```
C
      CLOSE6.FOR
                   (ErikSoft 5 June 2001)
C
c Author: John E. Chambers
c Makes output files containing details of close encounters that occurred
c during an integration using Mercury6 or higher.
c The user specifies the names of the required objects in the file close.in
C-
C
     implicit none
     include 'mercury.inc'
C
     integer itmp,i,j,k,l,iclo,jclo,precision,lenin
     integer nmaster,nopen,nwait,nbig,nsml,nsub,lim(2,100)
     integer year,month,timestyle,line_num,lenhead,lmem(NMESS)
     integer nchar, algor, allflag, firstflag, ninfile
     integer unit(NMAX), master_unit(NMAX)
     real*8 time, t0, t1, rmax, rcen, rfac, dclo, mcen, jcen(3)
     real*8 mio_c2re, mio_c2fl,fr,theta,phi,fv,vtheta,vphi,gm
     real*8 x1(3), x2(3), v1(3), v2(3), m(NMAX)
     real*8 a1,a2,e1,e2,i1,i2,p1,p2,n1,n2,l1,l2,q1,q2
     logical test
     character*250 string,fout,header,infile(50)
     character*80 mem(NMESS),cc,c(NMAX)
     character*8 master_id(NMAX),id(NMAX)
     character*5 fin
     character*1 check,style,type,c1
C-----
     allflag = 0
c Read in output messages
     inquire (file='message.in', exist=test)
     if (.not.test) then
       write (*,'(/,2a)') ' ERROR: This file is needed to continue: ',
         ' message.in'
     end if
     open (14, file='message.in', status='old')
  10 continue
       read (14,'(i3,1x,i2,1x,a80)',end=20) j,lmem(j),mem(j)
     goto 10
  20 close (14)
c Open file containing parameters for this programme
     inquire (file='close.in', exist=test)
     if (test) then
       open (10, file='close.in', status='old')
     else
      call mio_err (6,mem(81),lmem(81),mem(88),lmem(88),' ',1,
        'close.in',9)
     end if
c Read number of input files
  30 read (10, '(a250)') string
     if (string(1:1).eq.')') goto 30
     call mio_spl (250,string,nsub,lim)
     read (string(lim(1,nsub):lim(2,nsub)),*) ninfile
c Make sure all the input files exist
     do j = 1, ninfile
       read (10, '(a250)') string
       if (string(1:1).eq.')') goto 40
       call mio_spl (250,string,nsub,lim)
       infile(j)(\bar{1}:(lim(2,1)-lim(1,1)+1)) = string(lim(1,1):lim(2,1))
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```
inquire (file=infile(j), exist=test)
        if (.not.test) call mio_err (6,mem(81),lmem(81),mem(88),
          lmem(88),' ',1,infile(j),80)
c Read parameters used by this programme
      timestyle = 1
      do j = 1, 2
        read (10,'(a250)') string
        if (string(1:1).eq.')') goto 50
        call mio_spl (250,string,nsub,lim)
        c1 = string(lim(1,nsub):lim(2,nsub))
        if (j.eq.1.and.(c1.eq.'d'.or.c1.eq.'D')) timestyle = 0
        if (j.eq.2.and.(c1.eq.'y'.or.c1.eq.'Y')) timestyle = timestyle+2
      end do
c Read in the names of the objects for which orbital elements are required
      nopen = 0
      nwait = 0
      nmaster = 0
      call m_formce (timestyle,fout,header,lenhead)
  60 continue
        read (10,'(a250)',end=70) string
        call mio_spl (250,string,nsub,lim)
        if (string(1:1).eq.')'.or.lim(1,1).eq.-1) goto 60
c Either open an aei file for this object or put it on the waiting list
        nmaster = nmaster + 1
        itmp = min(7, lim(2,1) - lim(1,1))
        master_id(nmaster)='
        master\_id(nmaster)(1:itmp+1) = string(lim(1,1):lim(1,1)+itmp)
        if (nopen.lt.NFILES) then
          nopen = nopen + 1
          master_unit(nmaster) = 10 + nopen
          call mio_aei (master_id(nmaster),'.clo', master_unit(nmaster),
            header,lenhead,mem,lmem)
        else
          nwait = nwait + 1
          master\_unit(nmaster) = -2
        end if
      goto 60
  70 continue
c If no objects are listed in CLOSE.IN assume that all objects are required
      if (nopen.eq.0) allflag = 1
      close (10)
C
C-
C
  LOOP OVER EACH INPUT FILE CONTAINING INTEGRATION DATA
С
  90 continue
      firstflag = 0
      do i = 1, ninfile
        line_num = 0
        open (10, file=infile(i), status='old')
c Loop over each time slice
        continue
        line_num = line_num + 1
        read (10, '(3a1)', end=900, err=666) check, style, type
        line_num = line_num - 1
        backspace 10
c Check if this is an old style input file
        if (ichar(check).eq.12.and.(style.eq.'0'.or.style.eq.'1'.or.
          style.eq.'2'.or.style.eq.'3'.or.style.eq.'4')) then
write (*,'(/,2a)') ' ERROR: This is an old style data file',
                     Try running m_close5.for instead.'
          stop
        end if
        if (ichar(check).ne.12) goto 666
```

```
C-
C
   IF SPECIAL INPUT, READ TIME, PARAMETERS, NAMES, MASSES ETC.
C
C
        if (type.eq.'a') then
          line_num = line_num + 1
          read (10,'(3x,i2,a62,i1)') algor,cc(1:62),precision
c Decompress the time, number of objects, central mass and J components etc.
          time = mio_c2fl (cc(1:8))
          if (firstflag.eq.0) then
            t0 = time
            firstflag = 1
          end if
          nbig = int(.5d0 + mio_c2re(cc(9:16), 0.d0, 11239424.d0, 3))
          nsml = int(.5d0 + mio_c2re(cc(12:19), 0.d0, 11239424.d0, 3))
          mcen = mio_c2fl (cc(15:22)) * K2
          jcen(1) = mio_c2fl (cc(23:30))
          jcen(2) = mio_c2fl (cc(31:38))
          jcen(3) = mio_c2fl (cc(39:46))
          rcen = mio_c2fl (cc(47:54))
          rmax = mio_c2fl (cc(55:62))
          rfac = log10 (rmax / rcen)
c Read in strings containing compressed data for each object
          do j = 1, nbig + nsml
            line_num = line_num + 1
            read (10,'(a)',err=666) c(j)(1:51)
          end do
c Create input format list
          if (precision.eq.1) nchar = 2
          if (precision.eq.2) nchar = 4
          if (precision.eq.3) nchar = 7
          lenin = 3 + 6 * nchar
fin(1:5) = '(a00)'
          write (fin(3:4),'(i2)') lenin
c For each object decompress its name, code number, mass, spin and density
          do j = 1, nbig + nsml
            k = int(.5d0 + mio_c2re(c(j)(1:8), 0.d0, 11239424.d0, 3))
            id(k) = c(j)(4:11)
            m(k) = mio_c2fl(c(j)(12:19)) * K2
c Find the object on the master list
            unit(k) = 0
            do 1 = 1, nmaster
              if (id(k).eq.master_id(l)) unit(k) = master_unit(l)
            end do
c If object is not on the master list, add it to the list now
            if (unit(k).eq.0) then
              nmaster = nmaster + 1
              master_id(nmaster) = id(k)
c Either open an aei file for this object or put it on the waiting list
              if (allflag.eq.1) then
                if (nopen.lt.NFILES) then
                  nopen = nopen + 1
                  master\_unit(nmaster) = 10 + nopen
                  call mio_aei (master_id(nmaster),'.clo',
                    master_unit(nmaster), header, lenhead, mem, lmem)
                  nwait = nwait + 1
                  master\_unit(nmaster) = -2
                end if
              else
                master\_unit(nmaster) = -1
              end if
              unit(k) = master_unit(nmaster)
            end if
```

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end do
C
C-
C
   IF NORMAL INPUT, READ COMPRESSED DATA ON THE CLOSE ENCOUNTER
C
         else if (type.eq.'b') then
           line_num = line_num + 1
           read (10,'(3x,a70)',err=666) cc(1:70)
c Decompress time, distance and orbital variables for each object
           time = mio_c2fl(cc(1:8))
           iclo = int(.5d0 + mio_c2re(cc(9:16),  0.d0, 11239424.d0, 3))
jclo = int(.5d0 + mio_c2re(cc(12:19), 0.d0, 11239424.d0, 3))
           if (iclo.gt.NMAX.or.jclo.gt.NMAX) then
             write (*,'(/,2a)') mem(81)(1:lmem(81)),
               mem(90)(1:lmem(90))
             stop
           end if
           dclo = mio_c2fl (cc(15:22))
                   = mio_c2re (cc(23:30), 0.d0, rfac,
           theta = mio_c2re (cc(27:34), 0.d0, PI,
                                                           4)
                   = mio_c2re (cc(31:38), 0.d0, TWOPI, 4)
           fv = mio_c2re (cc(35:42), 0.d0, 1.d0, vtheta = mio_c2re (cc(39:46), 0.d0, PI,
                                                           4)
                                                           4)
           vphi = mio_c2re(cc(43:50), 0.d0, TWOPI, 4)
           call mco_ov2x (rcen,rmax,mcen,m(iclo),fr,theta,phi,fv,
             vtheta, vphi, x1(1), x1(2), x1(3), v1(1), v1(2), v1(3))
C
           fr = mio_c2re (cc(47:54), 0.d0, rfac,
theta = mio_c2re (cc(51:58), 0.d0, PI,
                                                           4)
                   = mio_c2re (cc(55:62), 0.d0, TWOPI, 4)
                   = mio_c2re (cc(59:66), 0.d0, 1.d0, 4)
           vtheta = mio_c2re (cc(63:70), 0.d0, PI, 4)

vphi = mio_c2re (cc(67:74), 0.d0, TWOPI, 4)
           call mco_ov2x (rcen,rmax,mcen,m(jclo),fr,theta,phi,fv,
             vtheta, vphi, x2(1), x2(2), x2(3), v2(1), v2(2), v2(3))
c Convert to Keplerian elements
           gm = mcen + m(iclo)
           call mco_x2el (gm, x1(1), x1(2), x1(3), v1(1), v1(2), v1(3),
             q1,e1,i1,p1,n1,l1)
           a1 = q1 / (1.d0 - e1)
           gm = mcen + m(jclo)
           call mco_x2el (gm,x2(1),x2(2),x2(3),v2(1),v2(2),v2(3),
             q2,e2,i2,p2,n2,l2)
           a2 = q2 / (1.d0 - e2)
           i1 = i1 / DR
           i2 = i2 / DR
c Convert time to desired format
           if (timestyle.eq.0) t1 = time
           if (timestyle.eq.1) call mio_jd_y (time,year,month,t1)
           if (timestyle.eq.2) t1 = time - t0
           if (timestyle.eq.3) t1 = (time - t0) / 365.25d0
c Write encounter details to appropriate files
           if (timestyle.eq.1) then
             if (unit(iclo).ge.10) write (unit(iclo), fout) year, month,
     2
               t1, id(jclo), dclo, a1, e1, i1, a2, e2, i2
C
             if (unit(jclo).ge.10) write (unit(jclo), fout) year, month,
     %
               t1,id(iclo),dclo,a2,e2,i2,a1,e1,i1
           else
             if (unit(iclo).ge.10) write (unit(iclo), fout) t1, id(jclo),
               dclo,a1,e1,i1,a2,e2,i2
              if (unit(jclo).ge.10) write (unit(jclo), fout) t1, id(iclo),
               dclo,a2,e2,i2,a1,e1,i1
           end if
C
C
C
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IF TYPE IS NOT 'a' OR 'b', THE INPUT FILE IS CORRUPTED
       else
        goto 666
       end if
c Move on to the next time slice
       goto 100
c If input file is corrupted, try to continue from next uncorrupted time slice
       write (*,'(2a,/,a,i10)') mem(121)(1:lmem(121)),
        infile(i)(1:60),mem(104)(1:lmem(104)),line_num
       c1 =
       do while (ichar(c1).ne.12)
        line_num = line_num + 1
        read (10, '(a1)', end=900) c1
       end do
       line_num = line_num - 1
       backspace 10
c Move on to the next file containing close encounter data
       continue
       close (10)
     end do
c Close clo files
     do j = 1, nopen
      close (10+j)
     end do
     nopen = 0
C
c If some objects remain on waiting list, read through input files again
     if (nwait.gt.0) then
       do j = 1, nmaster
         if (master_unit(j).ge.10) master_unit(j) = -1
         if (master_unit(j).eq.-2.and.nopen.lt.NFILES) then
          nopen = nopen + 1
          nwait = nwait - 1
          master\_unit(j) = 10 + nopen
          call mio_aei (master_id(j),'.clo',master_unit(j),header,
            lenhead,mem,lmem)
        end if
       end do
       goto 90
     end if
C
     end
C
                    (ErikSoft 30 November 1999)
C
C
c Author: John E. Chambers
C
C
C
     subroutine m_formce (timestyle,fout,header,lenhead)
C
     implicit none
c Input/Output
     integer timestyle, lenhead
     character*250 fout, header
C
C
C
     if (timestyle.eq.0.or.timestyle.eq.2) then
       header(1:19) = ' Time (days)
```

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header(20:58) = 'Object dmin (AU) al header(59:90) = 'il a2 e2
       lenhead = 90
       fout = (1x, f18.5, 1x, a8, 1x, f10.8, 2(1x, f9.4, 1x, f8.6, 1x, f7.3))
     else
       if (timestyle.eq.1) then
         header(1:23) = ' Year/Month/Day '
header(24:62) = ' Object dmin (AU) a1
header(63:94) = ' i1 a2 e2
         lenhead = 94
         fout(1:37) = '(1x,i10,1x,i2,1x,f8.5,1x,a8,1x,f10.8,'
         fout(38:64) = '2(1x, f9.4, 1x, f8.6, 1x, f7.3))'
       else
         header(1:19) = ' Time (years)
         header(20:58) = 'Object dmin (AU) al
header(59:90) = 'il a2 e2 i2'
         fout = (1x, f18.7, 1x, a8, 1x, f10.8, 2(1x, f9.4, 1x, f8.6, 1x, f7.3))
         lenhead = 90
       end if
     end if
C-
C
     return
     end
MCO_OV2X.FOR (ErikSoft 28 February 2001)
C
c Author: John E. Chambers
c Converts output variables for an object to coordinates and velocities.
c The output variables are:
c r = the radial distance
c theta = polar angle
c phi = azimuthal angle
c fv = 1 / [1 + 2(ke/be)^2], where be and ke are the object's binding and
                           kinetic energies. (Note that 0 < fv < 1).
c vtheta = polar angle of velocity vector
c vphi = azimuthal angle of the velocity vector
C
C---
C
     subroutine mco_ov2x (rcen,rmax,mcen,m,fr,theta,phi,fv,vtheta,
    % vphi,x,y,z,u,v,w)
C
     implicit none
     include 'mercury.inc'
c Input/Output
     real*8 rcen,rmax,mcen,m,x,y,z,u,v,w,fr,theta,phi,fv,vtheta,vphi
C
    real*8 r, v1, temp
C
C-----
C
       r = rcen * 10.d0**fr
       temp = sqrt(.5d0*(1.d0/fv - 1.d0))
       v1 = sqrt(2.d0 * temp * (mcen + m) / r)
       x = r * sin(theta) * cos(phi)
       y = r * sin(theta) * sin(phi)
       z = r * cos(theta)
       u = v1 * sin(vtheta) * cos(vphi)
       v = v1 * sin(vtheta) * sin(vphi)
       w = v1 * cos(vtheta)
```

//25

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~/PlanetProject/mercury/
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```
return
C
      MCO_EL2X.FOR
                   (ErikSoft 7 July 1999)
C
c Author: John E. Chambers
c Calculates Cartesian coordinates and velocities given Keplerian orbital
c elements (for elliptical, parabolic or hyperbolic orbits).
c Based on a routine from Levison and Duncan's SWIFT integrator.
C
  mu = grav const * (central + secondary mass)
C
  q = perihelion distance
c e = eccentricity
  i = inclination
c p = longitude of perihelion !!! ) in
c n = longitude of ascending node ) radians
  1 = mean anomaly
c x,y,z = Cartesian positions (units the same as a)
  u,v,w = " velocities ( units the same as sqrt(mu/a) )
C
     subroutine mco_el2x (mu,q,e,i,p,n,l,x,y,z,u,v,w)
C
     implicit none
     include 'mercury.inc'
C
c Input/Output
     real*8 mu,q,e,i,p,n,l,x,y,z,u,v,w
C
     real*8 g,a,ci,si,cn,sn,cg,sg,ce,se,romes,temp
     real*8 z1,z2,z3,z4,d11,d12,d13,d21,d22,d23
     real*8 mco_kep, orbel_fhybrid, orbel_zget
C
C-
C
c Change from longitude of perihelion to argument of perihelion
     g = p - n
C
c Rotation factors
     call mco_sine (i,si,ci)
     call mco_sine (g,sg,cg)
     call mco_sine (n,sn,cn)
     z1 = cg * cn
     z2 = cg * sn
     z3 = sg * cn
     z4 = sg * sn
     d11 = z1 - z4*ci
d12 = z2 + z3*ci
     d13 = sg * si
     d21 = -z3 - z2*ci
     d22 = -z4 + z1*ci
     d23 = cg * si
C
c Semi-major axis
     a = q / (1.d0 - e)
C
c Ellipse
     if (e.lt.1.d0) then
       romes = sqrt(1.d0 - e*e)
       temp = mco_kep (e,1)
       call mco_sine (temp,se,ce)
       z1 = a * (ce - e)
```

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z2 = a * romes * se
      temp = sqrt(mu/a) / (1.d0 - e*ce)
       z3 = -se * temp
       z4 = romes * ce * temp
     else
c Parabola
       if (e.eq.1.d0) then
        ce = orbel_zget(1)
        z1 = q * (1.d0 - ce*ce)
        z2 = 2.d0 * q * ce
        z4 = sqrt(2.d0*mu/q) / (1.d0 + ce*ce)
        z3 = -ce * z4
      else
c Hyperbola
        romes = sqrt(e*e - 1.d0)
        temp = orbel_fhybrid(e,1)
        call mco_sinh (temp,se,ce)
        z1 = a * (ce - e)
        z2 = -a * romes * se
        temp = sqrt(mu/abs(a)) / (e*ce - 1.d0)
        z3 = -se * temp
        z4 = romes * ce * temp
      end if
     endif
C
     x = d11*z1 + d21*z2
     y = d12*z1 + d22*z2
     z = d13*z1 + d23*z2
     u = d11*z3 + d21*z4
     v = d12*z3 + d22*z4
     w = d13*z3 + d23*z4
C
C-----
     return
C
     MCO KEP.FOR
                 (ErikSoft 7 July 1999)
c Author: John E. Chambers
c Solves Kepler's equation for eccentricities less than one.
c Algorithm from A. Nijenhuis (1991) Cel. Mech. Dyn. Astron. 51, 319-330.
c e = eccentricity
c l = mean anomaly (radians)
c u = eccentric anomaly ( " )
C
C-----
    function mco_kep (e,oldl)
     implicit none
C
c Input/Outout
    real*8 oldl,e,mco_kep
C
c Local
    real*8 l,pi,twopi,piby2,u1,u2,ome,sign
    real*8 x,x2,sn,dsn,z1,z2,z3,f0,f1,f2,f3
     real*8 p,q,p2,ss,cc
     logical flag, big, bigg
C
C
     pi = 3.141592653589793d0
     twopi = 2.d0 * pi
     piby2 = .5d0 * pi
C
```

```
c Reduce mean anomaly to lie in the range 0 < 1 < pi
      if (oldl.ge.0) then
        1 = mod(oldl, twopi)
      else
       1 = mod(oldl, twopi) + twopi
      end if
      sign = 1.d0
      if (l.gt.pi) then
        l = twopi - 1
       sign = -1.d0
      end if
C
      ome = 1.d0 - e
C
      if (1.ge..45d0.or.e.lt..55d0) then
С
c Regions A,B or C in Nijenhuis
C -----
c Rough starting value for eccentric anomaly
        if (1.1t.ome) then
          u1 = ome
        else
          if (l.gt.(pi-1.d0-e)) then
           u1 = (1+e*pi)/(1.d0+e)
          else
           u1 = 1 + e
          end if
        end if
c Improved value using Halley's method
        flag = u1.gt.piby2
        if (flag) then
          x = pi - u1
        else
         x = u1
        end if
        x2 = x*x
        sn = x*(1.d0 + x2*(-.16605 + x2*.00761))
        dsn = 1.d0 + x2*(-.49815 + x2*.03805)
        if (flag) dsn = -dsn
        f2 = e*sn
        f0 = u1 - f2 - 1
        f1 = 1.d0 - e*dsn
        u2 = u1 - f0/(f1 - .5d0*f0*f2/f1)
      else
C
c Region D in Nijenhuis
C
c Rough starting value for eccentric anomaly
        z1 = 4.d0*e + .5d0
        p = ome / z1
        q = .5d0 * 1 / z1
        p^2 = p*p
z^2 = \exp(\log(\operatorname{dsqrt}(p^2*p + q*q) + q)/1.5)
        u1 = 2.d0*q / (z2 + p + p2/z2)
c Improved value using Newton's method
        z2 = u1*u1
        z3 = z2*z2
        u2 = u1 - .075d0*u1*z3 / (ome + z1*z2 + .375d0*z3)
        u2 = 1 + e*u2*( 3.d0 - 4.d0*u2*u2 )
      end if
c Accurate value using 3rd-order version of Newton's method
c N.B. Keep cos(u2) rather than sqrt( 1-sin^2(u2) ) to maintain accuracy!
c First get accurate values for u2 - \sin(u2) and 1 - \cos(u2)
      bigg = (u2.gt.piby2)
      if (bigg) then
        z3 = pi - u2
```

```
z3 = u2
                        end if
                        big = (z3.gt.(.5d0*piby2))
                        if (big) then
                               x = piby2 - z3
                        else
                            x = z3
                        end if
                        x2 = x*x
                        ss = 1.d0
                        cc = 1.d0
 C
                       ss = x*x2/6.*(1. - x2/20.*(1. - x2/42.*(1. - x2/72.*(1. - x2/72.*(1.
                    % x2/110.*(1. - x2/156.*(1. - x2/210.*(1. - x2/272.)))))) cc = x2/2.*(1. - x2/12.*(1. - x2/30.*(1. - x2/56.*(1. - x2/5
                     % x2/ 90.*(1. - x2/132.*(1. - x2/182.*(1. - x2/240.*(1. -
                     % x2/306.)))))))
                        if (big) then
                                z1 = cc + z3 - 1.d0
                                z2 = ss + z3 + 1.d0 - piby2
                        else
                               z1 = ss
                               z2 = cc
                        end if
C
                        if (bigg) then
                               z1 = 2.d0*u2 + z1 - pi
                               z2 = 2.d0 - z2
                        end if
                       f0 = 1 - u2*ome - e*z1
                        f1 = ome + e*z2
                        f2 = .5d0*e*(u2-z1)
                       f3 = e/6.d0*(1.d0-z2)
                        z1 = f0/f1
                        z2 = f0/(f2*z1+f1)
                        mco\_kep = sign*(u2 + f0/((f3*z1+f2)*z2+f1))
C
C-----
                       return
                       end
C
\mathbb{C}^{8}
                           MCO_SINE.FOR
                                                                                   (ErikSoft 17 April 1997)
C
c Author: John E. Chambers
 c Calculates sin and cos of an angle X (in radians).
C
C-----
C
                      subroutine mco_sine (x,sx,cx)
 C
                      implicit none
C
 c Input/Output
                     real*8 x,sx,cx
C
 c Local
                      real*8 pi,twopi
C-----
 С
                       pi = 3.141592653589793d0
```

```
twopi = 2.d0 * pi
C
    if (x.gt.0) then
     x = mod(x, twopi)
    else
     x = mod(x, twopi) + twopi
    end if
C
    cx = cos(x)
C
    if (x.gt.pi) then
     sx = -sqrt(1.d0 - cx*cx)
     sx = sqrt(1.d0 - cx*cx)
    end if
C
C.
C
    return
    end
(ErikSoft 12 June 1998)
C
    MCO_SINH.FOR
C
c Calculates sinh and cosh of an angle X (in radians)
C-----
    subroutine mco_sinh (x,sx,cx)
C
    implicit none
C
c Input/Output
    real*8 x,sx,cx
C
C-
C
    sx = sinh(x)
    cx = sqrt (1.d0 + sx*sx)
C
C-
C
    return
C
    MIO_AEI.FOR (ErikSoft 31 January 2001)
C
c Author: John E. Chambers
c Creates a filename and opens a file to store aei information for an object.
c The filename is based on the name of the object.
C--
C
    subroutine mio_aei (id,extn,unitnum,header,lenhead,mem,lmem)
C
    implicit none
    include 'mercury.inc'
c Input/Output
    integer unitnum,lenhead,lmem(NMESS)
    character*4 extn
    character*8 id
    character*250 header
    character*80 mem(NMESS)
```

```
c Local
     integer j,k,itmp,nsub,lim(2,4)
     logical test
     character*1 bad(5)
     character*250 filename
C
C-
C
     data bad/ '*', '/', '.', ':', '&'/
c Create a filename based on the object's name
     call mio_spl (8,id,nsub,lim)
     itmp = min(7, lim(2,1) - lim(1,1))
     filename(1:itmp+1) = id(1:itmp+1)
     filename(itmp+2:itmp+5) = extn
     do j = itmp + 6, 250
      filename(j:j) = '
C
c Check for inappropriate characters in the filename
     do j = 1, itmp + 1
       do k = 1, 5
        if (filename(j:j).eq.bad(k)) filename(j:j) = '_'
       end do
     end do
c If the file exists already, give a warning and don't overwrite it
     inquire (file=filename, exist=test)
     if (test) then
       write (*,'(/,3a)') mem(121)(1:lmem(121)),mem(87)(1:lmem(87)),
        filename(1:80)
       unitnum = -1
       open (unitnum, file=filename, status='new')
       write (unitnum, '(/,30x,a8,//,a)') id,header(1:lenhead)
C
C
     return
C
C
      MIO_C2FL.FOR
                  (ErikSoft 5 June 2001)
C
c CHARACTER*8 ASCII string into a REAL*8 variable.
c N.B. X will lie in the range -1.e112 < X < 1.e112
C ===
C-
C
     function mio_c2fl (c)
C
     implicit none
C
c Input/Output
     real*8 mio_c2fl
     character*8 c
C
c Local
     real*8 x,mio_c2re
     integer ex
C
C
     x = mio_c2re(c(1:8), 0.d0, 1.d0, 7)
     x = x * 2.d0 - 1.d0
```

```
ex = mod(ichar(c(8:8)) + 256, 256) - 32 - 112
                         mio_c2fl = x * (10.d0**dble(ex))
 С
 C
C
                         return
                         end
C
                            MIO_C2RE.FOR
                                                                                            (ErikSoft 5 June 2001)
C
C
c Author: John E. Chambers
c Converts an ASCII string into a REAL*8 variable X, where XMIN <= X < XMAX,
 c using the new format compression:
{\tt c}\ {\tt X}\ {\tt is}\ {\tt assumed}\ {\tt to}\ {\tt be}\ {\tt made}\ {\tt up}\ {\tt of}\ {\tt NCHAR}\ {\tt base-224}\ {\tt digits},\ {\tt each}\ {\tt one}\ {\tt represented}
 c by a character in the ASCII string. Each digit is given by the ASCII
c number of the character minus 32.
\ensuremath{\mathtt{c}} The first 32 ASCII characters (CTRL characters) are avoided, because they
c cause problems when using some operating systems.
C
                         function mio_c2re (c,xmin,xmax,nchar)
 C
                         implicit none
c Input/output
                         integer nchar
                          real*8 xmin,xmax,mio_c2re
                         character*8 c
c Local
                         integer j
                         real*8 y
 C
 C-
C
                          y = 0
                         do j = nchar, 1, -1
                               y = (y + dble(mod(ichar(c(j:j)) + 256, 256) - 32)) / 224.d0
 C
                         mio\_c2re = xmin + y * (xmax - xmin)
 C
C-
 C
                         return
                         end
\mathbb{C}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}
C
                                                                                            (ErikSoft 6 December 1999)
C
                             MIO_ERR.FOR
୍ଦର୍ଷ ବ୍ୟବ୍ୟ ପ୍ରତ୍ୟ ପର୍ୟ ପ୍ରତ୍ୟ ପର ପ୍ରତ୍ୟ ପର ପ୍ରତ୍ୟ ପର ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ ପ୍ରତ୍ୟ
c Author: John E. Chambers
c Writes out an error message and terminates Mercury.
 C-
 C
                         subroutine mio_err (unit,s1,ls1,s2,ls2,s3,ls3,s4,ls4)
 C
                         implicit none
C
 c Input/Output
                          integer unit, ls1, ls2, ls3, ls4
```

```
character*80 s1,s2,s3,s4
C
C--
    write (*,'(a)') ' ERROR: Programme terminated.'
    write (unit, '(/,3a,/,2a)') s1(1:ls1),s2(1:ls2),s3(1:ls3),
    % ' ',s4(1:ls4)
    stop
C----
C
MCO_H2B.FOR
                (ErikSoft 2 November 2000)
C
c Author: John E. Chambers
c Converts coordinates with respect to the central body to barycentric
C-
     subroutine mco_h2b (jcen,nbod,nbig,h,m,xh,vh,x,v)
C
    implicit none
C
c Input/Output
    integer nbod,nbig
    real*8 jcen(3),h,m(nbod),xh(3,nbod),vh(3,nbod),x(3,nbod),v(3,nbod)
C
c Local
    integer j
    real*8 mtot, temp
C
C
    mtot = 0.d0
    x(1,1) = 0.d0
    x(2,1) = 0.d0
     x(3,1) = 0.d0
     v(1,1) = 0.d0
     v(2,1) = 0.d0
     v(3,1) = 0.d0
c Calculate coordinates and velocities of the central body
     do j = 2, nbod
      mtot = mtot + m(j)
      enddo
    temp = -1.d0 / (mtot + m(1))
 x(1,1) = temp * x(1,1)
     x(2,1) = temp * x(2,1)
     x(3,1) = temp * x(3,1)
     v(1,1) = temp * v(1,1)
     v(2,1) = temp * v(2,1)
     v(3,1) = temp * v(3,1)
c Calculate the barycentric coordinates and velocities
     do j = 2, nbod
      x(1,j) = xh(1,j) + x(1,1)
      x(2,j) = xh(2,j) + x(2,1)
      x(3,j) = xh(3,j) + x(3,1)
```

```
v(1,j) = vh(1,j) + v(1,1)
        v(2,j) = vh(2,j) + v(2,1)
        v(3,j) = vh(3,j) + v(3,1)
C
C
      return
C
C
C
       MCO H2CB.FOR
                     (ErikSoft 2 November 2000)
c Author: John E. Chambers
c Convert coordinates with respect to the central body to close-binary
c coordinates.
C-
С
      subroutine mco_h2cb (jcen,nbod,nbig,h,m,xh,vh,x,v)
C
      implicit none
C
c Input/Output
      integer nbod,nbig
      real*8 jcen(3), h, m(nbod), xh(3, nbod), vh(3, nbod), x(3, nbod), v(3, nbod)
C
c Local
      integer j
      real*8 msum, mvsum(3), temp, mbin, mbin_1, mtot_1
C
C-
C
      msum = 0.d0
      mvsum(1) = 0.d0
      mvsum(2) = 0.d0
      mvsum(3) = 0.d0
      mbin = m(1) + m(2)
      mbin_1 = 1.d0 / mbin
      x(1,2) = xh(1,2)
      x(2,2) = xh(2,2)
      x(3,2) = xh(3,2)
      temp = m(1) * mbin_1
      v(1,2) = temp * vh(1,2)
      v(2,2) = temp * vh(2,2)
      v(3,2) = temp * vh(3,2)
C
      do j = 3, nbod
        msum = msum + m(j)
        mvsum(1) = mvsum(1) + m(j) * vh(1,j)
        mvsum(2) = mvsum(2) + m(j) * vh(2,j)

mvsum(3) = mvsum(3) + m(j) * vh(3,j)
      end do
      mtot_1 = 1.d0 / (msum + mbin)
      mvsum(1) = mtot_1 * (mvsum(1) + m(2)*vh(1,2))
mvsum(2) = mtot_1 * (mvsum(2) + m(2)*vh(2,2))
      mvsum(3) = mtot_1 * (mvsum(3) + m(2)*vh(3,2))
C
      temp = m(2) * mbin_1
      do \bar{j} = 3, nbod
        x(1,j) = xh(1,j) - temp * xh(1,2)

x(2,j) = xh(2,j) - temp * xh(2,2)

x(3,j) = xh(3,j) - temp * xh(3,2)
        v(1,j) = vh(1,j) - mvsum(1)
        v(2,j) = vh(2,j) - mvsum(2)

v(3,j) = vh(3,j) - mvsum(3)
      end do
```

```
C-
C
      end
(ErikSoft 2 November 2000)
c Author: John E. Chambers
c Converts coordinates with respect to the central body to Jacobi coordinates.
c Note that the Jacobi coordinates of all small bodies are assumed to be the
c same as their coordinates with respect to the central body.
C
      subroutine mco_h2j (jcen,nbod,nbig,h,m,xh,vh,x,v)
C
      implicit none
С
c Input/Output
      integer nbod,nbig
      real*8 jcen(3), h, m(nbig), xh(3, nbig), vh(3, nbig), x(3, nbig), v(3, nbig)
c Local
      integer i
      real*8 mtot, mx, my, mz, mu, mv, mw, temp
C
      mtot = m(2)
      x(1,2) = xh(1,2)
      x(2,2) = xh(2,2)
      x(3,2) = xh(3,2)
      v(1,2) = vh(1,2)
      v(2,2) = vh(2,2)
      v(3,2) = vh(3,2)
      mx = m(2) * xh(1,2)
      my = m(2) * xh(2,2)
      mz = m(2) * xh(3,2)
      mu = m(2) * vh(1,2)
      mv = m(2) * vh(2,2)
      mw = m(2) * vh(3,2)
C
      do j = 3, nbig - 1
        temp = 1.d0 / (mtot + m(1))
        mtot = mtot + m(j)
        x(1,j) = xh(1,j) - temp * mx

x(2,j) = xh(2,j) - temp * my
                               temp * mx
        x(3,j) = xh(2,j) = temp * mz

x(3,j) = xh(3,j) - temp * mz

v(1,j) = vh(1,j) - temp * mu

v(2,j) = vh(2,j) - temp * mv
        v(3,j) = vh(3,j) - temp * mw
        mx = mx + m(j) * xh(1,j)

my = my + m(j) * xh(2,j)
        mz = mz + m(j) * xh(3,j)
        mu = mu + m(j) * vh(1,j)

mv = mv + m(j) * vh(2,j)
        mw = mw + m(j) * vh(3,j)
      enddo
      if (nbig.gt.2) then
        temp = 1.d0 / (mtot + m(1))
        x(1,nbig) = xh(1,nbig) - temp * mx

x(2,nbig) = xh(2,nbig) - temp * my
        x(3,nbig) = xh(3,nbig) -
        v(1,nbig) = vh(1,nbig) - temp * mu
v(2,nbig) = vh(2,nbig) - temp * mv
v(3,nbig) = vh(3,nbig) - temp * mw
```

```
end if
C
                 do j = nbig + 1, nbod
                       x(1,j) = xh(1,j)
                       x(2,j) = xh(2,j)
                       x(3,j) = xh(3,j)
                       v(1,j) = vh(1,j)
                       v(2,j) = vh(2,j)
                      v(3,j) = vh(3,j)
                 end do
C
C-
C
                  return
                 end
(ErikSoft 2 November 2000)
C
C
c Author: John E. Chambers
c Makes a new copy of a set of coordinates.
C
C
                  subroutine mco_iden (jcen,nbod,nbig,h,m,xh,vh,x,v)
C
                 implicit none
C
c Input/Output
                 integer nbod,nbig
                 real*8 jcen(3), h, m(nbod), x(3, nbod), v(3, nbod), xh(3, nbod), vh(3, nbod)
c Local
                 integer j
C
C-
C
                 do j = 1, nbod
                       x(1,j) = xh(1,j)
                       x(2,j) = xh(2,j)
                       x(3,j) = xh(3,j)
                       v(1,j) = vh(1,j)
                       v(2,j) = vh(2,j)
                       v(3,j) = vh(3,j)
C
C--
C
                 return
C
                    MCO X2EL.FOR
                                                                 (ErikSoft 20 February 2001)
C
\mathbb{C}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}^{\frac{5}{6}}
c Author: John E. Chambers
c Calculates Keplerian orbital elements given relative coordinates and
c velocities, and GM = G times the sum of the masses.
c The elements are: q = perihelion distance
                                                           e = eccentricity
C
                                                            i = inclination
                                                           p = longitude of perihelion (NOT argument of perihelion!!)
C
                                                           n = longitude of ascending node
C
                                                            1 = mean anomaly (or mean longitude if e < 1.e-8)</pre>
С
```

```
C
С
     subroutine mco_x2el (gm,x,y,z,u,v,w,q,e,i,p,n,l)
C
     implicit none
     include 'mercury.inc'
C
c Input/Output
     real*8 gm,q,e,i,p,n,l,x,y,z,u,v,w
c Local
     real*8 hx,hy,hz,h2,h,v2,r,rv,s,true
     real*8 ci,to,temp,tmp2,bige,f,cf,ce
C
C-----
C
     hx = y * w - z * v
     hy = \overline{z} * u - x * w
     hz = x * v - y * u
     h2 = hx*hx + hy*hy + hz*hz
     v2 = u * u + v * v + w
     rv = x * u + y * v + z * w
     r = sqrt(x*x + y*y + z*z)
     h = sqrt(h2)
     s = h2 / gm
C
c Inclination and node
     ci = hz / h
     if (abs(ci).lt.1) then
       i = acos (ci)
       n = atan2 (hx, -hy)
       if (n.lt.0) n = n + TWOPI
       if (ci.gt.0) i = 0.d0
       if (ci.lt.0) i = PI
       n = 0.d0
     end if
c Eccentricity and perihelion distance
      temp = 1.d0 + s * (v2 / gm - 2.d0 / r)
      if (temp.le.0) then
       e = 0.d0
      else
       e = sqrt (temp)
     end if
     q = s / (1.d0 + e)
c True longitude
     if (hy.ne.0) then
       to = -hx/hy
       temp = (1.d0 - ci) * to
       tmp2 = to * to
       true = atan2((y*(1.d0+tmp2*ci)-x*temp),(x*(tmp2+ci)-y*temp))
     else
       true = atan2(y * ci, x)
     end if
     if (ci.lt.0) true = true + PI
     if (e.lt.3.d-8) then
       p = 0.d0
       1 = true
     else
       ce = (v2*r - gm) / (e*gm)
c Mean anomaly for ellipse
       if (e.lt.1) then
         if (abs(ce).gt.1) ce = sign(1.d0,ce)
         bige = acos(ce)
         if (rv.lt.0) bige = TWOPI - bige
         l = bige - e*sin(bige)
       else
```

```
c Mean anomaly for hyperbola
         if (ce.lt.1) ce = 1.d0
         bige = log( ce + sqrt(ce*ce-1.d0) )
         if (rv.lt.0) bige = - bige
         l = e*sinh(bige) - bige
       end if
c Longitude of perihelion
       cf = (s - r) / (e*r)
       if (abs(cf).gt.1) cf = sign(1.d0,cf)
       f = acos(cf)
       if (rv.lt.0) f = TWOPI - f
       p = true - f
       p = mod (p + TWOPI + TWOPI, TWOPI)
     end if
C
     if (1.lt.0.and.e.lt.1) 1 = 1 + TWOPI
     if (l.gt.TWOPI.and.e.lt.1) l = mod (l, TWOPI)
C
C-
C
     return
MIO_JD_Y.FOR
                   (ErikSoft 2 June 1998)
C
c Author: John E. Chambers
c Converts from Julian day number to Julian/Gregorian Calendar dates, assuming
c the dates are those used by the English calendar.
c Algorithm taken from 'Practical Astronomy with your calculator' (1988)
c by Peter Duffett-Smith, 3rd edition, C.U.P.
c Algorithm for negative Julian day numbers (Julian calendar assumed) by
c J. E. Chambers.
c N.B. The output date is with respect to the Julian Calendar on or before
c === 4th October 1582, and with respect to the Gregorian Calendar on or
      after 15th October 1582.
C
C
C
C-
С
     subroutine mio_jd_y (jd0,year,month,day)
C
     implicit none
C
c Input/Output
     real*8 jd0,day
     integer year, month
C
c Local
     integer i,a,b,c,d,e,g
     real*8 jd,f,temp,x,y,z
C
C
C
     if (jd0.le.0) goto 50
C
     jd = jd0 + 0.5d0
     i = sign( dint(dabs(jd)), jd )
     f = jd - 1.d0*i
c If on or after 15th October 1582
     if (i.gt.2299160) then
```

temp = (1.d0\*i-1867216.25d0) / 36524.25d0

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```
a = sign( dint(dabs(temp)), temp )
       temp = .25d0 * a
       b = i + 1 + a - sign(dint(dabs(temp)), temp)
     else
      b = i
     end if
C
     c = b + 1524
     temp = (1.d0*c - 122.1d0) / 365.25d0
     d = sign( dint(dabs(temp)), temp )
     temp = 365.25d0 * d
     e = sign( dint(dabs(temp)), temp )
     temp = (c-e) / 30.6001d0
     g = sign( dint(dabs(temp)), temp )
C
     temp = 30.6001d0 * g
     day = 1.d0*(c-e) + f - 1.d0*sign(dint(dabs(temp)), temp)
C
     if (g.le.13) month = g - 1
     if (g.gt.13) month = g - 13
     if (month.gt.2) year = d - 4716
     if (month.le.2) year = d - 4715
С
     if (day.gt.32) then
       day = day - 32
       month = month + 1
     end if
C
     if (month.gt.12) then
       month = month - 12
       year = year + 1
     end if
     return
 50 continue
c Algorithm for negative Julian day numbers (Duffett-Smith won't work)
     x = jd0 - 2232101.5
     f = x - dint(x)
     if (f.lt.0) f = f + 1.d0
     y = dint(mod(x,1461.d0) + 1461.d0)
     z = dint(mod(y, 365.25d0))
     month = int((z + 0.5d0) / 30.61d0)
     day = dint(z + 1.5d0 - 30.61d0*dble(month)) + f
     month = mod(month + 2, 12) + 1
C
     year = 1399 + int (x / 365.25d0)
     if (x.lt.0) year = year - 1
     if (month.lt.3) year = year + 1
C
C-
C
     return
     end
(ErikSoft 14 November 1999)
c Author: John E. Chambers
c Given a character string STRING, of length LEN bytes, the routine finds
c the beginnings and ends of NSUB substrings present in the original, and
c delimited by spaces. The positions of the extremes of each substring are
c returned in the array DELIMIT.
c Substrings are those which are separated by spaces or the = symbol.
C
C-
C
```

```
subroutine mio_spl (len,string,nsub,delimit)
C
     implicit none
C
c Input/Output
     integer len,nsub,delimit(2,100)
     character*1 string(len)
C
     integer j,k
     character*1 c
C
C-----
C
     nsub = 0
     j = 0
     delimit(1,1) = -1
c Find the start of string
 10 \quad j = j + 1
     if (j.gt.len) goto 99
     c = string(j)
     if (c.eq.' '.or.c.eq.'=') goto 10
c Find the end of string
     k = j
  20 k = k + 1
     if (k.gt.len) goto 30
     c = string(k)
if (c.ne.' '.and.c.ne.'=') goto 20
c Store details for this string
  30 \quad \text{nsub} = \text{nsub} + 1
     delimit(1,nsub) = j
     delimit(2,nsub) = k - 1
C
     if (k.lt.len) then
      j = k
      goto 10
     end if
 99 continue
C-
C
     return
     end
*******************
                 ORBEL_FHYBRID.F
******************
     PURPOSE: Solves Kepler's eqn. for hyperbola using hybrid approach.
            Input:
                        e ==> eccentricity anomaly. (real scalar)
                        n ==> hyperbola mean anomaly. (real scalar)
            Returns:
             orbel_fhybrid ==> eccentric anomaly. (real scalar)
     ALGORITHM: For abs(N) < 0.636*ecc -0.6, use FLON
              For larger N, uses FGET
     REMARKS:
     AUTHOR: M. Duncan
     DATE WRITTEN: May 26,1992.
     REVISIONS:
     REVISIONS: 2/26/93 hfl
                   ***********
      real*8 function orbel_fhybrid(e,n)
     include 'swift.inc'
```

```
c... Inputs Only:
      real*8 e,n
c... Internals:
      real*8 abn
      real*8 orbel_flon, orbel_fget
c... Executable code
       abn = n
       if(n.lt.0.d0) abn = -abn
       if(abn .lt. 0.636d0*e -0.6d0) then
        orbel_fhybrid = orbel_flon(e,n)
        orbel_fhybrid = orbel_fget(e,n)
       endif
       return
       end ! orbel_fhybrid
********************
                  ORBEL_FGET.F
*******************
     PURPOSE: Solves Kepler's eqn. for hyperbola using hybrid approach.
            Input:
                        e ==> eccentricity anomaly. (real scalar)
                     capn ==> hyperbola mean anomaly. (real scalar)
            Returns:
                orbel_fget ==> eccentric anomaly. (real scalar)
     ALGORITHM: Based on pp. 70-72 of Fitzpatrick's book "Principles of
          Cel. Mech. ". Quartic convergence from Danby's book.
     REMARKS:
     AUTHOR: M. Duncan
     DATE WRITTEN: May 11, 1992.
     REVISIONS: 2/26/93 hfl
    Modified by JEC
**************
      real*8 function orbel_fget(e,capn)
     include 'swift.inc'
c... Inputs Only:
      real*8 e,capn
c... Internals:
       integer i,IMAX
       real*8 tmp,x,shx,chx
       real*8 esh,ech,f,fp,fpp,fppp,dx
       PARAMETER (IMAX = 10)
c... Executable code
c Function to solve "Kepler's eqn" for F (here called
c x) for given e and CAPN.
c begin with a guess proposed by Danby
       if( capn .lt. 0.d0) then
         tmp = -2.d0*capn/e + 1.8d0
         x = -\log(tmp)
         tmp = +2.d0*capn/e + 1.8d0
         x = log(tmp)
       endif
```

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```
orbel_fget = x
       do i = 1, IMAX
         call mco_sinh (x,shx,chx)
         esh = e*shx
         ech = e*chx
         f = esh - x - capn
         write(6,*) 'i,x,f : ',i,x,f
         fp = ech - 1.d0
         fpp = esh
         fppp = ech
         dx = -f/fp
         dx = -f/(fp + dx*fpp/2.d0)
         dx = -f/(fp + dx*fpp/2.d0 + dx*dx*fppp/6.d0)
         orbel_fget = x + dx
  If we have converged here there's no point in going on
         if(abs(dx) .le. TINY) RETURN
         x = orbel_fget
       write(6,*) 'FGET: RETURNING WITHOUT COMPLETE CONVERGENCE'
       return
       end ! orbel_fget
*******************
                  ORBEL_FLON.F
*****************
    PURPOSE: Solves Kepler's eqn. for hyperbola using hybrid approach.
            Input:
                         e ==> eccentricity anomaly. (real scalar)
                      capn ==> hyperbola mean anomaly. (real scalar)
                orbel_flon ==> eccentric anomaly. (real scalar)
     ALGORITHM: Uses power series for N in terms of F and Newton,s method
     REMARKS: ONLY GOOD FOR LOW VALUES OF N (N < 0.636*e -0.6)
     AUTHOR: M. Duncan
     DATE WRITTEN: May 26, 1992.
     REVISIONS:
*******************
      real*8 function orbel_flon(e,capn)
     include 'swift.inc'
c... Inputs Only:
      real*8 e,capn
c... Internals:
      integer iflag, i, IMAX
       real*8 a,b,sq,biga,bigb
       real*8 x, x2
       real*8 f,fp,dx
       real*8 diff
       real*8 a0,a1,a3,a5,a7,a9,a11
       real*8 b1,b3,b5,b7,b9,b11
       PARAMETER (IMAX = 10)
       PARAMETER (all = 156.d0, a9 = 17160.d0, a7 = 1235520.d0)
       PARAMETER (a5 = 51891840.d0,a3 = 1037836800.d0)
       PARAMETER (b11 = 11.d0*a11, b9 = 9.d0*a9, b7 = 7.d0*a7)
       PARAMETER (b5 = 5.d0*a5, b3 = 3.d0*a3)
c... Executable code
c Function to solve "Kepler's eqn" for F (here called
c x) for given e and CAPN. Only good for smallish CAPN
       iflag = 0
```

```
11/03/2015
```

```
if( capn .lt. 0.d0) then
          iflag = 1
          capn = -capn
       endif
       a1 = 6227020800.d0 * (1.d0 - 1.d0/e)
       a0 = -6227020800.d0*capn/e
       b1 = a1
c \, Set iflag nonzero if capn < 0., in which case solve for -capn \,
  and change the sign of the final answer for F.
c Begin with a reasonable guess based on solving the cubic for small F
       a = 6.d0*(e-1.d0)/e
       b = -6.d0*capn/e
       sq = sqrt(0.25*b*b + a*a*a/27.d0)
       x = biga + bigb

write(6,*) 'cubic = ', x**3 + a*x + b
       orbel_flon = x
c If capn is tiny (or zero) no need to go further than cubic even for
c = 1.
       if( capn .lt. TINY) go to 100
       do i = 1, IMAX
         x2 = x*x
         f = a0 +x*(a1+x2*(a3+x2*(a5+x2*(a7+x2*(a9+x2*(a11+x2))))))
         fp = b1 + x2*(b3+x2*(b5+x2*(b7+x2*(b9+x2*(b11 + 13.d0*x2)))))
         dx = -f/fp
         write(6,*) 'i,dx,x,f : '
C
         write(6,432) i,dx,x,f
C
432
         format(1x,i3,3(2x,1p1e22.15))
         orbel_flon = x + dx
   If we have converged here there's no point in going on
         if(abs(dx) .le. TINY) go to 100
         x = orbel_flon
       enddo
c Abnormal return here - we've gone thru the loop
c IMAX times without convergence
       if(iflag .eq. 1) then
          orbel_flon = -orbel_flon
          capn = -capn
       endif
       write(6,*) 'FLON : RETURNING WITHOUT COMPLETE CONVERGENCE'
         diff = e*sinh(orbel_flon) - orbel_flon - capn
         write(6,*) 'N, F, ecc*sinh(F) - F - N : '
         write(6,*) capn,orbel_flon,diff
       return
c Normal return here, but check if capn was originally negative
       if(iflag .eq. 1) then
          orbel_flon = -orbel_flon
          capn = -capn
       endif
       return
       end
              ! orbel_flon
********************
                  ORBEL_ZGET.F
***************
     PURPOSE: Solves the equivalent of Kepler's eqn. for a parabola
          given Q (Fitz. notation.)
            Input:
                         q ==> parabola mean anomaly. (real scalar)
                 orbel_zget ==> eccentric anomaly. (real scalar)
```

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```
ALGORITHM: p. 70-72 of Fitzpatrick's book "Princ. of Cel. Mech."
     REMARKS: For a parabola we can solve analytically.
      AUTHOR: M. Duncan
     DATE WRITTEN: May 11, 1992.
    REVISIONS: May 27 - corrected it for negative Q and use power
             series for small Q.
       real*8 function orbel_zget(q)
     include 'swift.inc'
c... Inputs Only:
       real*8 q
c... Internals:
       integer iflag
       real*8 x,tmp
c... Executable code
        iflag = 0
if(q.lt.0.d0) then
         iflag = 1
         q = -q
        endif
        if (q.lt.1.d-3) then
           orbel_zget = q*(1.d0 - (q*q/3.d0)*(1.d0 - q*q))
          x = 0.5d0*(3.d0*q + sqrt(9.d0*(q**2) +4.d0))
           tmp = x**(1.d0/3.d0)
          orbel_zget = tmp - 1.d0/tmp
        endif
        if(iflag .eq.1) then
          orbel_zget = -orbel_zget
           q = -q
        endif
       return
       end ! orbel_zget
```