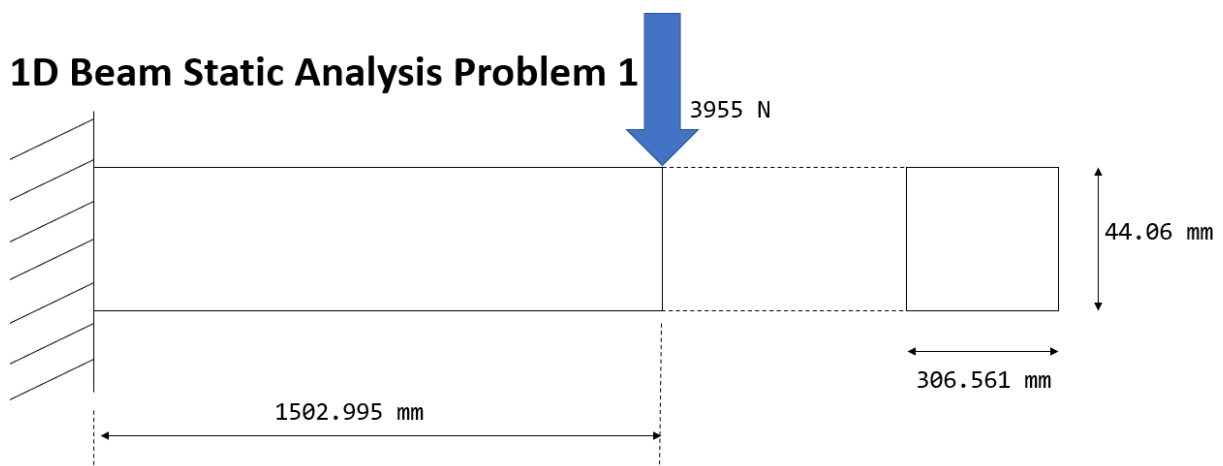


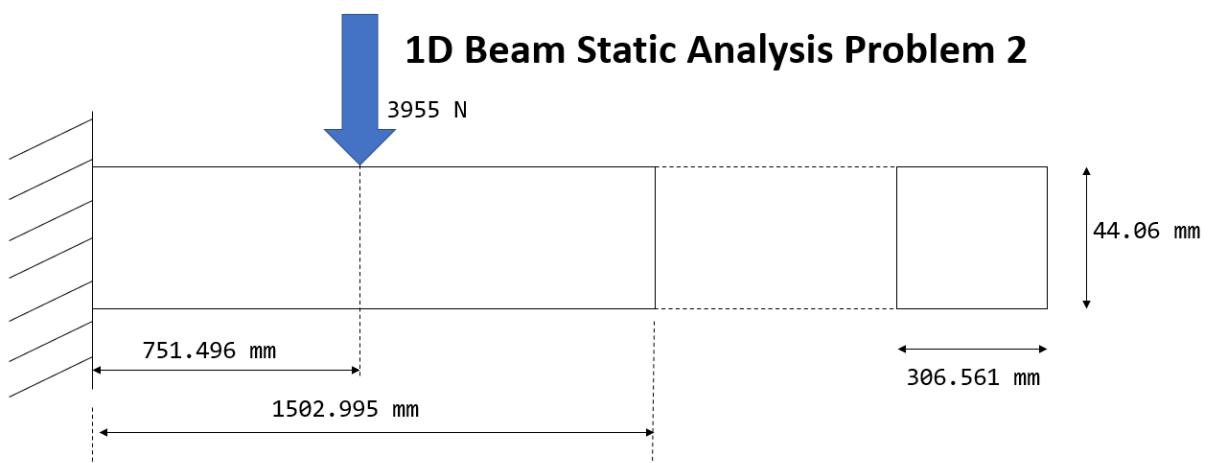
1D Beam Finite Element Method (FEM) Analysis

- This project contains Python code for performing Finite Element Method (FEM) analysis on 1D beam elements.
- The code includes implementations for Static-Linear, and modal analysis.
- Theoretical solutions for these analyses were derived based on established principles and then used to verify the accuracy of the FEM code.
- The Sympy library from Python was employed for symbolic mathematics to aid in the derivation and verification of theoretical solutions.
- This project demonstrates proficiency in FEM, numerical analysis, and symbolic computation using Python.

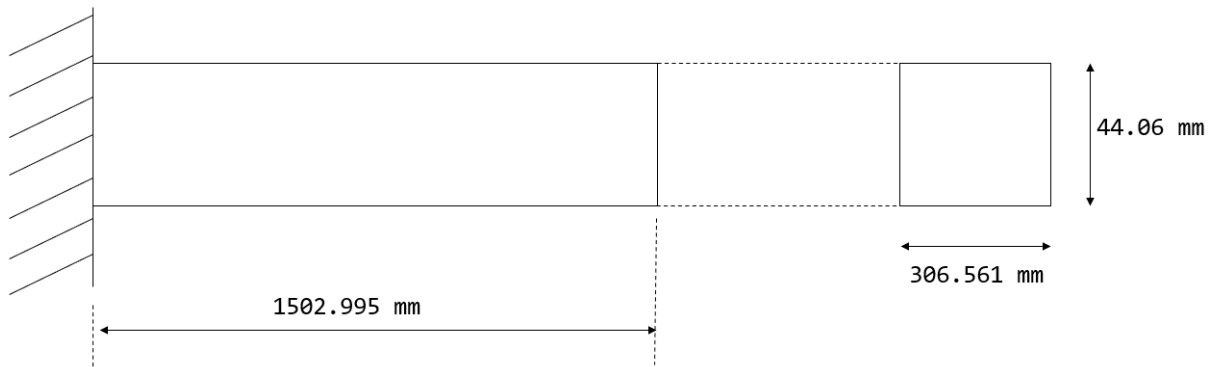
1D Beam Static Analysis Problem 1



1D Beam Static Analysis Problem 2



1D Beam Modal Analysis Problem



Derivation of Local Stiffness and Mass Matrix for FEM

$$\mathbf{M}\ddot{\mathbf{u}}(t) + \mathbf{K}_e\dot{\mathbf{u}}(t) = \mathbf{f}(t)$$

$$\mathbf{M} = \int_V \mathbf{\Psi}^T(\mathbf{x})\rho\mathbf{\Psi}(\mathbf{x})dV$$

$$\mathbf{K}_e = \int_V \mathbf{B}_e^T(\mathbf{x})\mathbf{S}_e(E, \nu)\mathbf{B}_e(\mathbf{x})dV$$

For 1D Bar the shape functions matrix is, $\mathbf{\Psi}(\mathbf{x}) =$

$$\begin{bmatrix} 1 - 3\left(\frac{x}{l_e}\right)^2 + 2\left(\frac{x}{l_e}\right)^3 & \left(\frac{x}{l_e} - 2\left(\frac{x}{l_e}\right)^2 + \left(\frac{x}{l_e}\right)^3\right)l_e & 3\left(\frac{x}{l_e}\right)^2 - 2\left(\frac{x}{l_e}\right)^3 & \left(-\left(\frac{x}{l_e}\right)^2 + \left(\frac{x}{l_e}\right)^3\right)l_e \end{bmatrix}$$

$$\mathbf{B}_e = \frac{\partial^2 \mathbf{\Psi}(\mathbf{x})}{\partial x^2}$$

$$\mathbf{M} = \int_{x=0}^L [\mathbf{\Psi}^T(x)\rho\mathbf{\Psi}(x)]_{6 \times 6} A dx$$

$$\mathbf{K}_e = \int_{x=0}^{l_e} \int_A [\mathbf{B}_e^T(x, y)E\mathbf{B}_e(x, y)]_{6 \times 6} dA dx$$

$$\epsilon_{xx} = \frac{\partial u_x}{\partial x} = \mathbf{B}_e \dot{\mathbf{u}}$$

$$\sigma_{xx} = E \epsilon_{xx} = E \frac{\partial^2 u_y}{\partial x^2} y = E \mathbf{B}_e \dot{\mathbf{u}}$$

Deriving Actual Solutions for Static-Linear Beam Analysis

Actual Stress on the beam

$$\sigma_{xx} = \frac{M}{I} y$$

Now, to calculate tip deflection on Beam for Problem 1

Problem 1

For a beam we can write,

$$EI \frac{d^2 u}{dx^2} = M$$

$$EI u'' = -F(L-x)$$

$$EI u' = -F\left(Lx - \frac{x^2}{2}\right) + C_1$$

$$u'(0) = 0$$

$$EI(0) = -F(0-0) + C_1 \Rightarrow \boxed{C_1 = 0}$$

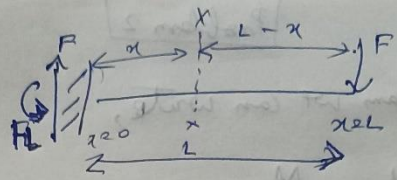
$$EI u = -F\left(L\frac{x^2}{2} - \frac{x^3}{6}\right) + C_2$$

$$u(0) = 0$$

$$EI(0) = -F(0-0) + C_2 \Rightarrow \boxed{C_2 = 0}$$

$$u(x) = -\frac{F}{EI} \left(L\frac{x^2}{2} - \frac{x^3}{6} \right)$$

At tip,

$$u(L) = u = \boxed{-\frac{FL^3}{3EI}}$$


For Fixed Free boundary condition

$$u(0) = 0$$

$$u'(0) = 0$$

$M = -F(L-x)$

Now, to calculate tip deflection on Beam for Problem 2

Problem 2

For a beam we can write,

$$EI \frac{d^2 u}{dx^2} = M$$

For case of $0 \leq x \leq a$

$$M = F(a-x)$$

$$EI u'' = F(a-x)$$

$$EI u' = F\left(ax - \frac{x^2}{2}\right) + C_1 \Rightarrow EI(0) = F(0+0) + C_1 \Rightarrow C_1 = 0$$

$$EI u = F\left(\frac{ax^2}{2} - \frac{x^3}{6}\right) + C_2 \Rightarrow EI(0) = F(0+0) + C_2 \Rightarrow C_2 = 0$$

At $x=a$, $EI(0) = F(0-0) + C_2 \Rightarrow C_2 = 0$

$$u(x) = \frac{F}{EI} \left(\frac{ax^2}{2} - \frac{x^3}{6} \right) \Rightarrow u(a) = \frac{Fa^3}{3EI}$$

$0 \leq x \leq a$

For case of $a \leq x \leq L$

$$M = Fa - Fx + F(x-a) = 0$$

$$M = 0$$

$$EI u'' = 0$$

$$EI u' = C_3 \Rightarrow EI u'(a) = C_3 \Rightarrow EI \left(\frac{Fa^2}{2EI} \right) = C_3 \Rightarrow C_3 = \frac{Fa^2}{2}$$

$$EI u = \frac{Fa^2 x}{2} + C_4 \Rightarrow EI u(a) = C_4 \Rightarrow EI \left(\frac{Fa^3}{3EI} \right) = \frac{Fa^2 a}{2} + C_4$$

$$C_4 = -\frac{Fa^3}{6}$$

$a \leq x \leq L$

$$U(x) = \frac{Fa^2}{EI} \left[\frac{Fa^2 x}{2} - \frac{Fa^3}{6} \right] = \frac{Fa^2}{6EI} (3x - a)$$

$$U(L) = U$$

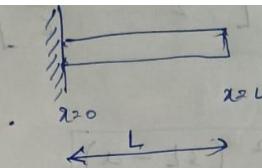
$$U = \frac{Fa^2}{6EI} (3L - a)$$

Deriving equation of Natural Frequencies in Beam

The Strong form of beam,

$$EI \frac{\partial^4 u}{\partial x^4} + \rho A \frac{\partial^2 u}{\partial t^2} = 0$$

$$c^2 \frac{\partial^4 u}{\partial x^4} + \frac{\partial^2 u}{\partial t^2} = 0 \quad \left[c^2 = \frac{EI}{\rho A} \right]$$



Let solution be, $u(x, t) = U(x) \cos(\omega_n t)$

$$c^2 \frac{\partial^4}{\partial x^4} [U(x) \cos(\omega_n t)] + \frac{\partial^2}{\partial t^2} [U(x) \cos(\omega_n t)] = 0$$

$$c^2 U^{(4)} \cos(\omega_n t) + U (-\omega_n^2 \cos(\omega_n t)) = 0$$

$$U^{(4)} - \frac{\omega_n^2}{c^2} U = 0 \quad \beta^4 = \frac{\omega_n^2}{c^2}$$

$$U^{(4)} - \beta^4 U = 0$$

$$(D^4 - \beta^4) U = 0 \Rightarrow (D^4 - \beta^4) = 0 \Rightarrow \underbrace{(D^2 - \beta^2)}_{\text{Hyperbolic}} \underbrace{(D^2 + \beta^2)}_{\text{Trigonometric}} = 0$$

$$U(x, t) = A \sinh(\beta x) + B \cosh(\beta x) + C \sin(\beta x) + D \cos(\beta x)$$

$$U(x, t) = [A \sinh(\beta x) + B \cosh(\beta x) + C \sin(\beta x) + D \cos(\beta x)] \cos(\omega_n t)$$

$$\beta^4 = \frac{\omega_n^2}{c^2} = \frac{\omega_n^2}{EI/\rho A} \Rightarrow \omega_n = \beta^2 \sqrt{\frac{EI}{\rho A}}$$

The boundary conditions for Fixed-Free are

$$u = 0, \quad \frac{\partial u}{\partial x} = 0 \quad \text{At } x = 0$$

$$\frac{\partial^2 u}{\partial x^2} = 0, \quad \frac{\partial^4 u}{\partial x^4} = 0 \quad \text{At } x = L$$

$$u = A \cosh(\beta x) + B \sinh(\beta x) + C \cos(\beta x) + D \sin(\beta x)$$

$$\frac{\partial u}{\partial x} = A\beta \sinh(\beta x) + B\beta \cosh(\beta x) - C\beta \sin(\beta x) + D\beta \cos(\beta x)$$

$$\frac{\partial^2 u}{\partial x^2} = A\beta^2 \cosh(\beta x) + B\beta^2 \sinh(\beta x) - C\beta^2 \cos(\beta x) - D\beta^2 \sin(\beta x)$$

$$\frac{\partial^3 u}{\partial x^3} = A\beta^3 \sinh(\beta x) + B\beta^3 \cosh(\beta x) + C\beta^3 \sin(\beta x) - D\beta^3 \cos(\beta x)$$

$$u(0) = 0 \Rightarrow 0 = A(1) + B(0) + C(1) + D(0) \Rightarrow \boxed{A + C = 0}$$

$$\frac{\partial u}{\partial x} \Big|_{x=0} = 0 \Rightarrow 0 = A(0) + B(1) - C(0) + D(1) \Rightarrow \boxed{B - D = 0}$$

$$\frac{\partial^2 u}{\partial x^2} \Big|_{x=L} = 0 \Rightarrow 0 = A \cosh(\beta L) + B \sinh(\beta L) - A \cos(\beta L) + B \sin(\beta L)$$

$$\frac{\partial^3 u}{\partial x^3} \Big|_{x=L} = 0 \Rightarrow 0 = A \sinh(\beta L) + B \cosh(\beta L) - A \sin(\beta L) + B \cos(\beta L)$$

$$0 = [\cosh(\beta L) + \cos(\beta L)] A + [\sinh(\beta L) + \sin(\beta L)] B$$

$$0 = [\sinh(\beta L) - \sin(\beta L)] A + [\cosh(\beta L) + \cos(\beta L)] B$$

$$\begin{bmatrix} \cosh(\beta L) + \cos(\beta L) & \sinh(\beta L) + \sin(\beta L) \\ \sinh(\beta L) - \sin(\beta L) & \cosh(\beta L) + \cos(\beta L) \end{bmatrix} \begin{Bmatrix} A \\ B \end{Bmatrix} = 0$$

$$\begin{vmatrix} \cosh(\beta L) + \cos(\beta L) & \sinh(\beta L) + \sin(\beta L) \\ \sinh(\beta L) - \sin(\beta L) & \cosh(\beta L) + \cos(\beta L) \end{vmatrix} = 0$$

$$\sin^2(\beta L) + \cos^2(\beta L) + \cosh^2(\beta L) - \sinh^2(\beta L) + 2 \cosh(\beta L) \cos(\beta L) = 0$$

$$\boxed{\cos(\beta L) \cosh(\beta L) = -1}$$

This roots (β) can be obtained by using Taylor's series expansion