

Problem 2)

$$\theta(r) = \int \frac{(l/r^2) dr}{\sqrt{2\mu(E + \frac{k}{r} - \frac{l^2}{2\mu r^2})}} - \frac{\pi}{2} \quad \alpha = \frac{l^2}{2\mu}, u = \frac{1}{r} : \quad \frac{du}{dr} = -\frac{1}{r^2} \quad \therefore dr = -r^2 du$$

$$\theta(r) = \int \frac{l \cdot u^2}{\sqrt{2\mu(E + k \cdot u - \alpha \cdot u^2)}} dr - \frac{\pi}{2} \rightarrow \int \frac{l \cdot u^2}{\sqrt{2\mu(E + k \cdot u - \alpha \cdot u^2)}} \cdot r^2 du - \frac{\pi}{2} = \frac{l}{\sqrt{2\mu}} \int \frac{du}{\sqrt{-\alpha u^2 + k u + E}} - \frac{\pi}{2}$$

$$\theta(r) = \frac{l^2}{2\mu} \int \frac{1}{\sqrt{-\alpha u^2 + k u + E}} du - \frac{\pi}{2} : \theta(r) = \int \frac{1}{\sqrt{-u^2 + \frac{k}{\alpha} u + \frac{E}{\alpha}}} du - \frac{\pi}{2} \quad a = -1, b = \frac{k}{\alpha}, c = \frac{E}{\alpha}$$

using appendix E): $\int \frac{dx}{\sqrt{ax^2 + bx + c}} = -\frac{1}{\sqrt{-a}} \sin^{-1}\left(\frac{2ax + b}{\sqrt{b^2 - 4ac}}\right)$

$$\theta(r) = -\frac{1}{\sqrt{-1}} \sin^{-1}\left(\frac{-2u + \frac{k}{\alpha}}{\sqrt{(\frac{k}{\alpha})^2 + 4(\frac{E}{\alpha})}}\right) - \frac{\pi}{2} : -\theta(r) - \frac{\pi}{2} = \sin^{-1}\left(\frac{\frac{k}{\alpha} - 2u}{\sqrt{(\frac{k}{\alpha})^2 + 4(\frac{E}{\alpha})}}\right)$$

$$\sin(-\theta(r) - \frac{\pi}{2}) = \frac{\frac{k}{\alpha} - 2u}{\sqrt{(\frac{k}{\alpha})^2 + 4(\frac{E}{\alpha})}} : -\cos(\theta(r)) = \frac{\frac{k}{\alpha} - 2u}{\sqrt{(\frac{k}{\alpha})^2 + 4(\frac{E}{\alpha})}}$$

↓

$$-\sin(\theta(r) + \frac{\pi}{2}) = -\cos(\theta(r))$$

$$\cos(\theta(r)) = \frac{2u - \frac{k}{\alpha}}{\sqrt{(\frac{k}{\alpha})^2 + 4(\frac{E}{\alpha})}} : \frac{\frac{2}{r} - \frac{k \cdot 2\mu}{l^2} \left(\frac{l^2}{2\mu k}\right)}{\sqrt{\frac{k^2 \cdot 4\mu^2}{l^4} + \frac{4E \cdot 2\mu}{l^2} \left(\frac{l^2}{4\mu k}\right)}} \quad u = \frac{1}{r} : \alpha = \frac{l^2}{2\mu}$$

$$\cos(\theta(r)) = \frac{\frac{l^2}{2\mu k} \cdot \frac{2}{r} - 1}{\sqrt{1 + \frac{2El^2}{\mu k^2}}} = \frac{\frac{l^2}{\mu k r} - 1}{\sqrt{1 + \frac{2El^2}{\mu k^2}}} \quad \text{Substitutions : NEW!!! } \alpha = \frac{l^2}{\mu k}$$

$$E = \sqrt{1 + \frac{2El^2}{\mu k^2}}$$

$$\cos(\theta(r)) = \frac{\frac{\alpha}{r} - 1}{E} : rE \cos(\theta(r)) = \alpha - r : rE \cos(\theta(r)) + r = \alpha : E \cos(\theta(r)) + 1 = \frac{\alpha}{r}$$

$$\therefore \boxed{\frac{\alpha}{r} = 1 + E \cos(\theta(r))}$$

Problem 3

$$a.) \frac{d^2 u}{d\sigma^2} + u = -\frac{\mu}{l^2} \cdot \frac{1}{u^2} F(1/u) \quad F(r) = \frac{-k}{r^3} \quad l^2 = \mu k, \quad l^2 > \mu k, \quad l^2 < \mu k$$

$$F(1/u) = -k u^3$$

$$\frac{d^2 u}{d\sigma^2} + u = \cancel{\frac{\mu}{l^2}} \cdot \frac{1}{\cancel{u^2}} \cdot k \cdot u^3 = \frac{\mu k}{l^2} \cdot u$$

$$\frac{d^2 u}{d\sigma^2} + u = \frac{\mu k}{l^2} u : u'' + u - \frac{\mu k}{l^2} u = 0 : u'' + u \left(1 - \frac{\mu k}{l^2}\right) = 0$$

$$i.) \quad l^2 = \mu k$$

$$\frac{d^2 u}{d\sigma^2} + u = \frac{\mu k}{\cancel{\mu k}} \cdot \frac{1}{\cancel{u^2}} \cdot (\cancel{\mu k}) u^3 : \frac{d^2 u}{d\sigma^2} + u = u \quad \therefore \frac{d^2 u}{d\sigma^2} = 0$$

$$u(\sigma) = C_1 \sigma + C_2 \quad \text{Let's take } C_1 = 2, C_2 = 5 \text{ through out}$$

$$ii.) \quad l^2 > \mu k : \nu = \frac{\mu k}{l^2} : u'' + u(1 - \nu) = 0 : r^2 + (1 - \nu) = 0 : r^2 = \nu - 1$$

$$r = \pm \sqrt{\nu - 1}$$

$$u(\sigma) = C_1 e^{\sqrt{\nu-1} \sigma} + C_2 e^{-\sqrt{\nu-1} \sigma}$$

$$u(\sigma) = C_1 e^{\sqrt{\frac{\mu k}{l^2} - 1} \sigma} + C_2 e^{-\sqrt{\frac{\mu k}{l^2} - 1} \sigma}$$

$$C_1 = 2, C_2 = 5, \frac{\mu k}{l^2} = 10$$

$$iii.) \quad l^2 < \mu k : \nu = -\frac{\mu k}{l^2} : u'' + u(1 + \nu) = 0 : r^2 + (1 + \nu) = 0 : r^2 = -(\nu + 1)$$

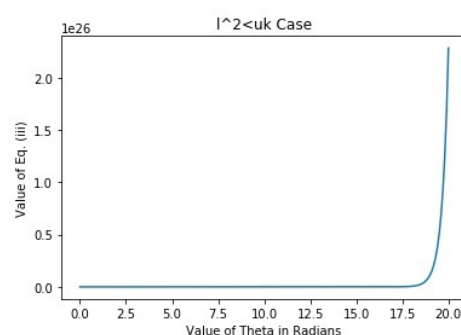
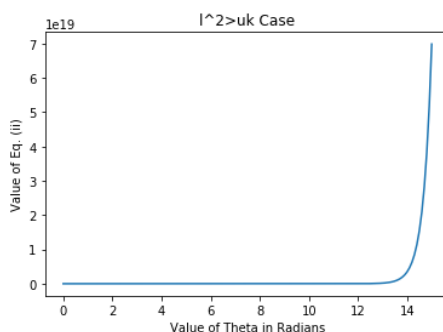
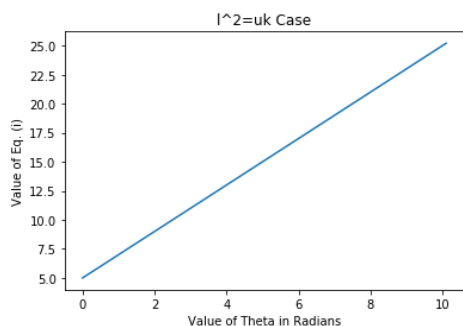
$$r = \pm i \sqrt{\nu + 1}$$

$$u(\sigma) = C_1 e^{i \sqrt{\nu+1} \sigma} + C_2 e^{-i \sqrt{\nu+1} \sigma}$$

$$u(\sigma) = C_1 e^{i \sqrt{\frac{\mu k}{l^2} + 1} \sigma} + C_2 e^{-i \sqrt{\frac{\mu k}{l^2} + 1} \sigma}$$

$$C_1 = 2, C_2 = 5, \frac{\mu k}{l^2} = 8$$

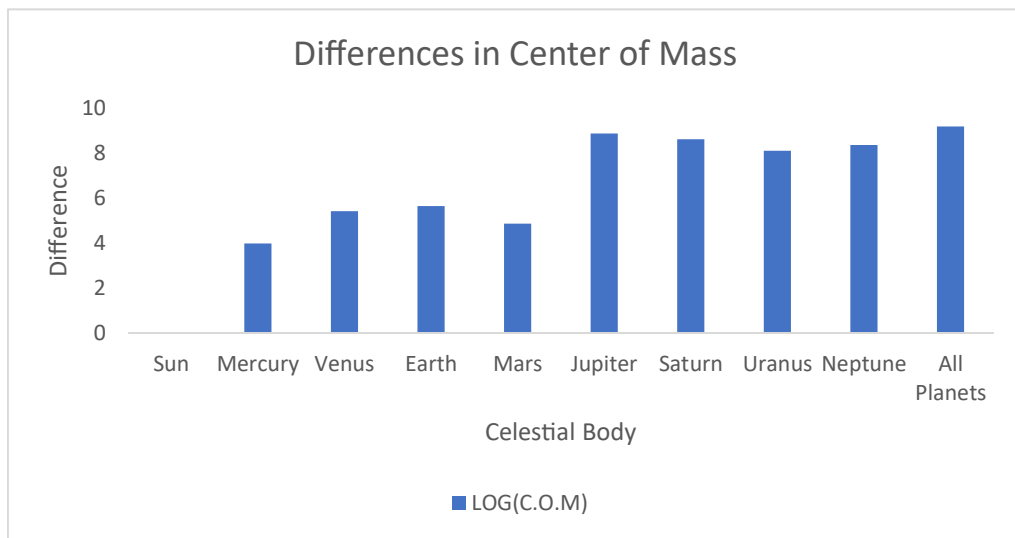
b.)



Problem 4

Celestial Body	Mass Of Celestial Body	Semi-Major Axis	Big M	Center of Mass
Sun	1.9885E+33			
Mercury	3.3011E+26	57909050000	1.99E+33	9613.454022
Venus	4.8675E+27	1.08208E+11	1.99E+33	264873.5986
Earth	5.97237E+27	1.49598E+11	1.99E+33	449309.5606
Mars	6.471E+26	2.27939E+11	1.99E+33	74176.21741
Jupiter	1.8982E+30	7.7857E+11	1.99E+33	742505481.6
Saturn	5.6834E+29	1.43353E+12	1.99E+33	409605051.7
Uranus	8.681E+28	2.87504E+12	1.99E+33	125507330.7
Neptune	1.02413E+29	4.50439E+12	1.99E+33	231976127.2
All Planets				1509210639

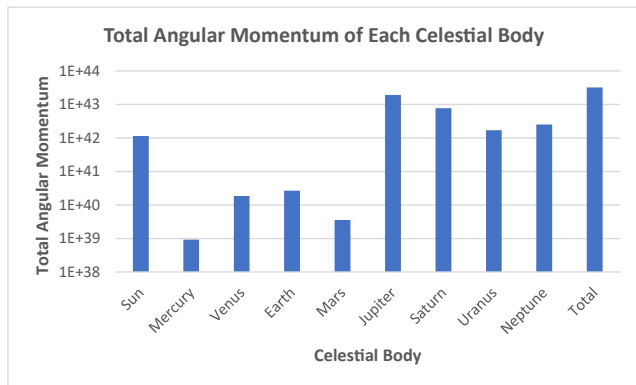
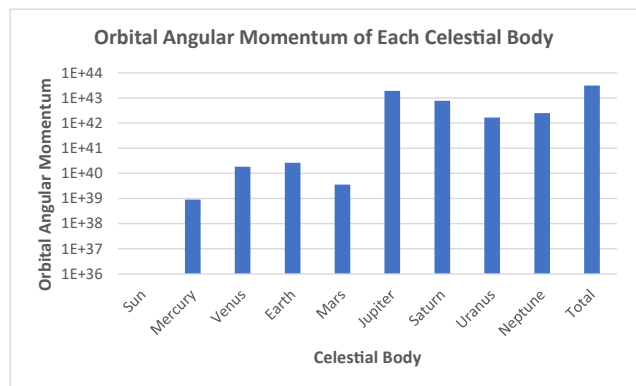
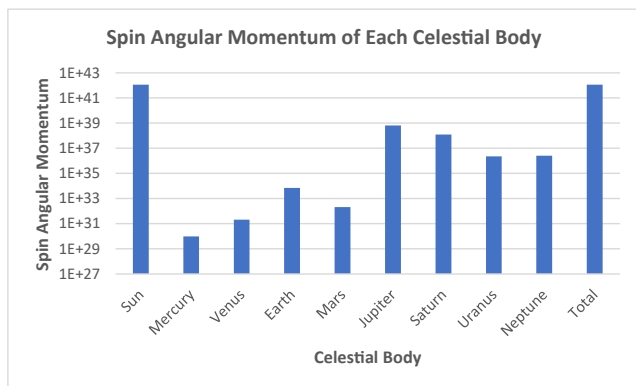
Planet	LOG(C.O.M)
Sun	0
Mercury	3.982879454
Venus	5.423038672
Earth	5.65254566
Mars	4.870264683
Jupiter	8.870699664
Saturn	8.612365304
Uranus	8.098669093
Neptune	8.365443294
All Planets	9.178749858



Problem 5

a.)

Celestial Body	Mass of Body	Radius of Body	Distance From Sun	I of Body	W Rotation	W Orbit	Spin L	Orbital L	Spin + Orbital L
Sun	1.9885E+30	695508000	0	3.8476E+47	2.9719E-06	0	1.1435E+42	0	1.14346E+42
Mercury	3.3011E+23	2440000	57909050000	7.86137E+35	1.2396E-06	8.264E-07	9.7448E+29	9.1482E+38	9.14819E+38
Venus	4.8675E+24	6052000	1.08208E+11	7.13122E+37	2.9927E-07	3.232E-07	2.1341E+31	1.8421E+40	1.84208E+40
Earth	5.97237E+24	6371000	1.49598E+11	9.69665E+37	7.2722E-05	1.992E-07	7.0516E+33	2.663E+40	2.663E+40
Mars	6.471E+23	3389000	2.27939E+11	2.97286E+36	6.9813E-05	1.059E-07	2.0754E+32	3.5589E+39	3.55893E+39
Jupiter	1.8982E+27	69911000	7.7857E+11	3.71102E+42	0.00017453	1.66E-08	6.4769E+38	1.9104E+43	1.91049E+43
Saturn	5.6834E+26	58232000	1.43353E+12	7.70889E+41	0.00015867	6.611E-09	1.2231E+38	7.7214E+42	7.72151E+42
Uranus	8.681E+25	25362000	2.87504E+12	2.23356E+40	0.00010267	2.346E-09	2.2931E+36	1.6833E+42	1.6833E+42
Neptune	1.02413E+26	24622000	4.50439E+12	2.48349E+40	0.00010267	1.208E-09	2.5497E+36	2.5101E+42	2.51014E+42
Total							1.1442E+42	3.1069E+43	3.22128E+43



b.)

$$E_{\text{binding}} = \frac{3}{5} \cdot \frac{6M^2}{r}, \quad T = \frac{L^2}{2I}$$

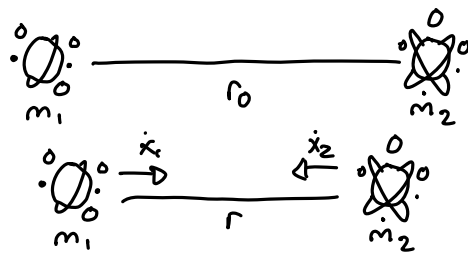
$$E_{\text{binding}} = \text{Kinetic Energy} : \quad \frac{3}{5} \cdot \frac{6M^2}{r} = \frac{L^2}{2I}$$

$$L^2 = \frac{6}{5} \cdot \frac{6M^2 I}{r} : \quad L = \sqrt{\frac{6}{5} \cdot \frac{6M^2 I}{r}}$$

$$L = \sqrt{\frac{6}{5} \cdot \frac{6M^2 I}{r}}$$

The Sun is not in danger of spinning itself out. Even if all the mass was put into the Sun, it would not be problematic.

Problem 6



$$E = T + u : T = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \dot{x}_2^2, \quad u = -\frac{G m_1 m_2}{r}$$

$$E_i = E_f : E_i = -\frac{G m_1 m_2}{r_0}, \quad E_f = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \dot{x}_2^2 - \frac{G m_1 m_2}{r}$$

$$p = m\dot{x} : m_1 \dot{x}_1 + m_2 \dot{x}_2 = 0 : m_1 \dot{x}_1 = -m_2 \dot{x}_2 : \dot{x}_1 = -\frac{m_2}{m_1} \dot{x}_2, \quad \dot{x}_2 = -\frac{m_1}{m_2} \dot{x}_1$$

$$M = m_1 + m_2$$

$$\dot{x}_1$$

$$-\frac{G m_1 m_2}{r_0} = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \dot{x}_2^2 - \frac{G m_1 m_2}{r} : \frac{G m_1 m_2}{r} - \frac{G m_1 m_2}{r_0} = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \dot{x}_2^2$$

$$G m_1 m_2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \left(\frac{m_1^2}{m_2^2} \right) \dot{x}_1^2 : G m_1 m_2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} m_1 \dot{x}_1^2 \left[1 + \frac{m_1}{m_2} \right]$$

$$G m_1 m_2^2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} m_1 \dot{x}_1^2 [m_2 + m_1] : G m_2^2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} \dot{x}_1^2 M, \quad \text{let } M = m_1 + m_2$$

$$\dot{x}_1^2 = \frac{2G m_2^2}{M} \left[\frac{1}{r} - \frac{1}{r_0} \right] \therefore \dot{x}_1 = m_2 \sqrt{\frac{2G}{M} \left(\frac{1}{r} - \frac{1}{r_0} \right)}$$

$$\dot{x}_2$$

$$G m_1 m_2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} m_1 \left(\frac{m_2^2}{m_1^2} \right) \dot{x}_2^2 + \frac{1}{2} m_2 \dot{x}_2^2 : G m_1 m_2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} \left(\frac{m_2^2}{m_1} \right) \dot{x}_2^2 + \frac{1}{2} m_2 \dot{x}_2^2$$

$$G m_1^2 m_2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} (m_2^2) \dot{x}_2^2 + \frac{1}{2} m_2 m_1 \dot{x}_2^2 : G m_1^2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} m_2 \dot{x}_2^2 + \frac{1}{2} m_1 \dot{x}_2^2$$

$$G m_1^2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} \dot{x}_2^2 [m_2 + m_1] : G m_1^2 \left[\frac{1}{r} - \frac{1}{r_0} \right] = \frac{1}{2} \dot{x}_2^2 M : \dot{x}_2^2 = \frac{2G m_1^2}{M} \left[\frac{1}{r} - \frac{1}{r_0} \right]$$

$$\dot{x}_1 = m_2 \sqrt{\frac{2G}{M} \left(\frac{1}{r} - \frac{1}{r_0} \right)}, \quad \dot{x}_2 = m_1 \sqrt{\frac{2G}{M} \left(\frac{1}{r} - \frac{1}{r_0} \right)}$$