

Dynamics - simple linear oscillator

Single elastic spring

Degree of freedom

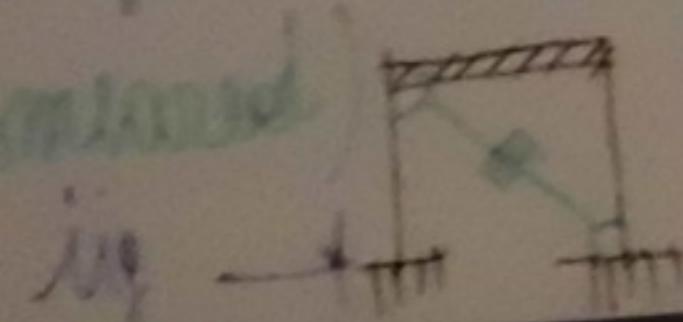
time-invariant

Freedom

systems (dynamic)

They represent physical properties of a structural system

- stiffness (example)
- damping (example)



Viscous damper

$$F_d = c\dot{u}$$

& damping different

$$[K] = \frac{F}{L} \quad [F] = \frac{dU}{L}$$

$$m = \frac{F}{a} \quad a = \frac{1}{L} F_d$$

$F(t)$ displacement
 $\dot{u}(t)$ velocity
 $\ddot{u}(t)$ acceleration
elastic energy quadratic function
external action

$$m \text{ mass} \quad [M] \quad F_e = Ku \quad \text{work} \quad \frac{1}{2} Ku^2$$

$$[F] \quad K = \frac{F_e}{u} \quad 2E = \frac{1}{2} F_e u = \frac{1}{2} \frac{1}{R} u^2$$

$$[L] \quad F_e = \frac{dE}{du} = Ku$$

$$\frac{d}{dt} \left(\frac{1}{2} Ku^2 \right) = \frac{1}{2} F_d \dot{u} \quad \text{power}$$

$$F_d = \frac{d}{dt} \left(\frac{1}{2} Ku^2 \right) \quad \text{ii } \ddot{u} = \frac{1}{L} F_d$$

$$\text{iii Rayleigh dissipation function} \quad F_d = \frac{dD}{du} = cu$$

Equation of motion (d'Alembert principle)

$$F_e = Ku \quad F = ma \quad F - ma = 0$$

$$F_d = c\dot{u} \quad \text{inertia force } F_i$$

dynamic equilibrium

$$Ku(t) + c\dot{u}(t) = F(t) - m\ddot{u}$$

$$m\ddot{u}(t) + c\dot{u}(t) + Ku(t) = F(t)$$

2nd order linear differential equation in $u(t)$

$t \geq t_0 = 0$
Initial conditions

$$\begin{cases} u(t=t_0=0) = u_0 \\ \dot{u}(t=t_0=0) = \dot{u}_0 \end{cases}$$

By Lagrange's equations

$$\text{Lagrange function } L(q, \dot{q}) = T - V + U$$

$$L = \frac{1}{2} m \dot{q}^2 - V(q)$$

$$q_1 = u \quad q_2 = \dot{u}$$

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_1} - \frac{\partial L}{\partial q_1} = \frac{\partial L}{\partial q_2} = \frac{d}{dt} F$$

$$\text{SDOF } u(t) = \frac{1}{2} m \dot{u}^2; \quad V = \frac{1}{2} K u^2$$

$$\frac{d}{dt} \left(\frac{1}{2} m \dot{u}^2 \right) - \frac{1}{2} K u^2 = F - F_d$$

$$m\ddot{u} + Ku = F - F_d$$

T = Kinetic energy relative

$$P = F \cdot v \quad a = \frac{dv}{dt}$$

$$\text{power} = m \cdot v \cdot v$$

$$= m \frac{1}{2} (\dot{v} \cdot v + v \cdot \dot{v})$$

$$\text{time invariant} \quad \frac{d}{dt} (v \cdot v)$$

$$= \frac{d}{dt} \left(\frac{1}{2} m v^2 \right) = \frac{dT}{dt}$$

conserv. forces T Kinetic energy theorem

dissipation viscous forces

- Equation of motion (d'Alembert principle) $\ddot{F}_e = -m\ddot{u}$
- By Lagrange's equation(s)

$$F = m a \quad \text{Lagrange function} \quad (q_k, \dot{q}_k) = T(\dot{q}_k) + V(q_k)$$

$$F - m a = 0 \quad \text{lagrangian coordinates} \quad U = L_f$$

$$V_f - L_f$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}_k} \right) - \frac{\partial L}{\partial q_k} = Q_k^* = \frac{\delta L(\delta q_k)}{\delta q_k}$$

$$\text{SDOF: } u(t) = \frac{1}{2} \dot{u}^2; \quad V_e = \frac{1}{2} K u^2$$

$$\frac{d}{dt} \left(\frac{1}{2} m \dot{u}^2 \right) + \frac{1}{2} K u^2 = Q_{f_k}^*$$

$$m \ddot{u} + Ku = F(t) - F_d = \frac{(F - F_d) \delta u}{\delta u}$$

$$Q_{d_k}^* = -\frac{\partial V_f}{\partial q_k} = \frac{\partial U_f}{\partial q_k}$$

$$Q_{d_k}^* = -\frac{\partial D}{\partial \dot{q}_k}$$

$$-F_d$$

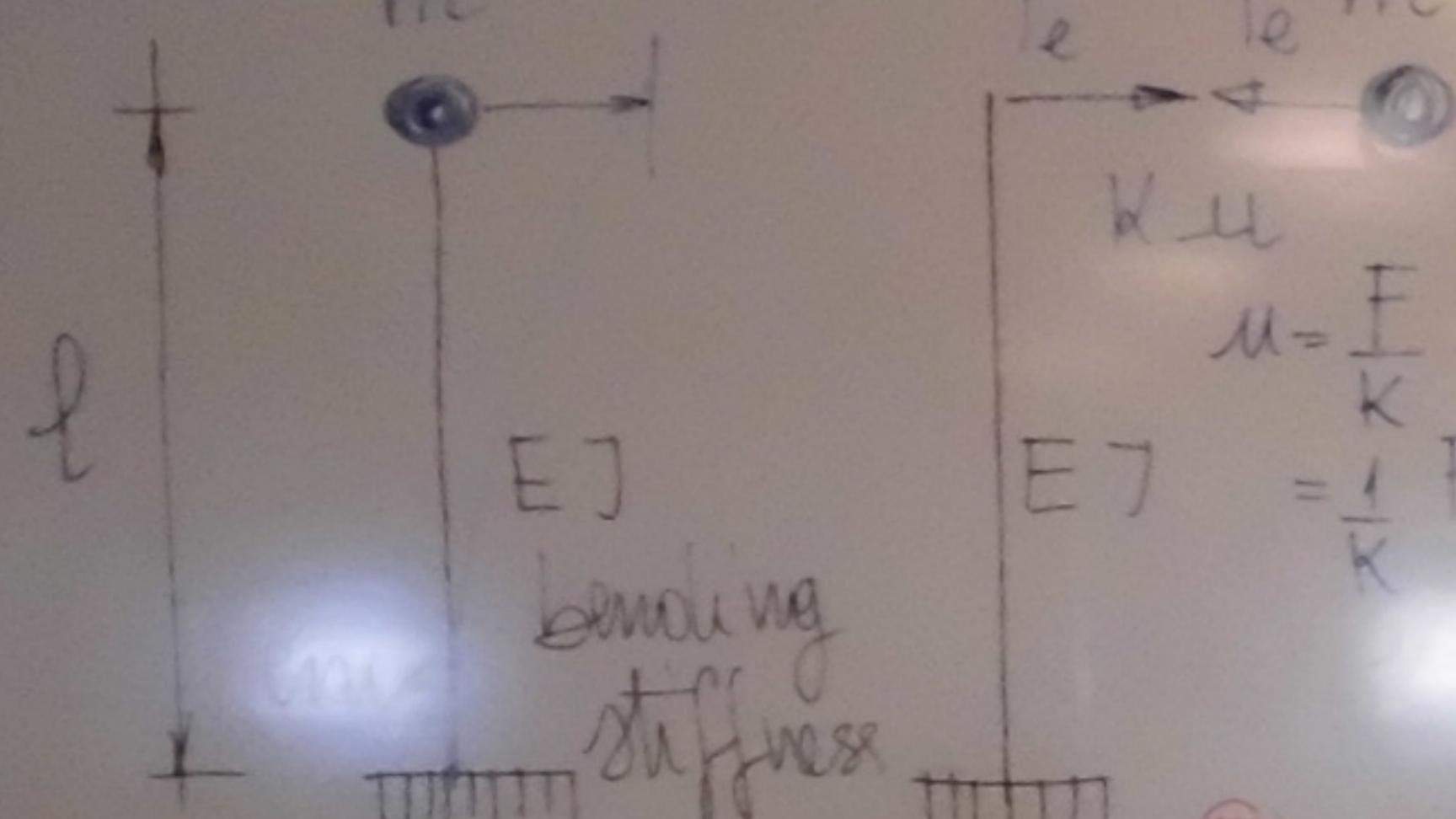
Initial conditions $t \geq t_0 = 0$

$$\begin{cases} u(t=t_0=0) = u_0 \\ \dot{u}(t=t_0=0) = \dot{u}_0 \end{cases}$$

$t \geq t_0 = 0$

Examples of structural systems

point mass m $u(t)$



EJ

bending

stiffness

For the evaluation of $K \rightarrow$

- Force method

\bullet PLV \rightarrow PVW

\bullet LE \rightarrow EL

$$\frac{F_e}{K} = \frac{1}{3} \frac{l}{EJ}$$

$$= \frac{1}{3} \frac{3EI}{l^3} \rightarrow \frac{1}{K}$$

(F_e) influence coefficients

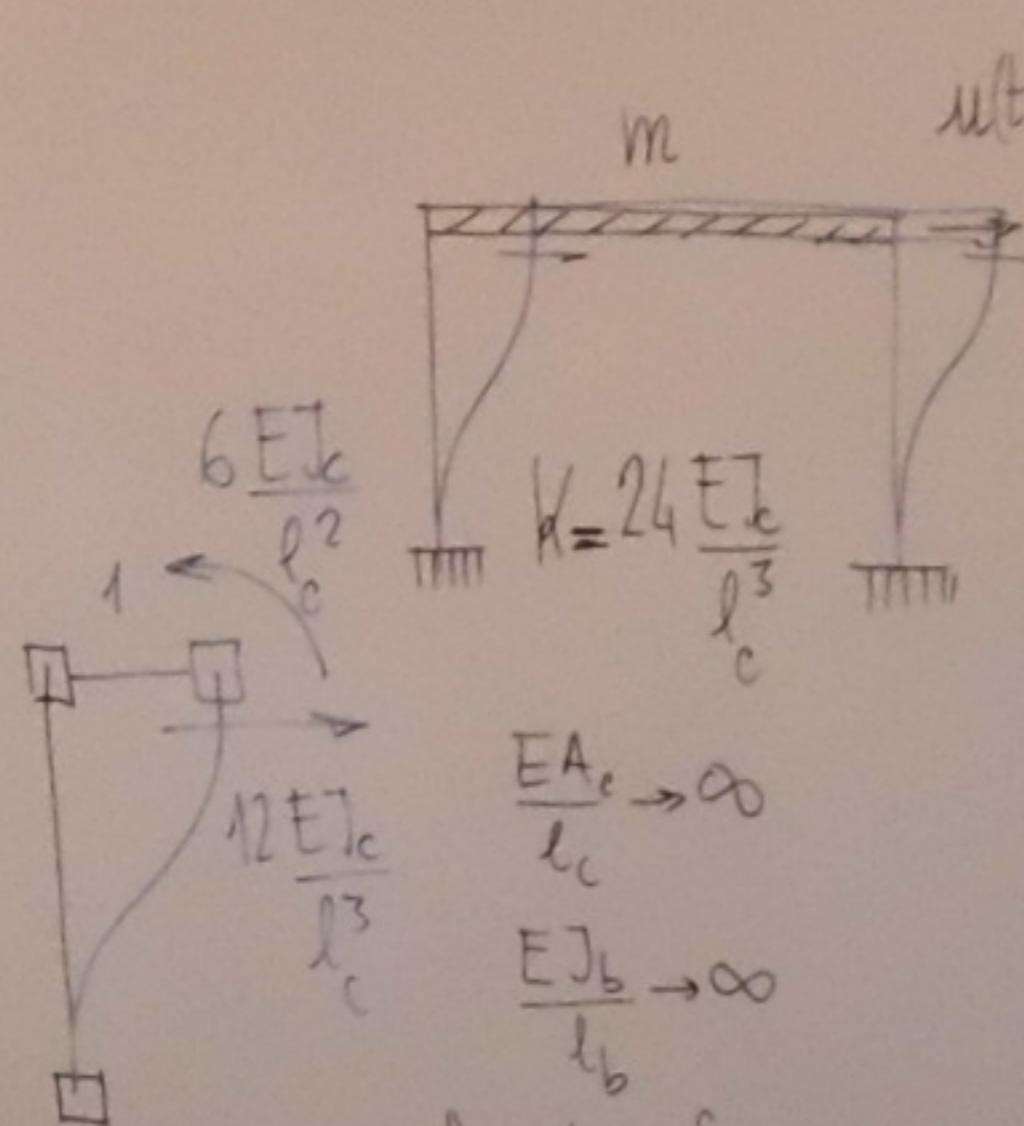
Displacement method

$$M = \frac{1}{3} \frac{F_e l^2}{EJ} \rightarrow K_{11} = \frac{3EI}{l^3}$$

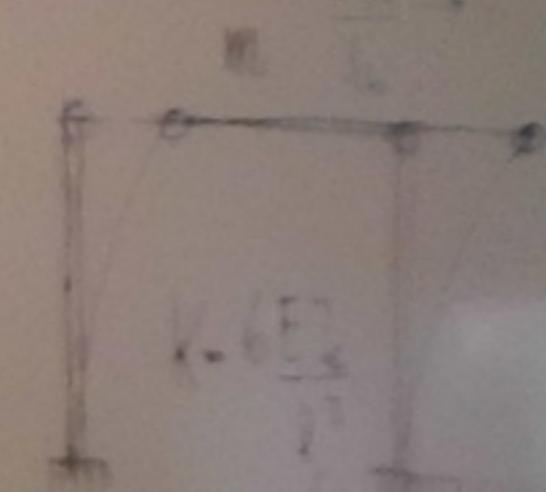
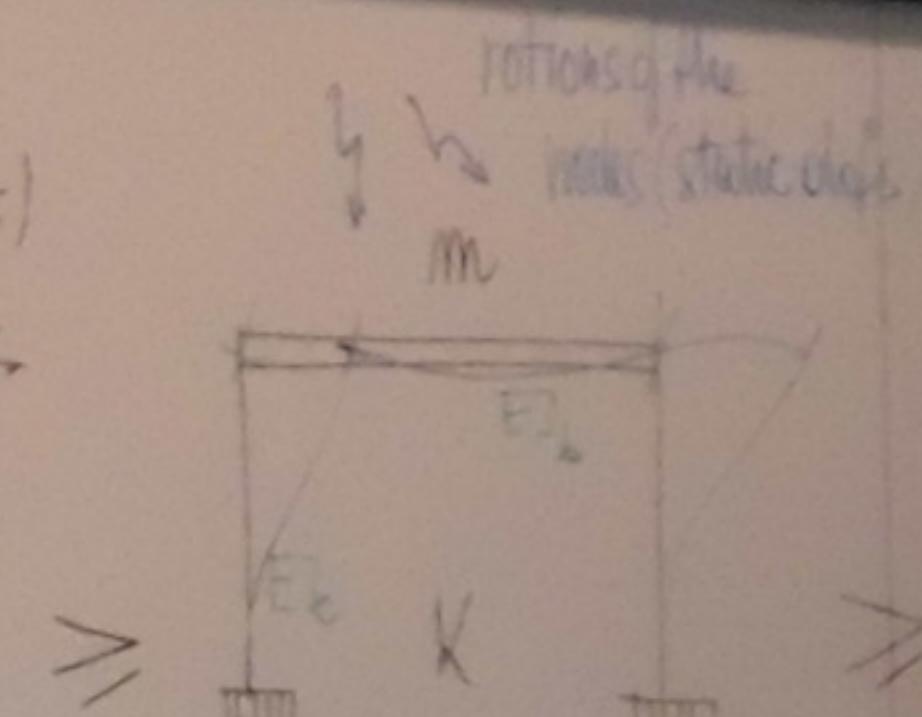
$$K = 3 \frac{EI_c}{l_c^3}$$

$$K = 48 \frac{EI_b}{l_b^3}$$

Frames



Shear-type frame



$$K = \sum \frac{EI}{l_i^3}$$

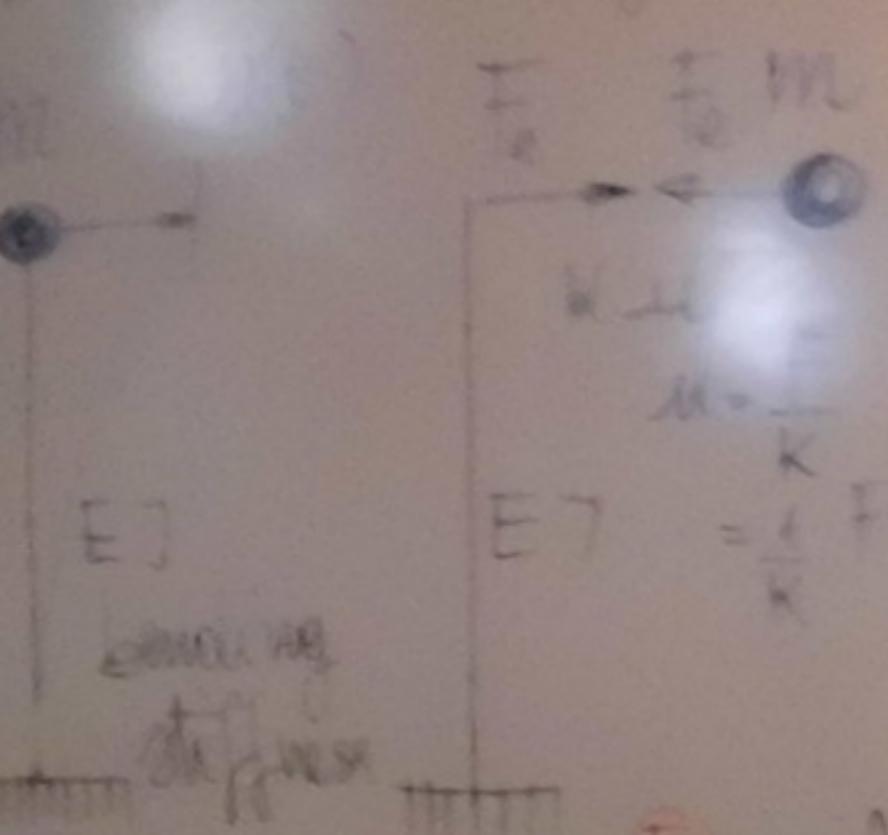
due to column stiffness

flexure

2 see Mathematics

Examples of structural systems

point masses



$$K = \frac{m}{\frac{EJ}{l^3}} = \frac{l^3}{K} F$$

beaming
stiffness

+ system stiffness

For the evaluation of $K \rightarrow$ influence coefficients

- Force method

$\rightarrow \frac{1}{F}$

$\rightarrow \frac{1}{F}$

$\rightarrow \frac{1}{F}$

$\rightarrow \frac{1}{F}$

Displacement method

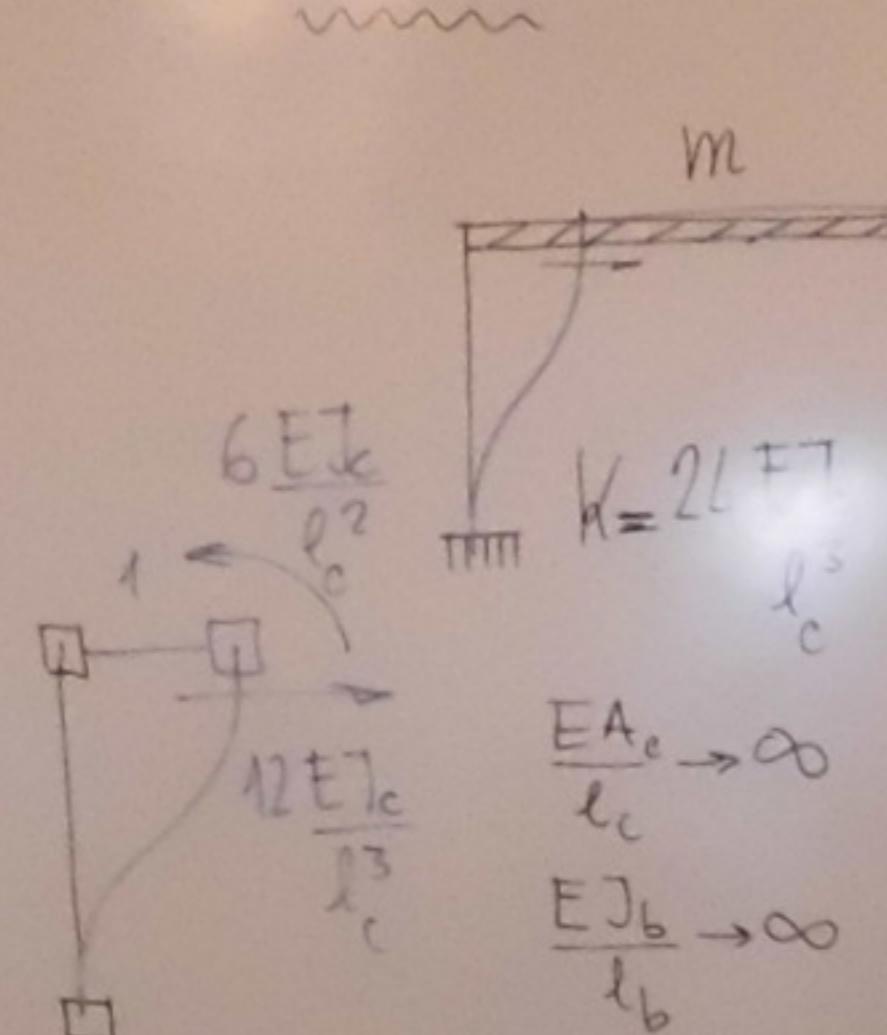
$M = \frac{F}{EJ} \cdot \frac{1}{l^3}$

$$K_{11} = \frac{5EJ}{l^3}$$

$$m \downarrow u(t)$$

$$K = 48 \frac{EJ}{l^3}$$

Frames



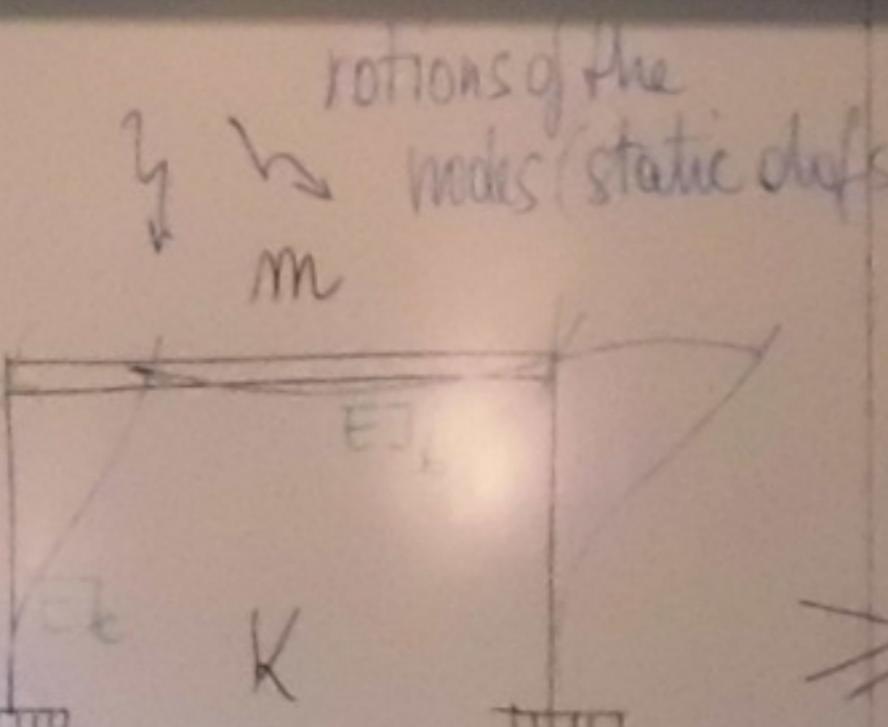
$$\frac{EA_c}{l_c} \rightarrow \infty$$

$$\frac{EJ_b}{l_b} \rightarrow \infty$$

shear-type frame

$$m \downarrow u(t)$$

$$K = 2L \frac{EJ}{l_c^3}$$



$$\rho = \sum \frac{EJ_b}{l_c}$$

$$\sum \frac{EJ_b}{l_c}$$

$$\frac{K}{6EJ_b}$$

$$\frac{120+1}{l_c^3}$$

$$\frac{6EJ_b}{l_c^3}$$

$$\frac{1}{30+1}$$

$$\frac{K}{6EJ_b}$$

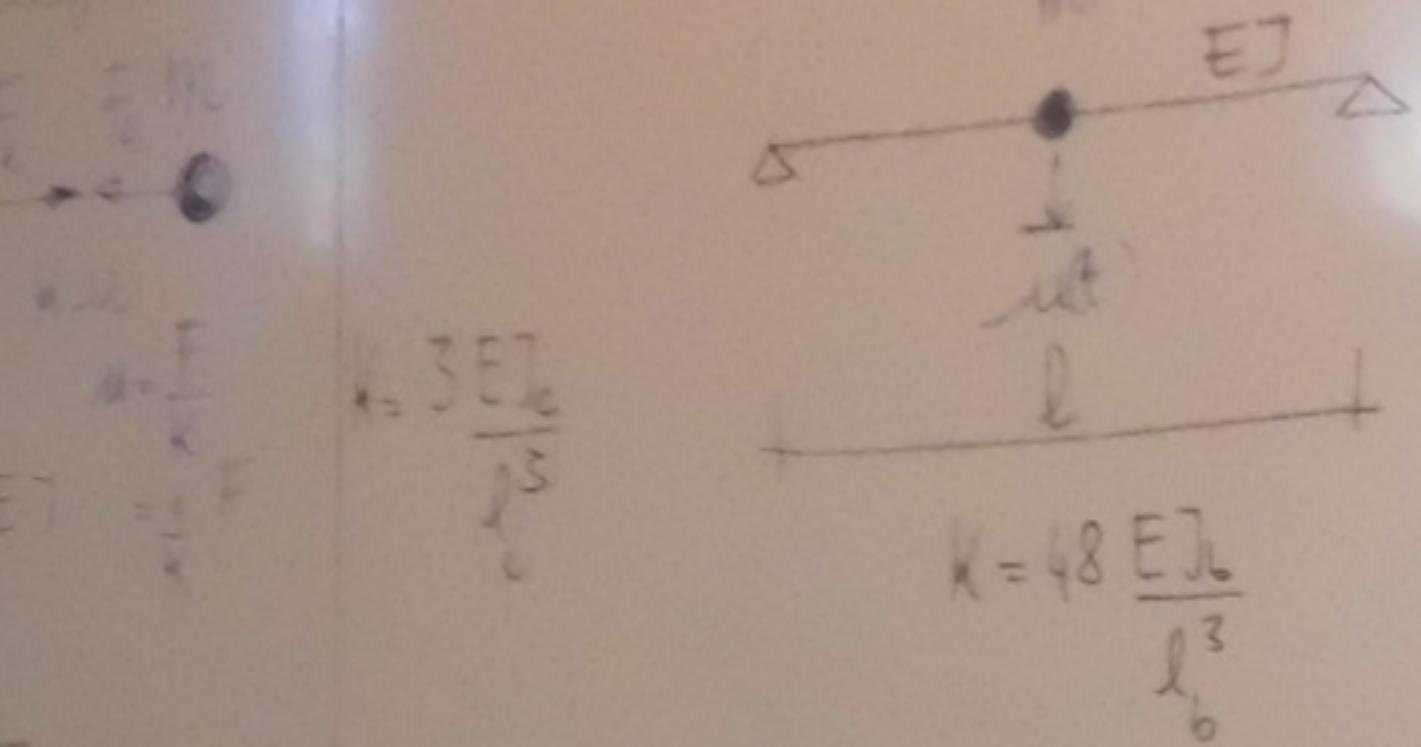
$$\frac{1}{30+1}$$

$$\frac{K}{6EJ_b}$$

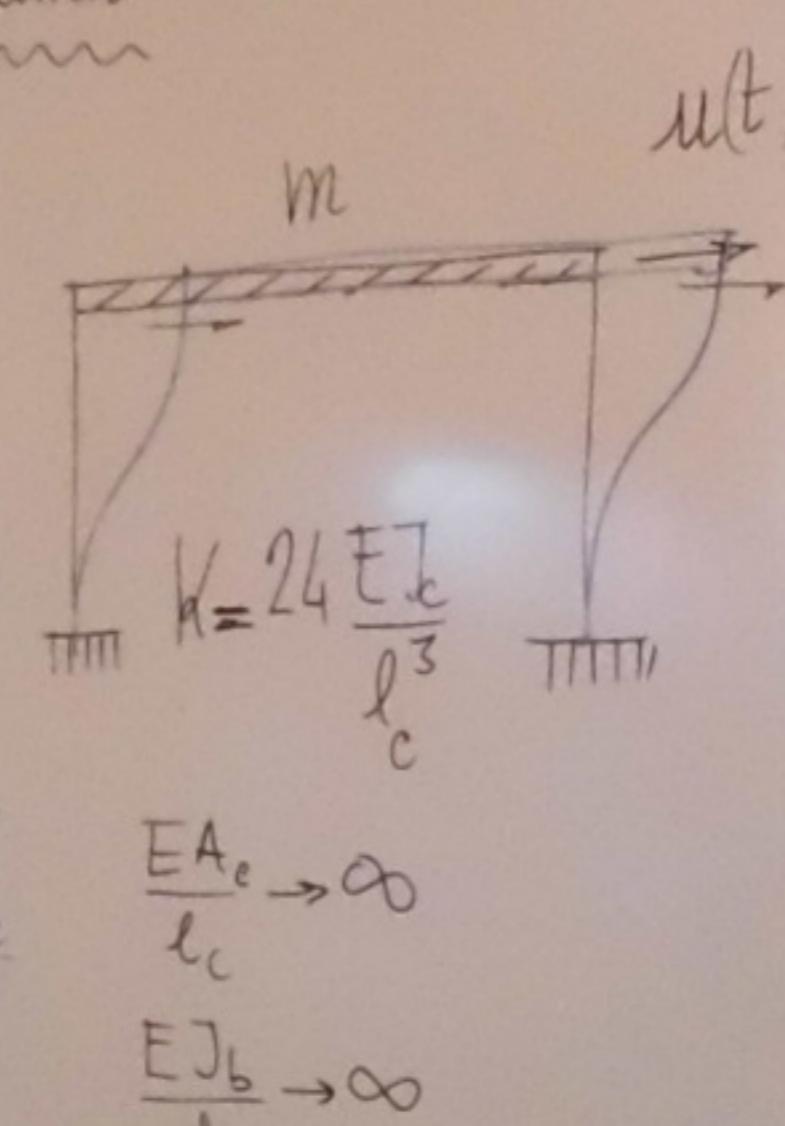
$$\frac{K}{6EJ$$

Simple structural systems

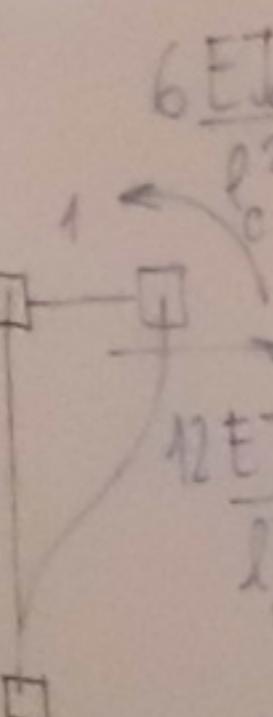
Frames



Frames



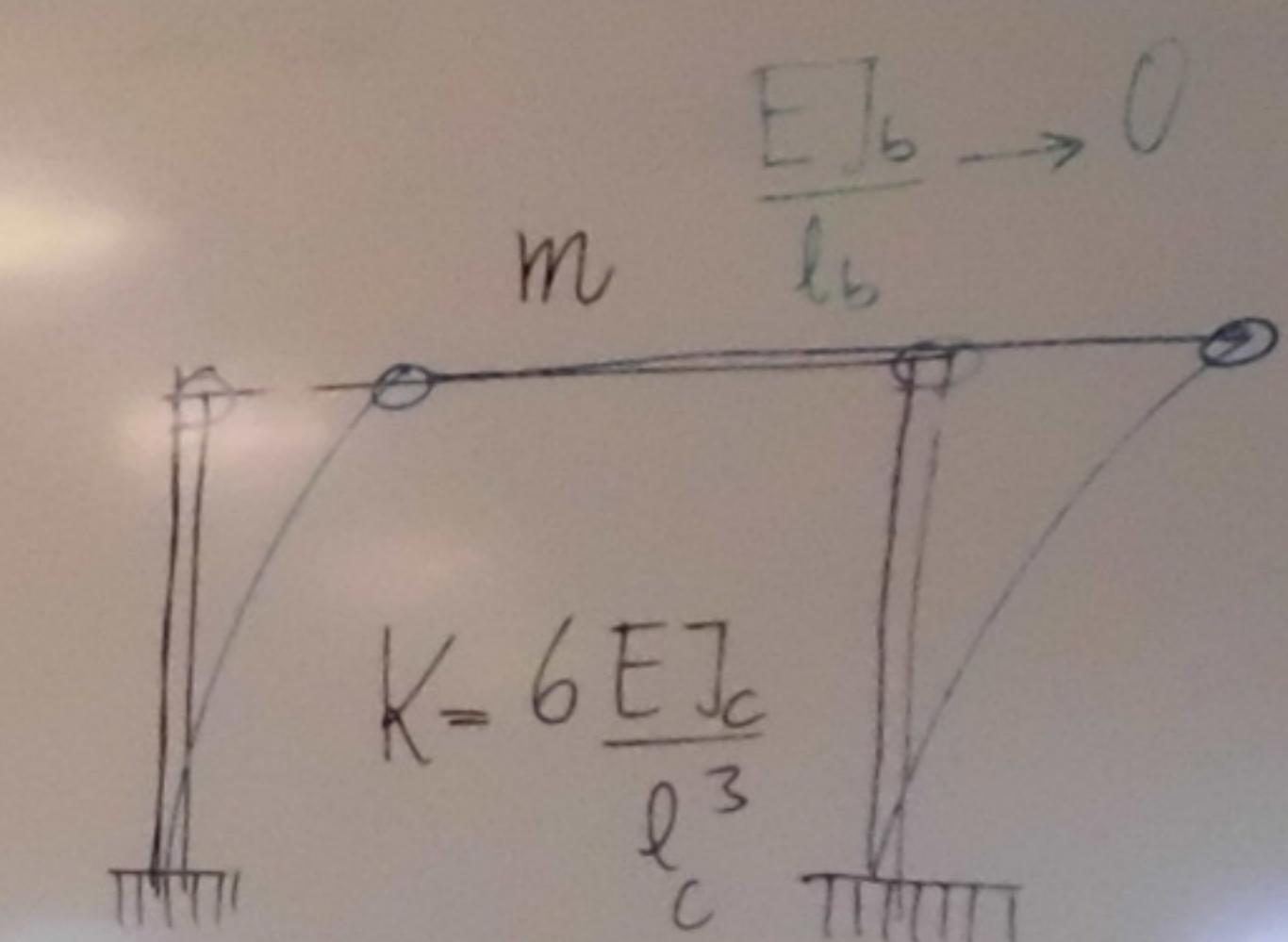
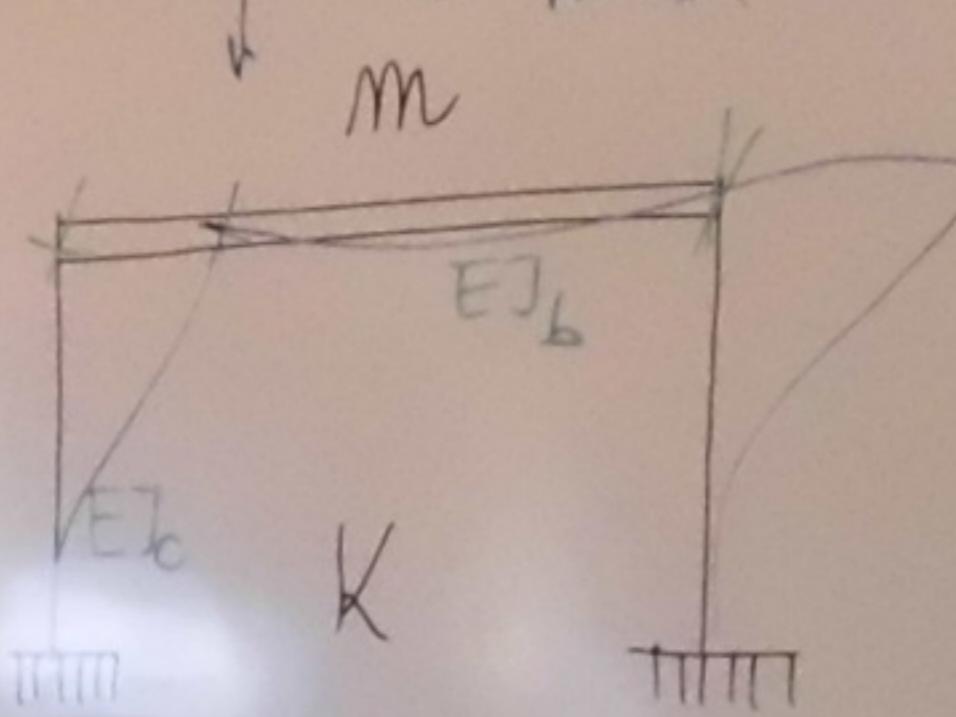
Shear-type frame



$$\frac{EA_c}{l_c} \rightarrow \infty$$

$$\frac{EJ_b}{l_b} \rightarrow \infty$$

rotions of the nodes (static dofs)



For the calculation of $K \rightarrow$

① Δ method

Frame method

different Deformation method

$M = \alpha \frac{\partial^2 U}{\partial x^2}$

$T_{uu} = \frac{M}{J}$

$EI \rightarrow EI$

$E \rightarrow EI$

$$\rho = \sum \frac{EJ_b}{l_b} \Rightarrow K = 24 \frac{EJ_c}{l_c^3} \frac{12\rho+1}{12\rho+4} \frac{K}{6EJ_c/l_c^3}$$

beam to column stiffness ratio (Chopra)

2 see Mathematica file