System Dynamics and Vibrations

Prof. Gustavo Alonso

Chapter 3: Single degree-of-freedom systems

Part 2

School of General Engineering Beihang University (BUAA)

Contents

- Introduction
- Undamped SDOF systems. Harmonic oscillator.
- Viscously damped SDOF.
- Coulomb damping. Dry friction.
- Response of SDOF systems to harmonic excitations.
- Frequency response plots.
- Systems with rotating unbalanced masses.
- Response to periodic excitation. Fourier series.
- Response to non-periodic excitation: step, ramp and impulse.

Introduction: types of excitations

- <u>Initial excitations</u>: Initial displacements, initial velocities, both
- Free vibration (free response): no further external factors affecting the system homogeneous equation
- Applied forces / moments → forced vibration / response
- → The response depends on the type of applied (external) forces / moments

- Steady-state excitations:
 - Harmonic
 - Periodic
- Steady-state excitations occur frequently in various areas of engineering
- The response is analysed in the frequency domain (rather than in time domain)

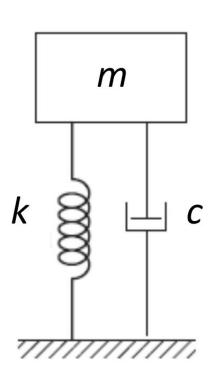
$$m\ddot{x} + c\dot{x} + kx = F$$

$$x(0) = x_0$$

$$\dot{x}(0) = v_0$$

initial conditions

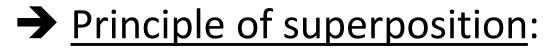
being x(t) the response and F(t) the excitation



$$m\ddot{x} + c\dot{x} + kx = F$$

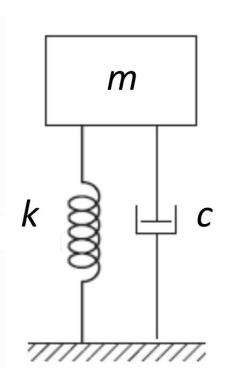
$$x(0) = x_0$$

$$\dot{x}(0) = v_0$$



- Response to initial excitations (already solved)
- Response to the applied force F(t)

 (obtained separately and combined linearly)



Harmonic force

$$F(t) = kf(t) = kA\cos\omega t$$
$$f(t) = A\cos\omega t$$

 ω is the excitation frequency, or driving frequency

$$m\ddot{x} + c\dot{x} + kx = F = kA\cos\omega t$$

$$\ddot{x}(t) + 2\varsigma\omega_n\dot{x}(t) + \omega_n^2x(t) = \omega_n^2A\cos\omega t$$

$$\omega_n = \sqrt{k/m}$$
 natural frequency of undamped oscillations

$$\varsigma = \frac{c}{2m\omega}$$
 viscous damping factor

- The response to harmonic excitation:
 - is also harmonic
 - has the same frequency as the excitation frequency

$$x(t) = C_1 \sin \omega t + C_2 \cos \omega t$$

 C_1 and C_2 are constants yet to be determined

$$\ddot{x}(t) + 2\varsigma\omega_n \dot{x}(t) + \omega_n^2 x(t) = \omega_n^2 A \cos \omega t$$

$$x(t) = C_1 \sin \omega t + C_2 \cos \omega t$$

$$\left[\left(\omega_n^2 - \omega^2 \right) C_1 - 2\varsigma \omega \omega_n C_2 \right] \sin \omega t + \left[2\varsigma \omega \omega_n C_1 + \left(\omega_n^2 - \omega^2 \right) C_2 \right] \cos \omega t$$
$$= \omega_n^2 A \cos \omega t$$

 \rightarrow coefficients of $\sin \omega t$ and $\cos \omega t$ must be equal

$$(\omega_n^2 - \omega^2)C_1 - 2\varsigma\omega\omega_n C_2 = 0$$
$$2\varsigma\omega\omega_n C_1 + (\omega_n^2 - \omega^2)C_2 = \omega_n^2 A$$



$$C_{1} = \frac{2\varsigma \mathscr{O}/\omega_{n}}{\left[1 - \left(\mathscr{O}/\omega_{n}\right)^{2}\right]^{2} + \left(2\varsigma \mathscr{O}/\omega_{n}\right)^{2}} A$$

$$C_{2} = \frac{1 - \left(\frac{\omega}{\omega_{n}}\right)^{2}}{\left[1 - \left(\frac{\omega}{\omega_{n}}\right)^{2}\right]^{2} + \left(2\varsigma\frac{\omega}{\omega_{n}}\right)^{2}} A$$

Steady-state solution:

$$x(t) = \frac{A}{\left[1 - \left(\frac{\omega}{\omega_n}\right)^2\right]^2 + \left(2\varsigma\frac{\omega}{\omega_n}\right)^2} \left\{\frac{2\varsigma\omega}{\omega_n}\sin\omega t + \left[1 - \left(\frac{\omega}{\omega_n}\right)^2\right]\cos\omega t\right\}$$

Steady-state solution:

$$x(t) = \frac{A}{\left[1 - \left(\frac{\omega}{\omega_n}\right)^2\right]^2 + \left(2\varsigma\frac{\omega}{\omega_n}\right)^2} \left\{\frac{2\varsigma\omega}{\omega_n}\sin\omega t + \left[1 - \left(\frac{\omega}{\omega_n}\right)^2\right]\cos\omega t\right\}$$

To find the physical interpretation we change the notation:

$$\sin \phi = \frac{2\varsigma \mathscr{O}/\omega_n}{\left\{ \left[1 - \left(\mathscr{O}/\omega_n \right)^2 \right]^2 + \left(2\varsigma \mathscr{O}/\omega_n \right)^2 \right\}^{1/2}}$$

$$\cos \phi = \frac{1 - \left(\mathscr{O}/\omega_n \right)^2}{\left\{ \left[1 - \left(\mathscr{O}/\omega_n \right)^2 \right]^2 + \left(2\varsigma \mathscr{O}/\omega_n \right)^2 \right\}^{1/2}}$$

$$x(t) = X \cos(\omega t - \phi)$$

$$X = X(\omega) = \frac{A}{\left[1 - \left(\frac{\omega}{\omega_n}\right)^2\right]^2 + \left(2\varsigma \frac{\omega}{\omega_n}\right)^2}$$

$$\phi = \phi(\omega) = \tan^{-1} \frac{2\varsigma \frac{\omega}{\omega_n}}{1 - \left(\frac{\omega}{\omega_n}\right)^2}$$

$$x(t) = X \cos(\omega t - \phi)$$

$$X = X(\omega) = \frac{A}{\left[1 - \left(\frac{\omega}{\omega_n}\right)^2\right]^2 + \left(2\varsigma \frac{\omega}{\omega_n}\right)^2}$$

amplitude

$$\phi = \phi(\omega) = \tan^{-1} \frac{2\varsigma \phi/\omega_n}{1 - \left(\frac{\omega}{\omega_n}\right)^2}$$

phase angle

Contents

- Introduction
- Undamped SDOF systems. Harmonic oscillator.
- Viscously damped SDOF.
- Coulomb damping. Dry friction.
- Response of SDOF systems to harmonic excitations.
- Frequency response plots.
- Systems with rotating unbalanced masses.
- Response to periodic excitation. Fourier series.
- Response to non-periodic excitation: step, ramp and impulse.

Frequency response plots
$$F(t) = kf(t) = kA\cos\omega t \qquad \Rightarrow \qquad \text{if} \qquad f(t) = A\cos\omega t \qquad \Rightarrow \qquad \text{Re } x(t)$$

$$f(t) = Ae^{i\omega t} \qquad \Rightarrow \qquad \text{if} \qquad f(t) = A\sin\omega t \qquad \Rightarrow \qquad \text{Im } x(t)$$



$$f(t) = A\sin\omega t \qquad \qquad \text{Im } x(t)$$

$$\ddot{x}(t) + 2\varsigma\omega_n\dot{x}(t) + \omega_n^2x(t) = \omega_n^2Ae^{i\omega t}$$

$$x(t) = X(i\omega)e^{i\omega t}$$

$$X(i\omega) = \frac{A}{1 - (\omega/\omega_n^2)^2 + i2\varsigma \omega/\omega_n}$$

$$G(i\omega) = \frac{X(i\omega)}{A} = \frac{1}{1 - (\omega/\omega_n^2)^2 + i2\varsigma \omega/\omega_n}$$



frequency response

$$x(t) = AG(i\omega)e^{i\omega t}$$

$$G(i\omega) = |G(i\omega)| e^{-i\phi(\omega)}$$

$$|G(i\omega)| = \left\{ \left[\operatorname{Re} G(i\omega) \right]^2 + \left[\operatorname{Im} G(i\omega) \right]^2 \right\}^{1/2}$$

$$\phi(\omega) = \tan^{-1} \left[\frac{-\operatorname{Im} G(i\omega)}{\operatorname{Re} G(i\omega)} \right]$$

$$x(t) = A |G(i\omega)| e^{i(\omega t - \phi)}$$

$$x(t) = A |G(i\omega)| e^{i(\omega t - \phi)}$$

if
$$f(t) = A\cos\omega t \implies x(t) = \operatorname{Re} A |G(i\omega)| e^{i(\omega t - \phi)} = A |G(i\omega)| \cos(\omega t - \phi)$$

$$f(t) = A\sin\omega t \implies x(t) = \operatorname{Im} A |G(i\omega)| e^{i(\omega t - \phi)} = A |G(i\omega)| \sin(\omega t - \phi)$$

$$x(t) = A |G(i\omega)| e^{i(\omega t - \phi)}$$

$$\dot{x}(t) = i\omega x(t)$$

$$\ddot{x}(t) = -\omega^2 x(t)$$

$$2\zeta \omega_n \dot{x}(t)$$

$$i \text{ Im } x(t)$$

$$\omega_n^2 x(t)$$

$$\omega_n^2 x(t)$$

$$\omega_n \dot{x}(t)$$

$$\ddot{x}(t) + 2\varsigma\omega_n\dot{x}(t) + \omega_n^2x(t) = \omega_n^2Ae^{i\omega t}$$

$$|G(i\omega)| = \left[G(i\omega)\overline{G}(i\omega)\right]^{1/2} = \frac{1}{\left\{\left[1 - \left(\omega/\omega_n\right)^2\right]^2 + \left(2\varsigma\left(\omega/\omega_n\right)^2\right)\right\}^{1/2}}$$

$$\phi(\omega) = \tan^{-1}\left[\frac{-\operatorname{Im}G(i\omega)}{\operatorname{Re}G(i\omega)}\right] = \tan^{-1}\frac{2\varsigma\omega/\omega_n}{1 - \left(\omega/\omega_n\right)^2}$$

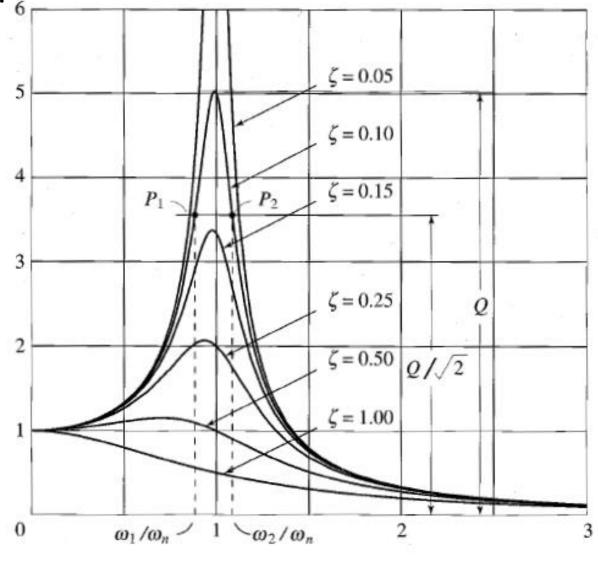
 $|G(i\omega)|$

$$|G(i\omega)| = \frac{1}{\left\{ \left[1 - \left(\omega/\omega_n \right)^2 \right]^2 + \left(2\varsigma \left(\omega/\omega_n \right)^2 \right) \right\}^{1/2}}$$

response peaks:

$$\frac{d|G(i\omega)|}{d(\omega/\omega_n)} = 0 \implies \frac{\omega}{\omega_n} = \sqrt{1 - 2\varsigma^2}$$

 $\varsigma > 1/\sqrt{2}$ \Longrightarrow no peaks



 ω / ω_n

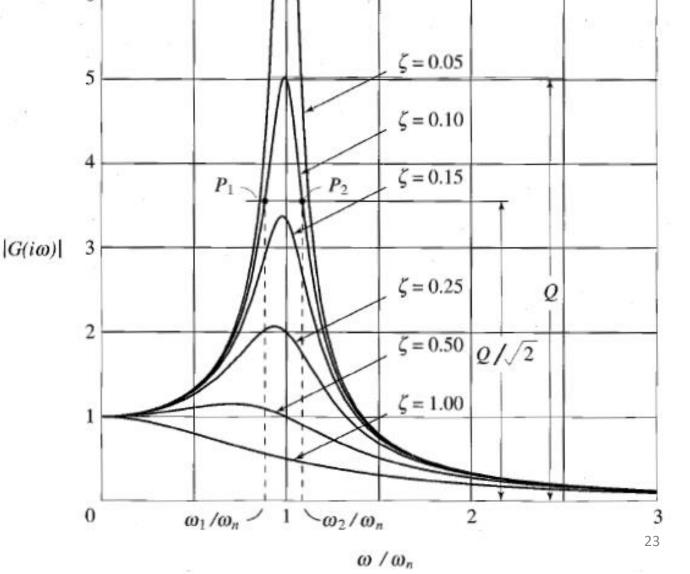
response peaks:

$$|G(i\omega)|_{\text{max}} = \frac{1}{2\varsigma\sqrt{1-\varsigma^2}}$$

light damping:

$$\varsigma \ll 1$$
 \Rightarrow $|G(i\omega)|_{\max} = Q \cong \frac{1}{2\varsigma}$

experimental way to estimate viscous damping: 1



 P_1 , P_2 : half-power points (the power absorbed by the damper is proportional to the square of the amplitude)

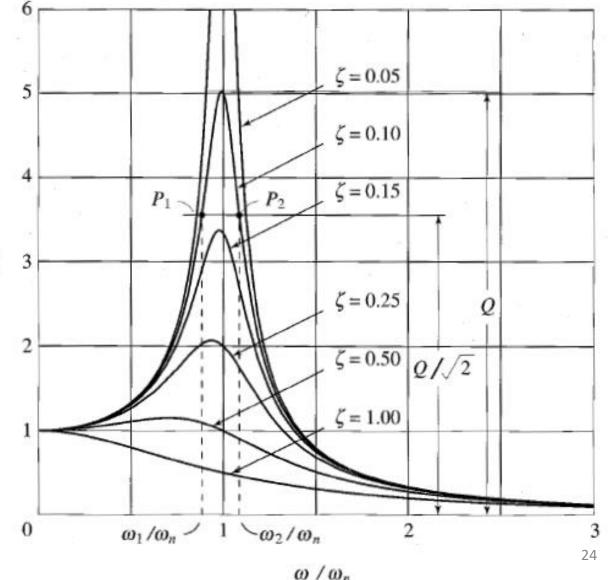
$$|G(i\omega)|_{\text{max}} = \frac{1}{\sqrt{2}Q} = \frac{1}{2\sqrt{2}\varsigma}$$

$$\frac{\left(\omega_{1}/\omega_{n}\right)^{2}}{\left(\omega_{2}/\omega_{n}\right)^{2}} \cong 1 - 2\varsigma^{2} \mp 2\varsigma$$

$$\left(\frac{\omega_{1}}{\omega_{2}}\right)^{2} = 1 - 2\varsigma^{2} + 2\varsigma$$

$$\varsigma \ll 1 \qquad \Longrightarrow \qquad \begin{array}{c} \omega_{1} + \omega_{2} \cong 2\omega_{n} \\ \Delta \omega = \omega_{2} - \omega_{1} \cong 2\varsigma \omega_{n} \\ \text{bandwith} \end{array}$$

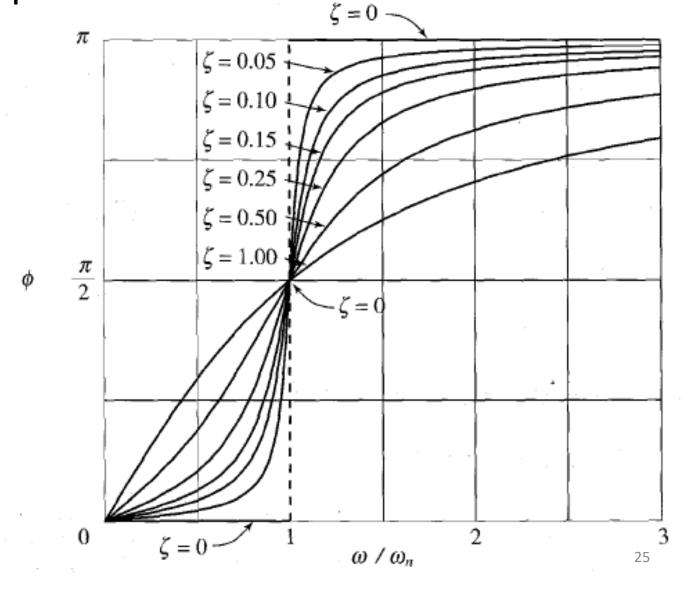
$$Q \cong \frac{1}{2} \cong \frac{\omega_n}{1}$$
 high Q implies small bandwith



$$\phi(\omega) = \tan^{-1} \frac{2\varsigma \, \omega / \omega_n}{1 - \left(\omega / \omega_n\right)^2}$$

$$\omega/\omega_n = 1$$
 $\phi = \frac{\pi}{2}$

for any value of $\, \, \varsigma \,$



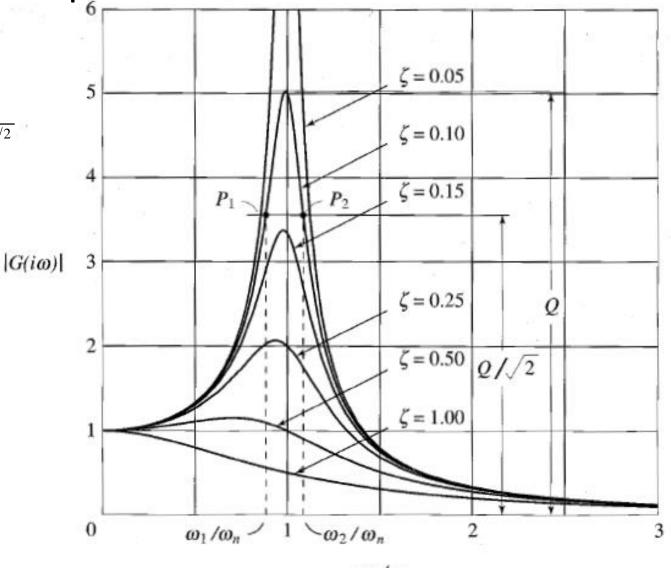
$$\left|G(i\omega)\right| = \frac{1}{\left\{\left[1 - \left(\omega/\omega_n\right)^2\right]^2 + \left(2\varsigma\left(\omega/\omega_n\right)^2\right)\right\}^{1/2}}$$

undamped case:

$$\varsigma = 0 \quad \Longrightarrow \quad |G(i\omega)| \to \infty$$

at
$$\omega = \omega_n$$

resonance



$$\phi(\omega) = \tan^{-1} \frac{2\varsigma \, \omega / \omega_n}{1 - \left(\omega / \omega_n\right)^2}$$

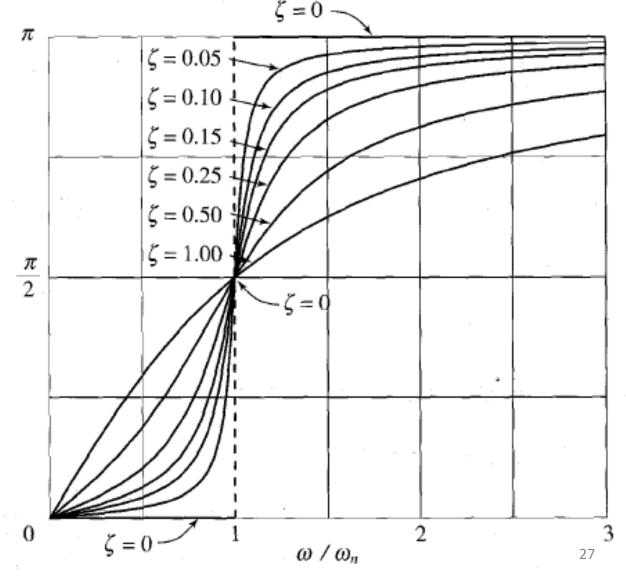
undamped case:

$$\varsigma = 0$$
 discontinuity in the phase angle

$$\omega/\omega_n < 1 \quad \Longrightarrow \quad \phi = 0$$

$$\omega/\omega_n = 1$$
 \Rightarrow $\phi = \frac{\pi}{2}$ (resonance)

$$\omega/\omega_n > 1 \quad \Longrightarrow \quad \phi = \pi$$



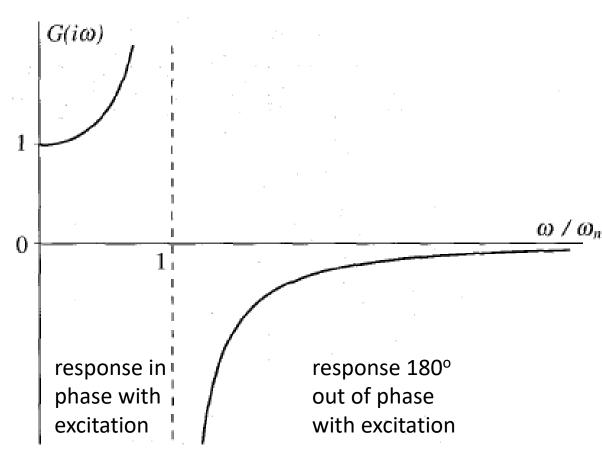
Response of SDOF systems to harmonic excitations: undamped case

$$\ddot{x}(t) + 2\dot{z}\omega_n\dot{x}(t) + \omega_n^2x(t) = \omega_n^2A\cos\omega t$$

$$x(t) = AG(\omega)\cos\omega t$$

$$G(\omega) = \frac{1}{1 - \left(\omega/\omega_n\right)^2}$$

real function (different than in the damped case (complex function)



Response of SDOF systems to harmonic excitations: undamped case

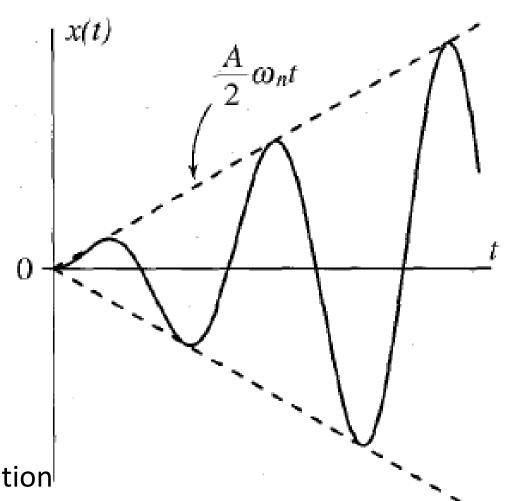
At resonance: $\omega = \omega_n$

$$\ddot{x}(t) + \omega_n^2 x(t) = \omega_n^2 A \cos \omega_n t$$

$$x(t) = \frac{A}{2} \omega_n t \sin \omega_n t$$

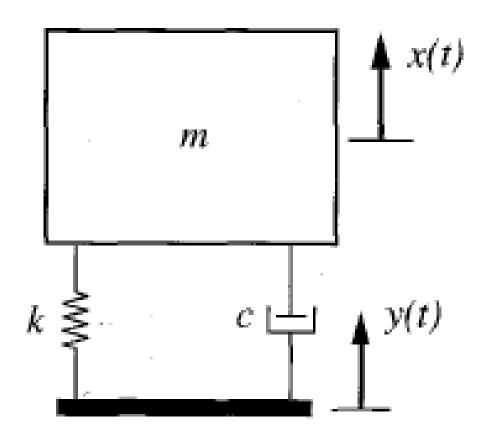
is a particular solution

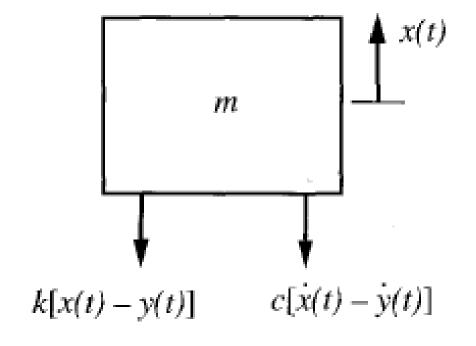
→ response 90° out of phase with excitation

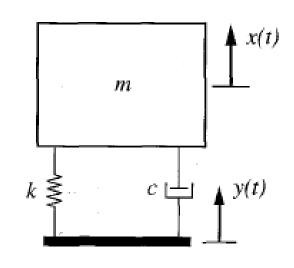


- Equipment placed on a vibrating foundation
- Vehicle traveling on a bumpy road
- Engine mounted on an aircraft wing

•







$$-c(\dot{x} - \dot{y}) - k(x - y) = m\ddot{x}$$

$$\ddot{x} + 2\varsigma\omega_n\dot{x} + \omega_n^2x = 2\varsigma\omega_n\dot{y} + \omega_n^2y$$

$$y(t) = \operatorname{Re} Ae^{i\omega t}$$

$$x(t) = X(i\omega)e^{i\omega t}$$

If the excitation is $A\cos\omega t$ the response is $\operatorname{Re} x(t)$ If the excitation is $A\sin\omega t$ the response is $\operatorname{Im} x(t)$

$$X(i\omega) = (1 + i2\varsigma \omega/\omega_n)G(i\omega)A$$
$$x(t) = |X(i\omega)|e^{i(\omega t - \phi)}$$

$$|X(i\omega)| = \left[1 + \left(\frac{2\varsigma\omega}{\omega_n}\right)^2\right]^{1/2} |G(i\omega)|A$$

$$\phi(\omega) = \tan^{-1} \left[\frac{-\operatorname{Im} G(i\omega)}{\operatorname{Re} G(i\omega)} \right] = \tan^{-1} \frac{2\varsigma \left(\omega/\omega_n\right)^3}{1 - \left(\omega/\omega_n\right)^2 + \left(2\varsigma \omega/\omega_n\right)^2}$$

$$G(i\omega) = \frac{1}{1 - \left(\omega/\omega_n^2\right)^2 + i2\varsigma \,\omega/\omega_n}$$

