

# A Conceptual Distributed Embedded Architecture for Modular Unmanned Aerial Systems

## A Conceptual Distributed Embedded Architecture for Modular Unmanned Aerial Systems

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### ABSTRACT

The increasing operational complexity of unmanned aerial systems (UAS) places concurrent demands on real-time flight control and computationally intensive perception workloads—demands that monolithic embedded architectures are ill-suited to satisfy. This paper presents a conceptual distributed embedded architecture for modular UAS platforms in which flight-critical control functions are partitioned from sensor processing and perception tasks across independent processing domains. The proposed design facilitates deterministic control loop execution while enabling the integration of heterogeneous sensing modalities, including inertial, optical, thermal, and barometric sources. System-level principles of modularity, fault isolation, graceful degradation, and inter-node communication are examined at an architectural abstraction level. Extensibility toward multi-agent swarm configurations is also considered. This work does not provide implementation-specific schematics, firmware, or hardware configurations; rather, it establishes a conceptual framework intended to inform future research in resilient, scalable embedded UAS design.

**Index Terms**— Unmanned Aerial Systems, Distributed Embedded Systems, Sensor Fusion, Fault Tolerance, Modular Architecture, Multi-Agent Systems

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### I. INTRODUCTION

Unmanned aerial systems have evolved from specialized research instruments into platforms supporting a broad range of civil and commercial applications, including environmental monitoring, precision agriculture, search and rescue, and infrastructure inspection [1]. This expansion of mission scope has introduced requirements for increased autonomy, rich environmental awareness, and high operational reliability—requirements that place substantial and often competing demands on onboard embedded computing resources.

Monolithic embedded architectures, in which a single processing unit governs all system functions, are subject to well-documented limitations in multi-workload environments. Resource contention between deterministic control loops and computationally intensive perception tasks can degrade timing guarantees, jeopardize stability margins, and reduce fault containment. As UAS platforms incorporate denser sensor suites and higher levels of onboard autonomy, these limitations become increasingly consequential [4].

Distributed embedded architectures offer a principled approach to managing this complexity. By partitioning workloads across dedicated processing nodes, flight-critical functions can be afforded deterministic execution environments free from interference by non-critical subsystems [3]. This separation simultaneously promotes modularity and scalability, enabling mission-specific reconfiguration without wholesale system redesign.

This paper presents a conceptual distributed architecture for a modular UAS platform. The system partitions operations into a Flight Control Domain and a Sensor and Perception Domain, each instantiated on an independent processing node and communicating through an abstracted inter-node interface. The remainder of this paper is organized as follows: Section II describes the overall system architecture; Sections III and IV detail the flight control and sensor domains, respectively; Section V addresses sensor fusion; Section VI examines inter-node communication; Section VII discusses fault tolerance; Section VIII addresses modularity and scalability; Section IX considers swarm extensibility; Section X discusses security; Section XI presents ethical context; and Section XII concludes the paper.

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## II. SYSTEM ARCHITECTURE

### *A. Architectural Overview*

The proposed system partitions all UAS functions into two primary embedded domains:

1. **Flight Control Domain** — responsible for real-time vehicle dynamics and system stability.
2. **Sensor and Perception Domain** — responsible for data acquisition, preprocessing, and environmental awareness.

Each domain operates on an independent processing node. Inter-domain communication is mediated through an abstracted data interface layer that decouples the timing behavior of the two domains. This structural separation ensures that computational bursts arising from perception workloads cannot propagate into the timing-critical control path, preserving the determinism required for stable flight operations.

### *B. Conceptual Architecture Diagram*

Fig. 1 (see `Diagram.md`) illustrates the hierarchical organization of the system. The Ground Interface defines the external command and telemetry boundary. The Flight Control Node forms the primary processing domain and manages actuation and stability. The Data Interface Layer provides the inter-node communication abstraction. The Sensor Processing Node aggregates and conditions all perceptual data prior to delivery to the control domain.

This diagram represents a conceptual model only and does not reflect any specific hardware implementation.

### III. FLIGHT CONTROL DOMAIN

The flight control domain constitutes the primary embedded node and is responsible for all time-critical vehicle operations. Core functions include:

- **Attitude stabilization** — closed-loop regulation of pitch, roll, and yaw via control effectors.
- **Control loop execution** — periodic execution of stabilization algorithms at fixed rates to maintain deterministic timing behavior.
- **Power management and distribution** — monitoring and arbitration of onboard power resources across all subsystems.
- **System health monitoring** — detection of anomalous operating conditions and initiation of appropriate responses.
- **Fail-safe fallback modes** — layered degraded-mode operation strategies that allow the platform to maintain baseline flight capability in the event of secondary subsystem failure.

The domain is architected to prioritize execution determinism above all other scheduling concerns. The fail-safe mode hierarchy ensures that degradation of the perception domain does not precipitate loss of flight control, preventing cascading failures from compromising vehicle safety.

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### IV. SENSOR AND PERCEPTION DOMAIN

The sensor and perception domain constitutes the secondary embedded node and is dedicated to computationally intensive data acquisition and preprocessing. Key functions include:

- **Thermal data acquisition and analysis** — capture and conditioning of infrared sensor data for environmental and thermal awareness.
- **Optical imaging and visual processing** — acquisition and initial processing of visible-spectrum imagery for situational awareness and navigation support.
- **Inertial measurement aggregation** — integration and preprocessing of inertial measurement unit (IMU) data streams.
- **Altitude estimation** — barometric and sensor-fused estimation of vehicle altitude above ground.
- **Data conditioning** — filtering and normalization of sensor outputs prior to transmission across the inter-node interface.

By hosting these functions on a dedicated node, the architecture prevents resource contention with the time-critical control path. Computational spikes associated with image processing or multi-sensor aggregation are fully isolated to the perception domain and do not affect control loop timing.

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## V. SENSOR FUSION

Sensor fusion integrates observations from multiple sensing modalities to produce state estimates that are more accurate and robust than those attainable from any individual sensor [2]. Within the proposed architecture, fusion operations are hosted within the Sensor and Perception Domain, preserving the timing integrity of the flight control node.

Conceptually, fusion operations include the following:

- **Inertial-barometric integration** — combining IMU measurements with barometric altitude data to improve vertical state estimation accuracy and robustness.
- **Visual-inertial coupling** — incorporating optical flow or feature-based visual feedback to augment inertial navigation, particularly in GPS-degraded environments.
- **Sensor noise filtering** — application of prediction and correction models to attenuate measurement noise and compensate for sensor latency and bias.

High-level fusion strategies such as Kalman-family estimators are well-established in the UAS literature [2][3]. Specific algorithm implementations, tuning parameters, and data schemas are outside the scope of this work.

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## VI. INTER-NODE COMMUNICATION

Communication between the flight control and sensor processing domains is mediated by a data interface layer designed to decouple the temporal behavior of the two nodes. The interface supports asynchronous data exchange, permitting each node to operate at its native task rate without synchronization dependencies that could introduce latency into the control path.

Design principles governing the interface include:

- **Asynchronous data exchange** — nodes produce and consume data independently at their respective native rates.
- **Fault-tolerant messaging** — the interface is designed to tolerate transient packet loss without corrupting control state or triggering spurious fault conditions.
- **Graceful degradation** — under sustained data loss or interface fault conditions, the flight control node defaults to previously valid state data or predetermined safe fallback values.

Specific communication protocols, bus topologies, message framing conventions, and transmission parameters are implementation-dependent and are not defined in this work.

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## VII. FAULT TOLERANCE AND SYSTEM RESILIENCE

A key architectural benefit of distributed processing is the containment of faults within defined domain boundaries. The proposed system is structured such that failure of the sensor and perception domain

does not propagate to the flight control domain. This property is maintained through several complementary mechanisms:

- **Domain isolation** — independent processing environments and power domains prevent fault propagation across the inter-node boundary.
- **Independent control loops** — flight stabilization functions execute autonomously within the control domain, with bounded dependency on sensor domain outputs.
- **Power segmentation** — independent power regulation for each domain allows one domain to degrade or lose power without compromising the other.
- **Layered fallback modes** — the flight control node implements multiple levels of degraded-mode operation, scaling functionality down progressively in response to subsystem failures rather than failing catastrophically.

This layered approach to resilience is consistent with established principles of safety-critical embedded system design [3]. The architecture prioritizes the preservation of basic flight stability over the completeness of perception and sensing functions.

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## VIII. MODULARITY AND SCALABILITY

The domain-partitioned architecture provides a natural basis for modular system extension. The separation of concerns between control and perception domains enables independent evolution of each subsystem:

- **Incremental sensor integration** — additional sensing modalities can be incorporated into the perception domain without modification to the control domain, provided they conform to the inter-node interface specification.
- **Processing unit substitution** — individual processing nodes can be upgraded or replaced independently, enabling performance scaling without full system redesign.
- **Subsystem reconfiguration** — responsibilities within the perception domain can be redistributed across additional nodes as mission requirements evolve or processing demands increase.

This modularity enables platform adaptation across diverse mission profiles—from lightweight inspection tasks with minimal sensor payloads to sensor-dense environmental mapping configurations—without requiring architectural redesign.

## IX. SWARM SYSTEM CONSIDERATIONS

Although this work focuses on a single-platform architecture, the node-partitioned design principle extends naturally to multi-agent swarm configurations. Several architectural properties support swarm scalability:

- **Replicable node structure** — the dual-domain architecture can be instantiated across multiple platforms with minimal per-unit customization, reducing the engineering overhead of fleet deployment.
- **Abstracted coordination layers** — an inter-platform communication abstraction, analogous in structure to the inter-node interface, can mediate distributed coordination without requiring centralized control.
- **Decentralized state sharing** — telemetry and state data can be exchanged across platforms in non-centralized formats, supporting emergent collective behaviors without single points of coordination failure.

Detailed swarm coordination algorithms, consensus protocols, and formation control strategies are beyond the scope of this paper and represent substantive directions for future research.

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## X. SECURITY AND SYSTEM PROTECTION CONSIDERATIONS

Distributed architectures introduce inter-node communication surfaces that must be considered in any security-conscious design. At a conceptual level, the system design addresses the following concerns:

- **Interface hardening** — mechanisms to prevent unauthorized injection of data into the inter-node communication layer.
- **Hardware access protection** — physical and logical measures to prevent unauthorized access to or exploitation of onboard processing resources.
- **Non-recoverability provisions** — system-level safeguards designed to prevent sensitive hardware or data from being accessed or extracted under adverse operational conditions.

No implementation-specific security mechanisms, cryptographic schemes, or access control configurations are described in this work.

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## XI. ETHICAL AND LEGAL CONSIDERATIONS

This research is presented exclusively for educational and conceptual purposes within the domain of distributed embedded systems design. The work does not contain classified information, export-controlled technical data, operational or tactical methodologies, or hardware and firmware specifications sufficient for system replication.

All readers bear individual responsibility for ensuring compliance with applicable laws, regulations, export controls, and safety standards governing the development, deployment, and operation of unmanned aerial systems within their respective jurisdictions.

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## XII. CONCLUSION

This paper has presented a conceptual distributed embedded architecture for modular unmanned aerial systems, partitioning flight-critical control functions from sensor processing and perception workloads into independent processing domains. The architecture promotes deterministic control execution, fault domain isolation, heterogeneous sensor integration, and modular system expansion without mandating specific hardware or firmware implementations.

The design demonstrates that workload partitioning is a sound architectural strategy for managing the competing computational demands of modern UAS platforms. Fault isolation through domain separation reduces the risk of cascading failures, while the abstracted inter-node interface facilitates scalable, mission-adaptive system design. The architecture’s replicable structure further positions it as a candidate foundation for multi-agent swarm system research.

Future work may explore formal analysis of inter-node timing properties, adaptive degradation policy design, AI-assisted perception within the sensor domain, and the application of this partitioned architecture to distributed swarm coordination frameworks.

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*Author’s Note: All system descriptions in this paper are intentionally abstracted. No schematics, firmware, or detailed hardware configurations are provided.*