

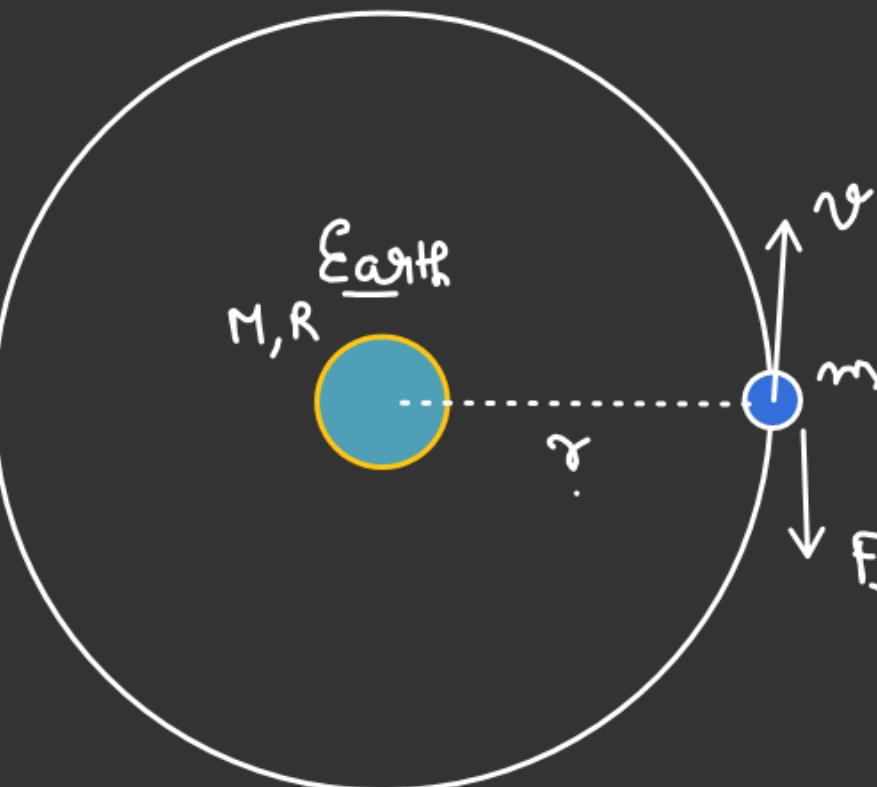
GRAVITATION

A Satellite moving in a Circular orbit of radius ($r = nR$). around Earth. Resistive force acting on the Satellite $F_r = \alpha v^2$.

How long it takes for Satellite to collide with earth.

Total Energy of Satellite at an orbital radius r is $E_T = \left(-\frac{GMm}{2r} \right)$
 (Rate of loss of Energy of Satellite) = $- (F \cdot v)$

$$-\left(\frac{dE_T}{dt}\right) = -Fv$$





GRAVITATION

$$\left(\frac{dE_T}{dt} \right) = - \underline{Fv} \quad \checkmark$$

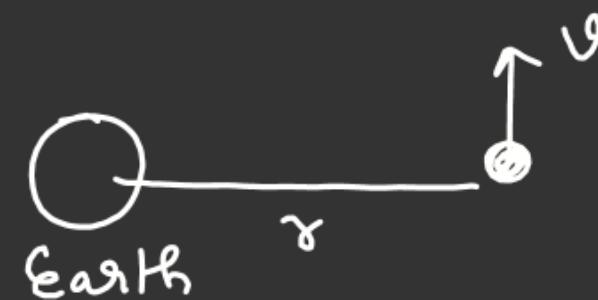
$$\frac{d}{dt} \left(-\frac{GMm}{2r} \right) = -(\propto v^2)v$$

$$+ \frac{GMm}{2} \left(-\frac{1}{r^2} \right) \left(\frac{dr}{dt} \right) = \propto v^3$$

$$-\frac{GMm}{2r^2} \left(\frac{dr}{dt} \right) = \propto \left(\frac{GM}{r} \cdot \sqrt{\frac{GM}{r}} \right)$$

$$-\frac{1}{2} \sqrt{\frac{1}{GM}} \int_{nR}^R r^{-1/2} dr = \frac{\propto}{m} \int_0^t dt \quad t = \frac{m}{\propto} \sqrt{\frac{R}{GM}} \left(\sqrt{n} - 1 \right)$$

Ans ✓



$$v_r = \sqrt{\frac{GM}{r}}$$

orbital
velocity of Satellite
at any radial distance

GRAVITATION

Assume Circular Orbit of Earth around Sun. Assume Earth Suddenly Stop rotating. Find time when earth collapse in Sun. $T =$ (Time period of Earth about sun)

$$-\frac{GMm}{r} = -\frac{GMm}{x} + \frac{1}{2}mv^2$$

$$\frac{1}{2}mv^2 = GMm\left(\frac{1}{r} - \frac{1}{x}\right)$$

$$v^2 = \frac{2GM}{r}\left(\frac{r-x}{x}\right)$$

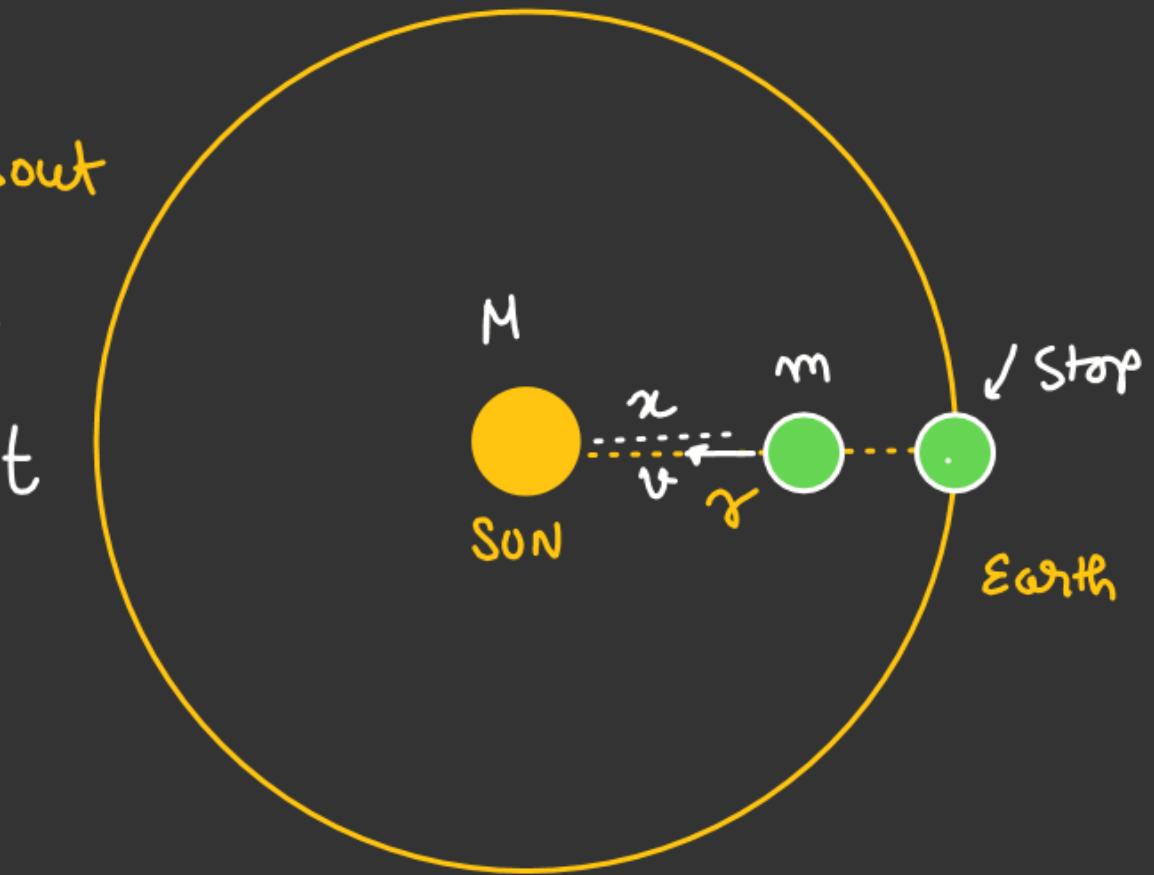
$$v = \sqrt{\frac{2GM}{r}} \sqrt{\frac{r-x}{x}}$$

$$-\frac{dx}{dt} = \sqrt{\frac{2GM}{r}} \sqrt{\frac{r-x}{x}}$$

$$\int_{r}^{x} \sqrt{\frac{x}{r-x}} dx = -\sqrt{\frac{2GM}{r}} \int_{0}^{t} dt$$

↓
put
 $x = r \sin^2 \theta$

$$(t = \frac{T}{2\sqrt{g}})$$



GRAVITATION

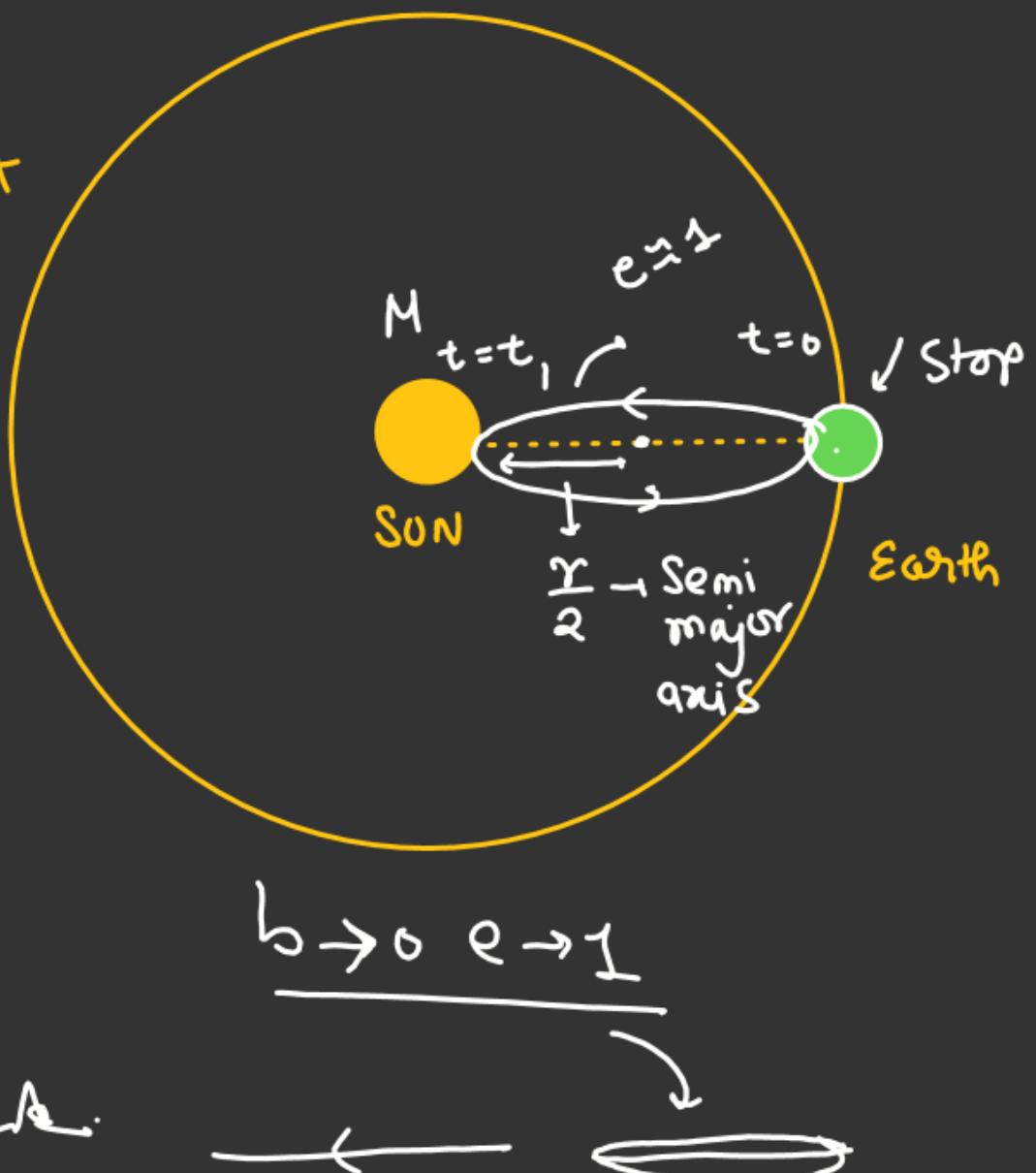
Assume Circular Orbit of Earth around Sun. Assume Earth Suddenly Stop rotating. Find time when earth Collapse in Sun. $T =$ (Time period of Earth about sun)

Assume motion of Earth as elliptical
let, T' be its time period and
 $\frac{r}{2}$ be the Semimajor axis.

By Kepler's 3rd law.

$$\frac{(T')^2}{(T)^2} = \frac{\left(\frac{r}{2}\right)^3}{(r)^3} \quad t_1 = \frac{T'}{2}$$

$$\frac{(T')^2}{T^2} = \frac{1}{8} \Rightarrow T' = \frac{T}{\sqrt[2]{8}}$$



GRAVITATION

Q.: A Body is projected at an angle θ from horizontal from the surface of earth with velocity v_0 .

Show that the semi-major axis of ellipse is independent of θ .

Let, r be either min or maximum distance of the body.

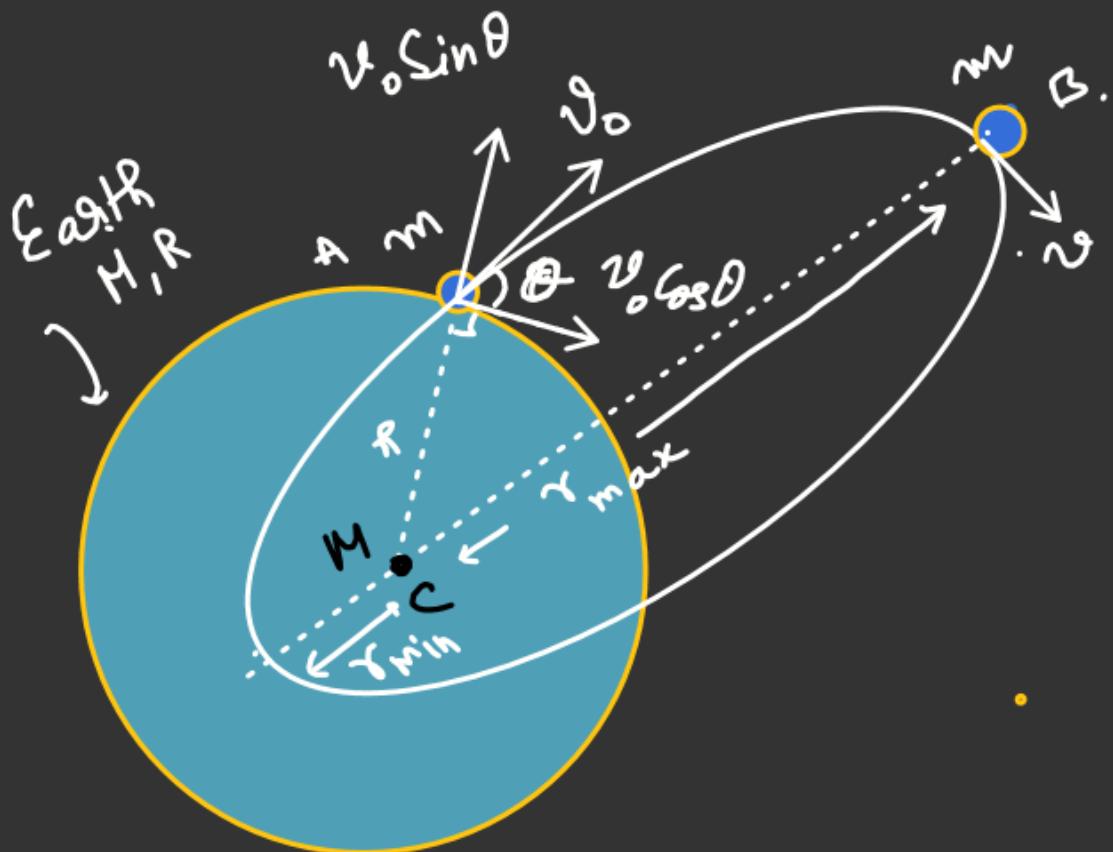
A.M.C about C from A to B.

$$(mv_0 \cos \theta) R = mv r - \textcircled{1}$$

Energy Conservation from A to B.

$$-\frac{GMm}{R} + \frac{1}{2}mv_0^2 = -\frac{GMm}{r} + \frac{1}{2}mv^2 - \textcircled{2}$$

$$\text{From } \textcircled{1} \quad v = (Rv_0 \cos \theta)$$



Semimajor axis = $\left(\frac{r_{max} + r_{min}}{2} \right)$

GRAVITATION

$$(m v_0 \cos \theta) R = m v r - ①$$

Energy Conservation from A to B.

$$-\frac{GMm}{R} + \frac{1}{2}mv_0^2 = -\frac{GMm}{r} + \frac{1}{2}mv^2 - ②$$

From ① $v = \left(\frac{Rv_0 \cos \theta}{r} \right)$

$$-\frac{GMm}{R} + \frac{m v_0^2}{2} = -\frac{GMm}{r} + \frac{m}{2} \left(\frac{R^2 v_0^2 \cos^2 \theta}{r^2} \right)$$

$$GMm \left(\frac{1}{r} - \frac{1}{R} \right) = \frac{mv_0^2}{2} \left(\frac{R^2 \cos^2 \theta}{r^2} - 1 \right)$$

↙

$r \rightarrow$ quadratic

$r_{\max} + r_{\min} = \text{sum of root}$

Semi major

$$\text{axis} = \left(\frac{r_{\max} + r_{\min}}{2} \right)$$

Check

$$= \left(\frac{g R^2}{2gR - v_0} \right) \text{ Ans}$$

GRAVITATION

A Satellite revolving around earth in circular orbit.
of radius r with velocity v_0 .

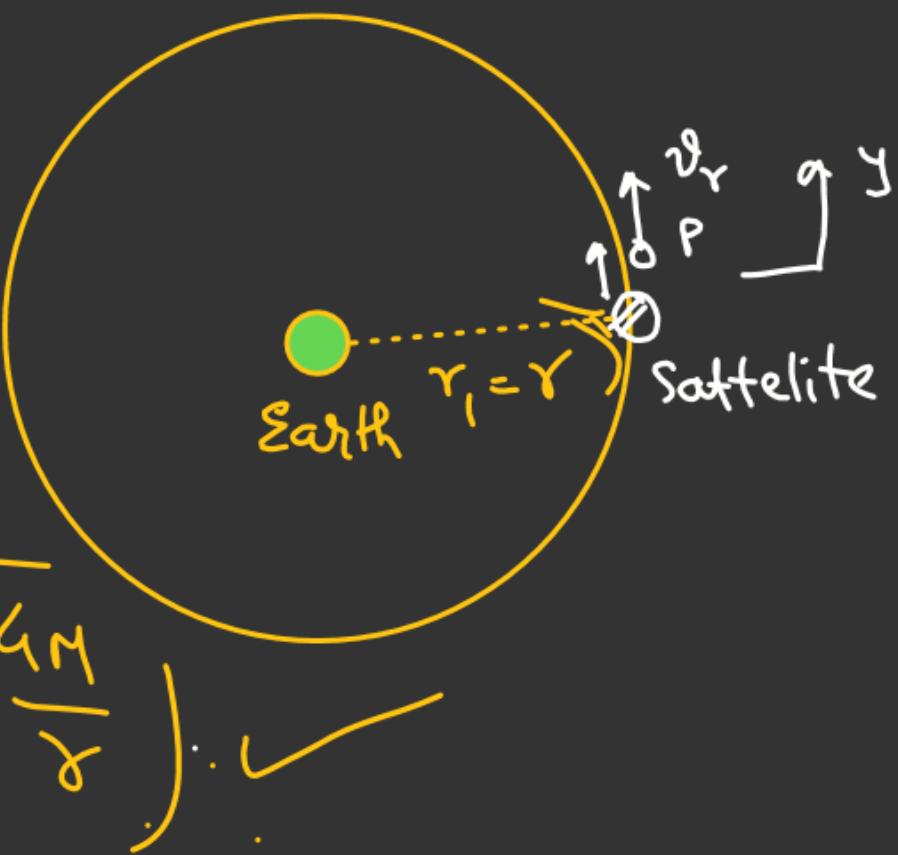
A particle is projected from the satellite in forward direction with velocity $(\sqrt{\frac{5}{4}} - 1)v_0$.

Find max and min distance of the particle w.r.t
earth during its subsequent motion

Solⁿ
Absolute velocity of particle

$$\begin{aligned}\vec{v}_{P/E} &= \vec{v}_{P/\text{satellite}} + \vec{v}_{\text{satellite}/E} \\ &= \left((\sqrt{\frac{5}{4}} - 1)v_0 + v_0 \right) \hat{j} \\ &= \left(\sqrt{\frac{5}{4}} v_0 \right)\end{aligned}$$

$v_p > v_0 \Rightarrow$ (Trajectory is ellipse) $\left(v_0 = \sqrt{\frac{GM}{r}} \right)$. ✓



GRAVITATIONA.M.C.

$$v_0 = \sqrt{\frac{GM}{r}} \quad \checkmark$$

$$\sqrt{\frac{5}{4}} v_0(r) = v_2 \gamma_2 \quad \textcircled{1}$$

$$-\frac{GMm}{r} + \frac{1}{2} m \left(\sqrt{\frac{5}{4}} v_0 \right)^2 = -\frac{GMm}{r_2} + \frac{1}{2} m v_2^2 \quad \textcircled{2}$$



$$\gamma_2 = \left(\frac{5r}{3} \right), \quad \gamma_1 = r$$

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