Ishar Singh Saini

Aerospace Engineer

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About Me

Presented hypersonic CFD analysis at SAROD 2024 and validated TBCC propulsion performance using OEM data. I'm an Aerospace graduate specializing in high-speed aerodynamics, CFD, and propulsion systems. Known for rapid problem-solving and hands-on troubleshooting in high-pressure environments; applied across CFD simulations, CAD modelling, and UAV prototyping. My work bridges theory, simulation, and real-world application, and I'm now ready to solve cutting-edge challenges in next-gen propulsion and aerothermal systems.

Experiences / Internships

TATA Advanced Systems Limited

Bengaluru, India

Concept Design Intern

June 2024 - Aug 2024

- Worked in the Concept Design Department under Dr. A. K. Ghosh analyzing the conceptual designs capabilities with the help of CFD.
- Simulated hybrid UAV performance under crosswind and hover conditions; observed that rear rotors produced up to 50% greater thrust and power due to equal RPM distribution. Analyzing crosswinds and test data conditions required iterative RPM input adjustments to balance mechanical power.
- o Operated TASL's Linux-based High-Performance Computer (HPC) for computational solving using HiFUN and Python.
- Investigated Pitot tube accuracy when in proximity to the fuselage, identifying stagnation-induced errors when static inlets are close to the fuselage **up to 2%**, with greater inaccuracies at high speeds conditions like headwinds or dives.
- Conducted aerodynamic analysis of radome modules of aircraft. Results highlight that the new leeward-curved design led to 7% drag reduction; suggested curvature if applied to the older yet more streamlined design could yield even lower drag.
- Generated high-quality meshes and mesh-compatible geometries from complex NMGs (Numerical Master Geometries) of the UAVs and aircrafts, ensuring solver compatibility and convergence.

Semester Long Intern

FOSSEE, IIT Bombay

(Remote) Mumbai, India March 2024 - June 2024

- Conducted CFD simulations using OpenFOAM, an open-source CFD software for the research migration project titled, 'High-speed compressible flow over blunt and re-entry bodies'.
- Validated turbulent compressible flow behavior over blunt re-entry capsules against experimental data; identified limitations up to 13% discrepancy of the sonicFoam solver in hypersonic regimes.
- Performed mesh independence studies to determine optimal grid structure, ensuring solver stability and convergence across Mach regimes.
- Analyzed shock structure and aerodynamic effects as a function of Mach number, identifying key trends relevant to re-entry vehicle design.
- Secured 1st Rank in the national screening for project selection.
- o This project was presented in SAROD 2024 symposium.

Project & Research Intern

Mumbai, India

IIT Bombay, Department of Aerospace Engineering

June 2023 - Aug 2023

- Worked on a Precision Landing Drone project under Prof. Arnab Maity, in the Department of Aerospace Engineering, focusing on FEM simulations, structural integrity analysis of composite materials, assembly and testing.
- Conducted detailed static and localized load analyses using Ansys, **validating** carbon fiber composite performance against aluminum through stress-strain and deformation comparisons resulting in almost **50% weight reduction** of the drone.
- o Designed PLA components in SolidWorks; oversaw machining, assembly, and integration of electronics and sensor calibration.
- Implemented a dual-phase precision landing system using RTK GPS and ArUco markers, achieving autonomous landings with up to **2 cm accuracy**.

Skills

Technical Skills: Prototype Designing, Research and Analytical skills, Unmanned Aerial Systems Piloting (Fixed-Wing and Rotary-Wing), Wind Tunnel Testing

Software Tools: Ansys, HiFUN, FreeFlyer, OpenFOAM, SimScale, OpenVSP, XFLR5, GasTurb, MatLab, Scilab, Visual Studio Code, CAD, SolidWorks, CATIA, GridPro, ICEM

Programming Tools: C, C++, Python, Linux, GitHub, LaTex

Interpersonal Skills: Leadership qualities, Innovation, Creativity, Strong verbal and written, Communication skills, Critical-Thinking, Problem-Solving, Teamwork

Languages: English, German (A2), Hindi, Punjabi, Marathi

Education

Amity University Mumbai

2021 - 2025

B. Tech in Aerospace Engineering

- o CGPA: 8.33/10.0
- $\circ~50\%$ merit scholarship awarded at admission
- o GATE (AE) 2025: 21.0/100.0

Awards

July 17, 2025 Emerging Talent in Open Source FOSSEE, IIT Bombay IIT Bombay National FOSSEE Awards 2025 (Edition II) 1st Consolation Prize Feb 20, 2025 FOSSEE, IIT Bombay Scilab Case Study Hackathon 1st Place Sept 26, 2024 Project Presentation - Innovation Day Amity University Mumbai 3rd Place Feb 24, 2023 Poster Presentation - Innovation Day Amity University Mumbai

Certifications

Aircraft Design
NPTEL
NPTEL
Supersonic and Hypersonic Flows
NPTEL — Elite + Silver Medal
Supersonic Flows
April 2023

Publications

- 1. Saini, I. S., Chourushi, T., et al. (2025). Assessment of Existing OpenFOAM Solvers for Incompressible Flow Over a 2D Prism. CFD Letters 17(10):184-197, 2025. https://doi.org/10.37934/cfdl.17.10.184197

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- 2. Saini, I. S., Khadka, B., & Chourushi, T. (2024). Numerical Simulation of High–Speed Compressible Flow over Re-entry Vehicles using the OpenFOAM. 10th Symposium on Applied Aerodynamics and Design of Aerospace Vehicles & SPICES Workshop (SAROD 2024). (Paper ID 57)

Projects

Conceptual & Numerical study of Turbine-based Combined Cycle Engines and their mode transition Final Year Project (Dissertation) Amity University Mumbai

- Investigated a TBCC propulsion configuration integrating an afterburning low-bypass turbofan (**GE F404-IN20**) with a ramjet, enabling **three operational modes** from subsonic to high supersonic (Mach 0.8–4) and identified the critical mode transition zone between **Mach 2.8–3.2** marked by a thrust imbalance.
- Employed a coupled simulation approach using GasTurb for combustion analysis and custom MATLAB thermodynamic codes to model thrust sharing and air-splitting strategy; results showed increased specific thrust and improved TSFC performance post-transition, peaking near Mach 2.75.

Conceptual Design of Supersonic Unmanned Combat Aerial Vehicle

National Aerospace Conceptual Design Competition (NACDeC-VIII)

Aeronautical Society of India

- \circ Led and mentored a team of sophomores in the conceptual design of a supersonic UCAV, targeting a 5,000 kg payload delivery over a 1,000 km combat radius within a one-hour mission cycle.
- Employed Aerodynamic modelling and Wing configuration, Performance and stability analysis, Propulsion sizing, Stealth design, Structural considerations, Sub-systems design, etc. to finalize a 18800 kg take-off gross weight and mission fuel fraction of 0.346, with performance analysis conducted in IRA conditions.
- \circ Selected a delta wing configuration and a turbofan with afterburner for Mach 2 cruise, enabling high-speed penetration and precision strike while ensuring $\mathbf{sub-2,000}$ m take-off/landing performance.

Conceptual Design for a TLAM (Tomahawk Land Attack Missile) $Personal\ Project$

- Designed a long-range, subsonic cruise missile for precision land strikes, launched from ships and submarines, offering strategic versatility and deep strike capability.
- Built to carry a warhead payload of up to **500 kg** and has an operational range of at least 1500 km and operational up to speeds of Mach 0.8 at altitude between **30–50 meters AGL**, to evade radar detection.
- \circ Additional high altitude cruise mode enabling long-range efficiency. Optimized fuselage employing Whitcomb Area Rule to reduce total drag up to 8% while in the transonic cruise stage.

- Participated in the IRoC-U challenge with a focus on implementing LiDAR-based 3D mapping, SLAM algorithms, and machine learning models for autonomous navigation and landing in GPS-denied environments.
- Engineered a modular, dual-mode (manual/autonomous) system with a Emergency Response System (ERS), with high-accuracy terrain analysis and precision landing under dynamic conditions with a 50 cm landing radius accuracy, supported by real-time telemetry, redundancy systems.

Minimum Length Rocket Nozzle Design using Method of Characteristics & Scilab Case-Study Hackathon

FOSSEE, IIT Bombay

- Developed a 2D minimum-length rocket nozzle design using the Theory of Characteristics, aiming for optimal Mach number and uniform flow at the exit and implemented a Scilab-based numerical solver for supersonic, inviscid, and irrotational flow equations.
- Incorporated a surrogate-based optimization approach to refine the nozzle profile, improving aerodynamic efficiency while reducing computational load; emphasized performance optimization in the diverging section.

Validation and Verification of Aerospike Designs in Supersonic Flow CopenFOAM Case Study Project

FOSSEE, IIT Bombay

Conducted a numerical study in OpenFOAM to analyze supersonic flow over blunt bodies with and without an aerospike,
 evaluating the effects of varying Mach numbers with a constant Reynold's number on pressure distribution, velocity profiles,
 and temperature gradients; validated results against Crawford's experimental data.