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Assg 1 AER121B

Date:

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- (1.)
- Symmetrical Aerofoil
  - $\alpha = 3^\circ = 0.05233 \text{ Rad}$
  - $U_\infty = 40 \text{ m/sec}$
  - $c = 1 \text{ m}$

$$\rho_\infty = 1.225 \text{ kg/m}^3$$
$$\mu = 1.789 \times 10^{-5} \frac{\text{kg m}}{\text{s}}$$

To find  $\rightarrow$  Lift,  $C_m|_{x=0}$ , Re

According to Thin Airfoil theory,

For Unumbered Airfoil,

$$L' = \rho_\infty U_\infty \Gamma = \rho_\infty U_\infty^2 \pi \alpha c$$

$$= 1.225 \times (40)^2 \times 3.14 \times 0.05233 \times 1$$

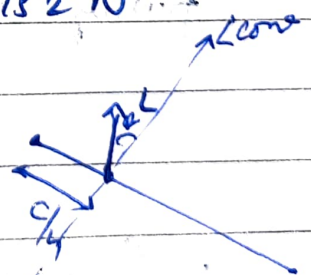
$$\Rightarrow 322.059752 \text{ N}$$

$$M|_{x=0} = (L' c) \times \frac{1}{4}$$

$$M_{LE} = \frac{-L}{4}$$

$$\Rightarrow L \times \frac{1}{4}$$

$$\Rightarrow \underline{80.5149 \text{ Nm}}$$



Also Can be derived from

$$M_{LE} = -\rho_\infty U_\infty \int_0^c x \gamma(x) dx$$

(2.) • Circular Line

$$\left(x - \frac{c}{2}\right)^2 + \left(z + \frac{c}{8K} - \frac{Kc}{2}\right)^2 = \left(\frac{c}{8K} + \frac{Kc}{2}\right)^2$$

•  $V_{\infty}, \alpha$  known

•  $K < 1$

•  $\gamma = f(V_{\infty}, \alpha, \theta, K)$

$$\rightarrow \text{Bare } \xi^h \rightarrow \frac{1}{2\pi} \int_0^c \frac{\gamma(\xi) d\xi}{x - \xi} = V_{\infty} \left( \alpha - \frac{dz}{dx} \right)$$

$$\text{putting } \xi \Rightarrow \frac{c}{2} (1 - \cos \theta)$$

$$x \Rightarrow \frac{c}{2} (1 - \cos \theta_0)$$

$$\frac{1}{2\pi} \int \frac{\gamma(\theta) \sin \theta d\theta}{\cos \theta - \cos \theta_0} = V_{\infty} \left( \alpha - \frac{dz}{dx} \right)$$

Assuming  $\gamma(\theta) = 2V_{\infty} \left[ A_0 \left( \frac{1 + \sin \theta}{\cos \theta} \right) + \sum_{n=1}^{\infty} A_n \sin(n\theta) \right]$

Subst in  $\xi^n$

$$\frac{A_0}{\pi} \int_0^{\pi} \frac{1 + \cos \theta}{\cos \theta - \cos \theta_0} d\theta + \sum_{n=1}^{\infty} \frac{A_n}{\pi} \int_0^{\pi} \frac{\sin n\theta \sin \theta}{\cos \theta - \cos \theta_0} d\theta$$

$$= \alpha - \frac{dz}{dx}$$

Solving

$$A_0 - \sum_{n=1}^{\infty} A_n \cos n\theta_0 = \alpha - \frac{dz}{dx}$$

$$(\alpha - A_0) + \sum_{n=1}^{\infty} A_n \cos n\theta_0 = \frac{dz}{dx}$$

$B_0$        $B_1 \rightarrow B_n$       Fourier Exp<sup>n</sup>

$$A_0 = \alpha - \frac{1}{\pi} \int_0^{\pi} \frac{dz}{dx} d\theta_0 ; A_n = \frac{2}{\pi} \int_0^{\pi} \frac{dz}{dx} \cos n\theta_0 d\theta$$



~~Star~~ Circular Arc Eq<sup>n</sup>



$$\underbrace{\left(x - \frac{c}{2}\right)^2}_{x'} + \underbrace{\left(z + \frac{c}{8K} - \frac{Kc}{2}\right)^2}_A = \underbrace{\left(\frac{c}{8K} + \frac{Kc}{2}\right)^2}_B$$

$$(x')^2 + (z + A)^2 = B^2$$

$$z = -A + \sqrt{B^2 - (x')^2} \quad \left[ -ve \text{ ignored} \right]$$

$$\frac{dz}{dx} = \frac{1 \times (-2x') \times \frac{dx'}{dx}}{2\sqrt{B^2 - (x')^2}}$$

$$\Rightarrow \frac{-x'}{\sqrt{B^2 - (x')^2}} \quad \left| \begin{array}{l} x' = x - \frac{c}{2} \\ x' = \frac{c}{2}(1 - \cos\theta_0) - \frac{c}{2} \\ \Rightarrow \frac{c}{2}(1 - \cos\theta_0 - 1) \\ x' \Rightarrow -\frac{c}{2} \cos\theta_0 \end{array} \right.$$

$$\frac{dz}{dp} \Rightarrow \frac{\frac{c}{2} \cos \theta_0}{\sqrt{B^2 - \frac{c^2}{4} \cos^2 \theta_0}}$$

$$\Rightarrow \frac{\frac{c}{2} \cos \theta_0}{\frac{1}{2}}$$

$$\sqrt{\left(\frac{c}{8K}\right)^2 + \left(\frac{Kc}{2}\right)^2 - \frac{c^2}{4} \cos^2 \theta_0}$$

$$\Rightarrow \frac{c \cos \theta_0}{\frac{1}{8K}}$$

$$\frac{dz}{dp} \Rightarrow 4K \cos \theta_0$$

$$\therefore A_0 \Rightarrow \alpha - \frac{1}{\pi} \int_0^\pi 4K \cos \theta_0 d\theta \Rightarrow \alpha$$

$$A_1 \Rightarrow \frac{2}{\pi} \int_0^\pi 4K \cos \theta_0 \times \cos \theta_0 d\theta \Rightarrow 4K$$

$$A_2, A_3, \dots, A_n \rightarrow 0 ; K \ll 1 \rightarrow \text{Camber small.}$$

$$r(\theta) \Rightarrow \left[ 2V_0 \left( \alpha \left( \frac{1 + \sin \theta}{\cos \theta} \right) + 4K \sin \theta \right) \right]$$

$$C_m|_{q_n} \Rightarrow -\frac{\pi}{4} A_1 \Rightarrow -5K \sin \theta$$

$$\alpha_{L=0} \Rightarrow -\frac{1}{\pi} \int_0^{\pi} \frac{dz}{dx} (\cos \theta - 1) d\theta$$

$$\Rightarrow -\frac{1}{\pi} \times \left( \frac{\pi}{2} \right) \times \left( \frac{2K}{2} \right)$$

$$\Rightarrow \underline{\underline{-2K}}$$

- (3) • Weight  $\rightarrow 60,000 \text{ N}$   
• Span  $\rightarrow 12 \text{ m}$   
• Speed  $\rightarrow 70 \text{ m/s}$

1. ~~Diagram~~ Sine Elliptical loading

$d_i \rightarrow$  Count for  $\gamma$ ,  $\Gamma = \Gamma_0 \sqrt{1 - \left(\frac{y}{b}\right)^2}$

$\therefore D_i \Rightarrow \frac{L^2}{90 \pi b^2}$

$\Rightarrow \frac{(60,000)^2}{(1.225 \times 70) \times \pi \times (12 \times 12)}$

$\Rightarrow \underline{\underline{92,848.786 \text{ N}}}$

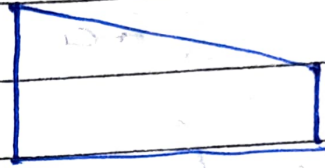


(4.)

$$AR = 8$$

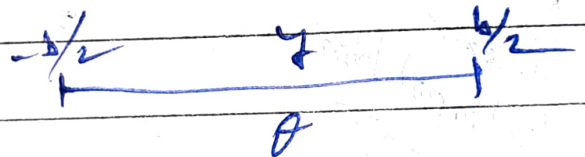
$$C_{tip} = 0.4 C_d$$

$$\underline{\underline{\alpha_{L=0} = -1.2}}$$



$$\alpha_{eff} = \frac{\Gamma(\omega)}{\pi V_{\infty} C(\omega)} + \alpha_{L=0}(\omega)$$

$$\Gamma(\omega) = 2bV_{\infty} \sum_{n=1}^N A_n \sin(n\omega)$$



$$\alpha(\omega) = \underbrace{\frac{2b}{\pi C(\omega)} \sum_{n=1}^N A_n \sin(n\omega)}_{\alpha_{eff.}} + \alpha_{L=0}(\omega)$$

$$+ \underbrace{\sum_{n=1}^N n A_n \frac{\sin(n\omega)}{\sin \omega}}_{\alpha'_0}$$

$$\alpha = \frac{2b}{\pi C(\omega)}$$

(talking for  $\underline{\underline{N=8}}$ )



$$(1) \left\{ A_1 \left( \frac{2b \sin \theta}{\pi c} + 1 \right) + A_2 \left( \frac{2b \sin 3\theta}{\pi c} + \frac{3 \sin 3\theta}{\sin \theta} \right) \right.$$

$$\left. + A_3 \left( \frac{2b \sin 5\theta}{\pi c} + \frac{5 \sin 5\theta}{\sin \theta} \right) + A_4 \left( \frac{2b \sin 7\theta}{\pi c} + \frac{7 \sin 7\theta}{\sin \theta} \right) \right\}$$

Variable  $c(\theta_0) \Rightarrow c_R \left( 1 - \frac{3}{4} \times \frac{(\pi/2 - \theta)}{\pi/2} \right)$

Direct Linear Interpolation

$$= c_R \left( 1 - \frac{3}{4} \times \frac{\pi/2 - \theta}{\pi} \right) \quad (2)$$

After Inputting this formulae of c(2) into (1) &

Substituting  $\pi/8, \pi/4, 3\pi/8, \pi/2$ , we get the matrix  
(Done in Matlab)

3.118	14.771	19.599	10.118	$A_1$	$x+b_2$
4.033	7.033	-9.033	-11.033	$A_2$	$x+b_2$
4.053	-2.941	-3.750	11.053	$A_3$	$x+b_2$
3.565	-6.565	8.565	-10.825	$A_4$	$x+b_2$

$$[A] \begin{bmatrix} A_1 / (\alpha + 1.2^\circ) \\ A_2 / (\alpha + 1.2^\circ) \\ A_3 / (\alpha + 1.2^\circ) \\ A_n / (\alpha + 1.2^\circ) \end{bmatrix} = \begin{bmatrix} 1 \\ 1 \\ 1 \\ 1 \end{bmatrix} \rightarrow B$$

$$\begin{bmatrix} A_1 \\ A_2 \\ A_3 \\ A_n \end{bmatrix} = (\alpha + 1.2^\circ) [A] [B]$$

$$\Rightarrow (\alpha + 1.2^\circ) \begin{bmatrix} 2.599 \times 10^{-1} \\ 2.2275 \times 10^{-2} \\ 8.6681 \times 10^{-3} \\ -1.3048 \times 10^{-3} \end{bmatrix}$$

$$\left( \alpha + 1.2 \times \frac{\pi}{180} \right)$$

(At all places)

$$L = \int_{-b/2}^{b/2} L(y) dy = \rho_\infty U_\infty \int_{-b/2}^{b/2} \Gamma(y) dy$$

$$\Rightarrow \rho_\infty U_\infty \int_0^\pi \Gamma(\theta) d\theta$$

$$\frac{2b}{\pi e} \times \rho_\infty U_\infty \int_0^\pi (A_1 \sin \theta + A_2 \sin 3\theta + A_3 \sin 5\theta + A_4 \sin 7\theta) d\theta \times (\alpha + 1.2^\circ)$$

$$C_L = A_1 \pi R$$

$$\Rightarrow (0.2599) \times (3.14) \times (8) \times (d + 1.2) \times \frac{\pi}{180}$$

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$$\Rightarrow \underline{\underline{0.4783}}$$

$$\left( \text{for } \alpha = 3^\circ \right)$$

$$C_{Di} \Rightarrow \frac{C_L^2}{\pi R e} ; e = \frac{1}{1 + 8} ; \delta = \sum_{n=2}^N \left( \frac{A_n}{A_m} \right)^2$$

$$\sum_{n=2}^N \left( \frac{A_n}{A_m} \right)^2 \Rightarrow 12.097 \times 10^{-4}$$

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$$\therefore e \approx 1$$

$$C_{Di} \Rightarrow \frac{(0.4783)^2}{(3.14) \times 8 \times 1} \Rightarrow \underline{\underline{0.0092}}$$