

# GATE - 2014- AE

EE1030 : Matrix Theory  
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- 1) For a given fuel flow rate and thermal efficiency, the take-off thrust for a gas turbine engine burning aviation turbine fuel (considering fuel-air ratio  $f \ll 1$ ) is
  - a) Directly proportional to exhaust velocity
  - b) Inversely proportional to exhaust velocity
  - c) Independent of exhaust velocity
  - d) Directly proportional to the square of the exhaust velocity
  
- 2) For a fifty percent reaction axial compressor stage, following statements are given:
  - I. Velocity triangles at the entry and exit of the rotor are symmetrical
  - II. The whirl or swirl component of absolute velocity at the entry of rotor and entry of stator are same.Which of the following options are correct?
  - a) Both I and II are correct statements
  - b) I is correct but II is incorrect
  - c) I is incorrect but II is correct
  - d) Both I and II are incorrect
  
- 3) A small rocket having a specific impulse of  $200s$  produces a total thrust of  $98kN$ , out of which  $10kN$  is the pressure thrust. Considering the acceleration due to gravity to be  $9.8m/s^2$ , the propellant mass flow rate in  $kg/s$  is
  - a) 55.1
  - b) 44.9
  - c) 50
  - d) 60.2
  
- 4) The thrust produced by a turbojet engine
  - a) Increases with increasing compressor pressure ratio
  - b) Decreases with increasing compressor pressure ratio
  - c) Remains constant with increasing compressor pressure ratio
  - d) First increases and then decreases with increasing compressor pressure ratio

- 5) The moment coefficient measured about the centre of gravity and about aerodynamic centre of a given wing-body combination are 0.0065 and -0.0235 respectively. The aerodynamic centre lies 0.06 chord lengths ahead of the centre of gravity. The lift coefficient for this wing-body is \_\_\_\_.
- 6) The vertical ground load factor on a stationary aircraft parked in its hangar is:
- 0
  - 1
  - Not defined
  - 1
- 7) Under what condition should a glider be operated to ensure minimum sink rate?
- Maximum  $\frac{C_L}{C_D}$
  - Minimum  $\frac{C_L}{C_D}$
  - Maximum  $\frac{C_D}{C_L^{\frac{2}{3}}}$
  - Minimum  $\frac{C_D}{C_L^{\frac{2}{3}}}$
- 8) In most airplanes, the Dutch roll mode can be excited by applying
- a step input to the elevators
  - a step input to the rudder
  - a sinusoidal input to the aileron
  - an impulse input to the elevators
- 9) Considering  $\mathbf{R}$  as the radius of the moon, the ratio of the velocities of two spacecraft orbiting moon in circular orbit at altitudes  $\mathbf{R}$  and  $\mathbf{2R}$  above the surface of the moon is \_\_\_\_.
- 10) If  $[A] = \begin{bmatrix} 3 & -3 \\ -3 & 4 \end{bmatrix}$ . Then  $\det(-[A]^2 + 7[A] - 3[I])$  is
- 0
  - 324
  - 324
  - 6
- 11) For the periodic function given by

$$f(x) = \begin{cases} -2, & -\pi < x < 0 \\ 2, & 0 < x < \pi \end{cases}$$

with  $f(x + 2\pi) = f(x)$ , using Fourier series, the sum

$$s = 1 - \frac{1}{3} + \frac{1}{5} - \frac{1}{7} + \dots$$

converges to

- a) 1
- b)  $\frac{\pi}{3}$
- c)  $\frac{\pi}{4}$
- d)  $\frac{\pi}{5}$

12) Let  $\Gamma$  be the boundary of the closed circular region  $A$  given by  $x^2 + y^2 \leq 1$ . Then

$$I = \int_{\Gamma} (3x^3 - 9xy^2) ds$$

(where  $ds$  means integration along the bounding curve) is

- a)  $\pi$
- b)  $-\pi$
- c) 1
- d) 0

13) Solution to the boundary-value problem

$$-9 \frac{d^2 u}{dx^2} + u = 5x, \quad 0 < x < 3$$

with  $u(0) = 0$ ,  $\left. \frac{du}{dx} \right|_{x=3} = 0$  is

- a)  $u(x) = \frac{15e}{1+e^2} \left( e^{-\frac{x}{3}} - e^{\frac{x}{3}} \right) + 5x$
- b)  $u(x) = \frac{15e}{1+e^2} \left( e^{-\frac{x}{3}} + e^{\frac{x}{3}} \right) + 5x$
- c)  $u(x) = -\frac{15 \sin(\frac{x}{3})}{\cos(1)} + 5x$
- d)  $u(x) = -\frac{15 \sin(\frac{x}{3})}{\cos(1)} - \frac{5}{54} x^3$