

# Nonlinear Observer for Tightly Coupled Integration of Pseudorange and Inertial Measurements

Guide and Navigation Systems

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# Introduction

This presentation shows the obtained results by the implementation of the article written by Tor. A. Johansen and Thor I. Fossen.

## Nonlinear Observer for Tightly Coupled Integration of Pseudorange and Inertial Measurements

Tor A. Johansen and Thor I. Fossen

# Introduction

The goal of navigation systems is to estimate the vehicle position within a certain area. This is let generally by a inertial navigation system, based on *IMU* (Inertial Measurement Unit).

Using an interconnection of a nonlinear attitude observer and a translational motion observer based on pseudorange and range-range measurements, a tightly coupled integrated aided inertial navigation system is designed!?!?!?

Due to the problem of measurements integration, this method is supported by other techniques. In this work, the inertial navigation system is aid by pseudorange measurements, obtained by transponders.

# Vehicle Kinematics

The vehicle kinematic model is given by

$$\dot{p}^n = v^n$$

$$\dot{v}^n = R_b^n f^b + g^n$$

$$\dot{R}_b^n = R_b^n S(\omega_{ib}^b)$$

Where  $p^n$ ,  $v^n$ ,  $f^n$  are position, velocity and proper acceleration in NED (North-East-Down), respectively, while the attitude is described by a rotation matrix  $R_b^n$  that represents the rotation from *body* to NED;  $\omega_{ib}^b$  represents the rotation rate of body with respect to ECI (Earth-Centered-Inertial) and  $g^n$  denotes the gravity vector. We also assume NED to be an inertial frame.

# Inertial Sensor Models

The inertial sensor model is based on the strapdown assumption

$$f_{IMU}^b = f^b + \epsilon_f$$

$$\omega_{ib,IMU}^b = \omega_{ib}^b + b + \epsilon_\omega$$

$$\dot{b} = \epsilon_b$$

$$m_{mag}^b = m^b + \epsilon_m$$

where  $\epsilon_f$ ,  $\epsilon_\omega$  and  $\epsilon_m$  account for noise, and  $b$  denotes the rate gyro bias that is driven by the noise  $\epsilon_b$  and assumed to be bounded. All sensors are 3-D.

# Pseudorange Measurement Model

The geometric range

$$\rho_i = \|p^n - p_i^n\|_2$$

is a nonlinear function of the vehicle position  $p^n$  and the  $i$ th transponder position  $p_i^n$ , given by their Euclidean distance.

The pseudorange measurement model is

$$y_i = \rho_i + \beta + \epsilon_{yi}$$

where  $\beta \in \mathbb{R}$  is a bias parameter due to unknown clock synchronization errors or other unknown effects and  $\epsilon_{yi}$  the noise.  $i = 1, 2, \dots, m$  where  $m$  is the number of measurements.

## Pseudorange Measurement Model

The nonlinear model can be approximated with a linear one by an algebraic transformation

$$2C_{\delta x}x = \delta + \varepsilon$$

where the matrix  $C_{\delta x} \in \mathbb{R}^{(m-1) \times 4}$  is

$$C_{\delta x} := \begin{pmatrix} (p_m^n - p_1^n)^T & y_1 - y_m \\ \vdots & \\ (p_m^n - p_{m-1}^n)^T & y_{m-1} - y_m \end{pmatrix}$$

$\varepsilon \in \mathbb{R}^{m-1}$  the noise and  $\delta \in \mathbb{R}^{m-1}$  the vector of squared range measurements.

$x := (p_{\Delta}^n; \beta)$  where  $p_{\Delta}^n = p^n - p_0^n$  ( $p_{\Delta}^n$  is a reference point in NED).



# Observer Design

Two observer are designed: one for the attitude estimation and one for the translational motion estimation.

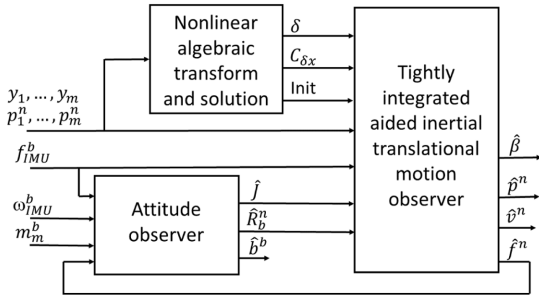


Figure : Overall block diagram for tightly integrated observer.

# Attitude Observer

The attitude variables to estimate are  $R_b^n$  and  $b$ .

$$\dot{\hat{R}}_b^n = \hat{R}_b^n S(\omega_{ib,IMU}^b - \hat{b}) + \sigma K_p J(t, \hat{R}_b^n)$$

$$\dot{\hat{b}} = Proj(-k_I vex(\mathbb{P}_a(sat(\hat{R}_b^n)^T K_p J(t, \hat{R}_b^n))), M_{\hat{b}})$$

where  $K_p > 0 \in \mathbb{R}^{3 \times 3}$  is a symmetric gain matrix,  $K_I > 0$  is a scalar gain and  $\sigma \geq 1$ .

The function  $sat(\cdot)$  is an element-wise saturation, while  $Proj(\cdot)$  is a parameter projection which ensures that  $\|\hat{b}\|_2$  is bounded.

## Attitude Observer

The function  $J(\cdot) \in \mathbb{R}^{3 \times 3}$  is a stabilizing injection term

$$J(t, \hat{R}_b^n) = (E^n - \hat{R}_b^n E^b)(E^b)^T$$

based on the vector measurements  $m_{mag}^b$  and  $f_{IMU}^b$  and their NED reference vectors  $m^n$  and  $\hat{f}^n$  used to define vectors scaled by nonzero terms

$$q_1^b = m_{mag}^b / \|m_{mag}^b\|_2 \quad q_2^b = f_{IMU}^b / \|g^n\|_2$$

$$q_1^n = m^n / \|m^n\|_2 \quad q_2^n = \hat{f}^n / \|g^n\|_2$$

and the  $3 \times 3$  matrices

$$E^b = (q_1^b, S(q_1^b)q_2^b, S^2(q_1^b)q_2^b)$$

$$E^n = (q_1^n, S(q_1^n)q_2^n, S^2(q_1^n)q_2^n)$$

## Translational Motion Observer

The variables to estimate are  $p^n, v^n, f^n, \beta$ .

$$\dot{\hat{p}}_{\Delta}^n = \hat{v}^n + K_{pp}(\delta - \hat{\delta})$$

$$\dot{\hat{\beta}} = K_{\beta p}(\delta - \hat{\delta})$$

$$\dot{\hat{v}}^n = \hat{f}^n + g^n + K_{vp}(\delta - \hat{\delta})$$

$$\dot{\xi} = -\sigma K_P J(t, \hat{R}_b^n) f_{IMU}^b + K_{\xi p}(\delta - \hat{\delta})$$

$$\hat{f}^n = \hat{R}_b^n f_{IMU}^b + \xi$$

where  $\hat{\delta} = 2C_{\delta x}\hat{x}$  and the gain matrix  $K \in \mathbb{R}^{10 \times (m-1)}$  is made of the matrices  $K_*$  and is in general time varying.

# Translational Motion Observer

Then it is possible to build the *estimated state* vector  $\dot{\tilde{\chi}} = (\tilde{p}_{\Delta}^n; \tilde{\beta}; \tilde{\hat{v}}^n; \tilde{\hat{f}}^n) \in \mathbb{R}^{10}$  and the relative LTV error system

$$\dot{\tilde{\chi}} = (A - KC)\tilde{\chi} + Bu + B\epsilon_u + K\epsilon$$

$$u = \tilde{R}_b^n \dot{f}^b + \tilde{R}_b^n S(\omega_{ib}^b) f^b - \hat{R}_b^n S(\tilde{b}) f^b$$

## Translational Motion Observer

The matrices  $A \in \mathbb{R}^{10 \times 10}$ ,  $B \in \mathbb{R}^{10 \times 3}$ ,  $C \in \mathbb{R}^{(m-1) \times 10}$  and  $K \in \mathbb{R}^{10 \times (m-1)}$  are described as follows

$$A := \begin{pmatrix} 0 & 0 & I_3 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & I_3 \\ 0 & 0 & 0 & 0 \end{pmatrix} \quad B := \begin{pmatrix} 0 \\ 0 \\ 0 \\ I_3 \end{pmatrix}$$

$$K := \begin{pmatrix} K_{pp} \\ K_{\beta p} \\ K_{vp} \\ K_{\xi p} \end{pmatrix} \quad C := (2C_{\delta x} \quad 0 \quad 0)$$

# Translational Motion Observer

The gain matrix  $K$  is time varying and calculated as

$$K := PC^T R^{-1}$$

where  $P$  is solution of the *Riccati* equation

$$\dot{P} = PA + A^T P - PC^T R^{-1} CP + Q$$

# Implementation Design

The whole model has been implemented with MATLAB/Simulink (R2016b).

In the following each observer is presented in terms of structure and results, i.e. the difference between the true value of the vehicle position and of the bias parameter and their estimation.



# Implementation Design

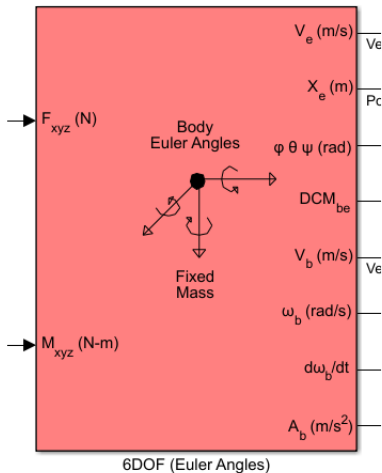
Data used by the authors are

- pseudorange measurements obtained by radio beacons;
- accelerations and angular velocities by IMU.

The inertial sensor model is based on the strapdown assumption, i.e. the inertial measurement unit is fixed to the body frame.

# Data Generation

In order to generate data the 6DOF Matlab block has been used

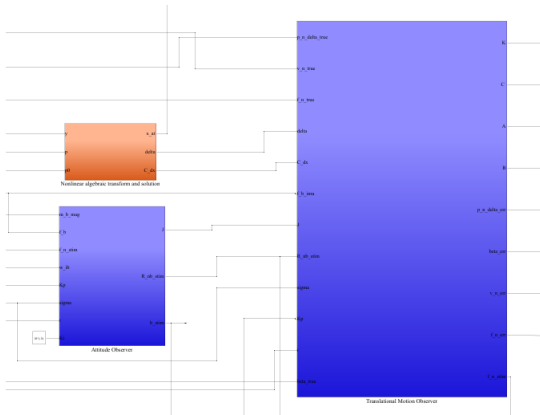


# Data Generation

- The real position of the vehicle is given from  $X_e(m)$ ;
- The pseudorange measurements always by  $X_e(m)$  but corrupted by noise;
- The accelerations from  $A_b(m/s^2)$ ;
- Angular velocities from  $\omega_b(rad/s)$ .

## Implementation Design

The Simulink implementation looks like the one in the following figure.



# Implementation Design

The *NonlinearAlgebraicTransform* block computes the  $C_{\delta x}$  matrix and the following variable:

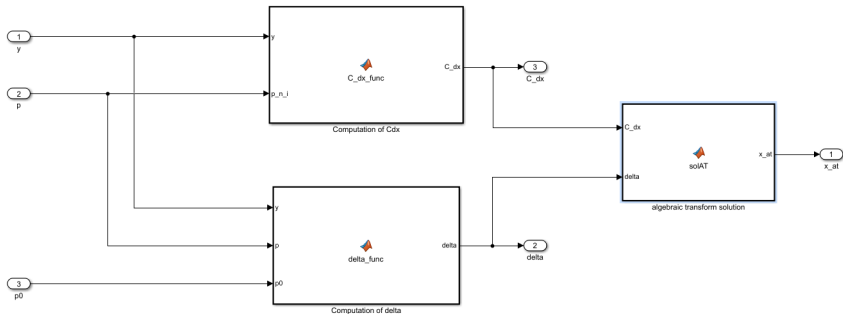
$$\hat{x} = \frac{C_{\delta x}^+ \delta}{2}$$

that is the unique solution to

$$2C_{\delta x}x = \delta + \epsilon$$

in the case of  $m \geq 5$  transponders, where  $x = \begin{pmatrix} p^n \\ \beta \end{pmatrix}$ .

# Nonlinear Algebraic Transform Implementation



figures

# Implementation Design

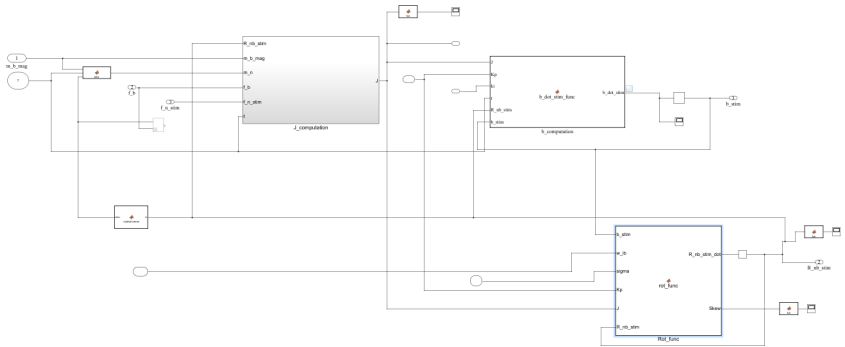
The Attitude Observer is made of three main function blocks:

- $J(\cdot)$  is a stabilizing injection term
- $b\_computation$  computes the dynamics of the bias by means of the  $Proj(\cdot)$  function
- $Rot\_func$  computes an estimate of the rotation matrix



# Attitude Observer Implementation

The following picture shows the Simulink scheme:

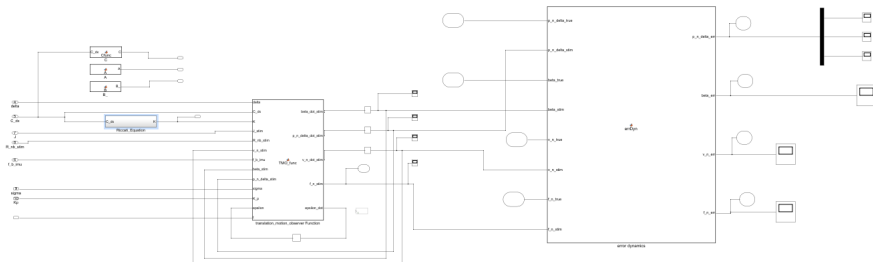


# Translational Motion Observer Design

The translational Motion Observer is made of three blocks:

- The first one solves the Riccati Equation to get the gain matrix  $K$ ;
- The *TMO\_fuction* computes the estimate of position, velocity, acceleration and bias;
- The last block computes the error dynamics.

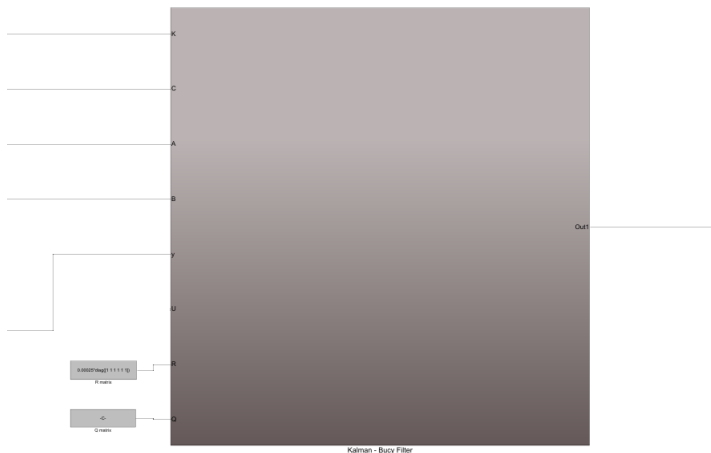
# Translational Motion Observer Design



figures

## Implementation Design

The estimates obtained by the Translational Motion Observer are then sent to a block that implements the Kalman-Bucy Filter in order to make a better estimation



# Kalman Equations

figures

## Confronto tra gli osservatori