

The solutions of this homework are entirely my own. I have discussed these problems with several classmates, they are: Shiming Liang and Yifei Shao

1. Discounted LQR

Consider the finite horizon discounted LQR problem

$$V_0^*(x_0) = \min_{x,u} \sum_{i=0}^{N-1} \alpha^i (x_i^T Q x_i + u_i^T R u_i)$$

$$\text{s.t. } x_{i+1} = A x_i + B u_i$$

with discount factor $\alpha \in (0, 1)$, $Q = Q^T, Q \succeq 0$, $R = R^T$ and $R \succ 0$

- (1) State the Bellman recursion (DP iteration) for this problem. [2 pts]
- (2) Assuming that the optimal cost-to-go at time $i+1$ is $V_{i+1}^*(x_{i+1}) = \alpha^{i+1} x_{i+1}^T P_{i+1} x_{i+1}$, find K_i as a function of P_{i+1} , such that the optimal input at time i is $u_i^*(x_i) = K_i x_i$ [4 pts]
- (3) Given that the optimal input at time i is $u_i^*(x_i) = K_i x_i$ and the optimal cost-to-go at time $i+1$ is $V_{i+1}^*(x_{i+1}) = \alpha^{i+1} x_{i+1}^T P_{i+1} x_{i+1}$, compute the matrix P_i such that the optimal cost-to-go at time i is $V_i^*(x_i) = \alpha^i x_i^T P_i x_i$ [4 pts]

Solution:

a. Starting from 1-step problem: solved at time $N-1$

$$V_{N-1}^*(x_{N-1}) = \min_{u_{N-1}} \alpha^{N-1} (x_{N-1}^T Q x_{N-1} + u_{N-1}^T R u_{N-1})$$

$$\text{s.t. } x_N = A x_{N-1} + B u_{N-1}$$

The optimal control input is derived by setting gradient of $V_{N-1}^*(x_{N-1})$ to be zero:

$$2R u_{N-1} = 0 \Rightarrow u_{N-1}^* = 0 = K_{N-1} x_{N-1}$$

Then the cost-to-go at time $N-1$, according to our notation convention, is:

$$V_{N-1}^*(x_{N-1}) = \alpha^{N-1} x_{N-1}^T Q_{N-1} x_{N-1} \triangleq \alpha^{N-1} x_{N-1}^T P_{N-1} x_{N-1}$$

Also note that $P_N = 0$ in this notation convention

b. Next, consider time step $N-2$

$$V_{N-2}^*(x_{N-2}) = \min_{u_{N-1}, u_{N-2}} \sum_{i=N-2}^{N-1} \alpha^{i+1} (x_i^T Q x_i + u_i^T R u_i)$$

$$\text{s.t. } x_{i+1} = A x_i + B u_i, \quad i = N-2, N-1$$

According to principle of optimality, since the solution at time $N-1$ is given, the above formulation is equivalent to:

$$V_{N-2}^*(x_{N-2}) = \min_{u_{N-2}} \alpha^{N-2} (x_{N-2}^T Q x_{N-2} + u_{N-2}^T R u_{N-2}) + \alpha^{N-1} x_{N-1}^T P_{N-1} x_{N-1}$$

$$\text{s.t. } x_{N-1} = A x_{N-2} + B u_{N-2}$$

The optimal control input is derived by setting gradient of $V_{N-2}^*(x_{N-2})$ to be zero:

$$2(\alpha^{N-1} B P_{N-1} B + \alpha^{N-2} R) u_{N-2} + 2\alpha^{N-1} B^T P_{N-1} A x_{N-2} = 0$$

$$\Rightarrow u_{N-2}^* = -\alpha (\alpha B^T P_{N-1} B + R)^{-1} B^T P_{N-1} A x_{N-2} = K_{N-2} x_{N-2}$$

Substituting u_{N-2}^* back into the expression, we can get that

$$\begin{aligned}
V_{N-2}^*(x_{N-2}) &= \alpha^{N-2} (x_{N-2}^T Q x_{N-2} + u_{N-2}^T R u_{N-2}) + \alpha^{N-1} (A x_{N-2} + B u_{N-2})^T P_{N-1} (A x_{N-2} + B u_{N-2}) \\
&= \alpha^{N-2} x_{N-2}^T (\alpha A^T P_{N-1} A + Q) x_{N-2} + 2\alpha^{N-2} x_{N-2}^T (\alpha A^T P_{N-1} B) u_{N-2} \\
&\quad + \alpha^{N-2} u_{N-2}^T (\alpha B^T P_{N-1} B + R) u_{N-2} \\
&= \alpha^{N-2} x_{N-2}^T \left[\alpha A^T P_{N-1} A + Q + \alpha^2 A^T P_{N-1} B (\alpha B^T P_{N-1} B + R)^{-1} B^T P_{N-1} A \right. \\
&\quad \left. - 2\alpha^2 A^T P_{N-1} B (\alpha B^T P_{N-1} B + R)^{-1} B^T P_{N-1} A \right] x_{N-2} \\
&= \alpha^{N-2} x_{N-2}^T \left[Q + \alpha A^T P_{N-1} A - \alpha^2 A^T P_{N-1} B (\alpha B^T P_{N-1} B + R)^{-1} B^T P_{N-1} A \right] x_{N-2} \\
&\triangleq \alpha^{N-2} x_{N-2}^T P_{N-2} x_{N-2}
\end{aligned}$$

c. Keep propagate backward, For any time step i , we can get the similar results as stated above. Therefore:

(1) The Bellman recursion (DP iteration) for this problem is:

$$V_i^*(x_i) = \min_{u_i} \alpha^i (x_i^T Q_i x_i + u_i^T R u_i) + V_{i+1}^*(x_i)$$

More specifically, starting from time $i = N - 1$, for any time step $i = n - 1, \dots, 1, 0$:

Given P_{i+1} , calculate the optimal control input:

$$u_i^* = K_i x_i = -\alpha (\alpha B^T P_{i+1} B + R)^{-1} B^T P_{i+1} A x_i$$

and the optimal value cost-to-go:

$$V_i^* = \alpha^i x_i^T P_i x_i \quad \text{where} \quad P_i = Q + \alpha A^T P_{i+1} A - \alpha^2 A^T P_{i+1} B (\alpha B^T P_{i+1} B + R)^{-1} B^T P_{i+1} A$$

Then keep propagate backward until time 0 (initial time)

(2) As is mentioned above, the expression of optimal gain matrix K_i as a function of P_{i+1} is:

$$K_i = -\alpha (\alpha B^T P_{i+1} B + R)^{-1} B^T P_{i+1} A$$

(3) As is mentioned above, the expression of P_i as a function of P_{i+1} is:

$$P_i = Q + \alpha A^T P_{i+1} A - \alpha^2 A^T P_{i+1} B (\alpha B^T P_{i+1} B + R)^{-1} B^T P_{i+1} A$$

2. Finite Horizon Optimal Control

Consider the discrete-time dynamic system with the following state space representation:

$$\begin{bmatrix} x_1(k+1) \\ x_2(k+1) \end{bmatrix} = \begin{bmatrix} 0.77 & -0.35 \\ 0.49 & 0.91 \end{bmatrix} \begin{bmatrix} x_1(k) \\ x_2(k) \end{bmatrix} + \begin{bmatrix} 0.04 \\ 0.15 \end{bmatrix} u(k) \quad (1)$$

We want to design a linear quadratic optimal control for this system with a finite horizon $N = 50$. We initially set the following cost matrices:

$$Q = \begin{bmatrix} 500 & 0 \\ 0 & 100 \end{bmatrix}, R = 1, P = \begin{bmatrix} 1500 & 0 \\ 0 & 100 \end{bmatrix}$$

and assume that the initial state is $x(0) = [1 \quad -1]^T$

(1) Determine the optimal set of inputs

$$U_0 = \begin{bmatrix} u_0 \\ u_1 \\ \vdots \\ u_{N-1} \end{bmatrix}$$

through the Batch Approach, i.e. by writing the dynamic equations as follows:

$$\begin{bmatrix} x_0 \\ x_1 \\ x_2 \\ \vdots \\ x_N \end{bmatrix} = \begin{bmatrix} I \\ A \\ A^2 \\ \vdots \\ A^N \end{bmatrix} x(0) + \begin{bmatrix} 0 & \cdots & \cdots & 0 \\ B & 0 & \cdots & 0 \\ AB & B & \cdots & 0 \\ \vdots & \ddots & \ddots & 0 \\ A^{N-1}B & \cdots & AB & B \end{bmatrix} \begin{bmatrix} u_0 \\ u_1 \\ u_2 \\ \vdots \\ u_{N-1} \end{bmatrix}$$

and using the formula:

$$U^*(x(0)) = -(\mathcal{S}^{uT} \bar{Q} \mathcal{S}^u + \bar{R})^{-1} \mathcal{S}^{uT} \bar{Q} \mathcal{S}^x x(0)$$

and calculate the optimal cost $J_0^*(x(0))$:

$$J_0^*(x(0)) = x(0)^T \left[\mathcal{S}^{xT} \bar{Q} \mathcal{S}^x x - \mathcal{S}^{xT} \bar{Q} \mathcal{S}^u (\mathcal{S}^{uT} \bar{Q} \mathcal{S}^u + \bar{R})^{-1} \mathcal{S}^{uT} \bar{Q} \mathcal{S}^x \right] x(0)$$

(2) Verify the results of the previous point by solving an optimization problem. In fact, the cost can be written as a function of U_0 as follows:

$$\begin{aligned} J_0(x(0), U_0) &= (\mathcal{S}^x x(0) + \mathcal{S}^u U_0)^T \bar{Q} (\mathcal{S}^x x(0) + \mathcal{S}^u U_0) + U_0^T \bar{R} U_0 \\ &= U_0^T H U_0 + 2x(0)^T F U_0 + x(0)^T \mathcal{S}^{xT} \bar{Q} \mathcal{S}^x x(0) \end{aligned}$$

where $H := \mathcal{S}^{uT} \bar{Q} \mathcal{S}^u + \bar{R}$ and $F := \mathcal{S}^{xT} \bar{Q} \mathcal{S}^u$, and then minimized by solving an unconstrained minimization problem. Check that the optimizer U_0^* and the optimum $J_0^*(x(0), U_0^*)$ correspond to the ones determined analytically in the previous point.

Note: Computing the optimal control trajectory in this way is a very basic version of model predictive control. More in the later lectures. [5 pts]

(3) Design the optimal controller through the recursive approach and determine the optimal state-feedback matrices F_k . Start from the Riccati Difference Equations, assuming that $P_N = P$, and compute recursively the P_k :

$$P_k = A^T P_{k+1} A + Q - A^T P_{k+1} B (B^T P_{k+1} B + R)^{-1} B^T P_{k+1} A \quad (2)$$

and then calculate F_k as a function of P_{k+1} :

$$F_k = - (B^T P_{k+1} B + R)^{-1} B^T P_{k+1} A x_k$$

Compare the optimal cost $J_0^*(x(0)) = x(0)^T P_0 x(0)$ with point (1) and check that they are equal. [5 pts]

- (4) We want to compare the robustness of the two approaches in simulations. Hence, let us add a process disturbance $Dw(k)$ to the right-hand side of Equation (1). Assume the matrix D to be:

$$D = \begin{bmatrix} 0.1 \\ 0.1 \end{bmatrix}$$

while the process $w(k)$ is a Gaussian white noise, with mean $m = 0$ and variance $\sigma^2 = 10$

Consider the simulation length to be equal to N time steps and that the input u_k to the system are defined as follows:

$$u_k = \begin{cases} U_0[k+1] & \text{for Batch Approach} \\ F_k x_k & \text{for Recursive Approach} \end{cases}$$

where $U_0[k]$ is the k -th component of the vector U_0 .

Plot a graph of the state evolution over time. What is the difference in the dynamic evolution? What happens if you modify the variance of the disturbance? Motivate your answer. [5 pts]

- (5) We assume now the horizon to be $N = 5$, and want to refine the definition of P . From the lectures we have seen two possible choices, corresponding to P solution of the ARE and of the Lyapunov equation.

Calculate the two possible final cost weight, P_{Ric} and P_{Lyap} or the recursive approach, then design two new optimal controllers. Simulate the dynamic behavior of the controller for N . steps with the code, considering no disturbance. Plot the evolution of the state variables and compare the two cases, motivating the results.

Solution:

The MATLAB code for solving all the problems are shown below, see code comments for more details

```

1      %% This is the script for ESE 6190 Homework 3 Ex2
2      clear all
3      clc
4
5      %% Basic Definition
6      % Define system matrices:
7      A = [0.77, -0.35; 0.49, 0.91];
8      % Define input matrices:
9      B = [0.04; 0.15];
10     % Horizon Length
11     N = 50;
12     % Define weight matrices:
13     Q = [500, 0; 0, 100];
14     R = 1;
15     P_N = [1500, 0; 0, 100];
16     % Define initial state:
17     x0 = [1; -1];
18
19     %% (1) Solution via Batch approach
20     [U_b, J_b] = UFHLQOC_Batch(A, B, N, Q, R, P_N, x0, "analytical");
21
22     %% (2) Solution via Batch approach, but use quadprog
23     [U_bopt, J_bopt] = UFHLQOC_Batch(A, B, N, Q, R, P_N, x0, "optimize");
24     %% verify the result
25     flag_U = norm(U_b - U_bopt) <= 1e-8;
26     flag_J = norm(J_b - J_bopt) <= 1e-8;
27
28     %% (3) Solution via Recursive approach
29     [U_r, J_r, F_r] = UFHLQOC_Recursive(A, B, N, Q, R, P_N, x0);
30     %% verify the result
31     flag_U_br = norm(U_b - U_r) <= 1e-8;
32     flag_J_br = norm(J_b - J_r) <= 1e-8;
33
34     %% (4) Simulate the disturbed dynamics

```

```

35 % Define disturbance process
36 rng(0)
37 n = size(A, 1);
38 D = [0.1;0.1];
39 w = sqrt(10)*randn(N);
40 x_b = zeros(n,N);
41 x_r = zeros(n,N);
42 % Simulate the system
43 for i = 0:N-1
44     if i == 0
45         x_b(:, i+1) = disturbed.system(x0, U_b(1), w(1), A, B, D);
46         x_r(:, i+1) = disturbed.system(x0, F_r(1,:)*x0, w(1), A, B, D);
47     else
48         x_b(:, i+1) = disturbed.system(x_b(:,i), U_b(i+1), w(i+1), A, B, D);
49         x_r(:, i+1) = disturbed.system(x_r(:,i), F_r(i+1,:)*x_r(:,i), w(i+1), A, ...
50                                     B, D);
51     end
52 end
53 x_b = [x0,x_b];
54 x_r = [x0,x_r];
55 t = 0:1:N;
56 % plot the figure
57 figure
58 plot(t, x_b(1,:), '-b', 'LineWidth', 2)
59 hold on
60 plot(t, x_b(2,:), '-r', 'LineWidth', 2)
61 legend('x_1b','x_2b',FontSize=12)
62 title('state evolution using bachth approach',FontSize=16)
63 xlabel('timestep k',FontSize=16)
64 ylabel('state',FontSize=16)
65 figure
66 plot(t, x_r(1,:), '-b', 'LineWidth', 2)
67 hold on
68 plot(t, x_r(2,:), '-r', 'LineWidth', 2)
69 legend('x_1r','x_2r',FontSize=12)
70 title('state evolution using recursive approach',FontSize=16)
71 xlabel('timestep k',FontSize=16)
72 ylabel('state',FontSize=16)
73 %% (5) Redesign the controller
74 % choose the terminal weight to be the solution of ARE
75 N = 5;
76 [P_ric,L_ric,G_ric] = idare(A,B,Q,R,0,eye(size(A)));
77 P_lya = dlyap(A,Q);
78 [U_ric,J_ric] = UFHLQOC.Batch(A, B, N, Q, R, P_ric, x0,"analytical");
79 [U_lya,J_lya] = UFHLQOC.Batch(A, B, N, Q, R, P_lya, x0,"analytical");
80 n = size(A, 1);
81 x_ric = zeros(n,N);
82 x_lya = zeros(n,N);
83 % Simulate the system
84 for i = 0:N-1
85     if i == 0
86         x_ric(:, i+1) = disturbed.system(x0, U_ric(1), 0, A, B, 0);
87         x_lya(:, i+1) = disturbed.system(x0, U_lya(1), 0, A, B, 0);
88     else
89         x_ric(:, i+1) = disturbed.system(x_ric(:,i), U_ric(i+1), 0, A, B, 0);
90         x_lya(:, i+1) = disturbed.system(x_lya(:,i), U_lya(i+1), 0, A, B, 0);
91     end
92 end
93 x_ric = [x0,x_ric];
94 x_lya = [x0,x_lya];
95 t = 0:1:N;
96 % plot the figure
97 figure
98 plot(t, x_ric(1,:), '-b', 'LineWidth', 2)
99 hold on
100 plot(t, x_ric(2,:), '-r', 'LineWidth', 2)
101 legend('x^{ric}_1','x^{ric}_2',FontSize=12)

```

```

102 title('state evolution with solution of ARE',FontSize=16)
103 xlabel('timestep k',FontSize=16)
104 ylabel('state',FontSize=16)
105 figure
106 plot(t, x_lya(1,:), '-b', 'LineWidth', 2)
107 hold on
108 plot(t, x_lya(2,:), '-r', 'LineWidth', 2)
109 legend('x^{lya}_1','x^{lya}_2',FontSize=12)
110 title('state evolution with solution of Lyapunov Equation',FontSize=16)
111 xlabel('timestep k',FontSize=16)
112 ylabel('state',FontSize=16)
113
114 function [U,J] = UFHLQOC.Batch(A, B, N, Q, R, P_N, x0, mode)
115     %% This is the function used to solve the unconstrained finite time
116     %% linear quadratic optimal control via batch approach
117     %% input:
118     % A: system matrix; B: Input matrix; N: control horizon;
119     % Q,R: weight matrices for states and inputs, respectively;
120     % P_N: terminal cost weight matrix; x0: initial condition
121     % mode: whether to use analytical method or optimization method.
122     %% output:
123     % U: the optimal control sequence U=[u0; u1; ...; uN-1];
124     % J: the optimal cost J=x0'*P0*x0
125
126     % first, rollout the dynamics to get Sx and Su:
127     n = size(A, 1);
128     m = size(B, 2);
129     Sx = zeros((N+1)*n, n);
130     Su = zeros((N+1)*n, N*m);
131     for i = 0:N
132         Sx(i*n+1:(i+1)*n, 1:n) = A^i;
133         for j = 0:N-1
134             Su(i*n+1:(i+1)*n, j*m+1:(j+1)*m) = (i-j-1 ≥ 0)*A^(i-j-1)*B;
135         end
136     end
137     % Then, get the stack-up weight matrices and rearrange as quadratic form
138     I = eye(N);
139     Q_bar = kron(I, Q);
140     Q_bar(end+1:end+n, end+1:end+n) = P_N;
141     R_bar = kron(I, R);
142     H = Su'*Q_bar*Su + R_bar;
143     F = Sx'*Q_bar*Sx;
144     % Calculate the optimal controller and optimal cost
145     if mode == "analytical"
146         U = -H^-1*F'*x0;
147         J = U'*H*U + 2*x0'*F'*U + x0'*Sx'*Q_bar*Sx*x0;
148     elseif mode == "optimize"
149         [U,J] = quadprog(2*H,2*F'*x0);
150         J = J + x0'*Sx'*Q_bar*Sx*x0;
151     end
152 end
153
154 %% functions for recursive approach
155 function [U,J,F] = UFHLQOC.Recursive(A, B, N, Q, R, P_N, x0)
156     %% This is the function used to solve the unconstrained finite time
157     %% linear quadratic optimal control via recursive approach
158     %% input:
159     % A: system matrix; B: Input matrix; N: control horizon;
160     % Q,R: weight matrices for states and inputs, respectively;
161     % P_N: terminal cost weight matrix; x0: initial condition
162     %% output:
163     % U: the optimal control sequence U=[u0; u1; ...; uN-1];
164     % J: the optimal cost J=x0'*P*x0
165     % F: the optimal gain matrices sequence F = [F0; F1; ...; FN-1]
166
167     % Start from the last time step k=N-1 to the initial state;
168     n = size(A, 1);
169     m = size(B, 2);

```

```

170     U = zeros(N, m);
171     F = zeros(N*m,n);
172     P = zeros(N*n,n);
173     J = zeros(N,1);
174     for i = N-1:-1:0
175         if i == N-1
176             P_prev = P_N;
177             F_curr = -(B'*P_prev*B+R)^(-1)*B'*P_prev*A;
178             F(i*m+1:(i+1)*m,1:n) = F_curr;
179             P_curr = A'*P_prev*A + Q + A'*P_prev*B*F_curr;
180             P(i*n+1:(i+1)*n,1:n) = P_curr;
181         else
182             P_prev = P((i+1)*n+1:(i+2)*n,1:n);
183             F_curr = -(B'*P_prev*B+R)^(-1)*B'*P_prev*A;
184             F(i*m+1:(i+1)*m,1:n) = F_curr;
185             P_curr = A'*P_prev*A + Q + A'*P_prev*B*F_curr;
186             P(i*n+1:(i+1)*n,1:n) = P_curr;
187         end
188     end
189     for i = 0:N-1
190         if i == 0
191             U(i*m+1:(i+1)*m) = F(i*m+1:(i+1)*m,1:n)*x0;
192             x = disturbed_system(x0,F(i*m+1:(i+1)*m,1:n)*x0,0,A,B,0);
193         else
194             U(i*m+1:(i+1)*m) = F(i*m+1:(i+1)*m,1:n)*x;
195             x = disturbed_system(x,F(i*m+1:(i+1)*m,1:n)*x,0,A,B,0);
196         end
197     end
198     J = x0'*P(1:n,1:n)*x0;
199 end
200
201 function xp = disturbed_system(xk, uk, wk, A, B, D)
202     %% This is the function used to simulate single-step dynamics of the diturbed ...
203     %% system
204     %% input:
205     %% A: system matrix; B: Input matrix; D: Disturbance matrix;
206     %% xk: current state, uk: current input, wk: gaussian white noise
207     xp = A*xk + B*uk + D*wk;
208 end

```

(1) The optimal controller and optimal cost calculated using bachth approach with analytical method:

```

1     U_b =
2         4.0818
3         -3.8039
4         -3.0017
5         -1.6477
6         -0.8106
7         -0.3813
8         -0.1757
9         -0.0801
10        -0.0364
11        -0.0165
12        -0.0075
13        -0.0034
14        -0.0015
15        -0.0007
16        -0.0003
17        -0.0001
18        -0.0001
19        -0.0000
20        -0.0000
21        -0.0000
22        -0.0000
23        -0.0000
24        -0.0000
25        -0.0000

```

```

26      -0.0000
27      -0.0000
28      -0.0000
29      -0.0000
30      -0.0000
31      -0.0000
32      -0.0000
33      -0.0000
34      -0.0000
35      -0.0000
36      -0.0000
37      -0.0000
38      -0.0000
39      -0.0000
40      -0.0000
41      -0.0000
42      -0.0000
43      -0.0000
44      -0.0000
45      -0.0000
46      -0.0000
47      -0.0000
48      -0.0000
49      -0.0000
50      -0.0000
51      -0.0000
52      J_b =
53      1.8723e+03

```

(2) The optimal controller and optimal cost calculated using bacth approach with optimization:

```

1      U_bopt =
2      4.0818
3      -3.8039
4      -3.0017
5      -1.6477
6      -0.8106
7      -0.3813
8      -0.1757
9      -0.0801
10     -0.0364
11     -0.0165
12     -0.0075
13     -0.0034
14     -0.0015
15     -0.0007
16     -0.0003
17     -0.0001
18     -0.0001
19     -0.0000
20     -0.0000
21     -0.0000
22     -0.0000
23     -0.0000
24     -0.0000
25     -0.0000
26     -0.0000
27     -0.0000
28     -0.0000
29     -0.0000
30     -0.0000
31     -0.0000
32     -0.0000
33     -0.0000
34     -0.0000
35     -0.0000
36     -0.0000

```



```

37         -0.0000
38         -0.0000
39         -0.0000
40         -0.0000
41         -0.0000
42         -0.0000
43         -0.0000
44         -0.0000
45         -0.0000
46         -0.0000
47         -0.0000
48         -0.0000
49         -0.0000
50         -0.0000
51         -0.0000
52     J_bopt =
53         1.8723e+03

```

We can see that the results we get are the same as what we get in (1) using batch approach with analytical method.

(3) The optimal cost using recursive method (dynamic programming):

```

1     J_r =
2         1.8723e+03

```

We can see that it is exactly the same as what we get in (1) and (2) using batch approach.

(4) Plot of the state evolution over time of the disturbed system:

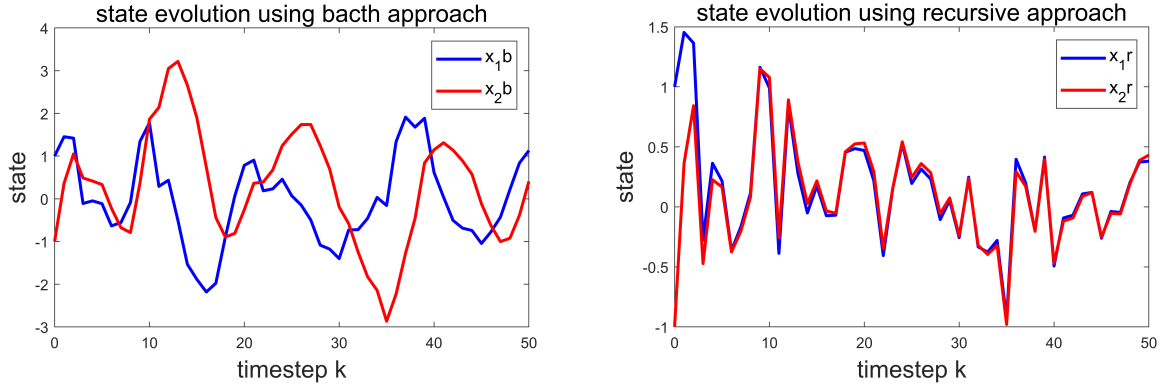


Figure 1: Evolution of the disturbed system using different approaches

From the figure we can see that recursive approach is much more robust than the batch approach. It still tend to converge even under disturbance. This is because batch approach generates the open loop control that relies on the prediction at the initial state, while recursive approach gives a closed-loop feedback policy at every time step. Therefore, when disturbance exists and largely influence the prediction of the state, batch approach can perform much better.

I also modify the variance of the disturbance, basically the results are very similar to the case where variance is 10 as shown above. Only when the variance become very small (near 0, i.e. constant disturbance rather than stochastic disturbance) could batch approach converges.

(5) Plot of the state evolution over time with different terminal cost weight matrices:

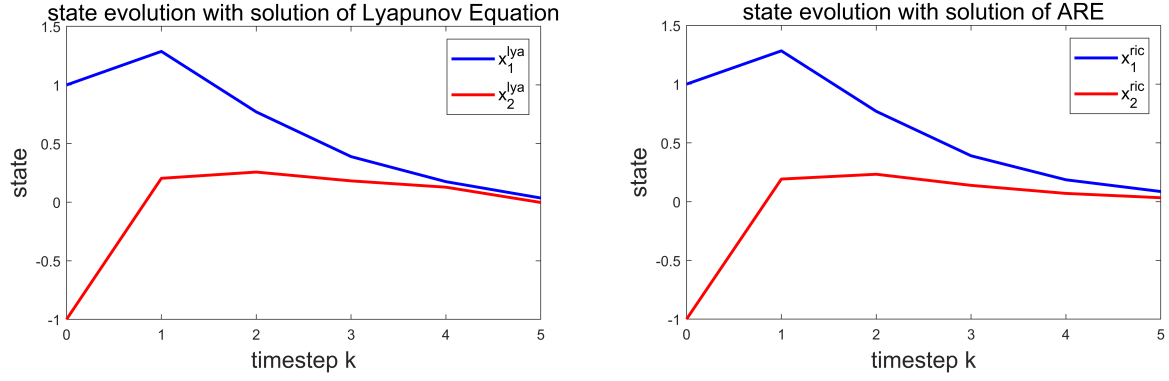


Figure 2: Evolution of the system with different terminal cost matrix

There is some differences between the two cases. Observing the matrices, final states and final control input, we can find that P_{Lyap} leads to smaller terminal state but larger control command compared to P_{Ric} . It is partly because using P_{Lyap} only make sense if the system is asymptotically stable, and hence it would hope to drive the state to original more aggressively, while using P_{Ric} is based on the assumption that no constraints will be active after the end of horizon, which is not that strict. I would say that these two cases don't show significant differences in this scenario.

3. Stability of LQR

Prove stability of an Infinite-Horizon LQR. Refer the slides of MPC chapter 5. [5 pts]

For infinite-horizon LQR, we assume that we already had a convergent Riccati Difference Equation with solution P_∞ :

$$P_\infty = A^T P_\infty A + Q - A^T P_\infty B (B^T P_\infty B + R)^{-1} B^T P_\infty A \quad (1)$$

and hence an valid optimal control law to be:

$$u_k^* = - (B^T P_\infty B + R)^{-1} B^T P_\infty A x_k = F_\infty x_k \quad (3)$$

Now, choose the candidate Lyapunov function $V(x_k) = x_k^T P_\infty x_k$

$$\textcircled{1} \quad V(0) = 0^T P_\infty 0 = 0$$

$$\textcircled{2} \quad P_\infty \succ 0 \Rightarrow x_k^T P_\infty x_k > 0, \quad \forall x_k \in \mathbb{R}^n \setminus \{0\}$$

$$\textcircled{3} \quad V(x_{k+1}) - V(x_k) = x_{k+1}^T P_\infty x_{k+1} - x_k^T P_\infty x_k = (Ax_k + Bu_k)^T P_\infty (Ax_k + Bu_k) - x_k^T P_\infty x_k$$

$$\stackrel{(2)}{\implies} x_k^T (A^T P_\infty A + A^T P_\infty B F_\infty + F_\infty^T B^T P_\infty A + F_\infty^T B^T P_\infty B F_\infty - P_\infty) x_k$$

$$\stackrel{(1),(2)}{\implies} x_k^T \left(\cancel{-Q + P_\infty - A^T P_\infty B F_\infty + A^T P_\infty B F_\infty} + F_\infty^T B^T P_\infty A + F_\infty^T B^T P_\infty B F_\infty - \cancel{P_\infty} \right) x_k$$

$$= x_k^T [-Q + F_\infty^T B^T P_\infty A + F_\infty^T (B^T P_\infty B + R) F_\infty - F_\infty^T R F_\infty]$$

$$\stackrel{(1)}{\implies} x_k^T \left[-Q + F_\infty^T B^T P_\infty A - \cancel{F_\infty^T (B^T P_\infty B + R) (B^T P_\infty B + R)^{-1} B^T P_\infty A x_k} - F_\infty^T R F_\infty \right] x_k$$

$$= x_k^T (-Q - F_\infty^T R F_\infty) x_k \stackrel{R \succ 0, Q \succeq 0}{\implies} < 0, \forall x_k \in \mathbb{R}^n \setminus \{0\}$$

Therefore, we prove that $V(x_k) = x_k^T P_\infty x_k$ (the infinite horizon cost) is the true Lyapunov function of the closed loop system and the system is stable.

4. Constrained Least Squares

Consider the least squares problem subject to linear constraints

$$\min \frac{1}{2} x^T Q x \quad \text{s.t.} \quad Ax = b$$

in which $x \in \mathbb{R}^n, b \in \mathbb{R}^p, Q \in \mathbb{R}^{n \times n}, Q \succeq 0, A \in \mathbb{R}^{p \times n}$. Show that this problem has a unique solution for every b and the solution is unique if and only if:

$$\text{rank}(A) = p, \text{rank} \begin{bmatrix} Q \\ A \end{bmatrix} = n$$

[5 pts]

Solution:

This is obviously a convex quadratic program, we can use KKT condition to find the least square solution we want, note that there are only equality constraints so the dual feasibility and complementarity slackness are not needed, we only need the stationary condition and primal feasibility, i.e.:

$$\begin{cases} Qx + A^T \nu = 0 \\ Ax = b \end{cases} \Rightarrow \begin{bmatrix} Q & A^T \\ A & 0 \end{bmatrix} \begin{bmatrix} x \\ \nu \end{bmatrix} = \begin{bmatrix} 0 \\ b \end{bmatrix}$$

And the coefficient matrix is also called the KKT matrix. A unique solution for every $b \in \mathbb{R}^p$ exists is equivalent to the KKT matrix to be nonsingular. Therefore, we are going to show that:

$$\begin{bmatrix} Q & A^T \\ A & 0 \end{bmatrix} \text{ nonsingular} \iff \text{rank}(A) = p, \text{rank} \begin{bmatrix} Q \\ A \end{bmatrix} = n$$

(Sufficiency \Rightarrow):

$$\begin{aligned} \begin{bmatrix} Q & A^T \\ A & 0 \end{bmatrix} \text{ nonsingular} &\iff \text{columns of } \begin{bmatrix} Q & A^T \\ A & 0 \end{bmatrix} \text{ linearly independent} \\ &\Rightarrow \text{columns of } \begin{bmatrix} A^T \\ 0 \end{bmatrix} \text{ linearly independent and} \\ &\quad \text{columns of } \begin{bmatrix} Q \\ A \end{bmatrix} \text{ linearly independent, respectively} \\ &\iff \text{rank} \begin{bmatrix} A^T \\ 0 \end{bmatrix} = p, \text{rank} \begin{bmatrix} Q \\ A \end{bmatrix} = n \iff \text{rank}(A) = p, \text{rank} \begin{bmatrix} Q \\ A \end{bmatrix} = n \end{aligned}$$

(Necessity \Leftarrow):

Firstly, the rank condition given is equivalent to the conditions as follows:

$$\begin{aligned} \text{rank} \begin{bmatrix} Q \\ A \end{bmatrix} = n &\iff x \in \mathbb{R}^n, \text{ if } \begin{cases} Qx = 0 \\ Ax = 0 \end{cases}, x = 0 \\ \text{rank}(A) = p &\iff \text{rank}(A^T) = p \iff y \in \mathbb{R}^p, \text{ if } A^T y = 0, y = 0 \end{aligned}$$

Now assume $\exists [w^T \ v^T]^T, w \in \mathbb{R}^n, v \in \mathbb{R}^p$ such that:

$$\begin{bmatrix} Q & A^T \\ A & 0 \end{bmatrix} \begin{bmatrix} w \\ v \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \Rightarrow \begin{cases} Qw + A^T v = 0 & \textcircled{1} \\ Aw = 0 & \textcircled{2} \end{cases}$$

Left-multiply w^T on both sides of $\textcircled{1}$, we get:

$$w^T Q w + w^T A^T v = 0 \xRightarrow{\textcircled{2}} \begin{cases} w^T Q w = 0 \\ Aw = 0 \end{cases} \xRightarrow{Q \succeq 0} \begin{cases} Qw = 0 \\ Aw = 0 \end{cases}$$

And consider one of the given rank condition, we actually can make inference as follows:

$$\exists \left[\begin{array}{c} w \\ v \end{array} \right], \left[\begin{array}{cc} Q & A^T \\ A & 0 \end{array} \right] \left[\begin{array}{c} w \\ v \end{array} \right] = \left[\begin{array}{c} 0 \\ 0 \end{array} \right] \Rightarrow \left[\begin{array}{c} Q \\ A \end{array} \right] w = 0 \left. \vphantom{\begin{array}{c} Q \\ A \end{array}} \right\} \begin{array}{c} \text{rank} \left[\begin{array}{c} Q \\ A \end{array} \right] = n \end{array} \Rightarrow w = 0$$

Still consider ①, given $w = 0$ we have:

$$Qw + A^T v = 0 \xrightarrow{w=0} A^T v = 0$$

Now consider the second given rank condition, the whole inference is as follows:

$$\exists \left[\begin{array}{c} w \\ v \end{array} \right], \left[\begin{array}{cc} Q & A^T \\ A & 0 \end{array} \right] \left[\begin{array}{c} w \\ v \end{array} \right] = \left[\begin{array}{c} 0 \\ 0 \end{array} \right] \Rightarrow \left[\begin{array}{c} Q \\ A \end{array} \right] w = 0 \left. \vphantom{\begin{array}{c} Q \\ A \end{array}} \right\} \begin{array}{c} \text{rank} \left[\begin{array}{c} Q \\ A \end{array} \right] = n \end{array} \Rightarrow w = 0 \Rightarrow \left. \begin{array}{c} A^T v = 0 \\ \text{rank} \left[\begin{array}{c} A^T \\ 0 \end{array} \right] = n \end{array} \right\} \Rightarrow v = 0$$

Therefore, in summary, this is stating that:

$$\exists \left[\begin{array}{c} w \\ v \end{array} \right], \left[\begin{array}{cc} Q & A^T \\ A & 0 \end{array} \right] \left[\begin{array}{c} w \\ v \end{array} \right] = \left[\begin{array}{c} 0 \\ 0 \end{array} \right] \Rightarrow \left[\begin{array}{c} Q \\ A \end{array} \right] w = 0 \left. \vphantom{\begin{array}{c} Q \\ A \end{array}} \right\} \begin{array}{c} \text{rank} \left[\begin{array}{c} Q \\ A \end{array} \right] = n, \text{rank} \left[\begin{array}{c} A^T \\ 0 \end{array} \right] = n \end{array} \Rightarrow \left[\begin{array}{c} w \\ v \end{array} \right] = \left[\begin{array}{c} 0 \\ 0 \end{array} \right] \Rightarrow \left[\begin{array}{cc} Q & A^T \\ A & 0 \end{array} \right] \text{ nonsingular}$$

With sufficiency and necessity proved, we hence prove that the KKT matrix is nonsingular if and only if the given rank condition holds.

5. Lagrange Multipliers

Consider the objective function $V(x) = \frac{1}{2}x^T Hx + h^T x$ and the optimization problem is:

$$\begin{aligned} \min_x \quad & V(x) \\ \text{subj. to} \quad & Dx = d \end{aligned}$$

in which $H \succ 0, x \in \mathbb{R}^m, d \in \mathbb{R}^m, m < n$. i.e. fewer constraints than decisions. Rather than partially solving for x using the constraint and eliminating it, we make use of the method of Lagrange multipliers for treating the equality constraints.

- (1) Show that the necessary and sufficient conditions are equivalent to the matrix equation,

$$\begin{bmatrix} H & -D^T \\ -D & 0 \end{bmatrix} \begin{bmatrix} x \\ \lambda \end{bmatrix} = - \begin{bmatrix} h \\ d \end{bmatrix}$$

The solution then provides the solution to the original problem.[6 pts]

Solution:

Since $H \succ 0$. This is obviously a strictly convex QP and the optimal solution can be decided by the KKT condition, note that there are only equality constraints so the dual feasibility and complementarity slackness are not needed, we only need the stationary condition and primal feasibility, i.e.:

$$\begin{cases} \nabla f - \nabla g^T \lambda = 0 \\ g = 0 \end{cases} \Rightarrow \begin{cases} Hx + h^T - D^T \lambda = 0 \\ Dx - d = 0 \end{cases} \Rightarrow \begin{bmatrix} H & -D^T \\ -D & 0 \end{bmatrix} \begin{bmatrix} x \\ \lambda \end{bmatrix} = - \begin{bmatrix} h \\ d \end{bmatrix}$$

- (2) We note one other important feature of the Lagrange multiplier, their relationship to the optimal cost of the purely quadratic case. For $h = 0$ the cost is given by:

$$V^0 = \frac{1}{2} (x^0)^T H x^0$$

Show that this can also be expressed in terms of λ^0 by the following:

$$V^0 = \frac{1}{2} d^T \lambda^0$$

Solution:

According to (1), with $h = 0$, we know that x^0 and λ^0 satisfy that:

$$\begin{cases} Hx^0 - D^T \lambda^0 = 0 \\ Dx^0 - d = 0 \end{cases} \Rightarrow \begin{cases} Hx^0 = D^T \lambda^0 \\ Dx^0 = d \end{cases}$$

Therefore, substituting the above relation into the objective, we get:

$$V^0 = \frac{1}{2} (x^0)^T H x^0 = \frac{1}{2} (x^0)^T D^T \lambda^0 = \frac{1}{2} d^T \lambda^0$$