

Chapter 9: The Geometry Outside of a Spherical Star

3-27-19

Schwarzschild Geometry

The line element is

$$\left[ds^2 = -\left(1 - \frac{2GM}{c^2r}\right)c^2dt^2 + \frac{dr^2}{\left(1 - \frac{2GM}{c^2r}\right)} + r^2(d\theta^2 + \sin^2\theta d\phi^2) \right]$$

- These coordinates are called the Schwarzschild Coordinates
- $g_{\alpha\beta}(x)$ is called the Schwarzschild Metric

Notes:

1. Time independent : $t \rightarrow t + \text{const.}$ Leaves ds^2 invariant

\Rightarrow Associated killing vector ξ^α where $\xi^\alpha = (1, 0, 0, \varphi)$

2. Spherically Symmetric for $t = \text{const.}$, $r = \text{const.}$

$$d\Sigma^2 = r^2(d\theta^2 + \sin^2\theta d\phi^2)$$

- Geometry of a sphere in flat 3D Space

3. $\varphi \rightarrow \varphi + \text{const.}$ Leaves ds^2 invariant

\Rightarrow Associated killing vector η^α w/

$$\eta^\alpha = (0, 0, 0, 1)$$

Note: r is not the distance from any center

$$A = 4\pi r^2 : r = \left(\frac{A}{4\pi}\right)^{\frac{1}{2}} \text{ where } A \text{ is the area of the 2-sphere}$$

4. Mass M

$$\text{when } \frac{2GM}{c^2} \ll r : ds^2 = -\left(1 - \frac{2GM}{c^2r}\right)c^2dt^2 + \left(1 + \frac{2GM}{c^2r}\right) + r^2(d\theta^2 + \sin^2\theta d\phi^2)$$

\Rightarrow Weak, Static metric w/ a Newtonian Gravitational Potential

$$\left[\Phi = -\frac{GM}{r} \right] \text{ (See 6.6)}$$

$\Rightarrow M$ is the total mass of the source of curvature

5. Schwarzschild Radius

- $r = 0 \Rightarrow$ Real physical Singularity
- $r = 2GM/c^2 \Rightarrow$ Schwarzschild radius : Not real

4-1-19

The Schwarzschild Line element....

$$ds^2 = - \left(1 - \frac{2GM}{c^2r}\right) c^2 dt^2 + \left(1 - \frac{2GM}{c^2r}\right)^{-1} dr^2 + r^2 (d\theta^2 + \sin^2\theta d\phi^2)$$

Killing vectors

1. ξ^α w/ $\xi^\alpha = (1, 0, 0, 0)$

2. η^α w/ $\eta^\alpha = (0, 0, 0, 1)$

Schwarzschild Radius

- $r=0 \Rightarrow$ Physical Singularity
- $r = \frac{2GM}{c^2}$ \Rightarrow Coordinate Singularity (Schwarzschild radius)
 - \Rightarrow Characteristic length scale for the curvature @ $r = \frac{2GM}{c^2}$
 - \hookrightarrow Surface joins different geometries

Define Geometrized Units

- A system of units convenient for GR, which also puts $G=1$ ($c=1$)

$$M(\text{in cm}) = M(\text{in g}) \cdot \frac{G}{c^2} = 0.742 \times 10^{-28} \frac{\text{cm}}{\text{g}} \times M(\text{in g})$$

i.e $M_\odot = 1.47 \text{ km}$

$M_\odot = 0.44 \text{ cm}$

In Geometrized units....

$$ds^2 = -(1 - 2M/r) dt^2 + (dr^2) / (1 - 2M/r) + r^2 (d\theta^2 + \sin^2\theta d\phi^2)$$

$$g_{\alpha\beta} = \begin{pmatrix} -(1 - 2M/r) & & & \\ & (1 - 2M/r)^{-1} & & \\ & & (r^2) & \\ & & & (r^2 \sin^2\theta) \end{pmatrix} \rightarrow \text{Zero's elsewhere}$$

The Gravitational Redshift

Observer @ $r=R$, emit signal ω^* (measured by the stationary observer)

Observer @ $r=\infty$, Absorb ω_∞ : $\omega_\infty < \omega^*$

Conserved Quantity is

$\xi \cdot \underline{P}$ where \underline{P} is the photon's four-momentum

The energy of a photon measured by an observer w/ Four-Velocity \tilde{u}_{obs}

$$E_\gamma = \hbar\omega = -\tilde{P} \cdot \tilde{u}_{\text{obs}}$$

Now

$$\tilde{u}_{\text{obs}}^i = 0$$

so

$$\tilde{u}_{\text{obs}} \cdot \tilde{u}_{\text{obs}} = -1 = g_{\alpha\beta} u^\alpha u^\beta = g_{tt} (u^t_{\text{obs}})^2 = -1 (1 - \frac{2M}{r}) (u^t_{\text{obs}})^2$$

$$\therefore u^t_{\text{obs}} = \frac{1}{\sqrt{1 - \frac{2M}{r}}} \quad \therefore u^\alpha_{\text{obs}} = \left(\frac{1}{\sqrt{1 - \frac{2M}{r}}}, 0, 0, 0 \right) = (1 - \frac{2M}{r})^{-\frac{1}{2}} \xi^\alpha$$

$$\tilde{u}_{\text{obs}}(r) = \frac{\xi}{\sqrt{1 - \frac{2M}{r}}}$$

@ $r=R$

$$\hbar\omega_* = \frac{1}{\sqrt{1 - \frac{2M}{R}}} (-\tilde{P} \cdot \xi)_R$$

@ $r=\infty$

$$\hbar\omega_\infty = (-\tilde{P} \cdot \xi)_\infty$$

However, $(\xi \cdot \tilde{P})_\infty = (\xi \cdot \tilde{P})_R$ Since $\xi \cdot \tilde{P}$ is conserved along the photon's geodesic

$$\therefore \omega_* = \frac{\omega_\infty}{\sqrt{1 - \frac{2M}{R}}} \Rightarrow \left[\omega_\infty = \sqrt{1 - \frac{2M}{r}} \omega_* \right] \rightarrow \text{Gravitational redshift}$$

Particle Orbits - Precession of the Perihelion

Since ds^2 is independent of $t : \phi$

$$-e \equiv \xi \cdot \tilde{u} \quad \text{for } \xi \text{ w/ } \xi^\alpha = (1, 0, 0, 0)$$

$$l \equiv \tilde{u} \cdot \tilde{u} \quad \text{for } \tilde{u} \text{ w/ } \tilde{u}^\alpha = (0, 0, 0, 1)$$

\tilde{u} is the four-velocity of the particle

$$c = -g_{\alpha\beta} \xi^\alpha u^\beta = -g_{tt} \xi^t u^t = (1 - \frac{2M}{r}) \frac{dt}{d\tau}$$

$$l = g_{\alpha\beta} u^\alpha u^\beta = g_{\varphi\varphi} u^\varphi u^\varphi = r^2 \sin^2 \theta \frac{d\varphi}{d\tau} \Rightarrow \text{Particle lies in a plane}$$

• orientate axes such that $[\theta = \frac{\pi}{2} : u^\theta = \frac{du^\theta}{d\tau} = 0]$

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Another "first integral" comes from

$$-1 = \tilde{u} \cdot \tilde{u} = g_{\alpha\beta} u^\alpha u^\beta = -\left(1 - \frac{2M}{r}\right) (u^r)^2 + \frac{1}{\left(1 - \frac{2M}{r}\right)} (u^\theta)^2 + r^2 (u^\phi)^2 + r^2 \sin^2 \theta (u^\varphi)^2$$

$$-1 = -\left(1 - \frac{2M}{r}\right) \left(\frac{dt}{d\tau}\right)^2 + \left(1 - \frac{2M}{r}\right)^{-1} \left(\frac{dr}{d\tau}\right)^2 + r^2 \left(\frac{d\phi}{d\tau}\right)^2$$

$$\therefore \frac{dt}{d\tau} = e^{\left(1 - \frac{2M}{r}\right)^{-1}}, \quad \frac{d\phi}{d\tau} = \frac{l}{r^2}$$

Plugging into (*) yields

$$-1 = -\left(1 - \frac{2M}{r}\right)^{-1} e^2 + \left(1 - \frac{2M}{r}\right)^{-1} \left(\frac{dr}{d\tau}\right)^2 + \left(\frac{l}{r}\right)^2$$

$$-\left(1 + \frac{l^2}{r^2}\right)\left(1 - \frac{2M}{r}\right) = -e^2 + \left(\frac{dr}{d\tau}\right)^2 : \quad e^2 = \left(\frac{dr}{d\tau}\right)^2 + \left(1 + \frac{l^2}{r^2}\right)\left(1 - \frac{2M}{r}\right)$$

Defining a new constant

$$E = \left(\frac{1}{2}\right)(e^2 - 1) = \left(\frac{1}{2}\right)\left(\frac{dr}{d\tau}\right)^2 + \left(\frac{1}{2}\right)\left[\left(1 + \frac{l^2}{r^2}\right)\left(1 - \frac{2M}{r}\right) - 1\right]$$

$$E = \left(\frac{1}{2}\right)\left(\frac{dr}{d\tau}\right)^2 + \left(\frac{1}{2}\right)\left[1 - \frac{2M}{r} + \frac{l^2}{r^2} - \frac{2Ml^2}{r^3} - 1\right] \xrightarrow{\text{Newtonian}}$$

$$E = \left(\frac{1}{2}\right)\left(\frac{dr}{d\tau}\right)^2 + \underbrace{\left[-\frac{M}{r} + \frac{l^2}{2r^2} - \frac{Ml^2}{r^3}\right]}_{V_{\text{eff}}(r)} \xrightarrow{\text{GR correction term}}$$

In Newtonian Gravitation

- $E > 0$, unbounded particle (Parabolas, Hyperbolas)
- $E < 0$, bounded particles (Ellipses, circle)

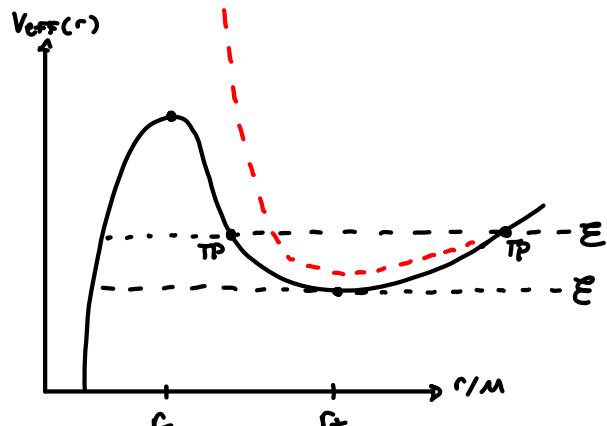
Put back C : G's

Let $t: \tau \rightarrow ct : C\tau$

$$M \rightarrow \frac{6M}{C^2}$$

$$l \rightarrow \frac{l}{C}$$

$$V_{\text{eff}}(r) = \frac{1}{c^2} \left[-\frac{6M}{r} + \frac{l^2}{2r^2} - \frac{6Ml^2}{C^2 r^3} \right]$$



Equilibrium points?

$$\frac{dV_{\text{eff}}}{dr} = 0 \rightarrow V_{\text{eff}} = -\frac{M}{r} + \frac{l^2}{2r^2} - \frac{Ml^2}{r^3}$$

$$\frac{dV_{\text{eff}}}{dr} = \frac{M}{r^2} - \frac{l^2}{r^3} + \frac{3Ml^2}{r^4} = 0 \quad \frac{1}{r^4} \left[Mr^2 - l^2r + 3Ml^2 \right] = 0$$

$$r_{\pm} = \frac{l^2 \pm \sqrt{l^4 - 4M(3Ml^2)}}{2(M)} = \frac{l^2}{2M} \left[1 \pm \sqrt{1 - \frac{12M^2}{l^2}} \right]$$

Notice:

- If $1 - 12M^2/l^2 < 0 \Rightarrow \frac{l}{M} < \sqrt{12} = 3.46$

\Rightarrow There are no real extrema: $V_{\text{eff}} < 0$ for all r

- If $1 - 12M^2/l^2 > 0 \Rightarrow \frac{l}{M} > \sqrt{12}$

$\Rightarrow 1 \text{ max : } 1 \text{ min}$

$$r_{\text{min}} = \frac{l^2}{2M} \left[1 \pm \sqrt{1 - \frac{12M^2}{l^2}} \right] \quad \left[E = \frac{1}{2} \left(\frac{dr}{dT} \right)^2 + V_{\text{eff}}(r) \right]^{(*)}$$

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Turning points? $\frac{dr}{dT} = 0$

• When $l/M < \sqrt{12}$, there are no r_{tp} for $E > 0$

Radial Plunge Orbit

$$l=0, \text{ start particle at } r \rightarrow \infty \text{ w/ } \left. \frac{dr}{dT} \right|_{r \rightarrow \infty} = 0 \quad \therefore E=0, e=1$$

$$e = (1 - 2M/r)(dt/dr)$$

(*) Becomes

$$0 = \frac{1}{2} \left(\frac{dr}{dT} \right)^2 - M/r \quad \therefore \frac{dr}{dT} = -\sqrt{\frac{2M}{r}} \quad \text{negative for radially infalling trajectory}$$

$$= \frac{2}{3} r^{\frac{3}{2}} = -\sqrt{2M} (\tau - \tau_*) : \tau_* \rightarrow \text{Arbitrary integration constant}$$

$$\left[r(\tau) = \left(\frac{3}{2} \right)^{\frac{2}{3}} (2M)^{\frac{1}{3}} (\tau_* - \tau) \right] \longrightarrow \begin{aligned} \tau_* \text{ such that} \dots \\ \tau \rightarrow -\infty, r \rightarrow \infty \\ \tau = \tau_* \therefore r = 0 \end{aligned}$$

$$e = 1 = (1 - 2M/r) dt/dr \quad \therefore \frac{dt}{dr} = (1 - 2M/r)^{-1}$$

$$\frac{dr}{dT} = \frac{dr}{dt} \frac{dt}{dT}$$

$$\left[r = 2Mx \right] \quad \text{Homework hint}$$

So

$$-\sqrt{\frac{2m}{r}} = \frac{dr}{dt} \left(1 - \frac{2m}{r}\right)^{-1} \quad \therefore \quad \frac{dr}{dt} = -\sqrt{\frac{2m}{r}} \left(1 - \frac{2m}{r}\right) : \int dt = -\int \left(\frac{2m}{r}\right)^{\frac{1}{2}} \left(1 - \frac{2m}{r}\right)^{-1} dr$$

Integrating Yields

$$t = t_* + 2M \left[-\frac{2}{3} \left(\frac{r}{2m}\right)^{\frac{3}{2}} - 2 \left(\frac{r}{2m}\right)^{\frac{1}{2}} + \ln \left(\frac{\left(\frac{r}{2m}\right)^{\frac{1}{2}} + 1}{\left(\frac{r}{2m}\right)^{\frac{1}{2}} - 1} \right) \right]$$

Notice : $T(r=2m)$: Finite

$$t(r=2m) = t_* + 2M \left[-\frac{2}{3} - 2 + \ln(2) - \ln(0) \right] \xrightarrow{1 \rightarrow \infty}$$

Stable Circular Orbits

$$r_{\min} = \frac{\ell^2}{2m} \left[1 + \sqrt{1 - 12 \left(\frac{m^2}{\ell^2}\right)} \right]$$

as (ℓ/m) gets smaller, r_{\min} also gets smaller

$$1 - 12 \left(\frac{m^2}{\ell^2}\right)^2 = 0 \quad \therefore \quad \ell/m = \sqrt{12} \quad \longrightarrow \quad r_{\min} \quad \left| \begin{array}{l} \frac{\ell}{m} = \sqrt{12} \\ \frac{\ell^2}{m^2} = 12 \end{array} \right. = \frac{\ell^2}{2m^2} m = \frac{1}{2} \cdot 12m = 6m^2$$

△ → Innermost Stable Circular Orbit (ISCO)

$$\left[V_{\text{ISCO}} = 6M \right]$$

The angular velocity of a particle in a circular orbit, Ω ,W.R.T. The Schwarzschild coordinate time, t , is

$$\Omega \equiv \frac{d\phi}{dt} = \frac{d\phi}{dT} \frac{dT}{dt} = \frac{d\phi/dT}{dt/dT} = \frac{\ell}{r^2} \cdot \frac{(1 - 2m/r)}{e} = \frac{1}{r^2} \left(1 - \frac{2m}{r}\right) \frac{\ell}{e}$$

Circular orbits of radius r_{\min} have $\ell:e$ determined by

$$1. V_{\text{eff}}(r_{\min}) = E$$

$$2. r_{\min} = \frac{\ell^2}{2m} \left[1 + \sqrt{1 - \frac{12m^2}{\ell^2}} \right]$$

One finds

$$\left[\Omega^2 = \frac{m}{r^3} \right] - \text{Angular Velocity in stable circular orbit} \Rightarrow \text{Nonrelativistic Kepler's 3rd Law}$$

$$ds^2 = - \left(1 - \frac{2m}{r}\right) dt^2 + \frac{dr^2}{\left(1 - \frac{2m}{r}\right)} + r^2(d\theta^2 + \sin^2\theta d\varphi^2)$$

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Light Ray Orbits

$$u^\alpha \equiv \frac{dx^\alpha}{d\lambda} \quad \text{Affine parameter}$$

Symmetry in $t : \varphi$ Directions \Rightarrow Killing vector $\xi : \eta$ w/ Conserved quantities

- Two "first integrals" . . .

$$c \equiv -\xi \cdot u = (1-2m/r) dt/d\lambda$$

$$l = \eta \cdot u = r^2 \sin^2 \theta \frac{d\varphi}{d\lambda}$$

- Third first integral from requirement that four velocity vector is null

$$u^\alpha \cdot u_\alpha = 0 = g_{\alpha\beta} u^\alpha u^\beta = g_{\alpha\beta} \frac{dx^\alpha}{d\lambda} \frac{dx^\beta}{d\lambda} = g_{tt} \left(\frac{dt}{d\lambda} \right)^2 + g_{rr} \left(\frac{dr}{d\lambda} \right)^2 + g_{\theta\theta} \left(\frac{d\theta}{d\lambda} \right)^2 + g_{\varphi\varphi} \left(\frac{d\varphi}{d\lambda} \right)^2$$

$$0 = -(1-2m/r) \left(\frac{dt}{d\lambda} \right)^2 + (1-2m/r)^{-1} \left(\frac{dr}{d\lambda} \right)^2 + r^2 \left(\frac{d\theta}{d\lambda} \right)^2 + r^2 \sin^2 \theta \left(\frac{d\varphi}{d\lambda} \right)^2 \quad \theta = \pi/2$$

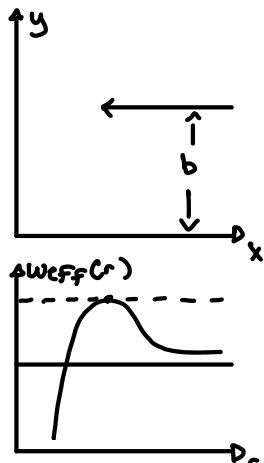
$$= -(1-2m/r) \left(\frac{dt}{d\lambda} \right)^2 + (1-2m/r)^{-1} \left(\frac{dr}{d\lambda} \right)^2 + r^2 \left(\frac{d\varphi}{d\lambda} \right)^2 \quad : \quad \frac{dt}{d\lambda} = e(1-2m/r)^{-1}$$

$$\frac{d\varphi}{d\lambda} = \frac{l}{r^2}$$

$$\left[0 = -e^2 (1-2m/r)^{-1} + (1-2m/r)^{-1} \left(\frac{dr}{d\lambda} \right)^2 + \frac{l^2}{r^2} \right] (1-2m/r)$$

$$\left[e^2 = \left(\frac{dr}{d\lambda} \right)^2 + \frac{l^2}{r^2} (1-2m/r) \right] \frac{1}{g^2} \quad \frac{e^2}{l^2} = \frac{1}{l^2} \left(\frac{dr}{d\lambda} \right)^2 + W_{\text{eff}}(r) \quad : \quad W_{\text{eff}}(r) = \frac{1}{r^2} (1-2m/r) \quad \therefore b = \left| \frac{e}{l} \right|$$

b is the impact parameter of a light ray that reaches infinity



$$W_{\text{eff}}(r) = \frac{1}{r^2} \left(1 - \frac{2m}{r} \right)$$

$$\text{where } x = r \cos \varphi \quad y = r \sin \varphi$$

$$\frac{e^2}{l^2} = \frac{1}{b^2}$$

Notice: $\lim_{r \rightarrow \infty} W_{\text{eff}}(r) \rightarrow 0$

$$\left. \frac{dW_{\text{eff}}}{dr} \right|_{r=r_1} = \frac{d}{dr} \left(\frac{1}{r^2} - \frac{2m}{r^3} \right)$$

$$\frac{dw_{\text{eff}}}{dr} = -\frac{2}{r^3} + \frac{6M}{r^4} = \frac{2}{r^4} (-r' + 3M) : [r' = 3M]$$

$$w_{\text{eff}}(r=3M) = \frac{1}{(3M)^2} \left(1 - \frac{2M}{3M} \right) = \frac{1}{27M^2}$$

Circular orbit? yes! $r = 3M$

$$\left[\frac{1}{b^2} = \frac{1}{27M^2} \right]$$

For our Sun

$$r' = 3M_\odot \approx 4.54 \text{ km} \ll R_\odot$$