The University of Derby Faculty of Arts, Design and Technology

Efficient Acoustic Modelling of Large Spaces using Time Domain Methods

Analysis of Time Domain Numerical Methods for Acoustic Modelling of Large Spaces

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April 12, 2017

Submitted for in part-fulfilment of the requirements for the MSc in Audio Engineering.



Acknowledgements

I would like to dedicate this work to anyone of remote importance.

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Acronyms

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ABC Spelled-out abbreviation and definition BABI Spelled-out abbreviation and definition CABR Spelled-out abbreviation and definition

Chapter 1 Introduction

The intro Text

1.1 Context

1.2 Problem Definition

Real time acoustic modelling could be of significant benefit to many applications; Engineers could make design changes and see results 'on the fly', and entertainment users could have more realistic experiences. These benefits should be possible for an arbitrary number of sources and receivers, in proportionally large environments with high quality results. Is it possible to further reduce computation time for simulations of large acoustic problems, to provide results in real time for the full human audio frequency range? There are two 'branches' of computation solution that should be considered: the direct solution i.e. direct outputs or audio samples from the simulation, and indirect solutions i.e. a system impulse response that may be convolved with mixed source signals in order to create an auralization of the system.//

Fig. 1.1 A visualisation of a 2D explicit FDTD simulation [?]

Chapter 2

Loudspeaker Systems & Large Room Acoustics

Abstract Acoustics is a branch of physics that aims to characterise Newton's law of motion applied to wave propagation, while obeying the physical conservation law and often focussing on propagation in an audible spectrum. This characterisation of sound propagation is intrinsically linked to many other branches of physics, as well as psychoacoustics and perception. Many aspects of acoustic modelling may be of interest when considering the design and application of loudspeaker systems. Both small and large scale simulations may allow a user to make informed decisions about the design and placement of a loudspeaker system, so that the performance of the system may be validated and optimised before application. In this chapter we will evaluate the lossless acoustic wave equation for gasses, and consider the application of the wave equation in bounded space. We will then consider some specific use cases for applying such an equation for modelling loudspeaker system performance.

2.1 The Acoustic Wave Equation

In the Mcgraw-Hill Electronic and Electrical Engineering Series of books, the late L Beranek authored the Acoustics volume. This volume contains an elegant summary of the wave equation, that will be the subject of paraphrase in the following section.

2.1.1 The Wave Equations

Acoustic waves are classified as fluctuations of pressure in a given medium. In room acoustics and loudspeaker system engineering, these fluctuations are often cyclical in nature around an ambient pressure, as opposed to the jets described in aeroacoustic study. Similar to the behaviour of heat convection or fluid diffusion, these cyclical fluctuations propagate and spread through the medium of interest. As these fluctuations of pressure propagate energy is often lost, and eventually the medium will often come to a state of relative rest where the energy of the propagating waves have been almost entirely dissipated. It is possible to calculate an approximate solution the the propagation of pressure through a space, by solving a system of second

order partial differential equations that can be collectively lumped into a 'Wave Equation'. Below, we will introduce the three building blocks of the wave equation in both one dimension, and three dimensions (based on vector notation). These building blocks are Newton's Second Law of Motion, the gas law, and the laws of conservation of mass.

To consider the wave equation, we should use the analogy of a small¹ volume of gas, within a larger homogeneous medium. The faces of the volume are frictionless, and only the pressure at any face impacts on the gas inside the volume.

One Dimension	Three Dimensions
medium like a plane wave, from one side	Sound pressure <i>p</i> propagates across the medium like a spherical wave, from one
	side to the other at a rate of grad $p = \mathbf{i} \frac{\delta p}{\delta x} + \mathbf{j} \frac{\delta p}{\delta y} + \mathbf{k} \frac{\delta p}{\delta z}$ where i, j and k are unit vectors in the directions x, y and z.
	Force acting on the volume in the positive x direction can thus be described as $-[i(\frac{\delta p}{\delta x}\Delta x)\Delta y\Delta z) + j(\frac{\delta p}{\delta y}\Delta y)\Delta x\Delta z) + k(\frac{\delta p}{\delta z}\Delta z)\Delta x\Delta y)]$
A positive gradient causes acceleration in the $-x$ direction	←
	Force per unit volume is given by dividing both sides of the previous equation by the volume V , $\frac{f}{V} = -\mathbf{grad}p$
Newton's second law of motion dictates that the rate of change of momentum in the volume must balance with force per unit volume, and we can assume the mass of gas in the volume is constant.	
	The force mass balance can be described as $\frac{f}{V} = -\mathbf{grad}p = \frac{M}{M}\frac{Dq}{Dt} = \rho'\frac{Dq}{Dt}$

¹ rectilinear

| u is the velocity of gas in the volume, ρ' | where q is the vector velocity, ρ' is the is the density of the gas, and $M = \rho'V$ is density of gas in the volume, $M = \rho'V$ the mass of gas in the volume.

is the total mass of gas in the volume. $\frac{D}{Dt}$ represents the total rate of change of velocity of a section of gas in the volume, and can be composed as $\frac{Dq}{Dt} = \frac{\delta q}{\delta t} + q_x \frac{\delta q}{\delta x} + q_y \frac{\delta q}{\delta y} + q_z \frac{\delta q}{\delta z}$ where q_x , q_y and q_z are the components of the particle velocity q in each direction. As this is a linear wave equation approximation, these velocity components have no cross terms.

If the change in density of gas in the vol-If the particle velocity vector is suffiabove to $-\frac{\delta p}{\delta x} = \rho_0 \frac{\delta u}{\delta t}$

ume is sufficiently small, the ρ' will be ciently small, the change of momentum of approximately equal to the average den-the gas is approximately the same as the sity ρ_0 , thus simplifying the equations momentum of the volume at any arbitrary point, and the density of gas within the volume ρ' will be approximately equal to the average density ρ_0 . Thus the above can be written as $-grad p = \rho_0 \frac{\delta q}{\delta t}$

This kind of approximation may be appropriate as long as the maximum pressure is appropriately low, so that the behaviour of the air is linear, often quoted to be at or under the threshold of pain for human hearing or 120dB SIL.

Assuming that the gas of the volume is ideal, then the gas law PV = RT should hold true. Here, T is the temperature in degrees Kelvin, and R is a constant based on the mass of the gas. For this approximation we assume that the system is adiabatic, and that T and R are lumped into a gas constant which for air is $\gamma = 1.4$.

In differential form, the relationship between pressure and volume for an adiabatic expansion the volume is $\frac{dP}{P}$ = $\frac{-\gamma dV}{V}$ i.e. changes in pressure scale with changes in volume by this γ value.

If perturbations in pressure and volume due to a sound wave, p for pressure and τ for volume respectively, are sufficiently small compared to the rest values P_0 and V_0 : the time based derivative of the above equation can be written as follows: $\frac{1}{P_0} \frac{\delta p}{\delta t} = \frac{-\gamma}{V_0} \frac{\delta \tau}{\delta t}$

As the wave equation being derived is concerned with the transport of pressure within a volume, a continuity expression must be applied. The conservation of mass states that the total mass of gas in the volume must remain constant. This conservation law brings a unique relationship between discrete velocities at the boundary of the volume:

is displaced with a velocity, in a given as $\tau = V_0 \operatorname{div} \varepsilon$ time step the particles at the right hand boundary must also be displaced. This can be written as $\varepsilon_x + \frac{\delta \varepsilon_x}{\delta x} \Delta x$ The difference between this velocity and a subsequent change in volume τ multiplied by the volume gives $\tau = V_0 \frac{\delta \varepsilon_x}{\delta r}$.

The one dimensional wave equation in The three dimensional wave equation in $\left| \frac{\delta p}{\delta t} = -\gamma P_0 \frac{\delta u}{\delta x} \right|$

If the volume is displaced by some rate If the mass of gas within the box must ε_x , air particles at either boundary of the remain constant, the vector displacement volume must be displaced at an equal rate will directly change the volume by some for the mass of the volume to remain con-rate, as the two must balance to satisfy the stant. As such if the left side of the volume continuity equation. This can be written

Differentiating this with respect to time gives: $\frac{\delta \tau}{\delta t} = V_0 \frac{\delta u}{\delta x}$ where u is the instantaneous particle velocity

Differentiating this with respect to time gives: $\frac{\delta \tau}{\delta t} = V_0 \ div \ q$ where q is the instantaneous particle velocity

rectangular coordinates can be created by rectangular coordinates can be created by combining the above statements about the combining the above statements about the equation of motion, the gas law and the equation of motion, the gas law and the continuity equation. The combination of continuity equation. The combination of the gas law and continuity equation gives the gas law and continuity equation gives $\frac{\delta p}{\delta t} = -\gamma P_0 div \mathbf{q}$

When differentiated with respect to time, When differentiated with respect to time this gives: $\frac{\delta^2 p}{\delta t^2} = -\gamma P_0 \frac{\delta^2 u}{\delta t \delta x}$

this gives: $\frac{\delta^2 p}{\delta t^2} = -\gamma P_0 div \frac{\delta q}{\delta t}$

Differentiating the momentum equation The divergence of the momentum equaderived above with respect to time gives tion derived above gives: $-div = \rho_0 div \frac{\delta q}{\delta t}$ $-\frac{\delta^2 p}{\delta t^2} = \rho_0 \frac{\delta^2 u}{\delta r \delta t}$

Replacing the divergence (gradp) term with the Lapacian operator $\nabla^2 p$ produces $-\nabla^2 p = \rho_0 div \frac{\delta^2 p}{\delta t}$

 $\frac{\delta^2 p}{\delta x^2} = \frac{\rho_0}{\gamma P_0} \frac{\delta^2 p}{\delta t^2}$

Combining the above equations gives: Combining the above equations gives: $\nabla^2 p = \frac{\rho_0}{\gamma P_0} \frac{\delta^2 p}{\delta t^2}$

If we define c as the speed of propagation \leftarrow in the medium of interest, then $c^2 \approx \frac{\gamma P_0}{\rho_0}$ due to the fact that the speed of sound $c \approx (1.4 \frac{10^5}{1.18})^{\frac{1}{2}}$ where the ambient air pressure at sea level is $10^5 Pa$, 1.4 is the adiabatic constant γ (ratio of specific heats) for air, and ρ_0 is the density of air is approximately $1.8kg/m^3$

Finally we find that the 1 dimensional Finally we find that the 3 dimensional wave equation is: $\frac{\delta^2 p}{\delta x^2} = \frac{1}{c^2} \frac{\delta^2 p}{\delta t^2}$

wave equation is: $\nabla^2 p = \frac{1}{c^2} \frac{\delta^2 p}{\delta t^2}$ An explicit 3 dimensional expression of the pressure component of this equation is: $\nabla^2 p = \frac{\delta^2 p}{\delta x^2} + \frac{\delta^2 p}{\delta x^2} + \frac{\delta^2 p}{\delta z^2}$

This equation can also be expressed in This equation can be expressed velocity terms of the instantaneous velocity in the vector $\nabla^2 q = \frac{1}{c^2} \frac{\delta^2 q}{\delta t^2}$ where $\nabla^2 q$ represents the gradient of pressure (velocity) in the volume.

2.2 Loudspeaker Systems

Instead of simply listing headings of different levels we recommend to let every heading be followed by at least a short passage of text. Furtheron please use the LATEX automatism for all your cross-references and citations.

Please note that the first line of text that follows a heading is not indented, whereas the first lines of all subsequent paragraphs are.

Use the standard equation environment to typeset your equations, e.g.

$$a \times b = c \,, \tag{2.1}$$

however, for multiline equations we recommend to use the equarray environment².

$$a \times b = c$$

$$\mathbf{a} \cdot \mathbf{b} = \mathbf{c}$$
(2.2)

2.2.1 Subsection Heading

Instead of simply listing headings of different levels we recommend to let every heading be followed by at least a short passage of text. Furtheron please use the LaTeX automatism for all your cross-references and citations as has already been described in Sect. 2.2.

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2.2.1.1 Subsubsection Heading

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- Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development.
 - a. Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development.

 $^{^2}$ In physics texts please activate the class option vecphys to depict your vectors in **boldface-italic** type - as is customary for a wide range of physical subjects.

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- Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development.
- 2. Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development.

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For unnumbered list we recommend to use the itemize environment – it will automatically render Springer's preferred layout.

- Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development, cf. Table 2.2.
 - Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development.
 - Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development.
- Livelihood and survival mobility are oftentimes coutcomes of uneven socioeconomic development.

Run-in Heading Boldface Version Use the LATEX automatism for all your cross-references and citations as has already been described in Sect. 2.2.

Run-in Heading Italic Version Use the LATEX automatism for all your cross-references and citations as has already been described in Sect. 2.2.

Table 2.2 Please write your table caption here

Classes	Subclass	Length	Action Mechanism
Translation	mRNA ^a	22 (19–25)	Translation repression, mRNA cleavage
Translation	mRNA cleavage	21	mRNA cleavage
Translation	mRNA	21–22	mRNA cleavage
Translation	mRNA	24–26	Histone and DNA Modification

^a Table foot note (with superscript)

2.3 Section Heading

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- Type 1 That addresses central themes pertaining to migration, health, and disease. In Sect. 2.1, Wilson discusses the role of human migration in infectious disease distributions and patterns.
- Type 2 That addresses central themes pertaining to migration, health, and disease. In Sect. 2.2.1, Wilson discusses the role of human migration in infectious disease distributions and patterns.

2.3.1 Subsection Heading

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2.3.1.1 Subsubsection Heading

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Please note that the first line of text that follows a heading is not indented, whereas the first lines of all subsequent paragraphs are.

Theorem 2.1. Theorem text goes here.

Definition 2.1. Definition text goes here.

Proof. Proof text goes here. \Box

2.3 Section Heading 11

Paragraph Heading Instead of simply listing headings of different levels we recommend to let every heading be followed by at least a short passage of text. Furtheron please use the LATEX automatism for all your cross-references and citations as has already been described in Sect. 2.2.

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Theorem 2.2. Theorem text goes here.

Definition 2.2. Definition text goes here.

Proof. Proof text goes here.

Acknowledgements If you want to include acknowledgments of assistance and the like at the end of an individual chapter please use the acknowledgement environment – it will automatically render Springer's preferred layout.

Appendix

When placed at the end of a chapter or contribution (as opposed to at the end of the book), the numbering of tables, figures, and equations in the appendix section continues on from that in the main text. Hence please *do not* use the appendix command when writing an appendix at the end of your chapter or contribution. If there is only one the appendix is designated "Appendix", or "Appendix 1", or "Appendix 2", etc. if there is more than one.

$$a \times b = c \tag{2.3}$$

Problems

2.1. A given problem or Excercise is described here. The problem is described here. The problem is described here.

2.2. Problem Heading

- (a) The first part of the problem is described here.
- (b) The second part of the problem is described here.

Methods

Difference Time Domain Method

Finite Difference Time Domain Method

Time Domain Method

Modelling Strategies

Work