

# Technical Variance



# Rolls-Royce

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This document is issued specifically to address the issue raised in the associated Technical Variance request. Authorisation for repeated application can be covered by the issuance of further documents and may be requested

This document is supplemental to the Manual(s) specified below and should be suitably stored in accordance with local airworthiness requirements.

Technical Variance No.	124851	Issue	2	Date	04 JUL 2012
Operator / Applicant	DEUTSCHE LUFTHANSA ROLLS-ROYCE			Original Request No.	TV_27_06_2012 DK
Engine Type	RB211	Engine Mark(s)	TRENT 970-84,972-84		
Part Description	TAIL BEARING HOUSING		Engine Serial No.	91057	
Part No.	FW 35923,FW27718,FW 51434		Part Serial No.	VARIOUS	
Manual Title	EM & AMM	Ref.	Airbus-A380 AMM & E-TRENT-9RR	ATA Ref.	72-52-51
TV Title	Fluorescent Penetrant Inspection of the Tail Bearing Housing, Mount Lug Run-outs				
Hours	N/A		Cycles	N/A	
<b>Existing Requirement</b>  The Aircraft Maintenance Manual (AMM) and the Engine Manual (EM) do not contain a specific procedure for Fluorescent Penetrant Inspection (FPI) of the Mount Lug Run-outs. This Technical Variance details the procedure for FPI of the TBH Mount Lug Run-outs.					
<b>Requested Variance</b>  ISSUE 2 DUE TO CHANGE IN ENGINE SERIAL NUMBER  A Technical Variance is requested to detail the procedure for the TBH inspection for both on wing and in shop inspections (in shop inspection procedure included for reference).  Engine 91057 is to be inspected on wing.					
<b>Summary of Investigation and Conclusions</b>  Rolls-Royce Engineering have reviewed the above requirement and an on wing trial of the FPI of the TBH Mount Lug Run-outs has been conducted.  This Technical Variance is issued to instruct the inspection of the TBH Mount Lug Run-outs for engine 91057 pending issue of a planned Non-modification Service Bulletin.					
<b>Approval on Behalf of Rolls-Royce</b>  Title <u>PRINCIPAL TECHNOLOGIST - SERVICE ENGINEERING</u> Service Engineering Team Leader / CVE Signature <u>[Signature]</u> Printed name <u>J D CLARKE</u> Date <u>04/07/12</u> 04 <sup>th</sup> July 2012 Document Created by <u>Daniel Kebell</u> Daniel Kebell					

## Declaration of Approval

This document is approved by Rolls-Royce plc under the authority of EASA Design Organisation Approval no. EASA.21J.035.

Rolls-Royce plc. Proprietary Information – Not for Manufacture



### Manpower Requirements

#### 1) On-wing

Time to gain access

0.5 hour (1 person)

Time to inspect Tail Bearing Housing (TBH) Mount Lug Run-outs

2 hours (2 persons)

Time to restore to serviceable condition

0.5 hour (1 person)

#### 2) In Shop

Time to inspect Tail Bearing Housing (TBH) Mount Lug Run-outs

2 hours (2 persons)

The Fluorescent Penetrant Inspection (FPI) defined in this Technical Instruction shall only be performed by personnel qualified to FPI Level 2 or higher and in possession of an up to date eyesight record in accordance with EN4179/NAS410 or alternative standards/guidelines accepted by the applicable National Aviation Authority.

### Materials

OMat 651F - F2B - AMS2644 Type 1, Method D, Level 3 (1D3) compliant aerosol penetrant (Fluorescent, Post-Emulsified, Medium Sensitivity).

OMat 617 – AMS2644 Form d – Non-Aqueous Wet Developer – Aerosol.

Ultra Violet Light Source capable of providing a minimum UV radiation intensity of 1200  $\mu\text{W}/\text{cm}^2$  at the inspection surface.

Dark Non-Reflective cover to darken the area for inspection.

OMat 150 – Acetone.

OMat 2/101 – Lint Free Cloths.

OMat 677 – Penetrant Test Panel.

Visual Aids as per AMM Task 70-20-02, Fluorescent Penetrant Inspection.



### PROCEDURE – ON WING

**WARNING:** YOU MUST BE CAREFUL WHEN YOU WORK ON THE ENGINE AFTER THE ENGINE IS SHUTDOWN. THE ENGINE CAN STAY HOT FOR UPTO ONE HOUR.

**WARNING:** YOU MUST NOT TOUCH HOT PARTS WITHOUT APPLICABLE GLOVES. HOT PARTS CAN CAUSE INJURY. IF YOU GET AN INJURY, TREAT WITH COLD WATER FOR 10 MINUTES, THEN SEEK MEDICAL AID.

1) To get access to the Tail Bearing Housing (TBH) Mount Lug Run-outs

- i. Open the Fan Cowl Doors in accordance with AMM Task 71-13-00, Maintenance Practices to get access
- ii. Make the Thrust Reverser unserviceable for maintenance in accordance with AMM Task 78-30-00 Deactivation/Activation.

**WARNING:** YOU MUST MAKE THE THRUST REVERSER UNSERVICEABLE (INSTALL AND SAFETY THE INHIBITION DEVICE) BEFORE YOU DO WORK ON OR AROUND THE THRUST REVERSER. IF YOU DO NOT INSTALL AND SAFETY THE INHIBITION DEVICE, THERE IS A RISK THAT THE THRUST REVERSER CAN OPERATE ACCIDENTALLY. THIS CAN CAUSE INJURY TO PERSON AND/OR DAMAGE TO THE EQUIPMENT.

- iii. Open the Fan Exhaust Cowl in accordance with the AMM, Task 78-26-00
- iv. Use appropriate staging to gain access to the Tail Bearing Housing TBH

2) Carry out the inspection on the Mount Lug Run-outs

**WARNING:** PROTECTIVE GLOVES AND SAFETY GLASSES MUST BE WORN AT ALL TIMES DURING THE INSPECTION. IF YOU GET AN INJURY, SEEK MEDICAL AID IMMEDIATELY.

- i. This inspection must be performed in accordance with the requirements of AMM 70-20-02, Fluorescent Penetrant Inspection.
- ii. Local cleaning of the inspection areas must be completed prior to FPI.
- iii. Apply OMat 150 – Acetone to a clean OMat 2/101 – Lint Free Cloth.
- iv. Clean the inspection surfaces (see Figures 1 and 2) with the moistened cloth.
- v. Ensure that all contaminants have been removed. If necessary repeat steps (iii) and (iv) in order to achieve this.

**NOTE:** THE INSPECTION SURFACE, MUST BE CLEAN, DRY AND FREE FROM SOILS, OIL, GREASE, PAINT, COATINGS, CORROSION PRODUCTS, SCALE, SMEARED METAL, WELDING FLUX, CHEMICAL RESIDUES OR ANY OTHER MATERIAL THAT COULD PREVENT PENETRANT FROM ENTERING A DISCONTINUITY, AFFECT PROCESS PERFORMANCE OR PRODUCE AN UNACCEPTABLE BACKGROUND.

- vi. Allow 10 minutes for the area to completely dry prior to penetrant application.



- vii. Penetrant Application - Apply OMat 651F – 1D3 Penetrant to the inspection areas (Figures 1 and 2) by brush or sponge applicator. The penetrant must be dispensed from a sealed aerosol can and any unused penetrant must go to waste.
- viii. Ensure that the inspection area is fully covered with penetrant with the aid of a UV lamp.
- ix. Apply OMat 651F – 1D3 Penetrant to the OMat 677 – Penetrant Test Panel.
- x. Leave the penetrant in contact with the components for 30 minutes.

**NOTE:** THE PENETRANT CONTACT TIME MUST NOT EXCEED 1 HOUR. IF IT DOES THE PENETRANT MUST BE REAPPLIED AND 5 MINUTES CONTACT TIME ALLOWED.

- xi. Penetrant Removal - After the 30 minutes contact time, remove the excess penetrant from the inspection surface and the OMat 677 – Penetrant Test Panel by wiping thoroughly with a clean OMat 2/101 – Lint Free Cloth.
- xii. Whilst illuminating with UV light, wipe off the penetrant with a clean OMat 2/10 - Lint Free Cloth moistened with OMat 150 – Acetone. Use the minimum application necessary to achieve acceptable background fluorescence and ensure that acetone is not allowed to flow on the inspection surface.
- xiii. Ensure that all traces of the OMat 150 – Acetone have been completely removed.
- xiv. Develop - Apply a light, even coating of OMat 617 – Non-Aqueous Wet Developer from a sealed aerosol can to the inspection area and the OMat 677 – Penetrant Test Panel.
- xv. Allow 10 minutes contact time for the developer.
- xvi. Cover the region of the engine with a dark non-reflective cover in order to reduce the white light levels.
- xvii. Ensure there are no white light sources or fluorescent items (including clothing) within the inspection area or the inspector's field of vision.
- xviii. Do not wear darkened or light sensitive glasses.
- xix. Allow 1 minute for eye adaptation to the darkened area.
- xx. Illuminate the OMat 677 – Penetrant Test Panel with UV light at an intensity of no less than  $1200 \mu\text{W}/\text{cm}^2$  in the darkened area.
- xxi. Inspect the OMat 677 – Penetrant Test Panel and check that the 4 largest indications are visible and match the control photograph for that individual panel.
- xxii. Illuminate the inspection surface with UV light at an intensity of no less than  $1200 \mu\text{W}/\text{cm}^2$  in the darkened area.
- xxiii. Inspect the Mount Lug Run-Outs for crack indications.



- xxiv. The "Wipe Off" technique can be used to aid the evaluation of any indications and must be performed as directed by AMM 70-20-02, Fluorescent Penetrant Inspection.
- xxv. The cause of all indications must be evaluated under white light using visual aids, such as magnification, as appropriate.
- xxvi. Record any suspected crack indication in Appendix x. Report any suspected crack indications to Rolls-Royce local representative.
- xxvii. Post Clean - Once all indications have been evaluated and recorded, remove the penetrant processing materials from the inspection surface by wiping with a clean OMat 2/101 – Lint Free Cloth moistened with OMat 150 – Acetone.

### 3) Close access to the Tail Bearing Housing (TBH)

- i. Make sure the work area is clean and clear of tools and other items.
- ii. Close the Fan Exhaust Cowl in accordance with AMM 78-26-00, Maintenance Practices.
- iii. Make the Thrust Reverser serviceable after maintenance in accordance with AMM 78-30-00, Deactivation/Activation as necessary.
- iv. Close the Fan Cowl Doors in accordance with AMM Task 71-13-00, Maintenance Practices.



### PROCEDURE – IN SHOP

1) Carry out the inspection on the Mount Lug Run-outs

**WARNING:** PROTECTIVE GLOVES AND SAFETY GLASSES MUST BE WORN AT ALL TIMES DURING THE INSPECTION. IF YOU GET AN INJURY, SEEK MEDICAL AID IMMEDIATELY.

- i. This inspection must be performed in accordance with the requirements of 70-00-00, Overhaul Process 210, Fluorescent Penetrant Inspection.
- ii. Local cleaning of the inspection areas must be completed prior to FPI.
- iii. Apply OMat 150 – Acetone to a clean OMat 2/101 – Lint Free Cloth.
- iv. Clean the inspection surfaces (see Figures 1 and 2) with the moistened cloth.
- v. Ensure that all contaminants have been removed. If necessary repeat steps (iii) and (iv) in order to achieve this.

**NOTE:** THE INSPECTION SURFACE, MUST BE CLEAN, DRY AND FREE FROM SOILS, OIL, GREASE, PAINT, COATINGS, CORROSION PRODUCTS, SCALE, SMEARED METAL, WELDING FLUX, CHEMICAL RESIDUES OR ANY OTHER MATERIAL THAT COULD PREVENT PENETRANT FROM ENTERING A DISCONTINUITY, AFFECT PROCESS PERFORMANCE OR PRODUCE AN UNACCEPTABLE BACKGROUND.

- vi. Allow 10 minutes for the area to completely dry prior to penetrant application.
- vii. Penetrant Application - Apply OMat 651F – 1D3 Penetrant to the inspection areas (Figures 1 and 2) by brush or sponge applicator. The penetrant must be dispensed from a sealed aerosol can and any unused penetrant must go to waste.
- viii. Ensure that the inspection area is fully covered with penetrant with the aid of a UV lamp.
- ix. Apply OMat 651F – 1D3 Penetrant to the OMat 677 – Penetrant Test Panel.
- x. Leave the penetrant in contact with the components for 30 minutes.

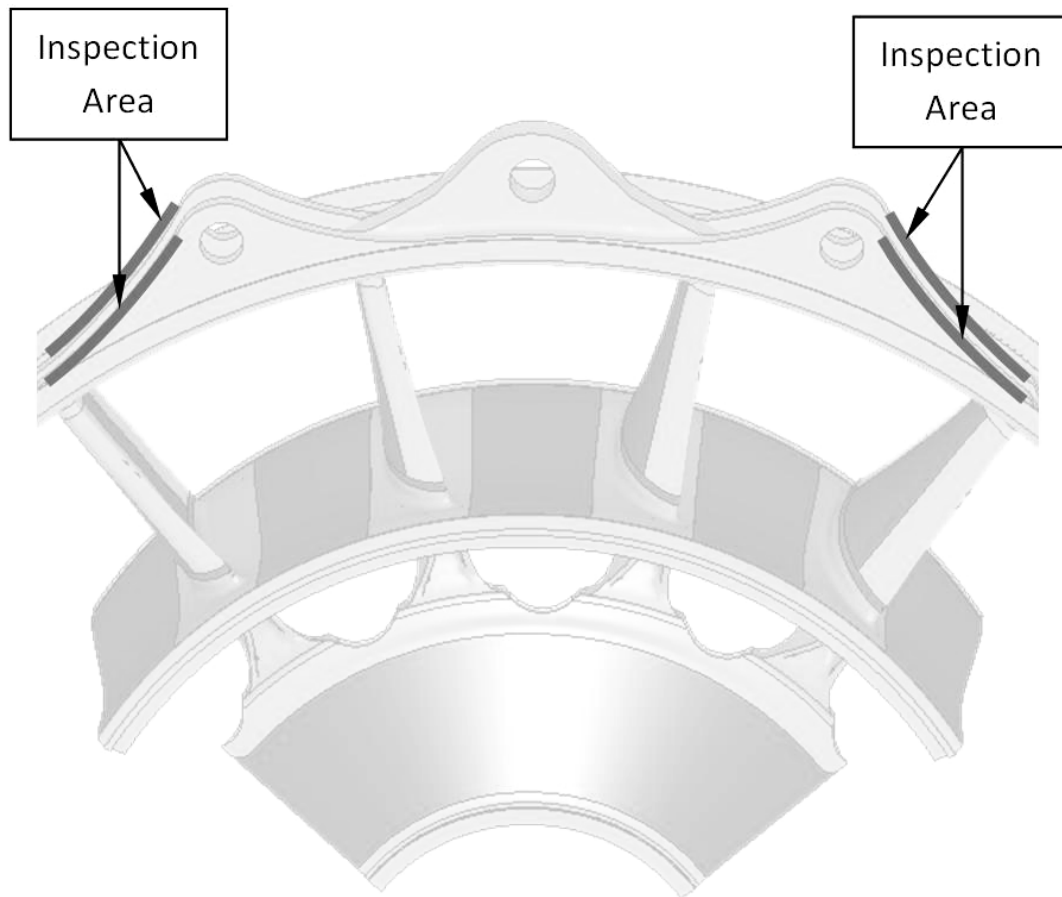
**NOTE:** THE PENETRANT CONTACT TIME MUST NOT EXCEED 1 HOUR. IF IT DOES THE PENETRANT MUST BE REAPPLIED AND 5 MINUTES CONTACT TIME ALLOWED.

- xi. Penetrant Removal - After the 30 minutes contact time, remove the excess penetrant from the inspection surface and the OMat 677 – Penetrant Test Panel by wiping thoroughly with a clean OMat 2/101 – Lint Free Cloth.
- xii. Whilst illuminating with UV light, wipe off the penetrant with a clean OMat 2/10 - Lint Free Cloth moistened with OMat 150 – Acetone. Use the minimum application necessary to achieve acceptable background fluorescence and ensure that acetone is not allowed to flow on the inspection surface.



- xiii. Ensure that all traces of the OMat 150 – Acetone have been completely removed.
- xiv. Develop - Apply a light, even coating of OMat 617 – Non-Aqueous Wet Developer from a sealed aerosol can to the inspection area and the OMat 677 – Penetrant Test Panel.
- xv. Allow 10 minutes contact time for the developer.
- xvi. Cover the region of the engine with a dark non-reflective cover in order to reduce the white light levels.
- xvii. Ensure there are no white light sources or fluorescent items (including clothing) within the inspection area or the inspector's field of vision.
- xviii. Do not wear darkened or light sensitive glasses.
- xix. Allow 1 minute for eye adaptation to the darkened area.
- xx. Illuminate the OMat 677 – Penetrant Test Panel with UV light at an intensity of no less than  $1200 \mu\text{W}/\text{cm}^2$  in the darkened area.
- xxi. Inspect the OMat 677 – Penetrant Test Panel and check that the 4 largest indications are visible and match the control photograph for that individual panel.
- xxii. Illuminate the inspection surface with UV light at an intensity of no less than  $1200 \mu\text{W}/\text{cm}^2$  in the darkened area.
- xxiii. Inspect the Mount Lug Run-Outs for crack indications.
- xxiv. The "Wipe Off" technique can be used to aid the evaluation of any indications and must be performed as directed by 70-00-00, Overhaul Process 210, Fluorescent Penetrant Inspection.
- xxv. The cause of all indications must be evaluated under white light using visual aids, such as magnification, as appropriate.
- xxvi. Record any suspected crack indication in Appendix 1. Report any suspected crack indications to Rolls-Royce local representative.
- xxvii. Post Clean - Once all indications have been evaluated and recorded, remove the penetrant processing materials from the inspection surface by wiping with a clean OMat 2/101 – Lint Free Cloth moistened with OMat 150 – Acetone.

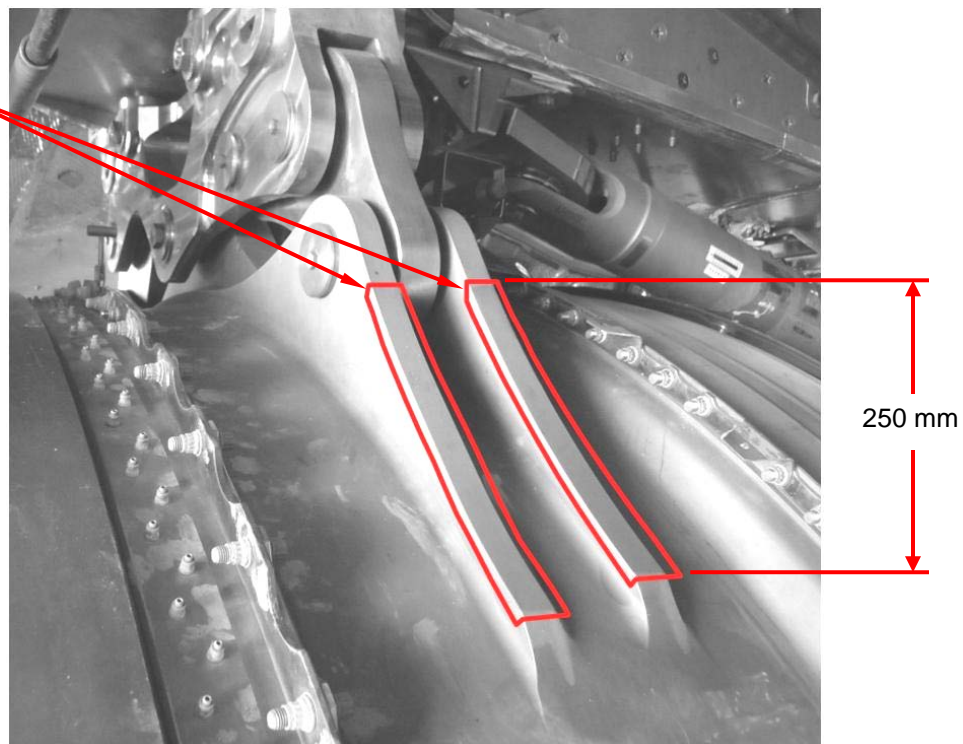




**Figure 1 - Mount Lug Run Outs**

**Inspection Areas:**

The curved faces in these 2 positions on each side of the engine must be inspected. In addition, approximately 5 mm coverage is required around the corners.



**Figure 2 - Inspection Areas**





### APPENDIX 1

This form to be completed for every engine/component inspected  
Results to be sent to Rolls-Royce via your Service Representative

Feedback Sheet TV124851	
Overhaul Base (if applicable)	
Operator	
Engine Mark	
Date of Inspection	
Engine Serial Number	
Engine Hours Since New	
Engine Cycles Since New	
Component Part Number	
Component Serial Number	
Component Hours Since New	
Component Cycles Since New	
	Inspection result and comments
On Wing Inspection of Mount Lug Run-outs (if applicable)	
In Shop Inspection of Mount Lug Run-outs (if applicable)	