

Declaration & Statement of Authorship

I, bearing Registration number, 103117086, agree and acknowledge that:

1. The assessment was answered by me as per the instructions applicable to each assessment and that I have not resorted to any unfair means to deliberately improve my performance.
2. I have neither impersonated anyone, nor have I been impersonated by any person for the purpose of assignments.

Signature of the student: Sanishma Bhandari
+1/12/2020

Full name: Sanishma Bhandari

Roll number: 103117086

Sub code: CEPC27, Advanced Steel Structural Elements

Mobile number: +977 9841605599

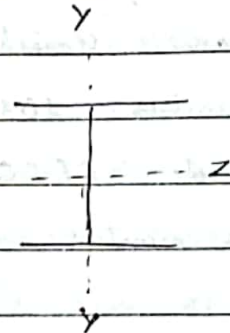
2. Solution.

(Given:-

Axial load (P) = 1500 kN (Tensile)Major axis bending moment = 80 kNm. (M_z)Minor axis bending moment = 30 kNm. (M_y)effective length = 4 m. (KL)

The section used is ISHB 350 (666.1 N/m)

The dimensions of ISHB 350 are given below.

Sectional area = 85.91 cm^2 $h = 350 \text{ mm}$ $b_f = 250 \text{ mm}$ $t_f = 11.6 \text{ mm}$ $t_w = 8.3 \text{ mm}$ $r_{zz} = 14.93 \text{ cm}$ $r_{yy} = 5.34 \text{ cm}$ $Z_{zz} = 1094.8 \text{ cm}^3$ $Z_{yx} \text{ (plastic modulus)} = 1213.53 \text{ cm}^3$ $Z_{yy} = 196.1 \text{ cm}^3$ 

The section has to satisfy both member level and section level interaction equations:

A) Section Level.

$$\frac{P}{P_{cap}} + \frac{M_y}{M_{ycapacity}} + \frac{M_z}{M_{zcapacity}} \leq 1.$$

$$i) P_{capacity} = \frac{f_y \times A_g}{\phi_{mo}} = \frac{250 \times 85.91 \times 100}{1.1} = 1952.5 \text{ kN}$$

ii) M_z , capacity
From table 2;

$$\frac{b}{t_f} = \frac{125}{11.6} = 10.776 \leq 15.7$$

\therefore our section is semi-compact.

$$M_z, capacity = \frac{f_y (Z_e)_z}{\phi_{mo}} = \frac{250 \times 1094.8 \times 10^3}{1.1} = 248.818 \text{ kN-m}$$

iii) M_y , capacity.

$$M_y, cap = \frac{f_y (Z_e)_y}{\phi_{mo}} = \frac{250 \times 196.1 \times 10^3}{1.1} = 44.5682 \text{ kN-m}$$

Checking the interaction equation for section level,

$$\begin{aligned} & \frac{P}{P_{cap}} + \frac{M_y}{M_{y, cap}} + \frac{M_z}{M_{z, cap}} \\ &= \frac{1500}{1952.5} + \frac{30}{44.5682} + \frac{80}{248.818} \\ &= 0.7682 + 0.673 + 0.3215 \\ &= 1.763 > 1 \end{aligned}$$

\therefore The member doesn't satisfy section level,

B) Member level.

$$\frac{P}{P_{dy}} + K_y \frac{C_{my} M_y}{M_{dy}} + K_{LT} \frac{M_z}{M_{dz}} \leq 1.0$$

$$\frac{P}{P_{dz}} + 0.6 K_y \frac{C_{my} M_y}{M_{dy}} + K_z \frac{C_{mz} M_z}{M_{dz}} \leq 1.0$$

For P_{dy} ,

$$\frac{KL}{D_y} = \frac{4000}{53.4} = 74.906 \sim 80 (1st)$$

fcd value; First check, buckling class,

$$\text{Table 10, } \frac{h}{b_f} = \frac{350}{250} = 1.4.$$

$$\frac{h}{b_f} > 1.2.$$

$$t_f = 11.6 \text{ mm} \leq 40 \text{ mm},$$

buckling class, y-y \rightarrow 'b' buckling class,

$$\therefore f_{cd} = 150 \text{ N/mm}^2.$$

$$P_{dy} = \frac{f_{cd} \times A}{\gamma_{mo}} = 1171.5 \text{ kN}$$

For, P_{dz} ,

$$\frac{KL}{D_z} = \frac{4000}{149.3} = 26.79 \sim 30$$

$$\therefore f_{cd} = 216 \text{ N/mm}^2$$

$$P_{dz} = \frac{f_{cd} \times A}{\gamma_{mo}} = 1686.96 \text{ kN}$$

For M_{dz}

$$M_{dz} = \beta_b Z_p f_{bd}$$

Now,

$$\frac{K L}{D_x} = \frac{4000}{149.3} = 26.79 \sim 30$$

$$\phi_{LT} = 0.21$$

$$\beta = 1$$

$$\frac{h}{t_f} = \frac{350}{11.6} = 30.17 \sim 30$$

From Table 14,

$$f_{cr,b} = 2,457.1 \text{ N/mm}^2$$

From Table 13(a)

$$f_{bd} = 227.3 \text{ N/mm}^2$$

$$Z_p = 1213.53 \text{ cm}^3$$

$$M_{dz} = 1 \times 1213.53 \times 10^3 \times 227.3$$

$$= 275.835 \text{ kN-m}$$

For KLT

since both top and bottom are same $\psi = 1$.

$$C_{mLT} = 0.6 + 0.4(1) = 1.0 \geq 0.4 \text{ okay,}$$

$$C_{my} = C_{mL} = 1.0$$

$$\eta_y = \frac{P}{P_y} = 1.28$$

$$\lambda_{LT} = \sqrt{\frac{f_y}{f_{cr,b}}} = 0.319$$

$$K_{LT} = 1 - \frac{(0.1 \lambda_{LT} \eta_y)}{(C_{mLT} - 0.25)} = 0.9455 > 0.83 \text{ so okay!!!}$$

$$1 - \frac{0.1 \eta_y}{C_{mLT} - 0.25} = 0.83$$

$$\therefore K_{LT} = 0.9455$$

For K_z

$$n_z = \frac{P}{P_z} = 0.889$$

$$K_z = 1 + (\lambda_z - 0.2)n_z \leq 1 + 0.8n_z$$

or,

$$\lambda_z = \sqrt{\frac{F_y}{I_{cc}}} = \sqrt{\frac{F_y \left(\frac{K L}{P}\right)^2}{\pi^2 E}} = \sqrt{\frac{250 \times 30^2}{\pi^2 \times E}} = 0.3376$$

$$\therefore K_z = 1 + (0.3376 - 0.2) \times 0.889 = 1.122 < 1.7112 \text{ okay!!}$$

For K_y

$$K_y = 1 + (\lambda_y - 0.2)n_y$$

$$n_y = \frac{P}{P_y} = 1.28$$

$$\lambda_y = \sqrt{\frac{F_y \left(\frac{K L}{P}\right)^2}{\pi^2 E}} = \sqrt{0.9} < 2.024 \text{ okay!!}$$

$$K_y \leq 1.896 < 2.024$$

Interaction equations

$$\frac{P}{P_d} + K_y \frac{C_{my} M_y}{M_{dy}} + K_{LT} \frac{M_z}{M_{dz}} \geq 1$$

=

\therefore Member is not susceptible acceptable

1

Solution,

1. Truss configuration: Let the inclination of the roof with the horizontal be ' α '.

$$\tan \alpha = 2/5$$

$$\therefore \alpha = 21.8^\circ$$

$$\text{Length of rafter} = \sqrt{5^2 + 2^2} = 5.38 \text{ m}$$

$$\text{Length of each panel} = \frac{5.38}{2} = 2.69 \text{ m}$$

Loads on panel points

Dead load:

$$\text{Dead load of C.I sheets with fixings} = 200 \text{ N/m}^2$$

$$\text{self weight of truss} = 60 \text{ N/m}^2$$

$$\text{self weight of purlin} = 70 \text{ N/m}^2 = 70 \times 3 = 210 \text{ N}$$

$$\text{panel length} = 2.69 \text{ m}$$

Now,

load on each intermediate node because of the dead load is given as,

$$= (200 + 60) \times (3 \times 2.69) + 210$$

$$= 2308.2 \text{ N} = 2.308 \text{ kN}$$

$$\text{Load on the end panel points of the rafter} = \frac{2308.2}{2} = 1154.1 = 1.154 \text{ kN}$$

$$\text{Load on intermediate purlin} = (200 + 60 + 70) \times 2.69 = 887.7 \text{ N} = 0.887 \text{ kN/m}$$

$$\text{Load on the end purlin} = \frac{0.887}{2} = 0.4435 \text{ kN/m}$$

3. For the analysis of beams.

$$W_x (\text{purlin}) = 0.867 \cos 21.8^\circ \\ = 0.82 \text{ kN/m.}$$

$$W_y (\text{purlin}) = 0.867 \sin 21.8^\circ \\ = 0.33 \text{ kN/m.}$$

Now, for the moment,

$$M_z = \frac{0.82 \times 3^2}{8} = 92.25 \times 0.9225 \text{ kN-m}$$

$$M_y = \frac{0.33 \times 3^2}{8} = 0.37125 \text{ kN-m.}$$

Now,

Live load.

$$L.L = 300 \text{ N/m}^2$$

$$\therefore \text{L.L on intermediate panel} = 300 \times 3 \times 2.69 = 2421 \text{ N} = 2.421 \text{ kN.}$$

2. Wind load calculations.

→ The given location is tricky and building location has large openings.

Therefore, the basic speed is,

$$V_b = 47 \text{ m/s.}$$

The terrain and topography of location are open with well scattered obstructions and therefore,

span of building = 5 years.

$$\text{wind load} = (C_{pe} - C_{pi}) \times A \times P_d$$

$$\text{The roof angle} = \tan^{-1}(2/5)$$

$$= 21.8^\circ.$$

$$\therefore \text{The design wind speed } (V_3) = K_1 \times K_2 \times K_3 \times V_b$$

K_1 - a/c to Table 1 (IS 875 part 3)

for 5 years and basic speed of 47 m/s,

$$K_1 = 0.71$$

$$K_2 = 1 \quad [\text{For terrain category of 2}]$$

For flat land, $K_3 = 1$

$$\text{The design wind speed} = K_1 \times K_2 \times K_3 \times 47$$

$$\text{or, } V_d = 1 \times 0.71 \times 1 \times 47$$

$$\therefore V_d = 33.37 \text{ m/s.}$$

$$\text{Design wind pressure } (P_d) = 0.6 (V_d)^2$$

$$= 0.6 \times (33.37)^2$$

$$= 668.134 \text{ N/m}^2$$

Now,

Wind pressure coefficient

→ Skinning building with large openings (greater than 20% area)
 $C_{pi} = \pm 0.7$

since our $h=5$, $w=10$

$$\therefore h/w = 0.5 \leq 1/2(0.5)$$

i) From Table 6, θ wind angle as 0°

our,

$$\begin{aligned} C_{pe} \text{ for } 20^\circ &= -0.4 \text{ (on windward direction)} \\ &= -0.4 \text{ (on leeward direction)} \end{aligned}$$

For,

$$\begin{aligned} C_{pe} \text{ } 30^\circ, \quad C_{pe} &= 0 \text{ (for windward)} \\ &= -0.4 \text{ (for leeward)} \end{aligned}$$

Doing interpolation,

For 21.8° ,

$$\text{our, } C_{pe} \text{ for windward} = -0.32$$

$$C_{pe} \text{ for leeward} = -0.4.$$

ii) For wind angle $= 20^\circ$,

$$\begin{aligned} C_{pe} \text{ for } 20^\circ &= -0.7 \text{ (windward)} \\ &= -0.6 \text{ (leeward)} \end{aligned}$$

$$\begin{aligned} C_{pe} \text{ for } 30^\circ &= -0.7 \text{ (windward)} \\ &= -0.6 \text{ (leeward)} \end{aligned}$$

Using interpolation for 21.8° ,

$$C_{pe} \text{ (windward)} = -0.7$$

$$C_{pe} \text{ (leeward)} = -0.6$$