### Objectives of the WorkShop

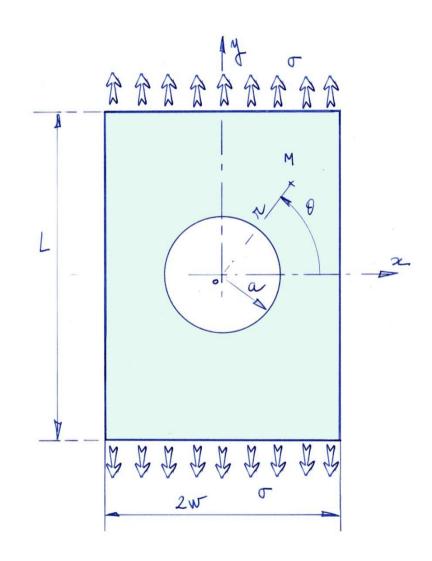
- Learn to use the elements of membrane,
- Learn to create a MeshSeed,
- Learn to create an accurate meshing,
- Learn to read the BDF File,
- Learn to use the F06 File,
- Learn to create images with the results,
- Learn to verify the accuracy of the modelling,
- Learn to create images and curves with the results.

### Structure studied

### A plate with a hole:

- Thickness e = 1 mm
- Lenght L = 600 mm
- Width 2W = 200 mm
- Drilling a = 50 mm

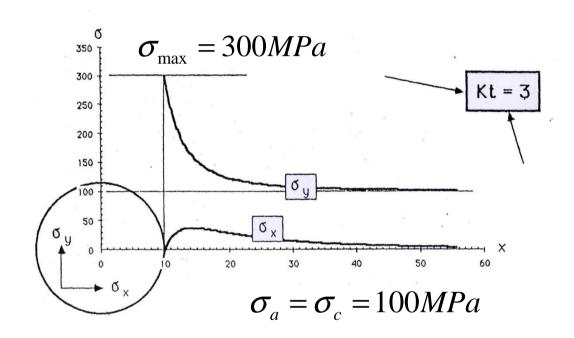
The load is a uniform tensile stress of 100 Mpa on two opposite edges



# Particular Case for an infinite plate (r<<W)

$$\sigma_{y} = \frac{\sigma_{a}}{2} \left[ 2 + \left( \frac{a}{x} \right)^{2} + 3 \left( \frac{a}{x} \right)^{4} \right]$$

$$\sigma_{x} = \frac{3\sigma_{a}}{2} \left[ \left( \frac{a}{x} \right)^{2} - \left( \frac{a}{x} \right)^{4} \right]$$



### Infinite Plate: Solution in cylindrical axis

If the point M of the plate is described in a cylindrical axis system with the two coordinates r and  $\theta$ , then the solution is :

$$\sigma_{rr} = \frac{\sigma}{2} \left( 1 - \frac{a^2}{r^2} \right) + \frac{\sigma}{2} \left( 1 - 4 \frac{a^2}{r^2} + 3 \frac{a^4}{r^4} \right) \cos 2\theta$$

$$\sigma_{\theta\theta} = \frac{\sigma}{2} \left( 1 + \frac{a^2}{r^2} \right) - \frac{\sigma}{2} \left( 1 + 3 \frac{a^4}{r^4} \right) \cos 2\theta$$

$$\sigma_{r\theta} = -\frac{\sigma}{2} \left( 1 + 2 \frac{a^2}{r^2} - 3 \frac{a^4}{r^4} \right) \sin 2\theta$$

### Case of a finite Plate

If the plate is a real plate with finite dimensions, as the one studied in this workshop, the results described below are accurates.

$$\sigma_{net} = \sigma_{\infty} \frac{w}{w - r} = \sigma_{\infty} \frac{100}{100 - 50} = 100 \sigma_{\infty} = 200 \quad MPa$$

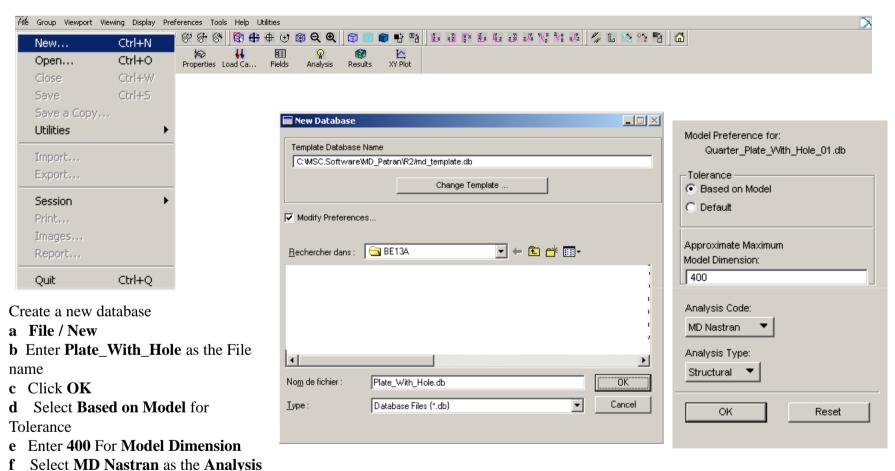
$$\sigma^{Max} = K_t \sigma_{net}$$

$$with$$

$$K_t = 2 + \left(1 - \frac{r}{w}\right)^3 = 2 + \left(1 - \frac{50}{100}\right)^3 = 2.125$$

$$\sigma^{Max} = 425 \quad MPa$$

### Step 1: Creation of a New Database

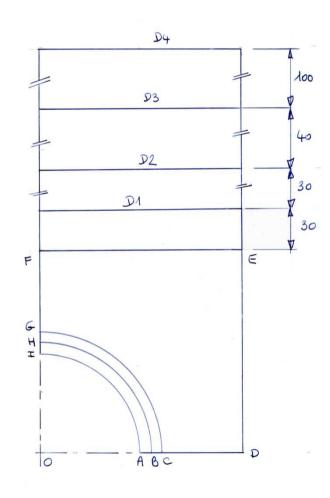


- **g** Select **Structural** as the Analysis Type
- h Click **OK**

Code

# Points O, A to I and Lines D1 to D4

Description of the geometry of a quarter of the plate.



### **Step 1: Creation of Geometry - Points**

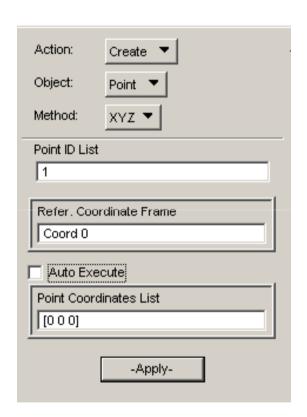


#### Create the first point:

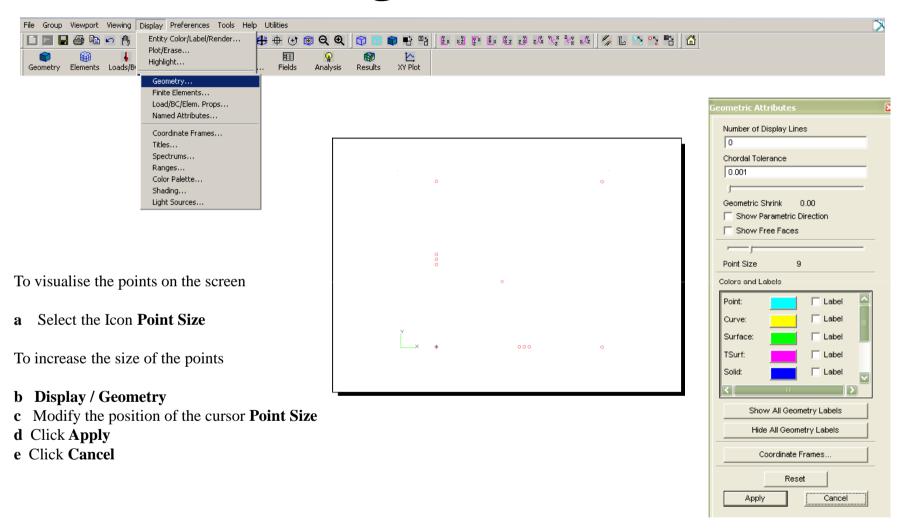
- a Geometry: Create / Point / XYZ
- **b** Unselect **Auto Execute**
- **c** Enter [0,0,0] for the **Point Coordinate List**.
- d Click Apply

Create the other points (A....I) with the same method.

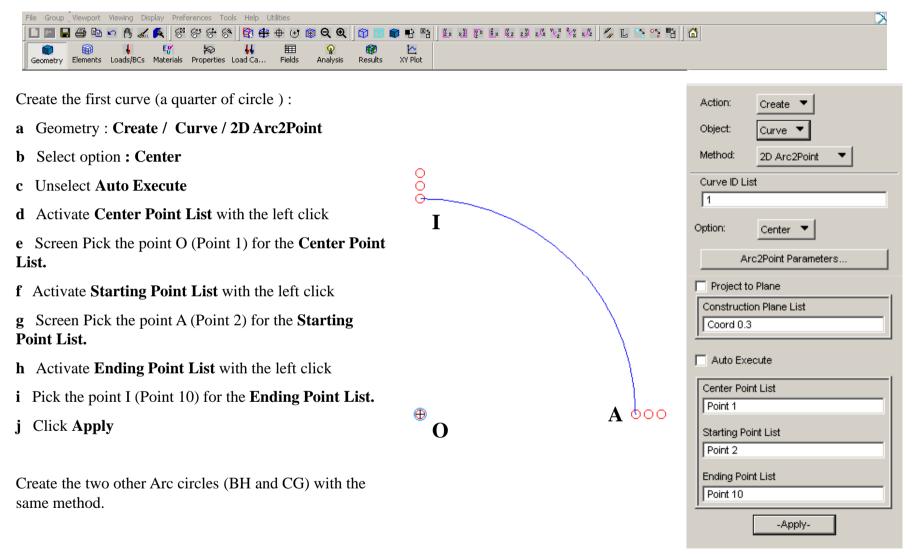
Point	х	у	z	Nastran Id
Α	50	0	0	1
В	53	0	0	3
С	56	0	0	4
D	100	0	0	5
E	100	100	0	6
F	0	100	0	7
G	0	56	0	8
Н	0	53	0	9
I	0	50	0	10



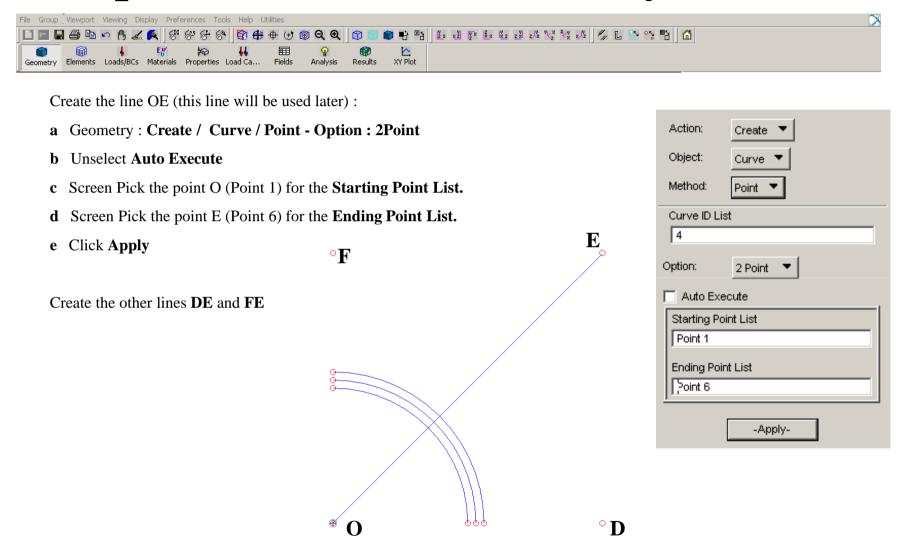
### Viewing of the Points



### **Step 1: Creation of Geometry – Curves**



### **Step 1: Creation of Geometry – Curves**



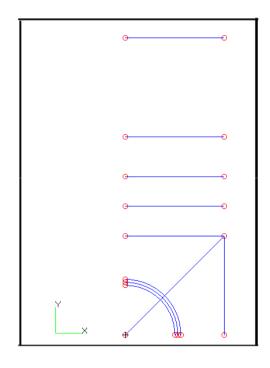
### **Step 1: Creation of Geometry – Curves**

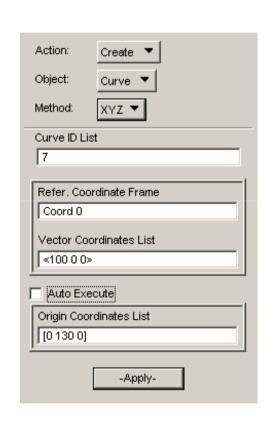


#### Create the line D1:

- a Geometry: Create / Curve / XYZ
- **b** Unselect **Auto Execute**
- c Enter [0,130,0], the coordinates of the beginning point of the line in **Origin Coordinates List**
- **d** Enter [100,0,0], the projected lengths of the line in **Vector Coordinates List**
- e Click Apply

Create the other lines D2, D3 & D4

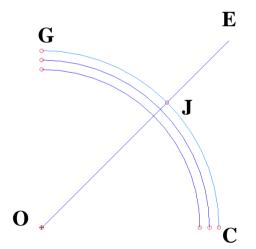


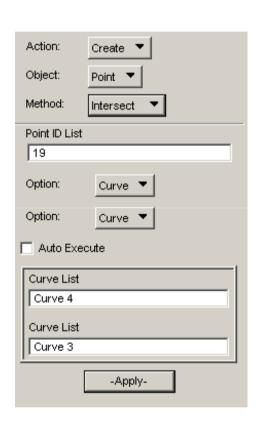




#### Create the point J:

- a Geometry: Create / Point / Intersect -
- **b** Select Option 1 : Curve Option 2 : Curve
- c Unselect Auto Execute
- **d** Pick the Curve 4 (Line OE) in **Curve List**.
- e Pick the Curve 3 (Cercle CG) in Curve List.
- g Click Apply

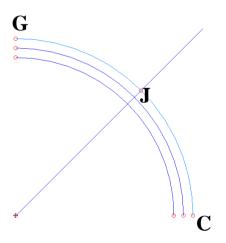


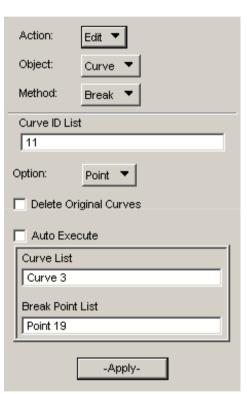




#### Break the circle on the point J:

- a Geometry: Edit / Curve / Break
- **b** Select Option : **Point**
- c Unselect Delete Original Curve
- d Unselect Auto Execute
- e Screen Pick the Curve 3 (Cercle CG) in Curve List.
- f Screen Pick the Point J in Break Point List.
- g Click Apply



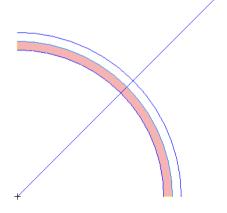


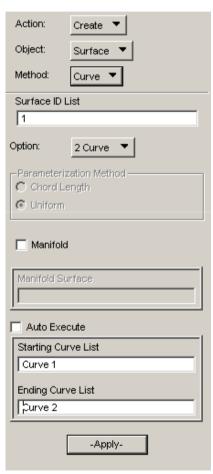


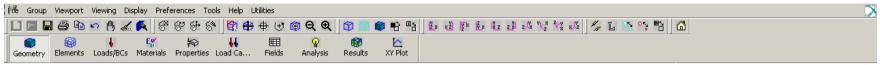
Create Surfaces with 2 curvilinear boundaries:

- a Geometry: Create / Surface / Curve
- **b** Select Option : **2 Curves**
- c Unselect Auto Execute
- d Screen Pick the Curve 1 (Cercle AI) in Starting Curve List.
- e Screen Pick the Curve 2 (Cercle BH) in Ending Curve List.
- f Click Apply
- **g** Use the shading Icon to visualize the surface

Create the surface 2 with curvilinear boundaries



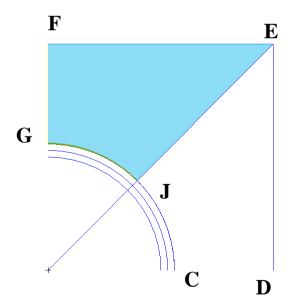


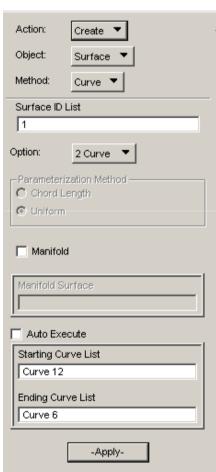


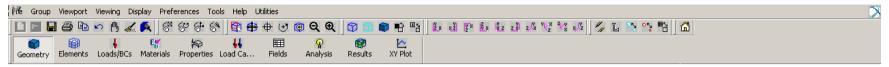
Create 2 New Surfaces as indicated below:

- a Geometry: Create / Surface / Curve
- **b** Select Option : 2 Curves
- c Unselect Auto Execute
- **d** Pick the Curve 12 (Circle JG) in **Starting Curve List**.
- e Pick the Curve 6 (Line EF) in **Ending Curve List**.
- f Click Apply

Create the surface 4 with Circle CJ and line DE







Create 4 New Surfaces with the different lines as indicated below:

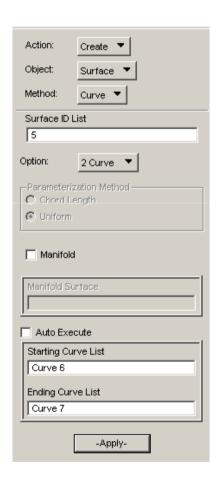
- a Geometry: Create / Surface / Curve
- **b** Select Option : **2 Curves**
- c Unselect Auto Execute
- d Screen Pick the Curve 6 (Line EF) in Starting Curve List.
- e Screen Pick the Curve 7 (Line D1) in Ending Curve List.
- f Click Apply

Create the other surfaces:

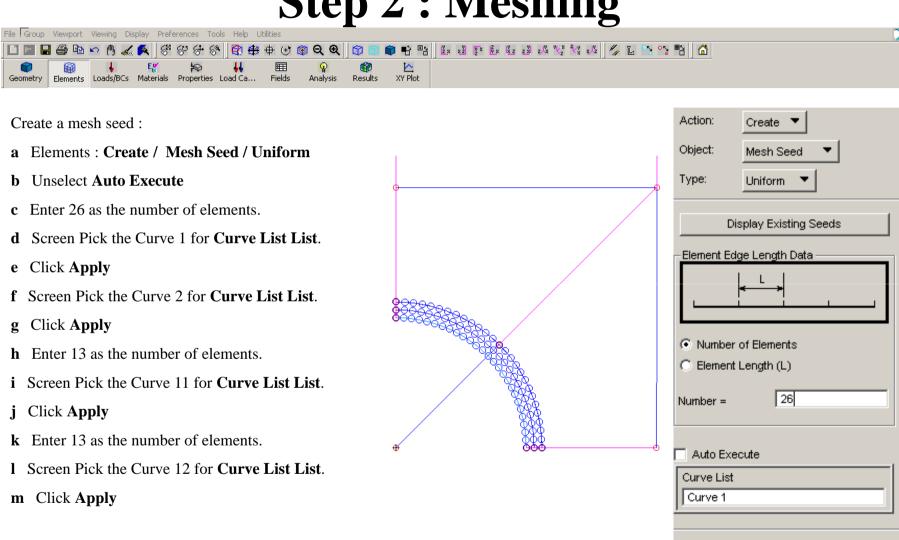
Surface 6 with D1 & D2

Surface 7 with D2 & D3

Surface 8 with D3 &D4



### Step 2 : Meshing



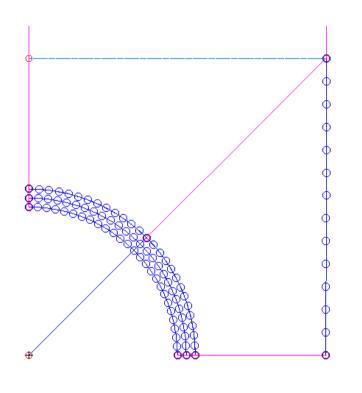
-Apply-

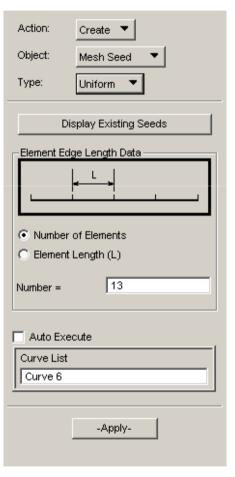
### Step 2: Meshing



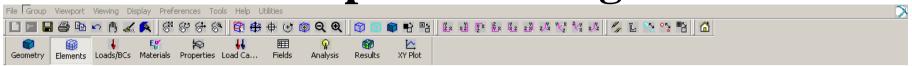
#### Continue the mesh seed:

- a Elements: Create / Mesh Seed / Uniform
- **b** Enter 13 as the number of elements.
- c Screen Pick the Curve 5 for Curve List
- d Click Apply
- e Screen Pick the Curve 6 for Curve List
- f Click Apply
- **g** Enter 11 as the number of elements.
- h Screen Pick the Curve 7 for Curve List
- i Click Apply
- **j** Enter 9 as the number of elements.
- k Screen Pick the Curve 8 for Curve List
- l Click Apply
- **m** Enter 7 as the number of elements.
- n Screen Pick the Curves 9 & 10 for Curve List
- o Click Apply



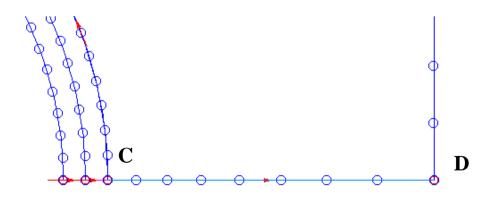


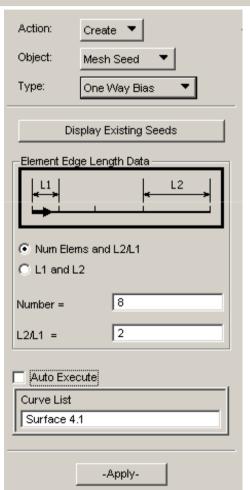
### Step 2: Meshing



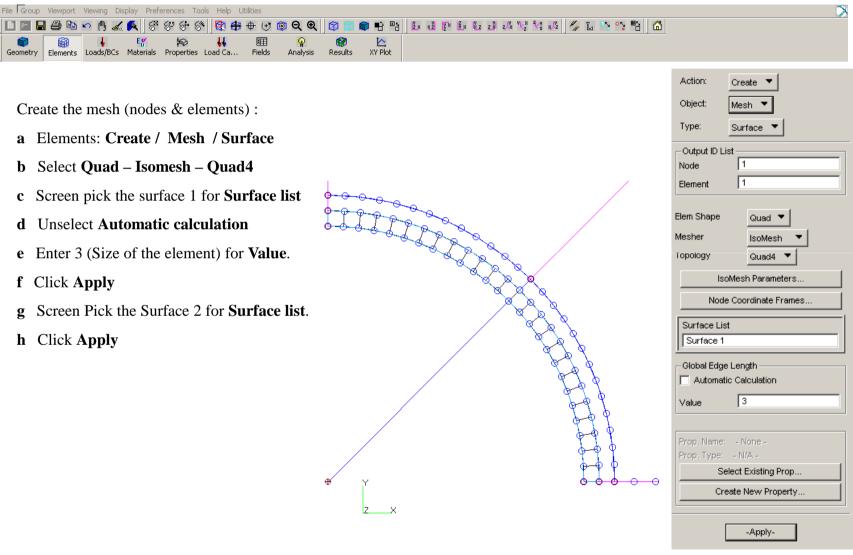
#### Continue the mesh seed:

- a Elements: Create / Mesh Seed / One Way Bias
- **b** Unselect Auto execute
- **c** Enter 8 as the number of elements.
- **d** Enter 2 (or 0.5) as the ratio **L2/L1**
- e Screen Pick the Curve between C & D for Curve List
- f Click Apply
- g Screen Pick the Curve between G & F for Curve List
- h Click Apply

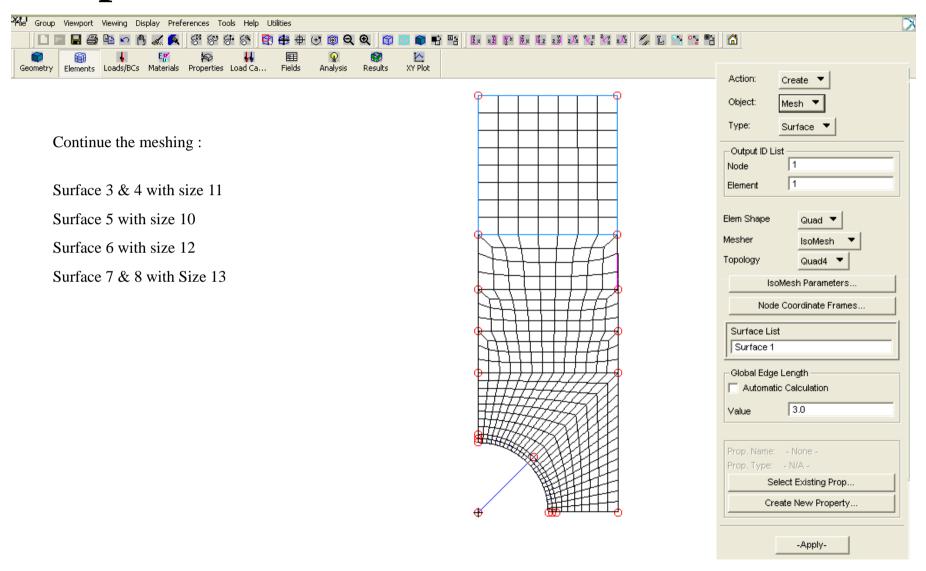




### Step 2: Meshing



### Step 3: Creation of the Nodes & Elements

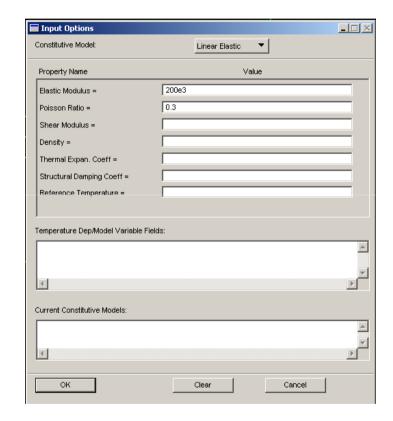


### **Step 4 Create material**



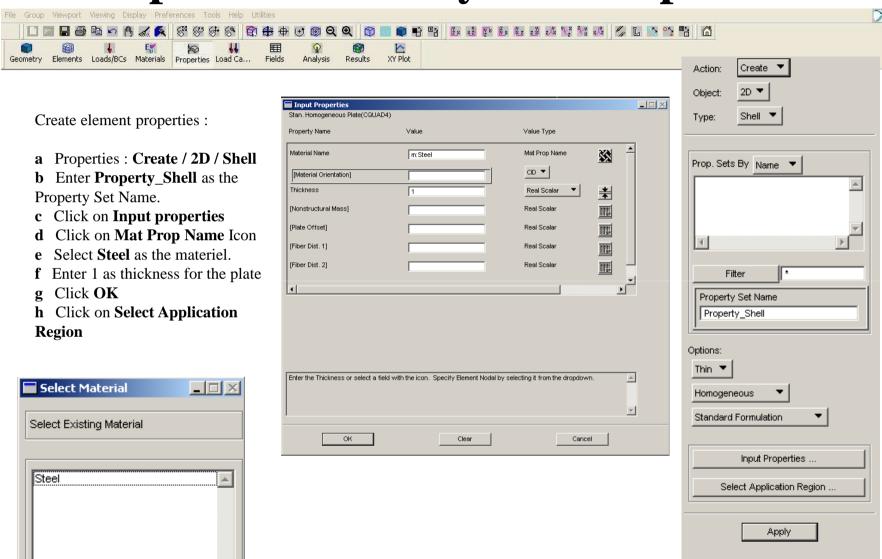
#### Create an isotropic material:

- a Material : Create / Isotropic / Manual Input
- **b** Enter **Steel** as the Material Name
- c Click on **Input properties**
- d Enter 200e3 for Elastic Modulus
- e Enter 0.3 for Poisson's Ratio
- f Click OK
- g Click Apply
- **h** Verify that the material has been created in the field **Existing Materials**

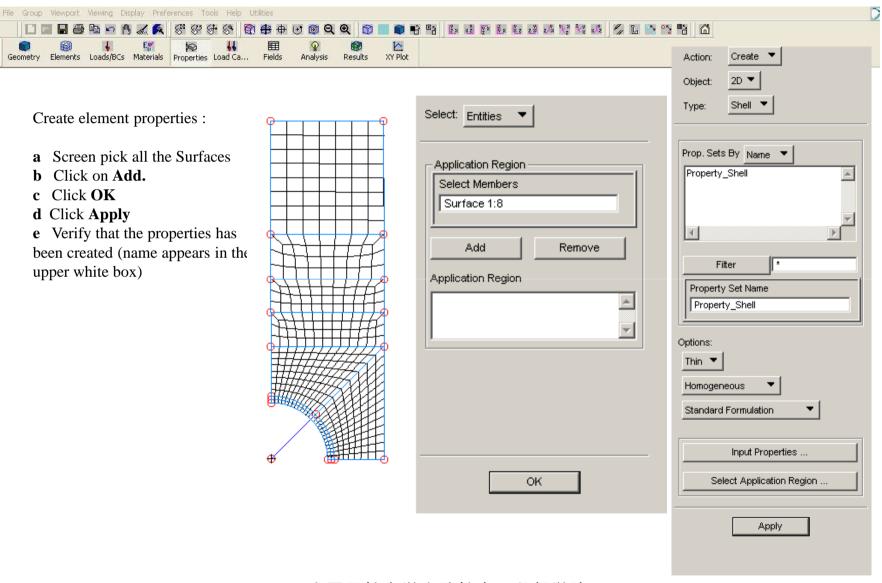




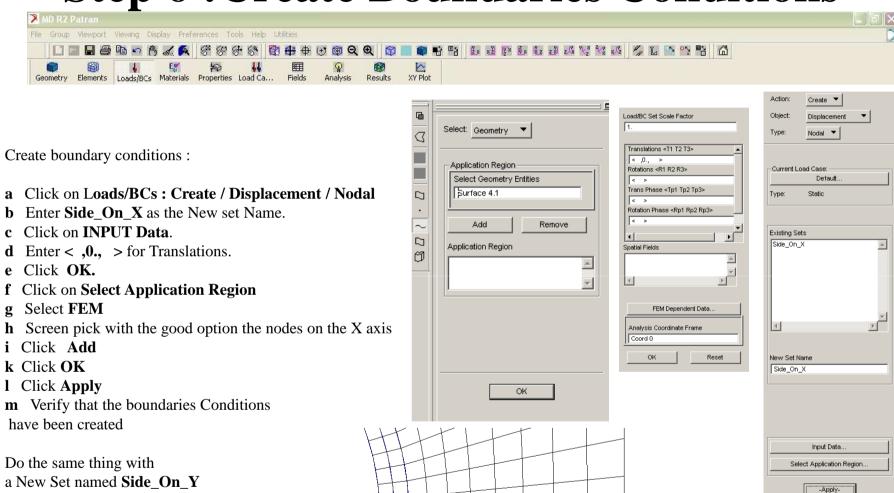
### **Step 5 : Create Physical Properties**



# **Step 5: Create Physical Properties**



### **Step 6: Create Boundaries Conditions**

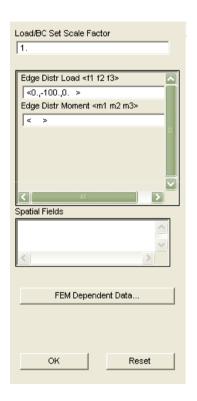


### Step 7: Create Load



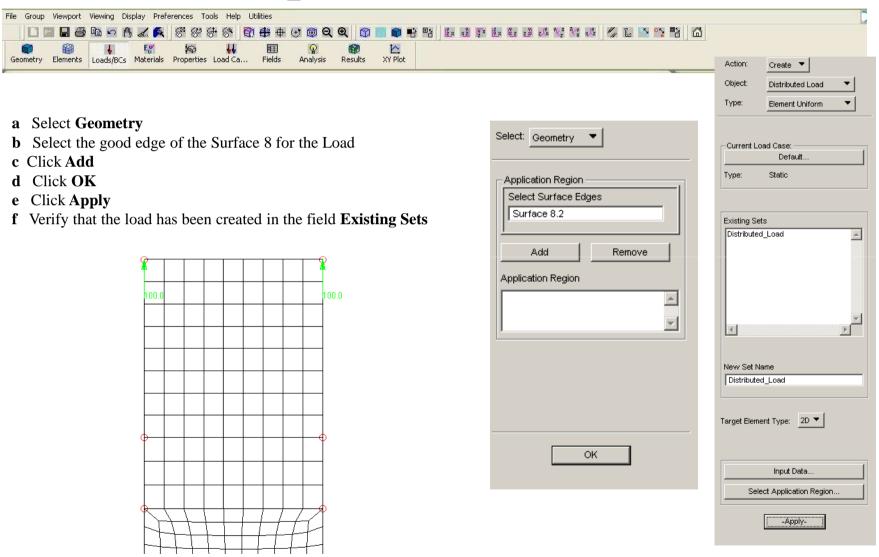
#### Create a constant uniform load:

- a Loads/BCs: Create / Distributed Load / Element Uniform
- **b** Enter **Distributed** Load as New Set Name
- c Select 2D with icon Target Element Type
- d Click on INPUT Data.
- e Enter <0., -100., 0.> for components of the distributed load.
- f Click OK
- g Click on Select Application Region





### **Step 7: Create Load**

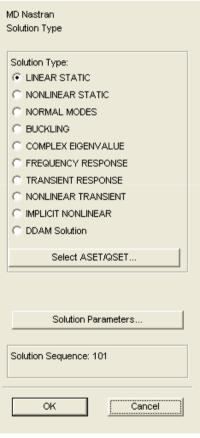


### Step 8: Run Nastran



#### Computation

- a Analysis: Analyse / Entire Model / Full Run
- **b** Select **Solution Type**
- c Select Linear Static
- d Click OK
- f Click Apply



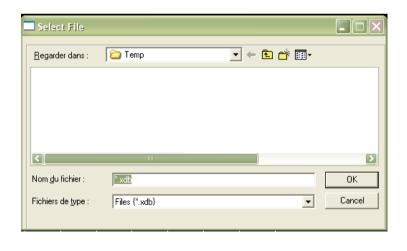


### Step 8: Load Nastran Results in Patran



#### Load the results file of NASTRAN

- a Analysis: Access Results / Attach XDB / Result Entities
- **b** Click on **Select Results File**
- c Select the file results Plate\_With\_Hole.XDB
- d Click OK
- e Click Apply





### Analysys of Files \*.F06 and \*.BDF

- Open the file Plate\_With\_Hole.F06
- Verify the run and the results
- Open the file Plate\_With\_Hole.XDB
- Verify the NASTRAN Input file

### Verification

The model doesn't perform !!

You have to find a solution to solve the problem !!

Modify the modelling and run it again.

### Where did the errors come from

- Verify if the equivalence has been made,
- Verify if there are rigid body movements, and removed them if necessary,
- Verify if the properties of the elements are the good one,
- Verify if the load has been applied,
- Verify if the boundary conditions are sufficient for the elements choosed for the modelling.

### Equivalence

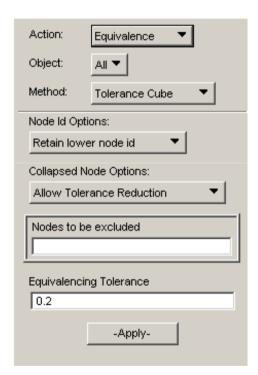


Never forget to equivalence the nodes of your model:

- a Elements: Equivalence / All / Tolerance Cube
- b Click Apply

Run the model again, Load the file results \*.XDB, and verify the

\*.F06 and \*.BDF files



### Rigid Body movement

Remove all the rigid body movements:

If you have a plane problem, like the one studied, don't forget that NASTRAN solve a 3 dimensional problem

# **Boundary Conditions**

### Don't forget the degrees of freedom of the element choosed:

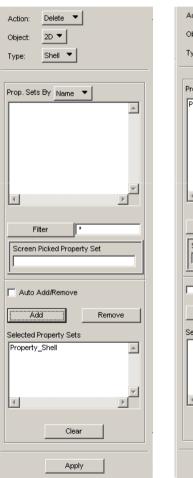
#	Membrane Element	Shell Element
Local Axis of the elemen	nt 3 DOF	5 DOF
Global Axis	3 DOF	6 DOF
DOF for Translation	TX, TY, TZ	TX, TY, TZ
DOF for Rotation	RX, RY, RZ	RX, RY, RZ

### **Shell and Membrane Elements**



When the model will be good, change the Shell elements by Membrane elements and compare the boundary conditions needed for each type of elements. Before creating new properties (see previous pages) you must erase the old, like this:

- a Properties: Delate / 2D / Shell
- **b** Select the properties by the name in the good box.
- c Click ADD
- **d** Verify the name properties is in the box Selected property Sets
- e Click Apply
- f Create Membrane properties as described in previous pages





### Analysis of the results

Before creating images, graph and data files to store the results you must:

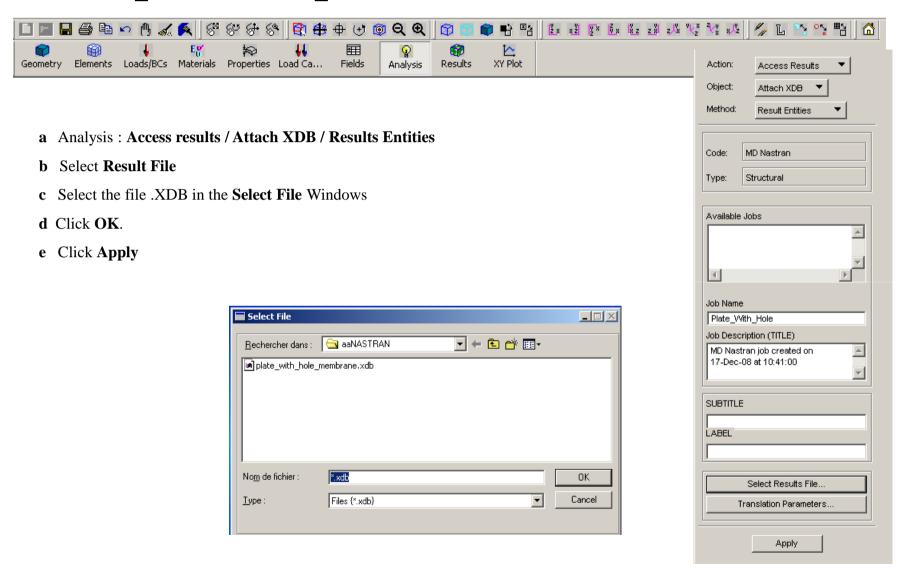
- \* Open the BDF file to verify if you modelling seem good
- \* Open the F06 file to verify if your model has been computed correctly, with the good accuracy.

After that you can import a specific output file inside Patran to analyse the results. Two different output files can be generated by NASTRAN:

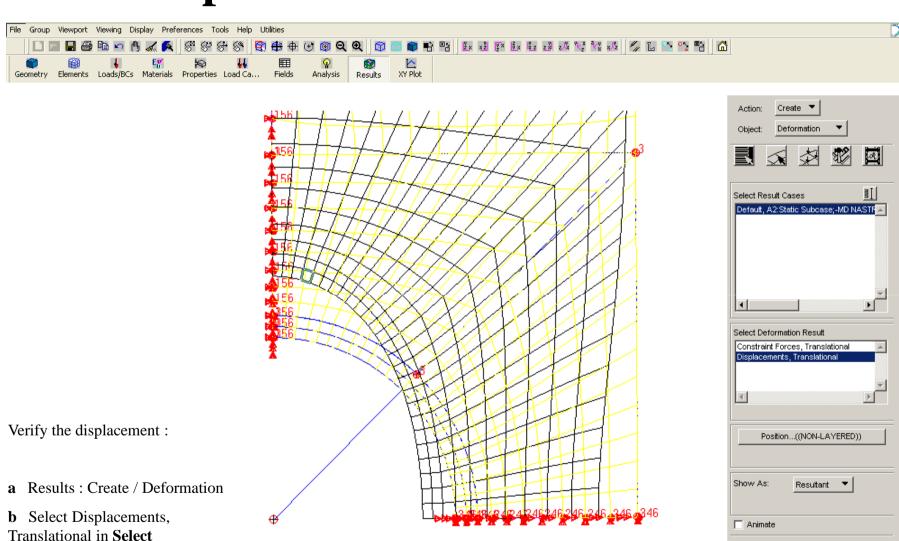
- \*. XDB output file (used at ISAE)
- \*.OP2

To import the output file in Patran see the following slide

### Import Output BDF file in PATRAN



## Step 9: Results - Deformation



c Click Apply

**Deformation Result** 

Reset

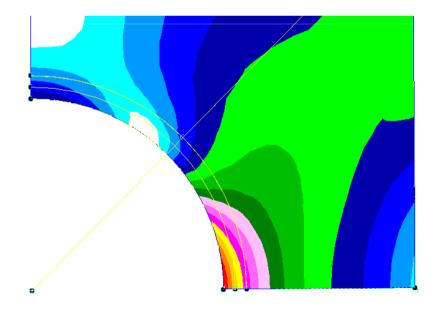
Apply

## Step 9: Results - Von Mises



Verify the Von Mises' Criteria:

- a Results : Create / Fringe
- **b** Select Stress Tensor in **Select Fringe Result**
- c Select Von Mises fo Quantity
- d Click Apply



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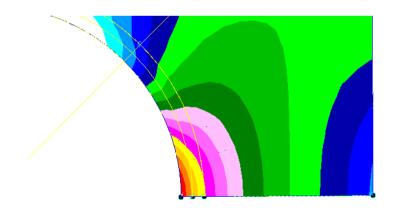
41

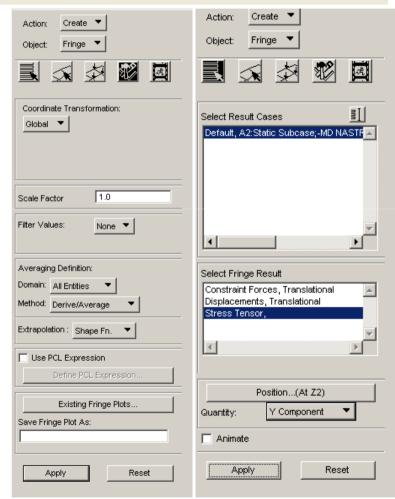
# Step 9 : Results - $\sigma_{x}$



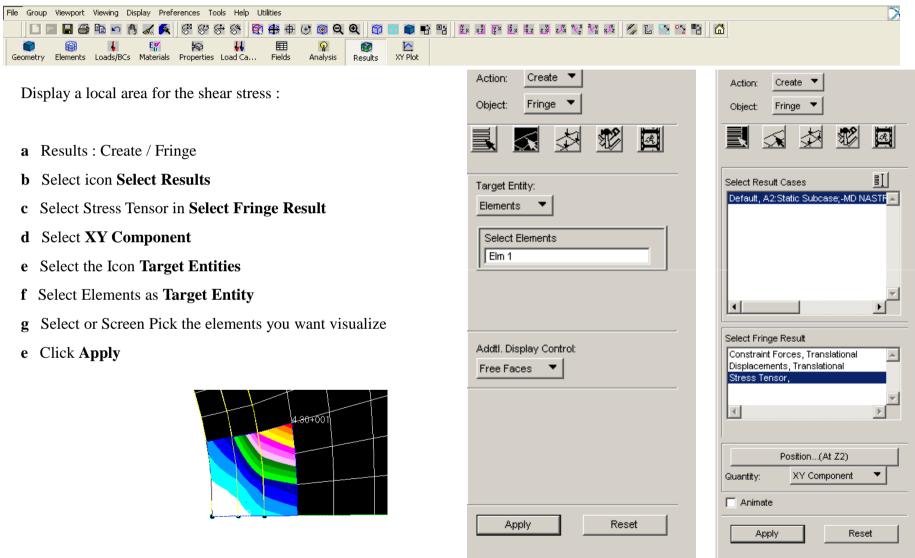
### Verify the Von Mises' Criteria:

- a Results : Create / Fringe
- **b** Select Stress Tensor in **Select Fringe Result**
- c Select Y Component for Quantity
- d Select the Icon Plot Options
- e Select Global as Coordinate Transformation
- d Click Apply





# Step 9: Local Results – $\sigma_x$

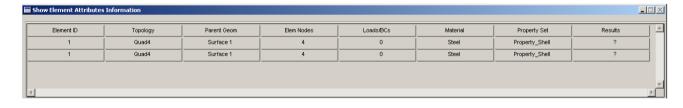


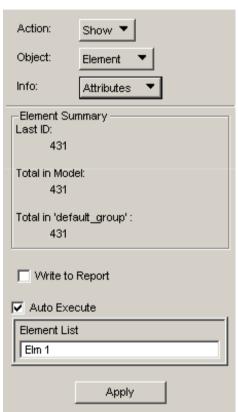
### Results: Stress in an Element



### Identify one (or More) element(s):

- a Elements: Show / Element / Attributes
- **b** Screen Select the element near the hole
- c Click Apply
- **d** Read the identification number of the element
- e Open the file job.F06
- **f** Find the data of the element(s)
- **g** Analyse the results





# Results File Job.F06 (Point A)

ELEMENT		FIBER	STRESSES	IN ELEMENT CO	ORD SYSTEM	PRINCIP	AL STRESSES (2	ERO SHEAR)	
ID	GRID-ID	DISTANCE	NORMAL-X	NORMAL-Y	SHEAR-XY	ANGLE	MAJOR	MINOR	VON MISES
1	CEN/4	-5.000000E-01	4.013723E+02	1.076666E+01	1.475449E+00	0.2164	4.013778E+02	1.076109E+01	3.961069E+02
		5.000000E-01	4.013723E+02	1.076666E+01	1.475449E+00	0.2164	4.013778E+02	1.076109E+01	3.961069E+02
	1	-5.000000E-01	4.366468E+02	1.044953E+01	1.475600E+00	0.1984	4.366519E+02	1.044442E+01	4.315245E+02
		5.000000E-01	4.366468E+02	1.044953E+01	1.475600E+00	0.1984	4.366519E+02	1.044442E+01	4.315245E+02
	2	-5.000000E-01	4.366471E+02	1.108379E+01	1.475603E+00	0.1987	4.366523E+02	1.107867E+01	4.312196E+02
		5.000000E-01	4.366471E+02	1.108379E+01	1.475603E+00	0.1987	4.366523E+02	1.107867E+01	4.312196E+02
	29	-5.000000E-01	3.667758E+02	1.106584E+01	1.475300E+00	0.2376	3.667819E+02	1.105972E+01	3.613790E+02
		5.000000E-01	3.667758E+02	1.106584E+01	1.475300E+00	0.2376	3.667819E+02	1.105972E+01	3.613790E+02
	28	-5.000000E-01	3.667761E+02	1.046748E+01	1.475297E+00	0.2372	3.667822E+02	1.046138E+01	3.616650E+02
		5.000000E-01	3.667761E+02	1.046748E+01	1.475297E+00	0.2372	3.667822E+02	1.046138E+01	3.616650E+02

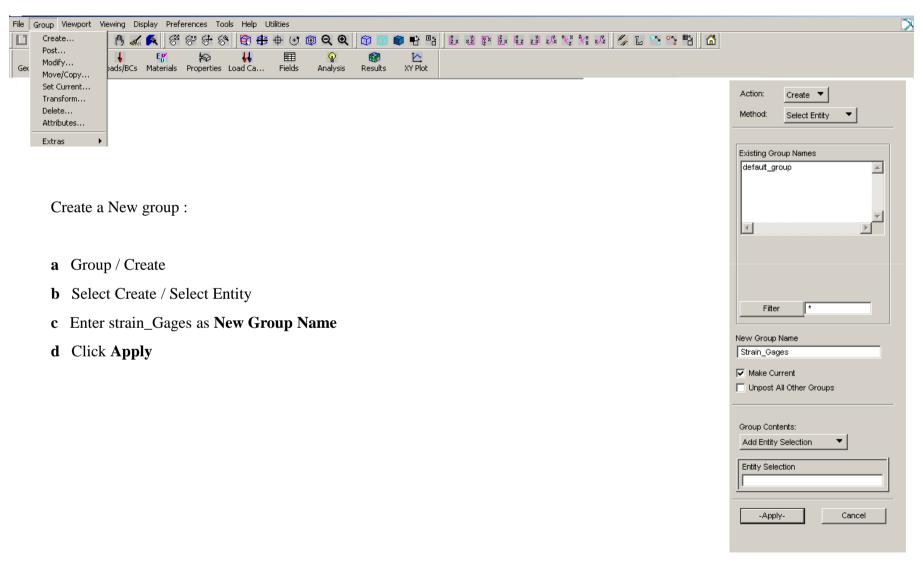
## Results File Job.F06 (Point A) (Excell)

ELEMENT		FIBER	STRESSES IN ELEMENT COOR SYSTEM			PRINCIPAL STRESES (ZERO SHEAR)			
ID	GRID-ID	DISTANCE	NORMAL-X	NORMAL-Y	SHEAR-XY	ANGLE	MAJOR	MINOR	VON MISES
1	CEN/4	-5,00E-01	401,4	11	1	0,2	401	11	396
		5,00E-01	401	11	1	0,2	401	11	396
	1	-5,00E-01	437	10	1	0,2	437	10	432
		5,00E-01	437	10	1	0,2	437	10	432
	2	-5,00E-01	437	11	1	0,2	437	11	431
		5,00E-01	437	11	1	0,2	437	11	431
	29	-5,00E-01	367	11	1	0,2	367	11	361
		5,00E-01	367	11	1	0,2	367	11	361
		_							
	28	-5,00E-01	367	10	1	0,2	367	10	362
		5,00E-01	367	10	1	0,2	367	10	362

### Strain gauges on the edge of the hole

- Create a new group,
- Create ROD elements on the edge of the hole,
- Create properties for these ROD elements,
- Run a new Job,
- Load the results and analyse the results in the ROD elements,
- Draw the curve of  $\sigma_{\theta}$  on the edge of the hole

# Creation of a New Group

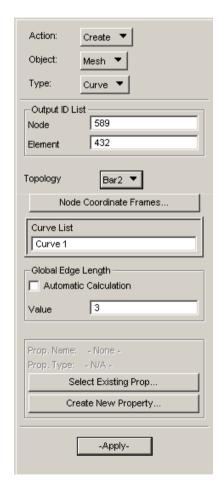


### **Creation of a New Mesh**



#### Create a new Mesh:

- a Elements: Create / Mesh / Curve
- **b** Select **Bar2** as **Topology**
- c Unselect Automatic Calculation
- **d** Enter 3 as Value
- e Verify that 26 elements have been created

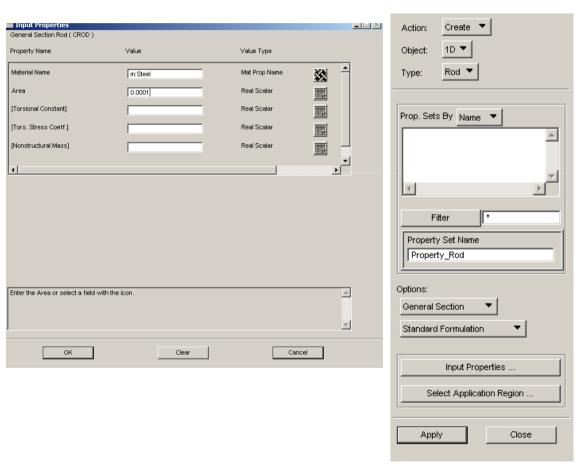


## **Create New Properties**

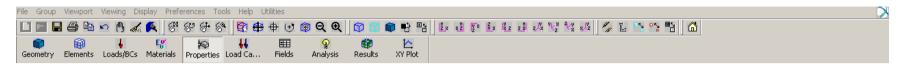


### Create element properties:

- a Properties: Create / 1D / Rod
- b Enter Property\_Rod as the name ofProperty Set Name
- c Select Input Properties
- d Select Steel as Material
- e Enter 0.0001 for Area
- f Click OK
- g Select Application Region

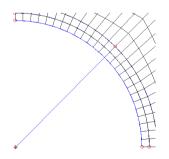


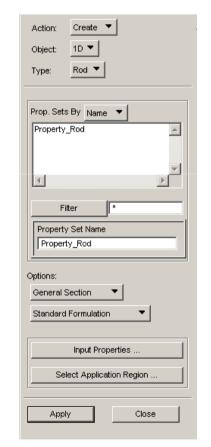
## **Create New Properties**

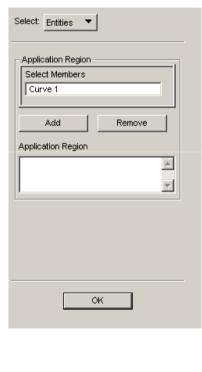


### Create element properties:

- a Activate the Field Select Member
- **b** Screen Select the Curve 1
- b Click ADD
- c Click OK
- d Click Apply
- e Verify that the Property has been created







## **Results: Creation of images**

- Create a Cylindrical axis system to analyse the stresses on the edge of the hole,
- Create an image of  $\sigma_{\theta}$  near the hole,
- Create an image of  $\sigma_r$  near the hole,
- Create an image of  $\tau_{r\theta}$  near the hole,
- Refine the mesh near the hole to increase the accuracy

## **Results: Creation of images**

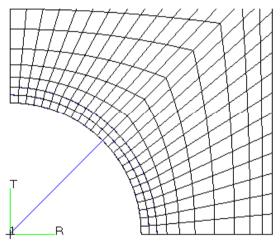
- Analyse the stresses in an element with:
- The local axis of the element,
- The global axis of the plate
- The cylindrical axis,

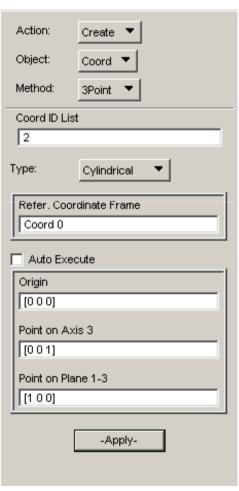
### Create a Cylindrical Axis System



There are several solutions to create a cylindrical axis system, for instance :

- a Geometry: Create / Curve / Coord / 3Point
- **b** Select **Cylindrical** as Type
- c With the **KeyBoard** enter the coordinates of the center of the axis system: [0 0 0]
- **d** With the **KeyBoard** enter the coordinates of a pronthe third axis: [0 0 1]
- **e** With the **KeyBoard** enter the coordinates of a po on the plane 1-3 : [ 1 0 0 ]
- f Click Apply
- g Verify that the axis system has been created



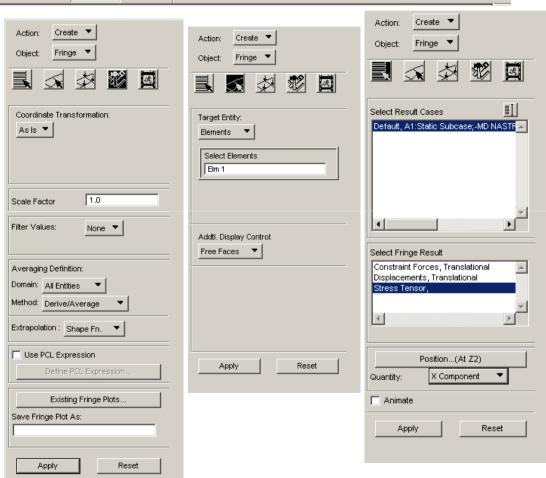


### Analysis in the local axis system



- a Results: Create / Fringe
- **b** Select the Icon **Select results**
- c Select Stress tensor in Field Fringe Result
- **d** Select the quantity X component for instance
- e Click Apply
- f Select the Icon Target Entities
- g Select Elements as Target Entity
- **h** Select the element on the model
- i Click Apply
- j Select the Icon Plot options
- k Select As is as Coordinate Transformation
- 1 Click Apply



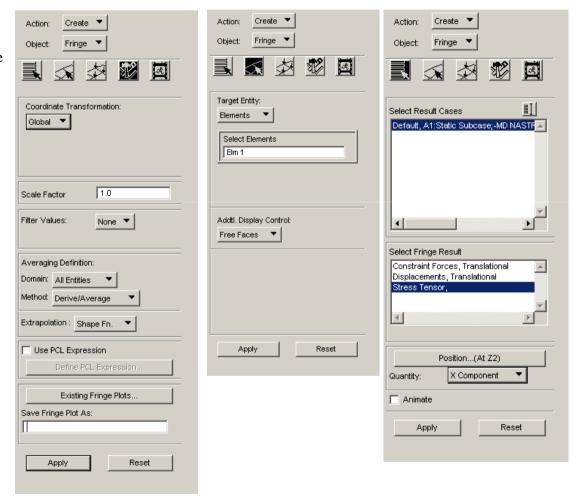


### Analysis in the global axis system



**REMARK:** The steps a to j have already been done in the previous slide. Go directly to the point k.

- a Results : Create / Fringe
- **b** Select the Icon **Select results**
- c Select Stress tensor in Field Fringe Result
- **d** Select the quantity X component for instance
- e Click Apply
- f Select the Icon Target Entities
- g Select Elements as Target Entity
- **h** Select the element on the model
- i Click Apply
- j Select the Icon Plot options
- k Select Global as Coordinate Transformation
- i Click Apply



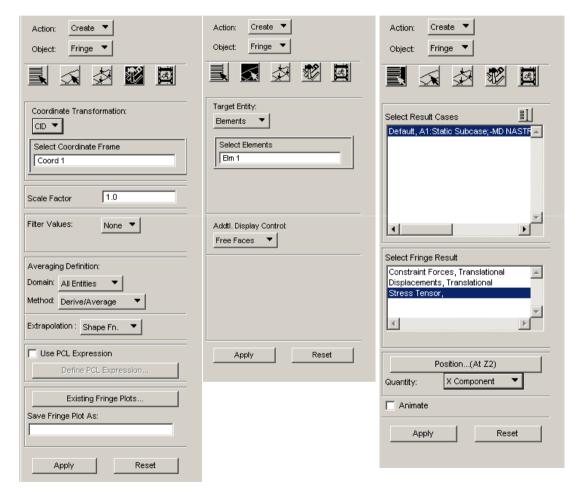
中国民航大学中欧航空工程师学院

### Analysis in the Cylindrical axis system

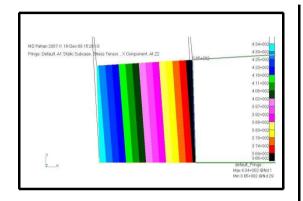


**REMARK:** The steps a to j have already been done in the previous slide. Go directly to the point k.

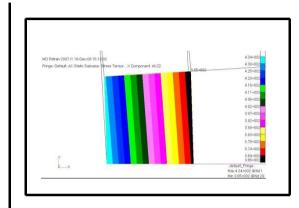
- a Results: Create / Fringe
- **b** Select the Icon **Select results**
- c Select Stress tensor in Field Fringe Result
- **d** Select the quantity X component for instance
- e Click Apply
- f Select the Icon Target Entities
- g Select Elements as Target Entity
- **h** Select the element on the model
- i Click Apply
- j Select the Icon Plot options
- k Select CID as Coordinate Transformation
- 1 Select the axis system on the model
- m Click Apply



### **Comparison of the different results**







**Local Axis System** 

**Global Axis System** 

**Cylindrical Axis System** 

## Local Axis of the shell elements

In the output file of Nastran all the results are calculated and printed in the local axis system of the element. It's very hard to visualize the field of stresses from these results. The best solution is to ask the computer output result in a specific axis system.

To visualize the local axis system of each element process like explained in the following pages

## Local axis of the Shell Elements

