahlung

**EI** Etalon Integral

**FL** Field Limiter

**CGT** Caustic Gain Truncation

**DPRs** Distributed Polarization Rotators

**RTI** Rayleigh–Taylor Instability

**IFAR** In-Flight Aspect Ratio

**ADI** Alternating-Direction Implicit

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**DFT** Discrete Fourier Transform

**NIF** National Ignition Facility

**MIT** Massachusetts Institute of Technology

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# **1 SOLAS, a 3-D Laser Ray-Trace and Cross-Beam Energy Transfer Model**

This chapter describes the SOLAS code, a 3-D laser ray-tracing module implemented in the Radiative-Magnetohydrodynamics (Rad-MHD) code CHIMERA. The chapter begins with a discussion of why ray-tracing is frequently used to model ‘long-pulse’ lasers for High Energy Density Physics (HEDP) experiments and why the standard framework is inadequate to model Laser-Plasma Instabilities (LPIs). It then goes on to describe the ray-trajectory solver, electric-field reconstruction and CBET components of the model in detail, including discussion of the validity of the model components. The numerical methods are also presented alongside an extensive set of validation problems to verify the implementation.

## 1.1 Ray-Tracing for Hydrodynamic Simulations of Fusion Plasmas

Nanosecond length ('long-pulse') lasers are often used as an external energy source in the field of HEDP, for example in laboratory-astrophysics experiments [9–11], equation of state studies [12, 13] or Inertial Confinement Fusion (ICF) implosions [14–16]. Experimental design and analysis for these experiments must often be supported with fluid Rad-MHD simulations. The laser must therefore be described in these codes by a theoretical framework that is both valid and computationally tractable. The physical processes by which lasers interact with plasma are a result of microscopic couplings between the laser field and particles or plasma waves. The detailed microphysics of these interactions are often studied using Particle in Cell (PiC) codes [17] or wave-based solvers [18], typically for scales and durations of tens of micrometres and hundreds of picoseconds. Coupling these tools directly to multidimensional Radiative-Hydrodynamics (Rad-Hydro) simulations, which are often used for millimetre and nanosecond scales, is usually computationally intractable.

The Geometric Optics (GO) assumptions are often applicable for HEDP laser-plasma interactions and therefore ray-tracing offers a computationally tractable approach to modelling lasers in these experiments<sup>1</sup>. By assuming static Rad-Hydro profiles (which is normally valid for the propagation time of light through the computational domain) a laser beam can be discretized into a bundle of rays. The ray equations can then be integrated along their path to solve for the trajectory of the light, assuming that refraction dominates over diffraction. A discrete amount of power can also be given to each ray. If there is a suitable model for the power-absorption rate, this can also be integrated along the ray to provide an energy source for the plasma. In many laser-driven HEDP experiments, frequency-doubled or -tripled lasers and long scale-length plasmas mean that Inv-Brem is the dominant deposition process [19, 20]. There are well established models for Inv-Brem that are suitable for implementation in ray-tracing codes, because they only require knowledge of the local plasma conditions, which are easily accessible via interpolation from the Rad-Hydro grid to ray locations [21, 22]. The combination of the ray equations and Inv-Brem deposition is therefore the basis for the vast majority of laser-modules coupled to Rad-Hydro codes.

In laser-driven ICF experiments however, another class of interaction, LPIs, are vitally important to the energetics of the implosion. For example, CBET leads to a zeroth-order correction to the energy deposited in direct-drive experiments at the OMEGA laser facility, reducing coupled power late in the implosion to  $\sim 50\%$  [23]. LPIs cannot be included in the simple framework described above for two reasons. Firstly, they are non-linear<sup>2</sup> and secondly, reduced theoretical models of the interaction rely on knowledge of the electric field or intensity of the light [24]. Implementation in a ray-tracing code therefore necessitates a method by which the separate light waves can 'talk' to each other. For example, the CBET code described in this chapter stores information for separate components of laser beams on a common grid, which can then interact via the 'pump-depletion iterations', described in Sec. 1.5.5. Additionally, calculating the electric field or intensity is problematic because this

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<sup>1</sup>Validity is discussed in detail for ICF plasmas in Sec. ??

<sup>2</sup>Non-linear in this context means that the interaction involves multiple light and plasma waves.

is not an attribute which can be obtained from a GO ray. Heuristically, rays have an associated power, so an area is required to obtain an intensity. The evolution of a portion of the beam front's area is governed by a first order expansion of the Helmholtz equation, rather than the zeroth order expansion, which is most commonly used in ray-tracing packages for Rad-Hydro codes [25]. The first order expansion introduces an equation for the ray amplitude, which can be solved in a variety of ways and used to obtain the electric field of the light. Sec. 1.4 describes the approach taken to solving for the amplitude of the rays in SOLAS, which is to track the area of a triangle<sup>3</sup> of rays around a standard GO ray.

For direct-drive ICF simulations, it is also desirable to have a 3-D ray-trace, where rays travel and refract in three dimensions. In some computational direct-drive studies, particularly in 1-D Rad-Hydro simulations, simplified laser models are employed in which rays travel radially toward the target [26]. This simplification can lead to significant deviations from reality, as it neglects any refractive losses, which become increasingly significant late in the implosion as the target converges. Assuming that rays travel radially inward, also leads to deposition occurring closer to the critical surface compared to a true 3-D ray-trace, resulting in an overestimation of the drive. The growth of LPIs also depends on vector summations of light and plasma wavevectors, so a 3-D ray-trace is necessary when modelling these effects. Predictive direct-drive simulations therefore necessitate a fully 3-D ray-trace, even when coupled to just 1-D hydrodynamics.

## 1.2 Existing Cross-Beam Energy Transfer Models

There is a variety of existing computational tools used to model CBET for ICF conditions. These include codes that are used to study the interaction from first principles, reduced models to investigate the effect of CBET when coupled to hydrodynamics and validation tools to test the implementation of these reduced models. A non-exhaustive list of existing codes is listed below, provided as context for the work presented later in this chapter.

### 1.2.1 Ray-Based Models

The most common tool to study the coupling of CBET effects into hydrodynamics are reduced ray-based models, which use the linear gain theory of Stimulated Brillouin Scattering (SBS), described in more detail in Sec. 1.5. A variety of different codes used to study this. The main difference between the models is the way that the electric field or intensity of the laser light is obtained.

**Inverse Ray Tracing (IRT)** This is an approach implemented in the IFRIT code, that creates a mapping between points on an initial beam front and arbitrary locations within the plasma [27, 28]. This is in contrast to Forward Ray Tracing (FRT), used in the other ray-based approaches listed below, where discrete points on the beam front are integrated forward to discrete points within the plasma. IRT is an efficient approach for convex, approximately

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<sup>3</sup>For a 2-D ray-trace, a co-planar, pair of rays is used rather than a triangle.

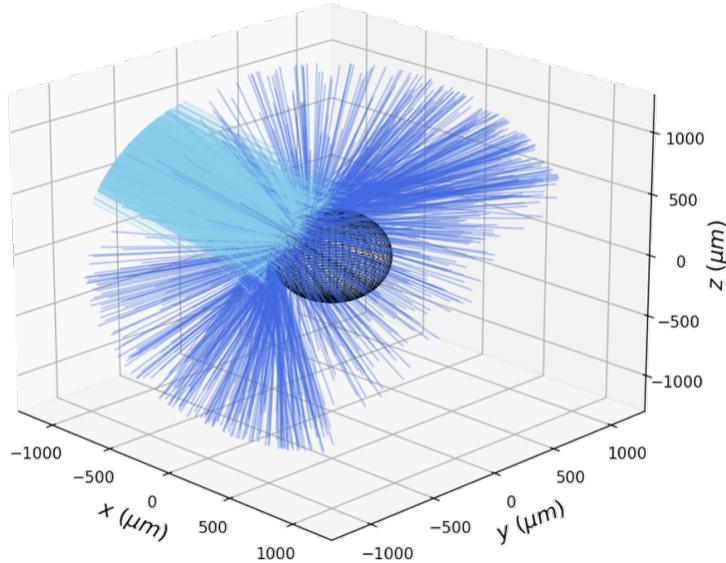


Figure 1.1: 3-D ray trajectories through a spherically symmetric, OMEGA direct-drive scale density profile. The incident rays are plotted in cyan and the reflected rays, which spread out over a large solid angle, are plotted in dark blue. The critical density is represented by the black mesh.

spherically symmetric plasma profiles, but cannot deal well with beams that have multiple caustics<sup>4</sup>, where the FRT approach is better suited. IFRIIT has been coupled to the 3-D Rad-Hydro code, ASTER from Laboratory for Laser Energetics (LLE), and prior to the development of CHIMERA-SOLAS, ASTER-IFRIIT was the only code combination capable of conducting 3-D direct-drive simulations with in-line CBET [23].

**Ray Statistics Approach** The LASNEX [29], TROLL [30] and DRACO codes [31], developed at Lawrence Livermore National Laboratory (LLNL), Commissariat à l’Énergie Atomique et aux Energies Alternatives (CEA) and LLE respectively, implement field-reconstruction methods, which depend heavily on having many rays per computational cell. LASNEX and TROLL, used mainly for indirect-drive hohlraum simulations, obtain the intensity of light by first propagating rays through the mesh and obtaining the deposited power in each grid cell. The intensity is then obtained from the power in each cell using electromagnetic energy conservation. Obtaining the intensity from deposition means that the field cannot be reconstructed in vacuum regions where no deposition occurs.

DRACO uses a similar approach, but the intensity is estimated by multiplying the ray power by the path length in a cell and dividing by the cell volume, which gives a dimensional estimate for the intensity. Both of these approaches require many rays-per-cell to accurately obtain the intensity,  $\mathcal{O}(100)$  [32]. For direct-drive simulations this is extremely computationally expensive, because backscatter CBET dominates over sidescatter. Therefore, the reflected field of each beam must be resolved, and each beam spreads out over a large solid angle after reflecting off a convex density profile, as is demonstrated in Fig. 1.1. This means that

<sup>4</sup>Caustics are defined and discussed in detail in Sec. 1.4.1.

orders of magnitude more rays are required, compared to approaches which can accurately obtain the field from a single ray per cell.

Another drawback of both of these approaches is that caustics of beams (regions where the amplitude of rays diverge) are not identifiable, and therefore fields cannot be capped to physically accurate, diffraction-limited values. If a significant amount of CBET occurs at caustics, such as in direct-drive ICF, then erroneous global multipliers to CBET gains must be applied which are effectively free parameters that must be tuned to obtain a pre-known reduction in absorbed energy. It is therefore difficult to trust this approach for predictive, pre-shot direct-drive simulations.

**Paraxial Complex Geometric Optics (PCGO)** In this approach, each ray has an associated Gaussian intensity profile, the width of which is integrated along the ray trajectory [25, 33]. A single ray can be used per cell because each ray therefore has an intensity which is interpolated to the mesh. However, the reconstructed field near caustics is not accurate and the width evolution is only valid for a short propagation distance. This approach was coupled to the CHIC 2-D Rad-Hydro code, but the approach is difficult to extend to 3-D as the implementation relied on interacting only rays whose centroids crossed, which does not occur in 3-D for non-coplanar rays [34].

**Neighbouring Rays** The BEAMCROSSER code obtains an area for each ray by co-tracing a triangle of neighbouring rays around it that can be converted into a ray amplitude and therefore electric field from electromagnetic energy conservation [35]. Integrating the amplitude along the ray trajectory means that the caustics can be identified and therefore fields in those regions capped to diffraction limited values [36]. Each ray also has an individual field value, and it is therefore less dependent on ray-per-cell statistics than geometric models, such as that used in the DRACO code described above [37]. Sec. 1.4 describes the implementation of this approach into the CHIMERA 3-D Rad-MHD code. A similar approach, simplified by assuming uniform illumination of a direct-drive target, has been implemented into the 1-D LILAC Rad-Hydro code [38, 39]. It had previously however, not been coupled to a multidimensional Rad-Hydro code.

### 1.2.2 Wave Solvers

Solving Maxwell's equations in a plasma background is a useful tool for the study of LPIs. The main code used in the ICF community that uses this approach is LPSE, which propagates light waves through a prescribed, spatially varying density, temperature and velocity profile in 1-D→3-D [18, 40]. It then solves the nonlinear coupling of electromagnetic waves by allowing first order plasma perturbations (obtained from the ponderomotive beat pattern between light waves), which then feed back into the wave propagation. The perturbative approach limits LPSE to linear problems and the temporal and spatial resolution required to resolve the beat frequency mean that coupling to multidimensional Rad-Hydro simulations is not feasible. However, it is an extremely useful tool to validate implementation of CBET

models, especially for laser caustics where diffraction is important, such as the test case presented in Sec. 1.5.7.2. It can also been used for many other important studies, such as the mitigation of LPIs via laser bandwidth and the effect of beam smoothing techniques on the growth rate of LPIs [41].

### 1.2.3 Particle in Cell Codes

Both ray-based codes and LPSE are ill-suited to the study of LPIs in the non-linear regime, where the laser intensity becomes sufficiently large that the ponderomotive imprint on the plasma can no longer be treated perturbatively. Understanding the growth and saturation of LPIs in this regime is particularly important for ICF schemes with high peak intensities, such as during the ignitor spike in shock ignition pulses [42]. Often kinetic effects such as ion-trapping become important in non-linear saturation, where ions are trapped and then accelerated in the CBET induced Ion Acoustic Wave (IAW), leading to changes in the IAW phase velocity and therefore a loss of resonance [17]. PiC codes operate by using a finite number of macro-particles to represent the distribution function of the plasma, coupled to an electromagnetic solver, discretised on a mesh. They are therefore well suited to model this kinetic saturation, albeit over short timescales and in simulations that are not truly representative of direct-drive conditions, due to computational expense [43]. Kinetic modelling of CBET has demonstrated that the growth of LPIs can be a much more complex, time-dependent problem than is assumed by the linear-models used in ray-based codes, leading to larger net energy transfers [44].

## 1.3 SOLAS 3-D Laser Ray-Trace

This section describes how the ray equations, derived from the Helmholtz equation in Sec. ?? are solved in the SOLAS module for the CHIMERA Rad-MHD code. Details of the mesh used for spherical simulations, load balancing options as well as validation problems are provided.

A short summary of the computational derived-types is presented here, to aid understanding of the following sections. In SOLAS, a domain balanced mesh is created from the hydrodynamic grid in CHIMERA. The mesh is made up of *laser-cells*, which hold information about the geometry, hydrodynamic-quantities, laser-quantities<sup>5</sup> and neighbouring cells on the SOLAS grid. *Beam* objects are created which hold user-defined geometric information about the beam-port and pointing location and also store a list of rays from the beam. Each ray is defined by a position ( $\mathbf{x}$ ), wavevector ( $\mathbf{k}$ ), phase ( $\varphi$ ), angular frequency ( $\omega$ ) and power ( $P$ ). If field reconstruction is performed, then each ray also stores up to three additional *neighbour-rays*, from which the electric field of the ray can be reconstructed, as described in Sec. 1.4.

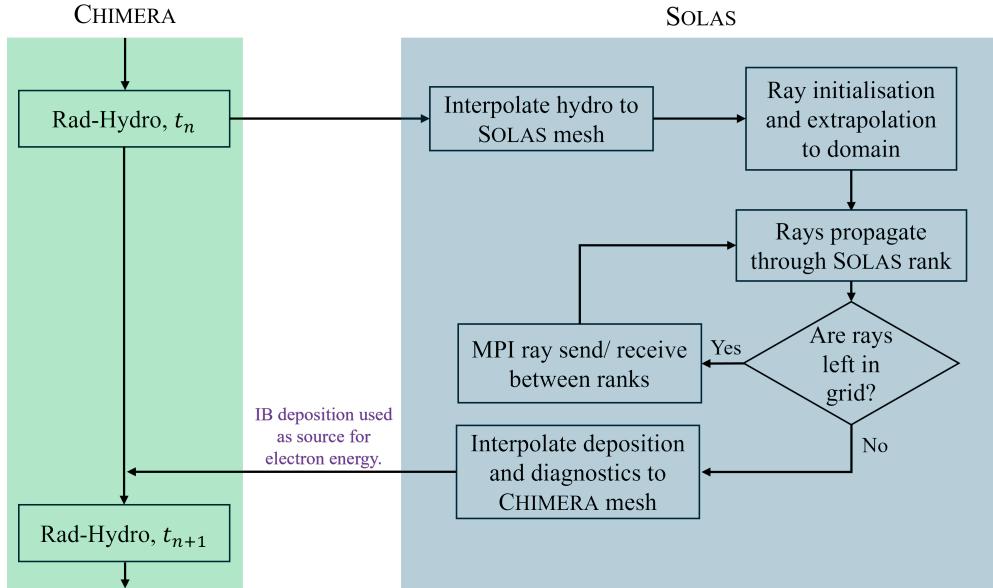


Figure 1.2: A flowchart illustrating the basic interfacing between CHIMERA and SOLAS. The interpolation steps refer simultaneously to the re-domain balancing and cell combination procedures described in Sec. 1.3.2.

### 1.3.1 Outline of SOLAS-CHIMERA Interfacing

A simple flowchart for the basic SOLAS operational loop and the interfacing with CHIMERA is shown in Fig. 1.2. At the start of a Rad-Hydro timestep,  $t_n$ , the hydrodynamic variables (stored on the Eulerian CHIMERA grid), which are necessary to compute ray trajectories and CBET gains, are interpolated to the SOLAS mesh. Rays are then initialised on the beam port locations and extrapolated to the computational domain, before being traced through the grid. SOLAS uses an Message Passing Interface (MPI) domain balanced approach to parallelisation, the reasons for which are discussed in more detail in Sec. 1.3.2, therefore rays are traced up to the internal borders of the MPI rank subdomain. Rays are then passed between ranks and moved through their new subdomain until they exit the entire simulation domain. As rays move through the domain, they lose energy via Inv-Brem, contributing to a running deposition total in each grid cell. When all rays have exited the domain, the SOLAS quantities, such as deposited energy which have been built up from the ray-trace, are then interpolated onto the CHIMERA grid, where the Inv-Brem deposition is used as a source term for the electron energy in the subsequent Rad-Hydro step,  $t_{n+1}$ . The interpolation, initialisation and propagation steps are outlined in sections 1.3.2, 1.3.3, 1.3.4 respectively.

### 1.3.2 SOLAS Mesh Structure

For direct-drive ICF simulations, which model the laser using ray-tracing (especially those which include CBET), the choice of an appropriate computational mesh on which to conduct the laser ray-trace is crucial to obtaining an accurate and noise-free energy deposition.

<sup>5</sup>Laser quantities include, for example, Inv-Brem deposition and electric field strength.

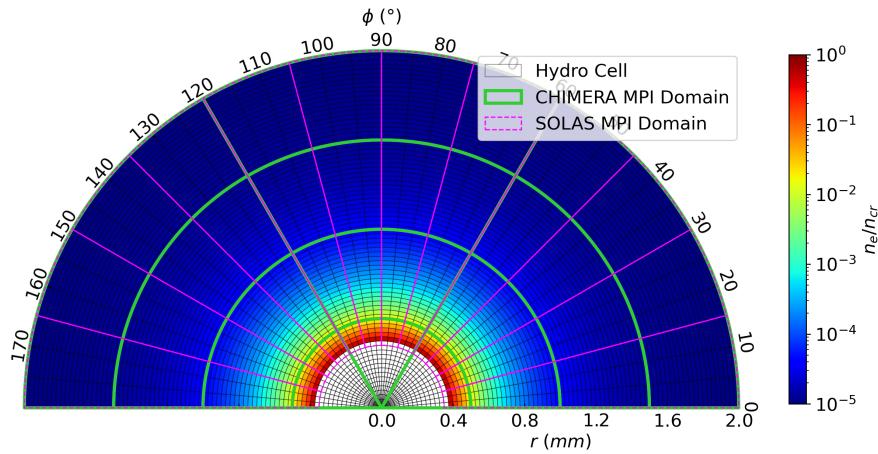


Figure 1.3: An illustrative diagram demonstrating the MPI re-domain balanced grid employed by SOLAS compared to the CHIMERA domain balanced grid for a cylindrical geometry.

A well-chosen grid can also significantly reduce the run-time and memory requirements of the calculation. The mesh used for hydrodynamic simulations of these experiments is often not well suited for ray-tracing calculations. For example, hydrodynamic grids often have excessive resolution in the corona to resolve the gradients in the laser quantities, which greatly increases the expense of tracing rays through the mesh, without significantly enhancing the accuracy of the solution. This section describes the approach taken in the SOLAS code to create a grid structure that accurately resolves laser energy deposition, both with and without CBET, in a memory-efficient manner.

### 1.3.2.1 Re-Domain Balance for Radial Geometries

Rad-MHD codes including CHIMERA, often employ a domain balanced approach to parallelisation, where each computational rank solves a portion of the entire spatial domain, for each discrete timestep. Additional ‘ghost-cells’ are stored in the subdomains and used to calculate gradients on the boundaries between ranks, which are updated with inter-rank MPI communications each timestep. The hydrodynamic grid for CHIMERA is Eulerian, with options for Cartesian, cylindrical or spherical-polar grids.

The optimal domain balancing minimises the number of ghost cells at subdomain boundaries, which leads to cubic<sup>6</sup> subdomains. If computing a laser ray-trace through this domain for a typical direct-drive calculation however, initially rays travel approximately radially and therefore regularly encounter processor boundaries where they must be transferred between ranks. A more optimal domain decomposition for the ray-trace minimises radial splitting to avoid excessive passing of rays. For spherical and cylindrical simulations, SOLAS therefore takes the hydrodynamic variables on the CHIMERA grid and re-domain balances the grid

<sup>6</sup>Cubic in cell dimension, not necessarily in physical space for non-Cartesian geometries.

for the ray-trace, such that the splitting does not occur in the radial coordinate. An example of this re-domain balancing is shown in Fig. 1.3 for an illustrative cylindrical mesh. The CHIMERA domain decomposition divides the radial and azimuthal extent into four and three ranks respectively, whereas SOLAS' division is purely in the azimuthal direction. For spherical and cylindrical direct-drive simulations where there is a defined, global minimum critical radius, SOLAS' mesh excludes the grid cells below this minimum radius beyond which the rays cannot reach in order to reduce the memory burden of the re-gridding.

An alternative approach to MPI-based domain-decomposition parallelisation of the module is to use the Open Multi-Processing (OpenMP) package, for which ranks share memory across a computational node. In this approach, the entire laser grid is stored once on the shared memory of the node and separate ranks trace rays through the entire domain without the need for transfers. While this is a preferable approach for a standard ray-trace to the MPI procedure described above, including a model for CBET is more challenging. This is because CBET requires communication between beams and therefore large amounts of information must be stored on the grid, which can lead to large memory overheads. Therefore, using multiple computing nodes is often a necessity for 3-D CBET calculations for which OpenMP-MPI hybrid approaches are required. This was deemed too significant an undertaking for the scope of the work presented in this thesis.

When a 3-D ray-trace is coupled to a lower dimensional hydrodynamics simulation (for example, a 1-D spherical direct-drive simulation) then a sparse 3-D mesh at a user-specified resolution is created for the ray-trace. This 3-D mesh allows the re-domain balancing routine to be employed to load balance the simulation. For CBET simulations, it also allows the field, which is required to compute the power change of rays due to CBET, to be stored as a function of angle when discretised on this sparse 3-D mesh. Storing the field discretised in angle enables the interaction between beams around the sphere to be correctly computed. After the ray-trace, if these extra dimensions have been added to the ray-trace grid, then the deposited power is integrated over the additional grid directions to obtain the source term on the hydrodynamic grid.

### **1.3.2.2 Semi-Structured Eulerian Grid with Combined Cells**

Ideal ICF implosions are spherically symmetric and departures from this symmetry are usually higher-order corrections. Simulations of these experiments therefore typically employ a computational grid with spherical symmetry. CHIMERA typically simulates direct-drive implosions with a spherical-polar Eulerian mesh, shown in Fig. 1.4.a. This grid has the advantage of simplicity to take gradients across cells, however it has the disadvantage that cell edge lengths tend to zero as the radial coordinate,  $r \rightarrow 0$  and the polar coordinate,  $\theta \rightarrow 0, \pi$ . This limits the hydrodynamic timestep, because stable explicit timesteps are inversely proportional to edge lengths, increasing the cost of 3-D spherical simulations. The issue is circumvented at late times in the implosion by remapping onto a Cartesian grid, which does not have vanishing edge lengths and face areas [45].

Computing CBET requires at least a single ray from each interacting beam to pass through

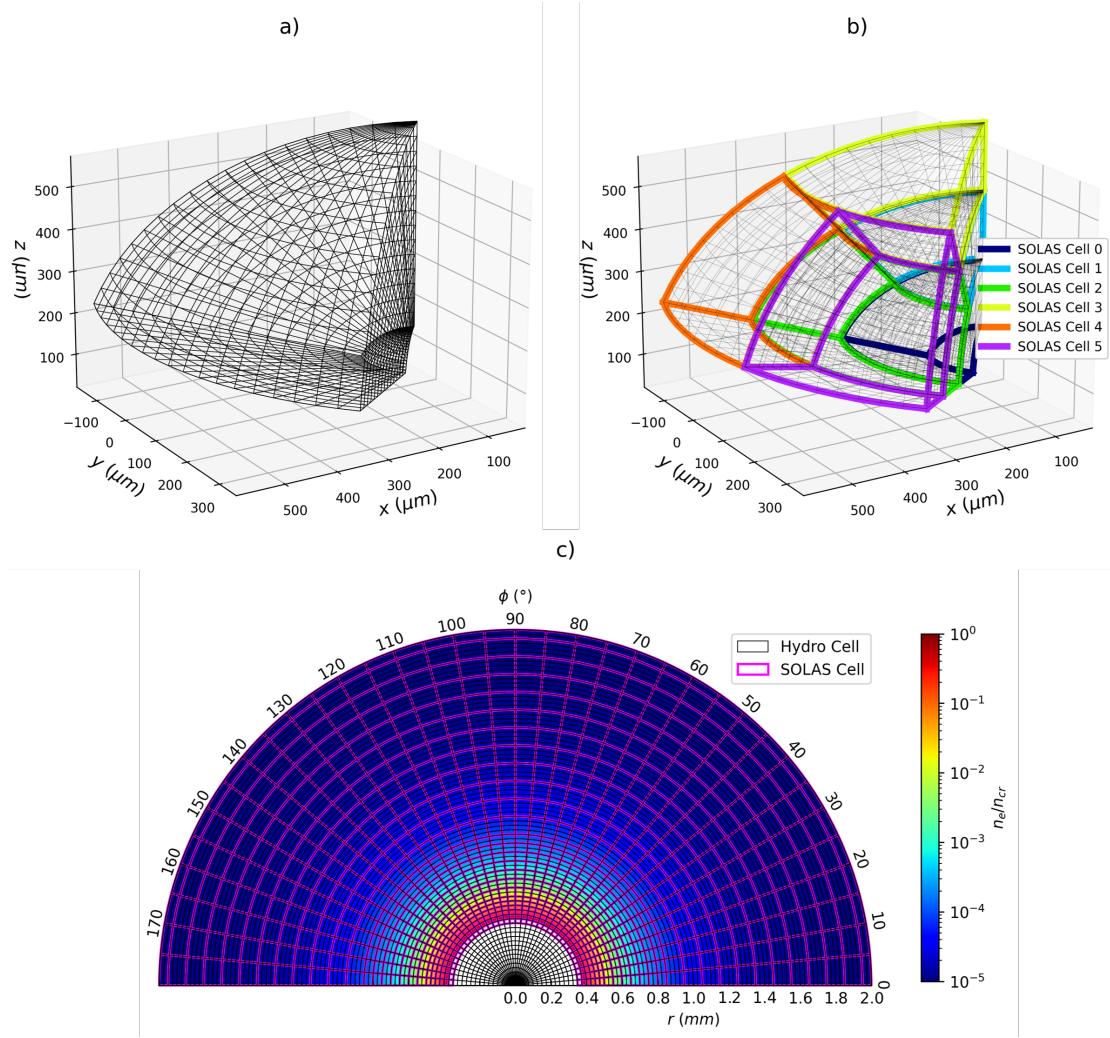


Figure 1.4: Illustrative diagrams demonstrating a) the spherical polar mesh used by CHIMERA for hydrodynamic calculations, b) the cell combination mechanism employed by SOLAS to obtain a roughly equal area grid for spherical simulations and c) the adaptive radial cell combination to reduce resolution in regions where CBET and refraction are unimportant.

each computational grid cell where the interaction should be important. For spherical-polar meshes the vanishing cell volume therefore sets extreme minimum ray number limits for the calculation to fully resolve all the fields, particularly the reflected field in direct-drive simulations, which spreads out over  $4\pi$  steradians, as is shown in Fig. 1.1. Hydrodynamic resolutions are often also excessive for ray-trace calculations to resolve the necessary refraction and energy exchange of the light. Rays must also stop at each cell boundary in order to deposit the correct amount of energy into each grid cell and therefore the expense of the ray-trace is directly proportional to the number of grid cells that the rays see.

To circumvent these issues, cells can be combined around the spherical grid angles in order to make a semi-structured Eulerian grid, as is shown in Fig. 1.4.b. Hydrodynamic grid cells are merged together on the SOLAS mesh in each grid direction until a pre-specified resolution, set by the user, is reached. In Fig. 1.4.b, many cells are combined in each direction to

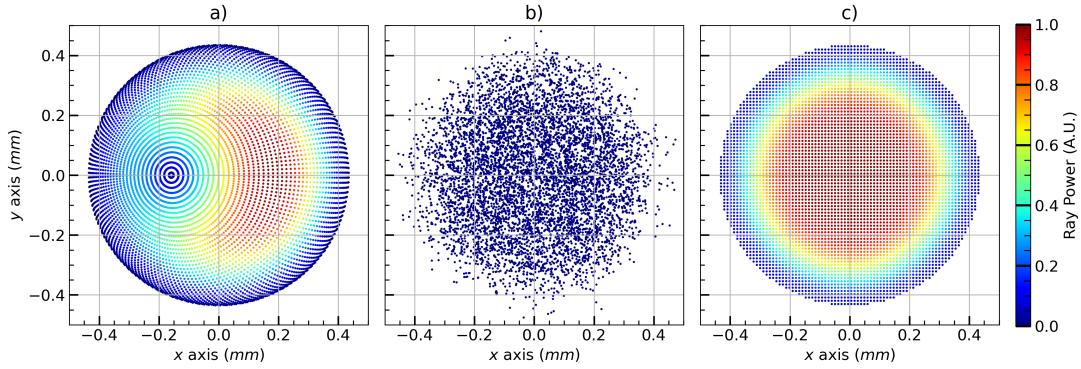


Figure 1.5: Example initial locations for  $\sim 4000$  rays for an OMEGA beam port with super-Gaussian intensity profile, with a shape defined by  $n_s = 5.2$  and  $\sigma = 352 \mu\text{m}$ . The inverse-projection ray locations are shown in sub-figure a) which demonstrate a higher ray density for rays pointed to the pole at  $x \sim -0.15\text{mm}$ . b) shows the randomly sampled rays where all rays are initialised with the same power, and c) shows uniformly sampled rays.

clearly display the effect. Typically, for spherical simulations however, the cells are not combined in the polar direction,  $\theta$  and  $n_\varphi$  cells are combined in the azimuthal direction until the azimuthal and polar resolutions match, explicitly  $n_\varphi r \Delta\varphi \sin\theta \approx r \Delta\theta$  at a given radius,  $r$ .

Fig. 1.4.c demonstrates the capability of the meshing algorithm to adaptively combine cells in the radial direction based upon density gradients. This allows large cells to exist in the coronal plasma where the light refracts minimally and deposits little energy, while the sharp turning point regions close to the critical surface can be well resolved. To find the number of radial cells to combine at a given radius  $n_r$ , maximum gradient length scale from the combined hydrodynamic cells,  $L_{n_e} = n_{crit}/|\nabla n_e|_{max}$  is calculated. The algorithm then finds the optimal number of cells to merge together such that the new cell resolution  $n_r \Delta r \approx C_L L_{n_e}$ , where  $C_L$  is a user parameter<sup>7</sup> which can be reduced to limit cell combining. A maximum cell size is also set to prevent excessively large cells in regions with small density gradients.

Note that many options for equal area computational grids exist which, unlike the semi-structured grid described here, are completely uncorrelated from the Rad-Hydro mesh [46, 47]. These have the advantage that the grid can be chosen purely based upon quantities important to the laser ray-trace. However, the mesh described in this section has the advantage that interpolation to the hydrodynamic mesh is extremely straightforward as the SOLAS grid cells completely overlap the CHIMERA cells. This also minimises artefacts from interpolation between the grids, which have the potential to introduce spurious high order modes to the deposition source term.

### 1.3.3 Ray Initialisation

To obtain a noise-free energy deposition source term from a ray-trace, both the grid and choice of initial ray number and location is crucial. The problems are closely related as a higher resolution grid requires a larger density of rays to get equivalent ray-per-cell statistics.

<sup>7</sup>Default value set to  $C_L = 0.05$ .

It is therefore wise to choose the number of rays used in a simulation to be some function of the grid used for the ray-trace. Several methods for ray initialisation which have been implemented are briefly outlined here, along with a summary of their strengths and weaknesses.

All beams profiles described in this thesis have circularly symmetric super-Gaussian intensity profiles described by the equation,

$$I(r) = I_0 \exp\left(-\left|\frac{r}{\sigma}\right|^{n_s}\right), \quad (1.1)$$

where  $r$  is the distance from the centre of the beam port, perpendicular to the beam normal,  $I_0$  is the peak intensity,  $\sigma$  is the beam width<sup>8</sup> and  $n_s$  is the super-Gaussian exponent.

**Uniform Sampling** In this method of ray initialisation, demonstrated in Fig. 1.5.c, each beam is assigned a number of rays and a maximum initialisation radius<sup>9</sup>, beyond which rays are not initialised. Rays are then placed on a uniform, square grid in this plane with a power proportional to the intensity value from by Eq. 1.1. For all three procedures described in this subsection, the total summed power of the rays is normalised to the incident beam power after the initialisation of all rays is complete. Typically, for direct-drive, the total number of rays used for spherical simulations is chosen to be,

$$N_{\text{ray}} = C_N \max(N_r N_\varphi, N_r N_\theta, N_\theta N_\varphi), \quad (1.2)$$

where  $N_{i=r,\varphi,\theta}$  is the number of SOLAS grid cells in each grid direction and  $C_N$  is a user-parameter multiplier which to give better ray statistics for CBET simulations when required.  $C_N \sim 2$  is found to give converged deposition when including CBET and so this value is used by default.

When there is sufficiently low ray-per-cell statistics, beat phenomena can occur between the ray spacing and the grid resolution. In this event, especially when using a nearest neighbour interpolation for ray power deposition onto the grid, significant spurious modes can be introduced to the power deposition. To resolve this, ray locations can be ‘dithered’ so that they take a random position within the polygon<sup>10</sup> defined by neighbouring ray locations. This option is always used for simulations using uniform sampling in this thesis.

**Random Sampling** Rays can also be randomly sampled according to the intensity profile. Example ray locations from this method are shown in Fig. 1.5.b. Note that the intensity profile purely emerges from the ray locations in this method as all rays have equal power. For no-CBET simulations this is a useful method as it minimises coherent build-up of noise from ray-spacing, grid-resolution beating. However, the wings of the intensity profile have poor ray statistics so for direct-drive CBET calculations, resolving the reflected field and therefore the dominant backscatter CBET is excessively expensive.

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<sup>8</sup>The radius at which  $I = I_0 e^{-1}$ .

<sup>9</sup>The maximum initialisation radius is usually set to be the radius at which  $I = I_0 e^{-3}$ .

<sup>10</sup>This polygon is a square for uniform sampling, but not for inverse-projection

**Inverse-Projection** Inverse-projection is the final method that has been implemented for ray initialisation in SOLAS. The algorithm is described in detail in Appendix A of Ref. [31]. The method works by finding several surfaces, defined by fractions of the critical density and then creating aim points in each cell on the surface. These aim points and their associated area on the surface are back-projected onto each beam port and rays are created if the aim point is not obscured by the surface on which it was created. The power of the ray is then the back-projected area multiplied by the intensity at the beam port location. For a spherical polar grid, this gives a ray distribution which varies in ray density according to which region on the beam port maps to regions on the grid with higher or lower cell-face areas. The higher ray density region on the beam port maps to the polar region of the grid, at  $x \sim -0.16\text{ mm}$  on Fig. 1.5.a have a correspondingly lower power compared to rays at equal radii to give the same intensity.

While inverse-projection does guarantee good ray statistics at given surfaces for the in-bound component of the field, the algorithm does not extend to beyond ray caustics, or the reflected field component. Therefore, ray statistics are not guaranteed to be good for the reflected field, so the advantages of this method are not evident for direct-drive CBET simulations when backscatter must be resolved. However, for no-CBET, direct-drive simulations, this is a useful option for ray initialisation.

For all simulations in this thesis, the uniform sampling method with ray dithering is used. This method is found to give the best ray-per cell statistics across the semi-structured Eulerian grid described in Sec. 1.3.2 while minimising the overall number of rays used required to resolve CBET.

After rays are initialised on each of the beam ports, they are extrapolated to the edge of the computational domain. Beam focussing is typically neglected, so all rays are assigned velocities parallel to the beam normal. This approximation is widely used for codes that simulate OMEGA-scale direct-drive implosions, where the lasers are approximately collimated on the implosion scale [23, 31]. It is assumed that outside of the simulation domain is vacuum and rays are therefore extrapolated to the edge of the computational domain in straight lines, using a simple root finding algorithm to obtain the intersection of the rays with either a spherical, cylindrical or rectangular domain depending on the simulation geometry.

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### 1.3.4 Equations of Rays and Adaptive Integration

The partial differential equations that are integrated along the trajectory of each ray are,

$$\begin{aligned}\frac{d\mathbf{x}}{d\tau} &= \mathbf{k}, \\ \frac{d\mathbf{k}}{d\tau} &= \frac{1}{2}\nabla\varepsilon(\mathbf{x}), \\ \frac{d\varphi}{d\tau} &= \varepsilon(\mathbf{x}), \\ \frac{d\omega}{d\tau} &= \frac{\omega}{2c}\frac{\partial(n_e/n_{cr})}{\partial t}, \\ \frac{dP}{d\tau} &= -\kappa_{IB},\end{aligned}\tag{1.3}$$

where  $\varepsilon = 1 - n_e/n_{cr}$  is the dielectric permittivity of the plasma,  $n_e$  and  $n_{cr}$  are the electron and critical densities,  $c$  is the speed of light,  $t$  is time,  $\kappa_{IB}$  is the Inv-Brem absorption kernel, given explicitly in Eq. ?? and  $\tau$  is the ray path length. Note that the phase here is defined such that it has the same dimension as path length and therefore has spatial units. The wavevector is normalised such that  $|\mathbf{k}| = \sqrt{\varepsilon}$ . The optical path length is related to the arc length,  $ds$  by the variable change  $ds = d\tau\sqrt{\varepsilon}$ . In some formulations of the ray-tracing equations, the differential equations are written in terms of the arc length [31, 48], which is equivalent, but more complicated and without benefit. Note that the time,  $t$  is not related to the ray path length  $\tau$ , because an operator split approach is taken to the ray-trace, such that the hydrodynamic variables are frozen and the ray-trace finds the time independent trajectory of the light through these profiles. The path length  $\tau$  is therefore best seen as a parameterisation of the ray curve.

The frequency shift arises due to light propagating through a time varying refractive index. This causes successive wavefronts to bunch up (blue-shift) or rarefy (red-shift), which occurs for a temporally increasing or decreasing  $n_e$  respectively [49]. This effect is only significant for the work presented in this thesis when calculating CBET. For direct-drive calculations at the OMEGA laser facility scale, this results in wavelength shifts of  $\Delta\lambda/\lambda_0 \sim \mathcal{O}(0.1\%)$ , which is typically negligible for computing ray trajectories. However, the induced wavelength shift is relatively significant when calculating the CBET resonance and in direct-drive can alter the spatial location where scattering is significant [23]. Neglecting time-dependent ionisation effects, which is a good assumption during peak-power for the approximately steady-state direct-drive corona where CBET occurs, the time derivative can be obtained by assuming that  $n_e$  is advected with the bulk fluid,

$$\frac{\partial(n_e/n_{cr})}{\partial t} = -\frac{\iiint_V \nabla \cdot [(n_e/n_{cr})\mathbf{u}] dV}{\iiint_V dV},\tag{1.4}$$

where  $\mathbf{u}$  is the fluid velocity and the integral is over a computational cell volume. Using the Divergence Theorem, this can be rewritten,

$$\frac{\partial(n_e/n_{cr})}{\partial t} = -\frac{\oint_S (n_e/n_{cr})\hat{\mathbf{n}} \cdot \mathbf{u} dS}{\iiint_V dV},\tag{1.5}$$

where the integral on the numerator is now over the cell bounding area, with normal  $\hat{\mathbf{n}}$  [31].

Two algorithms have been implemented to integrate Eqs. 1.3, an adaptive RK45 algorithm and the Kaiser algorithm. These are described below, alongside the method for ray-deposition-to-cell interpolation and time-stepping considerations.

**Adaptive RK45 Algorithm** The default ray evolution algorithm in SOLAS is to solve Eq. 1.3 using an adaptive RK45 algorithm with stepsize control [50]. Ray steps are either limited by the error from this algorithm, or by distance to the next impact with a cell face. Evaluations of the right-hand side of these equations employ trilinear interpolation for  $n_e$ ,  $\nabla n_e$  and  $T_e$  to obtain varying values for these quantities at different locations throughout the cell. All other quantities are either defined at the ray location, or use nearest neighbour interpolation. The `bspline-fortran` library has also been implemented to allow tricubic interpolation of  $n_e$ , which yields a quadratically varying  $\nabla n_e$  across the grid [51]. This is slow to evaluate many times per ray step as is required in an RK45 algorithm and therefore limits performance, but is used throughout this chapter as a higher order solution to validate accuracy of the default interpolation.

Linear interpolation of  $n_e$  and  $T_e$  is required for accurate computation of  $\kappa_{IB}$ , especially for ‘cold-start’ simulations where light impacts upon a solid target and the plasma profiles have steep gradients. Interpolation of  $\nabla n_e$  was found to be necessary to obtain low-noise ray amplitudes from neighbouring rays for CBET evaluation, which is described in more detail in 1.4. Linear interpolation of both  $n_e$  and  $\nabla n_e$  is technically inconsistent, because  $\nabla n_e$  should be the gradient of  $n_e$ , however the test cases presented in Sec. 1.3.4.1 found that this did not reduce accuracy compared to the Kaiser algorithm, which employs a self-consistent linear interpolation of  $n_e$  and constant  $\nabla n_e$  within a computational cell.

**Kaiser Algorithm** The Kaiser algorithm for ray integration has also been implemented [48]. In this algorithm, each cell has a single  $\nabla n_e$  and an  $n_e$  value defined at the cell centre, allowing for linear interpolation of  $n_e$  throughout the cell. In a constant  $\nabla n_e$  region, the trajectory of rays can be analytically shown to follow a parabola. The Kaiser algorithm thus evolves rays over parabolic segments between cells by finding the intersection of the parabola with cell faces, using a root finding algorithm. Assuming constant  $\nabla n_e$  cells leads to discontinuities in  $n_e$  at cell interfaces when the true gradient is not linear and thus Snell’s Law is used to refract rays upon cell exit. While this method is often slightly more efficient compared to the adaptive RK45, it was found that when attempting to reconstruct the ray amplitude from the area of neighbouring rays, as outlined in Sec. 1.4, excessive noise was introduced in the amplitude from the discontinuous refraction. Therefore, the adaptive algorithm was used for all work in this thesis, other than where explicitly outlined.

**Shape Functions for Ray Deposition** Nearest neighbour interpolation of power deposition, which is defined along the ray trajectory, onto the computational grid can result in significant levels of noise in the power deposition profile when ray statistics are not sufficiently high. This is similar to problems experienced in PiC codes when interpolating macro parti-

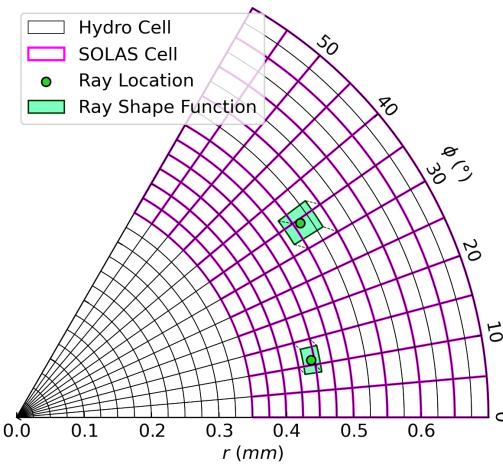


Figure 1.6: Shape function smearing for ray power interpolation to the SOLAS grid for a cylindrical mesh. The shape function has the same size and geometry as the SOLAS grid cell which the ray is located in. A top-hat shape function is used, so the fraction of the deposited power interpolated to a given cell is proportional to the volume overlap of the shape function with the cell.

cles to the grid without the use of shape functions [52, 53]. A PiC-inspired shape function approach was therefore taken, where the power deposition from rays is smeared across neighbouring cells. A top hat shape function is used, with a shape function that has the size and shape of the cell that the ray is located within, as is shown in Fig. 1.6. The power deposited into each cell that has an overlap with the shape function bounds is therefore proportional to the volume overlap of the shape function with the cell. This approach is more explicitly outlined, particularly for the case of the non- logically rectilinear grids which are present for SOLAS, in Ref. [54]. This smearing is somewhat numerically diffusive and therefore can be optionally disabled. For the work presented in this thesis, the shape function smearing is employed for multidimensional direct-drive simulations, but otherwise disabled.

**Hydrodynamic Time Step Limiter** Laser heating of a plasma results in a temperature increase of the material, which also reduces the Inv-Brem kernel. If the hydrodynamic timestep,  $\delta t$  is too large, excessive heating occurs in initially cold cells, which should have the magnitude of absorption gradually reduced as they warm up. This can lead to spurious oscillations in temperature at fine grid resolutions. A limiter on the hydrodynamic time-step is therefore introduced to prevent large heating in any cell within one timestep. The limit detailed by Haines in Ref. [55] is enforced, which estimates the increase in temperature of a cell from Inv-Brem heating by assuming an ideal gas equation of state to derive,

$$\delta t = \min_{n \in \text{cells}} \left( \tilde{\delta}t, C_{\delta t} \frac{3}{2} \frac{n_{e,n} e T_{e,n} V_n}{P_n} \right), \quad (1.6)$$

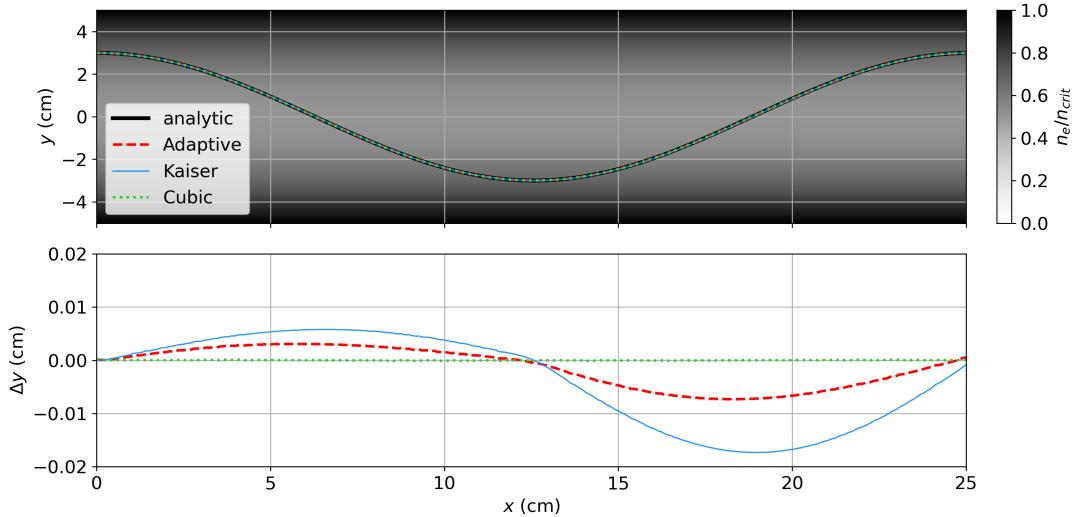


Figure 1.7: Results of the quadratic trough ray trajectory test problem for the adaptive solver with 3 different cases. These are default settings (linear interpolation of both  $n_e$  and  $\nabla n_e$ ), the Kaiser algorithm (linear interpolation of  $n_e$  and uniform  $\nabla n_e$  in a cell) and the adaptive algorithm with tricubic interpolation of  $n_e$ ). The top plot shows ray trajectories and analytic trajectory and the bottom plot shows absolute errors for the 3 cases.

where  $\tilde{\delta t}$  is the stable hydrodynamic timestep from other physical processes,  $n_{e,n}$ ,  $T_{e,n}$ ,  $V_n$  and  $P_n$  are the electron density, electron temperature, volume and deposited power into cell  $n$ , and  $C_{\delta t}$  is a user defined stability parameter. By default, the parameter  $C_{\delta t}$  is set to 1, which is found to be adequate for most situations [55]. Note that Eq. 1.6 is only applied in cells where  $n_{e,n} > 1 \times 10^{-4} n_{cr}$ . Small amounts of deposition in these cells can lead to large temperature increases, which severely limits  $\delta t$ , despite the cells often being far from the region of interest, normally near critical, where the maximum deposition occurs. For simulations where a laser heats a solid, initially cold target, Eq. 1.6 mostly only affects the hydrodynamic timestep early in the simulation, after which the laser-heated plasma corona remains in a mostly steady-state unless there are steep changes in incident power.

#### 1.3.4.1 Ray Solver Validation

Several validation problems have been conducted to verify that the ray solver has been implemented correctly. Here the quadratic trough and cylindrical helix test problems are presented, which compare the path of a single ray in to an analytic solution in order to verify that the ray solvers correctly obtain the trajectory of light. The blast wave problem is also presented which is a test of Inv-Brem absorption to an analytic solution in the absence of thermal conduction and hydrodynamic motion.

**Quadratic trough** The quadratic trough is a test of ray trajectory in a quadratic density trough, which admits an analytic solution of a periodically oscillating ray [48, 55]. The den-

sity profile used for the test is defined as,

$$n_e(y) = \frac{n_{\text{cr}}}{2} \left( 1 + \frac{y^2}{y_c^2} \right), \quad (1.7)$$

which for light with wavelength  $\lambda = 351$  nm, is initialised with  $n_{\text{cr}} = 9.049 \times 10^{21}$  cm $^{-3}$  and  $y_c = 5$  cm. The domain has bounds  $x \in [0, 25]$  cm and  $y \in [-5, 5]$  cm and a ray enters the domain at  $[x_0, y_0] = [0, 3]$  cm. Analytic integration of the first 2 lines from Eq. 1.3, yields the analytic trajectory as a function of ray path length  $\tau$ ,

$$\begin{aligned} x(\tau) &= \tau \sqrt{1 - \frac{n_e(y_0)}{n_{\text{cr}}}}, \\ y(\tau) &= y_0 \cos \left( \frac{\tau}{\sqrt{2} y_c} \right). \end{aligned} \quad (1.8)$$

The analytic trajectory is compared to the solution from the adaptive solver (using both default interpolation and tricubic interpolation of  $n_e$ ) and the Kaiser algorithm in Fig. 1.7. The associated error, defined as the difference in  $y$  from the analytic value at a given  $x$  is also plotted. Note that all results were obtained using a grid resolution of  $100 \times 100$  cells. The top panel demonstrates that all trajectories are identical to the analytic solution by eye. The corresponding errors show that the tricubic interpolation effectively perfectly recreates the analytic solution and the Kaiser error is slightly more significant compared to the default adaptive error. The tricubic error is insignificant because the cubic interpolation perfectly recreates the true density profile, so the errors are purely numeric, not due to the resolution. Errors from the other algorithms would therefore decrease more quickly with increasing resolution as the density profile the ray sees becomes more similar to a quadratic trough. The error from the default adaptive interpolation (linear interpolation of both  $n_e$  and  $\nabla n_e$ ) is lower than that of Kaiser (linear  $n_e$  and uniform  $\nabla n_e$ ), indicating that the inconsistent interpolation is not a significant issue for resolving ray trajectories.

**Cylindrical-Helix** The quadratic trough test ensures that the ray evolution algorithm functions correctly in Cartesian geometry. An additional test was desired in non-Cartesian geometry, because CHIMERA also operates with cylindrical and spherical grids. The cylindrical-helix test was therefore devised, where a ray enters an axially symmetric density trough that keeps the ray at a constant, cylindrical radius,  $r$ , as is shown in Fig. 1.8.b. As is shown in Fig. 1.8.a, the  $z$  extent of the simulation is chosen such that the ray has performed one complete helical spiral when it exits the domain. By requiring that a ray undergo circular motion in the  $x - y$  plane and specifying the initial ray direction such that  $k_{z0} = \sqrt{k_{x0}^2 + k_{y0}^2}$ , it can be derived that a ray entering the domain at  $r_0$  will undergo one complete helical spiral over a  $z$  length  $\Delta z = 2\pi r_0$ , if the density profile has the shape,

$$n_e(r) = n_{\text{cr}} \left( 1 - \exp \left( \frac{-r^2}{2r_0^2} \right) \right). \quad (1.9)$$

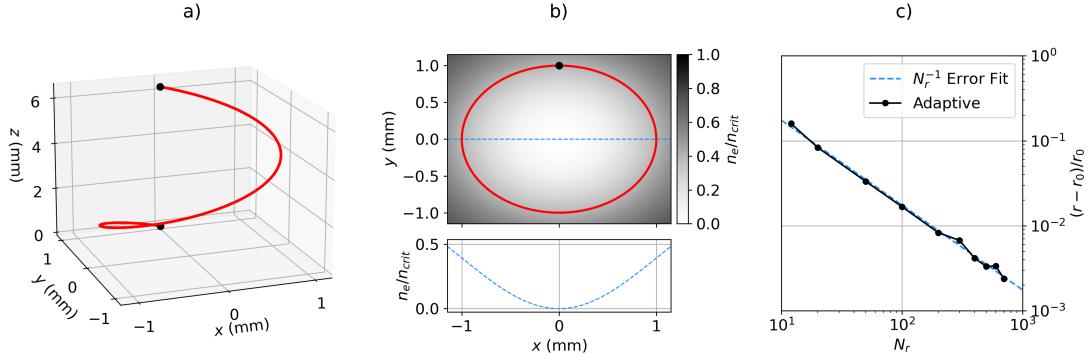


Figure 1.8: Results from the cylindrical-helix test problem, obtained using the default adaptive RK45 algorithm. a) shows the trajectory of a ray from a simulation with 100 radial cells in 3-D space. b) shows the trajectory of the ray projected on the  $x - y$  plane, along with the  $n_e$  profile. c) shows the error of the test, defined as the fractional difference in radius of the ray as it exits the domain from the initial radius, as a function of number of radial cells.

For a ray entering the domain at  $[x_0 = 0, y_0 = 1, z_0 = 2\pi]$  mm, with initial wavevector  $[k_{x0} = \sqrt{\varepsilon_0/2}, k_{y0} = 0, k_{z0} = -\sqrt{\varepsilon_0/2}]$ , the ray position as a function of path length  $\tau$  is,

$$\begin{aligned} x(\tau) &= r_0 \sin\left(\sqrt{\frac{\varepsilon_0}{2}}\tau\right), \\ y(\tau) &= r_0 \cos\left(\sqrt{\frac{\varepsilon_0}{2}}\tau\right), \\ z(\tau) &= z_0 - \sqrt{\frac{\varepsilon_0}{2}}\tau, \end{aligned} \quad (1.10)$$

where  $\varepsilon_0 \equiv \varepsilon(r = r_0)$ .

This problem was run in cylindrical geometry for a variety of radial resolutions, employing the default adaptive RK45 solver with linear interpolation of both  $n_e$  and  $\nabla n_e$ . The domain bounds for the problem were  $r \in [0, 1.5]$  mm,  $\varphi \in [0, 2\pi]$  and  $z \in [0, 2\pi]$  mm, with a single cell in both the  $\varphi$  and  $z$  directions. For all radial resolutions, the ray was observed by eye to exit the domain at the expected location, verifying that the ray-trace is functional in non-Cartesian geometries. Plotting the fractional error in ray exit location as a function of number of radial cells,  $N_r$  in Fig. 1.8.c, demonstrates that the error in final position of the ray scales as  $N_r^{-1}$ . This is because larger  $N_r$  more closely recreates the density profile in Eq. 1.9. The error scaling is dictated by the lowest order interpolation used in the algorithm. This is the linear interpolation of  $n_e$  for the default adaptive RK45 solver, which leads to the  $N_r^{-1}$  scaling.

**Blast wave** The blast wave problem is a validation test of the implementation of Inv-Brem absorption [55, 56]. In this problem, a 527 nm, 2 ns laser is incident on a uniform density ( $\rho = 1 \text{ mg/cm}^{-3}$ ), cold gas ( $T_{e0} = 1 \text{ eV}$ ) with fixed ionisation at  $Z = 6$ , that has an ideal gas equation of state. The  $\ln \Lambda$  value used for the  $\kappa_{IB}$  coefficient is fixed at 7. The laser travels in the  $x$

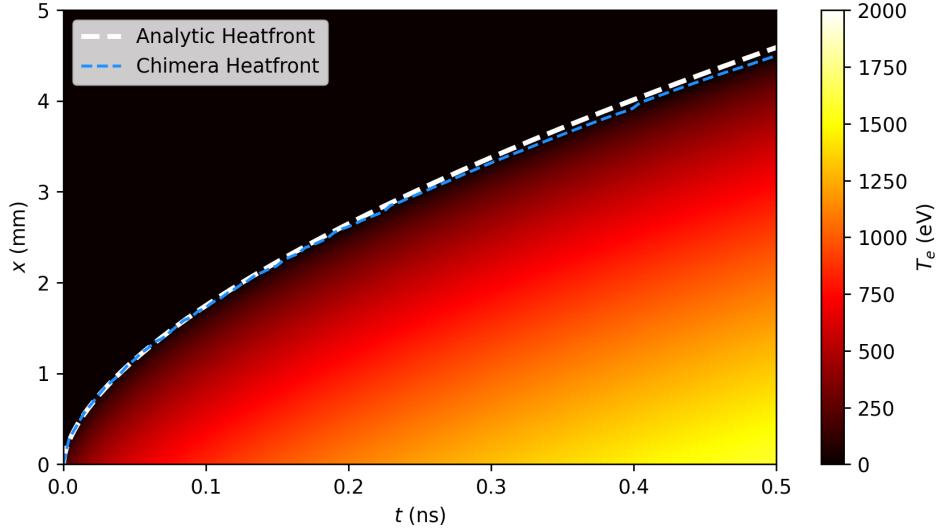


Figure 1.9: Results of the blast wave Inv-Brem absorption test problem. The colour-plot shows the temperature of the 1-D simulation as a function of time ( $x$ -axis) and space ( $y$ -axis). The heatfront obtained from CHIMERA, which is the maximum  $x$  location where  $T_e > T_{e0}$ , is in good agreement with the analytic solution.

direction and has an intensity of  $6.4 \times 10^{13} \text{ Wcm}^{-2}$ . Hydrodynamic motion and transport is disabled in the simulation. The laser is initially strongly absorbed in the gas, so energy is not transported beyond the initial layer, perpendicular to the laser propagation direction. As the gas heats up,  $\kappa_{\text{IB}}$  decreases and therefore more energy is transported further into the domain by the laser, moving the heatfront forward. The heatfront  $x_{\text{analytic}}(t)$ , defined as the largest  $x$  coordinate at a time,  $t$  where  $T_e > T_{e0}$ , has an analytic solution, derived by Denavit and Phillion in Ref. [56],

$$x_{\text{analytic}}(t) = \frac{2}{3\kappa_{\text{IB}}} \left( \frac{5}{3} \frac{\kappa_{\text{IB}} I}{n_e k_B} \right)^{3/5}, \quad (1.11)$$

which can be compared to the result from a ray-tracing simulation.

Fig. 1.9 shows the results for  $T_e(x, t)$  and the heatfront location from a CHIMERA-SOLAS simulation of the blast wave problem. The heatfront from the simulation compares well to the analytic heatfront, validating the implementation of the Inv-Brem absorption kernel.

## 1.4 Ray-Based Field Reconstruction and Ray Sheets

In order to compute CBET, or other LPIs in a ray-tracing calculation, additional information about the electric field or intensity of the light is required than can be obtained from the ray directly. This section outlines the method used in SOLAS to obtain the electric field of the light along the path of each ray, which broadly follows the implementation from Ref. [37].

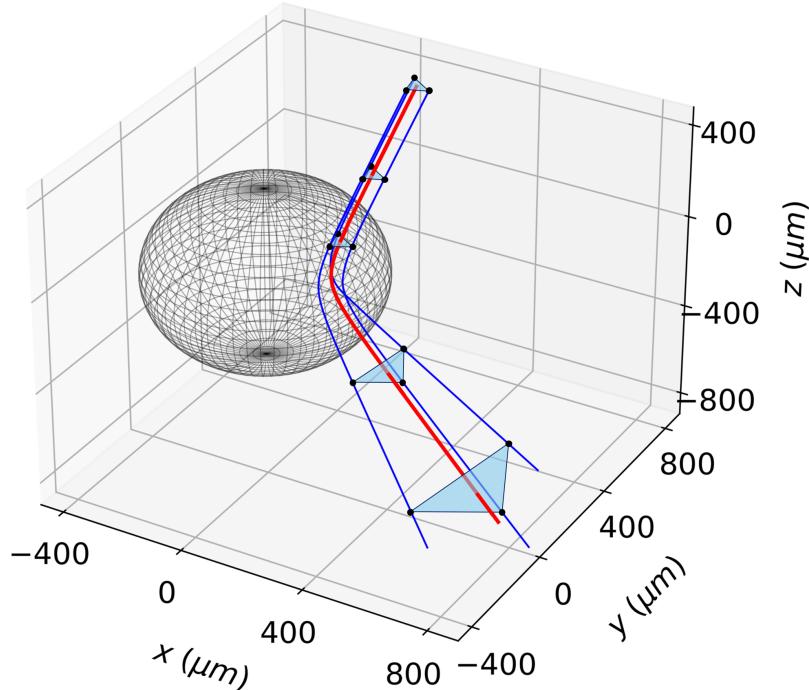


Figure 1.10: A 3-D illustration of the ray-amplitude construction for a single ray in a spherically-symmetric, direct-drive like plasma profile. The critical surface is shown as the grey wire mesh. The main-ray trajectory is plotted in red, and its three neighbour-ray trajectories are in blue. The area of the main-ray at 5 successive phase-fronts is plotted, *i.e.* when the main- and neighbour-rays all have the same phase,  $\varphi$ , is illustrated by the shaded triangles. This is inversely proportional to the ray-amplitude at this location.

#### 1.4.1 Ray-Amplitude and Field Estimate from Neighbour-Rays

Recalling the formula for the ray-amplitude,  $A$ , Eq. ?? from Sec. ??,

$$A(\tau) = A(0) \left| \frac{D(0)}{D(\tau)} \right|^{1/2}, \quad D(\tau) = \begin{bmatrix} \frac{\partial x}{\partial \zeta_1} & \frac{\partial x}{\partial \zeta_2} & \frac{\partial x}{\partial \tau} \\ \frac{\partial y}{\partial \zeta_1} & \frac{\partial y}{\partial \zeta_2} & \frac{\partial y}{\partial \tau} \\ \frac{\partial z}{\partial \zeta_1} & \frac{\partial z}{\partial \zeta_2} & \frac{\partial z}{\partial \tau} \end{bmatrix}, \quad (1.12)$$

where  $[x, y, z]$  and  $[\zeta_1, \zeta_2, \tau]$  are the ray real-space and phase-space coordinates respectively and  $D$  is the Jacobian for the coordinate transform from phase-space to real-space. When refraction forces rays closer together,  $|D(\tau)|$  will decrease and equally, when refraction forces rays to separate,  $|D(\tau)|$  will increase. The amplitude can therefore be approximated by assuming proportionality between the determinant of the Jacobian and the area,  $S$  of an infinitesimally small bundle of rays surrounding the main-ray,

$$A(\tau) = \varepsilon^{-1/4} \sqrt{\frac{S(\tau=0)}{S(\tau)}}, \quad (1.13)$$

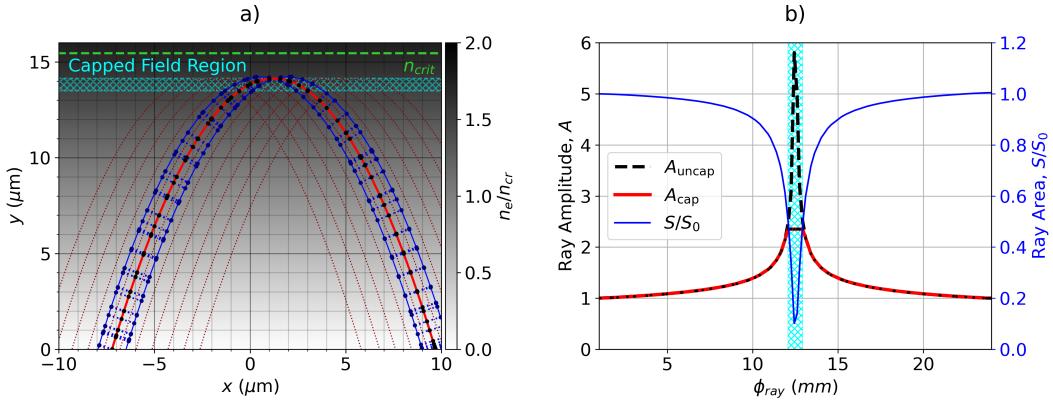


Figure 1.11: A demonstration of how the ray-amplitude is obtained from the area of neighbouring-rays for a beam propagating up a linear density gradient. Fig. a) explicitly shows the trajectory of a main-ray and its neighbour-rays in red and blue respectively, explicitly highlighting the beam's caustic region, where the amplitude of the light is capped. Fig. b) plots the area of the main-ray in blue (which goes through a minimum at the caustic location), as a function of the phase of the ray. Also shown are the uncapped and capped amplitudes of the ray.

where the  $\varepsilon^{-1/4}$  term accounts for swelling of the field due to the increased optical path through plasma with finite density [37]. Note that the amplitude is a purely geometric quantity, which holds no information about power changes of the light due to absorption or LPIs. The electric field outside the caustic region, which is discussed in more detail in Sec. 1.4.2, can then be obtained with the formula,

$$\frac{|E(\tau)|}{|E(\tau = 0)|} = \sqrt{\frac{P(\tau)}{P(\tau = 0)}} A(\tau), \quad (1.14)$$

where  $P$  is the power of the ray and the initial (vacuum) electric field  $|E(\tau = 0)| = \sqrt{2I_0/c\varepsilon_0}$ , where  $I_0$  is the intensity of the beam at the ray initialisation point on the beam port.

This method, which has been used successfully to model direct-drive CBET in the BEAM-CROSSER post-process code, is the approach taken to estimate the field in SOLAS. Explicitly, for every ray that initialised for a SOLAS calculation, an additional 3 'neighbour-rays' are initialised in an equilateral triangle around it, perpendicular to the initial direction of propagation. Note that when a 2-D ray-trace is used, only two neighbour-rays are required as the rays cannot change separation in the out of plane direction. It is important to distinguish the dimension of the ray-trace, which is the number of dimensions in which rays can move in a simulation, to the dimension of the hydrodynamics. For example, a 1-D spherical direct-drive Rad-Hydro simulation still requires a 3-D ray-trace where rays can move in  $x$ ,  $y$  and  $z$  for full accuracy.

These neighbour-rays are then co-traced up to the phase of the 'main-ray' to evaluate the area of the main-ray on its phase front. Fig. 1.10 plots an example of this co-tracing procedure for a single ray propagating in a 3-D direct-drive profile. It can be seen that as the

reflects away from the critical surface, the area diverges. Note that artificially large neighbour-ray separation was used for this plot as it is purely illustrative. Practically, the separation distance of the bundle of neighbour-rays (i.e. the side-length of the triangle,  $\Delta l$ ) does not need to be infinitesimally small, but small enough that it accurately captures the local change of  $|D(\tau)|$ . In terms of the hydrodynamic profiles, this means that  $\Delta l \ll L_{n_e}$ , where  $L_{n_e}$  is the density length scale of the plasma. For OMEGA scale direct-drive simulations, an initial separation of  $\Delta l \lesssim 1 \mu\text{m}$  is found to give converged behaviour. If the separation is too small however, floating point arithmetic can lead to noise in the ray-amplitude estimation. For all simulations presented in this thesis,  $\Delta l = \lambda_0/10 \ll 1 \mu\text{m}$  was used.

As mentioned in Sec. 1.3.4, it was found that the Kaiser algorithm for ray propagation, which uses a single  $\nabla n_e$  in each grid cell and discontinuously refracts rays at cell boundaries, led to large levels of noise in the ray-amplitude profiles. In order to obtain smooth profiles for  $A$ , linear interpolation of  $\nabla n_e$  was found to be necessary. The continuously varying  $\nabla n_e$ , which does not have discontinuities across cell interfaces, allows a smoothly varying area for each ray to be obtained.

Fig. 1.11.a demonstrates this amplitude reconstruction process in a purely illustrative example of a beam travelling up a linear density gradient at an angle, leading to a caustic at its turning point. The trajectory of several main-rays are plotted in dark-dashed red, and a single main-ray is highlighted in a brighter shade of red. The two neighbour-rays of this highlighted main-ray are plotted in blue. Note that in this figure, the default initial main-neighbour-ray separation of  $\Delta l = \lambda_0/10 = 0.035 \mu\text{m}$  was used, but the separation has been magnified on the plot to visualise the amplitude reconstruction from the area more clearly. The locations of the main- and neighbour-rays after a single step of each main-ray are plotted as black and dark-blue spots respectively. The ‘area’ of the neighbour-rays  $S$ , which in this 2-D ray-trace example here is simply a distance, is plotted on 1.11.a as a dashed blue line between the neighbour-rays. The neighbour-rays are pushed until they have the same phase as the main-ray, at which point the area is evaluated.

#### 1.4.1.1 Caustics and Ray-Sheets

Caustics are an important concept in GO, and the reconstructed field in their vicinity must be handled correctly in order to accurately model CBET in ray-based codes. Fig. 1.11.b plots the area of the main-ray as a function of main-ray phase in blue. At  $\varphi_{\text{ray}} \sim 12 \mu\text{m}$ ,  $S \rightarrow 0$ , *i.e.* neighbouring-rays cross over each other, which is known as the ray-caustic. In the absence of caustics, rays never cross over each other and therefore the mapping from ray phase-space (beam-port locations, where  $\tau = 0$ ) to real-space is single-valued, or in other words each point in space will be reached by (at most) one ray from each beam. In the presence of caustics, this single-valued mapping breaks down and a single real-space location is accessible by multiple ray launch locations. A single-valued mapping is still possible if each beam is separated into distinct ‘sheets’ after caustics, where each sheet still has a single-valued projection from phase-space to real-space.

This is illustrated in Fig. 1.12.a, where rays from a single beam are plotted propagating up

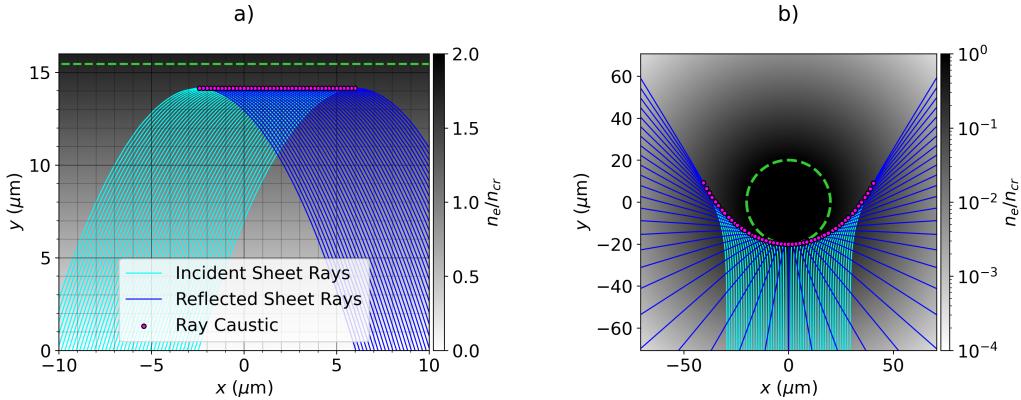


Figure 1.12: An illustration of the concept of caustics and ray sheets. In a), the trajectory of rays traversing up a linear density gradient are plotted. Separate colours are used for the incident and reflected sheets, which are separated by a caustic. In b), rays propagate up a density profile,  $n_e = n_{cr} \exp [-(r_{\mu\text{m}} - 20)/100]$ . Unlike in a), the ray caustics do not occur at their turning point, *i.e.* the minimum radius.

a linear density gradient with  $n_e = n_{cr} y_{\mu\text{m}} / 15.5$ . As the density gradient is purely in the  $y$  direction, each ray has translational symmetry in  $x$  and therefore each ray-caustic is located at the ray turning point,  $y \sim 14 \mu\text{m}$ . The portions of the ray trajectories that fall before and after the caustic are separated into the ‘incident’ and ‘reflected’ sheets respectively and plotted in separate shades of blue. Fig. 1.12.b also plots rays from a single beam, but now propagating through a cylindrically symmetric  $n_e$  profile. As can be seen from this figure, the caustic is not the same as the ray turning-point<sup>11</sup>, but it is the location that the amplitude diverges.

Unless the amplitude is tracked along each ray trajectory, the location of the caustic cannot be identified and rays cannot be separated into distinct sheets, both of which have important ramifications for ray-based CBET models. Firstly, because the amplitude diverges, the field values can become large and therefore must be capped to diffraction limited values (the SOLAS methods for this are discussed in more detail in Sec. 1.4.2). The power change of rays due to CBET scales exponentially with field strength squared, so erroneously large field values are extremely problematic to accurately computing power changes. Additionally, if the beam cannot be separated into distinct sheets, then self-CBET, where the reflected component of a beam interacts with the incident component of the same beam, must be neglected. In SOLAS, the neighbour-ray method allows the amplitude to be tracked along each ray, enabling caustic location identification and sheet-separation. Currently, it is assumed that each sheet has two components, *i.e.* an incident and reflected field, but it should be possible to extend this to multiple caustics to accurately model complex CBET interactions such as target-stalk simulations [57–59]. This is an advantage of the forward ray-tracing approach to CBET modelling, compared to inverse-ray-tracing, where only plasma profiles that result in a single caustic can be modelled [23].

The phase of the light is also evolved along each ray trajectory in equations 1.3, and therefore the coherent sum of all electric fields can be reconstructed which involves interference

<sup>11</sup>Defined here as the location of maximal  $n_e$  experienced along the ray trajectory.

between sheets. The total electric field is,

$$E = \sum_j^{\text{sheets}} |E_j| \exp(i(k_0\varphi_j - \pi\alpha_j/2)), \quad (1.15)$$

where  $k_0$  is the vacuum wavevector of the light and  $\alpha_j$  accounts for the  $\pi/2$  phase shift that occurs when a ray changes sheet, *i.e.*  $\alpha_j = 0$  for the incident sheet and  $\alpha_j = 1$  for the reflected sheet [37].

The fields are discretised on the SOLAS mesh by nearest neighbour interpolation from the ray to the cell. When multiple rays from the same sheet pass through a cell, then the inverse-distance weighted average of the mid-point along the ray step to the cell centre is used to compute the field value for the cell. This leads to a field structure,  $|E_j|(\mathbf{x}, \mathbf{k}, \omega, \phi)$ , where  $j$  refers to each sheet,  $\mathbf{x}$  is the spatial location of the field which refers to the cell the field is stored in, and  $\mathbf{k}, \omega, \phi$  are the wavevector, frequency and phase of the ray. For many beam simulations, there is a large associated memory cost to store this information on the mesh, which is discussed in more detail in Sec. 1.5.4. When computing the effect of CBET, it is also necessary to interpolate the fields stored at the cell centre to the ray trajectories. Nearest neighbour interpolation is also used for this step, so rays experience a constant value for  $|E_j|(\mathbf{x}, \mathbf{k}, \omega, \phi)$  throughout the cell. It is technically possible to use higher order interpolation for both stages, either from an unstructured mesh of the ray locations, or from the SOLAS mesh, which is close to logically rectilinear. The second of these options was tested but proved both to be the dominant computational cost for many beam CBET interactions and also challenging to robustly implement for non-Cartesian meshes, because the angular resolution for typical spherical direct-drive simulations was too coarse for the required accuracy.

### 1.4.2 Caustic Field Capping

In the vicinity of beam-caustics, the electric field magnitude of a beam is limited by diffraction which is neglected in GO. Therefore, ray-based models which do not cap fields near caustics can experience divergent electric fields. For direct-drive simulations, a significant amount of CBET occurs near caustics and therefore sensibly capping the field value is crucial for accurate simulations [28]. This cannot be achieved without knowledge of the caustic location, which is a severe limitation to the predictive capability of CBET codes that do not track the amplitude. This section describes two methods that have been implemented to cap the field values in the vicinity of laser caustics, the Field Limiter (FL) and Etalon Integral (EI) approaches.

**Field Limiter** This is a straightforward hard cap on the maximum value of the electric field reconstructed during the ray-trace. The electric field of light propagating up a linear density gradient has the analytic solution of an Airy function. The field limiter approach is to cap all

electric fields to this value,

$$\frac{|E|}{|E(\tau = 0)|} = \sqrt{\frac{P(\tau)}{P(\tau = 0)}} \min \left[ A, \sqrt{\zeta} \left( \frac{n_t}{n_{cr}} \right)^{1/4}, \sqrt{\zeta} \sqrt{\frac{S(\tau = 0)}{S(\tau)}} \right],$$

$$\zeta = 0.9 (\omega L/c)^{1/3},$$

$$L = n_{cr}/|\nabla n_e|_t,$$
(1.16)

where  $n_t$  is the electron density at the caustic and  $|\nabla n_e|_t$  is the magnitude of the electron density gradient at the caustic location [36, 39, 60]. The first term in the minimum is the standard field reconstruction from the amplitude away from caustics. The second term is the maximum of the Airy function and the third term is an improvement to the second term for near-normally incident rays [37]. Fig. 1.11.b plots the uncapped and capped amplitudes obtained from the field limiter approach for light propagating at an angle to a linear density gradient.

**Etalon Integral** The EI method is an improvement to the FL, which allows for deviations in the density profile away from linearity [27, 61, 62]. Several distinct types of caustic exist for different geometries of problem. For example in direct-drive, the light reflecting from the critical surface forms a ‘fold’-type caustic, whereas focussing light leads to a more complex ‘cusp’-type caustic [63]. If the form of the caustic is assumed, then an approximation to the total field in the vicinity of the caustic is formulated, which allows for deviations from the ideal case. In the example of the direct-drive fold caustic, the field is assumed to be the sum of an Airy function and its derivative, which accounts for the deviations from linearity. An expression for the total field can be derived in terms of the ray-amplitude and phase before ( $A_1, \varphi_1$ ) and after ( $A_2, \varphi_2$ ) the caustic,

$$E_T = \sqrt{\pi} [(-\xi)^{1/4} (A_1 + A_2) \text{Ai}(\xi) - i(-\xi)^{1/4} (A_1 - A_2) \text{Ai}'(\xi)] e^{i(k_0 \chi - \pi/4)},$$

$$\chi = \frac{1}{2}(\varphi_1 + \varphi_2),$$

$$\xi = - \left[ k_0 \frac{3}{4} (\varphi_2 - \varphi_1) \right]^{2/3},$$
(1.17)

where  $\text{Ai}$  and  $\text{Ai}'$  are the Airy function and its derivative. In SOLAS,  $\text{Ai}$  and  $\text{Ai}'$  are computed numerically using the `special-functions` library [64]. This total field can then be divided between the incident and reflected sheet by inversion of Eq. 1.15 assuming that  $|E_1| = |E_2|$ ,

$$\frac{|E_j|}{|E_{j,0}|} = \frac{\sqrt{W_j} |E_T|}{\sqrt{2} [1 + \sin(\varphi_2 - \varphi_1)]^{1/2}}.$$
(1.18)

The assumption of equal fields is approximately valid for OMEGA scale direct-drive conditions, where the caustic region is small compared to the plasma scales and therefore ray powers are assumed to not vary significantly over this distance. The Etalon integral is ap-

plied in the ‘caustic-region’, which in Refs. [27, 65, 66], is stated to be equivalent to,

$$|\varphi_1 - \varphi_2| \leq \lambda_0/2, \quad (1.19)$$

where  $\lambda_0$  is the vacuum wavelength of the light.

Applying the EI method requires evaluation of Eq. 1.19 to determine if a ray is inside its caustic region, which relies on interpolation of the phase of the other sheet from the same beam onto the ray location. If using nearest neighbour interpolation, such as in SOLAS for interpolation of all field quantities onto rays, a grid resolution of  $\Delta \ll \lambda_0$  must be used, which is a much higher resolution than is required for ray-tracing and CBET without the EI. Therefore, the EI method is only used in SOLAS simulations for high resolution grid test problems, such as those Sec. 1.4.3. The alternative approach is to implement linear interpolation from the field quantities of all beams onto ray locations, such as in Ref. [37]. This was not deemed a viable approach however, because the mesh resolution for spherical simulations was found to not be well-enough angularly resolved to interpolate the phase with sufficient accuracy to the rays. The standard procedure for CBET simulations in SOLAS is to use the FL approach, which was found to give satisfactory results compared to the EI method.

### 1.4.3 Field Reconstruction Validation

In this section, several test problems will be presented that compare electric fields obtained using the SOLAS field solver in the absence of CBET, to those from the wave based solver LPSE in direct-drive relevant plasma profiles. The LPSE results are presented in Ref. [37], and are available from the repository [67]. In all the following test problems, the laser has a vacuum wavelength,  $\lambda_0 = 0.351 \mu\text{m}$  and a super-Gaussian intensity profile, defined in Eq. 1.1. All 1-D and 2-D validation problems assume lasers polarised out of the simulation plane along the  $z$  axis. The plasma density profiles are given by,

$$n_e(r) = n_0(d/r)^m, \quad (1.20)$$

where  $n_0/n_{\text{cr}} = 1.165$ ,  $d = 343 \times S \mu\text{m}$ ,  $m = 3.78$  and  $S$  changes the scale of the density profile, where  $S = 1$  corresponds to scale lengths from implosions using the full OMEGA laser-energy. These values were obtained from fitting the equation to 1-D Rad-Hydro calculations using the LILAC code.

#### 1.4.3.1 1-D Reflected Beam

The first validation problem is for the case of light propagating normally up a density gradient. For this problem, 1-D Cartesian geometry was used, so the density profile was obtained by transforming the spatial coordinate from equation 1.20,  $r \rightarrow x - 36.42 \mu\text{m}$ . The beam is travelled in the  $+\hat{x}$  direction, with  $I_0 = 14 \text{ W/cm}^2$ . Only a single ray was used because the ray-trace is completely 1-D. A scale factor of  $S = 1/16$  was used, resulting in a critical surface at  $x \sim 14.14 \mu\text{m}$ . The simulation had bounds  $x \in [-16, 20] \mu\text{m}$  with a resolution of  $\Delta x \sim 1 \text{ nm}$ . A high resolution was used here in order to resolve the highly oscillatory coher-

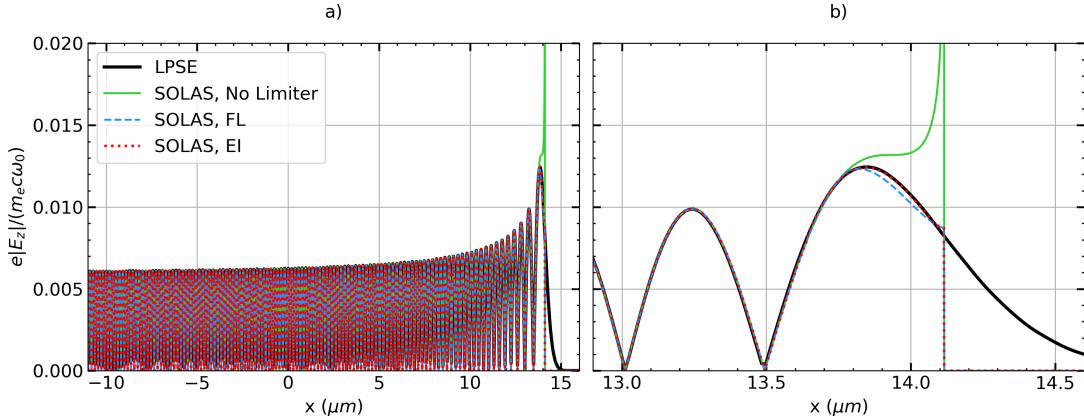


Figure 1.13: Results of the 1-D field reconstruction test, comparing SOLAS fields with different caustic field capping methods to the LPSE results. Panel a) shows the oscillatory behaviour of the field solution which arises due to the interference between the incident and reflected sheet. Panel b) plots the same results, but zoomed in on the caustic region to demonstrate the differences in fields obtained from the different methods.

ent field sum and also enabled use of the EI field capping method. Power changes of the rays due to Inv-Brem and CBET were also neglected.

The laser in this problem travelled up the density gradient, reflecting from the critical surface. For light travelling normally up a density gradient, the critical density is the location of the caustic, because the ray area,  $S(\tau)$ , is constant and therefore the amplitude diverges when  $\epsilon \rightarrow 0$ . The total field is given by the coherent sum between the incident and reflected sheets. Fig. 1.13.a shows the LPSE result compared to the field obtained from SOLAS without a cap on the caustic field; using the FL approach; and using the EI approach. The only difference between the SOLAS methods occurs in the caustic region<sup>12</sup>, shown more clearly in Fig. 1.13.b, which in this problem is for  $x > 13.64 \mu\text{m}$ . Failure to cap the caustic field leads to a divergent  $|E_z|$  at the critical surface, but both the EI and the FL methods show good agreement with LPSE. This justifies the default use of the simpler FL method for SOLAS CBET calculations.

Standard GO cannot capture the field beyond the critical surface because rays cannot propagate beyond  $\epsilon = 0$ . The evanescent field in this region can be reproduced in the complex-GO framework, which integrates rays which have complex-valued properties and can thus be evolved beyond the critical surface [27]. This, however, is not deemed to be necessary for accurate CBET modelling of OMEGA-scale direct-drive implosions.

#### 1.4.3.2 2-D Reflected Beam

A second field reconstruction test was also implemented in 2-D geometry. Unlike the previous 1-D test, the reconstructed field for this problem also depends on the divergence of neighbouring rays. This is because apart from normally incident rays at the centre of the beam,  $\mathbf{k} \nparallel \nabla n_e$ , so the area,  $S(\tau)$ , of the rays vary. Simulations at 1/64 and 1/4 scales are

<sup>12</sup>i.e. where equation 1.19 is satisfied.

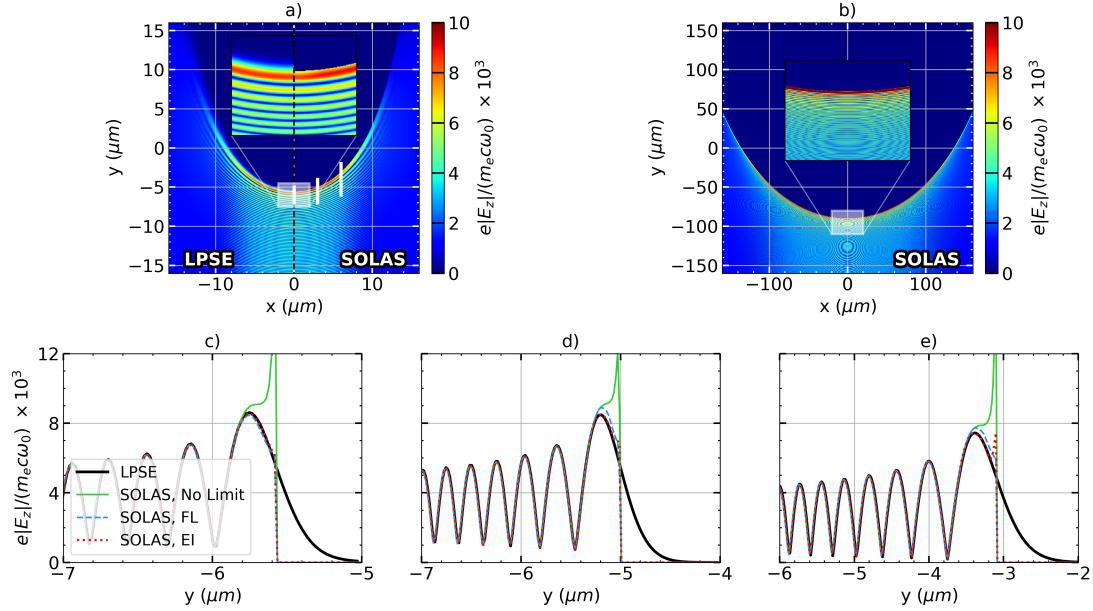


Figure 1.14: Results of the 2-D field reconstruction test. Panel a) plots the field from LPSE on the left side and the SOLAS field on the right for the 1/64 OMEGA scale simulation. Panel b) shows the SOLAS field from the 1/4 scale simulation which, when compared to a), demonstrates that as scale increases, the relative size of the caustic region decreases. Panels c), d) and e) are lineouts near the caustic region along  $y$  from the 1/64 scale simulations at  $x = 0, 3$  and  $6$   $\mu\text{m}$  respectively, the positions of which are indicated by the white lines on a). Note that the circular features in the field in panels a) and b) arise due to aliasing artefacts between the true solution scale, with the resolution of the image and/ or simulation grid.

presented here. The  $S = 1/64$  simulation bounds were  $x, y \in [-20, 20]$   $\mu\text{m}$  and resolution  $\Delta \sim 20$  nm. The  $S = 1/4$  simulation bounds were  $x, y \in [-200, 200]$   $\mu\text{m}$  and resolution  $\Delta \sim 100$  nm. For both scales, a single beam propagated parallel to  $+\hat{y}$  and was centred on  $x = 0$   $\mu\text{m}$ . The beam widths were  $\sigma = 8$  and  $115$   $\mu\text{m}$  for the 1/64 and 1/4 scales respectively. The standard uniform sampling procedure described in Sec. 1.3.3 was used with  $C_N = 2$ . CBET and Inv-Brem were again both neglected.

Fig. 1.14.a shows the LPSE and SOLAS, FL fields for the  $S = 1/64$  setup. Qualitative agreement is good between LPSE and the ray-based field in the sub-critical plasma. The field from the SOLAS FL  $S = 1/4$  simulation is plotted in Fig. 1.14.b, from which it can be seen that the caustic region, which has a characteristic width  $\sim \lambda_0$ , shrinks in relative size as the scale increases. This suggests that at increasing scale, the importance of accurate caustic modelling may decrease in importance. Plotted in Figs. 1.14.c, 1.14.d and 1.14.e are lineouts from the 1/64 simulations taken at  $x = 0, 3$  and  $6$   $\mu\text{m}$  respectively. They all include LPSE and SOLAS results with FL, EI caustic field capping along with no limiting. These plots demonstrate that caustic field limiting is necessary to compare favourably to the LPSE solution and that the FL approach slightly overestimates the caustic field compared to LPSE and the EI method. The discrepancy between the FL and EI results is again deemed sufficiently small to justify the default use of the simpler FL approach. To remind the reader, this choice is because the

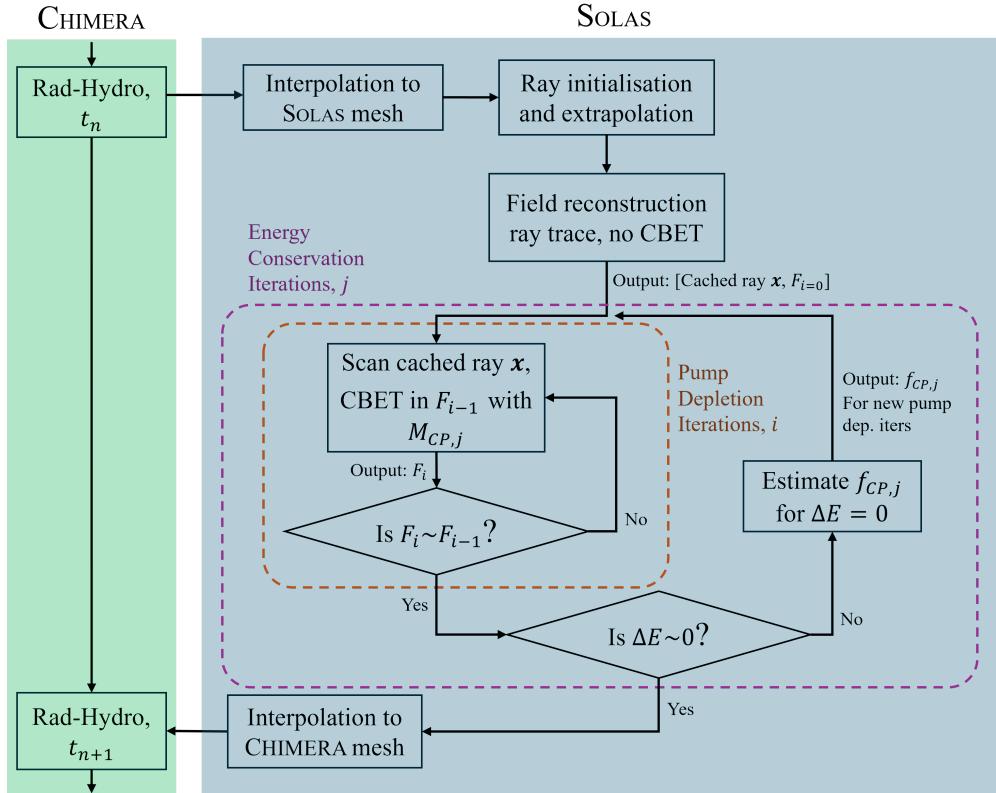


Figure 1.15: The SOLAS CBET model operation loop.  $F_i$  is shorthand for the total field from all sheets at pump depletion iteration  $i$  and  $f_{CP}$  is the modifier to the caustic pump multiplier, described in Sec. 1.5.6 of the  $j^{\text{th}}$  energy conservation iteration.  $\Delta E$  denotes the energy conservation error, which is the total incident power minus deposited power exiting the grid.

implementation of the EI method in SOLAS requires grid resolution  $\Delta \ll \lambda_0$ , the reason for which is described at the end of Sec. 1.4.2.

## 1.5 Ray-Based CBET Model

This section describes the implementation of the CBET model in SOLAS. Fig. 1.15 shows a flowchart which illustrates a broad overview of the operational loop of the CBET model when coupled to CHIMERA. The initial ‘field reconstruction’ ray-trace in the absence of CBET, follows the procedure described in Sec. 1.4 to obtain the field for every sheet on the SOLAS mesh. Ray-based CBET models, including the model in SOLAS, typically must perform repeated ray-traces through the same hydrodynamic profiles to account for pump depletion and optionally energy conservation.

The pump depletion loop, described in more detail in Sec. 1.5.5, is a fixed point iteration method to obtain the new CBET laser fields. This loop is necessary as each iteration produces a new background field which leads to a different CBET interaction, which converges when CBET has been fully accounted for. The resultant solution will not necessarily conserve energy if CBET occurs in the presence of laser caustics, due to errors in field reconstruction

and CBET scattering in these locations. To remedy this, additional energy conservation iterations can be performed, where a multiplier on the CBET gain in caustics can be varied to conserve laser energy. Although this is an ad-hoc correction, it is found to lead to better agreement with higher fidelity solver, as is explicitly demonstrated in Sec. 1.5.7.3.

CBET does not affect the trajectory of light, only its power and therefore repeating the calculations of ray trajectories is both computationally costly and unnecessary. Ray locations are therefore stored in a memory-efficient, linked-list to be re-used in both loops. It was found in SOLAS that caching the ray locations in this manner typically leads to an order of magnitude speed up compared to performing a new ray-trace for every iteration.

### 1.5.1 Power Change of Rays due to CBET

Ray-tracing calculations assume steady-state hydrodynamic profiles for rays propagating through the plasma and that the field of a ray propagates as a plane wave along the ray path [68]. The linear gain theory of SBS, which describes the steady-state energy change of locally plane wave light which passes through uniform plasma conditions, can therefore be used [18, 24]. ‘Linear’ in this context means that the plasma response to the driving fields is sufficiently small to be treated by a linear expansion. The uniform assumption here means that the plasma conditions must not vary significantly over the interaction time or length scales. For direct-drive conditions, hydrodynamic time scales are typically  $\sim 100\text{ps}$  and CBET saturates on the order of  $\sim 10\text{ps}$ . Ray steps are also limited to grid cell boundaries, which are smaller than hydrodynamic length-scales, so it is safe to assume a locally uniform plasma background when this model is applied only over a single ray step at a time.

The steady-state interaction between two parallel-polarised sheets ( $i, j$ ) over a path length  $d\tau$  can be written in terms of the field magnitude of the sheet ( $|E_{i,j}|$ ) or equivalently ray power ( $P_{i,j} = A_{i,j}|E_{i,j}|^2$ ),

$$\begin{aligned}\frac{d|E_i|^2}{d\tau} &= (-\kappa_{IB} + \gamma_{ij}|E_j|^2)|E_i|^2, \\ \frac{dP_i}{d\tau} &= (-\kappa_{IB} + \gamma_{ij}A_j P_j)P_i,\end{aligned}\tag{1.21}$$

where  $\kappa_{IB}$  is the Inv-Brem absorption kernel,  $A_j$  is the amplitude of sheet  $j$  and  $\gamma_{ij}$  is the CBET gain of sheet  $i$  from the interaction with sheet  $j$ . The sheet/ ray which changes energy in a given interaction (here  $i$ ), is often called the *probe* while the other beam is termed the *pump*. The gain is a function of both the plasma conditions and the local field profiles for both sheets in the interaction, so a gain must be calculated between every pair of sheets at every location where both are present. Going from the first to the second equation assumes that  $dA_i/d\tau = 0$ , which is consistent with the assumption from the linear gain theory that the fields are approximately uniform over a small step. Different models exist for the gain, which either use a fluid or kinetic treatment for the plasma IAW response.

When multiple different probe beams are present, the total CBET interaction is treated simply as the sum of all interactions from different sheets,

$$\frac{dP_i}{d\tau} = (-\kappa_{IB} + \kappa_{CBET})P_i,\tag{1.22}$$

where the CBET kernel has been defined  $\kappa_{\text{CBET}} = \sum_{j \neq i} \gamma_{ij} |E_j|^2$ . This summation assumes that the fields from different sheets are uncorrelated. For direct-drive conditions, this assumption is broadly correct, apart from when a probe sheet undergoes a CBET interaction in the caustic region of the probe. In this case, the *Coherent Caustic* correction is applied, which is described in Sec. 1.5.3.

Note that when computing both Inv-Brem and CBET, the partition of power lost to each interaction can be computed as,

$$\begin{aligned}\Delta P_{i,\text{IB}} &= -\langle P_i \rangle \frac{\kappa_{\text{IB}}}{\kappa_{\text{CBET}} - \kappa_{\text{IB}}}, \\ \Delta P_{i,\text{CBET}} &= \langle P_i \rangle \frac{\kappa_{\text{CBET}}}{\kappa_{\text{CBET}} - \kappa_{\text{IB}}}, \\ \langle P_i \rangle &= P_i + P_i (\kappa_{\text{CBET}} - \kappa_{\text{IB}}) d\tau / 2,\end{aligned}\quad (1.23)$$

where  $\Delta P_{i,\text{IB}}$  and  $\Delta P_{i,\text{CBET}}$  are the powers lost to Inv-Brem and CBET respectively and  $\langle P_i \rangle$  is the average power of the ray over the step,  $d\tau$ . This can be derived by solving Eq. 1.21 for  $P_i$  and then taking the limit  $d\tau \rightarrow 0$  [31]. The deposition,  $\Delta P_{i,\text{IB}}$ , can then be interpolated to the grid as an electron energy source as is described in Sec. 1.3.4.

The two formulations of the gain are presented below. Broadly, the gain describes the plasma response to the 2 driving fields and therefore dictates the energy transfer between them. They differ in that the fluid formulation arises by deriving a gain from linearised fluid equations for the plasma response, whereas the kinetic formulation is obtained from a linearisation of the Vlasov equation. The kinetic gain is more complete and should be used for comparison to experimental observables, whereas the fluid gain is useful for comparison with LPSE, which directly solves the linearised fluid equations.

**Fluid CBET Gain** The linear fluid gain, derived by Randall *et al.* in Ref. [24] is,

$$\begin{aligned}\gamma_{ij} &= \frac{n_e e}{4m_e c \omega_i} \frac{1}{T_e (1 + 3T_i/ZT_e)} \frac{R(\eta_{ij})}{v_{ia}}, \\ R(\eta_{ij}) &= \frac{(v_{ia}/\omega_s)^2 \eta_{ij}}{(\eta^2 - 1)^2 + (v_{ia}/\omega_s)^2 \eta_{ij}^2}, \\ \eta_{ij} &= \frac{(\omega_j - \omega_i) - \mathbf{k}_s \cdot \mathbf{u}}{\omega_s},\end{aligned}\quad (1.24)$$

where  $T_e$  and  $T_i$  are the electron and ion temperatures in eV respectively,  $\omega_i$  and  $\mathbf{k}_i$  are the frequency and wavevector of sheet  $i$  respectively,  $Z$  is the ionisation state,  $\mathbf{u}$  is the fluid velocity,  $\omega_s = |\mathbf{k}_s|c_s$  is the IAW frequency,  $\mathbf{k}_s = \mathbf{k}_j - \mathbf{k}_i$  is the IAW wavevector,  $c_s = \sqrt{e(ZT_e + 3T_i)/m_i}$  is the sound speed and  $v_{ia}$  is the IAW damping rate. This formulation is useful for comparison with LPSE, which directly solves the linearised fluid equations and therefore in situations where the GO assumptions are valid, the two methods should give similar results. However, it is generally not considered to be a good choice of model to compare to experiment. The damping rate,  $v_{ia}$  does not have an analytic formula for arbitrary plasma conditions and therefore must be prescribed. Additionally, Eq. 1.24 assumes an average ion treatment of the

species in the plasma if multiple ion species are present. In reality, both fast and slow modes of IAW can be driven in a 2-species plasma [69], which can independently lead to CBET scattering. This effect cannot be captured using the fluid gain, where the plasma is assumed to be made of a single ion species with a number density averaged ion mass. Codes that use this formulation typically use a damping value  $v_{ia}/\omega_s = 0.2$  for CH plasmas, which is the value used for all fluid gain simulations in this chapter.

**Kinetic CBET Gain** The linear kinetic CBET gain, described by Michel in Ref. [70] is,

$$\begin{aligned}\gamma_{ij} &= \frac{e^2 |\mathbf{k}_s|^2}{4m_e^2 c \omega_i^3} \text{Im}(K_{ij}), \\ K_{ij} &= \frac{\chi_e(1 + \chi_i)}{(1 + \chi_e + \chi_i)}, \\ \chi_e &= \frac{-1}{2|\mathbf{k}_s|^2 \lambda_{De}^2} Z' \left( \frac{(\omega_j - \omega_i) - \mathbf{k}_s \cdot \mathbf{u}}{\sqrt{2} |\mathbf{k}_s| v_{Te}} \right), \\ \chi_i &= \frac{-1}{2|\mathbf{k}_s|^2 \lambda_{De}^2} \frac{T_e}{\langle Z \rangle T_i} \sum_{\alpha}^{\text{species}} f_{\alpha} Z_{\alpha}^2 Z' \left( \frac{(\omega_j - \omega_i) - \mathbf{k}_s \cdot \mathbf{u}}{\sqrt{2} |\mathbf{k}_s| v_{T\alpha}} \right),\end{aligned}\quad (1.25)$$

where  $\chi_e$  and  $\chi_i$  are the electron and ion susceptibilities respectively,  $f_{\alpha}$  and  $Z_{\alpha}$  are the number fraction and ionisation of ion species  $\alpha$  respectively,  $Z'$  is the derivative of the plasma dispersion function,  $v_{Tn} = \sqrt{eT_n/m_n}$  is the thermal velocity of species  $n$  with temperature  $T_n$  in eV and mass  $m_n$ ,  $\lambda_{De} = \sqrt{\epsilon_0 e T_e / n_e e^2}$  is the electron Debye length and  $\langle Z \rangle = \sum_{\alpha} f_{\alpha} Z_{\alpha}$  is the average ionisation. Note that the susceptibilities  $\chi_{i,e}$  are complex. In order to compute  $Z'$  in SOLAS, the CALGO library is used to numerically compute the value of the *Faddeeva function*,  $w(x)$  [71]. This is then related to the plasma dispersion function and its derivative by,

$$\begin{aligned}Z(x) &= i\sqrt{\pi}w(x), \\ Z'(x) &= -2[1 + xZ(x)],\end{aligned}\quad (1.26)$$

where  $i = \sqrt{-1}$  and  $w(x)$ ,  $Z(x)$  and  $Z'(x)$  are all complex numbers [72]. Unlike the fluid gain, this theory correctly captures the resonance and has no effective free parameters, making it the judicious choice for predictive simulations. The plasma dispersion function is however relatively slow to numerically evaluate and dominates computational runtime if evaluated every ray step. SOLAS therefore calculates  $\gamma_{ij}$  between each sheet in every grid cell at the end of the field reconstruction ray-trace (from Fig. 1.15) and then interpolates this pre-calculated value onto the ray locations for the pump-depletion iterations. This reduces computational runtimes by orders of magnitude for many beam direct-drive simulations, but has an associated memory overhead, which is discussed in more detail in Sec. 1.5.4. All simulations in this thesis use the kinetic CBET gain unless explicitly stated otherwise.

**Random Polarisation Correction** The gains calculated above all assume that the interacting fields have parallel polarisations. For many laser systems, beams may have arbitrary or random polarisations, such as at the OMEGA laser facility where Distributed Polarization Ro-

tators (DPRs) separate each beam into 2 orthogonally polarised sub-beams which overlap to effectively have one spot with random polarisation [73]. If a simulation is conducted with multiple beams of random polarisation, a *Polarisation Smoothing* multiplier,  $M_{PS}$ , must be applied to the value of  $\gamma_{12}$  which accounts for random polarisation angles between the field from different sheets,

$$M_{PS} = \frac{1}{4}(1 + \cos^2 \theta), \quad (1.27)$$

where  $\theta$  is the angle between the wavevectors of the interacting sheets.

MAYBE TRANSFER BELOW TO THEORY SECTION!

In reality, the DPRs on OMEGA do not actually create a single, mixed polarisation laser spot, but 2 slightly offset sub-beams on the target. CBET models which track polarisation or rays have been developed which account for the true polarisation angle between 2 sheets [74–77]. These models have demonstrated that polarised CBET on OMEGA creates a mode-1 which is always in the same direction. This persistent mode-1 was experimentally observed from neutron time of flight diagnostics before its origin was understood [78].

### 1.5.2 Caustic Gain Truncation

For rays propagating through SOLAS cells, nearest neighbour interpolation is used to interpolate field quantities to the rays. This is generally an adequate approximation for field magnitudes, frequencies and wavevectors, which vary over greater length scales than typical cell resolutions  $\mathcal{O}(1\mu\text{m})$ , apart from in the vicinity of a caustic [37]. Near caustics, rays turn sharply and fields have a sharp cut-off, dropping to zero sharply, as be seen in the lineouts from Fig. 1.14. The sharp turning of rays mean that  $\mathbf{k}$  can vary rapidly within a cell. Although SOLAS does have variable radial resolution, as described in Sec. 1.4, this resolution is limited to the minimum radial cell size from the hydrodynamic grid, which is not dictated from laser parameters and cells are typically larger than the caustic region width  $\mathcal{O}(\lambda_0)$ . The issue of  $\mathbf{k}$  variation is not addressed in SOLAS, but linear interpolation or a truly adaptive mesh, separate from the CHIMERA grid could be implemented to address this.

The field cut-off is an issue however, as it can significantly affect energy conservation in ray-based CBET models. CGT is an algorithm that effectively improves the resolution of the grid near laser caustics [36]. This method separates cells into *lit* and *unlit* regions. If rays travel in the *unlit* region, then the CBET gain is not applied. For example, Fig. 1.16.a shows a beam propagating up a linear density gradient, separated by the caustic into incident and reflected sheets. Fig. 1.16.b shows a zoomed in view of in the vicinity of the caustic of the incident sheet ray locations, plotted as grey dots. If a ray from another beam were present with  $14.3 \lesssim y < 15 \mu\text{m}$ , nearest neighbour interpolation would lead to the ray undergoing a CBET interaction in this *unlit* region, where no rays from the pump sheet are present. Field magnitudes in this caustic region are typically large, so even though this is a small region of space, it can lead to significant errors, especially at lower resolutions.

The CGT algorithm described by Follett in Ref. [36] defines the *lit* region for each sheet by creating a mesh of the caustic locations from every ray on the sheet and only applying CBET gains in on the side of this mesh that rays traversed. A different algorithm has been devel-

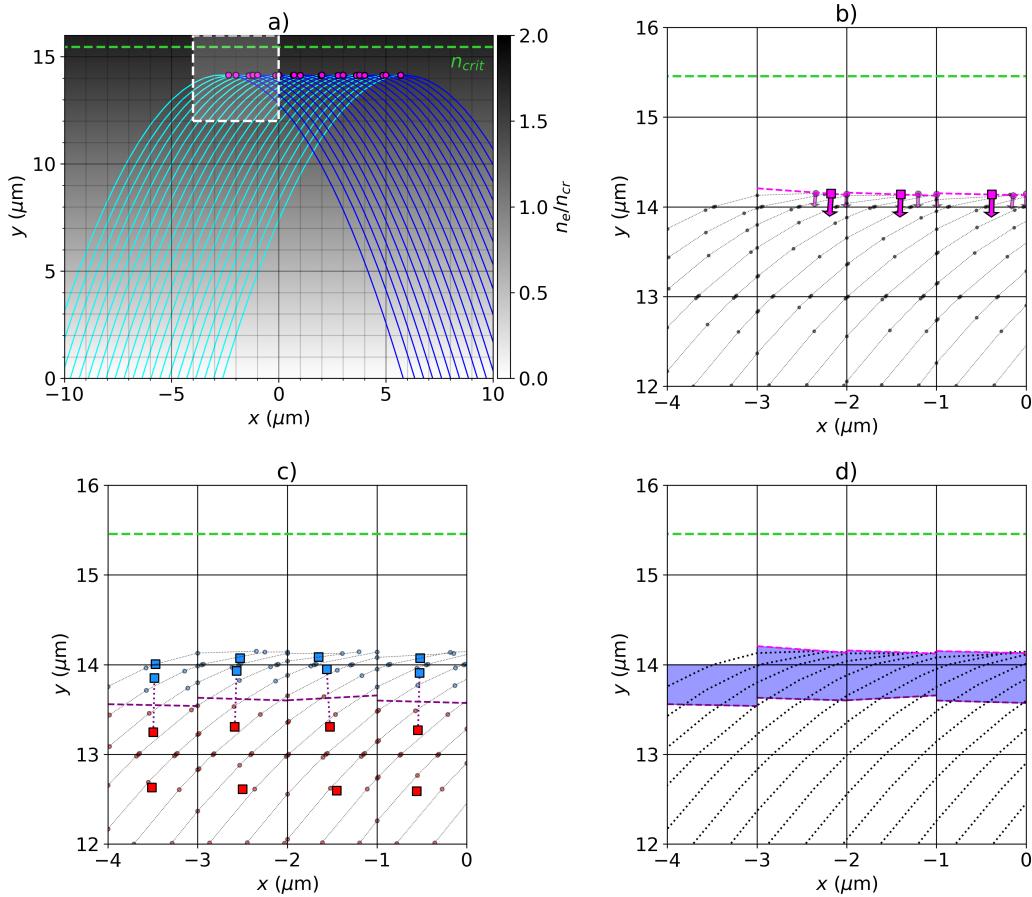


Figure 1.16: Illustration of the CGT and caustic region identification algorithms. Panel a) shows a laser, separated into incident and reflected sheets by a caustic, propagating up and reflecting in a linear density ramp. CGT is illustrated in b) where a cell is separated into a *lit* and *unlit* region by the reconstructed *caustic plane* (dashed magenta line). Panels c) and d) illustrate the method used to identify the *caustic region* (shaded blue), where the coherent caustic modifier is applied. c) shows that cells are geometrically separated into *capped* (blue squares) and *uncapped* (red squares) regions and the coherent caustic modifier to the gain is applied in this capped, caustic-region.

oped for SOLAS, which geometrically splits each cell into *lit* and *unlit* regions by a *caustic plane* if at least one ray caustic occurs in the cell. The *caustic plane*<sup>13</sup> is defined by a point  $\mathbf{x}_{caus}$  and normal  $\hat{\mathbf{n}}_{caus}$ . The location,  $\mathbf{x}_{caus}$ , is simply the average of all ray caustic locations from the sheet present in the cell, for example shown in Fig. 1.16.b by the small pink circles for each ray caustic and the larger pink square for the cell averaged location. The normal is estimated by the average over all ray caustics in the cell of the ray tangent vector just after the caustic  $\hat{\mathbf{t}}_{After}$ , minus the velocity just before  $\hat{\mathbf{t}}_{Before}$ ,

$$\hat{\mathbf{n}}_{caus} = \left\langle \frac{\hat{\mathbf{t}}_{After} - \hat{\mathbf{t}}_{Before}}{|\hat{\mathbf{t}}_{After} - \hat{\mathbf{t}}_{Before}|} \right\rangle, \quad (1.28)$$

<sup>13</sup>In cylindrical and spherical simulation geometries, the ‘planes’ are defined in the native coordinate system, which creates manifolds that better reflect the assumed geometry of the problem.

such that the normal points to the lit side. This is plotted in Fig. 1.16.b as small pink arrows for each ray caustic normal and the large arrows in each cell for the cell averaged value. The planes defined by these points and normals for each cell that contains a sheet caustic is shown as a pink dashed line. If a probe ray at location  $\mathbf{x}_{\text{probe}}$  is propagating in a cell where there is a pump sheet caustic, it will only experience a CBET interaction if  $\hat{\mathbf{n}}_{\text{caus}} \cdot (\mathbf{x}_{\text{probe}} - \mathbf{x}_{\text{caus}}) > 0$ . Rays from other sheets also have their steps limited by the caustic plane.

This is a slightly lower fidelity implementation of the more complete CGT model from Ref. [36]. The model implemented by Follett can represent caustics which, within a grid cell, are not locally well approximated by planes. The SOLAS model is however simpler to implement and also acts to effectively increase the resolution of the cells near caustics. The caustic test problem presented in Sec. 1.5.7.2 illustrates the effectiveness of the SOLAS-CGT model at reducing energy conservation errors for caustic CBET interactions where the grid cells are significantly larger than the caustic region.

### 1.5.3 Coherent Caustic Correction and Caustic Region Identification

In Sec. 1.5.1, in order to obtain Eq. 1.22, it was assumed that when a ray undergoes CBET interactions with multiple pump sheets, the field from the pump sheets can be treated independently. This is equivalent to assuming that over a ray propagation path,  $d\tau$ , the fields from the sheets do not add coherently,

$$\int d\tau \sum_j^{\text{sheets}} |E_j|^2 \approx \int d\tau \left| \sum_j^{\text{sheets}} E_j \right|^2, \quad (1.29)$$

where  $E_j$  is the field from the  $j^{\text{th}}$  sheet. Eq. 1.29 is valid for all regions of space where the phase from different sheets are uncorrelated. For direct-drive configurations, this assumption holds everywhere apart from the caustic region of the pump sheet, where fields add coherently. Thus, the coherent sum of fields in the caustic region is larger than the incoherent sum, so without correcting for this, CBET scattering through a pump beam in its caustic region will be underestimated.

The ratio of the coherent and incoherent sums of incident ( $E_{\text{inc}}$ ) and reflected ( $E_{\text{refl}}$ ) fields of the pump sheet in its caustic region is called the *coherent caustic multiplier*,

$$M_{\text{CP}} = \frac{|E_{\text{inc}}|e^{i\phi_{\text{inc}}} + |E_{\text{refl}}|e^{i\phi_{\text{refl}}}}{|E_{\text{inc}}|^2 + |E_{\text{refl}}|^2}. \quad (1.30)$$

By making the same assumption as used to obtain Eq. 1.18, that  $|E_{\text{inc}}| \sim |E_{\text{refl}}|$ , this simplifies to,

$$M_{\text{CP}} = 1 + \sin(\phi_{\text{refl}} - \phi_{\text{inc}}), \quad (1.31)$$

which should be applied in the caustic region, (defined in Eq. 1.19), as the region where  $|\varphi_1 - \varphi_2| \leq \lambda_0/2$ . As stated in the discussion in Sec. 1.4.2, grid cells are typically larger than the size of the caustic region,  $\mathcal{O}(\lambda_0)$ , and SOLAS does not include linear interpolation of field quantities. Therefore, Eq. 1.31 cannot robustly be evaluated at the relevant scales. The caus-

tic region is identified by a geometric algorithm, similar to the *caustic plane* method discussed in Sec. 1.5.2 and the ‘average’ value of the coherent caustic multiplier is used,

$$M_{\text{CP}} = 1 + 2/\pi, \quad (1.32)$$

when a ray undergoes CBET with a pump sheet in its caustic region. This coherent caustic correction increases the CBET scattering and is an important addition to make sure that ray-based CBET models are energy conserving.

The geometric approach that was developed to estimate when a ray is in the caustic region of another beam, ( $|\varphi_1 - \varphi_2| \leq \lambda_0/2$ ), is described here. Similar to the *caustic plane* method from Sec. 1.5.2, this algorithm works by finding planes that divide cells and thus effectively increase their resolution, improving the accuracy of the nearest neighbour interpolation. The key assumptions of the method are that the boundary of the caustic region can be treated locally in grid cells as a plane and that the caustic region is approximately where ray amplitudes are greater than the FL capped amplitude,  $A_{\text{ray}} > A_{\text{FL}}$ . Fig. 1.16.c plots the trajectory of incident sheet rays from Fig. 1.16.a, with the locations of capped and uncapped rays plotted as small transparent red and blue dots respectively. The average value of each of these locations is represented by the larger blue ( $\langle \mathbf{x}_{\text{cap}} \rangle$ ) and red squares ( $\langle \mathbf{x}_{\text{uncap}} \rangle$ ) respectively, which are stored (only where present), in each SOLAS grid cell. The region of the cell occupied by these capped (blue) ray locations is assumed to be equivalent to the caustic region. The validity of the assumption was tested by conducting high resolution simulations of the 2-D reflected beam field reconstruction test in Sec. 1.4.3 to accurately reconstruct the phase and amplitudes of incident and reflected sheets on a high resolution mesh. It was found that the assumption is adequate, although it typically slightly overestimates the width of the caustic region by  $\mathcal{O}(10\%)$ . This is not considered to be a significant issue, because energy conservation iterations are employed which minimise errors by varying  $M_{\text{CP}}$  over multiple iterations. Therefore, errors in the initial iteration are compensated by this minimisation.

To geometrically estimate the caustic region in cells which contain both capped (blue) and uncapped (red) ray locations, a *caustic region plane* is defined to separate these regions, defined by a point,  $\mathbf{x}_{\text{CR}}$ , and normal,  $\hat{\mathbf{n}}_{\text{CR}}$ ,

$$\begin{aligned} \mathbf{x}_{\text{CR}} &= \frac{\langle \mathbf{x}_{\text{cap}} \rangle + \langle \mathbf{x}_{\text{uncap}} \rangle}{2}, \\ \hat{\mathbf{n}}_{\text{CR}} &= \frac{\langle \mathbf{x}_{\text{cap}} \rangle - \langle \mathbf{x}_{\text{uncap}} \rangle}{|\langle \mathbf{x}_{\text{cap}} \rangle - \langle \mathbf{x}_{\text{uncap}} \rangle|}, \end{aligned} \quad (1.33)$$

where  $\hat{\mathbf{n}}_{\text{CR}}$  is defined such that it points toward the caustic region. For a probe ray propagating through a mesh which contains a pump field, the ray is inside the caustic region if  $\hat{\mathbf{n}}_{\text{CR}} \cdot (\mathbf{x}_{\text{probe}} - \mathbf{x}_{\text{CR}}) > 0$ , under which circumstances, the pump-probe CBET gain is multiplied by the average value of  $M_{\text{CP}}$  defined in 1.32. This is illustrated in Fig. 1.16.c by the purple dashed lines. If a cell contains both a *caustic region plane* and a *caustic plane* (from Sec. 1.5.2), then the ray must be inside the caustic and on the lit side of the caustic to undergo

CBET with  $M_{\text{CP}}$  applied. Fig. 1.16.d plots the caustic region, as obtained by the algorithm, as a transparent blue shaded area.

#### 1.5.4 Dynamic Memory for Storing Fields and CBET Gains

CBET gains must be calculated between every pair of sheets that cross at a given location. For  $N_b$  beams, each with 2 possible sheets, discretised on a grid with  $N_x$  cells in each dimension, a total of  $N_f \sim 2N_x^3 N_b$  *field data points* must be stored. Each of these data points must store the field magnitude, phase, wavevector and information used to evaluate CGT and the coherent caustic correction where necessary, as described in Sec. 1.5.2 and 1.5.3 respectively. This amounts to large computational memory requirements to store the fields. Additionally, SOLAS stores the CBET gains between each laser sheet to improve the code execution time. A separate gain exists between each sheet and therefore the memory cost to store this information scales as  $N_g \sim 2N_f^3 N_b^2$ .

$N_g$  is a single number and  $N_f$  represents all field information, so for simulations with small numbers of beams,  $N_f$  is the dominant memory cost, whereas for large number of overlapping beams such as direct-drive simulations,  $N_g$  dominates. It is therefore important to use a memory efficient method to store this data so that large scale, 3-D simulations can be completed. SOLAS utilises pointers in each grid cell to dynamically allocate the field data, such that only where a sheet is present is its corresponding field magnitude, phase *etc.* stored. Additionally, gains between sheets are an attribute of the field data and are stored as single-precision floats, rather than the native SOLAS double precision, which also reduces memory costs for large simulations. These measures significantly reduce memory overhead for multi-dimensional CBET simulations, however high resolution 3-D simulations are still extremely memory intensive, which is an issue also experienced by IFRIIT, the other 3-D CBET code coupled to hydrodynamics.

#### 1.5.5 Pump Depletion Iterations

The field reconstruction ray-trace (the initial ray-trace performed in the absence of CBET) returns the field from each sheet  $|E_j^{i=0}|(\mathbf{x}, \mathbf{k}, \omega, \phi)$ , where  $j$  is the sheet index and  $i = 0$  indicates the field reconstruction ray-trace. This field is discretised on the SOLAS mesh using nearest neighbour interpolation, so that a ray passing through a cell updates the field in that cell. Rays can then be re-traced through this mesh and undergo changes in power due to CBET interactions with the fields from other sheets. The solution obtained from a single ray-trace with CBET is not the true solution however, because the field that the rays see throughout the CBET ray-trace is not dynamically updated by the change in ray power due to CBET as they propagate. This necessitates iteratively solving the ray-trace to account for CBET in *pump depletion iterations*, whereby for iteration  $i$ , the rays from sheet  $j$  propagate through the mesh and undergo CBET through  $|E_{\text{sheet} \neq j}^{i-1}|$ , while building up a new field,  $|\tilde{E}_j^i|$ . The new *build-up* fields,  $|\tilde{E}_j^i|$ , are introduced for the next iteration. This loop is repeated until convergence of the field has been achieved. Convergence is tested each iteration ( $i > 0$ ) against the total power of rays  $\tilde{P}_{j,n}^i$ , from each sheet  $j$  travelling through each cell, explicitly defined by

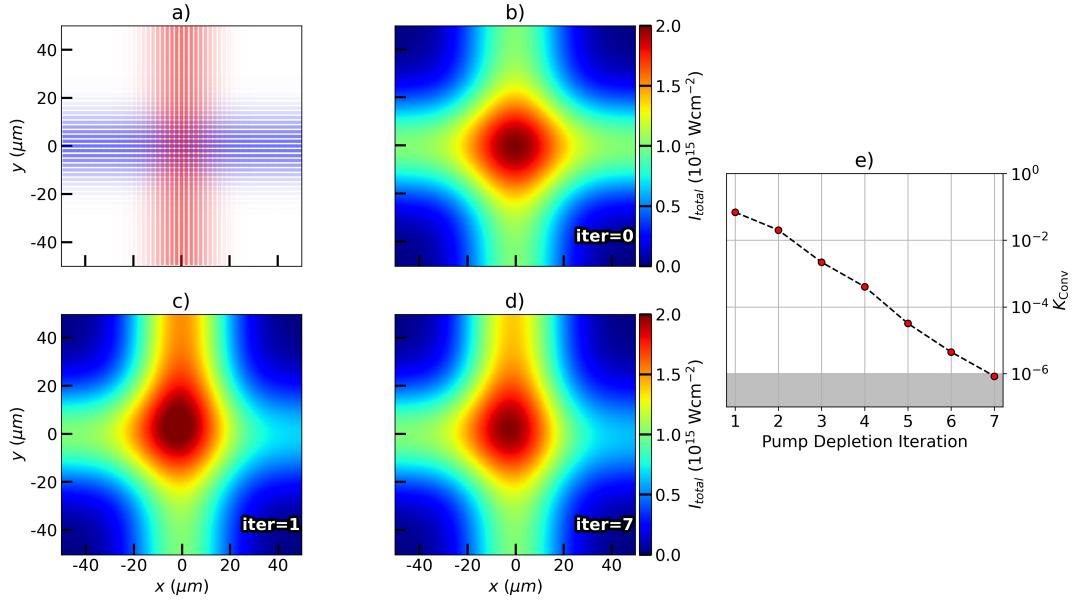


Figure 1.17: Results of successive pump depletion iterations for an illustrative simulation of two beams crossing and undergoing a CBET resonance in a uniform plasma background. Each beams' rays from the field reconstruction ray-trace are plotted in a), where more opaque rays indicate a higher power. Panel b) plots the reconstructed intensity from this ray-trace, where  $i = 0$  indicates the field reconstruction ray-trace and therefore no CBET occurs. Panels c) and d) plot the intensity reconstructed from the 1<sup>st</sup> and 7<sup>th</sup> pump depletion iterations respectively, where the effect of transfer of powers between beams due to CBET is evident. Panel e) plots the convergence parameter from the simulation over successive pump depletion iterations, with convergence defined here as  $K_{\text{Conv},0} = 10^{-6}$ .

the convergence parameter,

$$K_{\text{Conv}}^i = \max_{\substack{j \in \text{sheets} \\ n \in \text{cells}}} \left| \left( \tilde{P}_{j,n}^i / \tilde{P}_{j,n}^{i-1} \right) - 1 \right| < K_{\text{Conv},0}, \quad (1.34)$$

where  $K_{\text{Conv},0}$  is a user defined parameter, typically set to  $10^{-5}$  unless stated otherwise.

This procedure is illustrated in Fig. 1.17 for a simple, illustrative problem of two beams crossing in a uniform, stationary plasma background with a frequency difference between the beams tuned to give a CBET resonance. The effect of Inv-Brem deposition has been neglected in this simulation. Fig. 1.17.a shows the ray trajectories for the two crossing beams and Fig. 1.17.b plots the intensity from the field reconstruction iteration, without CBET. The intensities after one and seven pump depletion iterations are plotted in Fig. 1.17.c and Fig. 1.17.d respectively, which show that a single re-trace through the no-CBET fields does not return the converged solution. Fig. 1.17.e plots the convergence parameter after each iteration, illustrating that seven iterations were required to obtain the converged solution, plotted in Fig. 1.17.d.

For many beam simulations with complicated beam trajectories and many overlapping crossings such as direct-drive calculations, the problem is asymptotically unstable. This

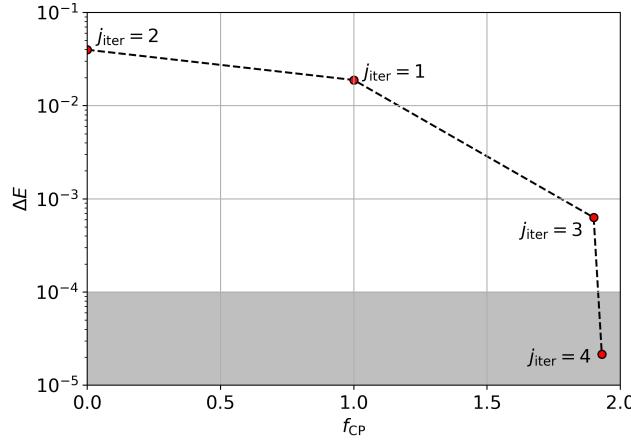


Figure 1.18: Energy conservation iterations for a two-beam caustic test problem, which is presented in Sec. 1.5.7.2. 4 iterations were required to achieve energy conservation, which was defined to be  $\Delta E < 1 \times 10^{-4}$ . The order of the iterations is indicated by the text next to each point.

means that for iteration  $i$ , allowing rays to undergo CBET in the full  $|\tilde{E}_{sheet \neq j}^{i-1}|$  causes unbounded gains on ray powers leading to divergent results for  $|\tilde{E}_j^i|$ . Numerical damping is therefore applied for these calculations to stabilise the solution. The actual fields that a ray will see for the subsequent iteration,  $i + 1$ , are composed of the weighted sum of the two *build-up* fields,

$$|E_j^i| = f_{damp} |\tilde{E}_j^i| + (1 - f_{damp}) |\tilde{E}_j^{i-1}|, \quad (1.35)$$

where  $f_{damp}$  is a user-defined damping value, which is by default set to a (relatively conservative) value of 0.75, to ensure that the solver does not diverge for complex problems.

### 1.5.6 Energy Conservation Iterations

The converged field solution from the pump depletion iterations returns the steady-state field and deposition profiles, accounting for CBET, given a field reconstruction algorithm, interpolation scheme *etc*. If there are errors in, for example, the field reconstruction algorithm, then the pump depletion iterations will return a steady-state solution, however this is not guaranteed to be accurate, *i.e.* the same as the result from a higher fidelity solver. The main cause of errors in field reconstruction is the treatment of laser caustics, as demonstrated for example in Fig. 1.14. This figure demonstrates that the SOLAS field reconstruction algorithm obtains essentially perfect agreement with LPSE everywhere apart from in the vicinity of the caustic. It can therefore be assumed that any errors in the CBET profiles, returned from the pump depletion iterations, occur due to errors in the field reconstruction and CBET scattering in the caustic region. The energy conservation error,  $\Delta E$ , is a good proxy for the accuracy of the fields obtained from CBET, as in the absence of laser caustics, energy is conserved in ray-based algorithms [37]. This is defined as,

$$\Delta E = \frac{P_{\text{dep}} + P_{\text{out}} - P_{\text{inc}}}{P_{\text{inc}}}, \quad (1.36)$$

where  $P_{\text{dep}}$ ,  $P_{\text{out}}$  and  $P_{\text{inc}}$  are the power deposited by Inv-Brem, the ray-power leaving the domain and the incident power respectively. If this value is not small, it suggests that caustics are not accurately treated in the CBET scattering.

As in the BEAMCROSSER code, the assumption in SOLAS is made that minimising  $\Delta E$  by altering the CBET gains in the caustic region will bring the result into better agreement with the true solution [37].  $M_{\text{CP}}$  is modified to include a new *caustic pump multiplier*,  $f_{\text{CP}}$ , which is applied to probe ray CBET gains, when they are in the caustic region of the pump sheet,

$$\tilde{M}_{\text{CP}} = \begin{cases} 1 + f_{\text{CP}} \left( \frac{2}{\pi} \right), & \text{if in caustic region} \\ 1, & \text{otherwise} \end{cases} \quad (1.37)$$

where the same, geometric identification for the pump sheet caustic region is used as in Sec. 1.5.3.  $f_{\text{CP}} = 1$  initially and is varied between  $f_{\text{CP}} \in [-\pi/2, 2]$  over several *energy conservation* iterations<sup>14</sup> to minimise  $\Delta E$ . The lower bound on  $f_{\text{CP}}$  is to prevent ‘negative scattering’ in the caustic region, where the sign of the gain flips. The upper bound was selected by experimentation to prevent excessively large gains which result in the exponential power gain of rays blowing up to infinity.

An implementation of the Secant method is used to estimate the next value of  $f_{\text{CP}}$ . The first and second energy conservation iterations in a simulation use  $f_{\text{CP}} = 1$  and 0, to obtain a gradient with which to estimate values of the multiplier. The Secant method was chosen for its simplicity and the fact that bracketing values of the root are not required. An example of the minimisation results are shown in Fig. 1.18. These results are from a two-beam caustic test problem, presented in Sec. 1.5.7.2, but the specifics of the problem are not important for the discussion here, simply that CBET in the presence of a caustic occurs. Therefore, a caustic region exists inside which  $f_{\text{CP}}$  will lead to modified gains which reduce energy conservation errors. The curve of  $\Delta E(f_{\text{CP}})$  is simple and monotonic, which is observed for all simulations conducted with SOLAS, regardless of the complexity of the laser and target configuration. This means a more complex algorithm than the Secant method is not required for minimisation. When the CBET model runs in-line with the hydrodynamics, the value of  $f_{\text{CP}}$  from the last timestep that CBET was calculated is used, which means that energy conservation iterations are often not necessary. This is shown explicitly in Fig. 1.24.c, which shows the results of a 1-D CHIMERA-SOLAS simulation of OMEGA shot 89224. The bottom panel in the figure shows that energy conservation is frequently achieved by simply re-using the value of  $f_{\text{CP}}$  from the previous timestep.

To summarise, the total CBET gain that a probe ray ( $i$ ) experiences in a pump sheet ( $j$ ) is,

$$\tilde{\gamma}_{ij} = \gamma_{ij} M_{\text{PS}} \tilde{M}_{\text{CP}}, \quad (1.38)$$

where  $\gamma_{ij}$  is the value defined in Sec. 1.5.1 (depending on if using the kinetic or fluid gain model), the modified coherent caustic multiplier  $\tilde{M}_{\text{CP}}$  is defined in Eq. 1.37 and the polarisation smoothing multiplier,  $M_{\text{PS}}$ , defined in Eq. 1.27, is applied if the beams are assumed to

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<sup>14</sup>Each energy conservation iteration contains a new set of pump depletion iterations, as is shown in Fig. 1.15.

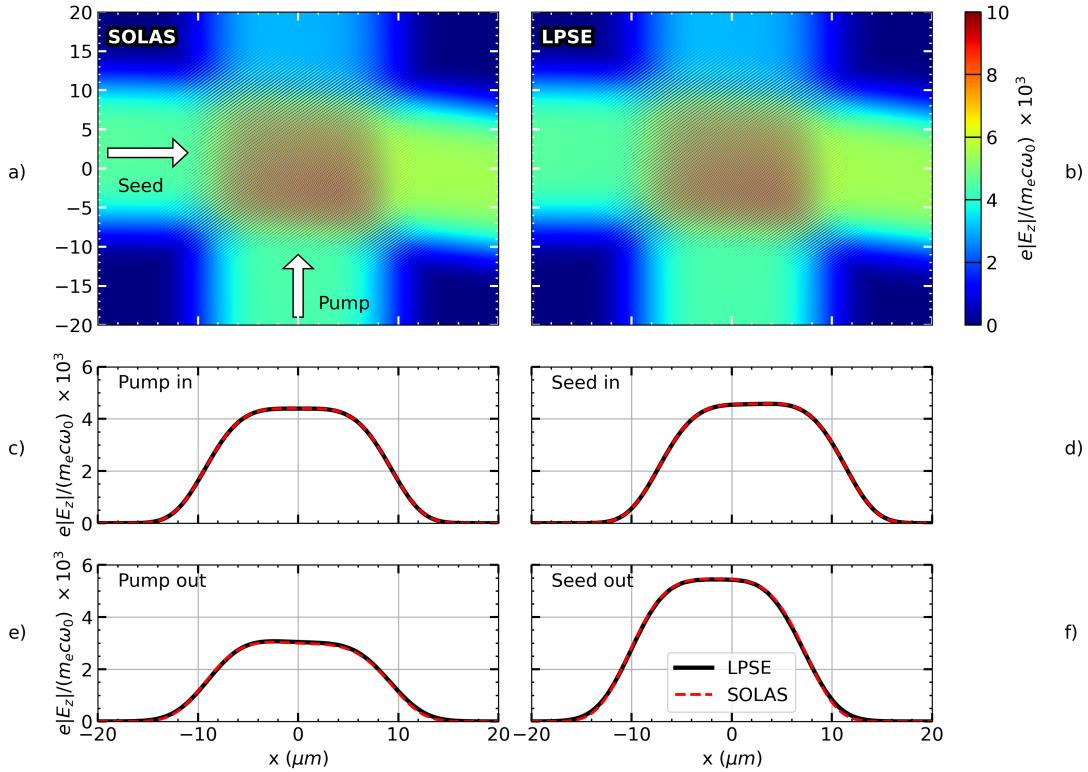


Figure 1.19: Results of the 2-D CBET test in the absence of laser caustics. Panels a) and b) show the electric field from the SOLAS and LPSE simulations respectively, after the pump depletion iterations have occurred. Panels c) and d) plot lineouts from the inbound fields of the pump and seed beams respectively for both SOLAS and LPSE. e) and f) plot the same for the outbound fields.

be randomly rather than parallel-polarised.

### 1.5.7 CBET Validation

This section presents results from several CBET test problems where the results from SOLAS are compared to the wave based solver LPSE. Because LPSE is a wave-solver coupled to a linearised fluid response, the fluid CBET gain was used for all simulations in this section. Beams in the simulations all had a vacuum wavelength,  $\lambda_0 = 0.351$  nm and a super-Gaussian spot profile, which was defined in Eq. 1.1. For consistency with LPSE, the Doppler shift of laser frequency was not calculated in these simulations so its effect on CBET is neglected, *i.e.* the right-hand side of Eq. 1.5 was forced to be zero. The plasma backgrounds for all initialisations were fully-ionised CH with  $Z = 3.1$  and  $T_i = T_e/2$ . All tests are presented in Ref. [37] and the LPSE data is available from the repository in Ref. [67].

#### 1.5.7.1 2-D CBET Without Caustics

The first test problem is for the case of two beams crossing in a non-uniform, non-stationary plasma in the absence of laser caustics. The grid used for the problem was  $x, y \in [-21, 22]$   $\mu\text{m}$

Table 1.1: The ratio of the power of the seed beam as it exits the domain to its initial power.

	$P_{\text{seed,out}}^{\text{LPSE}} / P_{\text{seed,in}}^{\text{LPSE}}$	$P_{\text{seed,out}}^{\text{BEAMCROSSER}} / P_{\text{seed,in}}^{\text{BEAMCROSSER}}$	$P_{\text{seed,out}}^{\text{SOLAS}} / P_{\text{seed,in}}^{\text{SOLAS}}$
CBET + Absorption	1.315	1.325	1.320
CBET only	1.551	1.561	1.558
Absorption only	0.856	0.856	0.852

and 2160 cells were used in each direction. This led to a resolution of  $\Delta x \sim \lambda_0/20$  such that the beat field between the two beams could be resolved. The electron temperature was uniform,  $T_e = 1$  keV. The density and fluid velocity profiles were,

$$n_e(y) = n_{\text{cr}}(0.006y_{\mu\text{m}} + 0.25), \quad (1.39)$$

$$\mathbf{u}(y) = -\hat{\mathbf{y}}(1.4 - 0.008y_{\mu\text{m}})c_s, \quad (1.40)$$

where  $y_{\mu\text{m}}$  is the coordinate in microns. The beams both had widths  $\sigma = 8.41 \mu\text{m}$  and peak intensities  $I_0 = 2 \times 10^{18} \text{ W/m}^2$ . The beams were defined as *pump* and *seed* and launched from  $\mathbf{x} = [-21, 2] \mu\text{m}$  and  $\mathbf{x} = [0, -21] \mu\text{m}$  along the  $+\hat{\mathbf{x}}$  and  $+ \hat{\mathbf{y}}$  directions respectively. The beam and plasma configuration were chosen to create a strong CBET resonance where the beams cross one another.

Fig. 1.19.a and Fig. 1.19.b show the electric field obtained from the SOLAS and LPSE simulations respectively. It can be seen that the density gradient in  $\hat{\mathbf{y}}$  causes the seed beam to refract somewhat and that CBET results in a significant energy transfer from the pump to the seed beam. The lineouts in Fig. 1.19.c, Fig. 1.19.d, Fig. 1.19.e and Fig. 1.19.f all show excellent agreement between SOLAS and LPSE. This is further demonstrated in Tab. 1.1, which lists the ratio of the outgoing seed beam power to the incident power for LPSE, SOLAS and the BEAM-CROSSER post-processor, the results from which are also presented in Ref. [37]. The energy conservation error after the pump depletion iterations was small,  $\Delta E \sim 10^{-6}$ , as expected because the CBET interaction does not involve any caustics.

### 1.5.7.2 2-D CBET With Caustics

Another two-beam problem was implemented to test CBET in the presence of a caustic. For this simulation, the grid was  $x, y \in [-21, 21] \mu\text{m}$  and  $x, y \in [-60, 60] \mu\text{m}$  with a minimum resolution of  $\Delta x = \Delta y \sim \lambda_0/20$ . A uniform electron temperature was used,  $T_e = 2$  keV. The pump beam travelled straight up a linear density gradient and therefore did not refract, whereas the seed beam reflected in the gradient and travels out. The beam geometry was chosen so that the pump beam crossed through the middle of the seed beam caustic and the density and flow profiles were tuned to create a resonance at this crossing location,

$$n_e(y) = \begin{cases} 0.02y_{\mu\text{m}} + 0.3, & x \geq -15 \mu\text{m}, \\ 0, & x < -15 \mu\text{m}, \end{cases} \quad (1.41)$$

$$\mathbf{u} = -1.14c_s\hat{\mathbf{y}}. \quad (1.42)$$

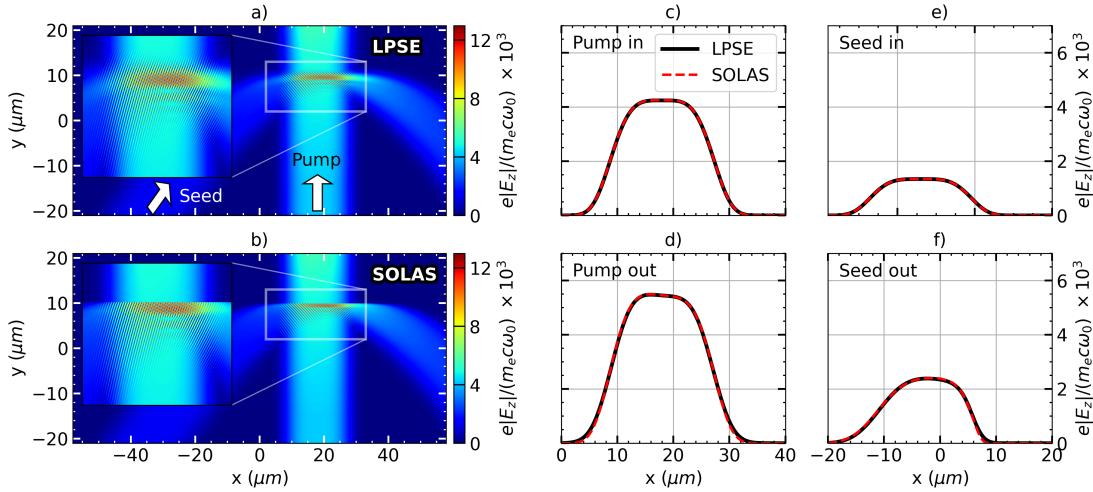


Figure 1.20: Results of the 2-D CBET test in the presence of a laser caustic. Panels a) and b) show the electric field from the LPSE and SOLAS simulations respectively. Also plotted is an inset zoom on the caustic region of the seed beam, which demonstrates that SOLAS does not capture the evanescent field. Panels c) and e) plot lineouts from the inbound fields of the pump and seed beams respectively for both SOLAS and LPSE. d) and f) plot the same for the outbound fields.

Both beams again had widths  $\sigma = 8.41\ \mu m$ . The pump and seed beams were launched from  $\mathbf{x} = [18, -21]\ \mu m$  and  $\mathbf{x} = [-35, -21]\ \mu m$  and pointed along  $\hat{\mathbf{k}} = [0, 1]$  and  $\hat{\mathbf{k}} = [0.707, 0.707]$  respectively. The peak intensities were  $I_0 = 2 \times 10^{18}\ W/m^2$  and  $I_0 = 2 \times 10^{17}\ W/m^2$  for the pump and seed respectively.

Fig. 1.20.a and Fig. 1.20.b plot the LPSE and SOLAS<sup>15</sup> total fields respectively. The inset zooms demonstrate that the evanescent fields are not captured in the GO framework. Fig. 1.20.c, Fig. 1.20.d, Fig. 1.20.e and Fig. 1.20.f plot the lineouts of the LPSE and SOLAS field solutions at their exit from the grid, which all demonstrate excellent agreement. The LPSE seed beam amplification was  $P_{\text{seed,out}}/P_{\text{seed,in}} = 1.94$  [37]. This test was run both with and without the CGT model described in Sec. 1.5.2 and at a variety of resolutions, to demonstrate how energy conservation at caustics are partially a resolution problem and how CGT mitigates for this somewhat.

Fig. 1.21 plots the results of the resolution scans both with and without CGT. The seed beam amplification after energy conservation iterations is plotted in Fig. 1.21.a, which demonstrates that even at lower resolutions, SOLAS is able to reproduce the LPSE result to an accuracy of  $\mathcal{O}(1\%)$ . The energy conservation error from the first energy conservation iteration (*i.e.* after the first set of pump depletion iterations, but before energy conservation was achieved), is plotted in Fig. 1.21.b. This plot shows convergence in the physics at higher resolutions, although the converged solution (without enforced energy conservation) indicates small errors in the handling of the caustics, which energy conservation is able to correct in an *ad-hoc* manner. Fig. 1.21.c plots the caustic pump multiplier that was required to achieve energy conservation. Both Fig. 1.21.b and Fig. 1.21.c show that CGT improves the conver-

<sup>15</sup>The field from SOLAS simulation with the minimum resolution,  $\Delta x \sim \lambda_0/20$  is plotted in Fig. 1.20.b.

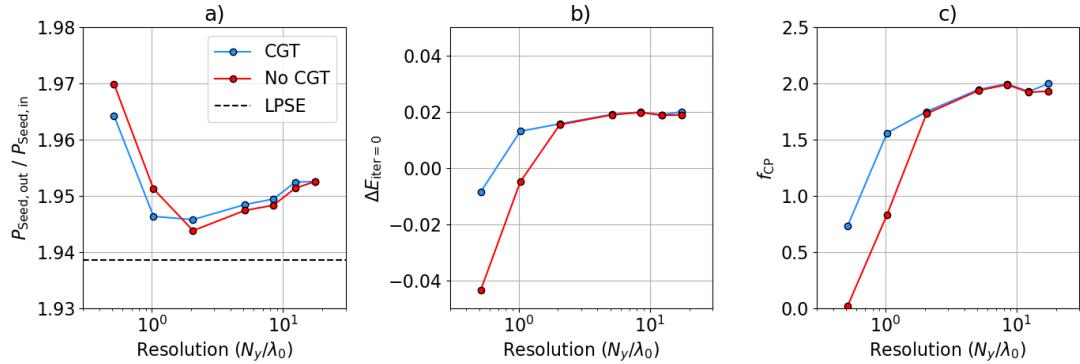


Figure 1.21: Results from a resolution scan of the two-beam caustic CBET test both with CGT and without. Plotted in a), b) and c) are the power amplification of the seed beam, initial energy conservation error before energy conservation iterations and caustic pump multiplier used to obtain energy conservation respectively.

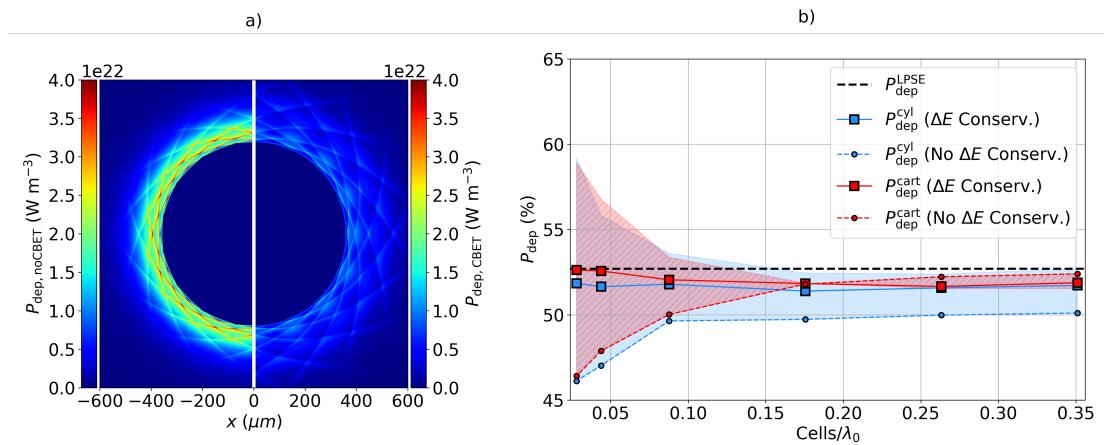


Figure 1.22: Results from the 16 beam CBET test problem. Panel a) plots the deposited power both without (left) and with (right) CBET from the simulation ran in cylindrical coordinates at  $1 \mu\text{m}$  radial resolution. Panel b) plots the results of the resolution scan for both Cartesian and cylindrical coordinates. The circles and squares are the values of power deposition without and with enforced energy conservation respectively. The shaded regions illustrate the magnitude of energy conservation error from the initial energy conservation iteration, which can be thought of as the (single-sided) error on the circular points.

gence of the SOLAS solution when the caustic region width  $\sim \lambda_0$  is not properly resolved by the mesh.

### 1.5.7.3 16 Beam CBET in 2-D

The final CBET test problem presented as validation is a planar, azimuthally symmetric CBET test with 16 beams in a direct-drive like plasma at the OMEGA scale [37]. This problem involves many beams, all of which undergo laser caustics and CBET scattering with multiple other laser sheets. It is also well suited to test the implementation of the SOLAS CBET model in non-Cartesian geometries, unlike the previous problems which did not have an angle of

symmetry. The same setup can be easily initialised in both Cartesian and cylindrical geometry to cross-compare the results.

All beams were pointed toward the origin with an initial starting radius of 1 mm, equally spaced around the circle with a width  $\sigma = 386.8 \mu\text{m}$ . The peak intensity of each beam was  $I_0 = 0.68 \times 10^{18} \text{ W m}^{-2}$  and each had an incident power  $P_{\text{inc}} = 76.3 \text{ TW}$ . The electron temperature was set to  $T_e = 2 \text{ keV}$  and the ion temperature  $T_i = T_e/2$ . The electron density used the profile from Eq. 1.20, with the scale factor,  $S = 1$ . A radial velocity profile was used, which was obtained from a fit to a 1-D LILAC simulation,

$$\mathbf{u}(r) = c_s \left[ M_0 + M_1 \log\left(\frac{r - x_0}{x_1}\right) \right], \quad (1.43)$$

where  $M_0 = 1.41$ ,  $M_1 = 1.37$ ,  $x_0 = 204 \mu\text{m}$  and  $x_1 = 343 \mu\text{m}$ . The Cartesian simulations had a bounding domain  $x, y \in [-800, 800] \mu\text{m}$ , with a resolution ranging from  $\Delta_{x,y} = 1 \rightarrow 12.5 \mu\text{m}$ . The cylindrical simulations had domain  $r \in [0, 800] \mu\text{m}$  and  $\phi \in [-\pi, \pi]$ , with radial resolutions  $\Delta_r = 1 \rightarrow 12.5 \mu\text{m}$  and a fixed  $\phi$  resolution of  $\Delta_\phi = 0.025$ . These simulations are compared to an LPSE simulation of the same problem below.

In the absence of CBET, LPSE calculated 90.5 % power deposition. All SOLAS simulations agree with this result to within  $\mathcal{O}(0.1\%)$ . Including CBET, LPSE obtained an absorption of 52.7 %. Deposited power from the  $\Delta_r = 1 \mu\text{m}$  simulation is plotted in Fig. 1.22.a, both without CBET, and with CBET after energy conservation was achieved. Power deposition from the Cartesian simulation at  $\Delta_x = 1 \mu\text{m}$  is not plotted, but looks almost identical, apart from the small grid artefacts. The reduction in magnitude of deposition is clearly visible, along with a slight radial broadening of the deposition region. The angled-line features in both plots are from the caustic region of the beam, where a 16-fold symmetry can be seen in both of the results, due to the beam geometry of the problem. The finer scale thin structures in both of these plots is due to enhanced deposition occurring at the beam caustic, where the ray amplitude swells to a maximum, enhancing the field strength.

Plotted in Fig. 1.22.b are the values of deposited power in the presence of CBET for both the cylindrical and Cartesian simulations, compared to the LPSE result. Circular markers are the values obtained for CBET deposition after the first set of pump-depletion iterations, *i.e.* before energy conservation was achieved. The larger square markers are the results after energy conservation was achieved. Shaded regions correspond to the magnitude and sign of the energy conservation error,  $\Delta E_{j=0}$ , *before* energy conservation was achieved, *i.e.*, for the circular points. This region corresponds to the deposition that could be achieved if any, arbitrary approach was taken to enforce conservation, for example if the value  $\Delta E_{j=0} = \pm 10\%$  for one simulation, then up to 10 % could be added to (subtracted from)  $P_{\text{dep}}$ . The trend, especially at lower resolution demonstrates that enforcing energy conservation brings the SOLAS result closer to LPSE, motivating the use of the approach in later chapters of this thesis. Although the cylindrical results typically appear to have a larger value for  $\Delta E_{j=0}$  compared to the Cartesian results, the post conservation results are all deemed sufficiently close to both the Cartesian result and LPSE. The saturation of error for cylindrical geometry at higher radial resolutions seem to indicate that  $\Delta E_{j=0}$  is dominated by the  $\phi$  resolution in this region,

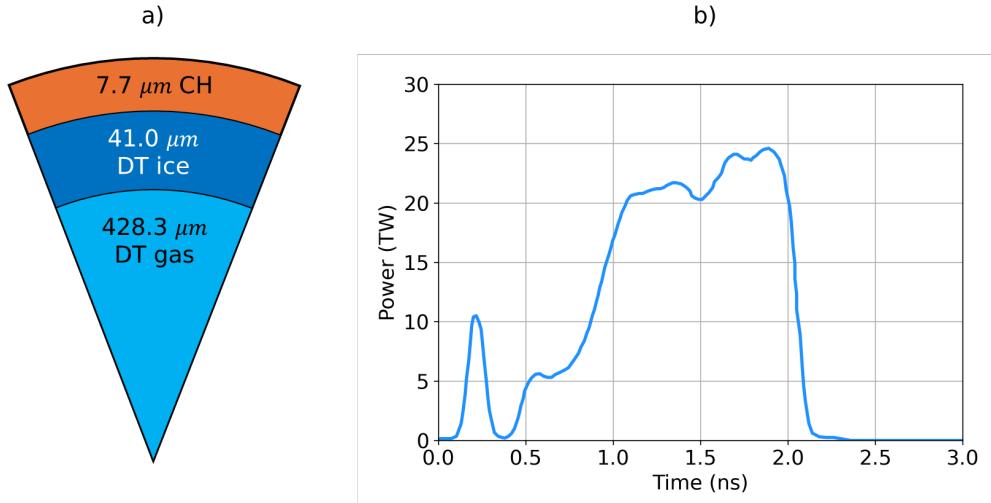


Figure 1.23: The initial conditions for OMEGA shot 89224. The shot was a cryogenic, DT implosion with materials and radii shown in panel a). Panel b) plots the incident laser pulse shape.

which was kept fixed across all simulations.

## 1.6 Simulations of CBET for OMEGA Direct-Drive Implosions

This Section presents CHIMERA-SOLAS simulations of OMEGA direct-drive implosions. In Sec. 1.6.1, 1-D Rad-Hydro simulations of OMEGA shot 89224 coupled to a 3-D ray-trace from SOLAS will be presented, both with and without the effect of CBET. This illustrates that the implementation of the CBET model in SOLAS correctly captures the loss of drive efficiency. In Sec. 1.6.1, 3-D SOLAS post-processes of spherically symmetric hydrodynamic data from the LILAC code are presented both with and without CBET. It is compared to results from the 3-D CBET codes IFRIIT and BEAMCROSSER. These simulations demonstrate that the SOLAS CBET model also correctly captures the enhancement of beam-mode drive asymmetry introduced by CBET in spherical illumination geometries.

### 1.6.1 1-D hydrodynamics, 3-D Ray-Trace Simulation of OMEGA Shot 89224

Table 1.2: Integrated metrics from simulations of OMEGA 89224. LILAC metrics are from Refs. [1–3].

	Yield ( $\times 10^{14}$ )	$\langle T_i \rangle$ (keV)	$t_{\text{bang}}$ (ns)	$V_{\text{imp}}$ (km/s)
LILAC, CBET	4.00	4.8	2.17	476
CHIMERA, CBET	1.78	4.2	2.16	490
CHIMERA, no CBET	6.77	6.2	1.94	661

A 1-D CHIMERA-SOLAS Rad-Hydro simulation of OMEGA 89224 was performed to illustrate the effect that CBET has on drive efficiency of direct-drive implosions. Drive efficiency is lost

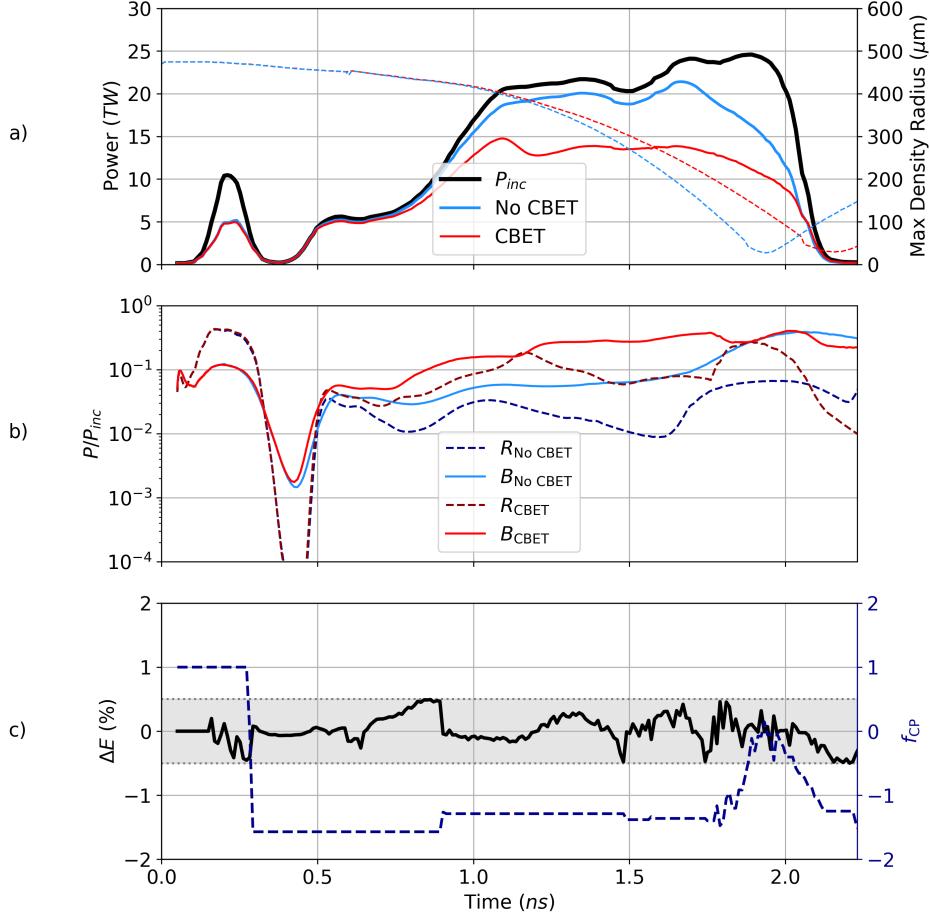


Figure 1.24: CHIMERA-SOLAS simulation of OMEGA shot 89224. Panel a) shows the deposited power both from 2 separate simulations of 89224, one with and one without CBET as red and blue solid curves respectively. The shell radius, defined at the radius of maximum density, is also plotted on the right axis as dashed curves. Panel b) plots the time resolved partition of non-absorbed light for these simulations. Plotted in c) is the energy conservation error after energy conservation was achieved, alongside the value of  $f_{CP}$  that was used to achieve conservation.

primarily due a reduction in the magnitude of deposited energy. A secondary effect is that the radii where deposition occurs is pushed slightly further away from the ablation surface. This happens because beams only lose energy inside the CBET resonance volume, which for backscatter CBET occurs at the Mach-1 surface, as explained in Sec. ???. Therefore, the deposited power vs. radius curve is shifted to slightly higher radii. Both of these effects are captured by a 3-D CBET model coupled to 1-D Rad-Hydro.

The initial conditions for shot 89224 are plotted in Fig. 1.23.a, with vacuum outside the outer radius. The incident laser pulse is also plotted in Fig. 1.23.b, which has a total incident energy of 28.7 kJ. An experimental yield of  $1.17 \times 10^{14}$  neutrons was observed for this shot. A tabulated Sesame equation of state was used for each material [79] and a  $P_{1/3}$  radiation transport algorithm was used with tabulated opacities and emissivities from the SPK code [80].

Thermal conduction was solved with using a fully implicit algorithm for flux-limited Spitzer conductivities [81]. The electron flux limiter was set by using the default CHIMERA direct-drive method,

$$f_{\text{lim},e} = \begin{cases} 0.06, & \text{if in picket pulse} \\ 0.15, & \text{otherwise.} \end{cases} \quad (1.44)$$

The simulation domain was  $r \in [0, 1680] \mu\text{m}$ , with a fixed radial resolution  $\Delta_r = 1 \mu\text{m}$ . Two simulations were conducted, one with the effect of CBET on power deposition and one without. In order to load balance the ray-trace and discretise CBET interactions between different beams around the spherical  $4\pi \text{ sr}$ , SOLAS used a sparse angular grid with  $N_\phi = 51$  and  $N_\theta = 30$  azimuthal and polar cells respectively. At times earlier than 10 ps, a *cold start* routine was employed, where 50 % of the incident power was deposited at the critical radius to create a finite scale length of underdense plasma, which rays could deposit and refract in. Ray-traces were conducted every hydrodynamic timestep, without computing CBET up to 150 ps<sup>16</sup> after which a full ray-trace routine, including the effect of CBET was conducted every 10 ps. For hydrodynamic steps in between these CBET ray-traces, the most recently computed  $P_{\text{dep}}$  profile (normalised to 1 and then scaled to the incident power), was used as the laser source term for the timestep. Parameter scans of the ray-trace frequency confirmed that the behaviour of OMEGA simulations is well converged with this frequency.

The spatially integrated absorbed power is plotted in Fig. 1.24.a as solid curves. 62.8 % and 83.9 % of the incident energy was absorbed for the CBET and no-CBET simulations respectively. Also plotted as dashed curves are the shell trajectories from the two simulations. It can be seen that the reduction in drive efficiency leads to a significantly slower implosion with a delayed bangtime. Tab. 1.2 gives temporally and spatially integrated implosion metrics from the CHIMERA-SOLAS simulations, along with a comparison to a post-shot LILAC simulation which includes a 1-D model for CBET. Including CBET in the CHIMERA-SOLAS simulation gives significantly better agreement across all metrics, which is primarily due to correctly capturing the reduction in absorption. The most interesting metrics to compare across the codes when looking at the effect of CBET in 1-D are the implosion velocity and bangtime, which are in excellent agreement with LILAC. This is because these metrics are primarily functions of the absorbed laser energy, whereas the yield and burn averaged temperature additionally depend strongly on shock timing and compression [82, 83]. Shock timing between codes is particularly sensitive to differences in the thermal conduction modelling, equation of states, grid resolution and preheat modelling, all of which are likely to have been different between the LILAC and CHIMERA simulations. The strong agreement between the CHIMERA CBET simulation and the LILAC results for the bangtime and implosion velocity metrics provide evidence that direct-drive implosions can be accurately modelled without using artificial multipliers to the laser power.

Fig. 1.24.b plots the partition of the non-absorbed power for both simulations. This is divided between *blowby* light ( $B$ ), which is light that left the computational domain travelling quasi parallel to the beam normal and *reflected* light, which leaves the domain travel-

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<sup>16</sup>Note that the low intensity from this section of the pulse means that the CBET gains are negligible.

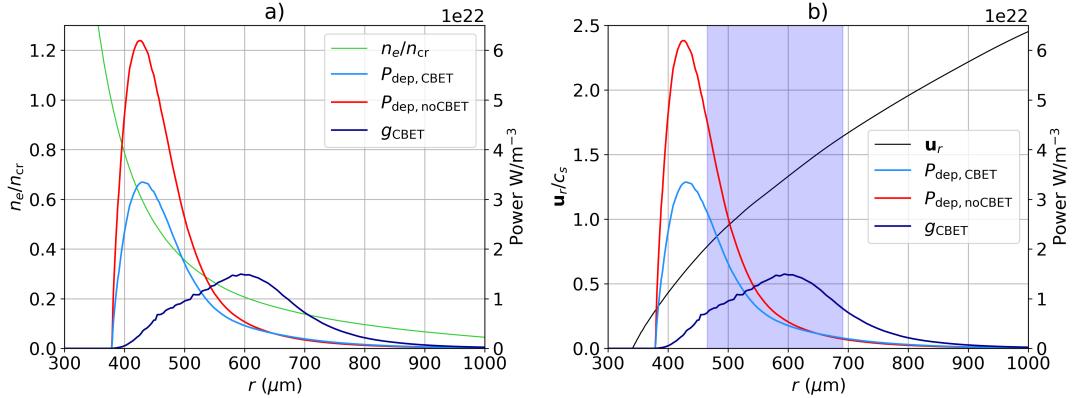


Figure 1.25: CBET deposition diagnostics from  $t = 1.6$  ns into the CHIMERA-SOLAS simulation of OMEGA 89224, with CBET. Both sub-plots plot the power deposition with and without CBET, along with the  $g_{CBET}$ , which is defined as the sum of the absolute value of all power lost or gained by rays via CBET in the cell. The electron density normalised to critical density and radial velocity normalised to the sound speed are also plotted in panel a) and b) respectively. The blue shaded region in b) represents the full-width half-maximum of the  $g_{CBET}$  curve.

ling quasi antiparallel. Note that the plot is on a log axis. Examining first the blue curves which are from the no-CBET simulation, it can be seen that initially during the main pulse, from  $t \sim 1.0 \rightarrow 1.6$  ns, the reflected light fraction decreases while the blowby goes up. This is because the critical radius converges, so the wings of the laser beam are less efficiently absorbed. The reflected light fraction increases sharply at  $t \sim 1.6$  ns and  $t \sim 1.8$  ns for the no-CBET and CBET simulations respectively. This corresponds to the time when the critical density crosses from the CH ablator material to DT fuel. DT has a significantly lower ionisation state compared to the ablator and therefore the electron-ion collision frequency decreases, corresponding to a drop in the Inv-Brem absorption and therefore an increase in the reflected light fraction. Both simulations show that blowby light is the most significant loss of coupling both with and without CBET, especially late in the implosion as the capsule converges.

Plotted in Fig. 1.24.c are the laser energy conservation,  $\Delta E$ , and the modification to the caustic pump multiplier  $f_{CP}$  that was found to achieve energy conservation. The grey bounds indicate the accepted criteria to satisfy energy conservation, which was 0.5 % of the incident power. The plot of  $f_{CP}$  illustrates that the value remained constant over many timesteps and therefore only a single set of pump depletion iterations was required for the majority of timesteps. This is not true late in the implosion, when the critical radius converges and CBET is responsible for  $\sim 40$  % reductions in deposited power. At these times, caustic CBET interactions are important and therefore  $\Delta E$  is sensitive to the value of  $f_{CP}$ . As shown in Sec. 1.5.7.3 however, the iterations are believed to improve the fidelity of the modelling which is validated here by strong agreement with tuned, post-shot LILAC simulations.

Plotted in Fig. 1.25.a are the instantaneous power deposition profiles from the CHIMERA-SOLAS simulation with CBET at  $t = 1.6$  ns, alongside the  $n_e$  profile in the under-dense region.

Note that the no-CBET profile plotted here is not from the no-CBET simulation of OMEGA 89224, but the power deposition from the field reconstruction ray-trace of the CBET simulation, *i.e.*, the deposition through the same hydrodynamic profiles, but without the effect of CBET on the power deposition. At this time, SOLAS estimates that CBET reduces the deposition versus by about 40 % compared to the no-CBET deposition. No rays go beyond  $n_{\text{cr}}$  and therefore all deposition occurs outside of the critical radius. Also plotted in dark blue is the CBET scattering,  $g_{\text{CBET}}$ , defined as the sum of the absolute value of all power lost or gained by rays due to CBET in a SOLAS cell,

$$g_{\text{CBET}}(\mathbf{x}) = \sum_i^{\text{rays}} |\Delta P_{i,\text{CBET}}(\mathbf{x})|, \quad (1.45)$$

where  $\mathbf{x}$  represents the position of the power gain/ loss, discretised within cells on the SOLAS grid. The deposition is broadly reduced only inside the peak of the scattering curve, because backscatter CBET transfers energy from inbound light to outbound light in direct-drive. Deposition is marginally increased outside the peak of the scattering curve, because the reflected and blowby light has been amplified due to CBET.

These power profiles are also plotted alongside the coronal radial velocity profile, normalised to the sound speed. The full-width half-maximum of the scattering curve is shown as the shaded blue region. Resonance occurs close to the Mach-1 surface as expected, which is where the plasma flow shifts the frequencies of radially inbound and outbound light such that they can excite an IAW. The peak scattering also is shifted slightly further out, because the outbound light mostly does not travel radially outward. Assuming the fluid CBET gain formulation and neglecting the  $(\omega_j - \omega_i)$  in Eq. 1.24, it can be shown that non-radial light shifts the resonance to larger  $u_r$  and therefore greater radii.

### 1.6.2 3-D Post Process of Absorption Non-Uniformity

A final problem used to validate the CBET model from SOLAS was to compute the instantaneous 3-D effect of CBET on power deposition through spherically symmetric hydrodynamic data from a LILAC simulation of an OMEGA implosion. The same problem has been computed with both the IFRIIT and BEAMCROSSER codes allowing cross-code validation of the 3-D deposition asymmetry. The LILAC profiles were taken from Ref. [23], and the IFRIIT and BEAMCROSSER results are presented in Ref. [76]. CBET is known to enhance the beam-mode asymmetry which arises from the number and geometry of laser beams used to illuminate the target [35]. Correctly capturing this enhancement of asymmetry relies on a field reconstruction algorithm with good treatment of laser caustics and sufficient ray statistics in every cell across the computational grid to reconstruct the field everywhere that CBET is important.

This problem is a particularly stringent test of the grid discretisation for field reconstruction. The use of an underlying spherical polar grid for the SOLAS semi-structured mesh leads to cells that vary in volume from pole to pole. Some differences in the reconstructed field are therefore inevitable between the capsule waist and pole purely from this grid discretisation.

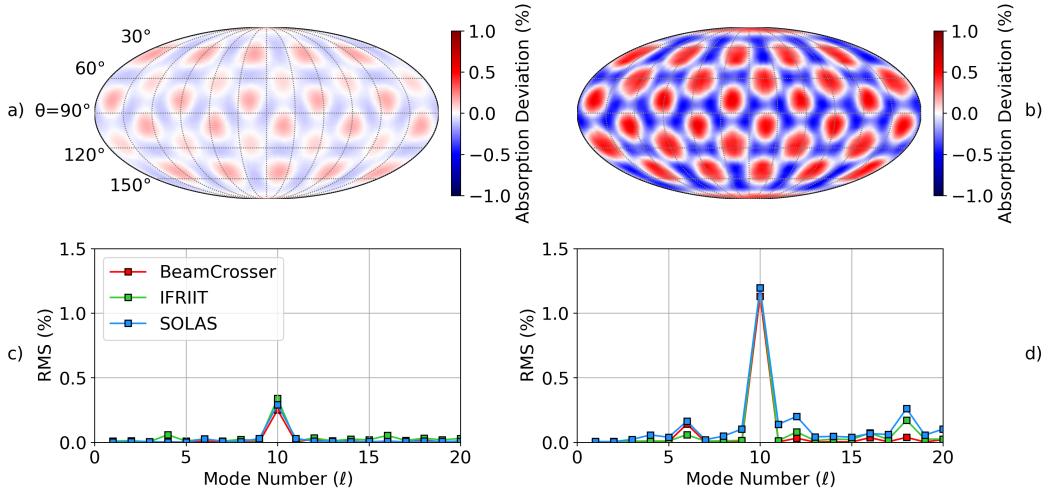


Figure 1.26: 3-D SOLAS post-process of spherically symmetric hydrodynamic data obtained from a LILAC simulation of an OMEGA direct-drive implosion. Figures a) and b) show the deviation in radially integrated deposited power without and with CBET respectively. Figures c) and d) plot the spherical harmonic modal decomposition for these maps alongside results from the IFRIIT and BEAMCROSSER codes without and with CBET respectively.

If this difference was insufficiently small, the power deposition with CBET would contain a conspicuous mode-2 which, for simulations integrated with the hydrodynamics, would lead to undesirable stagnation asymmetries, purely seeded from the computational mesh.

The simulation had a radial domain  $r \in [350, 1362] \mu\text{m}$  with underlying grid resolution  $\Delta_r = 1 \mu\text{m}$ <sup>17</sup>. The full  $4\pi \text{ sr}$  was simulated, where an underlying polar mesh was used for the SOLAS grid with  $N_\phi = 160$  and  $N_\theta = 82$ , which was then combined to give approximately equal area cells as described in Sec. 1.3.2. Fig. 1.26.a and Fig. 1.26.b show the deviation in radially integrated power deposition from the mean value for the no-CBET and CBET simulations respectively. The mode-10 pattern is clearly visible in both plots, although the magnitude is greatly increased by CBET, as expected. No clear mode-2 from the capsule pole to waist is seen which indicates that the SOLAS mesh is sufficiently close to equal-area to not result in significant grid artefacts.

A function,  $f$  discretised over azimuthal ( $\phi$ ) and polar ( $\theta$ ) angles can be decomposed into spherical harmonics,

$$f(\phi, \theta) = \sum_{\ell=0}^{\ell_{\max}} \sum_{m=-\ell}^{m=\ell} a_{\ell m} Y_{\ell}^m(\phi, \theta), \quad (1.46)$$

where  $Y_{\ell m}$  are the spherical harmonic basis vectors and  $a_{\ell m}$  are the associated coefficients [76]. The modal asymmetry of a given mode number,  $\ell$ , is then defined as,

$$\sigma_{\ell} = \sqrt{\frac{1}{4\pi} \sum_{m=-\ell}^{m=\ell} a_{\ell m} a_{\ell m}^*}, \quad (1.47)$$

<sup>17</sup>Note that the adaptive radial resolution of the SOLAS mesh had a maximum radial resolution  $\Delta_{r,\max} = 10 \mu\text{m}$ .

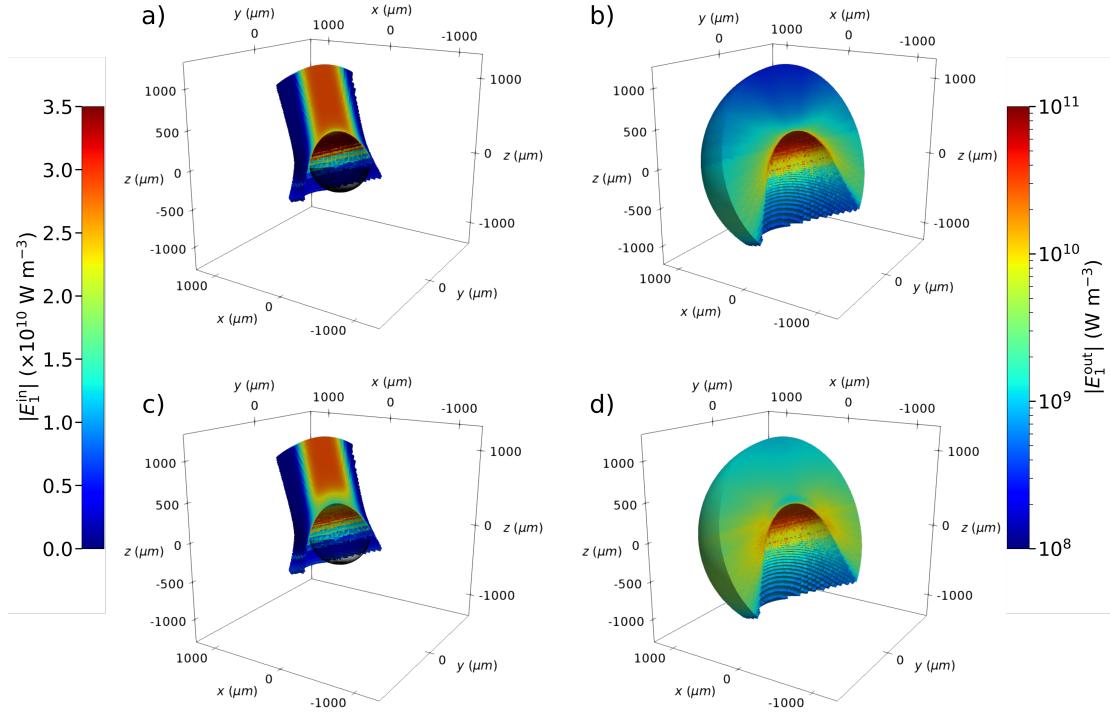


Figure 1.27: Electric fields from the 3-D post-process. Panels a), b), c) and d) plot the incident sheet without CBET, the reflected sheet without CBET, the incident sheet with CBET and the reflected sheet with CBET respectively. The critical surface is plotted with the inbound sheets as a transparent spherical wireframe. Note that the outbound sheets are plotted on a log scale.

where the  $a_{\ell m}^*$  is the conjugate of  $a_{\ell m}$ . Fig. 1.26.c plots the modal decomposition of the no-CBET radially integrated power deposition deviation from the mean, *i.e.*  $\sigma_\ell$  up to mode  $\ell = 20$ . Excellent agreement is observed between SOLAS and the other codes. The dominant mode is  $\ell = 10$ , as is expected due to the beam port geometry.

Fig. 1.26.d plots the same, but now for the power deposition including CBET. As is also seen in the skymaps, the asymmetry is significantly amplified by CBET. Good agreement is again observed between SOLAS, IFRIIT and BEAMCROSSER across the spectrum. CBET appears to induce small amplitude modes other than just the  $\ell = 10$  such as  $\ell = 6$ ,  $\ell = 12$  and  $\ell = 18$ . These are all multiples of 6, and therefore are likely induced by some OMEGA beam ports having 6 nearest neighbours [84, 85]. Small differences are observed between all codes, which is expected due to the different solvers and grids employed by each. SOLAS appears to slightly overestimate the modes 9 and 11, which have negligible values for both IFRIIT and BEAMCROSSER. This is potentially due to the ray shape function smearing routine used, that was described in Sec. 1.3.4, which smear power from a ray in one cell into neighbours. This could bleed some  $\ell = 10$  into neighbouring modes. In future, a more robust spherical harmonic smoothing could be introduced to address this issue, although the effect is not likely to be significant for integrated simulations due to the small amplitude.

The incident and reflected sheet fields<sup>18</sup> for a single beam both with and without CBET are

<sup>18</sup>Note that incident and reflected here refer to which sheet the ray belongs to, *i.e.* if it is before or after the caustic

plotted in Fig. 1.27. The incident fields without and with CBET are plotted in Fig. 1.27.a and Fig. 1.27.c respectively, and share the left colourbar. The inbound CBET sheet field shows a clear depletion of the field strength near the critical surface compared to the no-CBET field, which is the result of energy transfer from radially inward propagating light to outward travelling light via backscatter. This leads to the reduction in absorbed energy. Fig. 1.27.b and Fig. 1.27.d plot the reflected fields without and with CBET respectively, and share the right colourbar. The reflected sheet travels mostly radially outward and therefore has been significantly amplified by the CBET backscatter mechanism compared to the no-CBET case. In the absence of CBET, the field strength of the reflected sheet is significantly lower than the incident sheet field strengths for two reasons. Firstly, peak absorption occurs at the maximum radial penetration depth of the light into plasma, which is typically close to where rays experience their caustic and change sheet. Additionally, the rays diverge after reflecting from the critical surface leading to a reduction in ray amplitude and therefore field strength.

## 1.7 Conclusions

This chapter presented the implementation and validation of a 3-D ray-trace and CBET model, SOLAS, for the 3-D Rad-Hydro code CHIMERA. Prior to this work, a simple 1-D ray-trace was used to model the laser for direct-drive implosions which was strongly detrimental to the ability of the code to accurately model these experiments. Radially tracing rays inward, even for 1-D Rad-Hydro simulations, misses the important interplay between the convergence of the target and loss of deposition. The lack of a CBET model, which is responsible for significant energy coupling loss at OMEGA laser facility energy scale implosions, also meant that deposited power had to be reduced in an *ad-hoc* manner to obtain agreement with experiments. These factors severely limited the predictive capability of the code to model these experiments.

Sec. 1.3 described the implementation of the 3-D ray-trace in the SOLAS laser package. An adaptive solver is used to trace rays through a semi-structured Eulerian mesh in Cartesian, cylindrical or spherical geometries. Validation of this implementation was also presented. The 3-D laser ray-trace enables CHIMERA to model direct-drive implosions and other laser-driven HEDP experiments. It is also a crucial prerequisite to a ray-based CBET model.

Issues with ray-based CBET models often stem from a poorly reconstructed laser electric field profile. The approach taken in SOLAS to obtain the electric field was described in Sec. 1.4, which is to trace a set of neighbouring rays around each ray and then relate the area of the ray to the field strength. Validation problems were conducted for this section which illustrated that this approach yields excellent agreement with the wave-based solver LPSE. Caustics are also identified and the field near them is capped to a diffraction limited value. Validation problems showed that the field solver was robust and suitable for use within the CBET model.

The implementation of the SOLAS CBET model was described in Sec. 1.5. This model em-  
location.

ploys the linear kinetic or fluid gain formulations and the reconstructed field to estimate power changes of rays due to CBET. Fixed point iteration is then used to find the equilibrium solution and additional energy conservation iterations can be used to account for small errors in the caustic field reconstruction. Validation problems against LPSE and the ray-based BEAMCROSSER post-processor demonstrated that SOLAS correctly computes CBET for direct-drive conditions in Cartesian and cylindrical geometries.

A 1-D, spherical Rad-Hydro, CHIMERA-SOLAS simulation of the OMEGA shot 89224 was presented in Sec. 1.6.1, alongside a comparison to a LILAC simulation. The CBET model behaved as expected, significantly reducing the power deposition and slowing the implosion. Excellent agreement was obtained between CHIMERA-SOLAS and LILAC for integrated parameters which primarily depend on the amount of energy coupled to the target. Sec. 1.6.2 presented a 3-D post-process of spherically symmetric hydrodynamic data from a LILAC simulation. The power deposition asymmetry was compared to the IFRIIT and BEAMCROSSER codes and good agreement was found with those models. CBET was shown to significantly enhance the deposition asymmetries as expected. The results from this section showed that the SOLAS CBET model functions as expected in spherical geometry and therefore CHIMERA-SOLAS is capable of performing 3-D spherical direct-drive implosions with CBET which could previously only be conducted with the ASTER-IFRIIT code combination.

## 2 Cylindrical Simulations to Study the Effect of Beam Radius in Direct-Drive

This chapter describes a cylindrical, direct-drive implosion simulation platform and its use to study the effect of the beam radius initial condition on OMEGA laser facility experiments. Although results from the cylindrical simulations do not have the same convergence properties of spherical implosions, the essential physics for studying the effect of beam radius is preserved. The main benefit of the geometry is that a 2-D ray-trace can be used to model the lasers, which yields several orders of magnitude speed-up, compared to spherical 3-D implosions. The reduction in computational expense enables an ensemble of CBET simulations to be performed, which would be exceedingly expensive for 3-D spherical calculations. Beam radius strongly affects CBET and therefore including a model for the interaction in computational studies is crucial [4].

The chapter begins with a review of existing literature concerning the beam radius initial condition for direct-drive implosions, with an emphasis on the use of this parameter in statistical modelling of OMEGA campaigns. A description of the cylindrical simulations is then provided, which includes a discussion of its advantages, weaknesses and applicability to current OMEGA experiments. A tuning procedure which was followed to obtain hydrodynamically similar implosions at different beam radii is then described. The main results of the chapter are then presented, which include calculations of the power deposition asymmetry both with and without CBET and an explanation of why CBET typically amplifies the asymmetry. CBET is also shown to introduce *modal-flips* of the deposition, where the power deposition asymmetry inverts. Stagnation state asymmetries of the hydrodynamic profiles are then studied for all implosions and these demonstrate that while increasing beam radius in the absence of CBET reduces beam-mode asymmetries, the opposite behaviour is observed in the presence of CBET, although the exact relationship is complex. The chapter concludes with a summary of the simulations and suggestions for further work that could be undertaken using the same cylindrical platform are provided.

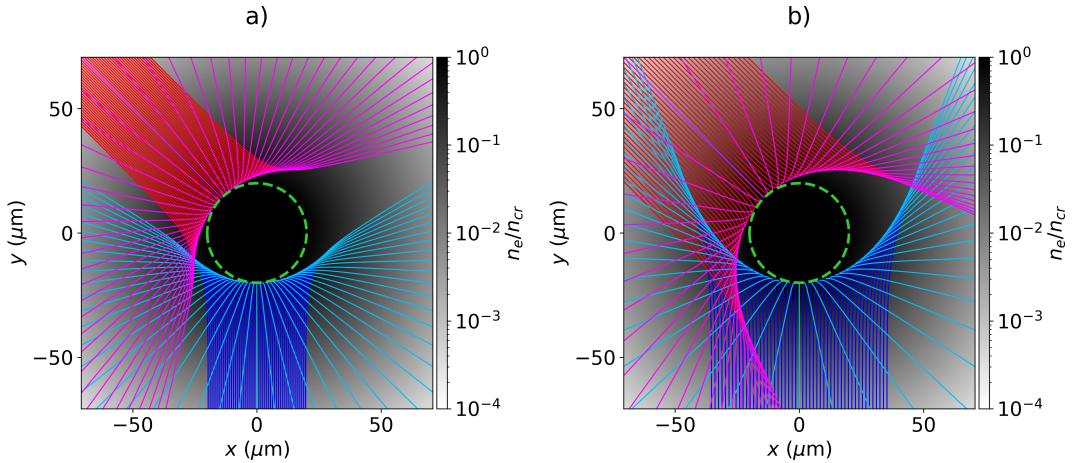


Figure 2.1: The trajectory of rays from two beams through a direct-drive like coronal plasma.

The density profile for both simulations is  $n_e = n_{\text{cr}} \exp [-(r_{\mu\text{m}} - 20)/100]$ . Panels a) and b) plot rays from beams with widths  $\sigma = 10$  and  $18 \mu\text{m}$ , respectively. Ray trajectories are separated for each beam by colour depending on their sheet. Rays coloured in red and dark blue are from the incident sheet (before the ray caustic) and rays coloured in magenta and light blue are from the reflected sheet (after the ray caustic).

## 2.1 Introduction to Beam Radius in Direct-Drive Inertial Confinement Fusion

Neglecting the effect of random shot-to-shot variations, the dynamics and performance of an idealised direct-drive implosion is defined by a limited number of initial conditions. The target can be described by a set of materials and their thicknesses. Initial target parameters are intimately coupled to the physics of the implosion and, in part, dictate the propagation time of shocks through the target, hydrodynamic stability and absorption of the laser energy. The pulse shape defines the laser power which is incident of the target as a function of time. It can be designed to, for example, drive shocks by introducing sharp rises in the incident power with time, which leads to large changes of ablation pressure [86]. A given facility also has a number of beam ports, each of which has a specific origin and pointing location, that influence the magnitude of the *beam-mode* asymmetry, arising from the uniformity of laser absorption. The focal-spot intensity profile of each laser, specifically the beam radius, is an additional parameter which can be varied. It plays an important role in defining both the power which can be coupled to the target and the magnitude of beam-mode asymmetry.

As shall be explored in this chapter, increasing the beam radius alters the magnitude of energy lost via CBET, leading to a reduction in the maximum target mass that can be imploded at a given speed [4]. The width of the beam relative to the target is therefore often effectively varied from shot to shot, by changing the outer radius (and therefore mass) of the target, while fixing the width of the beam. This defines a dimensional variable, which is the radius of the beam divided by the target radius,  $R_b/R_t$ . Typically, at the OMEGA laser facility,

this is explicitly defined as the radius of the beam which contains 95 % of the incident power divided by the initial outer target radius [4, 87, 88],

$$R_b/R_t = \frac{r_{95}}{R_t}, \quad (2.1)$$

where  $r_{95}$  is defined by the integral,

$$\int_0^{r_{95}} e^{-\left|\frac{r}{\sigma}\right|^{n_s}} dr = 0.95, \quad (2.2)$$

and the definition of a circular, super-Gaussian beam profile from Eq. 1.1 has been used. In the absence of CBET, it can intuitively be understood that increasing this parameter should improve the uniformity of the laser illumination, because beam spots overlap each other more on the target, and therefore reduce the beam-mode [5]. A larger  $R_b/R_t$  also leads to slightly less absorption in the absence of CBET, because a larger fraction of the incident light (especially at late time as the target converges) would reach lower density plasma and therefore not be absorbed. CBET significantly complicates this interpretation.

Fig. 2.1.a and Fig. 2.1.b plot results of ray-tracing calculations, in a direct-drive relevant, exponentially decaying density profile, with a smaller and larger beam, respectively. In direct-drive, backscatter CBET is the dominant mechanism that depletes absorption, which occurs when outbound light gains energy from inbound light<sup>1</sup>. The outward rays from the small beam radius simulation in Fig. 2.1.a do not overlap the incident light from the other beam and therefore limited CBET between these beams will occur. The trajectories from the larger radius simulation in Fig. 2.1.b, however, do cross the inbound rays from the other beam, which could lead to a resonant CBET interaction, and significant reduction of the absorbed power. As was shown in Fig. 1.26, CBET substantially increases absorption asymmetry on the OMEGA laser facility. This means that in the presence of CBET, the effect of increasing  $R_b/R_t$  on illumination asymmetry is not clear. While in the absence of CBET, it should lead to greater beam overlap, this increased overlap will result in more CBET, which could reduce the uniformity of absorption.

Isolating the contribution of CBET in particular is important to enable extrapolation of experimental results to future facilities. It is hoped that next generation direct-drive facilities will have broad bandwidth lasers<sup>2</sup> which are expected to almost entirely eliminate CBET [89–91]. Studying how CBET specifically affects absorption and stagnation symmetry can not currently be done experimentally, as significant backscatter CBET occurs at all laser facilities which are capable of conducting compression experiments. Therefore, computational studies are well suited to investigate how  $R_b/R_t$  influences the role of CBET.

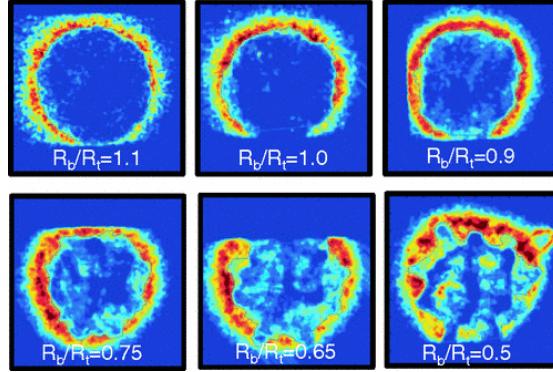


Figure 2.2: Soft x-rays emitted from the ablation surface of direct-drive implosions with various  $R_b/R_t$  values, as measured by an x-ray framing camera. All images were taken at a constant capsule radius of  $R = 175 \mu\text{m}$ . The figure has been reproduced with permission from Ref. [4].

### 2.1.1 Previous Work Studying the Effect of Beam Radius on OMEGA

Experimental and computational work has been conducted to explore the effect that the beam radius has on direct-drive implosions. Froula *et al.* conducted a series of implosions which systematically varied  $R_b/R_t$  to explore the balance between loss of coupling due to increasing CBET at large radii, and increased illumination non-uniformity at lower radii [4]. Soft x-ray emission data from a selection of implosions with different radii are plotted in Fig. 2.2, all of which are taken at the same target shell radius. These images show that at lower values of  $R_b/R_t$ , mid-mode perturbations<sup>3</sup> become increasingly significant. The results of these experiments found that neutron yield was maximised at  $R_b/R_t \sim 0.8$ . 1-D modelling using the CBET model in LILAC was in good agreement with the experimental results, verifying that CBET was responsible for the decrease in coupled energy to the target [39].

During direct-drive implosions, the target implodes radially inward, therefore the critical radius decreases with time. At the OMEGA laser facility, beam radii are fixed for a single shot, and therefore it makes sense to parameterise the initial condition by the ratio of the beam radius to the initial target radius. A promising laser optics technique known as *zooming*, could significantly enhance performance by reducing the focal spot of the laser to track the critical radius as the target implodes [92]. The reduced beam radius late in the implosion reduces blowby light and leads to more deposition closer to the ablation surface, which enhances the ablation pressure and overall performance. Simulations by Trickey *et al.* in Ref. [83] have demonstrated that, assuming full CBET mitigation, zooming can enhance ablation pressures by  $\sim 50\%$  for ignition scale direct-drive experiments. If zooming is employed without full CBET mitigation, the fractional increases would be substantially larger, because CBET losses scale very strongly with  $R_b/r_{\text{cr}}$ , which is the ratio of beam radius to critical radius [87]. 2-D modelling has also demonstrated that zooming the beam to  $\sim 60\%$  of its initial radius effec-

<sup>1</sup>'Outbound' here, means light travelling quasi-parallel to the approximately radially outward fluid velocity, and 'inbound' means quasi-anti-parallel.

<sup>2</sup>It is expected that  $\Delta\omega/\omega_0 \sim 1\%$  will be sufficient to eliminate CBET in direct-drive for frequency-tripled lasers.

<sup>3</sup>In direct-drive, *mid-modes* are loosely defined as modes similar to  $\ell = 10$ .

tively mitigates CBET while maintaining low-mode uniformity [93].

### 2.1.2 Statistical Modelling of OMEGA Direct-Drive Implosions

In recent years, a lot of work has been carried out to develop a statistical modelling capability for direct-drive implosions on the OMEGA laser facility. This modelling serves several critical purposes, including enhancing the predictive capability of simulations [94], guiding experimental design to achieve higher performance implosions [95], identifying important sources of degradation on current facilities [5, 96] and validating simulation codes to help ensure they produce physically relevant results [97]. The first generation statistical model, described by Lees *et al.* in Ref. [94], uses a high dimensional fit, to create a mapping between experimental and 1-D simulation results in order to explain significant differences in their results. A next-generation statistical model has recently been developed by Ejaz *et al.*, which uses auto-encoders to reduce the design-space dimensionality and subsequently transfer learning to calibrate simulation outputs to experimental data. The subsequent review shall however, focus on the more widely studied results of the first model.

In the model described by Lees *et al.* [94], 1-D simulation results are fed into the model and degraded by a series of power law multiplications, which returns a more physically accurate experimental yield. Each power law multiplication represents a physical process for yield degradation with respect to 1-D physics, not included in the simulation. Each of these is termed a *yield-over-clean* ( $\text{YOC}_i$ ), where the  $i$  refers to each source of degradation. The neutron yield from the 1-D simulation ( $Y_{1\text{-D}}^{\text{sim}}$ ) can thus be converted to a prediction of the experimental yield [94],

$$Y^{\text{exp}} = (\text{YOC}_h \text{YOC}_f \text{YOC}_{\ell=1} \text{YOC}_b \text{YOC}_{\text{res}}) Y_{1\text{-D}}^{\text{sim}}, \quad (2.3)$$

where  $\text{YOC}_h$  is a degradation term from hydrodynamics and instability growth,  $\text{YOC}_f$  is degradation due to radioactive decay of the tritium fill,  $\text{YOC}_{\ell=1}$  is degradation from  $\ell = 1$  modes,  $\text{YOC}_b$  is degradation from a finite number and radius of beams and  $\text{YOC}_{\text{res}}$  is a residual size scaling, which is required to reduce performance of hydrodynamically downscaled implosions [98]. Each of these terms and their functional forms shall be discussed briefly, in order to provide context for the utility of the model and highlight that understanding the relevant physical processes which lead to degradation can improve its performance.

**Hydrodynamic Degradation** 1-D simulations do not capture short wavelength perturbations which grow via the Rayleigh–Taylor Instability (RTI) and reduce the yield of experiments by puncturing and breaking up the shell as the capsule implodes inwards. Instabilities may be seeded by laser imprint [99–101], or small scale defects in the target materials [102, 103]. Degradation can be reduced either by altering implosion design to increase the shell adiabat which increases the ablative stabilisation of the RTI [104, 105], or by lowering the In-Flight Aspect Ratio (IFAR), which increases the distance that the instability must grow through to puncture the shell [106]. The degradation term includes scaling with the target convergence ratio,  $C_R \equiv R_0/R_{\text{stag}}$ , and with the ratio of outer to inner shell radius,

$\hat{D} \equiv R_{\text{out}}/R_{\text{in}}$ , which is believed to compensate for inaccuracies in modelling the shock propagation speed through the target. The hydrodynamic degradation term thus has the functional form,

$$\text{YOC}_h = \left[ \frac{(\alpha/3)^{1.1}}{\text{IFAR}/20} \right]^{\mu_1} C_R^{\mu_2} \hat{D}^{\mu_3}, \quad (2.4)$$

where  $\mu_i$  are the fitting parameters, which are obtained from nonlinear regression across many OMEGA shots. The fitting procedure demonstrated that experimental yields are very significantly reduced by these hydrodynamic degradations, with the most unstable shots yielding values of  $\text{YOC}_h \sim 0.1$  [5].

**Fill Age Degradation** OMEGA cryogenic implosions contain a DT fuel gas fill with a surrounding ice layer. The tritium in this fuel is unstable and undergoes radioactive decay to  ${}^3\text{He}$ . This occurs over a period of days to weeks, which typically elapse between the initial gas filling procedure to the shot day [107].  ${}^3\text{He}$  has a lower freezing temperature than DT and thus sublimates, accumulating in the fill region. The accumulation of helium in the gas reduces the final yield, both by increasing radiative losses due to its higher ionisation, and by increasing density of the vapour, which reduces compressibility and thus stagnation pressure in the hot-spot. Both of these effects can be captured by conducting 1-D simulations with a  ${}^3\text{He}$  concentration (and corresponding reduction of tritium density) which is a function of fill age. The yield over clean due to the fill age and radioactive decay can then be taken as the ratio of these 1-D simulation yields,

$$\text{YOC}_h = \left( \frac{Y_{1\text{-D},\text{He}}^{\text{sim}}}{Y_{1\text{-D}}^{\text{sim}}} \right)^{\mu_4}, \quad (2.5)$$

where  $Y_{1\text{-D},\text{He}}^{\text{sim}}$  is the yield from the 1-D simulation with accumulated  ${}^3\text{He}$  and  $\mu_4$  is a fitting parameter. Good agreement is observed with a fitted parameter value of  $\mu_4 = 1.3$ . The value is larger than 1 (and the 95 % confidence interval does not include 1), which suggests stronger degradation than observed in 1-D calculations. This could be due to radioactive decay damaging the shell, leading to increased hydrodynamic instability growth [5].

**Mode 1 Degradation** In direct-drive implosions on the OMEGA facility,  $\ell = 1$  modes can be introduced to an implosion by a global offset of the capsule from the target chamber centre, mispointing of the laser beams or a power imbalance. These are random and uncontrollable and therefore the statistical models can only account for their effect after the shot has occurred. By conducting post-shot analysis of an implosion, the model returns an estimated yield which could have been achieved, if no  $\ell = 1$  were present,  $Y^{\text{exp}}/\text{YOC}_{\ell=1}$ . Mode 1 asymmetries have a clear signature in the broadening of the neutron time-of-flight detector peaks, when observed from orthogonal lines of sight [78]. The width of the peaks from multiple lines of sight can be analysed to return an angularly resolved apparent ion temperature map, the asymmetry of which is dominated by the lowest mode of the hot-spot [108]. Thus, it is deduced that the ratio of the maximum to minimum apparent ion temperatures from the experimental neutron time-of-flight signal can be used as a proxy for the amplitude of the

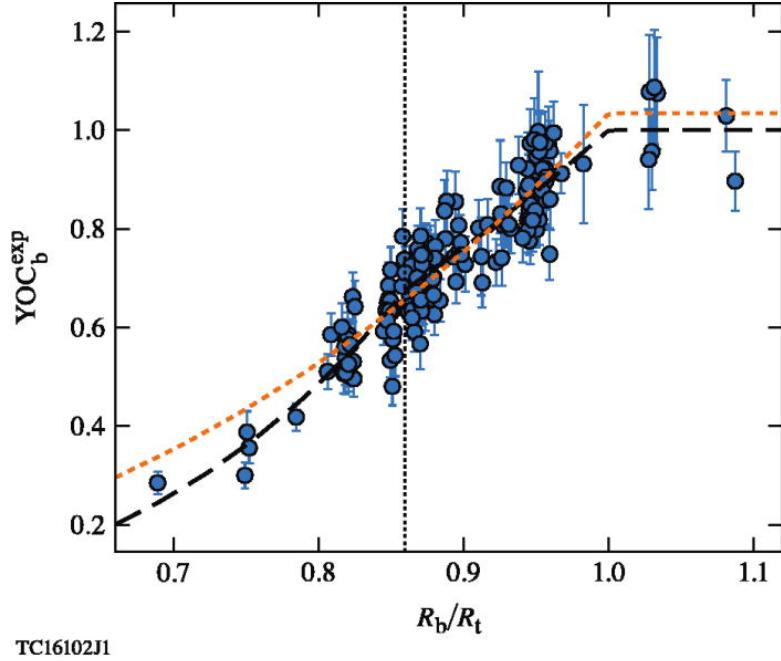


Figure 2.3: Experimentally inferred fusion yield degradation due to the finite beams on the OMEGA laser facility. The dotted orange curve is the fit obtained from just using the  $\bar{R}_{b/t}$  relation, while the black dashed curve uses the full relation in Eq. 2.7. The vertical dotted line indicates the critical threshold,  $\hat{R}_{b/t}^{\text{crit}} = 0.86$ , after which the  $\hat{R}_{b/t}$  also has an effect. The cross validation error from the  $\bar{R}_{b/t}$  and full fit is  $-1.0\%$  and  $-0.5\%$  respectively. The figure has been reproduced with permission from Ref. [5].

mode 1,  $R_T = T_{\max}/T_{\min}$ . This leads to a yield over clean expression for the  $\ell = 1$  degradation source,

$$\begin{aligned} \text{YOC}_{\ell=1} &= \hat{R}_T^{\mu_5}, \\ \hat{R}_t &\equiv \max\left(\frac{R_T}{R_T^{\min}}\right), \end{aligned} \quad (2.6)$$

where  $\mu_5$  is the fitting parameter and the minimum threshold value,  $R_T^{\min}$  is introduced due to imperfect reconstruction of the apparent ion temperature map and is fitted separately. Work has been conducted to minimise the effect of the  $\ell = 1$  on OMEGA by repositioning the target after several initial shots to minimise the asymmetry in the apparent ion temperature measurement and thus increase performance [78].

**Finite Beam Degradation** The OMEGA laser facility has 60 beams arranged around a sphere, which generally gives good illumination uniformity on a hard sphere surface, less than the 1 % deviation which is believed to be necessary to achieve ignition [109, 110]. An  $\ell = 10$  remains in the deposition however, as is demonstrated in Fig. 1.26, which is often referred to as the beam-mode. In the absence of CBET, increasing  $R_b/R_t$  increases the hard-sphere illumination uniformity [96]. As already described however, increasing beam radius leads to more blowby light and therefore more CBET. This reduces the coupled energy and potentially introduces additional asymmetry to the implosion. Additionally, increasing the over-

lap of beams on the target could reduce the amplitude of the imprint seed and therefore increase performance. The uncertainty as to which physical mechanisms are important, is highlighted by the complexity of the degradation parameter compared to those already described,

$$\text{YOC}_b = \left( \bar{R}_{b/t} \right)^{\mu_6} \left( \hat{R}_{b/t} \right)^{\mu_7},$$

$$\bar{R}_{b/t} = \begin{cases} R_b/R_t & \text{if } R_b < R_t, \\ 1 & \text{if } R_b \geq R_t, \end{cases}$$

$$\hat{R}_{b/t} = \begin{cases} \frac{R_b}{R_t R_{b/t}^{\text{crit}}} & \text{if } R_b/R_t < R_{b/t}^{\text{crit}}, \\ 1 & \text{if } R_b/R_t \geq R_{b/t}^{\text{crit}}, \end{cases}$$
(2.7)

where  $\mu_6$  &  $\mu_7$  are fitting parameters, the threshold behaviour in  $\bar{R}_{b/t}$  was chosen to fit a small number of shots ( $< 10$ ) at  $R_b/R_t > 1$  and the threshold behaviour at  $R_{b/t}^{\text{crit}}$  was introduced to fit a physically unexplained transition between two regimes in the data.

The fitted curve from the model is shown in Fig. 2.3, as the black dashed curve alongside the inferred values from experimental data points. Also plotted in orange is a fitted curve obtained from just using the simple  $\bar{R}_{b/t}$  degradation. Introducing the  $R_{b/t}^{\text{crit}}$  threshold significantly reduces the cross validation error of the fit. The switch between the two regimes is found from the fitting procedure to occur at  $R_{b/t}^{\text{crit}} = 0.86$ . This is close to value of minimum illumination asymmetry for beams incident on a hard hard-sphere ( $R_b/R_t = 0.82$ ), which suggests that the degradation at the lowest values of  $R_b/R_t$  is dominated by beam-mode, however this has not been experimentally or computationally verified. Experiments between  $R_{b/t}^{\text{crit}} < R_b/R_t < 1$  could be influenced by changing behaviour due to CBET or imprint, which is not properly captured by the 1-D LILAC simulations included in the model.

The hypothesis tested in this chapter is that the change in yield over clean at  $R_b/R_t = R_{b/t}^{\text{crit}}$  is due to increasing CBET as beam radius increases, which increases beam-mode asymmetry and therefore suppresses the  $\text{YOC}_b$  term. Although LILAC does include a model for CBET, it is a 1-D code and therefore the 3-D beam-mode perturbations cannot be inferred from its results. Qualitatively, this hypothesis can explain the observed behaviour in Fig. 2.3, which demonstrates that at  $R_{b/t}^{\text{crit}}$ , the gradient of  $\text{YOC}_b$  decreases. This behaviour should occur if CBET acted to amplify the asymmetry of the stagnation state, causing the simulation to be less similar to the 1-D LILAC results, than if CBET did not significantly amplify asymmetries.

## 2.2 Cylindrical Simulation Platform for Beam Radius Parameter Scan

The method employed to study whether CBET induced beam-mode asymmetry at  $R_b/R_t \gtrsim R_{b/t}^{\text{crit}}$  contributes to the second distinct regime from Eq. 2.7, was to conduct a series of cylindrical direct-drive simulations. These simulations are in a different convergence regime to spherical implosions, but retain much of the key physics relevant to the study, such as CBET, target convergence and beam-mode perturbations of the target. Crucially, it circumvents the

large computational run-times of 3-D spherical simulations which include a CBET model. Inspiration for the cylindrical implosion platform was taken from the work of Follett *et al.* in Ref. [37], where a similar, 16-beam, planar configuration was used for a 2-D CBET test problem. The SOLAS result for this validation problem is presented in Sec. 1.5.7.3.

### 2.2.1 Advantages and Validity Considerations of the Cylindrical Simulation Platform

Ideally, fully 3-D Rad-Hydro simulations, coupled to a 3-D CBET model would be used for this work. This would retain the spherical convergence of the implosion and the 3-D nature of the beam-mode perturbation growth through stagnation, while also capturing how CBET influences these effects. These simulations are extremely expensive however and can take months to complete. A 2-D ray-trace (where rays can only move in two cardinal directions rather than 3) leads to large reductions in cost.

Ray-based CBET models require a ray from every laser sheet to pass through every computational grid cell where CBET scattering is significant. Therefore, by reducing the dimensionality of the problem, savings are made proportional to the reduction in number of cells, which is  $\mathcal{O}(100)$  from 3-D spherical direct-drive calculations to the 2-D simulations presented in this chapter. Additionally, 10 beams were used to produce a mode-10, rather than the 60 on OMEGA, yielding another factor of 6 fewer rays. Each of the eight 2-D CBET simulation presented here took  $\mathcal{O}(10^3)$  CPU hours, whereas (assuming the above logic),  $\mathcal{O}(10^6)$  CPU hours would be required for each equivalent 3-D spherical simulation. On 1000 processors, each 3-D simulation would therefore take over a month to finish, as opposed to these simulations which all ran to completion within a day on 128 cores. Note that the above scaling assumes that the laser ray-trace and CBET computation costs dominate the simulation run-time, which is found to be the case for all CHIMERA-SOLAS simulations.

The physics of the implosion is however different in cylindrical as opposed to spherical geometry. Firstly, the mass converges in only two directions rather than three as the target implodes, which results in increased convergence at stagnation, potentially altering the beam-mode asymmetry growth. The 2-D cylindrical perturbations will also only evolve in the simulation plane, unlike the true 3-D case where they interact with material ‘above’ and ‘below’ them as well. This could lead to the cylindrical simulations overpredicting the beam-mode degradation compared to 3-D spherical simulations. In the corona, the expanding plasma also diverges in only two directions rather than three as it rockets away from the capsule, which leads to reduced density gradients in the corona, where the laser propagates and deposits energy. This could have the effect of shifting the deposition to greater radii above the critical surface, reducing the drive efficiency.

Despite these differences, the simulations were, qualitatively, sufficiently hydrodynamically similar to spherical implosions, to suggest that parallels can be drawn from the work done here to spherical implosion data. Cylindrical implosion experiments are also conducted on laser facilities, the results of which are often explicitly extrapolated to spherical implosions, suggesting that the approach is valid and accepted within the community [111,

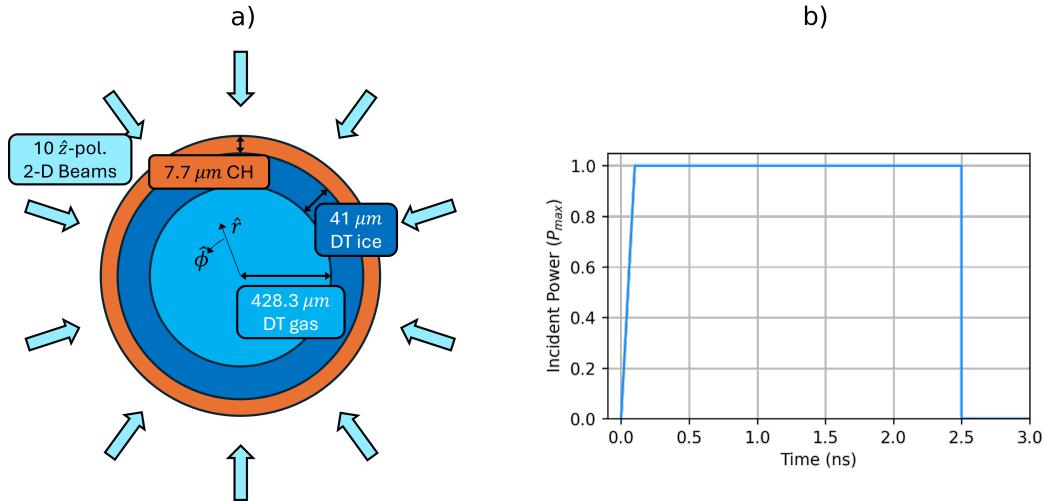


Figure 2.4: The a) target initial conditions with beam geometry, and b) pulse shape used for the 2-D cylindrical simulations. All beams were polarised out of the simulation plane, in the  $+\hat{z}$  direction. Initial layer radii were taken from the initial conditions for OMEGA shot 89224, presented in Fig. 1.23.a.

112]. Although the differences between the two regimes are not quantified in the work conducted in this chapter, future simulations could be performed, which extend the platform to a 2-D ‘plane’ in spherical rather than cylindrical geometry, which would capture the spherical convergence of the target and divergence of the coronal flow. This is discussed in more detail at the end of the chapter, in Sec. 2.5.

### 2.2.2 Pulse Shape and Target Initial Conditions

The base simulation initial conditions are plotted in Fig. 2.4. The initial conditions are symmetric about the azimuth and in the plane normal to the  $\hat{z}$  direction. A target with the same initial layer radii as OMEGA shot 89224 was constructed with a DT gas fill, a layer of DT ice and a CH plastic ablator with vacuum outside, shown in Fig. 2.4.a. Ten beam were placed around the target, equally spaced in azimuthal angle and all were polarised in the out-of-plane,  $\hat{z}$  direction. All beams had super-Gaussian spot profiles with  $n_s = 5.2$  and  $\sigma$  set by Eq. 2.2. A simple 2.5 ns square pulse (including a 0.1 ns ramp to full power) was used for all simulations, plotted in Fig. 2.4.b. The maximum power of the pulse for each 2-D simulation was tuned from a separate set of 1-D simulations, such that the bangtime occurred at 2.5 ns.

By tuning the simulations such that the bangtime was consistent across all simulations, the coupled energy and implosion velocity was kept the same across all implosions. The difference between implosions was therefore primarily due to differences in the spatial location of the deposited power. If the incident energy were fixed, increasing  $R_b/R_t$  would lead to more CBET, which would result in less energy coupled to the target. Therefore, to compare simulations, the target would also have to be altered to reduce the imploding mass with less coupled energy. This was deemed beyond the scope of the work presented in this chapter and therefore only the incident energy was altered to maintain the 1-D implosion hydrodynamics.

Table 2.1: Results of the 1-D Tuning Simulations.

$R_b/R_t$		$P_{\max}$ (TW/cm)	$I_0$ ( $10^{14}$ W/cm $^2$ )	$t_{\text{bang}}$ (ns)	$Y_{\text{DT}}$ ( $10^{13}$ /cm)
0.75	No CBET	54.44	0.85	2.49	1.53
	CBET			2.51	1.44
0.80	No CBET	58.25	0.83	2.49	1.56
	CBET			2.51	1.45
0.85	No CBET	63.44	0.83	2.48	1.67
	CBET			2.51	1.43
0.90	No CBET	70.00	0.85	2.47	1.82
	CBET			2.50	1.41
0.95	No CBET	77.94	0.89	2.46	1.99
	CBET			2.49	1.49
1.00	No CBET	87.25	0.93	2.45	2.15
	CBET			2.49	1.60
1.05	No CBET	97.94	0.99	2.46	2.27
	CBET			2.50	1.61
1.10	No CBET	110.00	1.06	2.47	2.31
	CBET			2.51	1.52

Every simulation in this chapter used a grid with radial extent  $r \in [0, 1600]$  with resolution  $\Delta r = 1 \mu\text{m}$  and 256 cells in the azimuthal direction. A tabulated Sesame table of state was used for each material [79] and thermal conduction routine was solved using an Alternating-Direction Implicit (ADI) method with flux-limited Spitzer conductivities [113]. The electron flux limiter was set using the default CHIMERA direct-drive setting, outlined in Eq. 1.44. Radiation transport was not included in simulations, because the small cells on the  $r = 0$  boundary led to significant computational expense. 1-D calculations showed that the effect of including radiation transport was relatively small. The dominant effect was that radiation transport reduced the bangtime by  $\sim 0.1$  ns, primarily due to temperature losses in the corona. Future work could therefore include the dominant radiation effect by using a radiative loss model rather than full transport.

### 2.2.3 1-D Implosion Tuning

As already mentioned, the energy of the laser was varied to maintain a consistent bang-time across all  $R_b/R_t$  values, so that the target parameters did not have to be separately optimised for each simulation. This was done via a series of 1-D, with-CBET simulations which varied the maximum power of each beam,  $P_{\max}$ , at each  $R_b/R_t$  to obtain an implosion with  $t_{\text{bang}} = 2.50 \pm 0.01$  ns. For no-CBET simulations, the absorbed power vs time from the CBET simulation with the same  $R_b/R_t$  was enforced. Thus, when comparing any two simulations, the absorbed energy is equivalent, but the spatial location of the deposition is

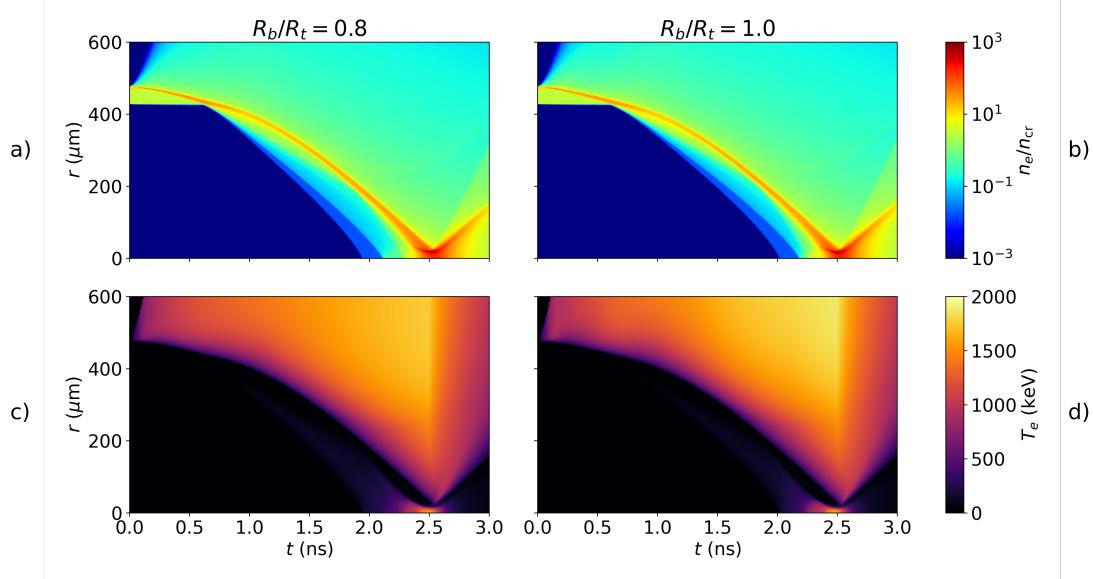


Figure 2.5: Streak plots from two of the 1-D tuning simulations including CBET. Panels a) and b) plot the electron density of the  $R_b/R_t = 0.8$  and  $R_b/R_t = 1.0$  simulations, respectively, as a function of time ( $x$ -axis) and radius ( $y$ -axis). Panels c) and d) plot the electron temperature for the  $R_b/R_t = 0.8$  and  $R_b/R_t = 1.0$  simulations, respectively.

different. This manifests both as different azimuthal asymmetries in the deposition, which alters the stagnation state asymmetry, and different radial location of absorption. For example, the with-CBET power deposition occurs at slightly larger radii compared to no-CBET profiles, as is shown in Fig. 1.25. This means that the no-CBET implosions have a slightly increased drive efficiency compared to CBET implosions, because thermal conduction does not have to transport energy as far from the absorption region to the ablation surface.

Tab. 2.1 shows implosion metrics from all the tuned 1-D implosions. As shown, the incident maximum power of each beam for the CBET simulations,  $P_{\max}$ , increases with increasing  $R_b/R_t$  because larger  $R_b/R_t$  leads to more CBET and therefore less absorption, so more incident power is required to maintain the same absorbed energy. The maximum intensity of each beam at peak power,  $I_0$  is non-monotonic, because although the maximum power increases, the beam radius also increases, which limits the increase in maximum intensity. Bangtimes and yields are similar across all simulations. Note that with increasing  $R_b/R_t$ , bangtime and yield difference between the CBET and no-CBET results at the same  $R_b/R_t$  increase. This is because more CBET occurs for the larger  $R_b/R_t$  simulations and therefore the difference in deposition radius increases between CBET and no-CBET simulations, marginally improving the effective drive efficiency of the no-CBET results.

Streaks, which plot hydrodynamic quantities as a function of radius and time, of  $n_e$  and  $T_e$  are shown in Fig. 2.5 for 1-D, with-CBET simulations at two separate  $R_b/R_t$  values. Qualitatively, the 1-D implosion trajectories from these plots are similar. Small differences in shock timing exist between simulations, as is evidenced by the initial shock for the  $R_b/R_t = 0.8$  hitting the  $r = 0$  axis at  $t \sim 1.9$  ns, which is about 0.1 ns earlier than the  $R_b/R_t = 1.0$  simula-

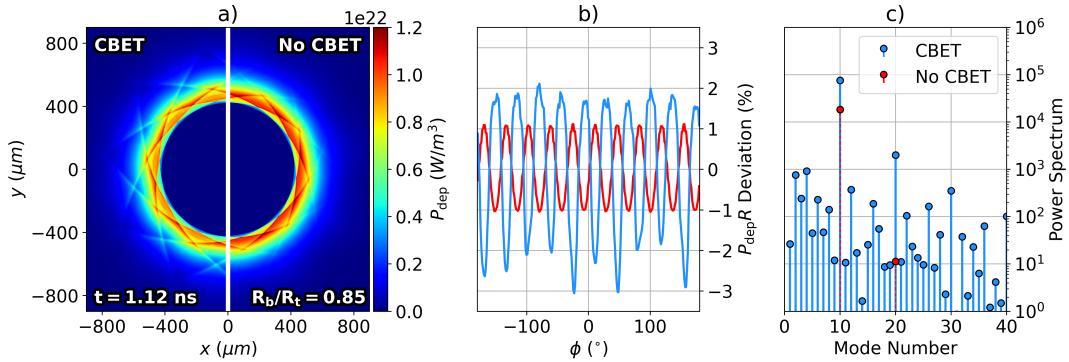


Figure 2.6: Demonstration of the analysis workflow to obtain the key results for this chapter.

The power deposition at  $t = 1.12 \text{ ns}$  from the CBET (left) and no-CBET (right) simulations are plotted in panel a) for the  $R_b/R_t = 0.85$  case. Panel b) plots the radially integrated deposition from the profiles in a) as a function of azimuthal angle. As can be seen from this plot, the CBET asymmetry (light-blue) is greater than the no-CBET asymmetry (red). The power spectrum of these profiles is then plotted in panel c). This demonstrates that the dominant modes in the spectrum are multiples of the number of beams.

tion ( $t \gtrsim 2.0 \text{ ns}$ ). Despite these small differences in both the metrics from Tab. 2.1 and the streaks from Fig. 2.5, all simulations were deemed sufficiently similar that implosions could be cross-compared.

## 2.3 Asymmetry of Deposited Power

This section describes the asymmetry of the deposited power profile for simulations with and without CBET. The effect of this asymmetry on the in-flight and stagnation state hot-spot profiles are discussed in Sec. 2.4. Analysis of the deposited power profile shows that the growth of asymmetries in the target is the result of a complex, space- and time-dependent evolution of the deposition. In the absence of CBET, *modal-flips* of the deposition occur, where the phase of the driving asymmetry flips in time. This is due to the overlapping beam intensity changing in the region where Inv-Brem deposition is important. The pattern of these modal flips depends on the width of the beams, the time-dependent convergence of the target and the time-dependent coronal plasma altering the radius above the target where deposition is important. It is observed that in the presence of CBET, due to the non-uniform resonance of CBET gains across inbound laser sheets, additional asymmetries in the deposition are seeded and lead to more modal-flips than are observed without CBET.

### 2.3.1 Analysis and Quantity Definitions

Initially, definitions of key variables used in the analysis of the results of the chapter shall be provided. These are introduced for the example of the with-CBET and no-CBET  $R_b/R_t = 0.85$  simulations, plotted instantaneously at  $t = 1.12 \text{ ns}$  in Fig. 2.6.

Fig. 2.6.a shows the volumetric deposited power for the CBET (left) and no-CBET (right)

simulations. Note that, as described in Sec. 2.2.3, the no-CBET simulation is forced to absorb the same magnitude of power as a function of time as the CBET simulation. Therefore, the total absorbed power is identical for both simulations, even though the no-CBET plot appears more saturated on the colour scale. This difference in saturation is partially due to the non-linear colour scale used for the plot, and also because the CBET result has more power deposited at larger radii, which widens the profile and reduces saturation on the colour scale. The mode-10 in the deposition due to the number of beams is clearly visible on both plots. Significant deposition in the caustic region, especially for the CBET result, is visible as the cross-structure in the deposited power. This suggests (and shall subsequently be shown explicitly in Sec. 2.3.3), that the caustic fields are strongly amplified by CBET, leading to more Inv-Brem in this region.

In order to quantify azimuthal asymmetry, radial integrals of the deposited power and fuel density are taken,

$$\begin{aligned} P_{\text{dep}} R(\phi) &= \int_{r=0}^{\infty} P_{\text{dep}}(r, \phi) dr, \\ \rho_{\text{DT}} R(\phi) &= \int_{r=0}^{\infty} \rho_{\text{DT}}(r, \phi) dr, \end{aligned} \quad (2.8)$$

where  $P_{\text{dep}}$  is power, in units [ $\text{W/m}^3$ ], unlike the volumetric power plotted in Fig. 2.6.a. The deviation from the mean of these profiles can then be taken,

$$\text{Deviation}(f[\phi]) = \frac{f[\phi] - \int_{-\pi}^{\pi} f[\phi] d\phi}{\int_{-\pi}^{\pi} f[\phi] d\phi}. \quad (2.9)$$

The deviation of the CBET and no-CBET deposited power profiles shown in Fig. 2.6.a is plotted in Fig. 2.6.b. It can be seen that at this time, CBET considerably amplifies the instantaneous deposition asymmetry. It also distorts the sinusoidal profile of the no-CBET simulation, marginally widening and narrowing the curve peaks and troughs, respectively. Interestingly, CBET has also resulted in a phase-inversion of the deposition profile, where the peaks of the CBET deviation occur at the angles of the troughs of the no-CBET curve. This behaviour shall be called a *modal-flip* throughout this chapter. Note that modal-flips are also observed in the same simulation through time, *i.e.* as the target converges, the beam overlap pattern changes which results in phase inversions of the deposition, relative to earlier deposition profiles.

A Discrete Fourier Transform (DFT) is then used to analyse the modes which contribute to the signal. A signal  $f(\phi)$ , which is sampled  $N$  times in the interval  $\phi \in [\phi_{\min}, \phi_{\max}]$  where  $n = 0 \rightarrow N - 1$ , has a DFT defined by,

$$F_{\ell} = \sum_{n=0}^{N-1} f_n \exp\left(-i2\pi \frac{\ell}{N} n\right), \quad (2.10)$$

where  $f_n$  is the sample at  $\phi = (\phi_{\max} - \phi_{\min})n/N$  and  $\ell$  is the frequency mode number. The power spectrum, which gives the power of each mode, is then given by,

$$P_{\ell} = \frac{1}{N^2} |F_{\ell}|^2. \quad (2.11)$$

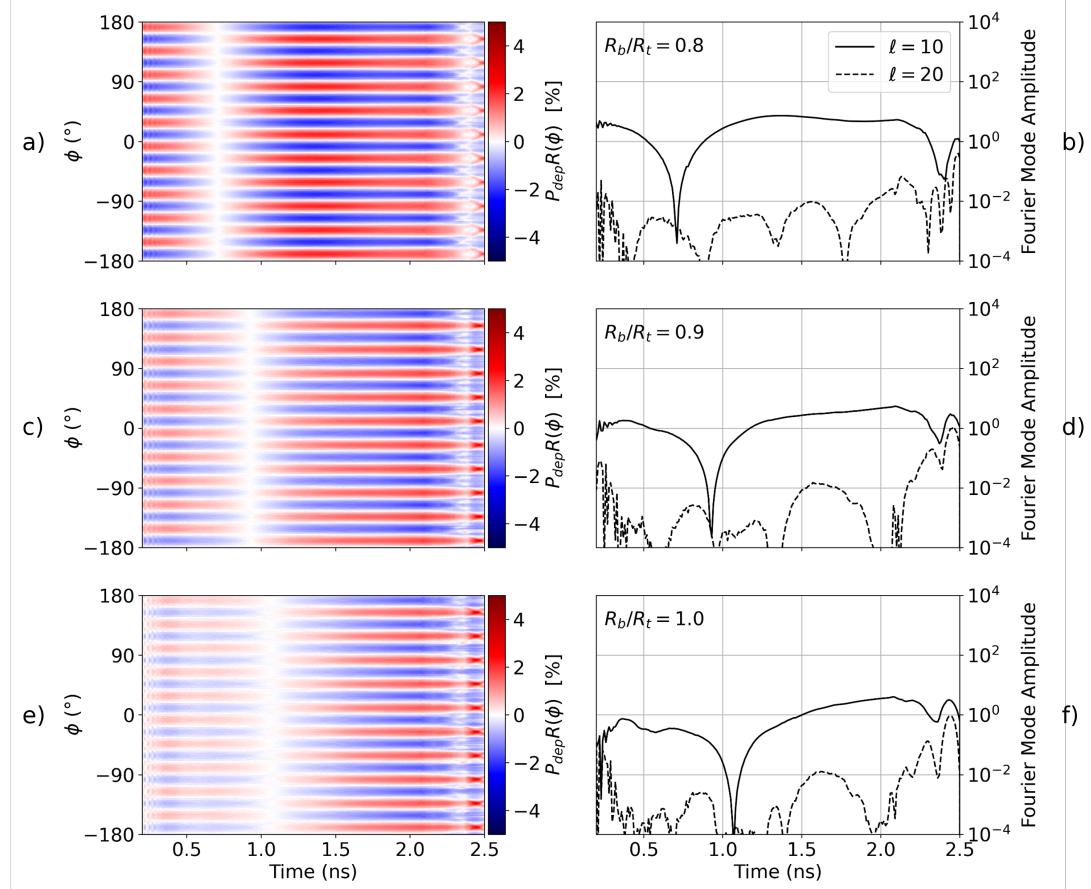


Figure 2.7: Radially integrated deposited power from no-CBET simulations as a function of time ( $x$ -axis) and angle ( $y$ -axis), alongside amplitudes of the dominant modes from a Fourier power spectrum. Panels a) and b) plot the radially integrated deposited power and Fourier modes, respectively, for the  $R_b/R_t = 0.8$  simulation. The same is plotted for the  $R_b/R_t = 0.9$  simulation in c) and d), and for the  $R_b/R_t = 1.0$  simulation in e) and f). The mode 10 from the number of beams is clearly visible in the radially integrated power plots as 10 peaks to troughs in angle at a given time, *i.e.* 10 cyclical perturbations along a vertical lineout.

The power spectra of the deposited power deviations from Fig. 2.6.b are plotted in Fig. 2.6.c on a log scale. The no-CBET profile is dominated by the  $\ell = 10$  mode, with only a small  $\ell = 20$  present. This yields the sinusoidal curve in Fig. 2.6.b. Many more modes are present for the CBET power spectrum, and (compared to the no-CBET results) a clear amplification of multiples of the  $\ell = 10$  is visible. The significant  $\ell = 20$  distorts the curve in Fig. 2.6.b, slightly widening the peaks and narrowing the troughs. Modes with  $\ell < 10$  are presumed to be mostly spurious and introduced by relatively small, instantaneous errors in the field reconstruction algorithm. Unlike the  $\ell = 10, 20, 30, \dots$ , the  $\ell < 10$  exhibit oscillatory, random growth from timestep to timestep. Therefore, they do not significantly imprint on the hydrodynamic profiles over the timescale of the implosion.

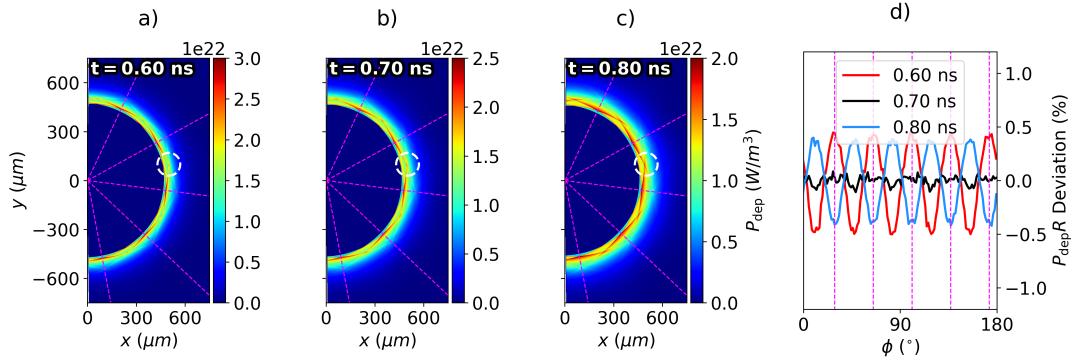


Figure 2.8: Demonstration of a mode-flip in  $P_{\text{dep}}$  for the no-CBET  $R_b/R_t = 0.8$  simulation. Panel a), b) and c) plot the power deposition just before, during and just after the mode-flip. Panel d) plots the deviation from the mean of the radially profiles around the azimuthal angle for all 3 times. It can be seen from this plot that the deposition is very symmetric at  $t = 0.7$  ns. In all four panels, the angles of the beams are shown by the dashed magenta lines. The overlap region where deposition rises as the scale length increases, is highlighted with a dashed circle in panels a), b) and c).

### 2.3.2 Deposition Asymmetries in the Absence of CBET

This section presents results of the deposited power in the absence of CBET for several implosions. Plotted in Figs. 2.7.a, 2.7.c and 2.7.e are  $P_{\text{dep}}R(t, \phi)$  for  $R_b/R_t = 0.8, 0.9$  and  $1.0$ , respectively. Explicitly, this is the radially integrated power from Eq. 2.8, plotted as a function of time ( $x$ -axis) and azimuthal angle ( $y$ -axis). The  $\ell = 10$  deposition asymmetry at a single time (for example the red curve in Fig. 2.6.b) is visible as ten peaks to troughs along a vertical lineout. As expected, comparison of the colour scale saturation between these three plots demonstrates that at smaller beam radii, no-CBET asymmetries are much more significant. Figs. 2.7.b, 2.7.d and 2.7.f plot the power spectrum amplitude of the dominant  $\ell = 10$  and  $20$  modes, as a function of time, on a log scale.

At times  $t \sim 0.7, 0.9$  and  $1.1$  ns for the  $R_b/R_t = 0.8, 0.9$  and  $1.0$  simulations respectively, a modal-flip of the deposited power is observed. This occurs because the plasma scale length increases in time, widening the plasma region above the critical surface where Inv-Brem is significant. Thus, the wings of the beams, which do not penetrate as far radially in, contribute more to deposition after this longer scale-length coronal plasma region has evolved. This eventually leads to a modal flip, when more deposition occurs between beam angles than at the angle of the beam itself. The flips occur later in time for wider beams, because the wings of the beams penetrate less far into the plasma, so a longer plasma scale length must develop before deposition from these edge rays becomes significant.

This is shown explicitly in Fig. 2.8, which plots the  $P_{\text{dep}}$  profiles for the  $R_b/R_t = 0.8$ , no-CBET simulation at  $t = 0.6$  ns,  $t = 0.7$  ns and  $t = 0.8$  ns, i.e. just before, during and after the modal-flip, respectively. Particularly, Fig. 2.8.d plots the radially integrated powers plotted in Figs. 2.8.a, 2.8.b and 2.8.c. Before the modal-flip, at  $t = 0.6$  ns, maximum deposition occurs at the angles of the beams, shown by dashed magenta lines. During the flip, at  $t = 0.7$  ns,

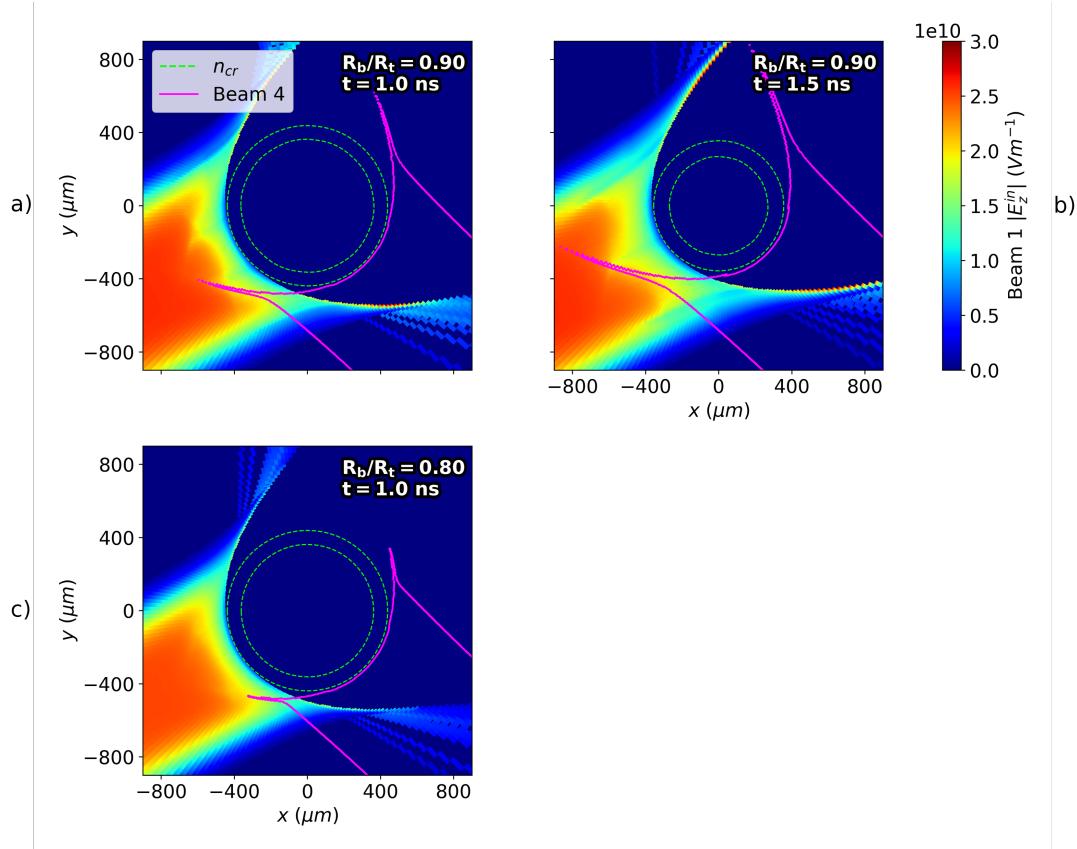


Figure 2.9: Field structure which leads to CBET induced asymmetry on power deposition and its dependence on  $R_b/R_t$  and target convergence. Each panel plots the incident sheet field, which includes the effect of CBET, along with contours of the critical electron density (green) and the incident field,  $|E_z^{\text{in}}| = 1 \times 10^{10} \text{ Vm}^{-1}$  contour of another beam (magenta). Panel a) and b) plot this for the  $R_b/R_t = 0.9$  simulation at  $t = 1.0 \text{ ns}$  and  $t = 1.5 \text{ ns}$  respectively. The convergence of the target in this time interval, leads to beams wrapping around the target more, and therefore a change in the spatial location of the resonant CBET interaction. Panel c) plots the same for the  $R_b/R_t = 0.80$  simulation at the same time as panel a). This demonstrates that the  $R_b/R_t = 0.80$  beam is not wide enough at this time to lead to a resonant CBET interaction, unlike the wider beam in panel a).

very symmetric deposition is observed and just after, at  $t = 0.8 \text{ ns}$ , maximum deposition occurs between beam angles. Examining Figs. 2.8.a, 2.8.b and 2.8.c, the highlighted ‘cross’ feature (white dashed circle) between beam angles becomes increasingly saturated as more Inv-Brem occurs here. This occurs due to the plasma scale length increasing, raising the density further away from the critical surface and thus increasing deposition where the wings of neighbouring beams overlap.

### 2.3.3 CBET Imprint on Incident Field

When CBET is included in these simulations, it acts to significantly alter the field structure of the inbound laser sheets. This leads to additional asymmetry in the deposition, which predominantly occurs closer to the critical surface than the CBET scattering. Extra asymme-

try is introduced, because the resonant CBET interaction is spatially localised, as opposed to azimuthally symmetric around the target. Fig. 2.9.a illustrates this by plotting the electric field magnitude from the incident sheet of ‘beam-1’ from the  $R_b/R_t = 0.9$  CBET simulation at  $t = 1.0$  ns, along with the  $|E_z^{\text{in}}| = 1 \times 10^{10} \text{ Vm}^{-1}$  contour of ‘beam-4’ (magenta line). These two beams are separated from each other by  $108^\circ$ . It can be seen that there are two ‘holes’ in the beam-1 incident field profile, on either side of the beam centre. These are due to CBET interactions with other beams, for example, the resonance with the caustic field region of beam-4 is responsible for the hole at  $[x, y] \sim [-500, -400] \mu\text{m}$ . Typically, a large fraction of CBET scattering occurs in the caustic field region of beams for direct-drive implosions, due to the refractive swelling of the ray amplitude and therefore large electric fields present [28]. Caustic regions are narrow structures, typically with width  $\sim \mathcal{O}(\lambda_0)$ , therefore the region of maximal CBET scattering is strongly localised, leading to the non-uniform imprint on the fields of inbound beams.

The trajectory that the light follows, and therefore the location of the caustic, depends on the electron density profile. For a direct-drive implosion, the critical density falls inward throughout the implosion, leading to a caustic field structure which ‘wraps around’ the target more, *i.e.* rays at an equivalent impact parameter are deflected less, as the critical radius converges. This can be seen by comparing Fig. 2.9.a ( $t = 1.0$  ns) and Fig. 2.9.b ( $t = 1.5$  ns), with critical radii  $r_{\text{cr}} \sim 430$  and  $360 \mu\text{m}$ , respectively. The shrinking of  $r_{\text{cr}}$  has allowed beam-4 to wrap around the target more, such that the caustic CBET interaction now occurs in the middle of the incident field of beam-1. Comparing the field structures from Fig. 2.9.a and Fig. 2.9.b, the deposition of beam-1 at  $t = 1.0$  ns will peak at the incident beam angle due to the depleted field on either side, whereas at  $t = 1.5$  ns, the depletion of the field in the beam centre will shift the  $P_{\text{dep}}R$  maxima in azimuthal angle. In other words a modal flip of the power deposition occurs between these times, which is due to the localised caustic CBET interaction changing location as the target converges.

Fig. 2.9.c plots the same as Fig. 2.9.a, but for the  $R_b/R_t = 0.80$  case, *i.e.* a much narrower beam. From this plot, it can be seen that unlike for the  $R_b/R_t = 0.9$  case, beam-4 does not wrap around the target enough for its caustic region to deplete beam-1, leading to a relatively unperturbed incident field profile. This demonstrates that the CBET induced modal flips are different for implosions with different  $R_b/R_t$ , because narrower (wider) beam will wrap around the target less (more) at an equivalent critical radius. Therefore, larger  $R_b/R_t$  values allow more modal flips to occur, because at a given critical radius, wider beams wrap around more, and therefore can undergo CBET with beams which have a larger angular separation. It is important to note that although this analysis only considers a single beam, the coronal hydrodynamic profiles are approximately azimuthally symmetric and therefore the beams and CBET interactions have rotational symmetry.

### 2.3.4 Deposition Asymmetries in the Presence of CBET

Results shall be presented in this section of the  $P_{\text{dep}}$  asymmetries for simulations including CBET, contrasted to Sec. 2.3.2, which presented results for the no-CBET simulations. Simi-

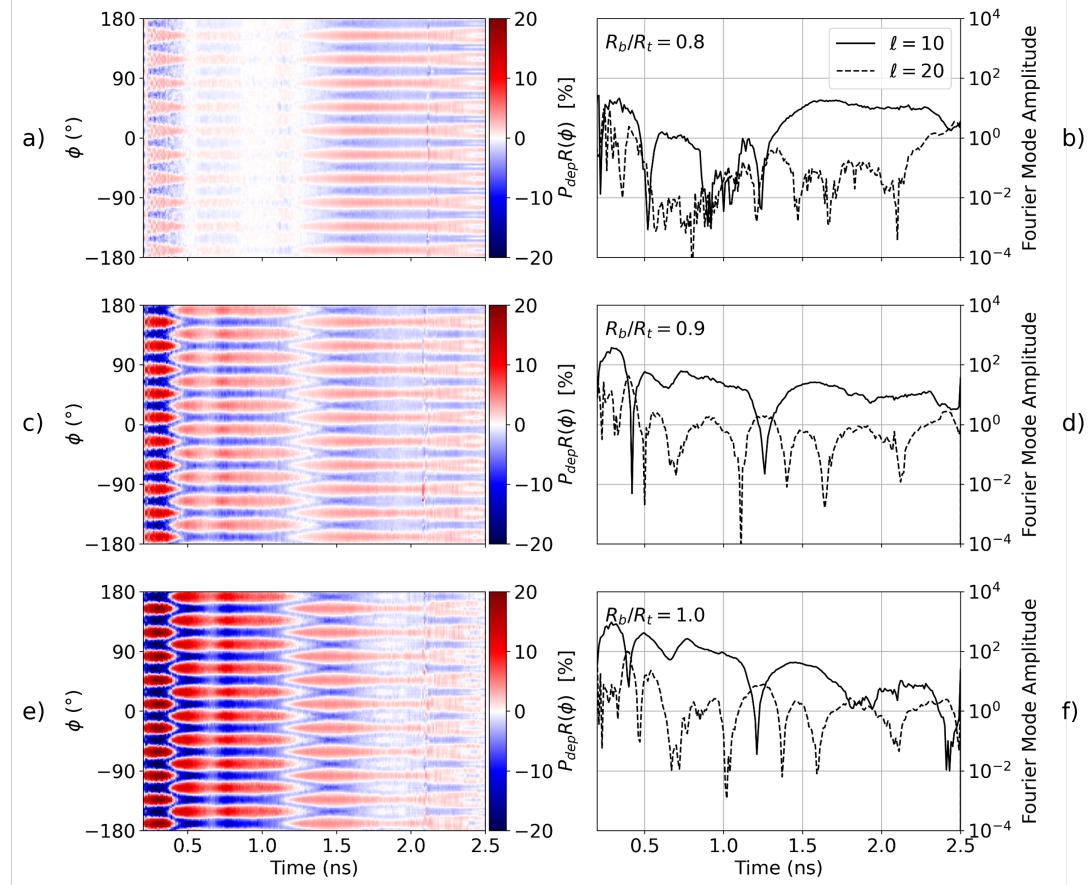


Figure 2.10: This figure plots the same as Fig. 2.7, but now for the equivalent simulations which include the effect of CBET. Comparing these results and those in Fig. 2.7 demonstrates that CBET introduces additional modal-flips of the deposition and amplifies the magnitude of asymmetries.

larly to Fig. 2.7, plotted in Figs. 2.10.a, 2.10.c and 2.10.e are  $P_{\text{dep}}R(t, \phi)$  for  $R_b/R_t = 0.8, 0.9$  and  $1.0$ , respectively. Note that in these plots, the colour scale saturates at 20 % rather than 5 %, which demonstrates that instantaneous deposition asymmetries are typically much larger when the effects of CBET are included. A small discontinuity is visible in these plots at  $t \sim 2.15$  ns, which is the time that the critical density passes from the CH ablator material into the DT fuel. Mix between these materials is not modelled in CHIMERA, so this critical density traversal leads to short-lived, potentially spurious, radially outward  $\nabla n_e$  near the critical radius, which are visible in the field reconstruction and deposition.

The  $P_{\text{dep}}R$  plots in Fig. 2.10 have significantly more structure than the equivalent no-CBET plots in Fig. 2.7. There is an increase in the number of mode-flips, increasing  $R_b/R_t$  appears to increase deposition asymmetry (rather than decrease for no-CBET) and more significant higher modes also appear to be present in  $P_{\text{dep}}R$ . Figs. 2.10.b, 2.10.d and 2.10.f plot the  $P_{\text{dep}}R$  Fourier power spectrum amplitudes of the modes  $\ell = 10$  and  $20$  throughout the implosion. The mode-flips can be identified in these plots as the times at which the  $\ell = 10$  sharply falls and then quickly rises again.

Comparing Fig. 2.7 and Fig. 2.10, CBET leads to a mode-flip in the deposition at  $t \sim 0.4$  ns,

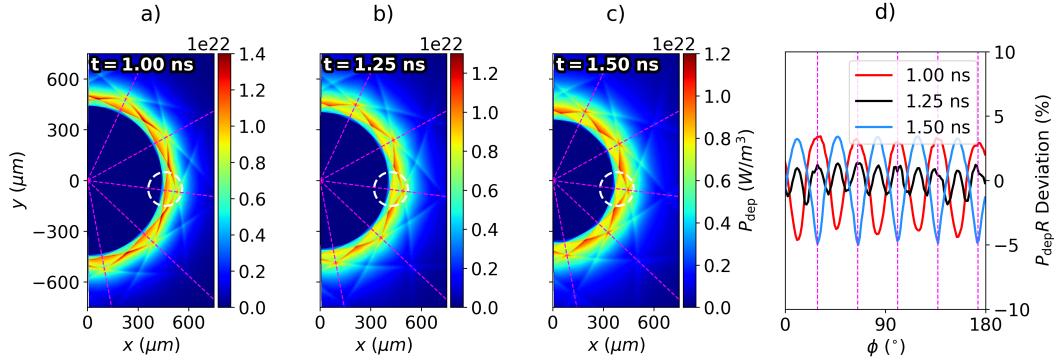


Figure 2.11: Demonstration of a mode-flip in  $P_{\text{dep}}$  for the CBET,  $R_b/R_t = 0.9$  simulation. Panels a), b) and c) plot the power deposition before, during and after the mode-flip. Panel d) plots the deviation from the mean of the radially profiles around the azimuthal angle for all 3 simulations. Fig. 2.9.a and Fig. 2.9.b plot a single, incident sheet field from panels a) and c) respectively. It can be seen that the centre of the beam depletion at  $t = 1.5$  ns leads to less deposition at the beam angles (magenta lines) on panel d). Panel b) shows that during the CBET induced mode-flip, a significant  $\ell = 20$  occurs in the deposition. The reducing absorption at the centre of one beam is highlighted in by the dashed circles.

which does not occur for the no-CBET case. The resulting  $\ell = 10$  in the CBET deposition profiles between  $t \sim 0.4$  ns and  $t \sim 1.2$  ns strongly depends on the  $R_b/R_t$  value. For the  $R_b/R_t = 0.8$ , CBET simulation, the  $\ell = 10$  amplitude is low compared to the larger  $R_b/R_t$ , CBET simulations. This is because the relatively narrow beams at this time do not sufficiently wrap around the target to give rise to strong caustic fields and subsequent CBET imprints on incident beams. This is seen explicitly, by comparing the incident field profiles at  $t = 1.0$  ns for the  $R_b/R_t = 0.9$  and 0.8 simulations plotted in Figs. 2.9.a and 2.9.c respectively.

Interesting behaviour is also observed later in the implosion. Comparing the  $R_b/R_t = 0.9$  (Fig. 2.10.c) and 1.0 (Fig. 2.10.e) plots at  $t \sim 1.8$  ns, it can be seen that an additional mode-flip is beginning to develop for the  $R_b/R_t = 1.0$  case. This is also seen in Figs. 2.10.d and 2.10.f, by the reduction in amplitude of the  $\ell = 10$  and rise of the  $\ell = 20$ . However, the  $R_b/R_t = 1.0$  beams are not quite wide enough to fully translate to an inversion of the deposition asymmetry, so this results in a net decrease in  $P_{\text{dep}}R$  asymmetry, compared to the  $R_b/R_t = 0.9$  simulation at the same time.

Fig. 2.11 plots the development of the mode-flip which occurs at  $t \sim 1.25$  ns. Figs. 2.11.a, 2.11.b and 2.11.c plot the deposited power from all beams before, during and after the flip, respectively. As can be seen in the area highlighted by the dashed circles in each plot, the deposition at the angle of the beams (magenta dashed lines) becomes less significant, compared to the angles between beams. Figs. 2.11.a and 2.11.c are from the same time as the single field profiles plotted in Figs. 2.9.a and 2.9.b, respectively. These field profiles demonstrate that CBET scattering in the caustic region of the beams separated by  $108^\circ$ , is responsible for the depletion of the inbound beams on either side of the beam centre at  $t = 1.0$  ns and in the beam centre at  $t = 1.5$  ns. The radially integrated deposition is plotted in Fig. 2.11.d, which

shows the  $\ell = 20$  of the deposited power during the mode-flip.

## 2.4 Stagnation State Asymmetry

This section describes how the asymmetry of the deposition profiles, described in Sec. 2.3, imprint upon the hydrodynamics. The stagnation profiles for various  $R_b/R_t$  are presented for CBET and no-CBET simulations. These demonstrate that CBET tends to increase the asymmetry of the stagnated state, due to the decreased absorption uniformity. It is observed that CBET changes the trend of the stagnation symmetry with  $R_b/R_t$ . While, for the no-CBET simulations, symmetry increases with increasing  $R_b/R_t$ , CBET generally reverses this behaviour, with the least symmetric stagnation states observed for the largest  $R_b/R_t$  value. The CBET trend is, however, non-monotonic which is due to development of additional mode-flips at specific values of  $R_b/R_t$ , slightly increasing the deposition uniformity late in the implosion.

### 2.4.1 Stagnation State Asymmetry Trend with Beam Radius

Fig. 2.12 plots the stagnation fuel density and ion temperature for both CBET and no-CBET simulations at four different  $R_b/R_t$ . The beam-mode  $\ell = 10$  is clearly identifiable in all plots. Recall that all simulations are tuned to have the same amount of absorbed laser energy, via the tuning process described in Sec. 2.2.3. Therefore, all profiles have approximately the same convergence properties. If 1-D effects of CBET were included, which would involve a much more complex, simultaneous optimisation of pulse and target, the stagnation state of all simulations would likely look far less similar. No-CBET and smaller  $R_b/R_t$  results would have significantly more coupled energy, and would therefore have more mass in the final fuel configuration. Instead, these results effectively show, for hydrodynamically similar implosions, how CBET and increasing  $R_b/R_t$  affects deposition and stagnation *symmetry* only. This simpler optimisation also has the advantage that the final fuel configurations do appear more similar and are therefore directly comparable.

The no-CBET results clearly show a trend of increasing stagnation symmetry with increasing  $R_b/R_t$ , as expected from the plots in Fig. 2.7, which showed increasing absorption uniformity with increasing beam radius. The pressure of the stagnation state is approximately isobaric, and therefore the temperature profiles are inversely proportional to the density. Peak densities increase and the radii of stagnation decrease marginally as  $R_b/R_t$  increases. This improved compression is partially due to increased symmetry providing better compression, but more optimal shock timing for the larger  $R_b/R_t$  implosions also contributes. As is seen in the streak plots in Fig. 2.5, the tuning process led to the first shock hitting the axis slightly earlier at narrower  $R_b/R_t$ , than for the larger  $R_b/R_t$  implosions. Although the streak plots are for 1-D with-CBET simulations, the same trend is observed for the 2-D with-and without-CBET simulations. The more optimal shock timing for larger  $R_b/R_t$  also has the signature of higher on-axis densities. The contribution of small differences in shock timing is not quantified in the following analysis, but it is assumed that it will have a second-order

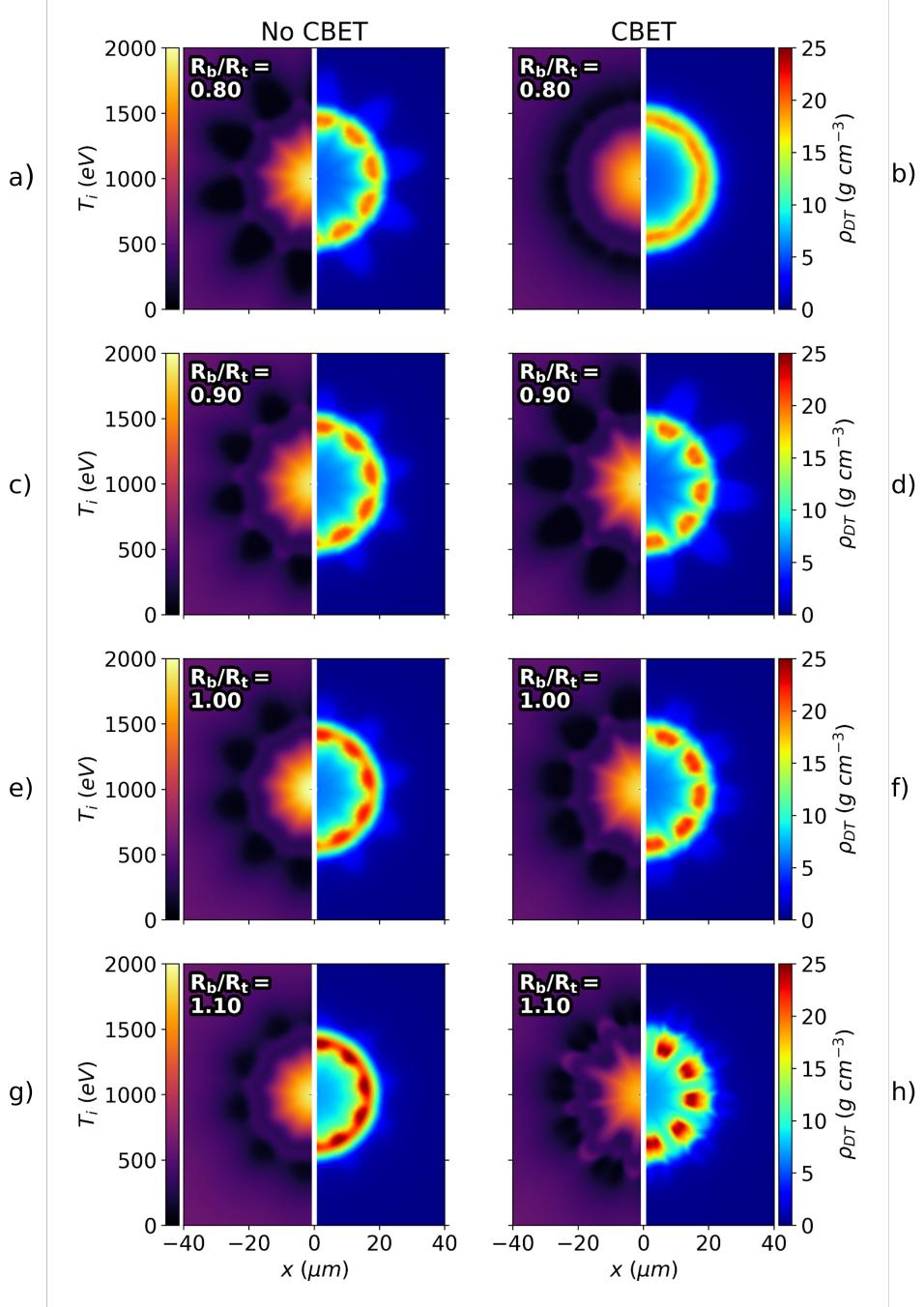


Figure 2.12: Densities of the DT fuel and ion temperatures for various  $R_b/R_t$  simulations both with and without CBET. Each row corresponds to a different  $R_b/R_t$  value; the left column contains simulations without CBET; and the right column contains simulations with CBET. It is visible from the density plots that increasing  $R_b/R_t$  improves stagnation symmetry for the no-CBET simulations, but degrades it for the CBET simulations.

effect on the stagnation uniformity, compared to asymmetry in deposition.

Increasing  $R_b/R_t$  when including CBET broadly shows the opposite behaviour to the no-CBET simulations, leading to highly non-uniform density profiles at large  $R_b/R_t$ . This trend was also observed in the deposition plots in Fig. 2.10, with the wider beam simulations lead-

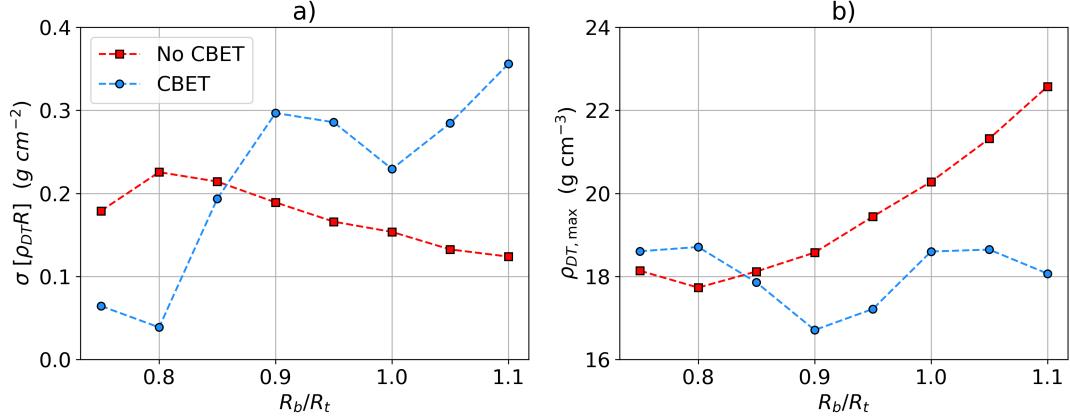


Figure 2.13: Trends of a) stagnation asymmetry and b) maximum (azimuthally averaged) fuel density for CBET and no-CBET simulations. The no-CBET improvement in symmetry with  $R_b/R_t$  is observed which also corresponds to improved compression. The symmetry trend including CBET is more complex, but broadly the stagnation state symmetry is worse with increasing  $R_b/R_t$ .

ing to a more saturated plot on the colour scale. Higher order modes than  $\ell = 10$  also become increasingly evident in the wider beam CBET simulations. Note that the  $R_b/R_t = 0.8$  simulation with CBET is more symmetric than the no-CBET simulation at the same beam radius. This is believed to be due to the developing mode-flip, visible in Fig. 2.10.a from  $t \sim 0.5 \rightarrow 1.3$  ns, which strongly reduces the absorption asymmetry and subsequent imprinting onto the density.

The standard deviations of the radially integrated stagnation state fuel density,

$$\sigma[\rho_{DT}R] = \sqrt{\langle \rho_{DT}R^2 \rangle - \langle \rho_{DT}R \rangle^2}, \quad (2.12)$$

for the CBET and no-CBET simulations are plotted as a function of  $R_b/R_t$  in Fig. 2.13.a. The no-CBET trend from this metric clearly follow the behaviour seen in Fig. 2.12, that increasing beam width results in better symmetry of the hydrodynamic profiles at stagnation. Note that the  $R_b/R_t = 0.75$  simulation value is more symmetric than the 0.8 result, suggesting that more complex behaviour may occur at smaller beam radii,  $R_b/R_t < 0.8$ . However, this is outside the range of typical OMEGA implosions and therefore not studied in detail. The with-CBET curve shows a broad trend of increasing asymmetry with wider beams, which is visible in the stagnation profiles in Fig. 2.12, however the behaviour is clearly more complex than the no-CBET simulations. A local maximum of asymmetry is observed at  $R_b/R_t = 0.9$  and minima occur at  $R_b/R_t = 0.8$  and  $R_b/R_t = 1.0$ . As shall be described in the subsequent section, Sec. 2.4.2, these features are explained by the complex, CBET-induced modal-flips of the deposition which were described in Sec. 2.3.4.

The azimuthally averaged maximum density,

$$\rho_{DT,\max} = \max \left( \int_{-\pi}^{\pi} \rho_{DT}(r, \phi) d\phi \right), \quad (2.13)$$

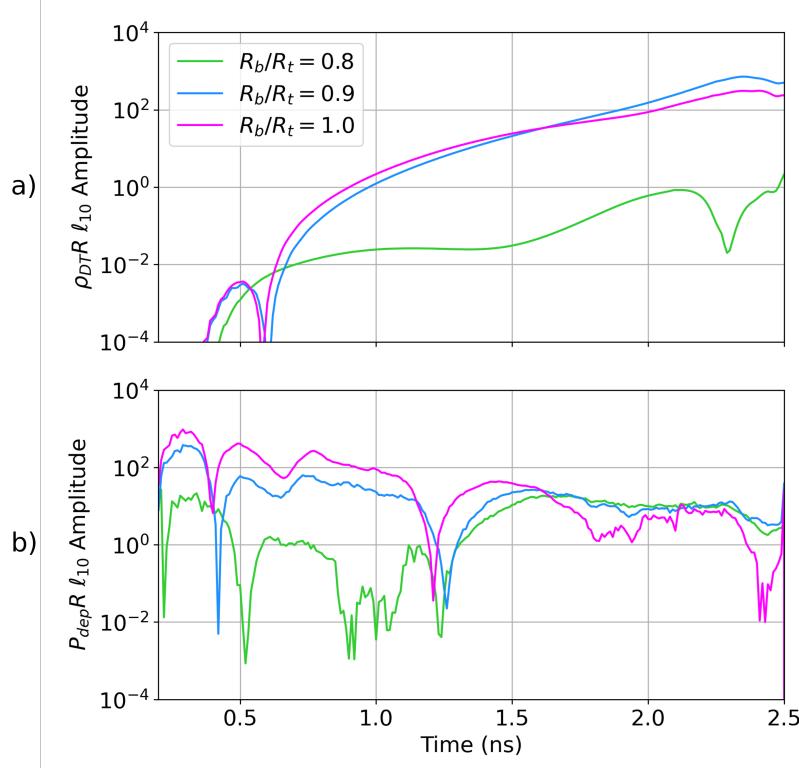


Figure 2.14: Time resolved  $\ell = 10$  Fourier power spectrum amplitude for a)  $\rho_{DT}R$  and b)  $P_{dep}R$  for CBET simulations with 3  $R_b/R_t$  values. The developing, but unrealised, modal-flip for  $R_b/R_t = 0.8$  from  $t \sim 0.5 \rightarrow 1.2$  ns reduces the  $P_{dep}R_{\ell=10}$ , leading to slow  $\rho_{DT}R_{\ell=10}$  growth and ultimately a relatively symmetric stagnation state. Despite large values of  $\rho_{DT}R_{\ell=10}$  initially, the developing modal flip of the  $R_b/R_t = 1.0$  simulation from  $t \sim 1.8 \rightarrow 2.1$  ns slows the density asymmetry growth.

is plotted in Fig. 2.13.b for both sets of simulations. The behaviour in this plot is inversely proportional to the standard deviation, which suggests that the asymmetry of the density has a detrimental effect on the compression of the target. Lower modes can prove detrimental to compression by introducing significant non-radial motion, resulting in less efficient compression [114, 115]. The (unintended) improved shock timing at larger beam radii does appear to somewhat compensate for the reduction in uniformity, *i.e.* the compression metric at  $R_b/R_t = 1.05$  is about the same as  $R_b/R_t = 0.8$ , despite far worse symmetry. It is therefore difficult to quantify the contribution to the degradation of compression due to each of these effects, from this set of simulations.

#### 2.4.2 Time Resolved Asymmetry Growth

The time-resolved,  $\ell = 10$  mode growth of the fuel density from the  $R_b/R_t = 0.8$ , 0.9 and 1.0, with-CBET simulations are examined more closely in this section in order to understand the non-monotonic behaviour of the CBET, fuel-symmetry curve from Fig. 2.13.a. These simulations were chosen as they are the local minima and maxima of the curve and therefore should most clearly display the important behaviour. Fig. 2.14.a plots the  $\ell = 10$  growth of

$\rho_{DT}R$  for each of these simulations. Note that all bangtimes were approximately  $t = 2.5$  ns, so the value at this time is the  $\ell = 10$  at stagnation. The  $\ell = 10$  growth of  $P_{dep}R$  for these simulations is plotted in Fig. 2.14.b, which are the same as the curves from the right panels of Fig. 2.10, but are plotted again here for direct comparison with the  $\rho_{DT}R$  values. The mode-flips are clearly identifiable in this plot as sharp falls and rises in the  $\ell = 10$ , which corresponds to an inversion of deposition asymmetry.

Firstly, the growth of the fuel density asymmetry from the  $R_b/R_t = 0.8$  simulation shall be studied. It can be seen from the green curve in Fig. 2.14.a, that the growth of this mode is slow from  $t \sim 0.5 \rightarrow 1.4$  ns, compared to the  $R_b/R_t = 0.9$  and  $R_b/R_t = 1.0$  simulations. This is explained due to the low deposition asymmetry in Fig. 2.14.b which is ultimately responsible for imprinting the asymmetry on the hydrodynamics. As was seen from the time-resolved, radially integrated deposition in Fig. 2.10.a, this low asymmetry of the deposition is due to the developing mode-flip. In this simulation, the developing mode-flip at this time ultimately reduces deposition asymmetry, because the beam is not wide enough for the target convergence to truly flip the deposition. Although the  $\ell = 10$  of the  $\rho_{DT}R$  starts to grow again after the second mode-flip at  $t \sim 1.2$  ns, the long period of highly symmetric deposition means that the stagnation state is ultimately relatively symmetric, compared to higher and lower beam radii simulations.

Comparison of the growth between the  $R_b/R_t = 0.9$  and  $1.0$  simulations is slightly more nuanced. As is seen in the radially integrated power deposition from Figs. 2.10.c and 2.10.e, the mode-flip at  $t \sim 0.5$  ns develops into a full inversion of the deposition asymmetry. Because the beams are wider in the  $R_b/R_t = 1.0$  simulation, more light wraps around the target. This leads to stronger caustic fields, more CBET and therefore greater distortion of the inbound field profiles. Thus, the  $R_b/R_t = 1.0$  has a less symmetric deposition profile than the narrower beam case, until the next mode-flip occurs at  $t \sim 1.2$  ns. This explains why the  $\ell = 10$  of the  $\rho_{DT}R$  is larger for the  $R_b/R_t = 1.0$  simulation up until  $t \sim 1.5$  ns.

This remains the case until the wider beam of the  $R_b/R_t = 1.0$  case begins to wrap around the target sufficiently to begin the development of an additional mode-flip of the deposition at  $t \sim 1.6$  ns. Similar to the  $R_b/R_t = 0.8$  case from  $\sim 0.5 \rightarrow 1.2$  ns, the development of a flip in the deposition profile acts simply to reduce the deposition asymmetry, since the beam is not sufficiently wide. Therefore, the deposition asymmetry of  $R_b/R_t = 1.0$  from  $\sim 1.6 \rightarrow 2.1$  ns is reduced compared to the  $R_b/R_t = 0.9$  simulation, for which the beams are not wide enough to develop another mode-flip. The reduction of deposition asymmetry is again clearly visible by comparing Figs. 2.10.c ( $R_b/R_t = 0.9$ ) and 2.10.e ( $R_b/R_t = 1.0$ ) from  $\sim 1.6 \rightarrow 2.1$  ns. More symmetric deposition during this period reduces the rate of  $\rho_{DT}R$  asymmetry growth. This results in a less symmetric  $R_b/R_t = 0.9$  stagnation state, as this simulation does experience the same disruption to the deposition asymmetry source. For  $R_b/R_t > 1.0$ , the beams are wide enough that the mode-flip at  $t \sim 1.6$  ns does lead to a full inversion of the deposition asymmetry and therefore the  $\ell = 10$  of the  $P_{dep}R$  is not reduced in the same manner for even wider beams. Ultimately, this leads to increasing asymmetry of the stagnation state at  $R_b/R_t > 1.0$ , as is observed in Fig. 2.13.a.

While this analysis does explain the trend observed in Fig. 2.13.a, subtleties exist which are

not explicitly addressed, but could offer additional insight to the growth of stagnation state asymmetries in direct-drive implosions. For example, the deposition all occurs below the critical density, which is separated from the ablation surface by a finite stand-off distance. Thus, the energy must be transported from the absorption region to the ablation surface by thermal conduction. This leads to a time-lag for asymmetry in Fig. 2.14.b (deposition) to propagate through to the curves in Fig. 2.14.a (fuel density). Changing coronal plasma conditions during the implosion will also change this time-lag in a non-obvious way. For example, early in the implosions, deposition occurs exclusively in the outermost CH material. As time progresses, the CH expands away and the ablation surface changes from CH to the DT layer. Even later, (at  $t \sim 2.1$  ns in these simulations), the critical surface moves from CH into DT. All of these changes in coronal and transport layer conditions will impact how conduction propagates asymmetry from the deposition to the fuel, *i.e.* there is additional complexity in mapping from  $P_{\text{dep}}R$  asymmetry to  $\rho_{\text{DT}}R$  asymmetry. Future work could perhaps investigate this further, by also comparing the heat flux just above the ablation surface, which may further elucidate the important physics.

### 2.4.3 Comparison of Results to Statistical Modelling

Ultimately, this work was conducted to explain why statistical modelling of OMEGA implosions demonstrates that distinct regimes exist in the  $R_b/R_t$  degradation of implosions from 1-D calculations, *i.e.* to explain the shape of the curve in Fig. 2.3. The fit to the experimental scatter plot has a transition at  $R_{b/t}^{\text{crit}} = 0.86$ , after which, improvement of the yield for increasing  $R_b/R_t$  slows. Neither 1-D codes which include CBET models, or 3-D codes without CBET, are able to reproduce this trend, which means that multidimensional effects, seeded by CBET, are considered a likely explanation [96]. The work conducted in this chapter is not directly comparable to these plots because a 1-D tuning process was conducted to ensure that all implosions had the same coupled energy. This was to prevent having to perform a far more in depth optimisation procedure for both the pulse and target simultaneously. Therefore, changes in the 1-D effects of CBET at different  $R_b/R_t$  are not included in these simulations (changes to magnitude of deposited energy), only the redistribution of deposited energy due to CBET, *i.e.* how CBET alters asymmetry. The differences in cylindrical and spherical geometry also complicate direct comparison between results due to different coronal plasma profiles and convergence properties. For example, the  $R_b/R_t$  values of transitions in behaviour would not be expected to be the same between geometries.

Despite this, the asymmetry and compression metrics in Figs. 2.13.a and 2.13.b, respectively, clearly demonstrate relatively sharp transitions in behaviour. For example, asymmetry and compression both get worse from  $R_b/R_t = 0.8 \rightarrow 0.9$ , but improve from  $0.9 \rightarrow 1.0$ . If this behaviour was also observed in spherical geometry simulations, which is due to CBET introducing mode-flips in the deposited power, it could potentially be the cause of the experimental data transition at  $R_{b/t}^{\text{crit}} = 0.86$ . Explicitly, if CBET acted to degrade deposition symmetry in spherical implosions from  $R_b/R_t = 0.86 \rightarrow 1.00$ , this would slow the rate of improvement in the yield over clean metric, plotted in Fig. 2.3. While this is not at all a conclu-

sive explanation of the behaviour in the statistical model, results from these simulations do not seem to rule it out. It therefore offers motivation for extending the simulation platform to a spherical geometry and including 1-D effects of CBET in  $R_b/R_t$  trends by changing the target and fixing the total incident laser energy. This should conclusively demonstrate that shape degradation due to CBET either is or is not responsible for the transitions.

## 2.5 Conclusions

### 2.5.1 Summary of work

This chapter has described the development of a novel, 2-D cylindrical simulation platform to study direct-drive implosion physics, and its use to investigate how altering  $R_b/R_t$  changes the stagnation symmetry both with, and without the effects of CBET. The work was conducted to test the hypothesis that changes to stagnation state symmetry due to changing CBET action at different  $R_b/R_t$  were responsible for unexplained behaviour in statistical modelling of OMEGA experiments. In order to avoid performing an in-depth optimisation process, where the target and pulse were simultaneously varied, only the incident laser energy was varied, to fix the coupled energy to the target. Therefore, only the changes to deposition symmetry due to CBET were studied at different  $R_b/R_t$ . This 1-D tuning procedure yielded eight implosions both with and without CBET effects.

By conducting 2-D simulations with the initial conditions obtained from the tuning, interesting behaviour was observed for both the CBET and no-CBET cases. In both cases, mode-flips of the deposition were observed, due to different effects. For no-CBET, these mode flips occurred due to increasing plasma scale lengths, which shifted maximum deposition from beam angles to the overlap region between beams. When including CBET, this scale-length behaviour was obscured by much more dominant mode-flips, which were induced due to caustic CBET interactions distorting the incident field profiles of beams.

Without CBET changes to deposition uniformity, the stagnation state asymmetry decreased with increasing  $R_b/R_t$ , as more beam overlap led to increasingly uniform target illumination. When CBET was included however, increasing CBET with wider beam radii resulted in less and less symmetric deposition, which broadly led to increasing asymmetry with  $R_b/R_t$ . The precise behaviour was, however, non-monotonic due to CBET mode-flips increasing the symmetry of illumination at specific  $R_b/R_t$  values. This led to distinct transitions regions in the trends of symmetry and compression, plotted against  $R_b/R_t$ , which were identified as a potential explanation of the transitions observed in OMEGA statistical modelling. Further work must however be conducted to obtain direct comparison to the statistical model and thus conclusively state whether CBET asymmetries are, or are not, the cause.

### 2.5.2 Future Work

There are many ways that the work presented in this chapter could be extended, through e.g. improving the simulation platform improved to better reflect reality, or studying additional physics on the existing platform. One interesting way to extend the analysis of the

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work would be to investigate how heat transport affects the conversion of deposition asymmetries into hydrodynamic perturbations. As was discussed at the end of Sec. 2.4.2, there is a lag in the conversion due to thermal conduction taking this information inward via a finite transport region. The transport region would also vary throughout the implosion as the hydrodynamic profiles evolve and the material changes from CH to DT. By including analysis of heat flux, the way that this information is propagated could be studied. Furthermore, a model for non-local heatflow could be incorporated into CHIMERA, such as SNB [116–118] or FAST-VFP [119]. Due to the strong temperature gradients in the transport layer, non-local effects could significantly affect how this information is propagated to the ablation surface and thus imprints on the hydrodynamics.

The geometry of the platform could also be modified to allow for spherical rather than cylindrical convergence. This would allow for better comparison between simulation results and spherical implosion experiments. Cylindrical geometry leads to greater target convergence for a given incident energy and also less divergence of the outflowing, coronal plasma, thus yielding higher coronal plasma densities for a given ablation velocity. By simulating a thin, equatorial ‘slice’ of a spherical domain, *i.e.* a spherical simulation with bounds  $\phi \in [-\pi, \pi]$  and  $\theta \in [\lesssim \pi/2, \gtrsim \pi/2]$ , a 2-D ray-trace could still be conducted so simulation run-times should be very similar to the cylindrical case. This was not done for the work in this chapter because the SOLAS 2-D ray-trace assumes that the simulation occurs in a plane, and the spherical ‘slice’ is not a plane. Minor development for this specific case could rectify this however and enable similar simulations in spherical geometry.

Similar initial conditions could also be used to study further multidimensional direct-drive physics, without the need to conduct, an expensive, full 3-D ray-trace. For example, laser imprint is a significant issue in direct-drive and means that implosions must be conducted at a high adiabat for optimal yields [120]. This limits the compressibility and therefore maximum performance of ICF experiments [121, 122]. Improved understanding of laser imprint requires high resolution simulations, due to the small spatial scale of the intensity modulations. Typically, high resolution, 2-D,  $r - z$  simulations with a 3-D laser ray-trace [123], or lower resolution, fully 3-D simulations [124] are conducted to study the effect of imprint. 2-D  $r - z$  Rad-Hydro simulations with a 3-D laser ray-trace average the deposition from the 3-D ray-trace around the azimuthal angle, and therefore artificially smooth out the imprint modulation to power deposition. While 3-D simulations at a sufficiently high resolution should resolve the imprint seed properly, these are extremely expensive to perform, particularly for the fixed angular resolution, Eulerian code ASTER which is typically used for these problems. By performing a spherical or cylindrical ‘slice’ simulation with an imprint model, the seed and the growth of imprint induced instabilities could be consistently modelled at high resolutions, providing a complimentary study to the existing literature. This setup would not include out-of-plane stabilisation flows however, and could therefore overestimate growth of these small modes.

### **3 Simulations of Cross-Beam Energy Transfer for Magnetised Direct-Drive**

This chapter describes a set of simulations which were conducted to understand the role of CBET in magnetised, direct-drive implosions. Magnetised ICF is a promising route to achieving higher target gains, due to the reduction of hotspot thermal energy loss at stagnation and additional confinement of the alpha particles responsible for burn propagation. For direct-drive implosions, magnetisation can significantly alter the coronal plasma conditions, due to the anisotropisation of thermal transport. The IAW dispersion relation, which mediates CBET interactions, depends upon the background plasma and therefore significantly altered temperature and density profiles could alter the action of CBET. Before the development of SOLAS, no direct-drive suitable CBET model existed, which was integrated into a Rad-MHD code. Therefore, the CHIMERA-SOLAS framework has enabled the effect of magnetisation on CBET to be studied for a direct-drive implosion.

The chapter begins with a review of experimental and computational work on magnetised ICF, with a particular focus on magnetised direct-drive. Work presented in this chapter focuses on the study of ‘exploding-pusher’ experiments. These are very different implosions to the typical central hotspot ignition designs, presented in previous chapters, so a short summary of exploding-pushers is also provided. 1-D and 2-D simulation results are presented of unmagnetised exploding-pushers, both with and without the effect of CBET, which demonstrate that CBET does significantly affect these implosions. This is followed by an investigation of how various extended-Magnetohydrodynamics (MHD) terms alter the implosion, including the Nernst effect, the Lorentz force and resistive diffusion of the magnetic field. Results are given of how magnetisation affects the CBET interaction and ultimately how it changes the stagnation shape of the target. The results demonstrate that redistribution of deposited power due to CBET, reduced the amplitude of the stagnation asymmetry, which originated from the polar beam configuration used. However, the reduction of asymmetry was consistent for different initial seed magnetic field values, and therefore CBET was not observed to be sufficiently strongly affected by magnetisation, to lead to observable signatures in experimental measurements. The chapter concludes with a summary of the work conducted, and suggestions of additional experimental configurations, which may leave a more significant signature of magnetisation altering CBET.

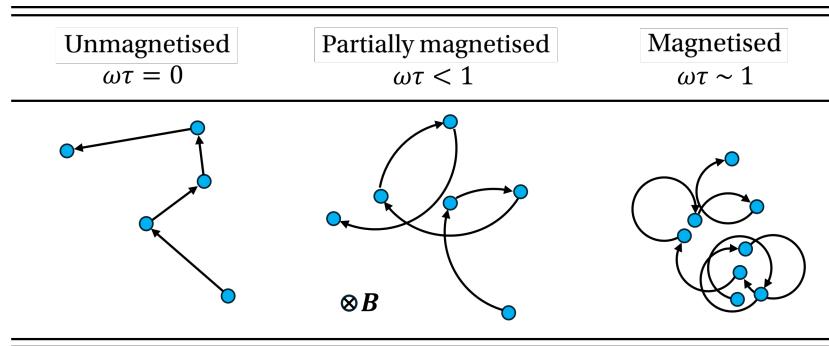


Figure 3.1: Cartoon to illustrate the effect of magnetisation on collisions, and therefore transport, of a test positive charge. Particle locations after collision are represented as blue circles and the path taken by the particle is shown by the black arrows. As the Hall parameter of the particle increases, diffusion is increasingly limited, and therefore collision transport is reduced.

### 3.1 Magnetised Inertial Confinement Fusion and Exploding-Pushers

This chapter begins with a review of published studies of relevance to the work conducted in this chapter. Firstly, a short review of magnetised-ICF is presented, which reviews both the key concepts, existing studies and potential challenges of the design. Both work on direct- and indirect-drive is summarised, alongside recent theoretical progress on understanding how magnetisation can effect LPs. The exploding-pusher concept is then briefly summarised to aid understanding of the implosion physics, which is markedly different to conventional hotspot ICF.

#### 3.1.1 Potential Benefits of Target Magnetisation

Magnetisation of an ICF target has long been thought of as a potential aid to ignition [125, 126]. It is still a relevant field of study in the context of regular ignition events on the NIF, because by relaxing the ignition threshold, magnetisation could make larger targets feasible at an equivalent laser energy, and therefore lead to higher gains than unmagnetised implosions. For a central hotspot ignition targets, ignition occurs when the heat source of alpha energy deposition balances the thermal and radiative losses in the hotspot. Thermal conduction is suppressed perpendicular to magnetic field lines, therefore a magnetic field can reduce thermal losses and aid the power balance required for ignition. Fig. 3.1 demonstrates the effect of increasing magnetisation on a unit positive test charge. By constraining charged particles to orbit field lines, collisional transport terms, such as thermal conduction, are reduced perpendicular to the field direction. Fits of transport coefficients to Fokker-Planck simulations, demonstrate that in a Hydrogen plasma, thermal conductivity perpendicular to field lines,  $\kappa_{\perp}$ , is reduced to  $\sim 30\%$  of the parallel value,  $\kappa_{\parallel}$ , at Hall parameter,  $\omega\tau = 1$ , and  $\sim 1\%$  at  $\omega\tau = 10$  [127]. Thus, for Hall parameters,  $\omega\tau \gtrsim 10$ , thermal conduction losses are almost negligible in the direction perpendicular to field lines.

Using an order of magnitude estimate for a below ignition threshold hotspot,  $T_e \sim 2.5$  keV

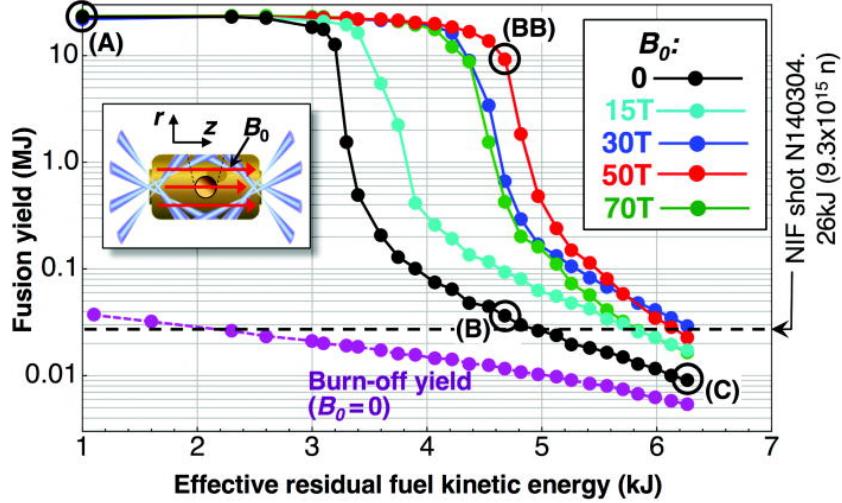


Figure 3.2: Simulated fusion yields versus effective residual fuel kinetic energy under imposed low-mode radiation flux perturbations for imposed fields in the range  $B_0 = 0 \rightarrow 70$  T. The plot demonstrates, that with increasing departure from ideal compression (moving to the right on the  $x$  axis), magnetisation can enable the onset of the ignition. Reused with permission from Ref. [6].

and  $\rho \sim 50 \text{ g cm}^{-3}$ , a field strength  $|\mathbf{B}| \sim 2.5 \text{ kT}$  is required to obtain  $\omega\tau \sim 1$  [128]. This field strength cannot be produced directly, but it is possible to produce a smaller field which, assuming frozen in magnetic field and a spherical compression, is amplified by the square of the convergence,

$$|\mathbf{B}_1| = |\mathbf{B}_0| \left( \frac{R_0}{R_1} \right)^2, \quad (3.1)$$

where  $|\mathbf{B}_0|$  and  $|\mathbf{B}_1|$  are initial and final magnetic fields, respectively and  $R_0$  and  $R_1$  are initial and final radii, respectively. Laboratory magnetic fields can be produced from pulsed power coils with field strength  $|\mathbf{B}| \sim \mathcal{O}(50)$  T [129], so even moderate convergence-ratio targets ( $R_0/R_1 \sim 10$ ) are able to produce strongly magnetised core plasma.

Fig. 3.2 plots results of magnetised indirect-drive simulations, for a target on the threshold of ignition [6]. Increasing magnitude of radiation perturbation were applied to the drive (moving to the right on the  $x$ -axis), which prevent the target from achieving ignition, which is visible as the steep increase in yield, below some threshold level of perturbation. The results demonstrate that when an initial magnetic field was applied to the target, it more robustly ignited with increasing field strength, due to reduced conduction losses. This simulation work, prior to the achievement of ignition on the NIF [14], motivated the development of a magnetised ICF campaign at LLNL [130].

The CHIMERA code has been used to study a wide array of physics relevant to magnetised ICF. Simulation work has been conducted, which has shown that magnetisation can alter instability growth of magnetised laser fusion implosions. While in the deceleration phase, magnetic tension can reduce low-mode perturbation growth [131], magnetisation of directly-driven targets inhibits heatflow in the plasma corona and thus limits thermal stabilisation of short wavelength modes from laser imprint [132]. Recent work has also demon-

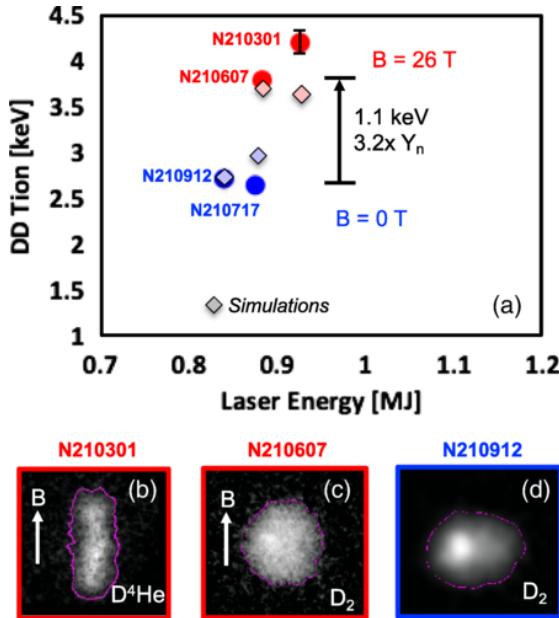


Figure 3.3: a) A 1.1 keV  $T_i$  increase was achieved by adding a 26 T  $B_0$  field to a  $D_2$  gas capsule implosion on the NIF. Also shown in the plot are the simulation results. b)–d) Equatorial shapes of the implosions. Reused with permission from Ref. [7].

strated that magnetisation of high-yield, indirect-drive targets must be carefully optimised, in order to avoid significant degradation to the implosion shape, due to anisotropic thermal conduction and inhibition of burn propagation from  $\alpha$  magnetisation [128].

### 3.1.2 Experimental Studies of Magnetised-ICF

Indirect-drive experiments have been conducted on the NIF to demonstrate the efficacy of magnetised targets, in reducing thermal conduction losses in the hotspot. Non-cryogenic, deuterium filled capsules were deployed with initial field strengths up to 26 T [7]. Results from this experimental campaign are show in Fig. 3.3.a. The magnetised targets demonstrated significantly enhanced ion temperatures and neutron yields and work is underway to explore non-uniform field configurations to further enhance the benefits of magnetisation [133]. Fig. 3.3.b, 3.3.c and 3.3.d plot x-ray images at stagnation of different experiments, showing that a shape-tuning process had to be conducted in order to optimise the sphericity of the target, due to the field leading to anisotropic thermal conduction.

Magnetisation of direct-drive targets has been investigated by experiments on the OMEGA laser facility for a number of years. Initial OMEGA experiments focussed on verification of magnetic flux compression, by applying an initial seed field along the axis of a cylinder that was imploded via laser irradiation [134]. The magnetised implosions validated predictions of flux compression and demonstrated enhanced neutron yields and core ion temperatures over unmagnetised implosions. Spherical targets were subsequently fielded, which also resulted in increased stagnation temperatures and yield compared to unmagnetised targets [135, 136]. No noticeable degradation to the implosion shape or performance was observed in these experiments, which was assumed to be due to the high ratio of plasma

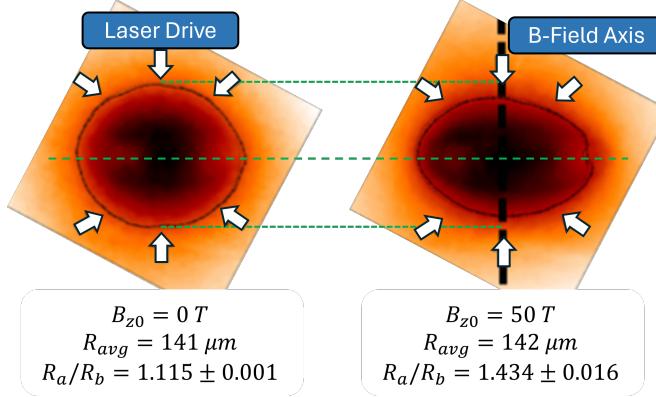


Figure 3.4: X-ray self emission images of (left) an unmagnetised and (right) a magnetised implosion. The average radius of the marked contour (corresponding to 40% of peak intensity), and the oblateness parameter  $R_a/R_b$  (ratio of major-to-minor axis) are listed below each image. The polar laser-drive is indicated by the white arrows, and the axis of the initial magnetic field by the black dashed line on the right. Applying an initial magnetic field demonstrated increased oblateness of the implosion. Adapted with permission from Ref. [8].

pressure to magnetic pressure,  $\beta \gg 1$ .

The most recent experimental, magnetised direct-drive work has focussed on exploring higher initial seed field values ( $|\mathbf{B}_0| \sim 50 \text{ T}$  compared to  $|\mathbf{B}_0| \sim 8 \text{ T}$ ), to understand the saturation of performance with increasing field. A shock-driven, exploding-pusher target configuration was used for these experiments, in order to create high ion temperatures and thus create a platform to study magnetised ions. Exploding-pushers are significantly different implosions compared to hotspot ignition targets discussed in previous chapters and shall be described in detail in Sec. 3.1.4. Creating these strong, 50 T fields at the target necessitated reducing the radius of the equatorial field coil compared to previous experiments. Therefore, a 40-beam configuration had to be used, without the 20 equatorial beams, leading to a pole-heavy laser drive. The high fields of these implosions led to strongly magnetised coronal electrons,  $\omega_e \tau_e \sim 50$ , resulting in strongly anisotropic thermal conduction  $\kappa_{\perp,e}/\kappa_{\parallel,e} \sim 10^{-4}$ . This is compared to previous experiments which produced  $\omega_e \tau_e \sim 1$  and therefore  $\kappa_{\perp,e}/\kappa_{\parallel,e} \sim 1/3$ . For direct-drive on OMEGA, laser deposition is transported to the ablation surface by electron thermal conduction, thus large electron Hall parameters led to an effective anisotropisation of the implosion drive.

Fig. 3.4 shows x-ray self-emission images of an unmagnetised (left) and magnetised (right) target with an initial  $|\mathbf{B}_0| = 50 \text{ T}$  seed field. The strongly magnetised coronal electrons led to decreased drive  $\perp \hat{\mathbf{B}_0}$ , markedly increasing the oblateness of the diagnostic image compared to the unmagnetised target. An ion magnetisation of  $\omega_i \tau_i \sim 7$  was also reported. Previous Rad-MHD modelling of these experiments, using the CHIMERA code, did not include the effects of CBET. The development of SOLAS, and particularly the CBET model, motivated further computational study of these experiments, to explore whether CBET played a significant role in dictating the shape of these implosions. This is because CBET is known to markedly compensate global,  $\ell = 1$  asymmetries [23, 137], therefore the anisotropy introduced from

magnetisation could affect the action of CBET.

### 3.1.3 Magnetised Laser-Plasma Instabilities

Sec. 3.4, aims to understand how magnetisation of a direct-drive implosion anisotropically changes the hydrodynamics, and how these altered coronal plasma conditions modify the calculated CBET gains, discussed in Sec. 1.5.1. For example, magnetisation restricts thermal conduction and therefore enhances coronal electron temperatures along the initial field axis. Approximately, the fluid CBET gain,  $\gamma_{ij} \propto T_e^{-1}$ , therefore anisotropic changes to  $T_e$  could result in reduced CBET gains around the target and therefore change CBET scattering compared to implosions without an applied field. This modification to CBET via the altered hydrodynamic profiles is called the *indirect* effect of magnetisation on LPIs.

Magnetisation can, however, also *directly* modify scattering from LPIs, in a number of ways. For ICF conditions, when the field strength is sufficiently high, electron cyclotron motion can become comparable to plasma wave frequencies, and therefore alter the dispersion relation of the mediating plasma wave in LPIs. In underdense, ICF relevant plasma ( $n_e \sim 10^{20} \text{ cm}^{-3}$  and  $T \sim 2 \text{ keV}$ ), the IAW, which mediates SBS and CBET, is significantly modified when  $|\mathbf{B}| \sim 100 \text{ T}$  and the Electron Plasma Wave (EPW), which mediates Stimulated Raman Scattering (SRS) and Two Plasmon Decay (TPD), when  $|\mathbf{B}| \sim 1000 \text{ T}$  [138]. Additionally, the (predominantly collisionless) damping of plasma waves can also be modified, because cyclotron motion of particles can affect their trapping in plasma waves [138]. Significant theoretical progress has been made in this field in recent years by Shi et al., who derived analytic formulae for 3 wave coupling in the presence of a magnetic field [139, 140]. This was a challenging problem, due to the lack of simple geometries for the interaction, when a field is applied to a plasma with an arbitrary direction.

The simulation results here neglect this direct affect of magnetisation on CBET, partially because the theory is not yet deemed to be significantly mature, to implement within a reduced, ray-based model. Additionally, coronal magnetic field strengths of  $|\mathbf{B}| \lesssim 50 \text{ T}$  were observed in the underdense coronal plasma so significant modifications to the IAW dispersion relation were not expected. It is noted, however, that altered damping of the waves from magnetisation may affect the results, but the focus of the study was predominantly to explore how magnetisation might indirectly affect CBET.

### 3.1.4 The Exploding-Pusher Configuration

Exploding-pushers are considered to be a highly reproducible platform, robust to instabilities and capable of producing large neutron yields. Although historically it had a slightly different meaning [109], the term ‘exploding-pusher’ is now, typically used for low convergence, thin-shell targets [141]. When irradiated with significant intensity, frequency-tripled laser light<sup>1</sup>, the thin shell rapidly heats and then explosively ablates, driving a strong shock radially inward, ahead of the majority of in-falling ablated material. This shock strongly

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<sup>1</sup>When frequency-tripled light is not used, suprathermal electrons, rather than ablation, is the dominant driver of the strong shock [142].

heats the ions as it propagates through gas-fill to large, fusion relevant temperatures. After rebounding from the axis, the shock recompresses the infalling exploded shell material, resulting in sufficient density for a significant number of fusion reactions.

Directly-driven exploding-pusher targets have the largest direct-drive fusion yields recorded on the NIF, resulting in  $E_{\text{fusion}} \sim 30$  kJ [142]. However, they are not suitable for high gain designs, as the low areal densities of the target are not sufficient to confine  $\alpha$  particles and thus enable burn propagation. A variety of interesting physics may be studied using the platform due to the significant ion temperatures that can be achieved, such as equilibration between electrons and ions [143] and kinetic effects in the ion population [144]. The strong shock is also highly kinetic, and thus accurate comparison to experimentally measurable variables, such as yields and ion temperatures, is expected to be difficult for Rad-Hydro codes which neglect kinetic effects [145]. However, much of the key dynamics and results can be studied more qualitatively.

## 3.2 Cross-Beam Energy Transfer in Unmagnetised Exploding-Pushers

This section presents simulation results, which demonstrate the effect of CBET in exploding-pushers on OMEGA. Both 1-D and 2-D CHIMERA-SOLAS simulations of 40-beam, pole-heavy drive, exploding-pusher experiments are presented, with a focus on how CBET acts to change the implosion. The 1-D results demonstrate that CBET significantly reduced the coupled laser energy to the implosion from 81.1% to 69.7%. Simulations conducted in 2-D, with a full 3-D raytrace and CBET model, clearly demonstrated that the polar drive configuration led to an oblate implosion.

### 3.2.1 Simulation Configuration

The simulations conducted for this chapter aimed to study experimental configurations similar to the results from Bose *et al.*, discussed in Sec. 3.1.2 [8]. Experimental data from these magnetised exploding-pushers, demonstrated a clear amplification of the mode-2 due to magnetisation, which could have affected CBET scattering. The beam configuration, capsule initial conditions and pulse shape used for all simulations in this chapter are shown in Fig. 3.5. The exact pulse shape and target specifications were from a set of follow-up experiments to Bose *et al.* and were provided by C. W. Chang and J. Frenje from Massachusetts Institute of Technology (MIT) [146]. 40 beams from the OMEGA laser, delivered a total of 17.7 kJ laser energy to a  $2.5 \mu\text{m}$  thick, glass ( $\text{SiO}_2$ )<sup>2</sup> capsule, filled with room temperature and pressure  $\text{D}_2$ . All experiments removed the 20 equatorial beams from the drive, because the presence of the small radius field coil in experiments precluded them. Magnetic fields of strength  $B_{z0} = 0, 25$  and  $50$  T were applied along the z-axis of the configuration for different simulations.

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<sup>2</sup>By ion number density, the material comprised 1/3 Si and 2/3 O.

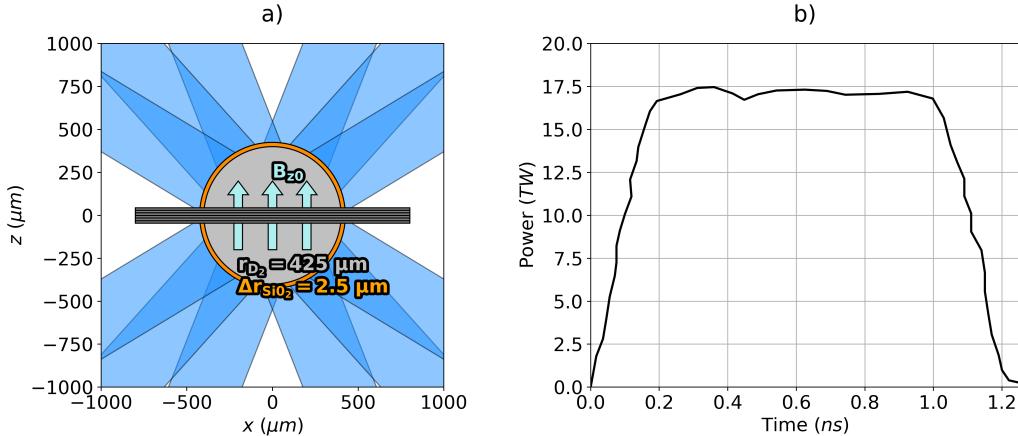


Figure 3.5: The initial conditions used for all simulations presented in this chapter. Panel a) plots the  $D_2$  filled, glass shell capsule and direction of the initial magnetic field. An example field coil (illustrative and not included in simulations) is also shown, the presence of which necessitated the polar laser drive in experiments. Panel b) plots the laser pulse shape used, which had a total of 17.7 kJ laser energy.

An explicit  $P_{1/3}$  radiation transport algorithm was used for all simulations, using tabulated opacities and emissivities from the SPK code [80]. A tabulated Sesame equation of state was employed for each material [79]. The CHIMERA extended-MHD package was used, which included resistive diffusion of the magnetic field, Lorentz force of the magnetic field on the hydrodynamics, Nernst advection of the field down temperature gradients and anisotropic thermal conduction [147]. Thermal conduction was solved using a Super-time-stepping, semi-implicit algorithm [148], with flux-limited, Spitzer conductivities [81]. An electron flux limiter of  $f_{lim,e} = 0.15$  was set for all simulations. Exploding-pushers exhibit high coronal temperatures due to the high- $Z$  ablator and large temperatures for the strong imploding shock, therefore simulation yields and bangtimes are highly sensitive to thermal conduction and the choice of flux limiter. Although not the focus of this work, if attempting to match simulation result to experimental yields, reducing the flux limiter value would strongly inhibit the drive and therefore reduce simulation bangtimes. The flux limiter would therefore be the primary parameter to vary, in order to obtain better match the integrated results to experimental values. In order to improve the speed of the thermal conduction algorithm, which severely limited the stable simulation timesteps on vacuum-plasma interfaces, an artificial material was placed outside the capsule, which had an enforced ionisation state of  $Z = 0$ . This had minimal impact on the implosion results, but significantly sped up simulation run-times. The presence of the material did create large viscous heating of the ions as the coronal plasma expanded into it, which is visible in results throughout the chapter. However, this viscously heated layer was well separated from the region of interest (near critical density) for the majority of the implosion, and therefore did not impact upon the results.

All simulations used a spherical-polar CHIMERA mesh, with a fixed radial resolution of 0.5  $\mu\text{m}$ , over the full  $4\pi$  str. 1-D simulations were run from beginning to end in spherical with  $r \in [0, 1400] \mu\text{m}$ . 2-D calculations assumed azimuthal symmetry of the hydrodynamics and

conducted an initial ‘drive-phase’ in spherical coordinates, with  $r \in [80, 1400] \mu\text{m}$ , where the central cutout region was removed to avoid taking small radiation transport timesteps due to the small cell faces close to the axis. The 2-D drive-phase grid used 120 cells in the polar direction. When the shock reached this cutout region, hydrodynamic variables were trilinearly interpolated onto a 2-D cylindrical  $(r_{\text{cyl}}, z)$  mesh for the ‘stagnation-phase’, with fixed resolution,  $\Delta_{r,\text{cyl}} = \Delta_z = 1 \mu\text{m}$ , and simulation bounds  $r_{\text{cyl}} \in [0, 1200] \mu\text{m}$  and  $z \in [-1200, 1200] \mu\text{m}$ . This grid contained no singularities, unlike a 2-D spherical grid without a cutout, and therefore excessively small radiation transport timesteps were not an issue.

All simulations used a 3-D laser ray-trace with a variety of CBET treatments. CBET was fully included for some simulations and neglected for others. Alternatively, partial-CBET simulations were conducted to explore the effect that CBET *spatial redistribution* of deposited power had on implosions. These simulations were conducted without the CBET model, but were forced to deposit the same magnitude of power that was absorbed from the equivalent CBET simulation. Explicitly, a no-CBET ray-trace was conducted, where the power of all rays was normalised to unity. When this raytrace was complete, it read in the absorbed power from the (previously conducted) CBET simulation at that time, and multiplied the (normalised) deposited power by this value. This created a hydrodynamically similar implosion to the CBET ray-trace, and via comparison of these two simulations, the impact of CBET redistribution-of-deposition upon the hydrodynamics was studied. Temporally-and-spatially integrated results from all simulations are presented in Tab. 3.1, and are referred to when relevant throughout the chapter.

Several metrics have been included in the table, which are explicitly defined here. The metrics are commonly used to compare implosions in ICF, because, apart from the oblateness parameter in the last column, they can all be directly computed from experimental neutron spectra [149]. As previously stated, bangtime is defined as the time of peak neutron production, which for this simulation, assumes only deuterium-deuterium reactions contribute,

$$t_b = \operatorname{argmax}_t \left( \int Y_{DD}(\mathbf{x}, t) \, dV \right), \quad (3.2)$$

where  $Y_{DD}$  is the deuterium-deuterium neutron production rate, which is calculated across the CHIMERA computational grid at all locations and times throughout the simulation, using Bosch-Hale fits to the reactivity [150]. The total neutron yield is the spatially and temporally integrated neutron production,

$$Y_n = \int \int Y_{DD}(\mathbf{x}, t) \, dV \, dt. \quad (3.3)$$

The burn-width,  $\Delta_b$  is the full width half maximum of the spatially integrated neutron production. One interpretation of this diagnostic, for this very two-dimensional configuration, is that a highly oblate implosion will have less temporally and spatially localised convergence. Therefore, neutron production for less symmetric implosions will likely happen over a longer timescale, because thermonuclear conditions are produced at different times in different locations. The burn-averaged ion temperature,  $\langle T_i \rangle$ , is defined as the ion temperature,

Table 3.1: Spatially and temporally integrated results from all Simulations. In the CBET column, ‘~’ indicates partial-CBET, where CBET reduced absorption magnitude, but *not* spatial location, of deposition.

Run	Dim.	CBET	Note	$B_{z0}$ (T)	$t_b$ (ns)	$\langle T_i \rangle$ (keV)	$Y_n$ ( $\times 10^{10}$ )	$\Delta_b$ (ps)	$\frac{R_{\text{equator}}}{R_{\text{pole}}} \Big _{t=t_b}$
1	1-D	Off	-	0	0.69	14.66	11.62	87	$1.00^{+0.00}_{-0.00}$
2	1-D	On	-	0	0.78	12.37	10.57	78	$1.00^{+0.00}_{-0.00}$
3	2-D	Off	-	0	0.71	8.44	6.20	148	$2.96^{+0.20}_{-0.19}$
4	2-D	~	-	0	0.75	7.61	5.23	153	$3.26^{+0.25}_{-0.23}$
5	2-D	On	-	0	0.75	7.77	5.46	148	$3.23^{+0.25}_{-0.23}$
6	2-D	Off	-	25	0.74	7.26	4.73	130	$3.80^{+0.41}_{-0.33}$
7	2-D	~	-	25	0.78	6.58	4.14	125	$4.55^{+0.50}_{-0.43}$
8	2-D	On	-	25	0.78	6.72	4.44	123	$4.32^{+0.47}_{-0.41}$
9	2-D	Off	-	50	0.73	6.82	3.73	134	$4.40^{+0.43}_{-0.38}$
10	2-D	~	-	50	0.78	6.30	3.30	130	$4.92^{+0.56}_{-0.48}$
11	2-D	On	-	50	0.78	6.37	3.52	129	$4.79^{+0.55}_{-0.47}$
12	2-D	Off	No Aniso.	25	0.81	6.40	5.02	118	$3.14^{+0.68}_{-0.50}$
13	2-D	Off	No Lor.	25	0.74	7.29	4.77	131	$3.80^{+0.41}_{-0.33}$
14	2-D	Off	No Nern.	25	0.73	7.41	4.85	130	$3.74^{+0.38}_{-0.32}$
15	2-D	Off	No Resis.	25	0.73	7.07	4.59	132	$3.84^{+0.42}_{-0.33}$

weighted by temporally and spatially resolved neutron production,

$$\langle T_i \rangle = \frac{\int \int T_i(\mathbf{x}, t) Y_{DD}(\mathbf{x}, t) dV dt}{Y_n}. \quad (3.4)$$

It is an integrated metric, which summarises the average temperature of the regions which are key in producing fusion-yield. Finally, the oblateness parameter in the final column of the table,  $(R_{\text{equator}}/R_{\text{pole}})|_{t=t_b}$ , was obtained by fitting an ellipse (with axes orientated along  $\hat{\mathbf{x}}$  and  $\hat{\mathbf{z}}$ ), to the spherical radius of maximum density at bangtime. The fitting procedure also returned asymmetric error bars, which are presented alongside the result.

### 3.2.2 1-D Simulations

In order to demonstrate how CBET typically affects exploding-pushers, two 1-D simulations were conducted, both with and without CBET included. The no-CBET and CBET simulation results are provided in Tab. 3.1, labelled as run 1 and run 2 respectively. Note that because the simulations were 1-D and therefore spherically symmetric, the ratio of equatorial radius to polar radius at bangtime was exactly unity. Both of these simulations used the beam configuration from Fig. 3.5, which unlike the hydrodynamics of the simulation, is not spherically symmetric. As described in Sec. 1.3.2, a sparse 3-D SOLAS mesh was created on which to

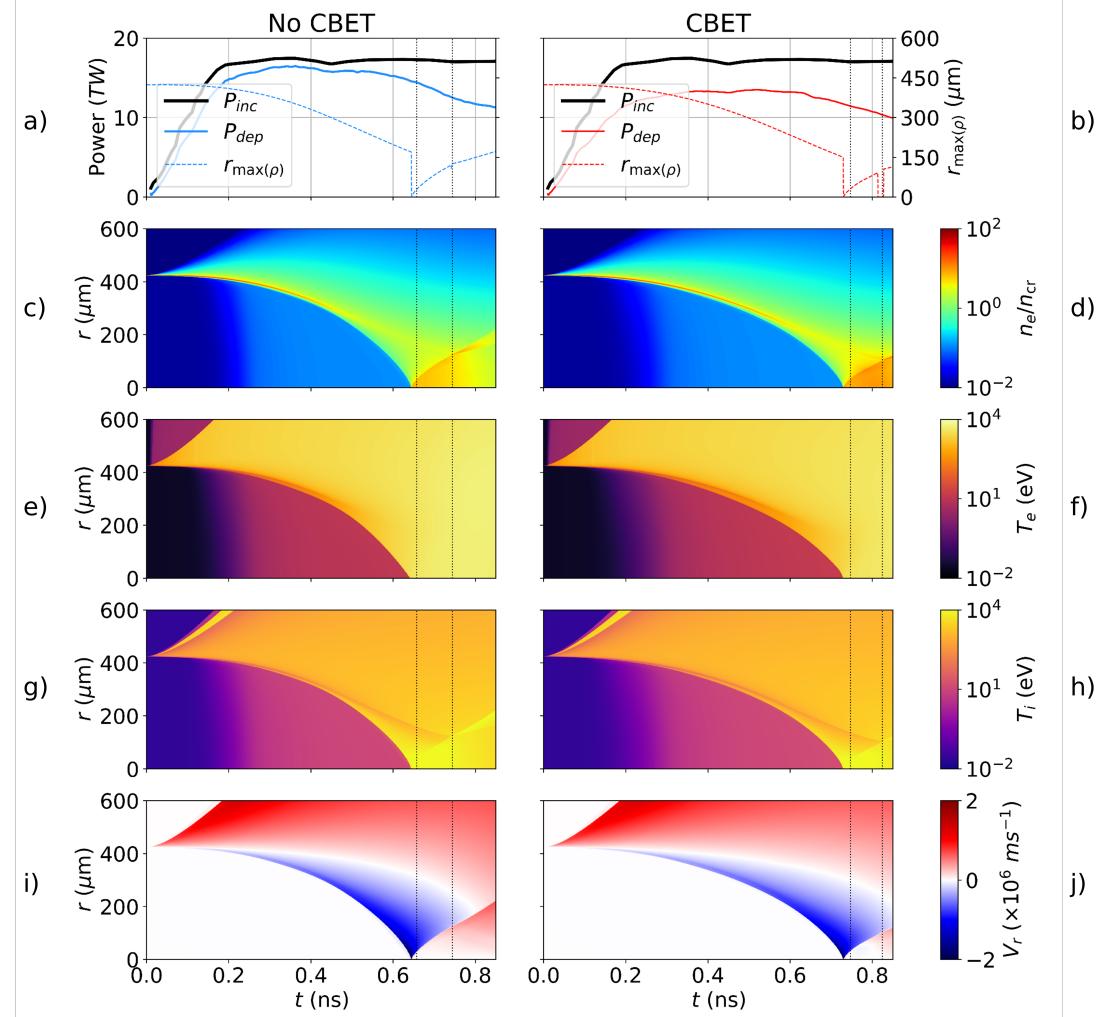


Figure 3.6: 1-D Simulation results both without (left) and with (right) CBET. The top row plots the incident and absorbed energy from the simulation on the left axis and the radius of maximum density on the right. In order, the next rows plot  $n_e$ ,  $T_e$ ,  $T_i$  and  $V_r$ . The full-width half-maximum times of the D<sub>2</sub> yield are plotted as dotted vertical black lines on all panels.

compute the correct beam interaction in 3-D. For the 1-D simulations, 23 cells were used for the underlying spherical polar mesh in the  $\theta$  direction and 54 cells in the  $\phi$  direction, which were then combined to create a semi-structured SOLAS mesh, with approximately equal area across each spherical surface.

Absorbed laser power as a function of time from these two simulations along with streak plots of hydrodynamic variables are plotted in Fig. 3.6. The left column of panels is from the no-CBET simulation and the right are from the CBET simulation. The top row demonstrates that CBET reduced the absorbed power by  $\sim 15\%$  during the implosion phase, up to about  $t \sim 0.6$  ns. Total absorbed laser energy was reduced from 81.1% to 69.7%, from the no-CBET to the CBET simulation. This reduced deposition, led to a slower and weaker shock being driven ahead of the imploding SiO<sub>2</sub> material, which is most clearly visible in the fourth panel, plotting  $T_i$  on a log scale. CBET reduced the speed of the shock, such that it hit the axis

for the with-CBET simulation at  $t \sim 0.73$  ns compared to  $t \sim 0.65$  ns without CBET. Bang-time occurred for both simulations after this ‘shock-flash’, as the rebounding Deuterium fuel compressed against the in-falling shell, creating thermonuclear densities. Greater densities are observed for the with-CBET simulation at bangtime, because the in-falling shell collides the rebounding shock at a smaller radius, *i.e.* the shock timing of the implosion is better. This somewhat compensated for the lower  $T_i$  values due to the weaker shock, but overall, the no-CBET simulation still produced a  $\sim 10\%$  higher neutron yield. Note, that the effect of the fictitious,  $Z = 0$  material, that was placed outside the capsule to speed up the super-time-stepping thermal conduction routine, is visible from the large  $T_i$  on the outer radius boundary of the expanding coronal plasma, which left the plot bounds at  $t \sim 0.2$  ns. Although not presented here, 1-D simulations with vacuum outside the glass ablator initially, showed minimal difference in bangtime hydrodynamic conditions and therefore integrated diagnostics. Thus, the substantially faster configuration with the fictitious ‘vacuum’ material was used for all 2-D simulations.

Compared to the streak plot of a more conventional, hotspot ignition implosion in Fig. ??, clear differences can be seen. Firstly, the simulations presented in Fig. ?? did not include radiation transport, which is the origin of the preheat ahead of the shock<sup>3</sup> in Fig. 3.6. Secondly, the hotspot ignition design maintained a cold dense shell throughout the implosion phase, whereas the initially thin shell of the exploding-pusher simulation was approximately volumetrically heated. Therefore, relatively little mass was left in the glass shell when the rebounding shock collided with it. Finally, the  $T_i$  increase in the exploding-pusher design was predominantly from the spherical convergence of the strong shock, compared to the stagnation heating in the hotspot design, which was localised to inside the decelerating shell when it compressed on axis, at  $t \sim 2.5$  ns in Fig. 2.5. The lack of the cold dense shell is the reason that the exploding-pusher design cannot scale to high yield, because they have insufficient areal density to confine  $\alpha$  particles and thus sustain a burn wave.

### 3.2.3 2-D Simulations

Simulations of this target configuration were also conducted in 2-D to explore how the non-uniformity of the laser-drive affected the implosion, both with and without CBET. 2-D simulations used a 3-D SOLAS mesh to resolve the beam overlap pattern, with 58 cells in the azimuthal direction. Fig. 3.7 provides plots, which illustrate the progression of the simulation during the drive-phase. The left column plots  $T_e$ ,  $T_i$  and  $n_e$  from the 2-D,  $B_{z0} = 0$  T simulation without CBET. Integrated results from this simulation are given in the row labelled run 3 in Tab. 3.1. The right column plots the same for the equivalent simulation, but with the full effect of CBET included, which corresponds to the row labelled run 5 in Tab. 3.1. Note that the viscous ion-heating of the fictitious material placed outside the glass shell initially, is again visible as large ion temperature in the layer immediately outside the coronal plasma expansion.

Particularly at the later times plotted in Fig. 3.7, the decreased, coronal  $T_e$  of the CBET

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<sup>3</sup>*i.e.* The temperature increase, which is visible in  $T_e$  and  $T_i$  plots, which hit the axis at  $t \sim 0.2$  ns.

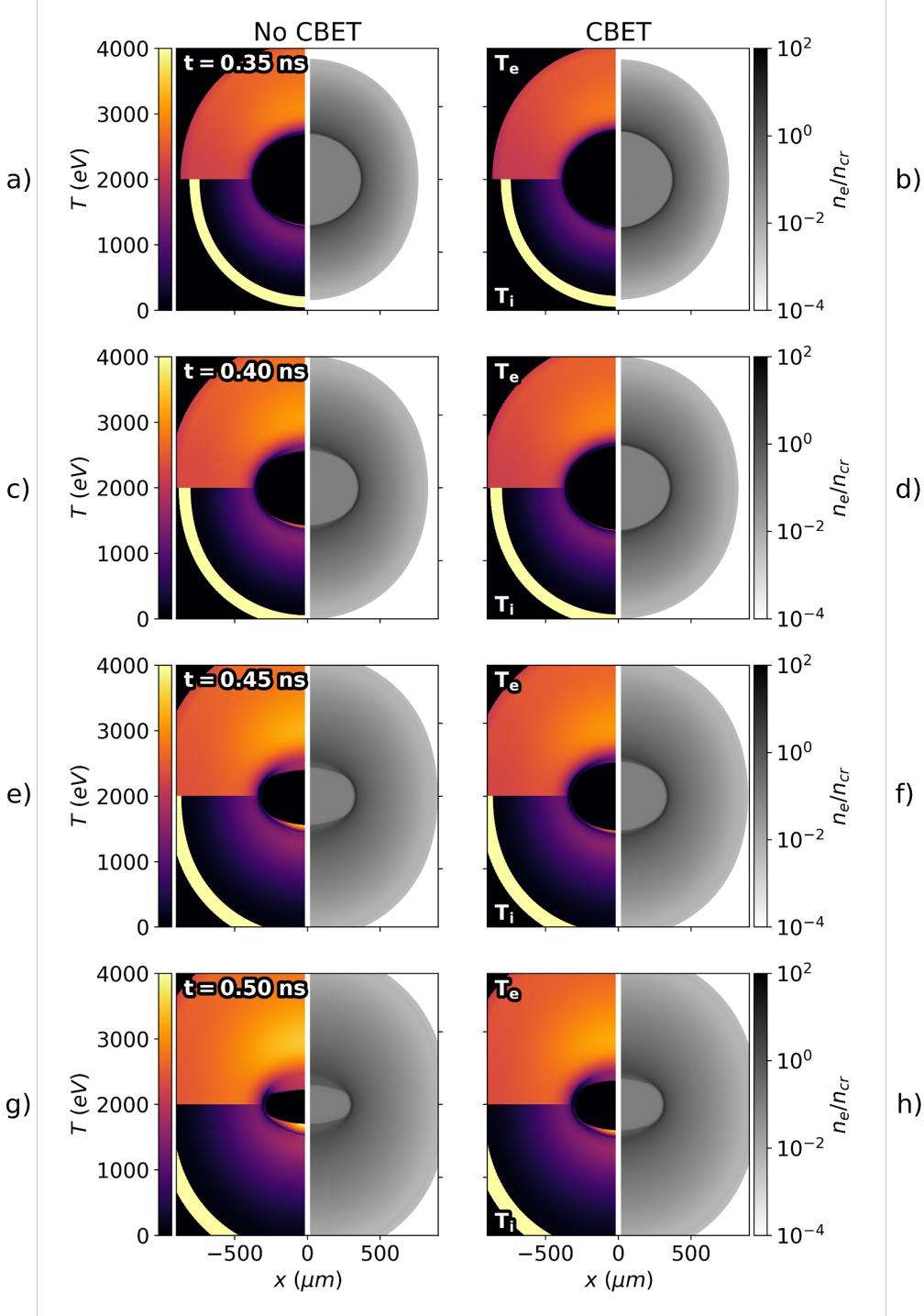


Figure 3.7:  $n_e$  (right-side),  $T_e$  (top-left-side) and  $T_i$  (bottom-left-side) plots from the 2-D, 0 T simulations without (left-column) and with (right-column) CBET at a variety of in-flight times. The decreased deposited energy due to CBET, results in lower coronal electron temperatures and therefore a slower, weaker shock being driven, which is especially evident at later times.

compared to the no-CBET simulation is visible. The lower  $T_e$  was due to reduced absorption, because of CBET scattering, and led to the shock from the no-CBET simulation imploding significantly faster than the CBET shock. Thus, the no-CBET implosion was more

oblate during the drive-phase, because the shock travelled primarily along the  $\hat{\mathbf{z}}$ -axis, due to the beam-geometry. However, when including CBET, the velocity of the shock was reduced more than the velocity of the imploding portion of the glass shell, *i.e.*  $|V_{r,\text{shock}} - V_{r,\text{shell}}|$  was larger for the no-CBET simulation than the CBET simulation. Therefore, the no-CBET shock, after rebounding from the axis, collided with the shell and underwent maximal neutron production at a larger radius than for the CBET result. Thus, the oblateness parameter at bangtime in Tab. 3.1,  $(R_{\text{equator}}/R_{\text{pole}})|_{t=t_b}$ , is significantly larger for the CBET calculation.

Both 2-D simulations exhibited substantially reduced burn-averaged ion temperatures, compared to their 1-D equivalents. This was primarily due to the strong shock travelling mainly along  $\hat{\mathbf{z}}$ , rather than radially inward as for the 1-D calculations. Therefore, the convergence of the shock was reduced, leading to lower temperatures, and therefore neutron yields. This interpretation is also corroborated by the increased  $\Delta_b$  of all 2-D calculations compared to 1-D. The burn-width was larger, because thermonuclear conditions were produced at different times throughout the hot fuel, as the rebounding D<sub>2</sub> compressed against the infalling shell material, compared to the 1-D where it was spherically symmetric.

In summary of these unmagnetised implosions in 1-D and 2-D, it has been shown that CBET substantially reduced the deposited power for these exploding-pusher calculations. 1-D spherical calculations, which averaged the deposited power across all angles, demonstrated that this substantially reduced deposition led to a weaker shock being driven, which reduced thermonuclear yield and delayed the bangtime. When 2-D effects were included, which better reflected the geometry of the laser-drive, this resulted in an oblate implosion, which reduced the convergence of the shock, and therefore the ion temperatures. The reduced drive in the CBET calculations also slowed the shock speed more than the infalling material, so the oblateness at bangtime (when the rebounding shock collided with the infalling material) was greater when including CBET.

### 3.3 Magnetisation in Exploding-Pusher Implosions

This section presents the effect that various extended-MHD terms had on the magnetised, 2-D exploding-pusher simulations. Simulations were conducted with an initial field strength  $B_{z0} = 25\text{T}$  and particular terms turned off, to deduce what the important physical processes were. The origin of the field structure is presented, which demonstrates that in the plasma corona, field lines are mainly radial due to the field being frozen in to highly conductive, radially outflowing plasma. Anisotropic thermal conduction in this highly magnetised coronal plasma acted to keep heat localised to the polar regions, which enhanced the drive on the pole relative to the waist. The results demonstrate that resistive diffusion and the Lorentz force have very little impact on the implosion physics, due to the bulk of the plasma being highly resistive and high- $\beta$  respectively. Nernst-advection of the magnetic field acted to significantly redistribute the field in the low Hall parameter, equatorial region of the capsule, which formed a ‘divot’ in density on the capsule waist. This divot was however well separated from the region where burn was important, and thus had minimal impact on integrated neutron diagnostics.

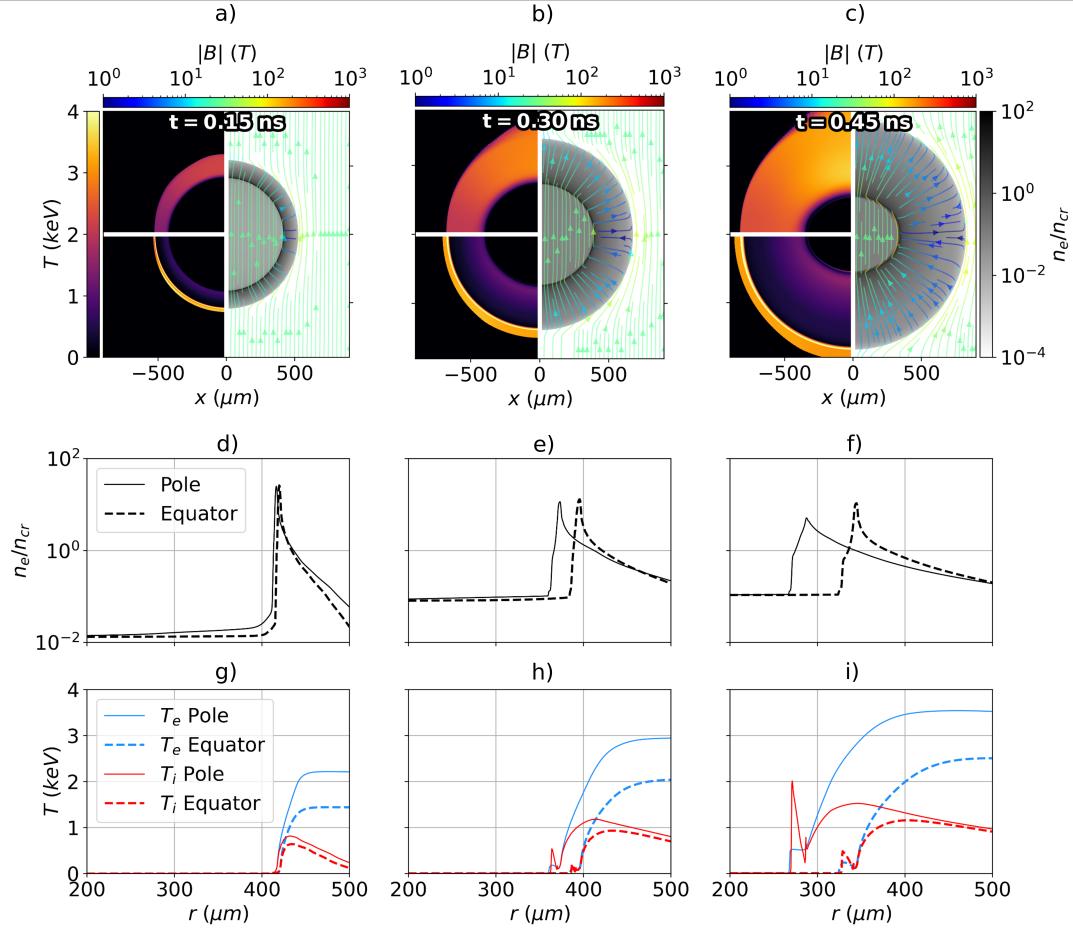


Figure 3.8: The development of the hydrodynamic variables and magnetic field structure from the  $B_{z0} = 25$  T, CBET simulation. Panels a), b) and c) plot  $T_e$  (top-left),  $T_i$  (bottom-left),  $n_e$  (right) and  $\mathbf{B}$  (streamlines) at  $t = 0.15, 0.30$  and  $0.45$  ns respectively. The approximately radially outward flowing, hot (and therefore highly conductive) ablating plasma pulled the magnetic field with it, resulting in radial  $\mathbf{B}$  field lines, which were weaker at the capsule equator. Panels d), e) and f) plot  $n_e$  lineouts along the pole ( $\theta = 0^\circ$ ) and equator ( $\theta = 90^\circ$ ). Panels g), h) and i) plot equivalent  $T_e$  and  $T_i$  lineouts. These all show that the increased polar temperatures, partially due to beam geometry and partially due to magnetisation, led to preferential ablation along the pole.

### 3.3.1 In-Flight Field Structure

Initially, the development of the coronal field structure from the  $B_{z0} = 25$  T, no-CBET simulation (labelled as run 5 in Tab. 3.1) is presented. The top row of Fig. 3.8, plots the drive phase hydrodynamic profiles, overlaid with streamlines of the magnetic field, coloured by its magnitude, at 3 different times. Lineouts of  $n_e$  and the temperatures are plotted in the middle and bottom rows respectively, along both the polar and equatorial directions. The hot coronal plasma was highly conductive, shown explicitly in Sec. 3.3.2, and thus the field remained frozen in to the plasma. As the coronal plasma expanded outward therefore, it dragged the field lines with it, leading to  $\mathbf{B} \sim \pm |\mathbf{B}| \hat{\mathbf{r}}$  in this, laser-heated region. The geomet-

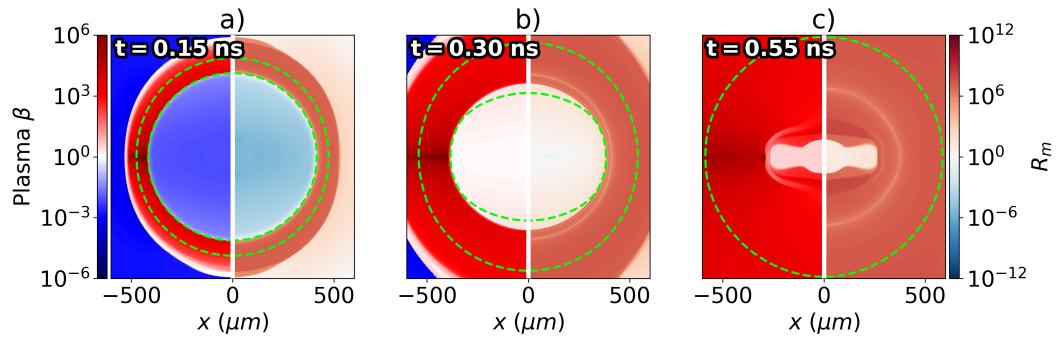


Figure 3.9: Plasma  $\beta$  (left-side) and Magnetic Reynolds Number,  $R_m$ , (right-side) at various in-flight times, throughout the  $B_{z0} = 25 \text{ T}$ , no-CBET simulation. Contours of the  $n_e = n_{cr}/10$  are plotted on all panels as dashed green lines to indicate the bounding region, which contained a significant amount of plasma. Broadly, the  $\beta$  and  $R_m$  values are  $\gg 1$  in all regions with an appreciable amount of material, which demonstrates that the Lorentz force and resistive diffusion should have minimal effect on the implosion dynamics.

ric stretching of the field lines at the target poles was less significant than at the equator, and therefore the coronal field strengths were smallest at the target equator and highest on the poles. As the target began to implode, the field compressed on the interior edge of the dense shell, resulting in non-radial field lines and an increase in field strength. This effect is most clearly visible at  $t = 0.45 \text{ ns}$ , in Fig. 3.8.c in the vicinity of the dense shell material region.

The lineouts clearly demonstrate that the preferential heating of the target on the pole, led to faster ablation of the shell along this direction. This led to a much stronger shock along the pole, which is seen most clearly by the discrepancy in ion temperature between the pole and equator at  $t = 0.45 \text{ ns}$  in Fig. 3.8.i. Increased polar electron temperature is partially due to the pole heavy drive from the beam geometry, and also due to anisotropic thermal conduction. The field structure plotted in Fig. 3.8, inhibited equilibration of temperature via thermal conduction in the polar direction, which increased the temperature asymmetry compared to the unmagnetised simulation. Anisotropic thermal conduction and its impact in this configuration is discussed in more detail in Sec. 3.3.3.

### 3.3.2 Resistive Diffusion and the Lorentz Force

As was stated in the above Sec. 3.3.1, the coronal plasma was highly conductive for all simulations, due to the combination of high temperature and relatively low density. The right side of each panel Fig. 3.9 plots the Magnetic Reynolds Number,  $R_m$ , at three times during the implosion of the  $B_{z0} = 25 \text{ T}$ , no-CBET simulation, labelled as run 6 in Tab. 3.1. As was stated in Sec. ??,  $R_m$  is the ratio of magnetic field advection to diffusion. Conservatively, a  $1 \mu\text{m}$  length scale was used to calculate this value, which was approximately the smallest length scale observed throughout the implosion across all times. Thus, for the majority of the simulation,  $R_m$  is likely underestimated, because  $R_m \propto L$ . The green dashed lines are contours of  $n_e = n_{cr}/10$ , which are included in the plots to illustrate the regions of the simulation that contained the majority of the plasma material. Although Fig. 3.9.a shows

that early in the simulation, the gas fill of the capsule had a small value of  $R_m$ , and therefore resistive diffusion dominated over frozen in flow, the dynamics of the implosion had not reached this region yet, *i.e.* it is not bounded by the  $n_e$  contours. The radiative preheat also raised the temperature, and therefore  $R_m$  value in this region, before the time plotted in Fig. 3.9.b. Therefore, the majority of the plasma material had a high value of  $R_m$  throughout the implosion, mainly due to the high temperatures, demonstrating that frozen-in-flow dominated over resistive diffusion. This is corroborated by comparing run 6 and run 15 in Tab. 3.1, which did not include the effects of resistive diffusion on the field transport. There is minimal difference across all metrics between these two rows, which shows that the global dynamics of both simulations were very similar.

The left side of the panels in Fig. 3.9 plots the plasma  $\beta$  at the same times for the same simulation. As was stated in Sec. ??, this is the ratio of thermal plasma pressure to magnetic pressure and thus describes whether thermal forces or the Lorentz force, is likely to predominantly influence the plasma dynamics. Just as was seen for  $R_m$ , the value of this parameter was generally high in all regions with appreciable plasma density throughout the implosion, so the Lorentz force was assumed to minimally influence the plasma dynamics. An additional simulation was performed, in which the effect Lorentz force was not included on the plasma, labelled by run 13 in Tab. 3.1. Again, there is minimal difference in integrated metrics between this and run 6 (the comparable run which included the Lorentz force), which verifies this analysis.

### 3.3.3 Anisotropic Thermal Conduction

In order to understand the effect of anisotropic thermal conduction, an additional simulation was performed, which had the same setup as the  $B_{z0} = 25$  T, no-CBET simulation, but thermal conduction was isotropically suppressed by the local magnetic field strength, regardless of its orientation. The anisotropic and isotropic simulations are labelled by run 6 and 12 respectively in Tab. 3.1. Explicitly, the parallel conductivity was forced to take the value of the perpendicular conductivity,  $\kappa_{\parallel} = \kappa_{\perp}$ . Comparison of this isotropically suppressed conduction simulation, with the  $B_{z0} = 25$  T, no-CBET simulation, elucidated the role of anisotropic conductivities relative to the orientation of the field. Fig. 3.10.a plots the ion (left-side) and electron (right-side) hall parameters from the  $B_{z0} = 25$  T, no-CBET, anisotropic thermal conduction simulation at  $t = 0.5$  ns. It can be seen that the polar region of the corona was significantly magnetised, with  $1 < \omega_e \tau_e < 10$ , therefore thermal conduction in this region was significantly suppressed perpendicular to the field in the anisotropic simulation, which was approximately the polar direction, as is seen in Fig. 3.8.c. Ion Hall parameters peaked at values of  $\omega_i \tau_i \sim 10^{-2}$  in-flight, thus ion transport was not expected to be significantly affected by the field.

The electron temperature profiles at the same time are plotted in Fig. 3.10.b for the isotropically magnetised conduction simulation (left-side) and anisotropic simulation (right-side). The  $T_e$  profile of the anisotropic case was significantly more oblate than the isotropic simulation and the polar regions of the isotropic region were also significantly hotter. This was

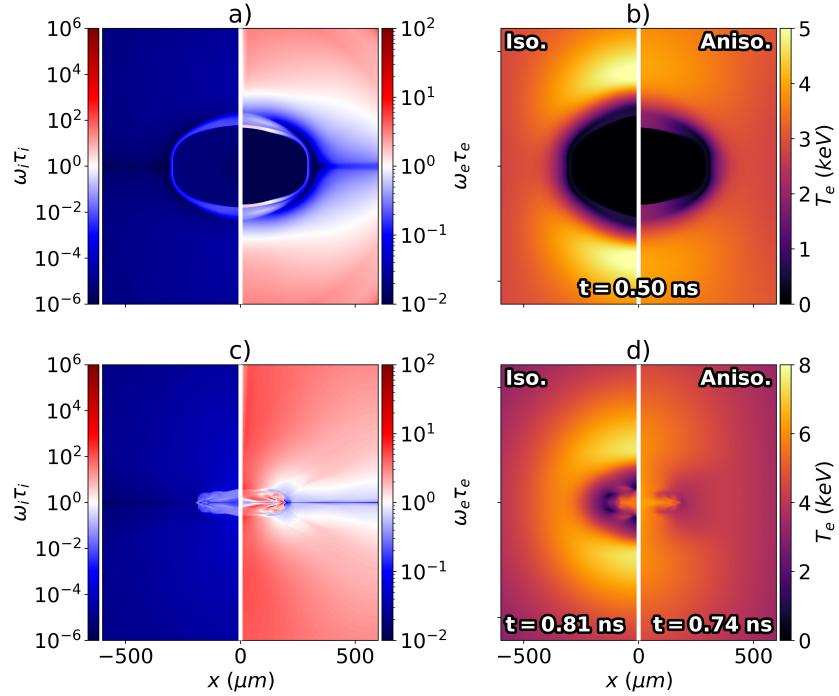


Figure 3.10: In-flight a) and bangtime c) Hall parameters, from the  $B_{z0} = 25$  T, no-CBET simulation. Panel b) plots the  $T_e$  from the isotropically magnetised simulation (left-side) and anisotropic conduction simulation (right-side) in-flight. Panel d) plots the same, but at bangtime. The electron Hall parameter was  $> 1$  at the poles due to high magnetic fields and temperatures, which led to significantly restricted thermal conduction from magnetised transport. Isotropically magnetised conduction, therefore resulted in a markedly different bangtime morphology, as is shown in panels b) and d). Ion hall parameters peaked at bangtime, when values reached about  $\omega_i \tau_i \sim 0.1$ .

due to decreased transport of deposited laser energy in the isotropic to the ablation region, because heat could not freely stream radially inward along the field lines when conductivity was isotropically suppressed by the field. Isotropic magnetisation of electron transport thus led to reduced ablation at the poles and a larger critical radius along  $\pm\hat{\mathbf{z}}$ , which gave the implosion and shock a rounder shape.

Bangtimes temperature profiles, plotted in Fig. 3.10.d, corroborate this analysis, which demonstrate that the isotropic simulation had a significantly rounder shape. The oblateness parameter in Tab. 3.1, obtained by fitting an ellipse to the radius of maximum density at bangtime, also demonstrates a less oblate profile. By isotropically magnetising the conduction,  $(R_{\text{equator}}/R_{\text{pole}})|_{t=t_b}$  was reduced from  $3.80 \rightarrow 3.14$ . The reduced ablation coupled less energy to the implosion, evident in the reduced bangtime from  $t_b = 0.74 \rightarrow 0.81$  ns, going from anisotropic to isotropic thermal conduction. Yield still increased for the isotropic simulation from  $4.72 \rightarrow 5.02 \times 10^{10}$ , which shows that integrated yield metrics are a fine balance of many factors, including coupled energy and implosion shape.

The bangtime Hall parameters are plotted in Fig. 3.10.c, which show highly magnetised electrons and moderately magnetised ions, with a maximum  $\omega_i \tau_i \sim 10^{-1}$ . Note that this is

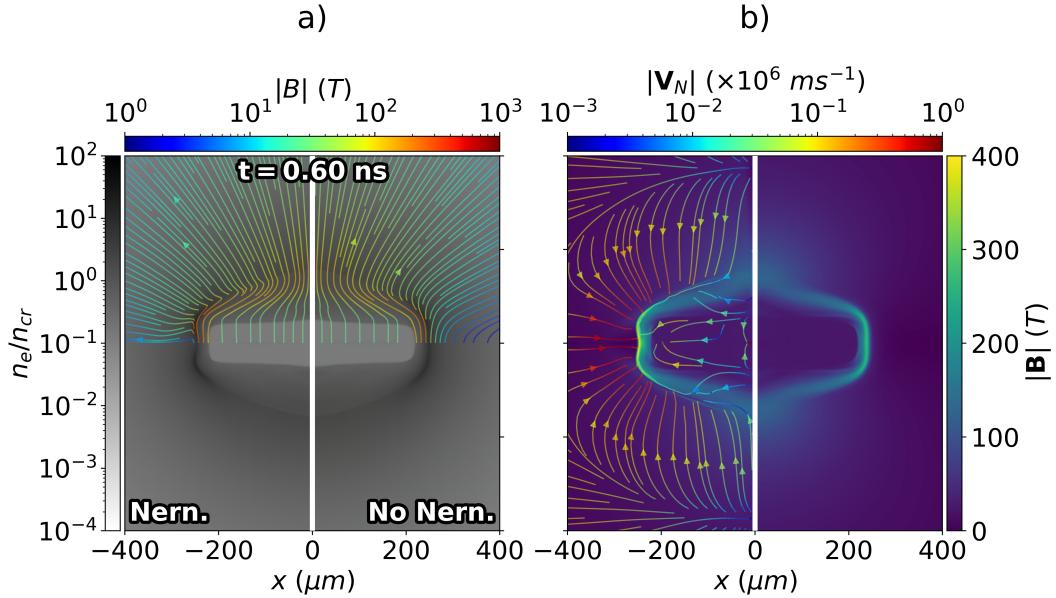


Figure 3.11: Panel a) plots in-flight electron density profiles from the  $B_{z0} = 25$  T, no-CBET simulations with (left-side) and without (right-side) Nernst advection of the magnetic field. The magnetic field is also plotted as streamlines in the top-half of each plot, coloured by magnitude. Panel b) plots magnetic field magnitude from the Nernst (left-side) and no-Nernst (right-side) simulation along with Nernst advection velocity streamlines from the Nernst-on simulation. Advection of the field is important in the low Hall parameter equatorial region, pulling  $\mathbf{B}$  down  $\nabla T_e$ , into the dense wall. Altered field at the equator impacts on the magnetised thermal conduction, which ultimately imprints on the density, as is seen in panel a).

also for an initial field strength  $B_{z0} = 25$  T simulation. For the  $B_{z0} = 50$  T, with-CBET simulation, which included CBET fully and is therefore the most realistic simulation to compare with experiment at this field strength, the maximum ion Hall parameter was also  $\omega_i \tau_i \sim 10^{-1}$ . This is lower than the reported value of  $\omega_i \tau_i \sim 1$  in the original paper [8]. However, improved sphericity of the converging shock, which could change slightly for example by changing  $f_{\text{lim},e}$ , would markedly boost this value. The hall parameter,  $\omega \tau \propto T^{3/2} |\mathbf{B}|$ , so a less oblate shock would increase the value this by both better flux compression of the field, and higher stagnation temperatures, due to greater convergence of the shock.

### 3.3.4 The Nernst Effect

The final extended-MHD term that was independently examined was the Nernst effect. In CHIMERA, Nernst is implemented as an advection of  $\mathbf{B}$  down  $\nabla T_e$  [147]. It is a collisional term, and therefore dominates in plasma with low values of  $\omega_e \tau_e$ . The plots of Hall parameter in Fig. 3.10 demonstrate that the coronal plasma has low Hall parameters at the equator, which suggests that the Nernst effect would be most important at the equator of the capsule and have minimal effect on the field profiles near the poles. An additional simulation, with  $B_{z0} = 25$  T and no-CBET, was conducted without Nernst advection of magnetic field to understand its impact upon the field and plasma dynamics. The simulation without the Nernst

effect is labelled in Tab. 3.1 as run 14.

Fig. 3.11.b plots the magnetic field strength from equivalent simulations with-Nernst (left-side) and without-Nernst (right-side) at  $t = 0.60$  ns. The simulation including Nernst (run 6 in Tab. 3.1), also plots the Nernst advection velocity as streamlines, coloured by the speed. The largest advection speeds are  $\sim 10^6 \text{ ms}^{-1}$ , which is comparable to the coronal fluid ablation speed, demonstrating that Nernst-advection is non-negligible in dictating the magnetic field structure. As expected, the equator of the target had the largest Nernst speeds, due to the low Hall parameter, and it acted to push field from the coronal plasma onto the interior edge of the shell. This created a localised field ‘pile-up’ in this region for the Nernst simulation, compared to the simulation on the right.

Fig. 3.11.a plots the electron density at the same time for the Nernst (left-side) and no-Nernst (right-side) simulations alongside magnetic field streamlines from the same simulations on the top-half of the plot. The streamlines show that the Nernst effect caused the coronal field lines at the equator to be radial, by pushing  $B_z$  down the mostly radial temperature gradient, towards the shell. It is visible that the altered magnetic field structure impacted on the plasma density indirectly, creating an equatorial bump in the shell density, which is not present in the right-side plot without Nernst included. This bump only appeared in simulations with both Nernst and anisotropic conduction, which shows that this altered field profile changed the anisotropic thermal conduction at the equator, which then impacted upon the density. The effective re-orientation of  $\mathbf{B}$  at the equator, meant that thermal conduction which transported laser energy from the deposition surface to the ablation surface, experienced the higher, parallel thermal conductivity  $\kappa_{\parallel}$ , rather than  $\kappa_{\perp}$ . This locally increased the drive along  $z \sim 0$ , leading to the bump at the equator.

Comparison of run 6 (Nernst) and run 14 (no-Nernst) in Tab. 3.1, shows minimal difference in the integrated metrics, despite the altered dynamics. This is mostly because the changes were localised to the equator, which was well separated from the polar region, which experienced the strongest shock convergence and therefore greatest temperatures and neutron production. It is expected however, that experiments with symmetric laser illumination, and therefore a less oblate implosion, may be more influenced by the Nernst effect, because the equator of the shell would partake more in neutron production. Additionally, it is worth noting that this bump is likely to only be observed in magnetised direct-drive implosion and not indirect-drive. This is because the main action of Nernst in these simulations was to change the field and therefore anisotropic transport in the conduction zone, which is the layer in direct-drive implosions that transports energy from the laser absorption region to the ablation surface. However, for indirect-drive, x-rays penetrate effectively all the way through to the ablation surface, so there is no conduction zone. Thus, the altered conduction zone field profile would have a much less significant impact on the density for indirect-drive.

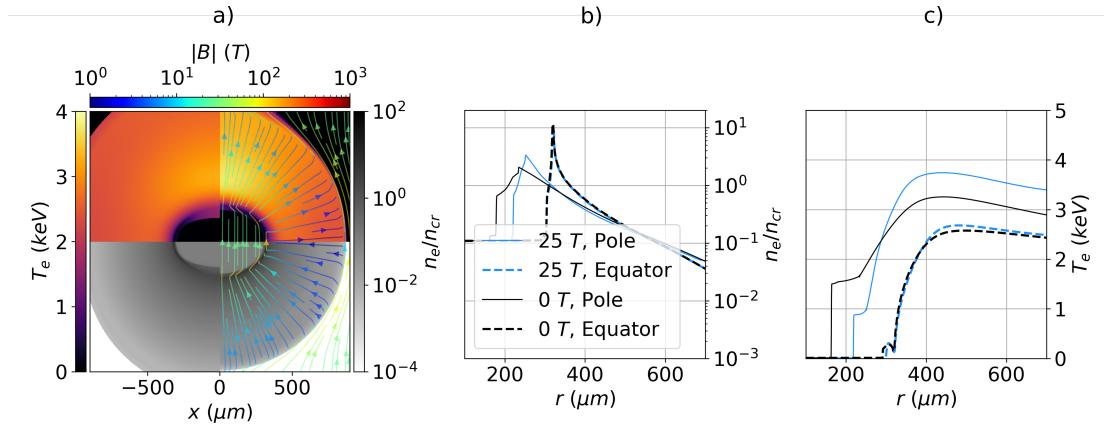


Figure 3.12: Comparison of  $n_e$  and  $T_e$  profiles from the  $B_{z0} = 0$  (panel a) left-side and 25 T (panel a) right-side, with-CBET simulations, both at  $t = 0.5$  ns. Panel a) also plots streamlines of  $\mathbf{B}$  for the  $B_{z0} = 25$  T simulation, coloured by the field magnitude. Panels b) and c) plot  $n_e$  and  $T_e$  lineouts respectively, along both the pole and equator. It is evident from these lineouts that magnetisation anisotropically affects hydrodynamic variables, which are used to calculate the CBET gain. Therefore, it is anticipated that magnetisation could anisotropically affect the CBET scattering volume.

### 3.4 The Effect of Magnetisation on Cross-Beam Energy Transfer and Stagnation

This section presents results on how the magnetisation of the corona affects both CBET scattering and the stagnation shape of the implosion. Fig. 3.12.a plots  $T_e$  and  $n_e$  from the  $B_{z0} = 0$  (left-side) and 25 T (right-side), with-CBET simulations at  $t = 0.5$  ns. Fig. 3.12.b and Fig. 3.12.c plot lineouts of  $n_e$  and  $T_e$  respectively, along both the poles and equator for both simulations. As was shown in Sec. 3.3, anisotropic thermal conduction is the dominant effect of magnetisation in these implosions. The equatorial lineouts are similar, due to low Hall parameters in this region and thus minimal impact of magnetisation upon the transport. However, the polar lineouts are significantly different in the magnetised and unmagnetised cases. Magnetisation has amplified the  $\ell = 2$  asymmetry of the coronal plasma  $n_e$  and  $T_e$  profiles, which are used to calculate the CBET gain. It was therefore hypothesised, that this amplified asymmetry in coronal profiles due to magnetisation, would leave a signature upon the CBET and therefore implosion dynamics, because CBET is known to interact strongly with low mode, coronal asymmetries [137].

In this chapter ‘partial-CBET’ simulations shall be presented, for which the coupled power as a function of time was kept the same as the equivalent with-CBET simulation, *i.e.* so CBET only acted to reduce the magnitude of the deposited power, rather than redistribute it around the target. It was hoped that, by comparison with the full-CBET simulations, this would demonstrate that CBET changed the shape of the implosion differently, for different levels of seed magnetic field. The results presented in this section demonstrate that the  $\ell = 2$  of the long-wavelength, density perturbation was slightly reduced by CBET, consistent

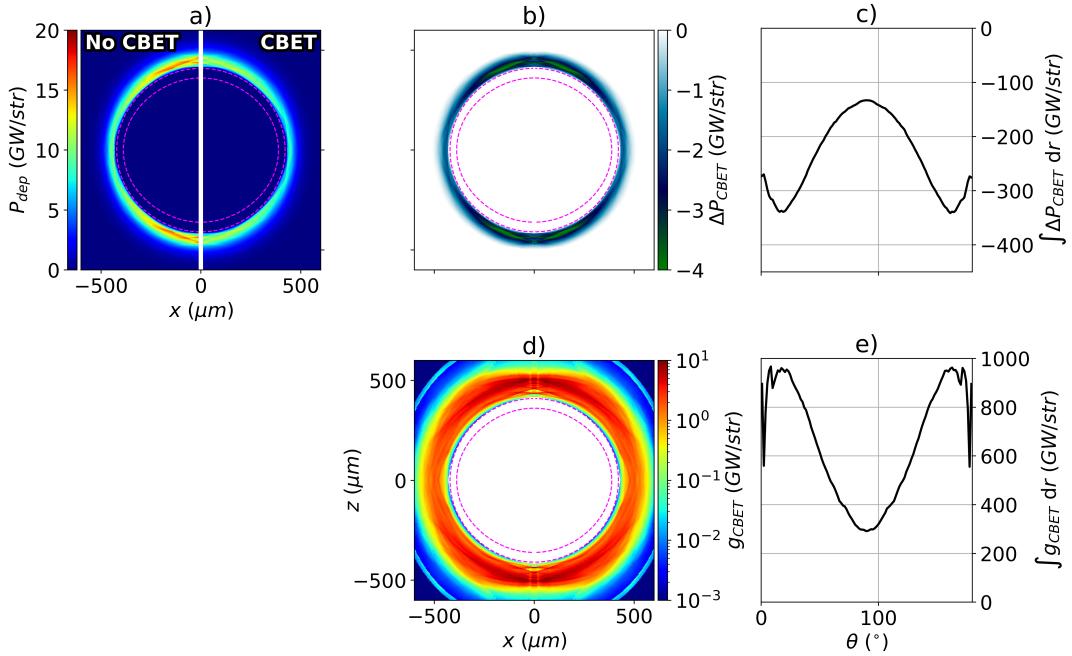


Figure 3.13: Various CBET diagnostics used in the analysis presented in this section. All plots are from the  $B_{z0} = 0$  T, with-CBET simulation at  $t = 0.30$  ns. Panel a) plots the instantaneous deposition with (right-side) and without (left-side) the effect of CBET. The ‘CBET-deficit’,  $\Delta P_{CBET}$ , which is the no-CBET deposition, subtracted from the with-CBET deposition, is plotted in panel b). Panel c) plots the radially integrated  $\Delta P_{CBET}$ , as a function of polar angle. The ‘CBET-scattering’, is plotted in panel d), and the radial integral is plotted in panel e). At this time, it is evident from panel e) that more CBET occurred at the capsule poles, resulting in a CBET-reduction of deposition near the poles, seen in panel c). The critical surface is shown in all 2-D panels by a dashed magenta line.

with existing literature on how CBET mitigates  $\ell = 1$  asymmetries [23]. Results also showed that the increasingly anisotropic coronal plasma profiles for increasing seed magnetic field strengths, did lead to changes in the CBET scattering. However, this was too small an effect to lead to experimentally observable changes in density and temperature. It is hypothesised that the polar beam geometry, the shock driven characteristics of the exploding-pusher implosions and the high  $Z$  shell (which results in increased coronal temperatures, and therefore reduced CBET [87]), minimised the impact on CBET scattering. Suggestions for an alternative implosion are provided, which would increase the likelihood of an experimentally observable signature.

### 3.4.1 Analysis and Key Definitions

Firstly, key quantities used for the analysis of this section shall be introduced. The key aim of this section was to discover if CBET acted differently in the magnetised coronal hydrodynamic profiles, compared to the unmagnetised profiles. Particular focus was dedicated to discovering if this affected the implosion shape, because this shape change would not

be captured by a laser-MHD model that did not have a full-CBET capability, but rather reduced the incident laser energy to compensate for CBET. Two quantities were used to understand how CBET redistributed the deposited power. The first was the ‘CBET-deficit’, which is defined as the difference in deposited power computed by a CBET and no-CBET raytrace through the *same* hydrodynamic profiles,

$$\Delta P_{\text{dep}}(\mathbf{x}, t) = P_{\text{dep}}^{\text{CBET}}(\mathbf{x}, t) - P_{\text{dep}}^{\text{no-CBET}}(\mathbf{x}, t), \quad (3.5)$$

where  $P_{\text{dep}}^{\text{CBET}}$  is the deposition from the calculation including CBET and  $P_{\text{dep}}^{\text{no-CBET}}(\mathbf{x}, t)$  is without CBET. As was described in Sec. ??, a CBET calculation in SOLAS always begins with a raytrace which does not include any CBET effects (the field-reconstruction raytrace), so  $P_{\text{dep}}^{\text{no-CBET}}(\mathbf{x}, t)$  is the deposited power from this raytrace.  $P_{\text{dep}}^{\text{CBET}}(\mathbf{x}, t)$  is the value after the pump depletion and energy conservation iterations, *i.e.* after CBET has been fully accounted for.

Fig. 3.13.a plots the deposited power without (left-side) and with-CBET (right-side), from the  $B_{z0} = 0$  T, with-CBET simulation at  $t = 0.30$  ns. The corresponding CBET deficit, which is the difference between these plots, is shown in Fig. 3.13.b. Note that the value is almost all negative, because for direct-drive configurations, CBET mainly acts to reduce the intensity of light near the critical surface where Inv-Brem predominantly occurs, and therefore it reduces the deposited power. There are small ( $\sim 1\%$ ) increases in deposition at radii outside the peak scattering surface, near Mach-1, but this increase is not be visible to the naked eye when the maximum colourbar scale is increased above zero. It can be seen from the CBET deficit, that at this time, CBET decreased the absorption at the poles more than at the equator of the capsule. This can be understood by considering the geometry of the pole heavy drive. The beam geometry meant that the intensity was greater in the polar coronal plasma, where many beams overlapped each other. Therefore, inbound laser sheets encountered stronger CBET resonances, because more reflected sheets were present in this region, thus their energy was depleted more than at the waist. The radial integral of the CBET deficit is plotted in Fig. 3.13.c. This conveniently illustrates the previous point, that the reduction in absorption due to CBET (at this time, of this simulation), was greater near the poles than the equator. It is also worth noting, that if CBET acted symmetrically to simply reduce the absorption rather than redistribute it, then Fig. 3.13.c would be uniform in  $\theta$ . This is the case for the ‘partial-CBET’ simulations, labelled by ‘~’ under the CBET column in Tab. 3.1, where the magnitude of the power was reduced to account for CBET, but the redistribution of power was not accounted for.

The second metric was the CBET scattering,  $G_{\text{CBET}}$ , defined in Sec. 1.6.1, but restated here for convenience,

$$g_{\text{CBET}}(\mathbf{x}, t) = \sum_i^{\text{rays}} |\Delta P_{i,\text{CBET}}(\mathbf{x}, t)|, \quad (3.6)$$

where the time dependence has now been made explicit and  $\Delta P_{i,\text{CBET}}$  is the power change of a ray due to CBET. This quantity shows where the majority of CBET power change occurred, across the simulation domain, at a given time. Fig. 3.13.d plots  $g_{\text{CBET}}$  on a log scale, from the

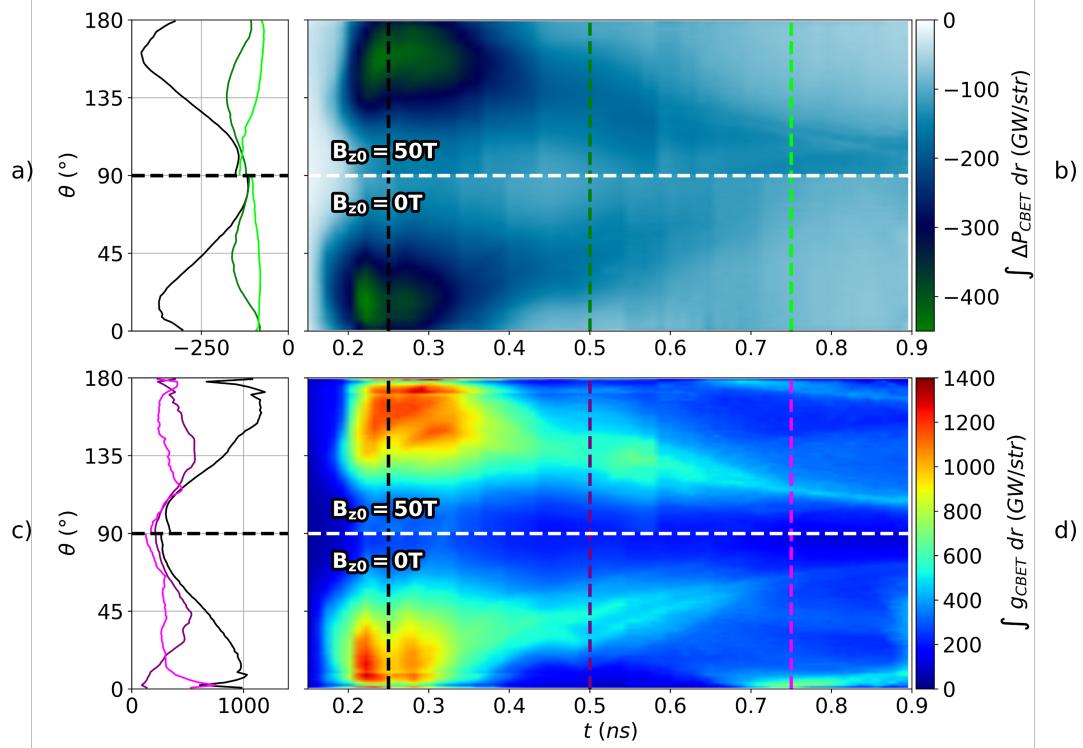


Figure 3.14: The radially integrated CBET-deficit and CBET-scattering, plotted as a function of angle and time for the  $B_{z0} = 0$  and  $50\text{ T}$ , with-CBET simulations. Panel b) plots the CBET-deficit from the  $50\text{ T}$  (top-half) and  $0\text{ T}$  (bottom-half) simulations. Lineouts in  $\theta$  at  $t = 0.25$  (black),  $0.50$  (dark-green) and  $0.75$  ns (light-green) are plotted in panel a). The same results, but for CBET-scattering are plotted in panels d) and c). It is evident from these plots that for both simulations, as time passed and the poles of the capsule fell in faster than the equator, the region where CBET mostly occurred, shifted in angle around the capsule. Differences are visible in all plots between the  $B_{z0} = 0$  and  $50\text{ T}$  simulations, indicating that magnetisation affected CBET indirectly, via the altered hydrodynamics.

same simulation and at the same time as Fig. 3.13.a. The radial integral of this plot is shown in Fig. 3.13.e, as a function of angle. It is evident by comparison of Fig. 3.13.c and Fig. 3.13.e, that the angles of maximal CBET scattering align with the most significant deficits, and vice-versa. It should be noted that there is some noise visible in both CBET scattering plots near the poles. This was presumed to be due to small cells and therefore poorer ray-per-cell statistics in this region. This noise was, however, localised to a few polar-cells, and therefore not thought to have a significant impact on the overall dynamics.

### 3.4.2 Spatial Change of CBET and Deposition from Magnetisation

This section shall explore differences in the CBET-deficit and CBET-scattering, between magnetised and unmagnetised implosions. As was illustrated in Fig. 3.4, magnetisation significantly amplified coronal plasma asymmetries in direct-drive implosions. Thus, an effect on the CBET gain calculations was expected. How changes to CBET affect the hydrodynamics and stagnation shape of the target, is discussed in the subsequent Sec. 3.4.3.

Fig. 3.14.b plots the CBET deficit as a function of time ( $x$  axis) and polar angle ( $y$  axis) for the with-CBET,  $B_{z0} = 50$  T (top-half) and  $B_{z0} = 0$  T (bottom-half) simulations respectively. Note that the colour scale has a maximum of zero, and is most saturated at negative values, because the radially integrated CBET-deficit was entirely negative. Lineouts in  $\theta$  at three times,  $t = 0.25, 0.50$  and  $0.75$  ns (times indicated by dashed lines on Fig. 3.14.b), are plotted in Fig. 3.14.a, where the  $x$  axis of this plot share the colour scale from Fig. 3.14.b. Broadly, both the top half and the bottom half of Fig. 3.14.b appear to be qualitatively similar, indicating that magnetisation of coronal profiles did not play a dominant role in altering CBET. The simulations featured a large CBET-deficit of deposition near the poles between  $t \sim 0.2 \rightarrow 0.3$  ns. As the implosion progressed, the asymmetry of the deficit reduced, which is visible as the reducing saturation of the plot, with increasing time. This was, however, also partially due to the lengthening of the plasma corona, which increased symmetry primarily through increasing deposition at larger radii. Large radius deposition is less efficient at coupling to ablation, compared to if the energy were deposited close to critical, due to thermal conduction having to transport it further. An improved metric, which appropriately weights deposition by distance from the critical surface, may further elucidate the role of CBET. Both plots show that the peak of the CBET-deficit shifted to the equator as the simulations progressed.

The CBET-scattering streak plot and lineouts in Fig. 3.14.d and 3.14.c respectively show similar behaviour. Before  $t \sim 0.2$  ns, the level of CBET-scattering was low, primarily due to the low incident power, and therefore laser field strengths. Maximal scattering occurred near the poles, at early times in both the magnetised and unmagnetised simulations, which then moved more to the equator later on. The movement of the scattering and deficit to the equator was due to the asymmetric convergence of the target. As can be seen from the time resolved  $n_e$  profiles from the with-CBET simulation in Fig. 3.7, the coronal plasma was relatively round up until  $t \sim 0.35$  ns, but then began to implode more quickly along the  $\pm\hat{z}$  axis than the equator. When this occurred, the trajectory of light changed asymmetrically, leading to a change of polar-angle where CBET predominates.

Firstly, the expected change to the bangtime profiles from the similar, qualitative behaviour shall be described. Bangtimes for these implosions were at  $t \sim 0.75$  ns, so the late time behaviour shortly before and after this, was unimportant in dictating the bangtime shapes. The most significant  $\Delta P_{\text{CBET}}$  values occurred between  $t \sim 0.2 \rightarrow 0.3$  ns at the poles, for both the magnetised and unmagnetised simulation. This CBET-induced power redistribution, would have reduced the strength of the shock launch along the poles, compared to the equator. Therefore, simulations with an equal magnitude of power deposition, but no spatial relocation of deposition (*i.e.* the partial-CBET simulations), were expected to exhibit a larger  $\ell = 2$  asymmetry, because CBET appeared to reduce the asymmetry of the drive.

Now, the change to the CBET deficit between the magnetised and unmagnetised simulations shall be explicitly compared. If the altered coronal plasma profiles due to magnetisation had played a dominant role in changing CBET, then the top and bottom plots would look markedly different. The similar qualitative behaviour seems to suggest therefore that this is not the dominant role in dictating implosion symmetry. However, minor differences are ob-

servable, for example the magnetised simulation exhibited more early-time CBET-scattering and a slightly larger CBET-deficit at the poles, which also lasted for a slightly longer period of time. This suggests that the shock launch in the  $B_{z0} = 50$  T case would be slightly weaker than the  $B_{z0} = 0$  T case. A significant reason for the relatively small changes, is that the pole-heavy drive obscures these, more subtle, changes to implosion dynamics. Explicitly, an unmagnetised, symmetric drive beam-configuration should not display an  $\ell = 2$  asymmetry in either of the diagnostics, and should thus result in a mostly symmetric bangtime profile. Changes to the shape, would therefore be much less obscured compared to these simulation which feature a global asymmetry, imposed by the 40-beam drive.

### 3.4.3 Stagnation Profiles

The bangtime  $n_e$ ,  $T_e$  and  $\mathbf{B}$  profiles from the 2-D simulations with all magnetisations (columns) and all CBET treatments (rows) are plotted in Fig. 3.15. The top row of figures correspond to full CBET treatment and the bottom row, no effect of CBET is included. As a reminder, to conduct the partial-CBET simulations in the middle row (labelled ‘~CBET’ for shorthand), the deposited power as a function of time was loaded in from the (already completed) full-CBET simulation with the same initial  $B_{z0}$  value. No-CBET ray-traces were then conducted with incident power normalised to unity, and the deposition was multiplied by the absorbed power from the full-CBET simulation at the same time. Thus, full-CBET and ~CBET simulations with equivalent  $B_{z0}$  have identical power deposition as a function of time. The only difference between them is that the full-CBET simulations, on the top row, include redistribution of deposition around the corona.

Rows labelled run 2 → 11 in Tab. 3.1 provide the integrated metrics for each of these simulations. The first thing to note about these results, is that all implosions were oblate, *i.e.* the radius of peak density was significantly larger in the equatorial direction than polar. Magnetisation also evidently flattened the implosion, regardless of CBET treatment. No-CBET simulations (bottom row) were also significantly less oblate than the top two rows, which both included reduced absorption due to CBET. To remind the reader, it was due to the larger  $|V_{r,\text{shock}} - V_{r,\text{shell}}|$  for the no-CBET simulation, compared to when CBET was included, as was described in 3.2.3. Thus, the collision between the rebounding shock and in-falling fuel material happened at larger radii, resulting in a rounder bangtime shape.

The stagnation field structure for the magnetised simulations exhibited field lines at  $z \sim 0 \mu\text{m}$  which point along  $\pm\hat{\mathbf{x}}$ . This was due to the oblate structure of the initial strong shock, which moved mainly along  $\pm\hat{\mathbf{z}}$  for a wide array of  $x$  values. When the shocks from the two poles met at  $z = 0 \mu\text{m}$ , a thin layer was superheated to extreme temperatures of several 10keV. This thin layer of highly conductive plasma, reoriented the un-shocked  $+\hat{\mathbf{z}}$  field to point in the  $\pm\hat{\mathbf{x}}$  directions. The large compressed field strengths also increased the peak temperatures significantly, by insulating thermal losses. Note however, that this increase was highly anisotropic and was localised to lower density regions of the hotspot, therefore did not directly translate to increased burn-averaged temperatures in Tab. 3.1. For simulations which exhibited larger hotspot temperatures, the higher pressures appeared to de-

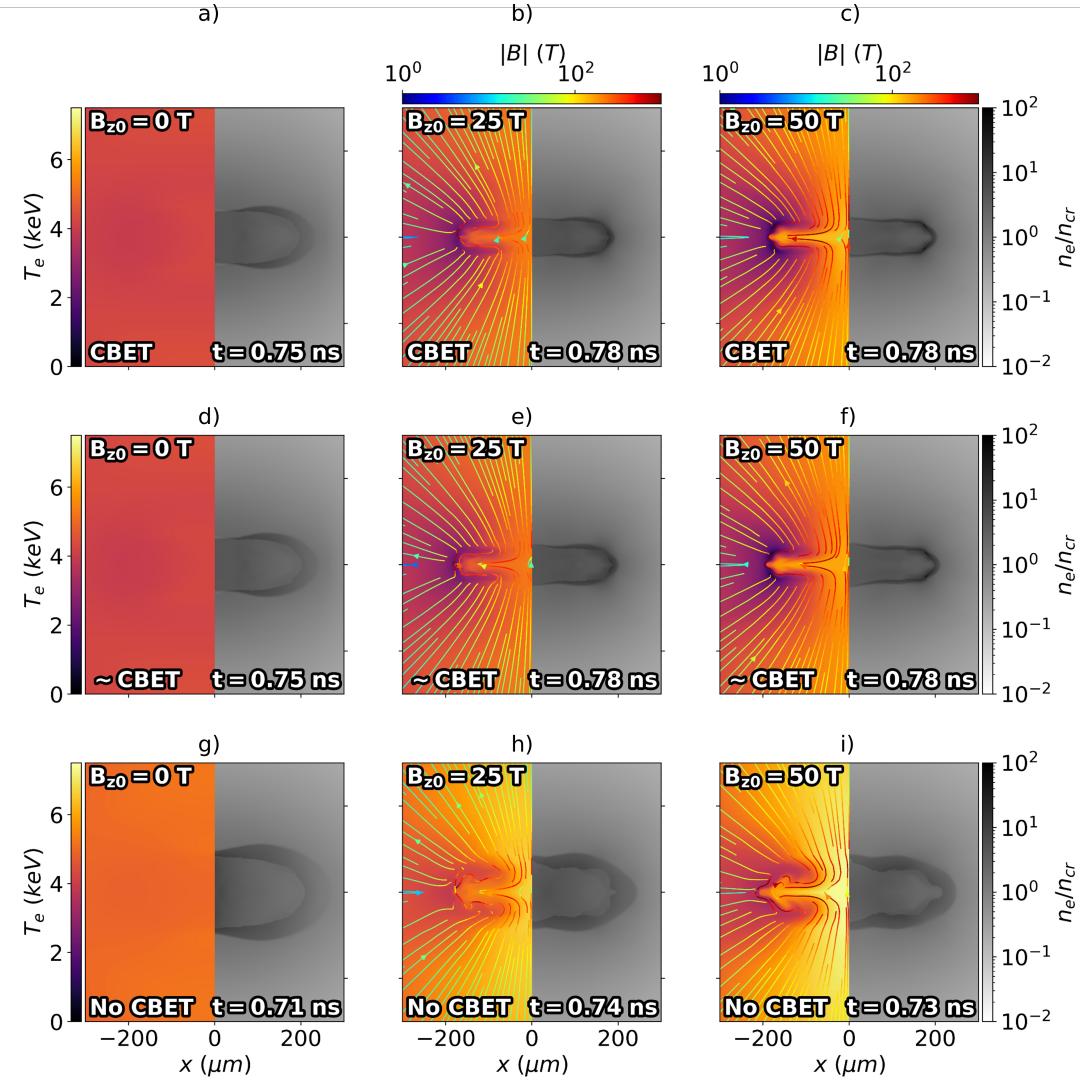


Figure 3.15:  $n_e$ ,  $T_e$  and  $\mathbf{B}$  bangtime profiles for different initial magnetisation (columns) and CBET treatments (rows). Panels a), b) and c) are from the full CBET simulations, which show that magnetisation increased the oblateness of the stagnation profile. Panels d), e) and f) are from the simulations where only the CBET effect on the magnitude, but not spatial location, of deposition was included. These simulations have identical coupled energy to the top row, and therefore have the same bangtimes. Panels g), h) and i) are from the no-CBET simulations, which had earlier bangtimes and increased temperatures, due to the greater coupled energy.

velop some form of magneto-RTI at the equator of the capsule, most evidently visible in  $T_e$  at  $x, z \sim [-150, 0] \mu\text{m}$ , although the exact dynamics of this have not been studied in detail.

If CBET was anisotropically affected by magnetisation, this would result in spatial differences in deposition location, between the full CBET (top row) and partial-CBET simulations (middle row), and therefore the bangtime profiles would be different. Additionally, higher initial magnetisation increased the coronal plasma anisotropy. Therefore, the difference between the top two rows would have changed as  $B_{z0}$  increased, *i.e.* from column to column.

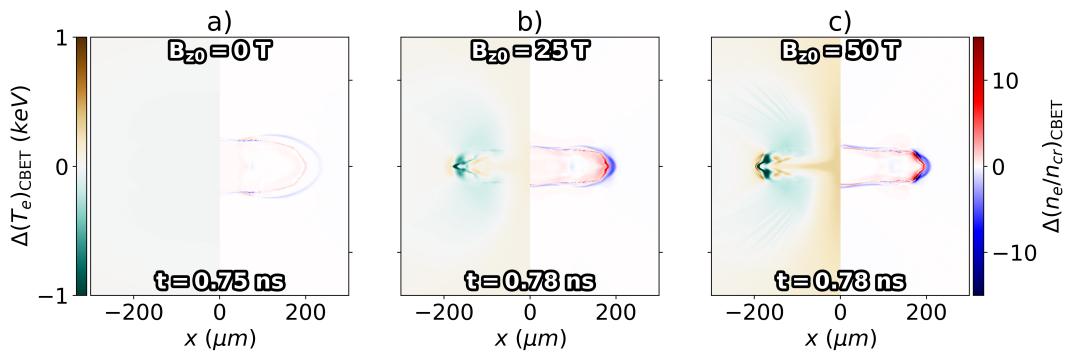


Figure 3.16: Difference in  $n_e$  and  $T_e$  bangtime profiles, between the full-CBET and CBET-magnitude a)  $B_{z0} = 0$ , b)  $B_{z0} = 25$  and c)  $B_{z0} = 50$  T simulations. The difference in variable  $v$ , is defined as  $\Delta v = v_{\text{full-CB}} - v_{\text{partial-CB}}$ , so higher colour scale values represent regions with increased  $v$  for the full-CBET calculation. These results primarily show that CBET slightly reduced the bangtime equatorial radius, and thus reduced the oblateness, compared to simulations where the spatial redistribution of power due to CBET was neglected.

Evidently, the top rows appear very similar to each other, which illustrates that anisotropic changes to CBET from magnetised hydrodynamic profiles was a small effect compared to both the polar nature of the drive and the anisotropic drive due to magnetised transport. Small changes are present however, which is far more clearly displayed in Fig. 3.16. This figure plots the difference in  $n_e$  and  $T_e$  between the full-CBET (top row of 3.15) and partial-CBET (middle row of 3.15) bangtime profiles. As can be seen, differences became more significant at larger magnetisations, demonstrating that anisotropic changes of CBET scattering do impact upon the hydrodynamics. The main difference can be seen at the equator of the  $n_e$  profile, of both magnetised simulations, where the full-CBET maximum density occurs at a lower  $x$  value. Thus, CBET redistribution of deposition in magnetised coronal profiles marginally reduced the oblateness of the bangtime density. This aligns with the expected behaviour, described at the end of Sec. 3.4.2. Restating here, the early, localised CBET-deficit reduced the strength of the polar shock launch compared to the equatorial shock. This improved the symmetry of the shock and therefore the stagnation state. It was also observed that this effect seemed to become marginally more significant in magnetised coronal profiles. By comparing Fig. 3.16.a with Fig. 3.16.b and 3.16.c, the differences are slightly more significant, which supports this analysis.

The oblateness parameters from Tab. 3.1 are plotted for these implosions as a function of initial field strength for different CBET treatments. These values were obtained by fitting an ellipse to the contour of maximum density at bangtime. The fitting procedure returned asymmetric error bars, which are also plotted. As was observed by examining the profiles in Fig. 3.15 directly, the no-CBET results are all significantly rounder, compared to simulations for which the effect of CBET on deposition magnitude is included. This was due to the faster shock launched with greater early time deposition. The partial-CBET simulations, which do not include redistribution of deposition are all marginally more oblate than the

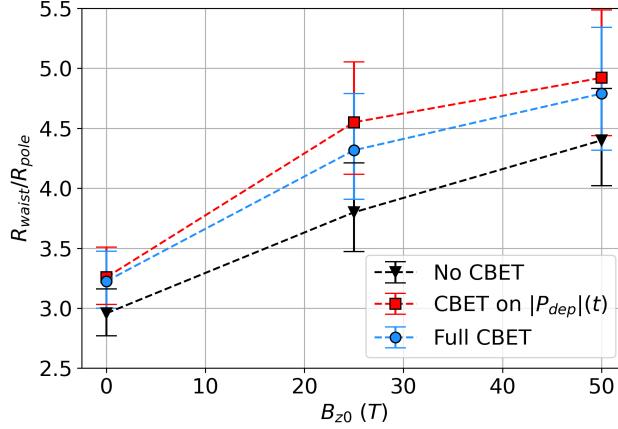


Figure 3.17: The oblateness of bangtime density profiles for different magnetisation and CBET effects. All values and errors were obtained by fitting an ellipse (with axes orientated along  $\hat{\mathbf{x}}$  and  $\hat{\mathbf{z}}$ ), to the radius of maximum density. No-CBET simulations were consistently more round, because the initial shock was stronger, and therefore travelled ahead of the pusher material more quickly than the CBET equivalent. Thus, after rebounding off the axis, it met the infalling mass and produced thermonuclear conditions at a larger radius. As was seen in Fig. 3.16, throughout the entire implosion, when the effect of CBET on spatial location of deposition was included, it acted to slightly move energy from the pole to the waist and thus marginally reduced the oblateness of the stagnation state.

full-CBET simulations, with the caveat this cannot be stated conclusively, because of the error bar magnitudes. Simply stated, CBET acts to slightly reduce the oblateness of the stagnation state by effectively redistributing a small quantity of power from the poles of the capsule to the equator. There is no evidence from this plot, however, that increasing magnetisation, anisotropically alters CBET sufficiently to leave an experimental signature. If, for example, the difference between the full- and partial-CBET curves had diverged (or converged) with increasing magnetisation, this would have suggested that the increasingly anisotropic hydrodynamic profiles from magnetisation, significantly impacted upon the CBET scattering.

In retrospect, however, it was noted that the configuration used for these simulations likely minimised any observable effect in several ways. Firstly, as was discussed at the end of Sec. 3.4.2, the polar beam geometry obscured the less significant CBET anisotropy from magnetisation. Secondly, the shock driven nature of the exploding-pusher configuration, could have minimised experimentally observable signatures. The shape of the stagnation state was predominantly dictated by the shock launch geometry from the shell explosion, which was mostly sensitive to the early laser-energy deposition. A more conventional, hotspot ignition target would have maintained a dense shell throughout the implosion, which would have developed density modulations due to CBET-induced deposition asymmetries, as was seen in the previous chapter, for example in Fig. 2.12. Experimentally, a mode  $\ell = 2$  is long wavelength enough to be inferred from neutron diagnostics, and would thus leave an experimentally observable signature in the neutron spectrum [151, 152]. Finally, higher  $Z$  ablator materials such as  $\text{SiO}_2$ , exhibit higher coronal plasma temperatures than CH shells, due to

increased Inv-Brem efficiency. CBET becomes less significant for direct-drive, as the coronal temperatures increase [87], therefore an experimental setup with a CH outer material, may have lead to more CBET scattering overall, and therefore a more easily observable change to hydrodynamic profiles from magnetisation.

## 3.5 Conclusions

### 3.5.1 Summary of Work

This chapter has described simulations of directly-driven, magnetised exploding-pusher implosions. For the first time, an in-line model for CBET was included in a magnetised direct-drive implosion simulation, in order to study how asymmetry in coronal plasma profiles from magnetisation, may affect the action of CBET. 1-D simulations of the exploding-pusher configuration without magnetisation demonstrated that CBET was dynamically significant to the implosion, reducing the coupled laser energy to the target by  $10 \rightarrow 15\%$ . The main effect of this reduced absorption was to slow the speed of the strong shock, which led to bangtime occurring at greater target convergence after the rebounding shock compressed against the in-falling material. When extended to 2-D, the simulations demonstrated that the polar nature of the drive was highly significant to the symmetry of the implosion, leading to a bangtime density profile with an aspect ratio  $R_{\text{eq.}}/R_{\text{pole}} \sim 3$ . The slowing of the shock due to the CBET reduction of absorption, led to bangtime occurring at a larger radius, and thus a less oblate profile was obtained, because the shock travelled mainly along the  $\pm\hat{\mathbf{z}}$ -axis. Simulations which included the reduced absorption due to CBET, but not spatial redistribution of energy, showed that CBET power redistribution marginally reduced the oblateness of the implosion, consistent with how it acts to reduce  $\ell = 1$  asymmetries [137].

Simulations were also conducted which included an initial magnetic field along the  $+\hat{\mathbf{z}}$ -axis. The key physical processes which were dynamically significant to the implosion dynamics were identified by conducting additional simulations which turned off the effect of each process in turn. Anisotropic thermal conduction was shown to be highly significant. It acted to insulate the deposited energy at the poles of the equator, preventing thermal equilibration in the polar direction and thus led to an increased asymmetry of the drive. The generally high temperatures of the target led to large  $R_m$  and Plasma  $\beta$  values, therefore the Lorentz force and resistive diffusion had minimal impact on the implosion. Nernst advection of the magnetic field effectively reoriented the coronal field structure at the low-Hall parameter, equator of the target. This field reorientation altered the anisotropic thermal conduction, leading to a ‘bump’ in the density at the target waist. Although this equatorial-bump had minimal impact on integrated neutron diagnostics, because it was well removed from the polar-regions, which produced the most thermonuclear reactions, it may significantly impact yields for symmetrically driven targets.

The impact of the magnetised coronal plasma profiles on CBET scattering of the laser energy was also studied. By examining how CBET reduced absorption instantaneously through both the magnetised and unmagnetised coronas, the dominant effect of CBET was observed

to be an early reduction of deposition at the capsule poles, regardless of seed field strength. This led to a marginally more symmetric drive, reducing the oblateness of the bangtime profiles. Increasing the level of the initial seed magnetic field appeared to slightly exaggerate this early, polar reduction of deposition, although the effect was minimal and did not significantly impact upon the hydrodynamics. Therefore, increased anisotropy of the coronal field profiles was not observed to significantly alter CBET scattering for these pole-heavy drive, exploding-pusher targets.

### 3.5.2 Future Work

Future alterations to the simulation configuration shall now be suggested, which may lead to a more obvious impact of magnetisation on CBET. The simulations that were conducted for this work all used a 40-beam OMEGA configuration, where the 20 equatorial beams were removed from the drive, due to the presence of a magnetic field coil in experiments. Simulations demonstrated that this was the dominant effect, which dictated the symmetry of the bangtime state. Magnetisation of thermal conduction also played a significant role, increasing the oblateness of the drive. The polar drive largely obscured the much more subtle changes to CBET from the altered hydrodynamic profiles, shown by comparing the top and bottom halves of panels in Fig. 3.14. A configuration with a more symmetric, 60-beam drive, would not have an underlying  $\ell = 2$  anisotropy without magnetisation, which may make the changes to CBET in magnetised profiles more pronounced.

Secondly, alterations to the target could be made which may lead to both increased diagnostic signatures of an anisotropic drive and a larger impact of CBET overall. Firstly, simulations of a more conventional, central hotspot ignition target could be conducted, which maintain a dense shell throughout the implosion. Asymmetry of the deposition seeds density perturbations in the shell throughout the duration of the implosion. This is in contrast to the exploding-pusher targets presented here, where the bangtime symmetry is predominantly dictated by the shape of the early, shock launch. Additionally, using an ablator material with a lower ionisation state would lower the coronal plasma temperatures significantly and thus enhance the levels of CBET [87]. More subtle changes to CBET scattering from magnetisation may thus have a larger impact on the implosion, due to more CBET occurring overall.

## **Appendices**

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