

Delhi | Bhopal | Hyderabad | Jaipur | Lucknow | Pune | Bhubaneswar | Kolkata

Web: www.madeeasy.in | **E-mail:** info@madeeasy.in | **Ph:** 011-45124612

STRENGTH OF MATERIALS

MECHANICAL ENGINEERING

Date of Test: 16/03/2023

ANSWER KEY ➤

1.	(b)	7.	(c)	13.	(c)	19.	(a)	25.	(d)
2.	(b)	8.	(a)	14.	(b)	20.	(b)	26.	(b)
3.	(c)	9.	(d)	15.	(a)	21.	(a)	27.	(c)
4.	(d)	10.	(b)	16.	(b)	22.	(a)	28.	(b)
5.	(c)	11.	(c)	17.	(c)	23.	(c)	29.	(d)
6.	(d)	12.	(c)	18.	(d)	24.	(b)	30.	(d)

DETAILED EXPLANATIONS

1. (b)

8

Thickness of cylindrical portion, $t_1 = 3 \text{ mm}$

Thickness of hemispherical ends = t_2

For no distortion of the junction under pressure,

$$\frac{t_2}{t_1} = \frac{1-\mu}{2-\mu} = \frac{1-0.3}{2-0.3}$$

$$t_2 = 1.235 \,\mathrm{mm}$$

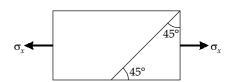
2. (b)

Given: P = 11 kN, $A = 150 \times 75 \text{ mm}^2$

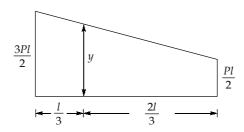
$$\sigma_x = \frac{P}{A} = \frac{11000}{150 \times 75} = 0.977 \text{ MPa}$$

$$(\sigma_n)_{\theta = 45^\circ} = \frac{\sigma_x}{2} + \frac{\sigma_x}{2} \cos 2\theta$$

$$=\frac{\sigma_x}{2} = \frac{0.977}{2} = 0.488 \text{ MPa}$$



3. (c)



$$At = \frac{2l}{3},$$

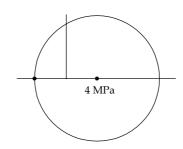
$$y = \frac{Pl}{2} + \frac{2Pl}{3} = \frac{7Pl}{6}$$

$$\theta_x - \theta_0 = \text{Area of } \frac{M}{EI} \text{diagram}$$

$$= \frac{1}{2} \left(\frac{3Pl}{2} + \frac{7Pl}{6} \right) \frac{l}{3EI}$$

$$= \frac{4}{9} \frac{Pl^2}{EI}$$

4. (d)



CT-2023-24

Radius of Mohr circle = 4 - (-4) = 8 MPa Maximum principal stress = 4 + 8 = 12 MPa

$$\frac{K}{G} = \frac{2(1+\mu)}{3(1-2\mu)}$$
$$= \frac{2(1.25)}{3(0.5)} = 1.67$$

$$R_A + R_B = P + wl$$

$$R_A + 6 = P + 6$$

$$R_A = P$$

$$\sum M_A = 0$$

$$6 \times 3 = P \times 2 + 2 \times \frac{3^2}{2}$$

$$18 = 2P + 9$$

$$P = 4.5 \text{ kN}$$

$$R_A = 4.5 \text{ kN}$$

:.

$$w = 5 \text{ kN/m}, q = (10 - 5) = 5 \text{ kN/m}$$

$$B.M. = \frac{wl^2}{2} + \frac{ql^2}{6}$$

$$= \frac{5 \times 4^2}{2} + \frac{5 \times 4^2}{6} = 53.33 \text{ kNm}$$

$$\frac{M}{I} = \frac{E}{R}$$

$$M = \frac{200 \times 10^{3}}{2000} \times 42 \times \frac{\pi}{180} \times \frac{18 \times 6^{3}}{12}$$

$$= 23760 \text{ Nmm} \simeq 23.76 \text{ N.m}$$

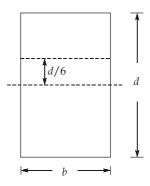
Minimum moment of inertia =
$$\frac{12 \times 6^3}{12}$$
 = 216 mm⁴

Radius of gyration = $\sqrt{\frac{I}{A}}$ = $\sqrt{\frac{216}{12 \times 6}}$ = $\sqrt{3}$

Slenderness ratio = $\frac{L_e}{k}$

$$L_e = 100 \times \sqrt{3} = 173.21 \text{ mm}$$

11. (c)



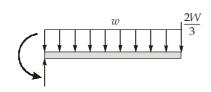
Maximum shear force in beam = $\frac{P}{2}$

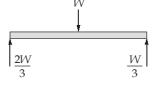
Shear stress,
$$\tau = \frac{FA\overline{y}}{Ib} = \frac{\frac{P}{2} \times \frac{bd}{3} \times \frac{d}{3}}{\frac{bd^3}{12} \times b} = \frac{2P}{3bd}$$

12. (c)

$$\begin{split} \delta_{impact} &= \delta_{st} \left(1 + \sqrt{1 + \frac{2h}{\delta_{st}}} \right) \\ \delta_{st} &= \frac{Pl}{AE} = \frac{500 \times 10^3 \times 1000}{50 \times 10^2 \times 200 \times 10^3} = 0.5 \text{ mm} \\ \frac{\delta_{impact}}{\delta_{gradually}} &= \left(1 + \sqrt{1 + \frac{2 \times 12}{0.5}} \right) = 8 \end{split}$$

13. (c)





$$\delta_{\text{centre}} = \frac{wa^4}{8EI} + \frac{2Wa^3}{9EI}$$

14. (b)

Strain energy stored in AB, CD

$$= \frac{P^2 L}{2\frac{\pi}{4}D^2 E}$$

Strain energy stored in BC= $\frac{P^2(2L)}{2\frac{\pi}{4}\frac{D^2}{4}E} = \frac{8P^2L}{2\frac{\pi}{4}\times D^2E}$

% Strain energy stored in BC =
$$\frac{\frac{8P^2L}{2\frac{\pi}{4}D^2E}}{\frac{P^2L}{2\frac{\pi}{4}D^2E} + \frac{8P^2L}{2\frac{\pi}{4}D^2E} + \frac{P^2L}{2\frac{\pi}{4}D^2E}} = \frac{8}{1+8+1} = \frac{8}{10}$$
$$= 80\%$$

15. (a)

As rod and tube are in parallel, so angle of twist will be same

$$\frac{T_R l}{G \frac{\pi}{32} d^4} = \frac{T_T l}{\frac{G}{3} \frac{\pi}{32} ((2d)^4 - d^4)}$$

$$5T_R = T_T$$

$$T_R + T_T = T$$

$$T_R + 5T_R = T$$

$$T_R = \frac{T}{6}$$

$$\theta_R = \theta_T = \frac{T}{6} \times \frac{32l}{G\pi d^4} = \frac{16Tl}{3\pi G d^4}$$

16. (b)

© Copyright: MADE EASY

Change of length of bar = Compression of spring

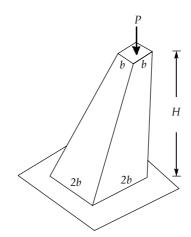
$$\int_{0}^{L} \alpha \Delta T(x) dx - \frac{R_A L}{AE} = \frac{R_A}{K}$$

$$\alpha \Delta T_0 \left(x - \frac{x^3}{3L^2} \right)_{0}^{L} = \frac{R_A L}{AE} + \frac{R_A}{K}$$

$$\frac{2}{3} \alpha \Delta T_0 L = R_A \left(\frac{L}{AE} + \frac{1}{K} \right)$$

$$R_A = \frac{2}{3} \frac{\alpha \Delta T_0 L KAE}{(KL + AE)}$$

17. (c)



$$\delta = \int_{0}^{H} \frac{Pdx}{AE} = \int_{0}^{H} \frac{Pdx}{\left(b + \frac{bx}{H}\right)^{2} E}$$

$$b + \frac{bx}{H} = t$$

$$\frac{dt}{dx} = \frac{b}{H}$$

$$dx = dt \left(\frac{H}{b}\right)$$

when x = 0, t = b; x = H, t = 2b

$$\delta = \frac{H}{b} \int_{b}^{2b} \frac{P}{t^2} \frac{dt}{E} = \frac{HP}{bE} \times \left(-\frac{1}{t}\right)_{b}^{2b}$$

$$\delta = \frac{HP}{Eb} \left(\frac{1}{b} - \frac{1}{2b} \right) = \frac{HP \times 1}{Eb \times 2b}$$

$$\delta = \frac{HP}{2b^2E}$$

18. (d)

$$\sigma_{x} = -\sigma$$

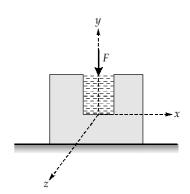
$$\sigma_{z} = -\sigma$$

$$\varepsilon_{x} = 0 = \frac{\sigma_{x}}{E} - \frac{\mu}{E} (\sigma_{y} + \sigma_{z})$$

$$= -\sigma - \mu (\frac{-F}{A} - \sigma)$$

$$\sigma = +\mu \frac{F}{A} + \mu \sigma = \frac{\mu}{1 - \mu} \frac{F}{A}$$

$$\varepsilon_{y} = -\frac{F}{AE} - \frac{\mu}{E} (-2\sigma)$$



$$\varepsilon_y = \frac{F}{AE} \left(\frac{2\mu^2}{1-\mu} - 1 \right)$$

$$\varepsilon_y = \frac{F}{AE} \left(\frac{2\mu^2 - 1 + \mu}{1 - \mu} \right)$$

$$\delta = \frac{Fl}{AE} \left(\frac{2\mu^2 + \mu - 1}{1 - \mu} \right)$$

19. (a)

Strain energy =
$$\frac{T^2 l}{2GJ}$$

= $\frac{T_0^2}{2G} \times \left(\frac{2l}{3}\right) \times \frac{32}{\pi(d^4)} + \frac{T_0^2}{2G} \times \left(\frac{4l}{3}\right) \times \frac{32 \times 16}{\pi \times d^4}$
= $\frac{T_0^2 l}{2Gd^4} \left[\frac{2}{3} \times \frac{32}{\pi} + \frac{4}{3} \times \frac{32}{\pi} \times 16\right]$
= $112 \frac{T_0^2 l}{Gd^4}$

20. (b)

The displacement of point 'C'

$$\begin{split} \delta_{C} &= \left(\delta_{AB}\right)_{self\ weight} + \left(\delta_{AB}\right)_{weight\ of\ BC} + \left(\delta_{BC}\right)_{Self\ weight} \\ &= \left.\frac{WL}{2AE}\right|_{AB} + \frac{P_{ext}L}{AE}\bigg|_{AB} + \frac{WL}{2AE}\bigg|_{BC} \end{split}$$

From given data,

$$W_{BC} = W_{AC} - W_{AB}$$

$$= 3W - W = 2W$$

$$W_{AB} = W$$

$$(P_{ext})_B = 2W$$
So,
$$\delta_C = \frac{WL}{2AE} + \frac{2WL}{AE} + \frac{(2W) \times L}{2(2A)E}$$

$$\delta_C = \frac{WL}{AE} \left[\frac{2 + 8 + 2}{4} \right] = \frac{12WL}{4AE}$$

$$\delta_C = \frac{3WL}{AE}$$

21.

The applied stress in the direction of thickness of plate,

$$\sigma_{z} = 0$$

Strain along thickness direction,

$$\varepsilon_z = \frac{\sigma_z}{E} - \frac{\mu}{E} (\sigma_x + \sigma_y) \qquad ...(i)$$

As we know that,

$$\sigma_{1} = \frac{E}{1-\mu^{2}} (\epsilon_{1} + \mu \epsilon_{2})$$
So, we can write,
$$\sigma_{x} = \frac{209 \times 10^{3}}{1-0.3^{2}} [50 \times 10^{-5} + 0.3 \times 15 \times 10^{-5}]$$

$$= 125.17 \text{ MPa}$$
Similarly,
$$\sigma_{y} = \frac{209 \times 10^{3}}{1-0.3^{2}} [15 \times 10^{-5} + 0.3 \times 50 \times 10^{-5}]$$

 $\sigma_{y} = 68.9 \text{ MPa}$

From equation (i)

$$\varepsilon_z = 0 - \frac{0.3}{209 \times 10^3} [125.17 + 68.9]$$

= 2.785 × 10⁻⁴

Reduction in thickness of plate,

$$\delta t = t \times \varepsilon_z$$

= 20 × 2.785 × 10⁻⁴
= 5.571 × 10⁻³ mm

22. (a)

Point *A* and *B* on the Mohr's circle represents the complementary planes. So shear stress will be same, i.e. $\tau_A = \tau_B$

Radius of the Mohr's circle, CA

$$CA = \sqrt{30^{2} + 40^{2}} = 50$$

$$CB' = \sqrt{50^{2} - 40^{2}} = 30$$

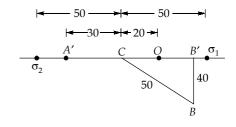
$$OB' = CB' - CO = 30 - 20 = 10$$

$$\sigma_{B} = OB' = 10$$

$$\sigma_{1} = 50 - 20 = 30$$

$$\sigma_{2} = -(50 + 20) = -70$$

$$\sigma_{A} = OA' = 30 + 20 = 50$$



23. (c)

Strain tensor =
$$\begin{bmatrix} \in_{xx} & \frac{\gamma_{xy}}{2} & \frac{\gamma_{xz}}{2} \\ \frac{\gamma_{yx}}{2} & \in_{yy} & \frac{\gamma_{yz}}{2} \\ \frac{\gamma_{zx}}{2} & \frac{\gamma_{zy}}{2} & \in_{zz} \end{bmatrix}$$

$$\gamma_{xy} = 0.004 \times 2 = 0.008$$

$$\gamma_{xz} = 0.006 \times 2 = 0.012$$

$$\tau_{xy} = G \gamma_{xy} = 100 \times 0.008 = 0.8 \text{ GPa}$$

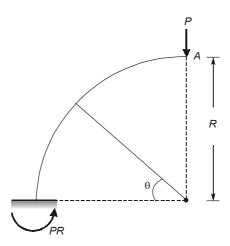
$$\tau_{xz} = G \gamma_{xz} = 100 \times 0.012 = 1.2 \text{ GPa}$$

$$\tau_{xy} + \tau_{xz} = 800 + 1200 = 2000 \text{ MPa}$$

and

So,

24. (b)



$$M = PR \cos\theta$$

$$\frac{\partial M}{\partial P} = R \cos \theta$$

Now,

$$\delta_v = \frac{\partial U}{\partial P}$$

(where U = strain energy)

$$\frac{\partial U}{\partial P} = \int_{0}^{\pi/2} \frac{M \times \left(\frac{\partial M}{\partial P}\right) R d\theta}{EI}$$
$$= \frac{PR^3}{EI} \int_{0}^{\pi/2} \cos^2 \theta d\theta$$
$$\delta_v = \frac{\pi PR^3}{4EI}$$

25. (d)

From equilibrium, $\Sigma V = 0$,

$$\Rightarrow R_A + R_B = 2 + 2 \times 4 = 10 \text{ kN}$$

$$\Sigma M = 0,$$

$$\Rightarrow 2 \times 4 \times 2 + 2 \times 6 = R_B \times 4$$

So,
$$R_B = \frac{16+12}{4} = \frac{28}{4} = 7 \text{ kN}$$

and
$$R_A = 3 \text{ kN}$$

Bending moment at a distance x' from end x' will be

$$(BM)_x = 3x - 2 \times x \times \frac{x}{2} = 3x - x^2$$

Now, the BMD changes sign in section AB, so the point of contraflexure is where the BM is zero.

So,
$$3x - x^2 = 0$$
$$x = 3 \text{ m}$$

26. (b)

$$I = \frac{\pi R^4}{4} - 2\left[\frac{\pi}{4}\left(\frac{R}{3}\right)^4 + \pi\left(\frac{R}{3}\right)^2\left(\frac{R}{2}\right)^2\right]$$

$$I = 0.592 R^4$$

Bending stress will be maximum at point *A* and *B*.

So,
$$\sigma_{\text{max}} = \frac{P \times 1.5 \times 0.1}{0.592 \times (0.1)^4}$$

$$\therefore \qquad P = 236.8 \text{ kN}$$

27. (c)

$$\sigma_{1} = \left(\frac{\sigma_{x} + \sigma_{y}}{2}\right) + \sqrt{\left(\frac{\sigma_{x} - \sigma_{y}}{2}\right)^{2} + \left(\tau_{xy}\right)^{2}}$$

$$= \frac{200 + 120}{2} + \sqrt{\left(\frac{200 - 120}{2}\right)^{2} + 30^{2}}$$

$$= 210 \text{ MPa}$$

28. (b)

Column I =
$$P_{cr} = \frac{2\pi^2 EI}{h_1^2}$$

Column II = $P_{cr} = \frac{4\pi^2}{3} \frac{EI}{h_2^2}$
 $\frac{2\pi^2 EI}{h_1^2} = \frac{4\pi^2 EI}{3h_2^2}$
 $\frac{h_2}{h_1} = \sqrt{\frac{2}{3}} = 0.82$

29. (d)

$$\gamma_{\text{max}} = r_{\text{max}} \frac{\theta}{l}$$

$$\gamma = 35 \times \frac{22 \times 1.8}{7 \times 180} \times \frac{1}{1500}$$

$$= 7.33 \times 10^{-4}$$

30. (d)

$$EI\frac{d^4y}{dx^4} = \sqrt{\frac{x}{L}}q$$

$$V_x = EI\frac{d^3y}{dx^3} = \frac{2x^{3/2}}{3\sqrt{L}}q + C_1$$

$$M_x = EI\frac{d^2y}{dx^2} = \frac{2}{3} \times \frac{2}{5} \times \frac{x^{5/2}}{\sqrt{L}}q + C_1x + C_2$$

$$x = 0$$

$$\Rightarrow \qquad V_x = 0$$

$$C_1 = 0$$

$$X =$$