The University of Derby Faculty of Arts, Design and Technology

# Efficient Acoustic Modelling of Large Spaces using Time Domain Methods

Analysis of Time Domain Numerical Methods for Acoustic Modelling of Large Spaces

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#### Acknowledgements

I would like to dedicate this work to anyone of remote importance.

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#### **Acronyms**

Use the template *acronym.tex* together with the Springer document class SVMono (monograph-type books) or SVMult (edited books) to style your list(s) of abbreviations or symbols in the Springer layout.

Lists of abbreviations, symbols and the like are easily formatted with the help of the Springer-enhanced description environment.

ABC Spelled-out abbreviation and definition BABI Spelled-out abbreviation and definition CABR Spelled-out abbreviation and definition

## Part I Background & Theory

## Chapter 1 Introduction

The intro Text

#### 1.1 Context

#### 1.2 Problem Definition

Real time acoustic modelling could be of significant benefit to many applications; Engineers could make design changes and see results 'on the fly', and entertainment users could have more realistic experiences. These benefits should be possible for an arbitrary number of sources and receivers, in proportionally large environments with high quality results. Is it possible to further reduce computation time for simulations of large acoustic problems, to provide results in real time for the full human audio frequency range? There are two 'branches' of computation solution that should be considered: the direct solution i.e. direct outputs or audio samples from the simulation, and indirect solutions i.e. a system impulse response that may be convolved with mixed source signals in order to create an auralization of the system.//

Fig. 1.1 A visualisation of a 2D explicit FDTD simulation [?]

#### Chapter 2

#### **Loudspeaker Systems & Large Room Acoustics**

Acoustics is a branch of physics that aims to characterise Newton's law of motion applied to wave propagation, while obeying the physical conservation law and often focussing on propagation in an audible spectrum. This characterisation of sound propagation is intrinsically linked to many other branches of physics, as well as psychoacoustics and perception. Many aspects of acoustic modelling may be of interest when considering the design and application of loudspeaker systems. Both small and large scale simulations may allow a user to make informed decisions about the design and placement of a loudspeaker system, so that the performance of the system may be validated and optimised before application. In this chapter we will evaluate the lossless acoustic wave equation for gasses, and consider the application of the wave equation in bounded space. We will then consider some specific use cases for applying such an equation for modelling loudspeaker system performance.

#### 2.1 The Acoustic Wave Equation

In the Mcgraw-Hill Electronic and Electrical Engineering Series of books, the late L Beranek authored the Acoustics volume. This volume contains an elegant summary of the wave equation, that will be the subject of paraphrase in the following section.

#### 2.1.1 The Wave Equations

Acoustic waves are classified as fluctuations of pressure in a given medium. In room acoustics and loudspeaker system engineering, these fluctuations are often cyclical in nature around an ambient pressure, as opposed to the jets described in aeroacoustic study. Similar to the behaviour of heat convection or fluid diffusion, these cyclical fluctuations propagate and spread through the medium of interest. As these fluctuations of pressure propagate energy is often lost, and eventually the medium will often come to a state of relative rest where the energy of the propagating waves have been almost entirely dissipated. It is possible to calculate an approximate solution the the propagation of pressure through a space, by solving a system of second order partial differential equations that can be collectively lumped into a 'Wave

Equation'. Below, we will introduce the three building blocks of the wave equation in both one dimension, and three dimensions (based on vector notation). These building blocks are Newton's Second Law of Motion, the gas law, and the laws of conservation of mass.

To consider the wave equation, we should use the analogy of a small<sup>1</sup> volume of gas, within a larger homogeneous medium. The faces of the volume are frictionless, and only the pressure at any face impacts on the gas inside the volume.

One Dimension	Three Dimensions
medium like a plane wave, from one side to the other in the x direction at a rate	Sound pressure $p$ propagates across the medium like a spherical wave, from one side to the other at a rate of <b>grad</b> $p = \mathbf{i} \frac{\delta p}{\delta x} + \mathbf{j} \frac{\delta p}{\delta y} + \mathbf{k} \frac{\delta p}{\delta z}$ where i, j and k are unit vectors in the directions x, y and z.
tive x direction can thus be described as $-(\frac{\delta p}{\delta x}\Delta x)\Delta y\Delta z$	Force acting on the volume in the positive $x$ direction can thus be described as $-[i(\frac{\delta p}{\delta x}\Delta x)\Delta y\Delta z) + j(\frac{\delta p}{\delta y}\Delta y)\Delta x\Delta z) + k(\frac{\delta p}{\delta z}\Delta z)\Delta x\Delta y)]$
A positive gradient causes acceleration in the $-x$ direction	<b>←</b>
	Force per unit volume is given by dividing both sides of the previous equation by the volume $V, \frac{f}{V} = -\mathbf{grad}p$
Newton's second law of motion dictates that the rate of change of momentum in the volume must balance with force per unit volume, and we can assume the mass of gas in the volume is constant.  The force mass balance can be described	The force mass balance can be described
as $\frac{f}{V} = -\frac{\delta p}{\delta x} = \frac{M}{M} \frac{\delta u}{\delta t} = \rho' \frac{\delta u}{\delta t}$	as $\frac{f}{V} = -\mathbf{grad}p = \frac{M}{M}\frac{Dq}{Dt} = \rho'\frac{Dq}{Dt}$

<sup>&</sup>lt;sup>1</sup> rectilinear

| u is the velocity of gas in the volume,  $\rho'$  | where q is the vector velocity,  $\rho'$  is the is the density of the gas, and  $M = \rho'V$  is density of gas in the volume,  $M = \rho'V$ the mass of gas in the volume.

is the total mass of gas in the volume.  $\frac{D}{Dt}$  represents the total rate of change of velocity of a section of gas in the volume, and can be composed as  $\frac{Dq}{Dt} = \frac{\delta q}{\delta t} + q_x \frac{\delta q}{\delta x} + q_y \frac{\delta q}{\delta y} + q_z \frac{\delta q}{\delta z}$  where  $q_x$ ,  $q_y$  and  $q_z$  are the components of the particle velocity q in each direction. As this is a linear wave equation approximation, these velocity components have no cross terms.

If the change in density of gas in the vol-If the particle velocity vector is suffiabove to  $-\frac{\delta p}{\delta x} = \rho_0 \frac{\delta u}{\delta t}$ 

ume is sufficiently small, the  $\rho'$  will be ciently small, the change of momentum of approximately equal to the average den-the gas is approximately the same as the sity  $\rho_0$ , thus simplifying the equations momentum of the volume at any arbitrary point, and the density of gas within the volume  $\rho'$  will be approximately equal to the average density  $\rho_0$ . Thus the above can be written as  $-grad p = \rho_0 \frac{\delta q}{\delta t}$ 

This kind of approximation may be appropriate as long as the maximum pressure is appropriately low, so that the behaviour of the air is linear, often quoted to be at or under the threshold of pain for human hearing or 120dB SIL.

Assuming that the gas of the volume is ideal, then the gas law PV = RT should hold true. Here, T is the temperature in degrees Kelvin, and R is a constant based on the mass of the gas. For this approximation we assume that the system is adiabatic, and that T and R are lumped into a gas constant which for air is  $\gamma = 1.4$ .

In differential form, the relationship between pressure and volume for an adiabatic expansion the volume is  $\frac{dP}{P}$  =  $\frac{-\gamma dV}{V}$  i.e. changes in pressure scale with changes in volume by this  $\gamma$  value.

If perturbations in pressure and volume due to a sound wave, p for pressure and  $\tau$  for volume respectively, are sufficiently small compared to the rest values  $P_0$  and  $V_0$ : the time based derivative of the above equation can be written as follows:  $\frac{1}{P_0} \frac{\delta p}{\delta t} = \frac{-\gamma}{V_0} \frac{\delta \tau}{\delta t}$ 

As the wave equation being derived is concerned with the transport of pressure within a volume, a continuity expression must be applied. The conservation of mass states that the total mass of gas in the volume must remain constant. This conservation law brings a unique relationship between discrete velocities at the boundary of the volume:

is displaced with a velocity, in a given as  $\tau = V_0 \operatorname{div} \varepsilon$ time step the particles at the right hand boundary must also be displaced. This can be written as  $\varepsilon_x + \frac{\delta \varepsilon_x}{\delta x} \Delta x$  The difference between this velocity and a subsequent change in volume  $\tau$  multiplied by the volume gives  $\tau = V_0 \frac{\delta \varepsilon_x}{\delta r}$ .

The one dimensional wave equation in The three dimensional wave equation in  $\left| \frac{\delta p}{\delta t} = -\gamma P_0 \frac{\delta u}{\delta x} \right|$ 

If the volume is displaced by some rate If the mass of gas within the box must  $\varepsilon_x$ , air particles at either boundary of the remain constant, the vector displacement volume must be displaced at an equal rate will directly change the volume by some for the mass of the volume to remain con-rate, as the two must balance to satisfy the stant. As such if the left side of the volume continuity equation. This can be written

Differentiating this with respect to time gives:  $\frac{\delta \tau}{\delta t} = V_0 \frac{\delta u}{\delta x}$  where u is the instantaneous particle velocity

Differentiating this with respect to time gives:  $\frac{\delta \tau}{\delta t} = V_0 \ div \ q$  where q is the instantaneous particle velocity

rectangular coordinates can be created by rectangular coordinates can be created by combining the above statements about the combining the above statements about the equation of motion, the gas law and the equation of motion, the gas law and the continuity equation. The combination of continuity equation. The combination of the gas law and continuity equation gives the gas law and continuity equation gives  $\frac{\delta p}{\delta t} = -\gamma P_0 div \mathbf{q}$ 

When differentiated with respect to time, When differentiated with respect to time this gives:  $\frac{\delta^2 p}{\delta t^2} = -\gamma P_0 \frac{\delta^2 u}{\delta t \delta r}$ 

derived above with respect to time gives tion derived above gives:  $-div = \rho_0 div \frac{\delta q}{\delta t}$  $-\frac{\delta^2 p}{\delta t^2} = \rho_0 \frac{\delta^2 u}{\delta r \delta t}$ 

 $\frac{\delta^2 p}{\delta x^2} = \frac{\rho_0}{\gamma \rho_0} \frac{\delta^2 p}{\delta t^2}$ 

If we define c as the speed of propagation  $\leftarrow$ in the medium of interest, then  $c^2 \approx \frac{\gamma P_0}{Q_0}$ due to the fact that the speed of sound  $c \approx (1.4 \frac{10^5}{1.18})^{\frac{1}{2}}$  where the ambient air pressure at sea level is  $10^5 Pa$ , 1.4 is the adiabatic constant  $\gamma$  (ratio of specific heats) for air, and  $\rho_0$  is the density of air is approximately  $1.8kg/m^3$ 

Finally we find that the 1 dimensional Finally we find that the 3 dimensional wave equation is:  $\frac{\delta^2 p}{\delta x^2} = \frac{1}{c^2} \frac{\delta^2 p}{\delta t^2}$ 

this gives:  $\frac{\delta^2 p}{\delta t^2} = -\gamma P_0 div \frac{\delta q}{\delta t}$ 

Differentiating the momentum equation The divergence of the momentum equa-Replacing the divergence (gradp) term with the Lapacian operator  $\nabla^2 p$  produces  $-\nabla^2 p = \rho_0 div \frac{\delta^2 p}{\delta t}$ 

Combining the above equations gives: Combining the above equations gives:  $\nabla^2 p = \frac{\rho_0}{\gamma P_0} \frac{\delta^2 p}{\delta t^2}$ 

> wave equation is:  $\nabla^2 p = \frac{1}{c^2} \frac{\delta^2 p}{\delta t^2}$  An explicit 3 dimensional expression of the pressure component of this equation is:  $\nabla^2 p = \frac{\delta^2 p}{\delta x^2} + \frac{\delta^2 p}{\delta x^2} + \frac{\delta^2 p}{\delta z^2}$

This equation can also be expressed in This equation can be expressed velocity terms of the instantaneous velocity in the vector  $\nabla^2 q = \frac{1}{c^2} \frac{\delta^2 q}{\delta t^2}$  where  $\nabla^2 q$  represents the gradient of pressure (velocity) in the volume.

In the above table we have derived wave equations, with forms of velocity and pressure as the independent variables. We have also shown that pressure, velocity, displacement and density are related within the system of equations, by differentiating and integrating with respect to space and time. As these forms of the wave equation are intrinsically coupled, it is possible to leverage this coupling when generating a numerical solution to the wave equation. It is also important to note that a significant number of assumptions have been taken when deriving these equations, and any solution to these equations may only be accurate when simulating a loss free, frictionless, homogeneous, ideal gas medium, where all perturbations are sufficiently small and fast that it is possible to reduce the complexity of the system.

## **Chapter 3 Finite Difference Time Domain Method**

The Finite Difference Time Domain Method is a numerical method for solving partial differential equations. The power of this method lies in its simplicity and flexibility, and it can be used to solve partial differential equations of varying complexity. In this chapter we will discuss the application of the finite difference time domain method to the acoustic wave equation, including the application of empirical partially absorbing boundary conditions.

### 3.1 Introduction to the Finite Difference Time Domain Method

Finite methods for solving partial differential equations have been of significant and continued research since the early 1900's, with mathematicians such as Courant, Fiedrichs and Hrennikof undertaking seminal work in the early 1920s, that formed a base for much of the finite methods used today. The Finite Difference Time Domain Method (FDTD) is a method for solving time domain problems (often wave equations) with loclised handling of time and space derivatives, and was first introduced for solving Maxwell's equations to simulate electromagnetic wave propagation by Yee [?]. Yee proposed a method for which Maxwell's equations in partial differential form were applied to matrices staggered in partial steps in time and space, these matrices representing the magnetic (H) and electric (E) fields. In this explicit formulation, partial derivatives were used to solve H and E contiguously in a 'leapfrog' style, executing two sets of computations to solve for one time step. Multiple time steps would be solved from current time t = 0, in steps of dt to the end of simulation time T.Each field is solved at half steps in time from each-other, thus H for a current time step  $t + \delta t$  is calculated using the H values one time step ago t, and the E values half a time step ago  $t + \frac{\delta t}{2}$ . These two fields are also solved using central finite differences in space, in a staggered grid format i.e. E at index x at time  $t + \delta t$ is calculated using E at index x at time t, and the finite difference between the local discrete values of H at  $x - \frac{\delta x}{2}$  and at  $x + \frac{\delta x}{2}$  at time  $t + \frac{\delta t}{2}$ . As such, it is possible to apply a simple kernel across many discretised points of a domain (H and E) to simulate electromagnetic wave propagation.

## 3.2 The Finite Difference Time Domain Method Applied To The Acoustic Wave Equation

The FDTD method applied to solving the acoustic wave equation, follows an almost identical form to that of solving Maxwells Equations with FDTD [?]<sup>1</sup>. Bottle-dooren's [?] seminal work applied the FDTD method to the acoustic wave equations for both Cartesian and quasi-Cartesian grid systems. As previously described in the room acoustics section, the linear acoustic wave equation is based on Newton's second law of motion, the gas law and the continuity equation, and follows the form for the changes in the pressure and velocity respectively within a volume:

$$\frac{\delta^2 p}{\delta t^2} = \frac{1}{c^2} \frac{\delta^2 p}{\delta t^2}$$
$$\frac{\delta^2 u}{\delta t^2} = \frac{1}{c^2} \frac{\delta^2 u}{\delta t^2}$$

. As pressure (p) and velocity (u) have a reciprocal relationship in a similar way to H and E, it is possible to rearrange the acoustic wave equation to reflect this relationship for a FDTD computation.

When treating the 1 dimensional linear acoustic wave equation with the FDTD method, it is possible to treat the p and u terms separately in time using the opposing terms for reciprocal calculation. As such, the p and u terms are reformulated as follows:

$$\frac{\delta^2 p}{\delta t^2} = p - \frac{\delta t}{\rho_0 \delta x} \frac{\delta^2 u}{\delta t^2}$$
$$\frac{\delta^2 u}{\delta t^2} = u - \frac{\delta t}{\rho_0 \delta x} \frac{\delta^2 u}{\delta t^2}$$

However, this formulation is incomplete as it does not consider spatial or temporal discretisation of the field of interest, when applying the FDTD method. As the FDTD method relies on solving local finite difference approximations across a domain of interest, it is important to define a space and time index referencing method. In many mathematical texts, time step indexing is often represented by an i value, and spatial indexing often uses a j,k,l or l,m,n convention. For the aim of simplicity and as we will not directly address other forms of input output system in this text, we will use t for the time step indexing, and x, y and z for spatial indexing in each dimension. Following an implementation of the acoustic FDTD method by Hill [?], we can generate the following p and u equations for FDTD applied to the acoustic wave equation:

$$u_x^{t+\frac{\delta t}{2}} = u_x^{t-\frac{\delta t}{2}} - \frac{\delta t}{\rho \delta x} [p_{x+\frac{\delta x}{2}}^t - p_{x-\frac{\delta x}{2}}^t]$$

<sup>&</sup>lt;sup>1</sup> In fact, the equations follow an almost identical form

$$p_x^{t+\frac{\delta t}{2}} = p_x^{t-\frac{\delta t}{2}} - \frac{c^2\rho\delta t}{\delta x} \left[ u_{x+\frac{\delta x}{2}}^t - u_{x-\frac{\delta x}{2}}^t \right]$$

Below, is a function written in the Matlab Rlanguage, used to solve one time step of the wave equation using the FDTD method, in 3 dimensions:

```
function [p, ux, uy, uz] = FDTD3Dfun(p, pCx, pCy, pCz, ux, uy, uz
      uCy, uCz, Rx, Ry, Rz, ZxN, ZxP, ZyN, ZyP, ZzN, ZzP)
  % Function that performs one timestep of FDTD method for acoustic
       simulation.
4 %
5 % This function performs central finite difference calculations
6 % matricies that represent pressure and velocity. This function
      assumes
  % that a linear acoustic wave equation is being solved, and so
      assumes that
 % the velocity terms are orthogonal and there are no cross-terms.
       This
9 % function solves empirical semi-absorbing boundary conditions,
      using the
10 % acoustic impedance of the boundary based on a normalised
      aproximation of
11 % absorption coefficient.
12 %
13 % Takes the following arguments:
14 % p = N:N:N matrix of pressure values
15 % ux = N:N+1:N matrix of velocity values
16 % uy = N+1:N:N matrix of velocity values
17 % uz = N:N:N+1 matrix of velocity values
18 % pCx = constant related to pressure calculation in x direction
_{19} % pCy = constant related to pressure calculation in y direction
20 % pCz = constant related to pressure calculation in z direction
21 % uCx = constant related to velocity calculation in x direction
22 % uCy = constant related to velocity calculation in y direction
23 % uCz = constant related to velocity calculation in z direction
\% Rx = (\text{rho0*dx})/(0.5*\text{dt}) Constant related to field constants
25 % Ry = (rho0*dy)/(0.5*dt) Constant related to field constants
_{26} % Rz = (\text{rho0*dz})/(0.5*\text{dt}) Constant related to field constants
27 % ZxN = acoutsite impedance term at boundary in -x direction
28 % ZxP = acoutsite impedance term at boundary in +x direction
29 % ZyN = acoutsite impedance term at boundary in -y direction
30 % ZyP = acoutsite impedance term at boundary in +y direction
31 % ZzN = acoutsite impedance term at boundary in -z direction
32 % ZzP = acoutsite impedance term at boundary in +z direction
33 %
34 % This functions returns the pressure and velocity field
      matricies
35 %
36
      % Calculate central difference approximation to velocity field
37
      % Velocity in a direction at current timestep excluding the
38
      boundarys
      % = velocity 1 time step ago - constants * pressure
```

```
% differential half a time step ago in that direction
40
             ux(:, 2:end-1, :) = ux(:, 2:end-1,:) - uCx*(p(:, 2:end,:) - p
41
             (:, 1: end -1, :));
             uy(2:end-1, :, :) = uy(2:end-1, :, :) - uCy*(p(2:end, :, :) - uC
42
              p(1:end-1, :, :));
             uz(:, :, 2:end-1) = uz(:, :, 2:end-1) - uCz*(p(:, :, 2:end) -
43
              p(:, :, 1: end -1));
44
             % update the velocity at the negative x boundary
45
             % Velocity at this boundary for all of y and z = time and
46
             space step
            % normalised by the lovel impedance condition * current
47
             velocity values
             \%-2 / time and space discretization * local pressure value
48
             ux(:, 1, :) = ((Rx - ZxN)/(Rx + ZxN))*ux(:, 1, :) ...
49
                      -(2/(Rx + ZxN))*p(:, 1, :);
50
51
             % update the velocity at the positive x boundary
52
             ux(:, end, :) = ((Rx - ZxP)/(Rx + ZxP))*ux(:, end, :) ...
53
                      + (2/(Rx + ZxP))*p(:, end, :);
54
55
             % update the velocity at the negative y boundary
56
             uy(1, :, :) = ((Ry - ZyP)/(Ry + ZyP))*uy(1, :, :) ...
57
                     -(2/(Ry + ZyP))*p(1, :, :);
58
             % update the velocity at the positive y boundary
60
             uy(end, :, :) = ((Ry - ZyN)/(Ry + ZyN))*uy(end, :, :) ...
61
                      + (2/(Ry + ZzN))*p(end, :, :);
62
            % update the velocity at the negative z boundary
64
             uz(:, :, 1) = ((Rz - ZzP)/(Rz + ZzP))*uz(:, :, 1) - ...
65
             (2/(Rz + ZzP))*p(:, :, 1);
66
             % update the velocity at the positive z boundary
             uz(:, :, end) = ((Rz - ZzN)/(Rz + ZzN))*uz(:, :, end) + ...
69
             (2/(Rz + ZzN))*p(:, :, end);
70
             % update the pressure at all nodes
             % new pressure across domain = pressure across domain 1 time
             step ago -
             % (space, time and wave speed constant) * central difference
74
             % velocities half a time step ago in all three dimensions
             p = p - pCx*(ux(:, 2:end, :) - ux(:, 1:end-1, :))...
76
                     - pCy*(uy(2:end, :, :) - uy(1:end-1, :, :))...
77
                      - pCz*(uz(:, :, 2:end) - uz(:, :, 1:end-1));
78
79 end
```

#### 3.2.1 Stability

#### 3.2.2 Handling Empirical Semi-Absorbing Boundary Conditions

## **3.3** A 2D Simulation with empirical partially absorbing boundary conditions

Instead of simply listing headings of different levels we recommend to let every heading be followed by at least a short passage of text. Furtheron please use the LATEX automatism for all your cross-references and citations.

Finite Difference Time Domain Method

Time Domain Method

Modelling Strategies

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Work

#### References