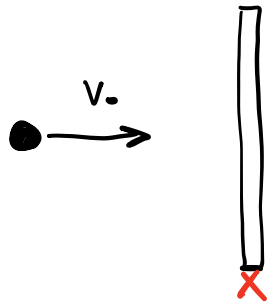


1)



a) Center of Mass

$$I_{xx} = I_{yy}$$

$$I_{xx} = \int y^2 + z^2 dm$$

$$= \rho \int_{-l/2}^{l/2} z^2 dz$$

$$= \rho \left[\frac{1}{3} z^3 \right]_{-l/2}^{l/2}$$

$$= \rho \frac{2}{3} \left(\frac{l}{2} \right)^3$$

$$= \frac{1}{12} m l^2$$



Rod symmetric
about x and y

$$\text{Let } \rho = \frac{dm}{dz}$$

$$I_{zz} = \int x^2 + y^2 dm = 0$$

Due to symmetry,

$$I_{xy} = I_{xz} = I_{yz} = 0$$

$$\vec{I}_{cm} = \begin{bmatrix} \frac{1}{12} m l^2 & 0 & 0 \\ 0 & \frac{1}{12} m l^2 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

Pivot

Using parallel axis theorem

$$I'_{xx} = I_{xx} + m\left(\frac{l}{2}\right)^2 = \frac{1}{3}ml^2$$

$$I'_{yy} = I_{yy} + m\left(\frac{l}{2}\right)^2 = \frac{1}{3}ml^2$$

$$I'_{zz} = 0$$

$$\vec{I}_{pivot} = \begin{bmatrix} \frac{1}{3}ml^2 & 0 & 0 \\ 0 & \frac{1}{3}ml^2 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

- b) Energy: collision is elastic
no external force applied over distance
 \Rightarrow conserved

Linear

Momentum: external force applied at pivot

\Rightarrow not conserved

Angular

Momentum: When taken about pivot, no external forces that cause moments.

\Rightarrow conserved about pivot.

c)

$$\text{Energy: } \frac{1}{2} m v_o^2 = \frac{1}{2} m v_f^2 + \frac{1}{2} \left(m \frac{l^2}{3} \right) \omega_f^2$$

$$\text{Ang. Momentum: } \frac{l}{2} m v_o = \frac{l}{2} m v_f + \left(m \frac{l^2}{3} \right) \omega_f$$

$$\Rightarrow v_o^2 = v_f^2 + \frac{l^2}{3} \omega_f^2$$

$$v_o = v_f + \frac{2}{3} l \omega_f \Rightarrow v_f = v_o - \frac{2}{3} l \omega_f$$

$$v_o^2 = \left(v_o - \frac{2}{3} l \omega_f \right)^2 + \frac{l^2}{3} \omega_f^2$$

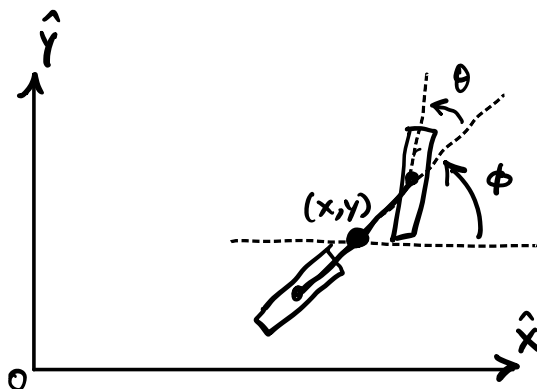
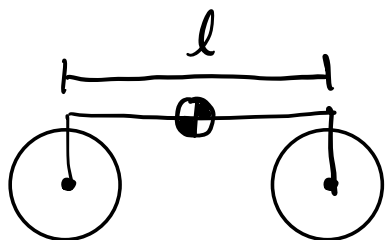
$$v_o^2 = v_o^2 - \frac{4}{3} l \omega_f v_o + \frac{4}{9} l^2 \omega_f^2 + \frac{l^2}{3} \omega_f^2$$

$$\Rightarrow \frac{7}{9} l^2 \omega_f^2 - \frac{4}{3} l v_o \omega_f = 0$$

$$\omega_f \left(\frac{7}{9} l \omega_f - \frac{4}{3} v_o \right) = 0$$

$$\Rightarrow \boxed{\omega_f = \frac{12}{7} \frac{v_o}{l}}$$

2)



a) Gen Coord: x, y, ϕ, θ

We can ignore the rotations of the wheels.

There are no holonomic constraints

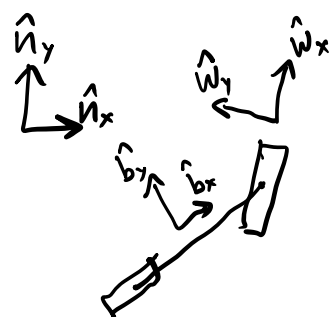
\Rightarrow 4 DOF

b) Define three frames:

Inertial frame N : \hat{n}_x, \hat{n}_y

Motorcycle frame B : \hat{b}_x, \hat{b}_y

Front wheel frame W : \hat{w}_x, \hat{w}_y



$$\vec{r}_{cm} = x \hat{n}_x + y \hat{n}_y$$

$$\vec{r}_{front} = x \hat{n}_x + y \hat{n}_y + \frac{l}{2} \hat{b}_x$$

$$\vec{r}_{rear} = x \hat{n}_x + y \hat{n}_y - \frac{l}{2} \hat{b}_x$$

\Rightarrow Take derivative for velocity.

$$\vec{V}_{\text{front}} = \dot{X} \hat{n}_x + \dot{Y} \hat{n}_y + \dot{\phi} \frac{l}{2} \hat{b}_y$$

$$\vec{V}_{\text{rear}} = \dot{X} \hat{n}_x + \dot{Y} \hat{n}_y - \dot{\phi} \frac{l}{2} \hat{b}_y$$

Constraints: $\vec{V}_{\text{front}} \cdot \hat{w}_y = 0$

$$\vec{V}_{\text{rear}} \cdot \hat{b}_y = 0$$

$$f_1 = -\dot{X} \sin(\phi + \theta) + \dot{Y} \cos(\phi + \theta) + \dot{\phi} \frac{l}{2} \cos(\theta) = 0$$

$$f_2 = -\dot{X} \sin(\phi) + \dot{Y} \cos(\phi) - \dot{\phi} \frac{l}{2} = 0$$

c)
$$T = \frac{1}{2} m \vec{V}_{\text{cm}} \cdot \vec{V}_{\text{cm}} + \frac{1}{2} I \dot{\phi}^2$$

$$= \frac{1}{2} m (\dot{X}^2 + \dot{Y}^2) + \frac{1}{2} I \dot{\phi}^2$$

$$V = 0$$

$$\mathcal{L} = T - V = \frac{1}{2} m (\dot{X}^2 + \dot{Y}^2) + \frac{1}{2} I \dot{\phi}^2$$

There are no non-conservative forces present $\Rightarrow \sum Q_j = 0$

$$x) \quad \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{x}} \right) - \frac{\partial L}{\partial x} + \mu_1 \frac{\partial f_1}{\partial \dot{x}} + \mu_2 \frac{\partial f_2}{\partial \dot{x}} = 0$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{x}} \right) = m \ddot{x} \quad \frac{\partial L}{\partial x} = 0$$

$$\frac{\partial f_1}{\partial \dot{x}} = -\sin(\phi + \theta) \quad \frac{\partial f_2}{\partial \dot{x}} = -\sin(\phi)$$

$$m \ddot{x} - \mu_1 \sin(\phi + \theta) - \mu_2 \sin(\phi) = 0$$

$$y) \quad \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{y}} \right) - \frac{\partial L}{\partial y} + \mu_1 \frac{\partial f_1}{\partial \dot{y}} + \mu_2 \frac{\partial f_2}{\partial \dot{y}} = 0$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{y}} \right) = m \ddot{y} \quad \frac{\partial L}{\partial y} = 0$$

$$\frac{\partial f_1}{\partial \dot{y}} = \cos(\phi + \theta) \quad \frac{\partial f_2}{\partial \dot{y}} = \cos(\phi)$$

$$m \ddot{y} + \mu_1 \cos(\phi + \theta) + \mu_2 \cos(\phi) = 0$$

$$\phi) \quad \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\phi}} \right) - \frac{\partial L}{\partial \phi} + \mu_1 \frac{\partial f_1}{\partial \dot{\phi}} + \mu_2 \frac{\partial f_2}{\partial \dot{\phi}} = 0$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\phi}} \right) = I \ddot{\phi} \quad \frac{\partial L}{\partial \phi} = 0$$

$$\frac{\partial f_1}{\partial \dot{\phi}} = \frac{l}{2} \cos \theta \quad \frac{\partial f_2}{\partial \dot{\phi}} = -\frac{l}{2}$$

$$I \ddot{\phi} + \mu_1 \frac{l}{2} \cos \theta - \mu_2 \frac{l}{2} = 0$$

$$\theta) \quad \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\theta}} \right) - \frac{\partial L}{\partial \theta} + \mu_1 \frac{\partial f_1}{\partial \dot{\theta}} + \mu_2 \frac{\partial f_2}{\partial \dot{\theta}} = 0$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\theta}} \right) = 0 \quad \frac{\partial L}{\partial \theta} = 0 \quad \frac{\partial f_1}{\partial \dot{\theta}} = 0 \quad \frac{\partial f_2}{\partial \dot{\theta}} = 0$$

\Rightarrow Degenerate equation (θ is a completely free input)
due to approximation (nothing to solve)

$$m \ddot{x} - \mu_1 \sin(\phi + \theta) - \mu_2 \sin(\phi) = 0$$

$$m \ddot{y} + \mu_1 \cos(\phi + \theta) + \mu_2 \cos(\phi) = 0$$

$$I \ddot{\phi} + \mu_1 \frac{l}{2} \cos \theta - \mu_2 \frac{l}{2} = 0$$

$$-\dot{x} \sin(\phi + \theta) + \dot{y} \cos(\phi + \theta) + \dot{\phi} \frac{l}{2} \cos(\theta) = 0$$

$$-\dot{x} \sin(\phi) + \dot{y} \cos(\phi) - \dot{\phi} \frac{l}{2} = 0$$

θ is by choice \Rightarrow no restrictions (not well-defined)

3)

ψ, θ, ϕ

$$\psi = 45^\circ$$

Spacecraft: 3, 2, 1

$$\theta = 30^\circ$$

Ground: 3, 1, 3

$$\phi = 45^\circ$$

a)

Rotation matrix mapping from Orbit frame to S/C frame

$$\Rightarrow \begin{bmatrix} \cos \phi & \sin \phi & 0 \\ -\sin \phi & \cos \phi & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \theta & \sin \theta \\ 0 & -\sin \theta & \cos \theta \end{bmatrix} \begin{bmatrix} \cos \psi & \sin \psi & 0 \\ -\sin \psi & \cos \psi & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$\Rightarrow \begin{bmatrix} \sqrt{2}/2 & \sqrt{2}/2 & 0 \\ -\sqrt{2}/2 & \sqrt{2}/2 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & \sqrt{3}/2 & 1/2 \\ 0 & -1/2 & \sqrt{3}/2 \end{bmatrix} \begin{bmatrix} \sqrt{2}/2 & \sqrt{2}/2 & 0 \\ -\sqrt{2}/2 & \sqrt{2}/2 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$\Rightarrow \begin{bmatrix} \sqrt{2}/2 & \sqrt{6}/4 & \sqrt{2}/4 \\ -\sqrt{2}/2 & \sqrt{6}/4 & \sqrt{2}/4 \\ 0 & -1/2 & \sqrt{3}/2 \end{bmatrix} \begin{bmatrix} \sqrt{2}/2 & \sqrt{2}/2 & 0 \\ -\sqrt{2}/2 & \sqrt{2}/2 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$\Rightarrow \begin{bmatrix} 1/2 - \sqrt{3}/4 & 1/2 + \sqrt{3}/4 & \sqrt{2}/4 \\ -1/2 - \sqrt{3}/4 & -1/2 + \sqrt{3}/4 & \sqrt{2}/4 \\ \sqrt{2}/4 & -\sqrt{2}/4 & \sqrt{3}/2 \end{bmatrix} \quad (\text{confirm in handout})$$

Take transpose for body \rightarrow orbit frame

$$\begin{bmatrix} \frac{1}{2} - \frac{\sqrt{3}}{4} & -\frac{1}{2} - \frac{\sqrt{3}}{4} & \frac{\sqrt{2}}{4} \\ \frac{1}{2} + \frac{\sqrt{3}}{4} & -\frac{1}{2} + \frac{\sqrt{3}}{4} & -\frac{\sqrt{2}}{4} \\ \frac{\sqrt{2}}{4} & \frac{\sqrt{2}}{4} & \frac{\sqrt{3}}{2} \end{bmatrix} \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} = \boxed{\begin{bmatrix} \frac{1}{2} - \frac{\sqrt{3}}{4} \\ \frac{1}{2} + \frac{\sqrt{3}}{4} \\ \frac{\sqrt{2}}{4} \end{bmatrix}}$$

b) space \rightarrow body

$$\begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos\phi & \sin\phi \\ 0 & -\sin\phi & \cos\phi \end{bmatrix} \begin{bmatrix} \cos\theta & 0 & -\sin\theta \\ 0 & 1 & 0 \\ \sin\theta & 0 & \cos\theta \end{bmatrix} \begin{bmatrix} \cos\psi & \sin\psi & 0 \\ -\sin\psi & \cos\psi & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$\begin{bmatrix} \frac{\sqrt{2}}{2}(\frac{\sqrt{3}}{2}) & \frac{\sqrt{2}}{2}(\frac{\sqrt{3}}{2}) & -\frac{1}{2} \\ -\frac{\sqrt{2}}{2}(\frac{\sqrt{2}}{2}) + \frac{\sqrt{2}}{2}(\frac{1}{2})(\frac{\sqrt{2}}{2}) & \frac{\sqrt{2}}{2}(\frac{\sqrt{2}}{2}) + \frac{\sqrt{2}}{2}(\frac{1}{2})(\frac{\sqrt{2}}{2}) & \frac{\sqrt{3}}{2}(\frac{\sqrt{2}}{2}) \\ \frac{\sqrt{2}}{2}(\frac{\sqrt{2}}{2}) + \frac{\sqrt{2}}{2}(\frac{1}{2})(\frac{\sqrt{2}}{2}) & -\frac{\sqrt{2}}{2}(\frac{\sqrt{2}}{2}) + \frac{\sqrt{2}}{2}(\frac{1}{2})(\frac{\sqrt{2}}{2}) & \frac{\sqrt{3}}{2}(\frac{\sqrt{2}}{2}) \end{bmatrix}$$

$$= \begin{bmatrix} \frac{\sqrt{6}}{4} & \frac{\sqrt{6}}{4} & -\frac{1}{2} \\ -\frac{1}{4} & \frac{3}{4} & \frac{\sqrt{6}}{4} \\ \frac{3}{4} & -\frac{1}{4} & \frac{\sqrt{6}}{4} \end{bmatrix}$$

\Rightarrow Take transpose for
body \rightarrow space

$$\begin{bmatrix} \frac{\sqrt{6}}{4} & -\frac{1}{4} & \frac{3}{4} \\ \frac{\sqrt{6}}{4} & \frac{3}{4} & -\frac{1}{4} \\ -\frac{1}{2} & \frac{\sqrt{6}}{4} & \frac{\sqrt{6}}{4} \end{bmatrix} \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} = \boxed{\begin{bmatrix} \frac{\sqrt{6}}{4} \\ \frac{\sqrt{6}}{4} \\ -\frac{1}{2} \end{bmatrix}}$$

c) $\vec{H} = \vec{I} \vec{\omega}$, but we need to match the frames.

Let ${}_B R_S$ be the rotation matrix mapping from Space to body frame. Then

$${}_B R_S \begin{bmatrix} 1 \\ 2 \\ 3 \end{bmatrix} = \begin{bmatrix} 4 & 0 & 0 \\ 0 & 3 & 0 \\ 0 & 0 & 3 \end{bmatrix} {}_B R_S \begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix} \quad \leftarrow \vec{\omega} \text{ in orbit frame coord.}$$

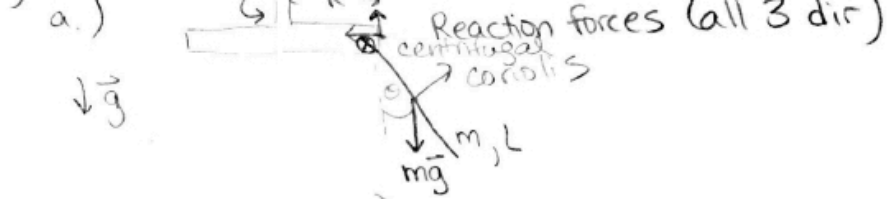
↖ inverse

$$\begin{aligned} {}_B R_S \begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix} &= \begin{bmatrix} 1/4 & 0 & 0 \\ 0 & 1/3 & 0 \\ 0 & 0 & 1/3 \end{bmatrix} \begin{bmatrix} \sqrt{6}/4 & \sqrt{6}/4 & -1/2 \\ -1/4 & 3/4 & \sqrt{6}/4 \\ 3/4 & -1/4 & \sqrt{6}/4 \end{bmatrix} \begin{bmatrix} 1 \\ 2 \\ 3 \end{bmatrix} \\ &= \begin{bmatrix} 1/4 & 0 & 0 \\ 0 & 1/3 & 0 \\ 0 & 0 & 1/3 \end{bmatrix} \begin{bmatrix} 3\sqrt{6}/4 - 3/2 \\ 5/4 + 3\sqrt{6}/4 \\ 1/4 + 3\sqrt{6}/4 \end{bmatrix} \\ &= \begin{bmatrix} \frac{3\sqrt{6}-6}{16} \\ \frac{3\sqrt{6}+5}{12} \\ \frac{3\sqrt{6}+1}{12} \end{bmatrix} \quad \leftarrow \vec{\omega} \text{ in body frame coordinates.} \end{aligned}$$

$$\begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix} = {}_B R_S^T \begin{bmatrix} (3\sqrt{6}-6)/16 \\ (3\sqrt{6}+5)/12 \\ (3\sqrt{6}+1)/12 \end{bmatrix}$$

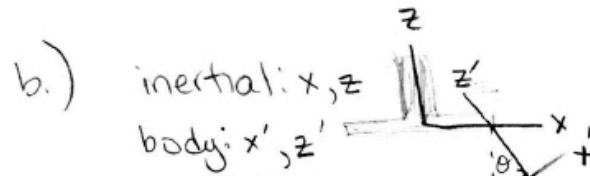
$$= \begin{bmatrix} \sqrt{6}/4 & -1/4 & 3/4 \\ \sqrt{6}/4 & 3/4 & -1/4 \\ -1/2 & \sqrt{6}/4 & \sqrt{6}/4 \end{bmatrix} \begin{bmatrix} (3\sqrt{6}-6)/16 \\ (3\sqrt{6}+5)/12 \\ (3\sqrt{6}+1)/12 \end{bmatrix}$$

From here on out this becomes painful,
So we will accept this as a solution.



$$F_c = \vec{\omega} \times \vec{\omega} \times \vec{r}$$

$$\vec{F}_c = 2(\vec{\omega} \times \vec{v}_r)$$



gen. coord: θ and ψ

$$\vec{\omega} = \dot{\psi} \hat{z} - \dot{\theta} \hat{y} \quad \text{and} \quad \hat{z} = -\sin\theta \hat{x}' + \cos\theta \hat{z}'$$

$$\vec{\omega} = -\dot{\psi} \sin\theta \hat{x}' - \dot{\theta} \hat{y}' + \dot{\psi} \cos\theta \hat{z}'$$

pivot: $\vec{v}_p = R \dot{\psi} \hat{y}'$

cm: $\vec{v}_{cm} = R \dot{\psi} \hat{y}' + [-\dot{\psi} \sin\theta \hat{x}' - \dot{\theta} \hat{y}' + \dot{\psi} \cos\theta \hat{z}'] \times \left(\frac{L}{2} \hat{z}'\right)$

$$\vec{v}_{cm} = \frac{L}{2} \dot{\theta} \hat{x}' + \dot{\psi} \left(R - \frac{L}{2} \sin\theta\right) \hat{y}'$$

$$\vec{H}_{cm} = I_{cm} (-\dot{\psi} \sin\theta \hat{x}' - \dot{\theta} \hat{y}')$$

Scanned with CamScanner

$$T = \frac{1}{2} m \vec{v}_{cm} \cdot \vec{v}_{cm} + \frac{1}{2} \vec{\omega} \cdot \vec{H}_{cm} + \frac{1}{2} I_1 \dot{\psi}^2$$

$$T = \frac{1}{2} m \left[\left(\frac{L}{2} \dot{\theta}\right)^2 + \dot{\psi}^2 \left(R + \frac{L}{2} \sin\theta\right)^2 \right] + \frac{1}{2} I_1 \dot{\psi}^2 + \frac{1}{2} I_{cm} (\dot{\psi}^2 \sin^2\theta + \dot{\theta}^2)$$

$$V = -mg \frac{L}{2} \cos\theta \quad \text{and} \quad \mathcal{L} = T - V$$

c.) $\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\psi}} - \frac{\partial \mathcal{L}}{\partial \psi} = 0$

$$\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\theta}} - \frac{\partial \mathcal{L}}{\partial \theta} = 0$$

5)

a) Measure \vec{I} for $\hat{x}, \hat{y}, \hat{z}$ about Center of mass ^{"body-fixed"}

$$I_{ij} = \sum_k m_k (\delta_{ij} \vec{r}_k \cdot \vec{r}_k - (\vec{r}_k \cdot \vec{e}_i)(\vec{r}_k \cdot \vec{e}_j))$$

$$\begin{aligned} \Rightarrow I_{xx} &= m_1(0^2 + b^2) + m_2(0^2 + (-b)^2) \\ &= 2mb^2 \end{aligned}$$

$$I_{yy} = 2mb^2$$

$$I_{zz} = 0$$

$$I_{xy} = I_{xz} = I_{yz} = 0$$

$$\vec{I} = \begin{bmatrix} 2mb^2 & 0 & 0 \\ 0 & 2mb^2 & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad \text{for } \hat{x}, \hat{y}, \hat{z} \text{ about center of mass}$$

$$\begin{aligned} \vec{\omega} &= \omega [\cos \alpha \hat{z} + \sin \alpha \hat{y}] \\ &= \omega \sin \alpha \hat{y} + \omega \cos \alpha \hat{z} \Rightarrow \end{aligned}$$

$$\omega_x = 0$$

$$\omega_y = \omega \sin \alpha$$

$$\omega_z = \omega \cos \alpha$$

c)

$$\vec{H} = \vec{I} \vec{\omega} = 2mb^2 \omega \sin \alpha \hat{y}$$

$$H_x = 0$$

$$H_y = 2mb^2 \omega \sin \alpha$$

$$H_z = 0$$

$$\vec{M} = \cancel{\vec{I} \dot{\vec{\omega}}} + \vec{\omega} \times \vec{I} \vec{\omega} \quad (\text{Euler's EOM})$$

$$= \begin{bmatrix} 0 \\ \omega \sin \alpha \\ \omega \cos \alpha \end{bmatrix} \times \begin{bmatrix} 2mb^2 & 0 & 0 \\ 0 & 2mb^2 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} 0 \\ \omega \sin \alpha \\ \omega \cos \alpha \end{bmatrix}$$

$$= \det \begin{bmatrix} \hat{x} & \hat{y} & \hat{z} \\ 0 & \omega \sin \alpha & \omega \cos \alpha \\ 0 & 2mb^2 \omega \sin \alpha & 0 \end{bmatrix}$$

$$= -2mb^2 \omega^2 \sin \alpha \cos \alpha \hat{x}$$

$$M_x = -2mb^2 \omega^2 \sin \alpha \cos \alpha$$

$$M_y = 0$$

$$M_z = 0$$