

MAE 298 Aeroacoustics – Homework #3

Generalized Differentiation and Farassat's Formulation

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Nomenclature

0	Subscript for undisturbed, quiescent parameters	$\bar{\partial}$	Generalized derivative
L	Subscript for loading parameters	δ	Dirac delta function
n	Surface normal direction	FW-H	Ffowcs Williams-Hawkings
s	Surface coordinate variable	ω	Wave oscillating frequency
t	time	i	Imaginary number $\sqrt{-1}$
\vec{x}	Observer location vector	exp	Exponential (e)
r	Distance between source and observer	E	Exponential term: $kz + n\theta - \omega t$
\hat{r}	Unit vector between source and observer	λ	Constant term in Bessel equation
θ	Angle between \hat{r} and \vec{x}	J	First-order Bessel function
f	Function of surfaces within a fluid space	Y	Second-order Bessel function
M	Mach number	$H^{(n)}$	nth-order Hankel function
$W(r)$	Radial distribution of mean axial velocity	x	Placeholder variable for λr
c	Speed of Sound	A, B	Arbitrary Bessel function constants
$\bar{\rho}$	Mean density	C, D	Arbitrary Hankel constants
γ	Specific heat ratio	\vec{V}	General velocity vector
p	Pressure	V_r	Velocity component in radial direction
\tilde{p}	Pressure (Discontinuous across data surface)	ν	Constant velocity parameter
\bar{p}	Mean pressure	χ	Constant position parameter
p'	Perturbation pressure	ζ	Position of vortex sheet dividing inner/outer solution
ΔP	Pressure difference from CFD solution	+/-	Outer/Inner solution, respectively
Φ	Piecewise-continuous variable		
k	Wavenumber		

Overview

In this assignment, we will first derive the wave equation for sound generated by a moving body using generalized differentiation.

Secondly, we will derive Farassat's Formulation 1A of the Ffowcs Williams-Hawkings (FW-H) equation for pressure due to loading noise beginning with Formulation 1

$$r = |\vec{x} - \vec{y}|$$

I. Problem 1 – Generalized Differentiation of Wave Equation

In this section, we will derive the wave equation for sound generated by a moving body, which is known as the Kirchhoff formula for moving surfaces. The general acoustic wave equation with no source term (homogeneous) is expressed as follows:

$$\frac{1}{c^2} \frac{\partial^2 p}{\partial t^2} - \nabla^2 p = 0 \quad (1)$$

We will use generalized differentiation to show that the wave equation whose sound is generated by an arbitrary moving body $f = 0$ can be expressed as follows:

$$\frac{1}{c^2} \frac{\partial^2 \tilde{p}}{\partial t^2} - \bar{\nabla}^2 \tilde{p} = - \left[\frac{M_n}{c} \frac{\partial p}{\partial t} + p_n \right] \delta(f) - \frac{1}{c} \frac{\partial}{\partial t} [M_n p \delta(f)] - \nabla \cdot [p \vec{n} \delta(f)] \quad (2)$$

where \vec{n} is the unit normal vector on the surface and $p_n = \nabla p \cdot \vec{n}$.

Now we can use the Greens function of the wave equation in the unbounded space, the so-called free-space Greens function, to find the unknown function $p(\vec{x}, t)$ everywhere in space. The result is the Kirchhoff formula for moving surfaces.

I.A. Data Surface Definition

To solve for the sound field of a moving surface, we will need to derive a source term to create the inhomogeneous version of Eqn 1. The moving surface will be defined as $f(\vec{x}, t) = 0$ and will be referred to as a “data surface”. The definition of f allows the fluid to be defined at all points in space, which can be divided into three regions:

Data Surface:	$f = 0$
Exterior Region:	$f > 0$
Interior Region:	$f < 0$

We will also assume that the gradient of f is the surface outward unit normal vector \hat{n} :

$$\nabla f = \hat{n} \quad (3)$$

Under this definition of f , the fluid will exist in all regions of unbounded space and thus enable the usage of Green’s function to solve for the sound field.

We will solve for the sound field in the exterior region, but we will assume that the fluid that extends into the interior region has the same conditions as an undisturbed, quiescent medium. We can define a pressure variable with these properties as \tilde{p} using the embedding technique as follows:

$$\tilde{p} = \begin{cases} p, & f > 0 \\ 0, & f < 0 \end{cases} \quad (4)$$

Thus, pressure will vary outside of the data surface and will be undisturbed inside. Though not continuous, this function of pressure will be defined at all points in an unbounded space and thus applicable to Green’s function.

I.B. Generalized Differentiation

Due to the discontinuity in \tilde{p} at the data surface $f = 0$, we will not be able to use standard differentiation to solve for the sound field and must instead use a technique called generalized differentiation, which will be denoted with a bar over the differential operator (e.g. $\bar{d}, \bar{\partial}, \bar{\nabla}$, etc).

To illustrate the principles of generalized differentiation, we will use the piecewise-continuous pressure \tilde{p} . The following rules will also apply to other piecewise-continuous fluid properties such as density $\tilde{\rho}$. Without a formal proof, we can state the rules of generalized differentiation as:

$$\frac{\bar{d}\tilde{p}}{dx} = \frac{d\tilde{p}}{dx} + \Delta\tilde{p}\delta(x - c) \quad (5)$$

where x is an arbitrary variable of differentiation, δ is the Dirac delta function, $x = c$ is the location of the discontinuity in \tilde{p} , and $\Delta\tilde{p} = \tilde{p}(c^+) - \tilde{p}(c^-) = p - p_0$ is the discontinuous difference in \tilde{p} at the location of

the discontinuity, or the difference in the parameter between the exterior and interior regions of the defined space.

For the case of the arbitrary moving body $f = 0$, where $x = f$ and $c = 0$, the Dirac delta function $\delta(x - c)$ becomes $\delta(f)$. Applying Eqn 5 to the gradient operator ∇ and using the chain rule, we obtain:

$$\begin{aligned}\bar{\nabla}\tilde{p} &= \frac{\bar{\partial}\tilde{p}}{\partial x_i} = \frac{\bar{\partial}\tilde{p}}{\partial f} \frac{\partial f}{\partial x_i} \\ &= \left(\frac{\partial\tilde{p}}{\partial f} + \Delta\tilde{p}\delta(f) \right) \frac{\partial f}{\partial x_i} \\ &= \frac{\partial\tilde{p}}{\partial f} \frac{\partial f}{\partial x_i} + \Delta\tilde{p} \frac{\partial f}{\partial x_i} \delta(f) \\ &= \frac{\partial\tilde{p}}{\partial x_i} + \Delta\tilde{p} \frac{\partial f}{\partial x_i} \delta(f)\end{aligned}$$

Substituting the gradient operator $\nabla = \frac{\partial}{\partial x_i}$:

$$\bar{\nabla}\tilde{p} = \nabla\tilde{p} + \Delta\tilde{p}\nabla f\delta(f) \quad (6)$$

Recall the assumption that gradient of f is the outward normal vector, as summarized in Eqn 3. Additionally, we can apply the definition of \tilde{p} to the pressure difference at the discontinuity $\Delta\tilde{p} = p - p_0 = p - 0 = p$. Making these substitutions, we can rewrite Eqn 6:

$$\bar{\nabla}\tilde{p} = \nabla\tilde{p} + p \cdot \hat{n}\delta(f) \quad (7)$$

II. Problem 2 – Farassat Formulation 1A for Loading Noise

Farassats formulation 1 for the loading noise is given as

$$4\pi p'_L = \frac{1}{c} \frac{\partial}{\partial t} \int_{f=0} \left[\frac{L_r}{r(1 - M_r)} \right]_{ret} ds + \int_{f=0} \left[\frac{L_r}{r^2(1 - M_r)} \right]_{ret} ds \quad (8)$$

where $L_r = \Delta P \vec{n} \cdot \hat{r} = \Delta P \cos \theta$. This formulation 1 is difficult to compute since the observer time differentiation is outside the integrals. A much more efficient and practical formulation can be derived by carrying the observer time derivative inside the integrals (formulation 1A). Show that formulation 1A for the loading noise becomes

$$4\pi p'_L = \frac{1}{c} \int_{f=0} \left[\frac{\dot{L}_r}{r(1 - M_r)^2} \right]_{ret} ds + \int_{f=0} \left[\frac{L_r - L_M}{r^2(1 - M_r)} \right]_{ret} ds + \frac{1}{c} \int_{f=0} \left[\frac{L_r[r\dot{M}_r + c(M_r - M^2)]}{r^2(1 - M_r)^3} \right]_{ret} ds \quad (9)$$

where $L_M = \vec{L} \cdot \vec{M}$.

Conclusion

what you could use moving surface formula for (airfoil)

how you would use formulation 1a in practice with CFD and numerical methods