









## Agenda (2/3)

- Module 4 Design requirements and Safety process
  - 4-1 Requirements
  - 4-2 Safety process
- Module 5 Aircraft power systems
  - 5-1 Hydraulic power systems
  - 5-2 Electric power systems
- Module 6 Aircraft Control systems Architectures
  - 6-1 Hydro Mechanical Systems
  - 6-2 Fly by wire systems
  - 6-3 Fly by wire systems new generation
  - 6-4 A320 FAL Visit















#### **Outlines**

- Generals : Control surfaces & Cockpit controls
- Mechanical control systems (ATR)
- Hydromechanical control systems (A300/A310/B737 etc)
- First generation Fly by Wire systems (A320/A330/ A340/B777)
- New generation, hybrid power sources, full Fly By Wire systems (A380, others)
- Future trends, smart and more electric actuation









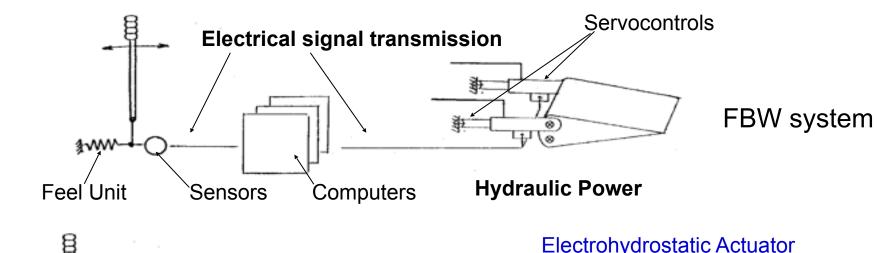


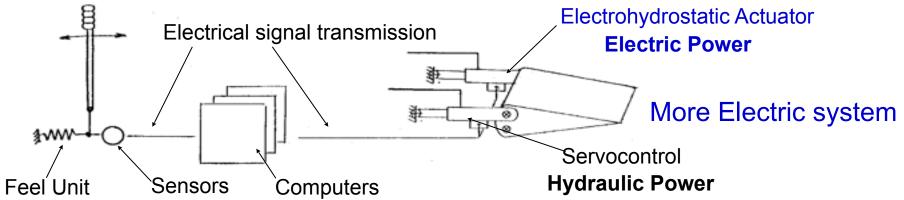




# New generation, hybrid power sources, full FBW systems

Airbus "More Electric" systems





















## New generation, hybrid power sources, full FBW systems

# Airbus "More Electric" systems Basic concept

- Three independent power sources are required to make the complete loss of the flight control actuation system Extremely Improbable
- Large transport category airplanes are currently fitted with three independent hydraulic systems plus two independent electric systems, which makes a total of five independent power sources
- The basic idea is that one hydraulic system can be eliminated and replaced by a set of electrically powered actuators with no detrimental impact to the probability of loosing the flight control actuation system.















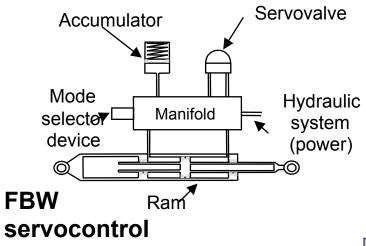


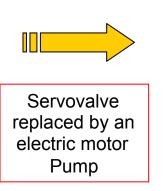
**Actuator** 

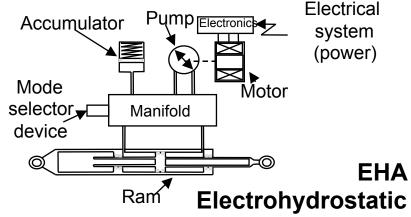
# GEA Tianjin / 中国民航大学中欧航空工程师学院

# New generation, hybrid power sources, full FBW systems

■ Key component, The Electrohydrostatic Actuator - EHA



















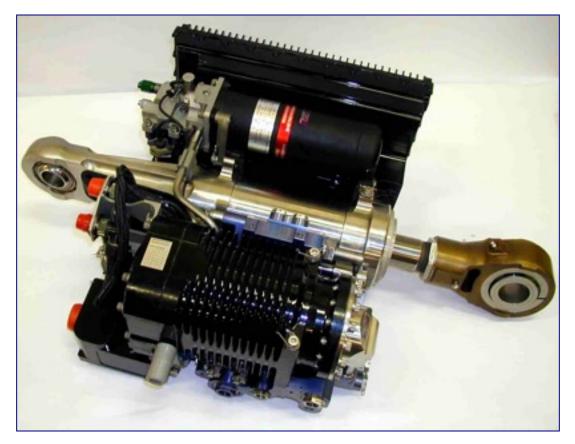






## New generation, hybrid power sources, full FBW systems

■ Key component, The Electrohydrostatic Actuator - EHA













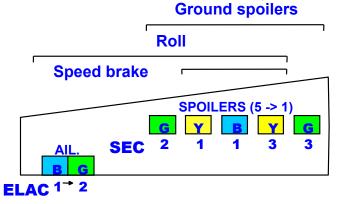


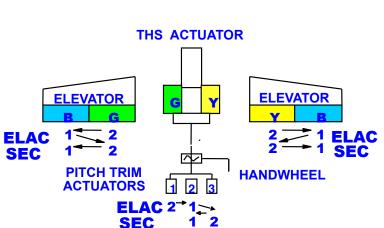


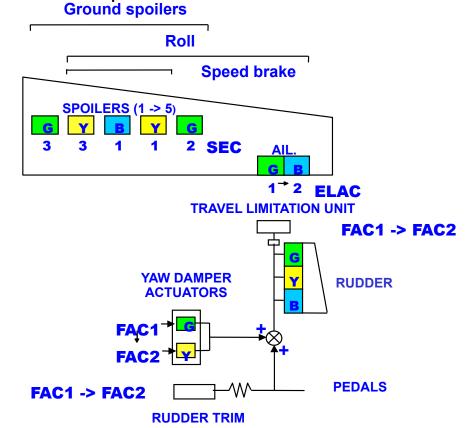


# First generation Fly by Wire systems

A320 Power Source, Actuator and Computer Distribution







**ELAC**: Elevator and Aileron Computer **SEC**: Spoiler and Elevator computer **FAC**: Flight Augmentation Computer







**ACTUATOR** 



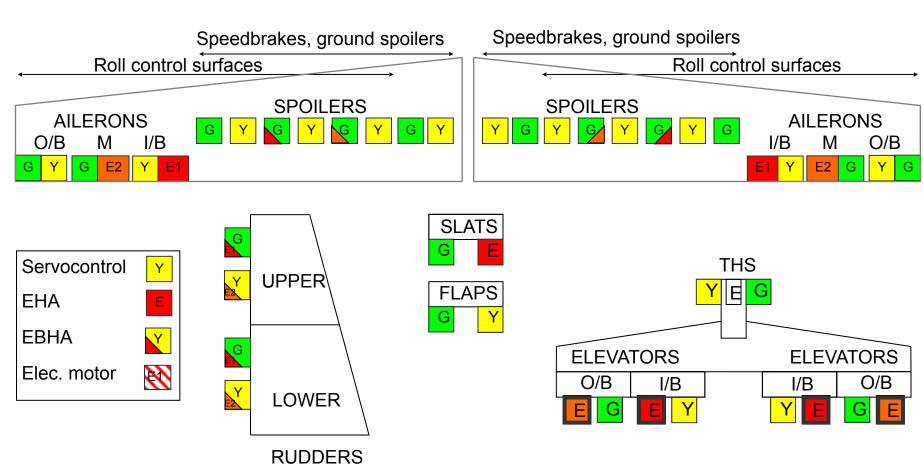








A380 « 2H/2E » Power Source and Actuator Distribution



Electrically Powered Actuators in stand-by positions











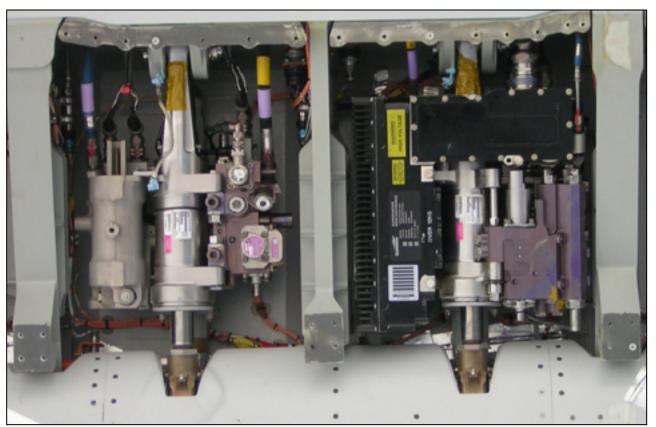






## New generation, hybrid power sources, full FBW systems

■ EHA and servocontrol installation, A380 aileron











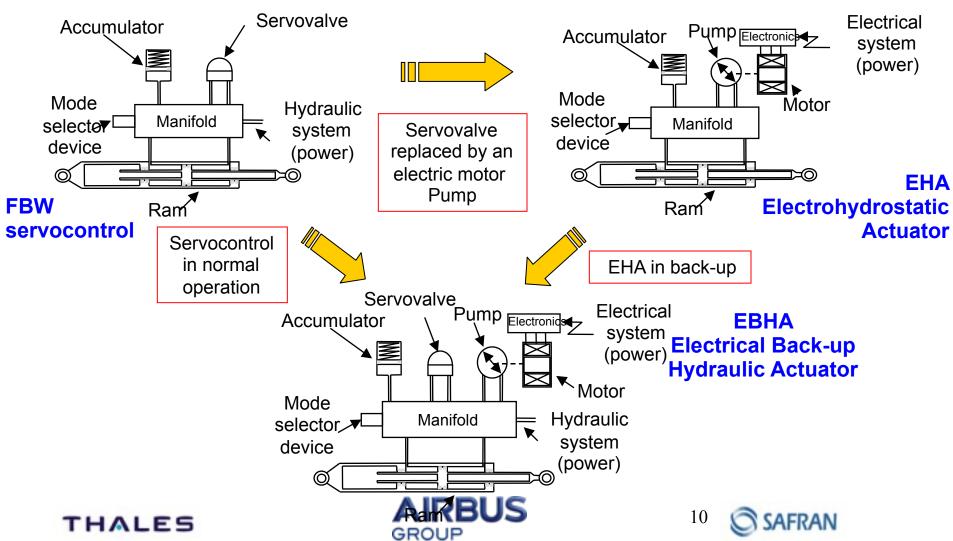








Electrical Back-up Hydraulic Actuator - EBHA







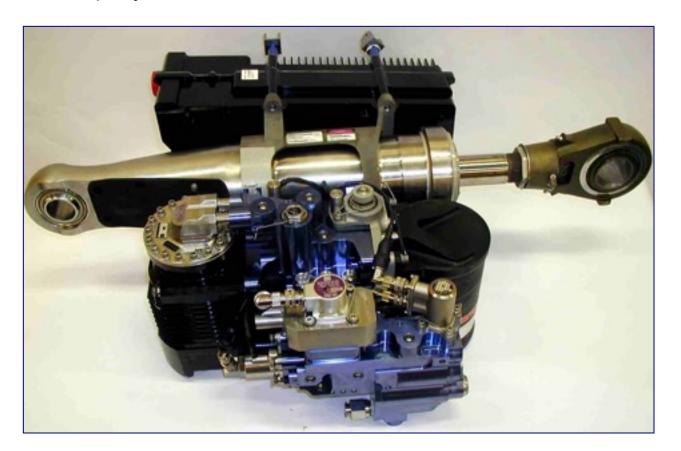






# New generation, hybrid power sources, full FBW systems

Electrical Back-up Hydraulic Actuator – EBHA, A380 rudder EBHA













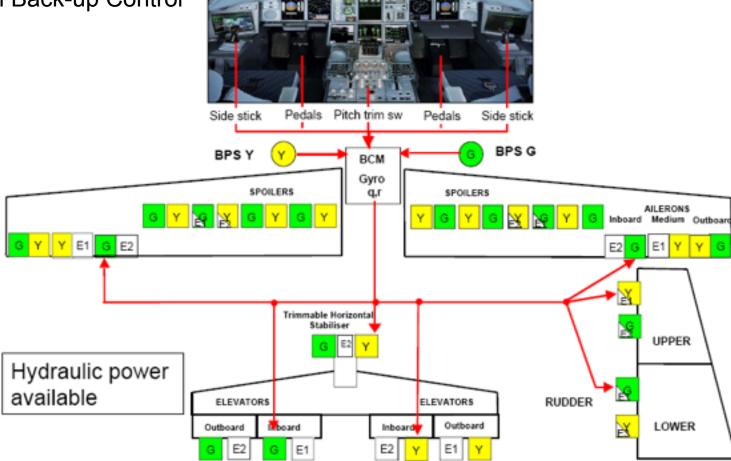






## New generation, hybrid power sources, full FBW systems





















## New generation, hybrid power sources, full FBW systems

- Electrical Back-up Control
- The objective of the electrical back up is to achieve control and stability in case of temporary loss of all PRIM and SEC or of their electrical supply.
- The Back-up Control Module BCM becomes automatically operative in absence of inhibition signal from: PRIM1, PRIM3, SEC1 and SEC3.
- BCM is electrically powered by the BPS Back-up Power Supply
- The BCM is able to damp the dutch roll using internal yaw gyrometer and to stabilise the pitch axis using internal pitch gyrometer.
- When the BCM is activated, it controls all axes.
- It is impossible to activate the BCM axis per axis.

















## New generation, hybrid power sources, full FBW systems

- Overall benefits of the 2H / 2E arrangement
- •Power source redundancy:
  - •Increased number of power sources: 2 electrical systems replace 1 hydraulic system
  - Dissimilarity of power sources
- •Survivability / robustness:
  - Make possible an efficient segregation of power distribution routes.
  - •Electrical isolation of a part of the system easier/more efficient than hydraulic fuse
  - •Reconfiguration in case of power generation failure more flexible
- Performance
  - •Reduction of electrical power required: EHA efficiency better than EMP/servovalve
  - Weight saving.
- Maintenance
  - Elimination of potential leakage sources

















#### **Actuation**

- End of session
- Thank You!
- Any question?







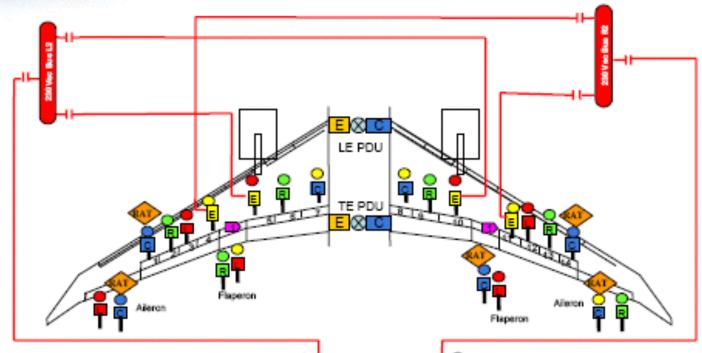






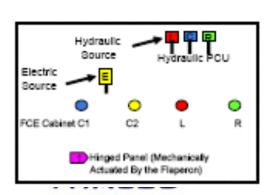


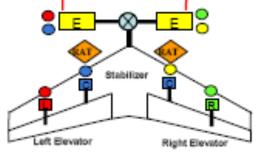
# ■The B787 approach

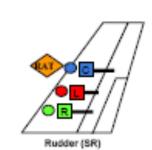


2 EMA spoiler Pairs

Electric horizontal stabilizer actuator



















# **A350 Flight Control Architecture**

