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%Roshan Jaiswal- %Section - 03 %Aero 300 Lab 2	-Ferri - Advance Data Loading and Plotting: 4/12/24	
clear all;	%Clears all %Clears Workspace %Clears Command Window	

# **PART 1: Creating CFD Output Plots**

```
load("Data\Data\x.txt")
load("Data\Data\y.txt")
a = x.'; %Creates a transpose of the imported CFD Data
b = y.';
figure('name', 'CFD Data')
plot(x, y)
grid on
hold on
plot(a,b)
xlabel('Chord Length')
vlabel('Height')
title('CFD Data')
%Grid lines are not straight because they represent calculated airflow
%around the airfoil, which is disrupting the air to create lift
%The domain is larger to provide more data like freesteam air to compare to
%A drawback to having a larger domain is that you (or a computer) has to do
%more work to get results
%-----Plot 2-----
%Source 1
columnx = x(:,1);
columny = y(:,1);
%Source 2
c = linspace(0,1,100); %position along chord length
t = 12/100;
```

```
yt =
(5*t)*((0.2969*sqrt(c))-(0.1260*c)-(0.3516*c.^2)+(0.2843*c.^3)-(0.1015*c.^4))
%Source 3
n = load("Data\Data\n0012-i1.txt");
n1 = n./100;
figure('name', 'Airfoils')
plot(c,yt, 'r')
hold on
plot(columnx,columny,'g')
plot(abs(n1(:,1)), abs(n1(:,2)), 'b')
xlabel('Chord Lenth')
ylabel('Height')
xlim([0 1])
legend('4-Digit Equation', 'CFD', 'N0012-il Data')
%There are slight differences between the the airfoils because the CFD
%model was slightly edited from its original form
%-----Plot 3-----
%third plot countourf() (F fills it in)
m = load("Data\Data\mach.txt");
%k = m(1:3,1:2);
figure('name', 'Mach')
subplot(2,2,1)
contourf (x, y, m)
hold on
title('Whole Domain')
xlabel('Chord Length')
ylabel('Height')
subplot(2,2,2)
contourf(x,y,m)
xlim([0 1])
ylim([0 1])
title('Wing Focus')
xlabel('Chord Length')
ylabel('Height')
subplot(2,2,3)
contourf (x, y, m)
xlim([-.2.5])
ylim([0.5])
title('Leading Edge')
```

```
xlabel('Chord Length')
vlabel('Height')
subplot(2,2,4)
contourf(x,y,m)
xlim([.5 1.2])
ylim([0.5])
title('Trailing Edge')
xlabel('Chord Length')
ylabel('Height')
colorbar('eastoutside')
clim([0 2])
% ------Plot 4-----
p = load("Data\Data\pressure.txt");
figure('name', 'Pressure')
subplot(2,2,1)
surf(x,y,p)%contourf(x,y,p)
shading interp
view(2)
hold on
title('Whole Domain')
xlabel('Chord Length')
ylabel('Height')
ylim([0 1.75])
subplot(2,2,2)
surf(x,y,p)%contourf(x,y,p)
shading interp
view(2)
xlim([0 1])
ylim([0 1])
title('Wing Focus')
xlabel('Chord Length')
ylabel('Height')
subplot(2,2,3)
surf(x,y,p)%contourf(x,y,p)
shading interp
view(2)
xlim([-.2.5])
ylim([0.5])
title('Leading Edge')
xlabel('Chord Length')
ylabel('Height')
subplot(2,2,4)
surf(x,y,p)%contourf(x,y,p)
shading interp
view(2)
```

```
xlim([.5 1.2])
ylim([0.5])
title('Trailing Edge')
xlabel('Chord Length')
ylabel('Height')
colorbar('eastoutside')
clim([0 2])
% -----Plot 5-----
Vx = load("Data\Data\vx.txt");
Vy = load("Data\Data\vy.txt");
d = load("Data\Data\rho.txt");
f = linspace(1,length(d),length(d));
figure
subplot(2,2,1)
surf(x, y, d)
shading interp
view(2)
hold on
colorbar('eastoutside')
title('Density')
ylabel('Height')
xlim([-.5 1.5])
ylim([0 1.75])
subplot(2,2,2)
surf(x,y,p)
shading interp
view(2)
colorbar('eastoutside')
clim([0 2])
title('Pressure')
ylabel('Height')
xlim([-.5 1.5])
ylim([0 1.75])
subplot(2,2,3)
surf(x, y, Vx)
shading interp
view(2)
colorbar('eastoutside')
clim([0 2])
title('Vx')
ylabel('Height')
xlabel('Velocity')
xlim([-.5 1.5])
ylim([0 1.75])
subplot(2,2,4)
surf(x, y, Vy)
```

```
shading interp
view(2)
colorbar('eastoutside')
clim([0 2])
title('Vy')
ylabel('Height')
xlabel('Velocity')
xlim([-.5 1.5])
ylim([0 1.75])
%You can clearly see on both the density and the pressure plots
%that there is a 'hot spot' on the leading edge of the airfoil, and that
%relative velocity also follows density and pressure.
%-----Plot 6-----
C = load("Data\Data\DENSITY iteration.mat");
q = input('Press 1 to skip Animation, or press return'); %asking for user
input
if q == 1
    disp('Skipping...')
else
    figure
    for k = 1:799 %Moving thru iterations
        h = C.C{1,k}; %Reading data from proper cell
        surf(h) %creating surf plot
        drawnow %redrawing in figure
    end
end
```

### **PART 2: File Manipulation**

```
if u == 1 %Deleting
    disp('Deleting...')
    delete("temp.txt")
    disp('Done')
end
```

## **PART 3: Vector Plotting**

```
xD = linspace(-2*pi, 2*pi, 30); %Creating domains
yD = linspace(-1, 1, 30);
sLine = 30; %Amount of streamlines
w = input('Would you like a streamline plot, quiver plot, or both?','s');
s1 = 'streamline';
s2 = 'quiver';
s3 = 'both';
if strcmp(w,s1) == 1 % strcmp compares strings to see if they are =
    type = 's';
elseif strcmp(w, s2) == 1
    type = 'q';
elseif strcmp(w, s3) == 1
    type = 'b';
else
    disp('Invalid Input')
    disp('Please type "streamline", "quiver", or "both".')
    type = 'f';
end
vPlotter(type,xD,yD,sLine) %Calling function plots
type = 's';
vPlotter(type,xD,yD,sLine) %Calling the function 3 times instead of input
type = 'q';
vPlotter(type,xD,yD,sLine)
type = 'b';
vPlotter(type,xD,yD,sLine)
```

#### **Function**

```
function [] = vPlotter(type, xD, yD, sLine)
[XX, YY] = meshgrid(xD, yD);
fx = XX;
fy = sin(XX);
sStartX = ones(1, sLine); %Creating starting stream lines alon 1 & -1
sStartX2 = ones(1, sLine) *-1;
sStartY = linspace(-1,1, sLine); %Creating starting y vector

    if type == 's' %Literally just a bunch of if statements graphing
required plot
        figure('Name', 'Streamlines')
        streamline(XX, YY, fx, fy, sStartX, sStartY);
```

```
hold on
   streamline(XX,YY,fx,fy,sStartX2,sStartY)
elseif type == 'q'
   figure('name','Quiver')
   quiver(XX,YY,fx,fy)
elseif type == 'b'
   figure('name','Quiver & Streamline')
   quiver(XX,YY,fx,fy)
   hold on
   streamline(XX,YY,fx,fy,sStartX,sStartY);
   streamline(XX,YY,fx,fy,sStartX2,sStartY)
elseif type == 'f'
   disp('You broke the code :(')
else
   disp('idek how u got here lol')
end
```

end

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