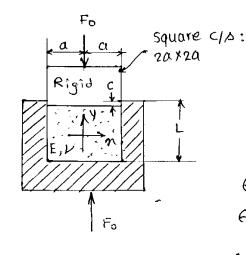
Solutions to H/W Problems Of

Chapter 5

Solution to problem (5.1):



Note:

$$\epsilon_{xx} = \epsilon_{zz} = 0 - 0$$

$$\epsilon_{yy} = -\frac{c}{L} - 0$$

$$\epsilon_{yy} = -\frac{F_0}{(2n)^2} - 0$$

Stress - strain relationship:

$$\begin{aligned} & \in_{XX} = \frac{1}{E} \left[\sigma_{NX} - \nu \left(\sigma_{YY} + \sigma_{ZZ} \right) \right] - \Theta \\ & \in_{YY} = \frac{1}{E} \left[\sigma_{YY} - \nu \left(\sigma_{NX} + \sigma_{ZZ} \right) \right] - \Theta \\ & \in_{ZZ} = \frac{1}{E} \left[\sigma_{ZZ} - \nu \left(\sigma_{NX} + \sigma_{YY} \right) \right] - \Theta \end{aligned}$$

eqns
$$\emptyset$$
, \emptyset , \emptyset \Rightarrow $\sigma_{xx} - \nu_{xy} - \nu_{xz} = 0$ $- \emptyset$

$$-\nu_{xx} - \nu_{xy} + \sigma_{zz} = 0 - \emptyset$$

$$= \epsilon_{yx} = \delta_{xx} = \delta_{xx} = \frac{\nu}{1-\nu} \delta_{yy} - 0$$

equations (5) and (4)
$$\Rightarrow \exists \exists \exists \exists \exists \exists \exists \forall y = \nu \left(\frac{2\nu}{1-\nu}\right) \exists \forall y \exists \exists \exists \exists \nu \in \exists \forall y \in \exists \exists \nu}\right)$$
.

Substituting (2), (3) into (6) we get $-\frac{C}{L} = -\frac{1-\nu-2\nu^2}{(1-\nu)E} \frac{F_0}{(20)^2}.$

$$F_0 = \frac{(1-\nu)E}{1-\nu-2\nu^2} \frac{(2\alpha)^2C}{L} = \frac{(1-\nu)E}{(1+\nu)(1-2\nu)} \frac{4c\alpha^2}{L}$$

Solution to problem (5.9);

Given: Flat steel plate loaded in my plane $\sigma_{m} = 145 \text{ MH/m}^2$, E = 210 GPW $T_{my} = 42 \text{ MH/m}^2$, V = 0.27 $E_{Z} = -3.6 \times 10^{-4}$.

Note: To = 0 (Became of the state of plane stress in xy plane)

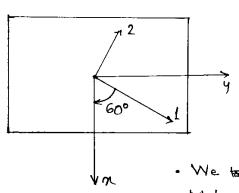
Stress - strain relationship:

$$\epsilon_z = \frac{1}{E} \left(\sigma_z - \nu (\sigma_n + \sigma_y) \right)$$

$$-3.6 \times 10^{4} = \frac{1}{210 \times 10^{9}} \left[0 - 0.27 \left(145 \times 10^{6} + 64 \right) \right]$$

= 135 ×106 H1m²

Solution to problem (5.10):



Given:

- · I'lat all 'Aluminium Plate' loaded in its plane.
- principal strains: $\epsilon_1 = 9.2 \times 10^{-4}, \ \epsilon_2 = -5.4 \times 10^{-4}$
 - · E = 75 GPA, V = 0-33
- · We will first find on and on and use Mohr's circle to find on, oy, o Tmy.
- · Since of = O (Plane stress in x-y plane),

$$\epsilon_1 = \frac{1}{E} (\sigma_1 - \nu \sigma_2) - 0$$

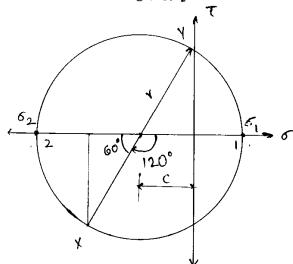
 $\epsilon_2 = \frac{1}{E} (\sigma_2 - \nu \sigma_1) - \emptyset$

equations o and o >

$$\sigma_{1} = \frac{E}{1-\nu^{2}} \left(E_{1} + \nu E_{2} \right) = \frac{75 \times 10^{9}}{1-0.33^{2}} \left[3.7 \times 10^{-4} - 0.33 \left(5.4 \times 10^{-4} \right) \right] = 11.9 \text{ MN/m}^{2}$$

$$\sigma_{2} = \frac{E}{1-\nu^{2}} \left(E_{2} + \nu E_{1} \right) = \frac{75 \times 10^{9}}{1-0.33^{2}} \left[-5.4 \times 10^{-4} + 0.33 \left(3.2 \times 10^{-4} \right) \right] = -36.5 \text{ MN/m}^{2}$$

· MOHR'S CIRCLE:



$$C = \frac{\sigma_1 + \sigma_2}{2} = \frac{-36.5 + 11.9}{2} = -12.3$$

$$r = \frac{\sigma_1 \cdot \sigma_2}{2} = \frac{11.9 + 36.5}{2} = 24.2$$

:. State of stress at XY.

$$6y = (+10000 = -12.3 + 24.2)$$

(Positive as pur the syn convention)

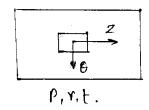
Solution to problem (5.12):

clused - evided

Given: Long, thin-walled & cyllinder.



To find: P.



• Stiesses:
$$G_{\theta} = \frac{p_r}{t}$$
, $G_{z} = \frac{p_r}{2t}$, $G_{r} \simeq 0$

$$E_{z} : E_{z} = \frac{1}{E} \left(G_{z} - \nu G_{\theta} \right)$$

$$= \frac{1}{E} \left(\frac{p_r}{2t} - \nu \frac{p_r}{t} \right).$$

 $=\frac{pr}{F}(0.5-\nu)$

$$\vdots \quad \varepsilon = \frac{pr}{tF} (0.5 - \nu)$$

$$\therefore p = \frac{E}{(0.5-\nu)} + \epsilon_0$$

Solution to problem (5.17)

- · Heating the pulley or cooling the shaft will be equally effective, although it may be easier to heat the pulley. Also, when pulley is to be fined on the shaft at some distance from the shaft ends, it will be much more convenient to heat the pulley.
- Pulley hole dia. = 24.950 mm.
 Shaft dia. = 25.000 mm.

Clearance after heating = 0.025 mm.

of hole diameter. Total thermal empansion = (25.000 - 24.950) +0.025

$$\therefore \epsilon_{\Theta} = \frac{\Delta d}{d} = \frac{0.075}{24.95} = 3 \times 10^{-3}$$

· Storin - temperature relation:

$$\epsilon_0 = \alpha \Delta T$$

$$\kappa = 12 \times 10^6 / {}^{\circ} c \text{ for Steel}$$

$$=\frac{60}{4}$$

$$=\frac{3\times 10^{3}}{12\times 10^{6}}$$

$$=250^{\circ} C$$

d = hole diameter

of the pulley

before healing,

d+ad = hole diameter

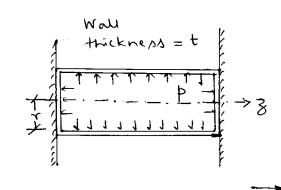
of the pulley

after healing,

$$\frac{1}{160} = \frac{1}{160} = \frac{1}$$

Solution to problem (5.20):

· Equilibrium :



Let F be the force emerted by the wall on the tank, then

$$\sigma_{ZZ}(2\pi rt) = P\pi r^2 - F$$
.

$$\sigma_{ZZ}(2\pi rt) = \rho \pi r^2 - F.$$

$$\sigma_{ZZ} = \frac{\rho r}{2t} - \frac{F}{2\pi rt}.$$

$$\sigma_{ZZ} = \frac{\rho r}{2t} - \frac{F}{2\pi rt}.$$
Also, $\delta \theta = \frac{\rho r}{t}$, $\delta_{rr} = 0$.

Stocks - Stockin Relations:

Also,
$$f_{\theta} = \frac{p_r}{t}$$
., $f_{rr} \cong 0$.

$$\begin{aligned}
&\mathcal{E}_{ZZ} = \frac{\sigma_{ZZ} - \nu \sigma_{\Theta\Theta}}{E} \\
&= \frac{1}{E} \left[\frac{pr}{2t} - \frac{f}{2\pi rt} - \nu \frac{pr}{t} \right] \\
&= \frac{1}{E} \left[\frac{pr}{t} (o.s - \nu) - \frac{F}{2\pi rt} \right].
\end{aligned}$$

Compatibility condition:

:
$$F = 2\pi rt \cdot \frac{pr}{t} (0.5-\nu)$$

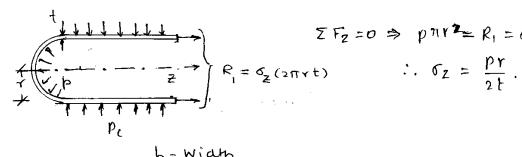
= $\pi r^2 p (1-2\nu)$.

Solution to problem (5.21):

Idealisation: Zero friction between tank and wall of cavity. Let Pc = contact pressure between tank and cavity.

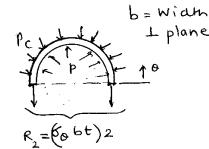
fo=0 in tank wall.

Equilibrium:



$$\Sigma F_{Z} = 0 \Rightarrow p \pi r^{2} = R_{1} = \delta_{Z} (2\pi r t).$$

$$\therefore \delta_{Z} = \frac{pr}{2t}.$$



or is between -p and -pc. Assuming that pc is of the order p we will neglect or (t/ <<1), ie o, =0

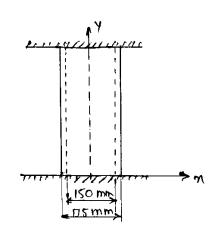
Stress - strain:

compatibility:

$$= \frac{U \, b \, \gamma}{2 \, t}$$

[Note: Then $p_c = (1 - \frac{y}{2}) p$. Thur p_c is of the order of p]

Solution to problem (5.40):



Away from ends, assume only normal stress in y-direction
i.e. 6n = 62 = 0, 2xy = 7y3 = 7y2 = 0. $fy = \frac{6y}{E} + d\Delta T$ $fy = \frac{6y}{E} + d\Delta T$

my = My7 = Mx7 = 0.

Because of constraint by wall,

$$\epsilon_{y=0} \Rightarrow \sigma_{y} = - \kappa \epsilon_{\Delta T}$$

 $E = 210 \times 10^9 \text{ N/m}^2$, $d = 12 \times 10^{-6} / {}^{\circ}_{\text{C}}$ v = 0.27 (sted.) $\Delta T = (-15) - 20 = -35 {}^{\circ}_{\text{C}}$.

 $\sigma_y = -(12 \times 10^{-6} \times 210 \times 10^{9} \times (-35))$ = 88.2 × 106 MN/m².

$$E_{M} = -\frac{V \delta y}{E} + \lambda DT = \frac{V \lambda E DT}{E} + \lambda DT$$

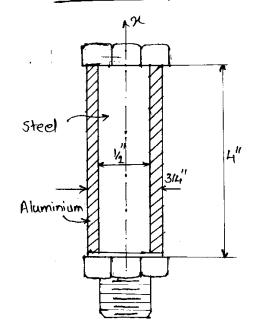
$$= \lambda DT (1+V)$$

$$= 12 \times 10^{-6} \times (-35) (1+0.27)$$

$$= -5.3 \times 10^{-4}$$

 $E_{9} = \text{XOT}(1+V) = 12 \times 10^{-6} \times (-35) (1+0.27)$ = -5.3 \times 10^{-4}

Solution to problem (5.41):



The problem is solved by superposition of following two problems -

- i) Determination of stresses due to turning of the nut.
- ii) Determination of stresses due to temperature rise from 60° to 100° F

In booth the cases the state of stress is 1-0 i.e. there is only on.

For Sleeve:

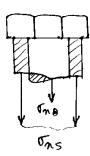
Es = 10 x 106 lb/in2 &s = 12 x 10-6 /0F

For Both: EB = 30 x 106 (b) in2, dg = 6.5 x 10-6 /EF.

(I) Stresses due to turning of nut:

<u>F. B.D.</u>

Equilibrium:



$$\int_{0}^{\infty} \int_{0}^{\frac{\pi}{4}} \left(\left(\frac{3}{4} \right)^{2} - \left(\frac{1}{2} \right)^{2} \right) + \int_{0}^{\infty} \int_{0}^{\frac{\pi}{4}} \left(\frac{1}{2} \right)^{2} = 0.$$

InsAt InaAn = 0

Stress strain Relations:

$$E_{ns} = \frac{\sigma_{ns}}{E_{s}}$$
, $E_{ng} = \frac{\sigma_{ng}}{E_{g}}$.

(Problem 5.41 contd.)

$$\therefore \epsilon_{ns} = \frac{\delta_{ns}}{10 \times 10^6} \qquad -2)$$

$$\epsilon_{\text{NB}} = \frac{\sigma_{\text{NB}}}{30 \times 106}$$
 — 3).

Compatibility:

$$\delta_{NB} - \delta_{NS} = \frac{1}{4} \times \frac{1}{16}$$
 (Sas to, $\frac{1}{4}$ turn, and $\frac{1}{16}$ threads

$$\therefore \frac{\delta n_B}{L} - \frac{\delta n_S}{L} = \frac{\frac{1}{4} \times \frac{1}{16}}{L}$$

:.
$$\epsilon_{nB} - \epsilon_{ns} = \frac{1}{256} - 4$$
 ($l = 4''$).

$$\frac{\sigma_{NB}}{30\times10^6} - \frac{\sigma_{NS}}{10\times10^6} = \frac{1}{256}$$

$$\frac{1}{3} = -3.9 \times 10^4 - 5)$$

$$\sigma_{MS} = -2.75 \times 10^4 \text{ (b)}$$

(II) Stresses due to temperature rise:

FBD and equilibrium tonditions are same as for 1st part.

(problem 2.41 contd.)

Stress strain temperature relations:
$$\Delta T = 4^{\circ}F$$

$$Ens = \frac{\sigma_{NST}}{E_S} + d_S \Delta T = \frac{\sigma_{MST}}{10 \times 10^6} + 4.8 \times 10^{-4} - b$$

$$Eng = \frac{\sigma_{MST}}{E_B} + d_B \Delta T = \frac{\sigma_{MST}}{30 \times 10^6} + 2.6 \times 10^{-4} - c$$

compatibility:

Solution:

b), (), d)
$$\Rightarrow \frac{\sigma_{MST}}{10 \times 10^6} + 4.8 \times 10^{-4} = \frac{\sigma_{MBT}}{30 \times 10^6} + 2.6 \times 10^{-4}$$

a) and e)
$$\Rightarrow$$

$$\sigma_{MST} = -1.55 \times 10^{3} \text{ lblin}^{2}.$$

$$\sigma_{MST} = 1.94 \times 10^{3} \text{ lblin}^{2}$$

(II) Superposition:

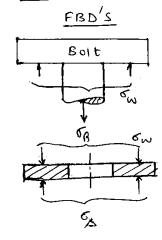
Stress in sleeve =
$$6ms + 6ms + = (-2.75 - 0.155) \times 10^4$$

= $-2.905 \times 10^4 \text{ psi}$

Stress in bolt =
$$Gmg + GmgT = 3.44 \times 10^4 + 0.194 \times 10^4$$

= 3.634 × 104 psi.

Solution to problem 5.44:



Stress - stown Relation:

$$f_B = \frac{\sigma_B}{E_B} \implies \sigma_B = f_B E_B.$$

Equilibrium in Bolt:

$$\frac{68AB}{AW} = \frac{68EBAB}{AW}$$
= $\frac{0.0005 \times 210 \times 10^{4} \times 310 \times 10^{-6}}{625 \times 10^{-6}}$
= $52 \times 10^{6} \, \text{N/m}^{2}$. (compressive) — A

(B) Stresses due to [temp. change]: Let temperature change Stress - Strain - Temp Relations: produce tensile stresses.

$$\epsilon_{BT} = \frac{\delta g_T}{E_B} + \lambda_B \Delta \tau$$
 — ∞

$$\epsilon_{WT} = \frac{\delta_{WT}}{\epsilon_W} + \alpha_W \Delta \tau - \emptyset$$

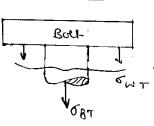
$$\epsilon_{ST} = \frac{\sigma_{ST}}{\epsilon_S} + d_S \Delta \tau$$
, — 3

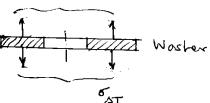
Equilibrium of Bolt:

$$\Rightarrow \sigma_{BT} = -\frac{\sigma_{WT} A_{W}}{A_{B}} - \Theta$$

Equilibrium of washer:

FBD's





Compatibility condition: SBT = SWT + SST.

.. FAT LB = EWTLW + FST LS. - 6

From equations (D, Q, 3), 6

$$L_{B}\left(\frac{\sigma_{BT}}{E_{B}} + \varkappa_{B}\Delta T\right) = L_{W}\left(\frac{\sigma_{WT}}{E_{W}} + \varkappa_{W}\Delta T\right) + L_{S}\left(\frac{\sigma_{ST}}{E_{S}} + \varkappa_{S}\Delta T\right) - \mathcal{O}$$

From equation (1), (1), (1)

$$\sigma_{WT}\left(-\frac{L_{B}A_{LO}}{E_{B}A_{B}}-\frac{L_{W}}{E_{W}}-\frac{L_{E}A_{W}}{E_{S}A_{S}}\right)=\left(L_{W}A_{W}+L_{S}A_{S}-L_{B}A_{A}\right)\Delta T_{s}$$

Lw = 1.5 mm, Ls = 150 mm, LB = 151.5 mm.

dw=dB = 12 x10-6 /00, ds = 22 x10-6 /00.

 $A_{10} = A_{5} = 625 \text{ mm}^{2}$ $A_{6} = 310 \text{ mm}^{2}$.

 $E_W = E_B = 210 \times 10^{+9} \text{ N/m}^2$ $E_S = 75 \times 10^{9} \text{ N/m}^2$ = $75 \times 10^{3} \text{ N/mm}^2$ = $75 \times 10^{3} \text{ N/mm}^2$.

DT = 95°C.

Substituting the values we get

$$\sigma_{\text{WT}} = -41.1 \cdot \text{N/mm}^2$$

$$= 41.1 \cdot \text{x} \cdot 10^6 \, \text{N/m}^2 \quad \text{(compressive)}. - \text{B}$$