Table of Contents

```
%Roshan Jaiswal-Ferri
%Section - 01
%Aero 320 HW 4 - Problem 2: 10/21/24
```

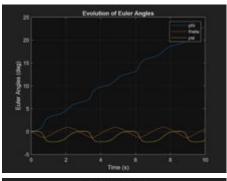
Workspace Prep

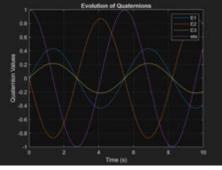
PART 1:

```
w21 = [1; -2; .5]; %angular vel
q0 = [0; 0; 0; 1]; %initial quaternion
eulA = [0; 0; 0]; %initial euler angles of 0
time = [0, 10]; %sec
[t1, q] = ode45(@quaternion, time, q0, [], w21); %
[t2, e] = ode45(@eula, time, eulA, [], w21);
figure ('Name', 'Evolution of Euler Angles')
plot(t2, e(:,1), t2, e(:,2), t2, e(:,3));
grid on
xlabel('Time (s)');
ylabel('Euler Angles (deg)');
legend('phi', 'theta', 'psi');
title('Evolution of Euler Angles');
figure ('Name', 'Evolution of Quaternions')
plot(t1, q(:,1), t1, q(:,2), t1, q(:,3), t1, q(:,4));
grid on
xlabel('Time (s)');
ylabel('Quaternion Values');
legend('E1', 'E2', 'E3', 'eta');
title('Evolution of Quaternions');
```

Functions:

```
function [dq] = quaternion(t,x,o)
    ep = x(1:3,1);
```





Published with MATLAB® R2023b