# **CP533:**Clean Combustion Technologies



# Lecture 1 Combustion Theory

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# **Overview**



### 1. Basic combustion reactions

- Calculating air fuel ratios
- Stoichiometric combustion,
- lean combustion and
- rich combustion
- Equivalence ratio & Excess air ratio

# 2. Term enthalpy (total energy associated)

- Absolute enthalpy
- Enthalpy of formation

# 3. Adiabatic Flame temperatures

### Basic combustion reaction

$$C_x H_y + a(O_2 + 3.76N_2) \rightarrow xCO_2 + \left(\frac{y}{2}\right)H_2O + bO_2 + 3.76aN_2$$

Air fuel mass ratios

 $\frac{m_{air}}{m_{fuel}} = \frac{a \cdot MW_{air}}{1 \cdot MW_{Fuel}}$ 

Stoichiometric combustion

b = 0

Equivalence ratio

$$\Phi = \frac{\left(m_{air}/m_{fuel}\right)_{stoic}}{\left(m_{air}/m_{fuel}\right)} = \frac{a_{stoi}}{a}$$

• Fuel lean mixture ( $\Phi < 1$ ); fuel rich combustion ( $\Phi > 1$ )

### Exercise 1: 1 mole $C_{10}H_{22}$ combustion with 16 mole air

$$a_{stoi} = 15.5$$
  $b = 0.5$   $\Phi = 0.969$ 

# **Basic combustion reaction**

$$C_x H_y + a(O_2 + 3.76N_2) \rightarrow xCO_2 + \left(\frac{y}{2}\right)H_2O + bO_2 + 3.76aN_2$$

Known: 1 mole  $C_{10}H_{22}$  combustion with 16 mole air

**To obtain:**  $a_{stoi}$ 

b

Ф

# **Absolute Enthalpy and Enthalpy of Formation**

$$\overline{h}_i(T) = \overline{h}_{f,i}^0(T_{ref}) + \Delta \overline{h}_{s,i}(T_{ref})$$

**Absolute enthalpy** at temperature *T* 

Enthalpy of formation at standard reference state  $(T_{ref} = 25 \, ^{\circ}\text{C}, P_{ref} = 1 \, atm)$ 

**Sensible enthalpy** change in going from  $T_{ref}$  to T

 $\Delta \overline{h}_{s,i}(T_{ref}) = \overline{h}_i(T) - \overline{h}_{f,i}^0(T_{ref})$ 

**Table A.2** Carbon dioxide (CO<sub>2</sub>), MW = 44.011, enthalpy of formation @ 298 K (kJ/kmol) = -393,546

T(K)	$\bar{c}_p$ (kJ/kmol-K)	$(ar{h}^o(T) - ar{h}^o_f(298)) \  ext{(kJ/kmol)}$	$ar{h}_f^o(T)$ (kJ/kmol)	$\bar{s}^{o}(T)$ (kJ/kmol-K)	$ar{g}_f^o(T) \  ext{(kJ/kmol)}$
200	32.387	-3,423	-393,483	199.876	-394,126
298	37.198	0	-393,546	213.736	-394,428
300	37.280	69	-393,547	213.966	-394,433
400	41.276	4,003	-393,617	225.257	-394,718

Table A.11 Oxygen  $(O_2)$ , MW = 31.999, enthalpy of formation @ 298 K (kJ/kmol) = 0

T(K)	$\bar{c}_p$ (kJ/kmol-K)	$\begin{array}{c} (\bar{h}^o(T) - \bar{h}^o_f(298)) \\ (\text{kJ/kmol}) \end{array}$	$ar{h}^o_f(T) \ ( ext{kJ/kmol})$	$\bar{s}^{o}(T)$ (kJ/kmol-K)	$ar{g}_f^o(T)$ (kJ/kmol)
200	28.473	-2,836	0	193.518	0
298	29.315	0	0	205.043	0
300	29.331	54	0	205.224	0
400	30.210	3,031	0	213.782	0
500	31.114	6,097	0	220.620	0

The standard enthalpy of formation of a pure element is **zero**.

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# **Absolute Enthalpy - mixtures**

$$\overline{h}_i(T) = \overline{h}_{f,i}^0(T_{ref}) + \Delta \overline{h}_{s,i}(T_{ref})$$

For ideal-gas mixtures, mixture enthalpies are calculated:

$$\bar{h}_{mix} = \sum_{i} \boldsymbol{\chi}_{i} \bar{h}_{i}$$

 $h_{mix} = \sum_{i} Y_i h_i$ 

Mole basis (kJ/kmol)  $\chi_i$ - mole fraction

Mass basis (kJ/kg)  $Y_i$ -mass fraction

### **Exercise 2:**

A gas stream at 1 atm contains a mixture of CO, CO2 and N2 in which the CO mole fraction is 0.1 and the CO2 mole fraction is 0.2. The gas-stream temperature is 1200K. Determine the absolute enthalpy of the mixture on both a mole basis (kJ/kmol) and a mass basis (kJ/kg).

### **Exercise 2**

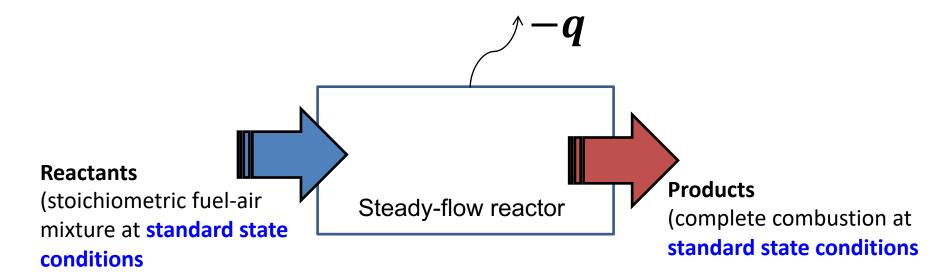
For ideal-gas mixtures, mixture enthalpies are calculated:

$$\begin{split} \bar{h}_{mix} &= \sum_{i} \chi_{i} \bar{h}_{i} = \chi_{CO} \left[ \bar{h}_{f,CO}^{0} + \left( \bar{h}(T) - \bar{h}_{f,298}^{0} \right)_{CO} \right] \\ &+ \chi_{CO2} \left[ \bar{h}_{f,CO2}^{0} + \left( \bar{h}(T) - \bar{h}_{f,298}^{0} \right)_{CO2} \right] \\ &+ \chi_{N2} \left[ \bar{h}_{f,N2}^{0} + \left( \bar{h}(T) - \bar{h}_{f,298}^{0} \right)_{N2} \right] \end{split}$$

$$\bar{h}_{mix} = 0.10[-110,541 + 28,440]$$
 $+0.20[-393,546 + 44,488]$ 
 $+0.70[0 + 28,118]$ 
 $= -58,339.1 \text{ kJ/kmol}$ 

$$Y_{CO} = ?$$
  $Y_{CO2} = ?$   $Y_{N2} = ?$   $h_{mix} = ?$ 

# **Enthalpy of combustion**



$$\Delta h_{R} = H_{prod} - H_{reac} = -\Delta h_{c}$$

**Enthalpy of combustion** 

**Heat of combustion (heating value)** 

per-mole-of fuel (often applied to gaseous/liquid fuel)

per-mass-of fuel (often applied to solid fuel)

# **Enthalpy of combustion**

### **Exercise 3**

Determine the upper and lower heating values at 298K of gaseous n-decane ( $C_{10}H_{22}$ ), per kilo-mole of fuel and per kilogram of fuel.

$$C_{10}H_{22} + 15.5(O_2 + 3.76N_2) \rightarrow 10CO_2 + 11H_2O + 58.28N_2$$

$$H_{reac} = H_{C10H22} + H_{air}$$

$$H_{prod} = H_{CO2} + H_{H2O} + H_{N2}$$

### Note:

- $\circ$  The numerical value of H<sub>prod</sub> depends on whether the H<sub>2</sub>O in the products is liquid or gaseous. Here <u>you could consider H<sub>2</sub>O as gaseous product firstly.</u>
- $\circ$  The sensible enthalpies for all species involved are <u>zero</u> as we desire  $\Delta h_c$  at the reference state (298K)
- $\circ$  Also, the enthalpies of formation of the O<sub>2</sub> and N<sub>2</sub> are zero at 298K.

# Adiabatic flame temperature

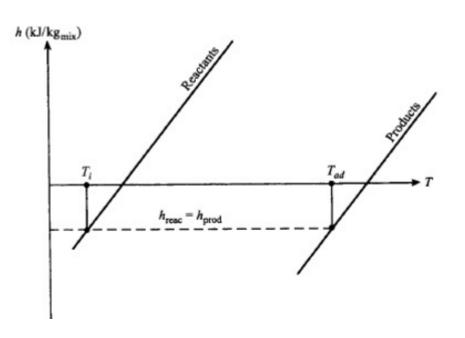
# @ constant pressure

Fuel-air mixture burns adiabatically:

$$H_{reac}(T_i, P) = H_{prod}(T_{ad}, P)$$

$$\overline{h}_i(T_{ad}) - \overline{h}_{f,i}^0(T_{ref}) = \int_{298}^{T_{ad}} \overline{c}_{P,i} dT$$

Sensible enthalpy



### **Exercise**

Estimate the constant-pressure adiabatic flame temperature for the combustion of stoichiometric CH<sub>4</sub>-air mixture. The pressure is 1atm and the initial reactant temperature is 298K. Assumptions: (1) Complete combustion, the product mixture consists of only CO<sub>2</sub>, H<sub>2</sub>O and N<sub>2</sub>; (2) The product mixture enthalpy is estimated using constant specific heats evaluated at 1200K ( $\approx 0.5(T_i + T_{ad})$ ), with a first guess of  $T_{ad}$  to be ca. 2100K.

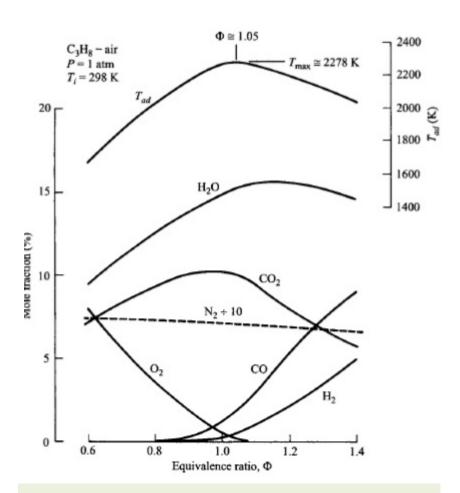
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# Discussions on adiabatic flame temperature

(text book 'An Introduction to Combustion : Concepts and Applications' – Chapter 2)

- How we could improve the accuracy of the calculated adiabatic flame temperature?
- What to consider for the calculation of adiabatic flame temperature when the equivalence ratio is less than or above one?

How pressure affects?



Equilibrium adiabatic flame temperature and major product species for propaneair combustion at 1 atm.

# Summary

- Combustion fundamentals
- Absolute enthalpy and sensible enthalpy
- Enthalpy of combustion
- Constant-pressure adiabatic flame temperature



### Nomenclature

a Molar oxygen–fuel ratio (kmol/kmol)

Constant-pressure specific heat (J/kmol-K)

 $h_f^o, \overline{h}_f^o$  Enthalpy of formation (J/kg or J/kmol)

 $H, h, \bar{h}$  Enthalpy (J or J/kg or J/kmol)

 $\Delta h_c$  Enthalpy of reaction (J or J/kg or J/kmol)  $\Delta H_R$ ,  $\Delta h$ ,  $\Delta \bar{h}_R$  Heat of combustion (heating value) (J/kg)

m Mass (kg)

MW Molecular weight (kg/kmol)

T Temperature (K)Y Mass fraction (kg/kg)

#### **Greek Symbols**

φ Equivalence ratioχ Mole fraction

#### Subscripts

ad Adiabatic

f Formation

*i*th species

*mix* Mixture

prodproductreacReactant

ref Reference

s Sensible

stoic Stoichiometric

#### **Superscripts**

o Denotes standard-state pressure (P° = 1 atm)

#### **Other Notation**

[X] Molar concentration of X (kmol/m $^3$ )