









04.09.2015

Agenda (2/3)

- Module 4 Design requirements and Safety process
 - 4-1 Requirements
 - 4-2 Safety process
- Module 5 Aircraft power systems
 - 5-1 Hydraulic power systems
 - 5-2 Electric power systems
- Module 6 Aircraft Control systems Architectures
 - 6-1 Hydro Mechanical Systems
 - 6-2 Fly by wire systems
 - 6-3 Fly by wire systems new generation incl future
 - 6-4 A320 FAL Visit















Outlines

04.09.2015

- Generals : Control surfaces & Cockpit controls
- Mechanical control systems (ATR)
- Hydromechanical control systems (A300/A310/B737 etc)
- First generation Fly by Wire systems (A320/A330/ A340/B777)
- New generation, hybrid power sources, full Fly By Wire systems (A380/A400M/A350/B787)
- Future trends, smart and more electric actuation





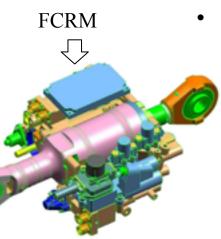






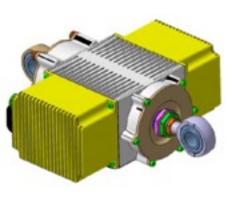






"Smart" actuators: Introduction of local electronics, A350 "Flight Control Remote Module" , B787 "Remote Electronic Unit"

- Airbus driver: Analog signals vulnerability to EMI when wiring installed in a composite fuselage
- Principle: achieve the closed loop control locally and concentrate data for bidirectional exchanges with flight control computers through digital data buses
- Generates weight saving
- Further reduction of the number of hydraulic systems
 - Driver: Bleedless aircraft electrical generation sized for the high power requirements of the Environment Control and Wing Ice Protection Systems, and able to power flight controls and other users with no significant over sizing, weight reliability and maintainability benefits
 - Issues to be solved: Jam free/tolerant landing gear actuation
- Introduction of Electromechanical Actuators EMA
 - Driver: simpler/lighter/cheaper than EHA
 - Issues to be solved: jamming and backlash
 - Spoiler EMA introduced on B787 for dissimilarity











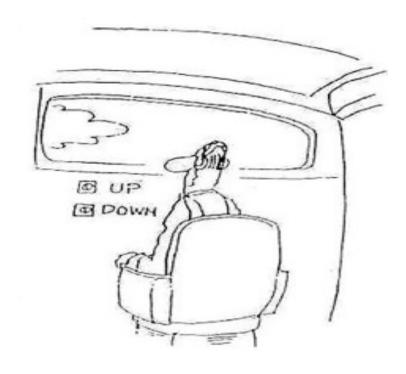






Future trends, more electric systems

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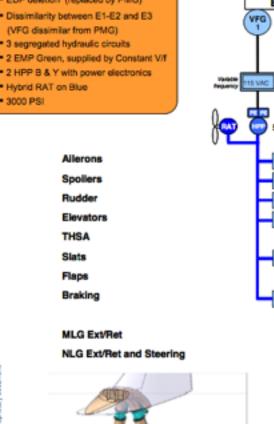




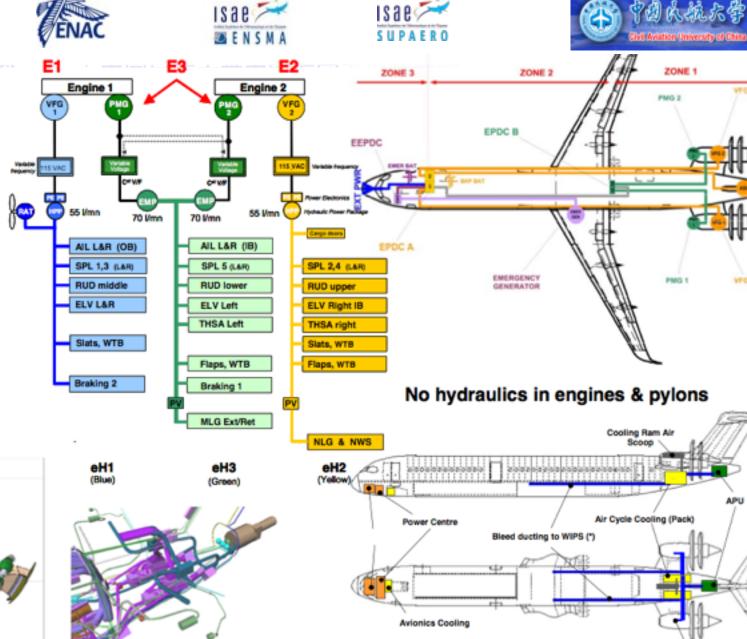




- 3 segregated electrical sources
- EDP deletion (replaced by PMG)
- Dissimilarity between E1-E2 and E3
- 2 EMP Green, supplied by Constant V/f
- 2 HPP B & Y with power electronics
- Hybrid RAT on Blue
- 3000 PSI























Towards more electric aircraft

- For years engineers have dreamed of all-electric aircraft. They have envisioned a concept called "power-by-wire," in which electrical power moves aircraft flight surfaces.
- The goal of power-by-wire is to eliminate the hydraulic connection, and its associated risks.
- The transition to all-electric aircraft is still many years in the future.
- Aircraft engineers have tested electro hydrostatic actuators (EHAs), which combine electrical and hydraulic power: hence the evolutionary "more electric aircraft" idea.

















EHA Concept

- EHAs are electrically powered but use small hydraulic pumps and reservoirs that transform electrical power into hydraulic power.
- Airbus has worked with EHA flight control technologies for more than a decade. A320 and A340 flight test beds have operated since 1993-94 and 2000, respectively.
- The U.S. military's Joint Strike Fighter and C-141 Electric Starlifter programs have tested EHA systems, as well.









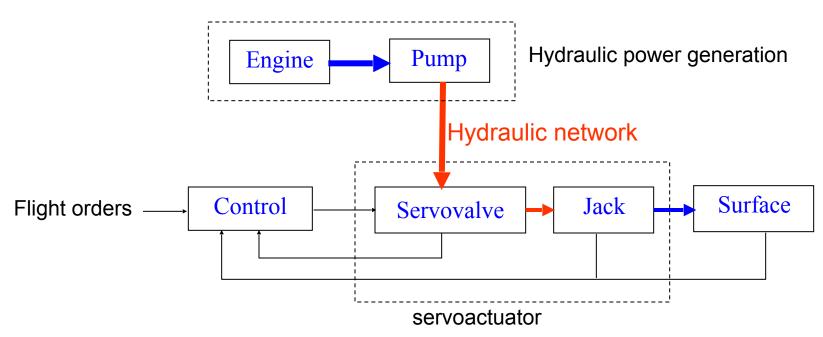








Typical architecture of actuation system



Mass proportion:

tubing fitting and attach fluid equipments 25% 15% 35% 25%









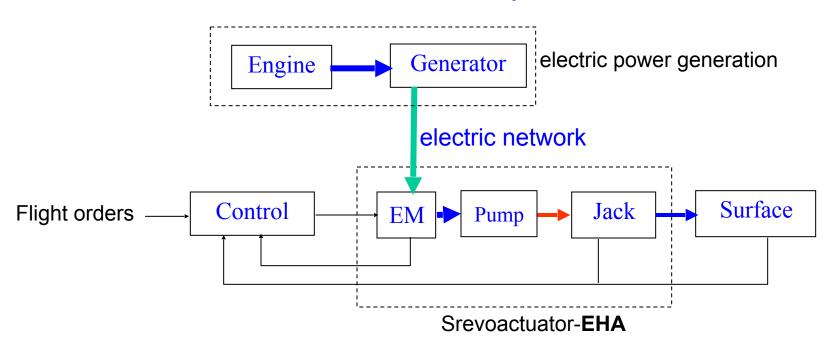








Alternative actuation system



• Mass proportion:











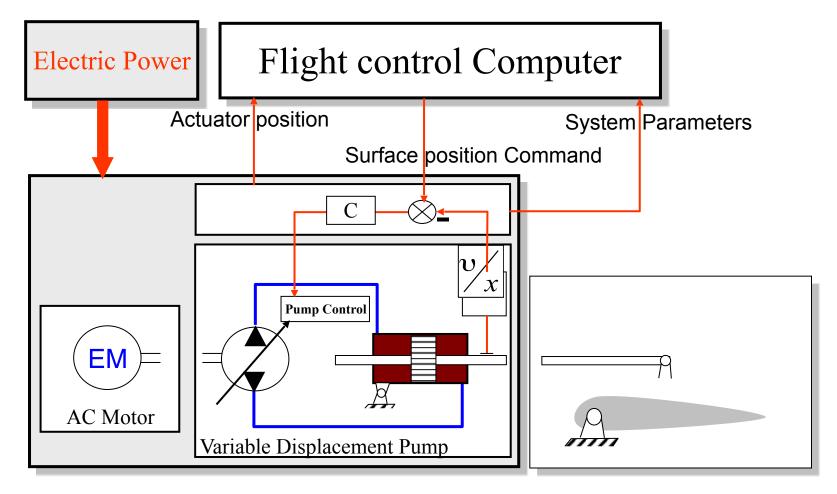








Pump Control EHA











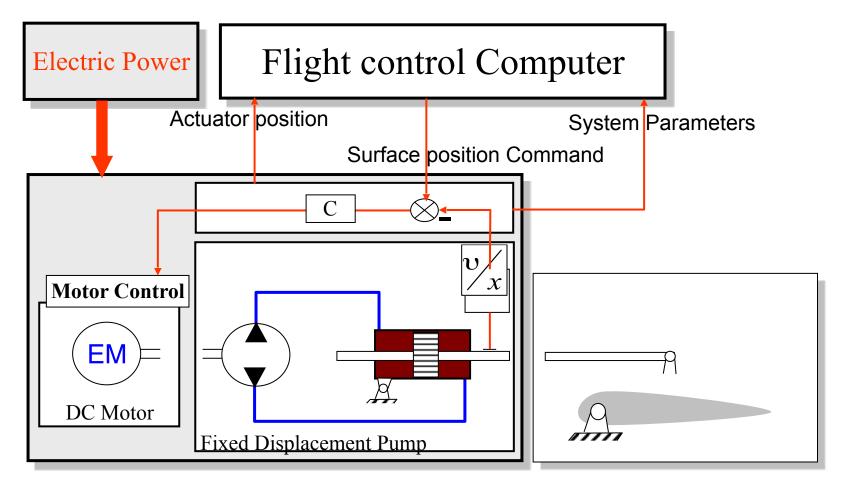








Motor Control EHA



















Electrohydrostatic Actuator



For: JSF (Joint Strike Fighter)

Designed by: Parker Aerospace Control Systems Division Military

















Future of hydraulics











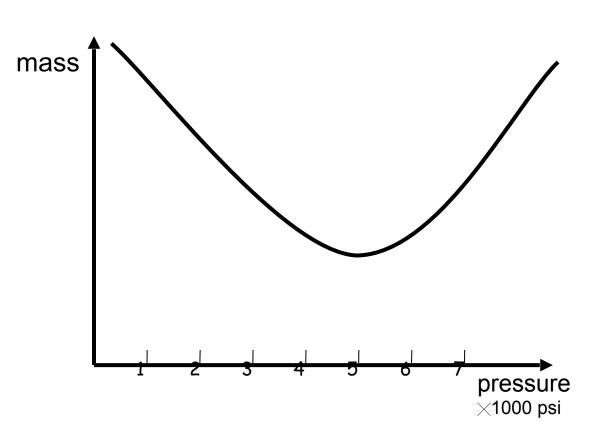






6.1 High pressure circuits

System pressure VS. system mass



The higher the pressure, the thinner the pipe, the thicker the pipe wall!

