



Structural Applications of Finite Elements

Chapter 2

Generalization of Finite Element Concepts

2018-09-01



Outline



- ❖ Principle of minimum potential energy
- ❖ Rayleigh-Ritz method
- ❖ Galerkin's method
- ❖ Structural element and structural system

Principle of Minimum Potential Energy

For conservative systems, of all the kinematically admissible displacement fields, those corresponding to equilibrium extremize the total potential energy. If the extremum condition is a minimum, the equilibrium state is stable.

$$\Pi = \frac{1}{2} \int_V \boldsymbol{\sigma}^T \boldsymbol{\epsilon} dV - \int_V \mathbf{u}^T \mathbf{f} dV - \int_S \mathbf{u}^T \mathbf{T} dS - \sum_i \mathbf{u}_i^T \mathbf{P}_i$$

Example



$$\Pi = \frac{1}{2}k_1\delta_1^2 + \frac{1}{2}k_2\delta_2^2 + \frac{1}{2}k_3\delta_3^2 + \frac{1}{2}k_4\delta_4^2 - F_1q_1 - F_3q_3$$

where δ_1 , δ_2 , δ_3 , and δ_4 are extensions of the four springs. Since $\delta_1 = q_1 - q_2$, $\delta_2 = q_2$, $\delta_3 = q_3 - q_2$, and $\delta_4 = -q_3$, we have

$$\Pi = \frac{1}{2}k_1(q_1 - q_2)^2 + \frac{1}{2}k_2q_2^2 + \frac{1}{2}k_3(q_3 - q_2)^2 + \frac{1}{2}k_4q_3^2 - F_1q_1 - F_3q_3$$

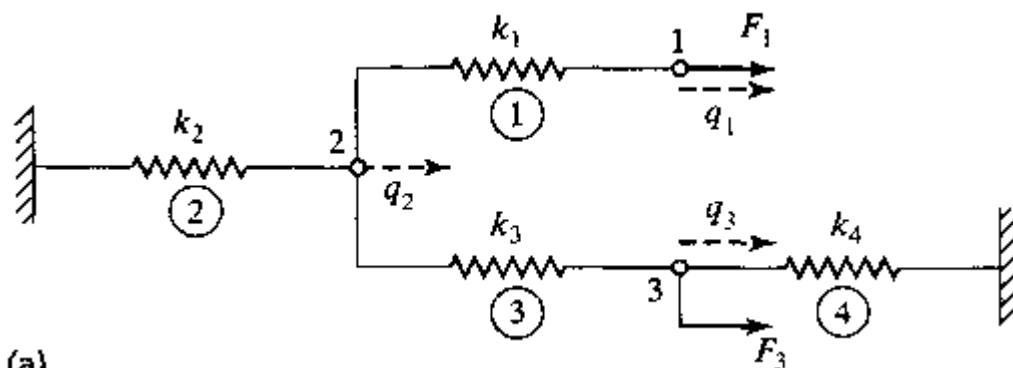
where q_1 , q_2 , and q_3 are the displacements of nodes 1, 2, and 3, respectively.

$$\frac{\partial \Pi}{\partial q_i} = 0 \quad i = 1, 2, 3$$

$$\frac{\partial \Pi}{\partial q_1} = k_1(q_1 - q_2) - F_1 = 0$$

$$\frac{\partial \Pi}{\partial q_2} = -k_1(q_1 - q_2) + k_2q_2 - k_3(q_3 - q_2) = 0 \quad (a)$$

$$\frac{\partial \Pi}{\partial q_3} = k_3(q_3 - q_2) + k_4q_3 - F_3 = 0$$



$$\begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 + k_3 & -k_3 \\ 0 & -k_3 & k_3 + k_4 \end{bmatrix} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ 0 \\ F_3 \end{Bmatrix}$$

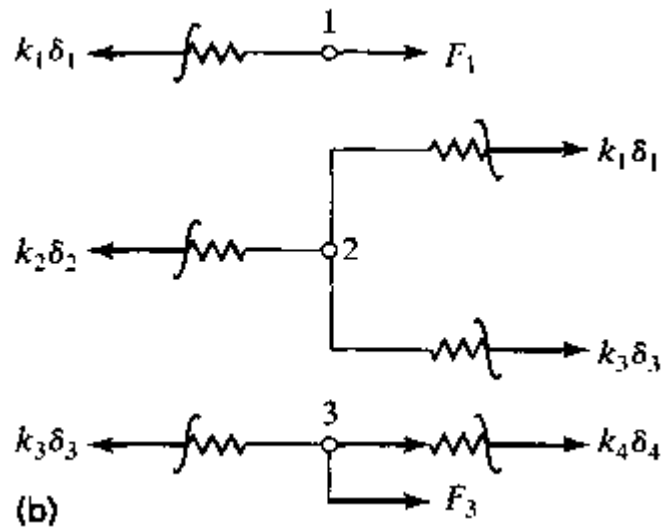
$$k_1 \delta_1 = F_1$$

$$k_2 \delta_2 - k_1 \delta_1 - k_3 \delta_3 = 0$$

$$k_3 \delta_3 - k_4 \delta_4 = F_3$$

$$\delta_1 = q_1 - q_2, \delta_2 = q_2,$$

$$\delta_3 = q_3 - q_2, \text{ and } \delta_4 = -q_3$$



$$\begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 + k_3 & -k_3 \\ 0 & -k_3 & k_3 + k_4 \end{bmatrix} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ 0 \\ F_3 \end{Bmatrix}$$

Rayleigh-Ritz method

Boundary conditions

Stress-strain relation

Strain-displacement relation



$$u = \sum a_i \phi_i(x, y, z) \quad i = 1 \text{ to } \ell$$

$$v = \sum a_i \phi_j(x, y, z) \quad j = \ell + 1 \text{ to } m$$

$$w = \sum a_k \phi_k(x, y, z) \quad k = m + 1 \text{ to } n$$

$$n > m > \ell$$

$$\Pi = \frac{1}{2} \int_V \boldsymbol{\sigma}^T \boldsymbol{\epsilon} dV - \int_V \mathbf{u}^T \mathbf{f} dV - \int_S \mathbf{u}^T \mathbf{T} dS - \sum_i \mathbf{u}_i^T \mathbf{P}_i$$

$$\Pi = \Pi(a_1, a_2, \dots, a_r)$$

$$\frac{\partial \Pi}{\partial a_i} = 0 \quad i = 1, 2, \dots, r$$

The potential energy for the linear elastic one-dimensional rod

$$\Pi = \frac{1}{2} \int_0^L EA \left(\frac{du}{dx} \right)^2 dx - 2u_1$$

where $u_1 = u(x = 1)$.

Let us consider a polynomial function

$$u = a_1 + a_2x + a_3x^2$$

This must satisfy $u = 0$ at $x = 0$ and $u = 0$ at $x = 2$. Thus,

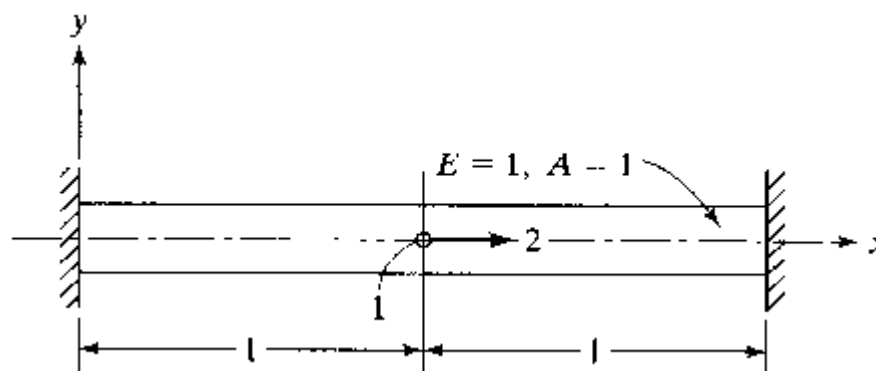
$$0 = a_1$$

$$0 = a_1 + 2a_2 + 4a_3$$

Hence,

$$a_2 = -2a_3$$

$$u = a_3(-2x + x^2) \quad u_1 = -a_3$$



Then $du/dx = 2a_3(-1 + x)$ and

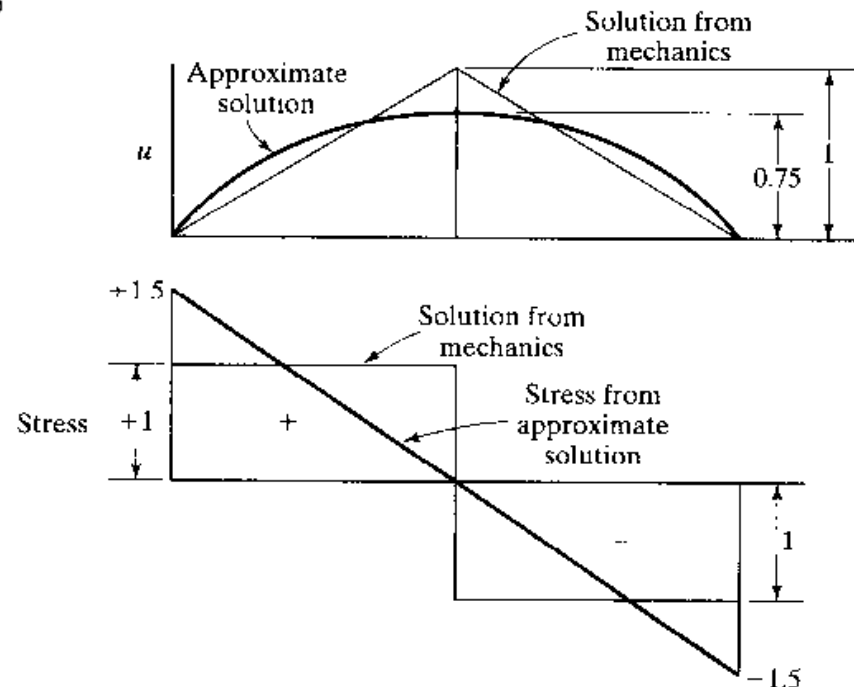
$$\begin{aligned}\Pi &= \frac{1}{2} \int_0^2 4a_3^2(-1 + x)^2 dx - 2(-a_3) \\ &= 2a_3^2 \int_0^2 (1 - 2x + x^2) dx + 2a_3 \\ &= 2a_3^2\left(\frac{2}{3}\right) + 2a_3\end{aligned}$$

We set $\partial \Pi / \partial a_3 = 4a_3\left(\frac{2}{3}\right) + 2 = 0$, resulting in

$$a_3 = -0.75 \quad u_1 = -a_3 = 0.75$$

The stress in the bar is given by

$$\sigma = E \frac{du}{dx} = 1.5(1 - x)$$



Residual of motion equation and boundary conditions

$$R_i = \sigma_{ij,j} + \bar{f}_i - \rho \ddot{u}_i \neq 0 \quad \text{在 } V \text{ 中}$$

$$\bar{R}_i = \sigma_{ij} n_j - \bar{T}_i \neq 0 \quad \text{在 } S_\sigma \text{ 上}$$

Equivalent weak form

$$\int_V R_i v_i dV + \int_{S_\sigma} \bar{R}_i \bar{v}_i dS = 0 \quad v_i, \bar{v}_i \text{ are the test functions}$$

Approximate solution

$$u_i = \sum_{I=1}^N \phi_I a_{iI} \quad \phi_I \text{ is the trial functions}$$

$$v_i = \sum_{I=1}^N W_I b_{iI}$$

$$\int_V R_i W_I dV = 0 \quad i = 1, 2, 3; \quad I = 1, 2, \dots, N$$



$$\int_V R_i W_I dV = 0 \quad i = 1, 2, 3; \quad I = 1, 2, \dots, N$$



Collocation method

$$W_I = \delta(\mathbf{x} - \mathbf{x}_I) \quad I = 1, 2, \dots, N$$

$$R_i(\mathbf{x}_I) = 0 \quad i = 1, 2, 3; \quad I = 1, 2, \dots, N$$

Subdomain method

$$W_I = \begin{cases} 1 & \mathbf{x} \in V_I \\ 0 & \mathbf{x} \notin V_I \end{cases} \quad I = 1, 2, \dots, N$$

Least square method

$$\frac{\partial}{\partial a_{iI}} \int_V R_i^2 dV = 2 \int_V R_i \frac{\partial R_i}{\partial a_{iI}} dV = 0$$

$$W_I = \frac{\partial R_i}{\partial a_{iI}} \quad I = 1, 2, \dots, N$$

Galerkin's method

$$\int_V R_i \phi_I dV = 0 \quad i = 1, 2, 3; \quad I = 1, 2, \dots, N$$

Example

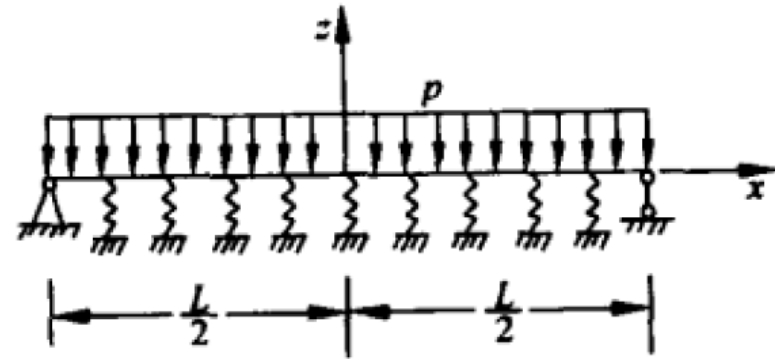


$$\begin{cases} \frac{d^4 w}{dx^4} + \alpha w + 1 = 0 & -1 \leq x \leq 1 \\ w(-1) = 0 \\ w(1) = 0 \end{cases}$$

$$\phi_1 = -\frac{1}{24}(5 - x^2)(1 - x^2)$$

$$w_1(x) = \phi_1 a_1 = -\frac{a_1}{24}(5 - x^2)(1 - x^2)$$

$$R_1(x, a_1) = -a_1 - \alpha \frac{a_1}{24}(5 - x^2)(1 - x^2) + 1$$



Collocation method

$$R_1(0, a_1) = -a_1 - \frac{5\alpha}{24}a_1 + 1 = 0$$

$$a_1 = \left(1 + \frac{5\alpha}{24}\right)^{-1}$$

Subdomain method

$$\int_{-1}^1 R_1 dx = -a_1 - \frac{2\alpha}{15}a_1 + 1 = 0 \quad a_1 = \left(1 + \frac{2\alpha}{15}\right)^{-1}$$

Least square method

$$\frac{\partial R_1}{\partial a_1} = -1 - \frac{\alpha}{24}(5 - x^2)(1 - x^2) \quad \int_{-1}^1 R_1 \frac{\partial R_1}{\partial a_1} dx = 0$$

$$a_1 = \left(1 + \frac{2\alpha}{15}\right) \left(1 + \frac{4\alpha}{15} + \frac{62\alpha^2}{2835}\right)^{-1}$$

Galerkin's method

$$\phi_1 = -\frac{1}{24}(5 - x^2)(1 - x^2) \quad \int_{-1}^1 R_1 \phi_1 dx = 0$$

$$a_1 = \left(1 + \frac{31\alpha}{189}\right)^{-1}$$

α	精确解	配点法	子域法	伽辽金法	最小二乘法
1	0.1788	0.1724	0.1838	0.1790	0.1832
10	0.07836	0.06757	0.08929	0.07891	0.08304
100	0.01134	0.00954	0.01453	0.01197	0.05818
1000	0.001025	0.000995	0.001551	0.001262	0.006068

$$\int_V \boldsymbol{\sigma}^T \boldsymbol{\epsilon}(\boldsymbol{\phi}) dV - \int_V \boldsymbol{\phi}^T \mathbf{f} dV - \int_S \boldsymbol{\phi}^T \mathbf{T} dS - \sum_i \boldsymbol{\phi}^T \mathbf{P} = 0$$

$$\int_V \left[\left(\frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \frac{\partial \tau_{xz}}{\partial z} + f_x \right) \phi_x + \left(\frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \sigma_y}{\partial y} + \frac{\partial \tau_{yz}}{\partial z} + f_y \right) \phi_y + \left(\frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \sigma_z}{\partial z} + f_z \right) \phi_z \right] dV = 0$$

where

$$\boldsymbol{\phi} = [\phi_x, \phi_y, \phi_z]^T$$

is an arbitrary displacement consistent with the boundary conditions of \mathbf{u} . If $\mathbf{n} = [n_x, n_y, n_z]^T$ is a unit normal at a point \mathbf{x} on the surface, the integration by parts formula is

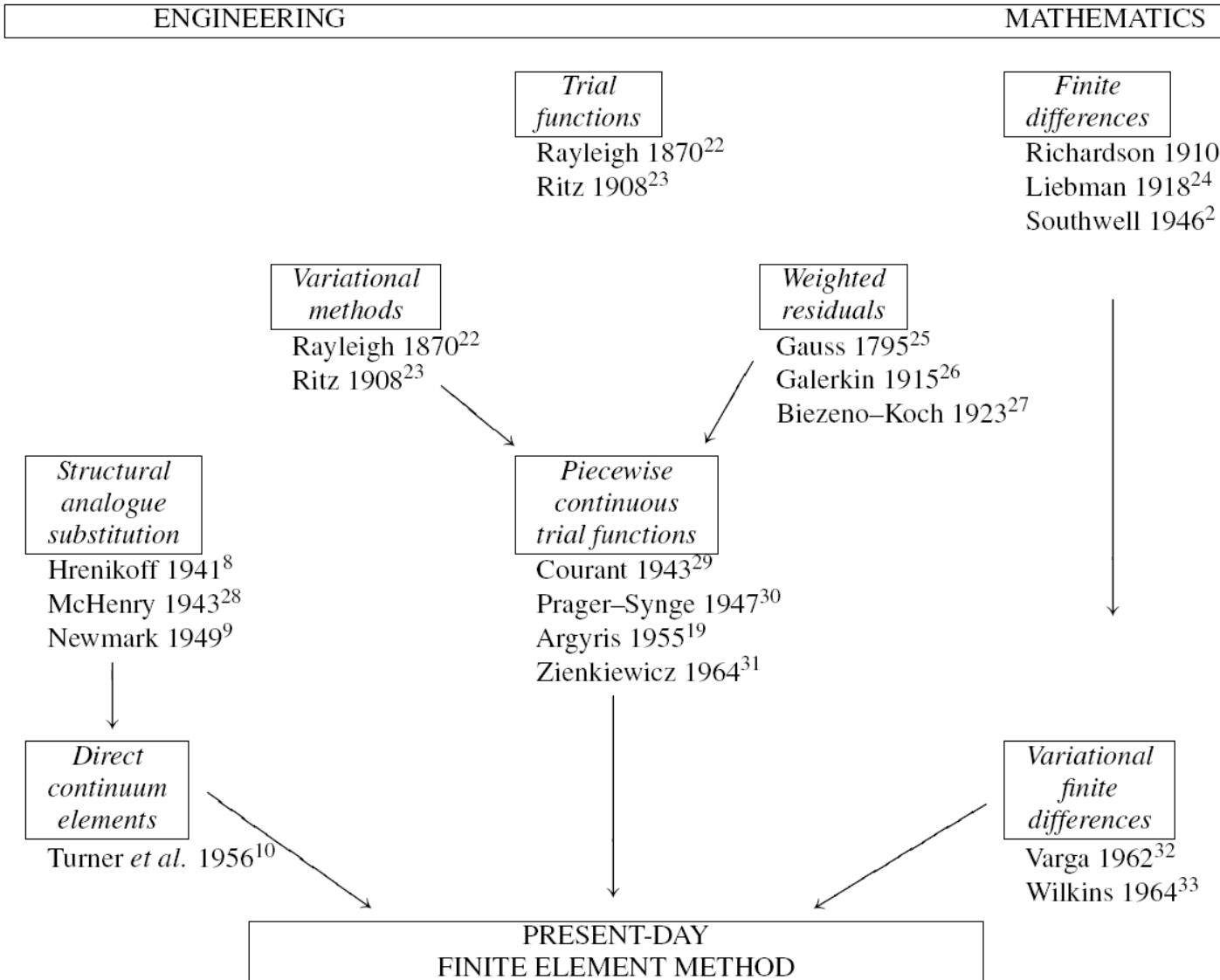
$$\begin{aligned} \int_V \frac{\partial \alpha}{\partial x} \theta dV &= - \int_V \alpha \frac{\partial \theta}{\partial x} dV + \int_S n_x \theta ds \\ &- \int_V \boldsymbol{\sigma}^T \boldsymbol{\epsilon}(\boldsymbol{\phi}) dV + \int_V \boldsymbol{\phi}^T \mathbf{f} dV + \int_S [(n_x \sigma_x + n_y \tau_{xy} + n_z \tau_{xz}) \phi_x \\ &+ (n_x \tau_{xy} + n_y \sigma_y + n_z \tau_{yz}) \phi_y + (n_x \tau_{xz} + n_y \tau_{yz} + n_z \sigma_z) \phi_z] dS = 0 \end{aligned}$$

where

$$\boldsymbol{\epsilon}(\boldsymbol{\phi}) = \left[\frac{\partial \phi_x}{\partial x}, \frac{\partial \phi_y}{\partial y}, \frac{\partial \phi_z}{\partial z}, \frac{\partial \phi_y}{\partial z} + \frac{\partial \phi_z}{\partial y}, \frac{\partial \phi_x}{\partial z} + \frac{\partial \phi_z}{\partial x}, \frac{\partial \phi_x}{\partial y} + \frac{\partial \phi_y}{\partial x} \right]^T$$

is the strain corresponding to the arbitrary displacement field $\boldsymbol{\phi}$.

History of approximate methods



❖ A typical structure built up from interconnected elements.

The forces acting on all the nodes

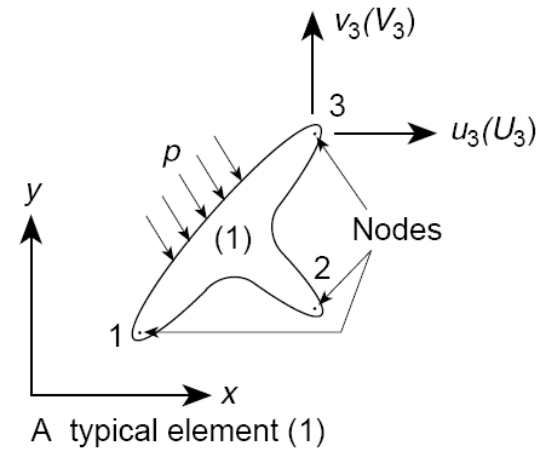
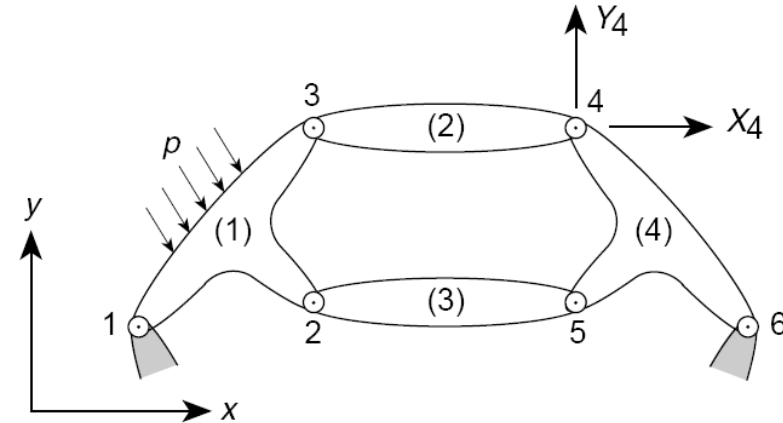
$$\mathbf{q}^1 = \begin{Bmatrix} \mathbf{q}_1^1 \\ \mathbf{q}_2^1 \\ \mathbf{q}_3^1 \end{Bmatrix} \quad \mathbf{q}_1^1 = \begin{Bmatrix} U_1 \\ V_1 \end{Bmatrix}$$

The corresponding nodal displacements

$$\mathbf{u}^1 = \begin{Bmatrix} \mathbf{u}_1^1 \\ \mathbf{u}_2^1 \\ \mathbf{u}_3^1 \end{Bmatrix} \quad \mathbf{u}_1^1 = \begin{Bmatrix} u_1 \\ v_1 \end{Bmatrix}$$

Assuming linear elastic behavior of the element, the characteristic relationship

$$\mathbf{q}^1 = \mathbf{K}^1 \mathbf{u}^1 + \mathbf{f}^1$$

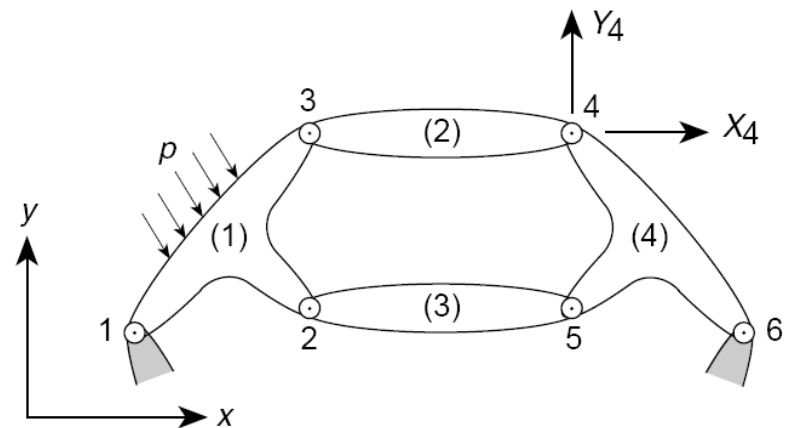


\mathbf{f}^1 represents the nodal forces balanced any concentrated or distributed loads acting on the element.

$\mathbf{K}^1 \mathbf{u}^1$ represents the forces induced by displacement of the nodes.

$$\mathbf{q}^e = \begin{Bmatrix} \mathbf{q}_1^e \\ \mathbf{q}_2^e \\ \vdots \\ \mathbf{q}_m^e \end{Bmatrix} \quad \text{and} \quad \mathbf{u}^e = \begin{Bmatrix} \mathbf{u}_1 \\ \mathbf{u}_2 \\ \vdots \\ \mathbf{u}_m \end{Bmatrix}$$

$$\mathbf{K}^e = \begin{bmatrix} \mathbf{K}_{11}^e & \mathbf{K}_{12}^e & \cdots & \mathbf{K}_{1m}^e \\ \mathbf{K}_{21}^e & \ddots & & \vdots \\ \vdots & \vdots & & \vdots \\ \mathbf{K}_{m1}^e & \cdots & \cdots & \mathbf{K}_{mm}^e \end{bmatrix}$$



Assembly and analysis of a structure



Consider again the hypothetical structure of Fig. 1.1. To obtain a complete solution the two conditions of

- (a) displacement compatibility and
- (b) equilibrium

Any system of nodal displacements \mathbf{u} :

$$\mathbf{u} = \begin{Bmatrix} \mathbf{u}_1 \\ \vdots \\ \mathbf{u}_n \end{Bmatrix}$$

$$\sum_{e=1}^m \mathbf{q}_a^e = \mathbf{q}_a^1 + \mathbf{q}_a^2 + \cdots = \mathbf{0}$$

$$\left(\sum_{e=1}^m \mathbf{K}_{a1}^e \right) \mathbf{u}_1 + \left(\sum_{e=1}^m \mathbf{K}_{a2}^e \right) \mathbf{u}_2 + \cdots + \sum_{e=1}^m \mathbf{f}_i^e = \mathbf{0}$$

$$\mathbf{K}\mathbf{u} + \mathbf{f} = \mathbf{0}$$

$$\mathbf{K}_{ab} = \sum_{e=1}^m \mathbf{K}_{ab}^e \quad \text{and} \quad \mathbf{f}_a = \sum_{e=1}^m \mathbf{f}_a^e$$

The boundary conditions



$$\mathbf{u}_1 = \mathbf{u}_6 = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$$\mathbf{K}_{11}\mathbf{u}_1 + \mathbf{K}_{12}\mathbf{u}_2 + \cdots + \mathbf{f}_1 = \mathbf{0}$$

$$\mathbf{K}_{21}\mathbf{u}_1 + \mathbf{K}_{22}\mathbf{u}_2 + \cdots + \mathbf{f}_2 = \mathbf{0}$$

etc.

The general pattern

