

## Critical deformed shapes - Example -

$$-\frac{1}{2} \delta V = \frac{1}{2} \delta q^T (K_E - P K_d) \delta q = 0 \Rightarrow \begin{cases} K_E \delta q_i^* = p_i^* K_d \delta q_i^* \\ \text{from 2-order theory } K \text{ loss of stability} \\ (\text{geometrically small displacements}) |V_i| \ll 1 \end{cases}$$

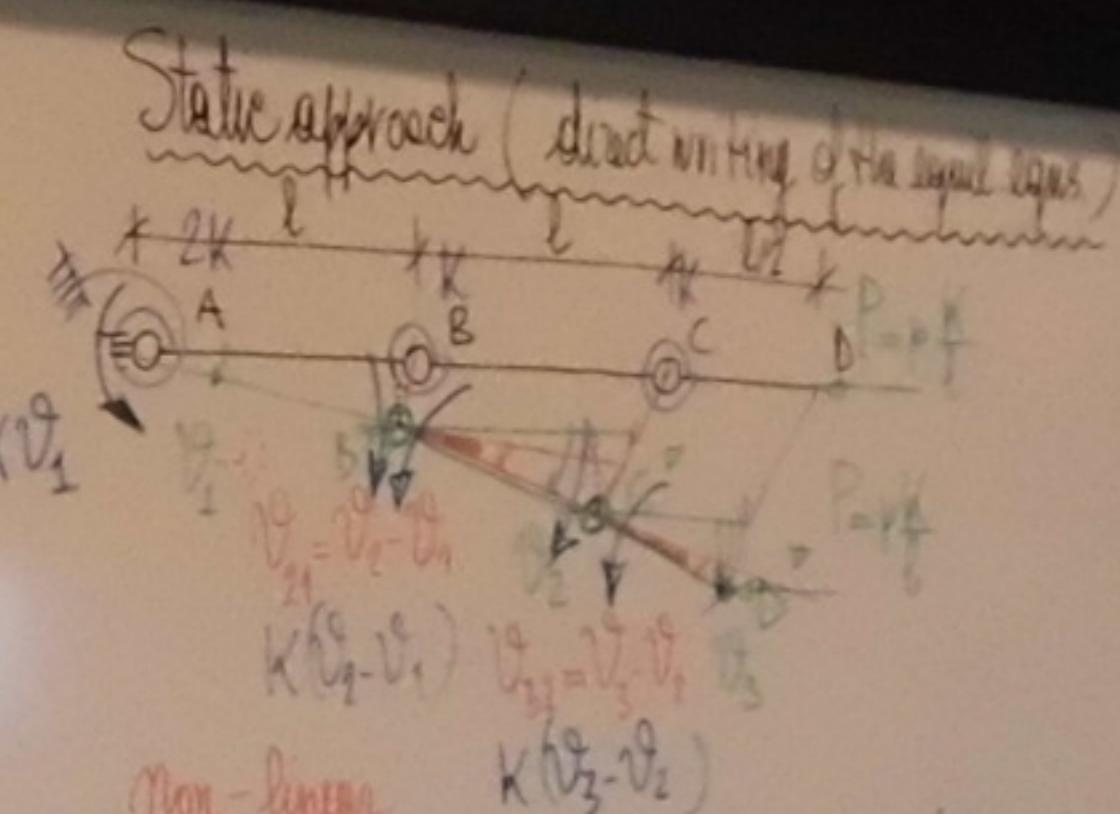
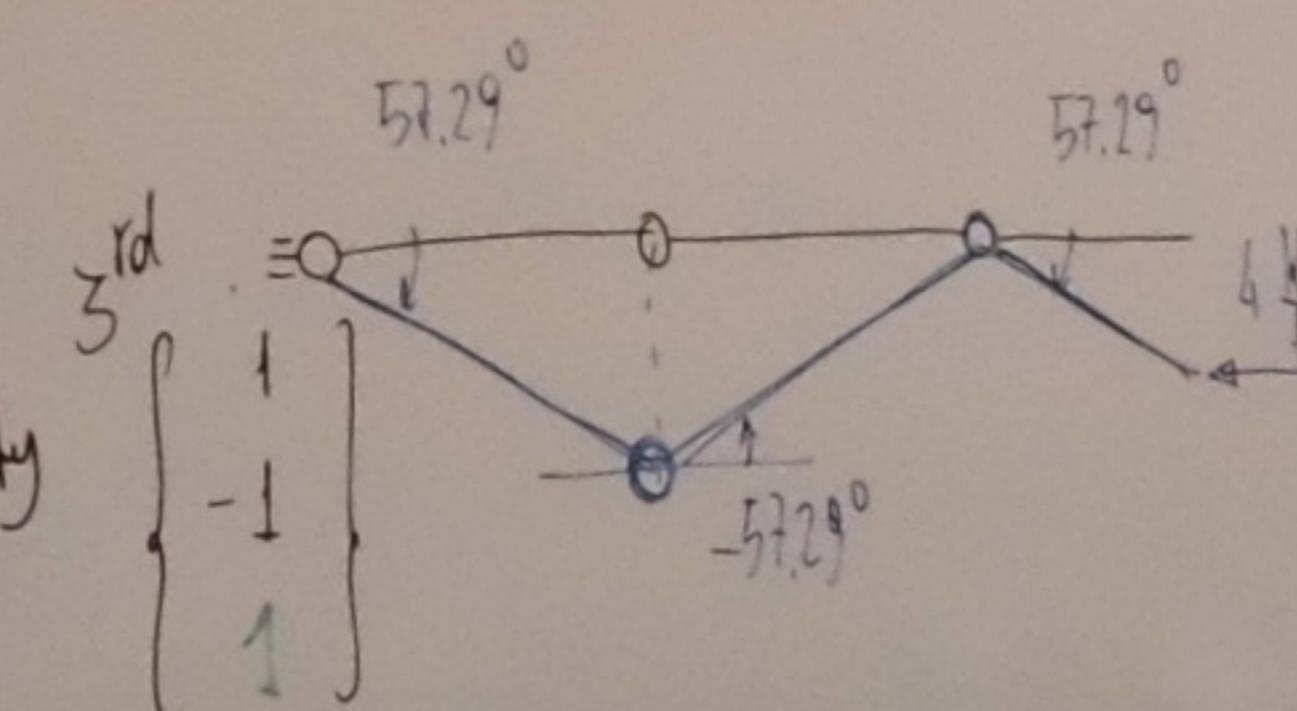
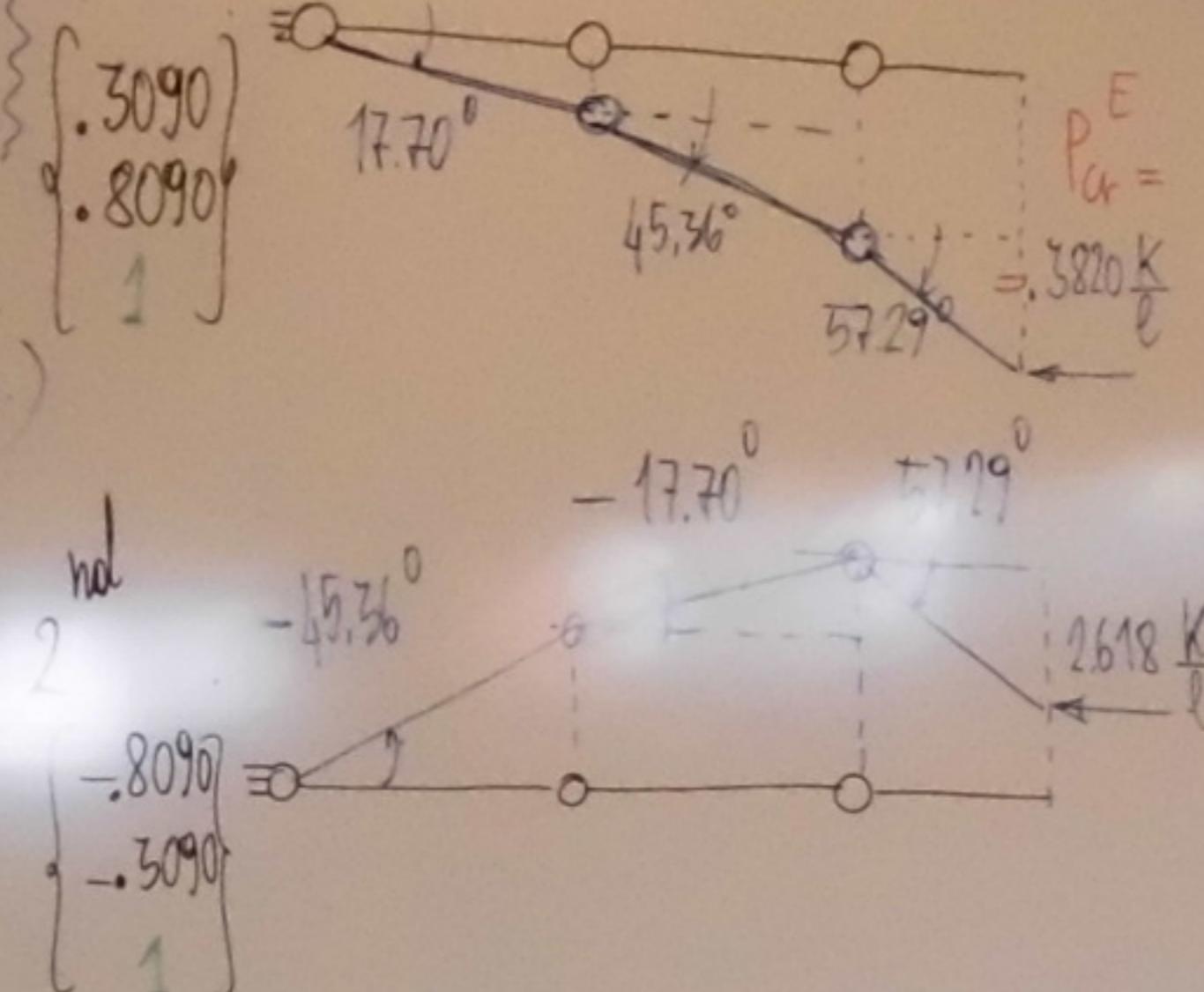
$$\begin{cases} (3-p)\vartheta_1 - \vartheta_2 = 0 \\ -\vartheta_1 + (2-p)\vartheta_2 - \vartheta_3 = 0 \\ -\vartheta_2 + (1-p)\vartheta_3 = 0 \end{cases} \Rightarrow \begin{cases} \delta q = q - q_0 = q \\ K \delta q = 0 \\ K \text{ singular} \end{cases}$$

+  $\det K = 0 \Rightarrow p_1^* = .3820, p_2^* = 2.618, p_4^* = 4$  critical load multipliers

- from 1st:  $\vartheta_1 = \frac{\vartheta_2}{3-p} = \frac{1}{2} \frac{2-p}{3-p} \vartheta_3$  Critical deformed  $\delta q_i^* = \begin{bmatrix} \frac{1}{2} \frac{2-p}{3-p} \\ \frac{2-p}{2} \\ 1 \end{bmatrix}$  "instability modes"

- from 3rd:  $\vartheta_2 = (1-p)\vartheta_3 = \vartheta_3 = \text{rad} = 57.29^\circ$

$$\begin{cases} \text{Dyn. } K \Phi = \omega_m M \Phi \\ \sum \delta q_i^* = 0 \end{cases} \quad \begin{bmatrix} .3090 \\ .8090 \\ 1 \end{bmatrix}$$



Non-linear Equilibrium eqns. (large displacements) in the deformed config.

$$\sum M_C^{CO} = 0 \Rightarrow K(\vartheta_3 - \vartheta_2) - p \frac{K}{2} l \sin \vartheta_3 = 0$$

$$\sum M_B^{BO} = 0 \Rightarrow K(\vartheta_1 - \vartheta_2) - p \frac{K}{2} l \sin \vartheta_2 = 0$$

$$\sum M_A^{ABCD} = 0 \Rightarrow 2K\vartheta_2 - p \frac{K}{2} l (\sin \vartheta_3 + \sin \vartheta_2) = 0 \quad \text{- Critical load}$$

Geometrically small displacements  $|V_i| \ll 1; \sin \vartheta_i \approx \vartheta_i$  to the first order, then  $K = \begin{bmatrix} 0 & 1 & 1 \\ 1 & 0 & 1 \\ 1 & 1 & 0 \end{bmatrix}$

$$2 \text{ linearized eqns. } K \delta q = (K_E - p K_d) \delta q = 0; \tilde{K} = \begin{bmatrix} 0 & 1 & 1 \\ 1 & 0 & 1 \\ 1 & 1 & 0 \end{bmatrix} \Rightarrow \begin{bmatrix} 1 & 1 & 1 \\ 1 & 0 & 1 \\ 1 & 1 & 0 \end{bmatrix} \delta q = 0$$

Critical deformed shapes - Example -  $K \dot{\phi}^2 = \omega_c M_B \phi$

$$\frac{1}{2} \delta V = \frac{1}{2} \delta q^T (K_E - p K_G) \delta q = 0 \Rightarrow K_E \delta q = p K_G \delta q$$

from 2-order theory  $K$  loss of stability  
geometrically small displacements

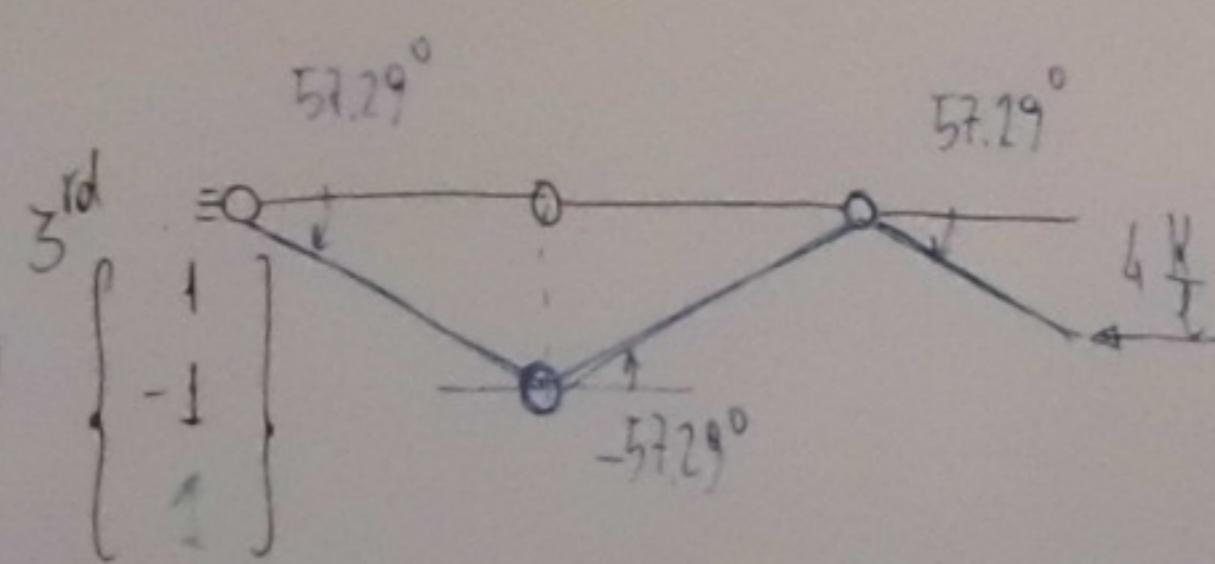
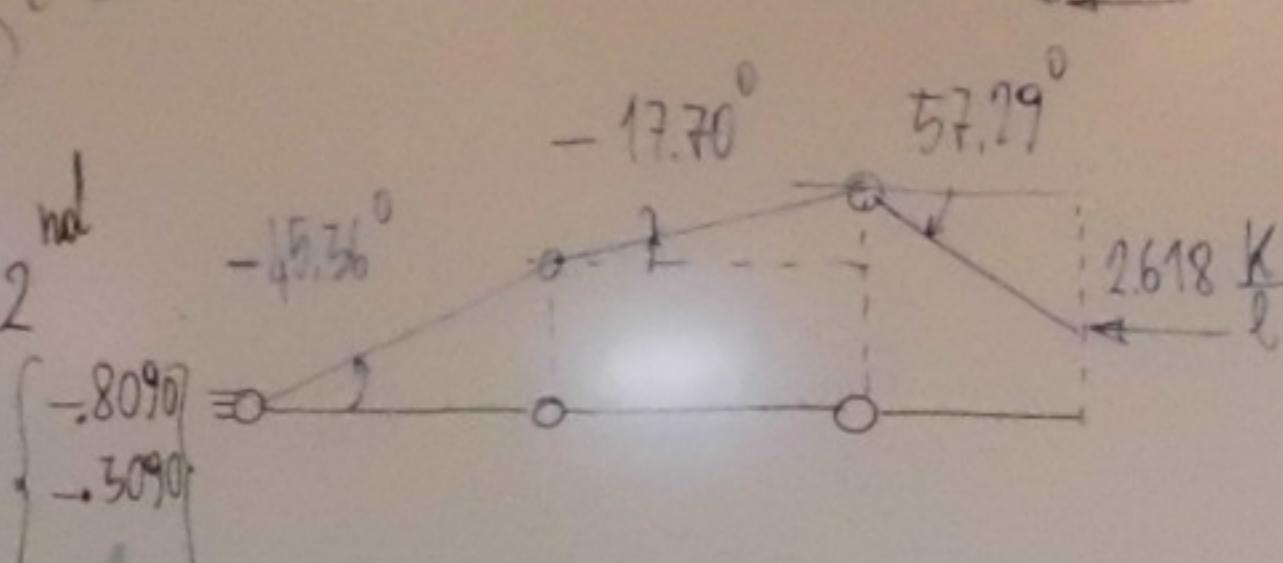
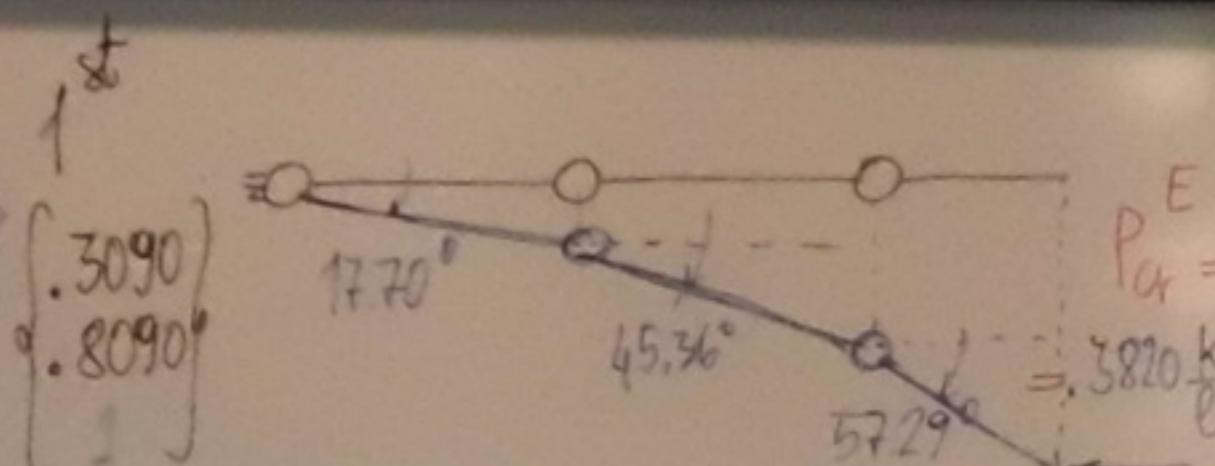
$$(3-p)\vartheta_1 - \vartheta_2 = 0$$

$$-\vartheta_1 + (2-p)\vartheta_2 - \vartheta_3 = 0 \Rightarrow K \delta q = 0$$

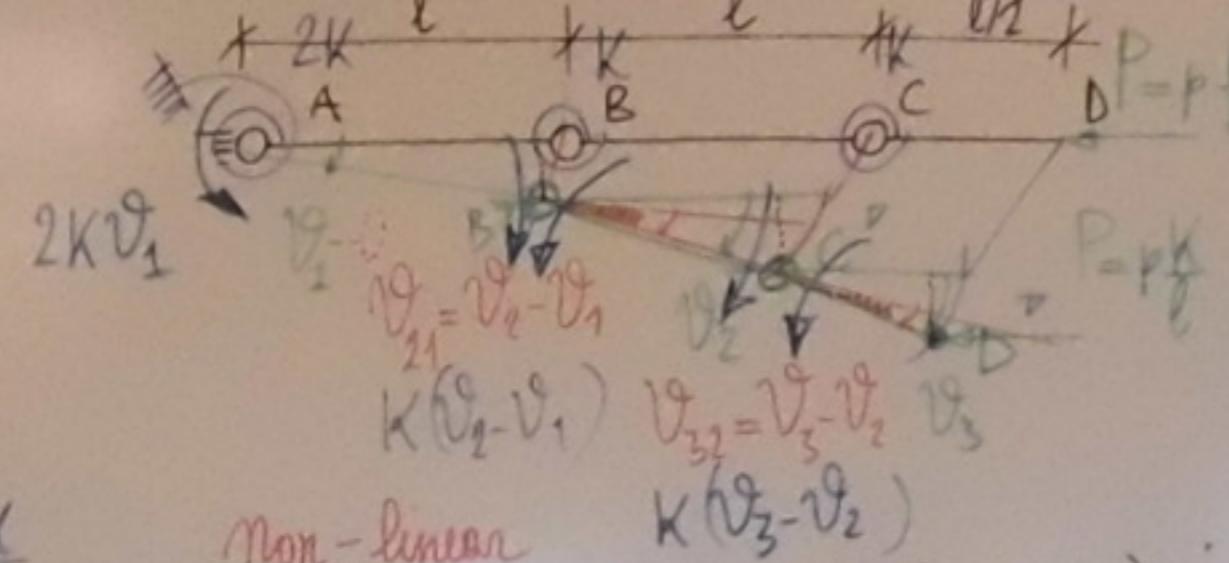
$$-\vartheta_2 + (1-p)\vartheta_3 = 0$$

$$\det K = 0 \Rightarrow p_1^+ = 3820, p_2^+ = 2618, p_4^+ = 4 \text{ critical load multipliers}$$

$$\begin{aligned} \vartheta_1 &= \frac{\vartheta_2}{3-p} = \frac{1-p}{3-p} \vartheta_3 \text{ Critical} \\ \vartheta_2 &= (1-p)\vartheta_3 = \vartheta_3 \text{ "modes"} \\ \vartheta_3 &= 57.29^\circ \end{aligned}$$



Static approach (direct writing of the equil. eqns.)



Non-linear Equilibrium eqns. (large displacements) in the deformed config. ( $\vartheta_1 = \vartheta_2 = \vartheta_3$  equal config.)

$$\sum M_C^{CO} = 0 \Rightarrow K(\vartheta_3 - \vartheta_2) - p \frac{K}{l} \frac{l}{2} \sin \vartheta_3 = 0$$

$$\sum M_B^{BO} = 0 \Rightarrow K(\vartheta_2 - \vartheta_1) - p \frac{K}{l} \left( \frac{l}{2} \sin \vartheta_3 + p \sin \vartheta_2 \right) = 0$$

$$\sum M_A^{ACCO} = 0 \Rightarrow 2K\vartheta_1 - p \frac{K}{l} \left( \frac{l}{2} \sin \vartheta_3 + l \sin \vartheta_1 + p \sin \vartheta_2 \right) = 0$$

Geometrically-small displacements  $|\vartheta_i| \ll 1$ ;  $\sin \vartheta_i \approx \vartheta_i$  to the first order  
2 linearized eqn. eqns.

$$K \delta q = (K_E - p K_G) \delta q = 0$$

Linearized  
equil.  
eqns:

$$-\vartheta_2 + (1-p)\vartheta_3 = 0$$

$$-\vartheta_1 + (1-p)\vartheta_2 - \frac{p}{2}\vartheta_3 = 0$$

$$(1-p)\vartheta_1 - p\vartheta_2 - \frac{p}{2}\vartheta_3 = 0$$

a combination of the eqns. coming from the  
non-linear approach - 2-order theory

$$\tilde{K}_E = \begin{bmatrix} 0 & -1 & 1 \\ -1 & 1 & 0 \\ 2 & 0 & 0 \end{bmatrix}; \tilde{K}_G = \begin{bmatrix} 0 & 0 & \frac{1}{2} \\ 0 & 1 & \frac{1}{2} \\ 1 & 1 & \frac{1}{2} \end{bmatrix}$$

in general, non-symmetric matrices

Critical loads:

$$\det \tilde{K} = \left[ \frac{p}{2} + p_2(2-p) \right] + (1-p_2)(p - (1-p)(2-p)) = 0$$

$$\Rightarrow \frac{1}{2} p - \frac{7}{2} p^2 + \frac{13}{2} p - 2 = 0 \Rightarrow p_c = \dots$$

Geometric shapes - Example

$$K = \frac{E}{L} I \cdot \frac{1}{3}$$

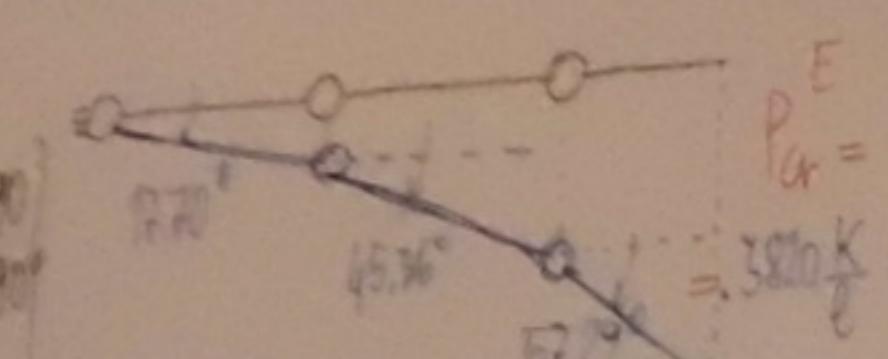
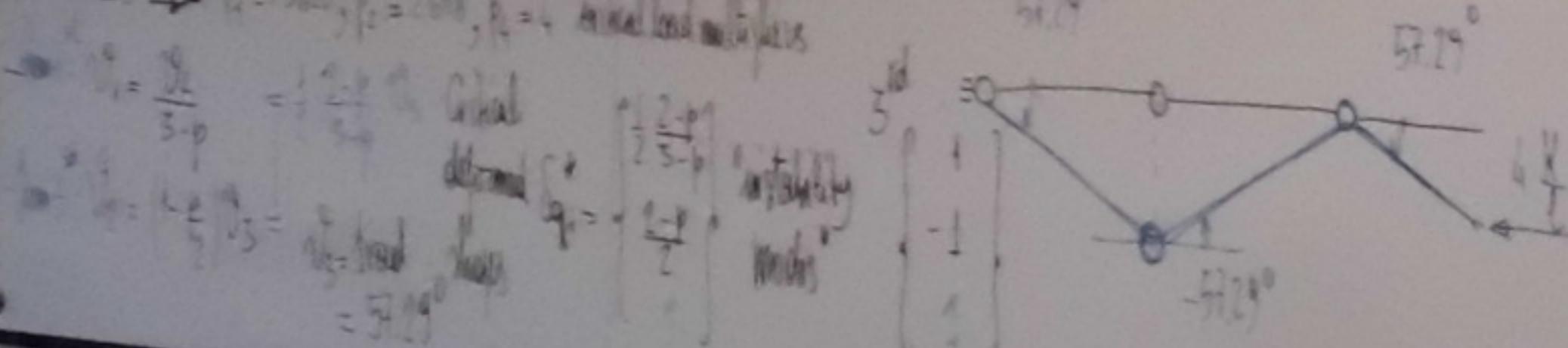
$$3 \cdot q - \theta_1 = 0 \quad \theta_1 = 3q$$

$$- \theta_1 + p\theta_2 - \theta_3 = 0 \Rightarrow K\delta q = 0$$

$$- \theta_2 + p\theta_3 = 0$$

$$\det K = 0 \Rightarrow \theta_1 = 3q, \theta_2 = 1.6q, \theta_3 = 4.33q \text{ initial load multipliers}$$

$$\begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix} \begin{bmatrix} 3q \\ 1.6q \\ 4.33q \end{bmatrix}$$



Static approach (direct writing of the equil. eqns.)

$$\begin{aligned} &+2K \quad +K \quad +K \quad P=pK \\ &A \quad B \quad C \quad D \\ &2K\vartheta_1 \quad \quad \quad \quad P=pK \\ &\theta_1 \quad \theta_2 \quad \theta_3 \end{aligned}$$

$$K(\theta_2 - \theta_1) \quad \theta_3 = \theta_1 - \theta_2$$

$$K(\theta_3 - \theta_2)$$

- Equilibrium eqns. (large displacements) in the deformed config. ( $\vartheta_1 = \vartheta_2 = \vartheta_3$  equal config.)

$$\sum M_{c=0}^{co} = 0 \Rightarrow K(\theta_3 - \theta_2) - p \frac{K}{L} \frac{L}{2} \sin \theta_3 = 0$$

$$\sum M_{B=0}^{so} = 0 \Rightarrow K(\theta_2 - \theta_1) - p \frac{K}{L} \left( \frac{L}{2} \sin \theta_3 + \frac{L}{2} \sin \theta_2 \right) = 0$$

$$\sum M_{A=A}^{ABCD} = 0 \Rightarrow 2K\vartheta_1 - p \frac{K}{L} \left( \frac{L}{2} \sin \theta_3 + \frac{L}{2} \sin \theta_2 + \frac{L}{2} \sin \theta_1 \right) = 0$$

- Geometrically-small displacements  $|\theta_i| \ll 1$ ;  $\sin \theta_i \approx \theta_i$  to the first order

→ linearized  
equil. eqns.

$$\tilde{K}\delta q = (\tilde{K}_E - p\tilde{K}_G)\delta q = 0; \quad \tilde{K} = \begin{bmatrix} 0 & -1 & 1-p \\ -1 & 1-p & -p \\ 2-p & -p & -\frac{p}{2} \end{bmatrix}$$

Linearized  
equil.  
eqns.:

$$-\vartheta_2 + (1-p)\vartheta_3 = 0 \quad \rightarrow \text{stability}$$

$$-\vartheta_1 + (1-p)\vartheta_2 - p\vartheta_3 = 0 \quad \rightarrow \text{stability}$$

$$(2-p)\vartheta_1 - p\vartheta_2 - p\vartheta_3 = 0 \quad \rightarrow \text{stability}$$

Linear combinations of the eqns. coming from the energy approach (2<sup>nd</sup> order theory)

$$\tilde{K}_E = \begin{bmatrix} 0 & -1 & 1 \\ -1 & 1 & 0 \\ 2 & 0 & 0 \end{bmatrix}; \quad \tilde{K}_G = \begin{bmatrix} 0 & 0 & \frac{1}{2} \\ 0 & 1 & \frac{1}{2} \\ 1 & 1 & \frac{1}{2} \end{bmatrix}$$

in general, non-symmetric matrices.

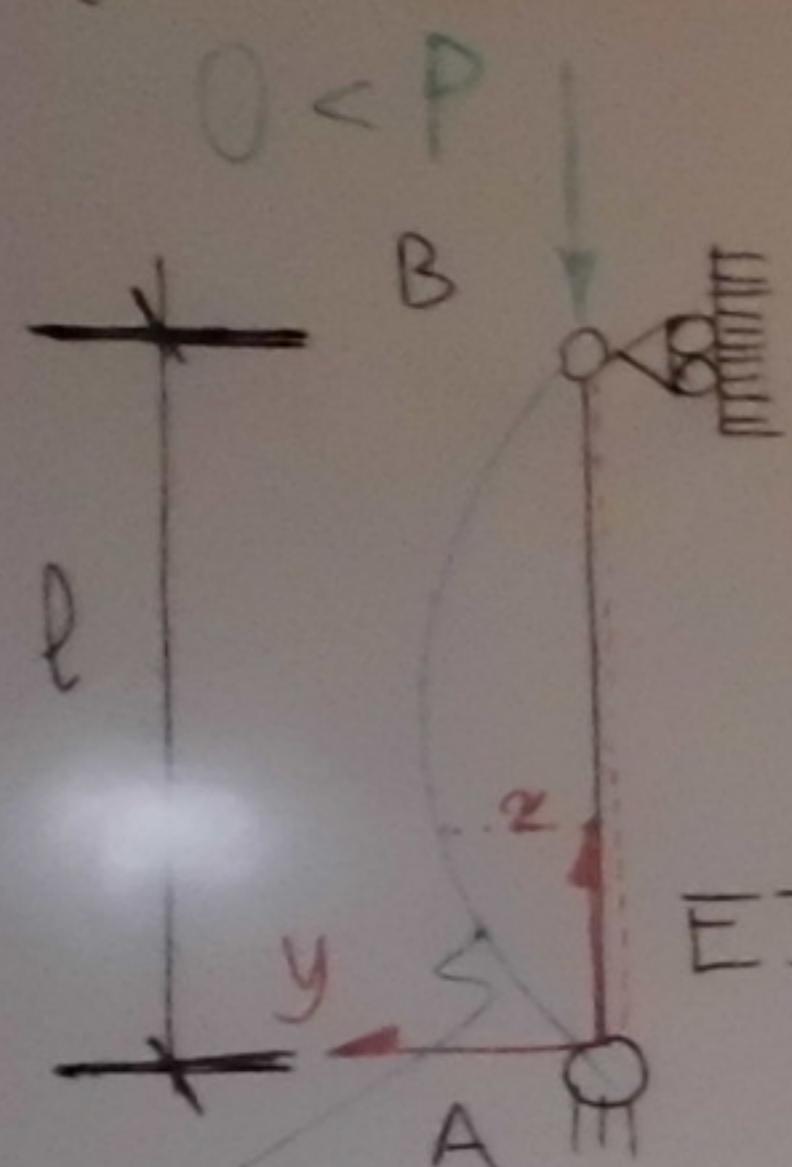
Critical loads:

$$\det \tilde{K} = 1 \left[ \frac{p}{2} + p(2-p) \right] + (1-p)(p-(1-p)(2-p)) = 0$$

$$\rightarrow \frac{1}{2}p^3 - \frac{7}{2}p^2 + \frac{13}{2}p - 2 = 0 \Rightarrow p_i^* \sim \star$$

some are by hand

Euler beam ( $\sim 1750$ )  $\rightarrow$  Continuous systems (static approach)  
 (Elastic bending instability of complex beams)



B-E-N theory (bending deflection of beams)

$$y(x) \approx X = \frac{M}{EI}$$

(small rotations)

$$EI y''' = M(x) = -Py(x)$$

$$EI = \text{const} > 0$$

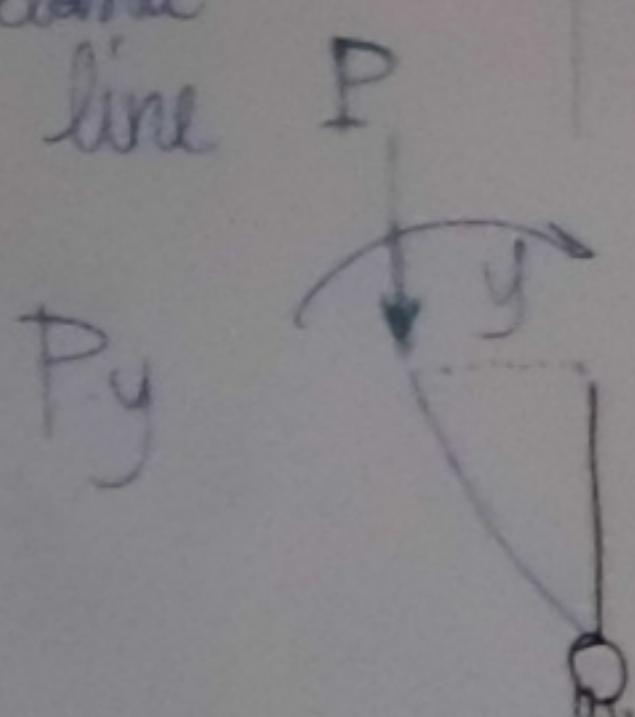
$$EI y'' + Py(x) = 0$$

$$y''(x) + \alpha^2 y(x) = 0 \quad (\text{harmonic motion eqn.})$$

boundary conditions

$$\begin{cases} y(0) = 0 \\ y(l) = 0 \end{cases}$$

elastic line



$$\alpha^2 = \frac{P}{EI} > 0$$

$$(dl)^2 = \frac{P l^2}{EI}; P = (\alpha l)^2 \frac{EI}{l^2}$$

By setting the b.c.s:

$$\begin{cases} y(0) = B = 0 \\ y(l) = As\sin\alpha l + Bs\cos\alpha l = 0 \Rightarrow As\sin\alpha l = 0 \end{cases}$$

$$\begin{bmatrix} 0 & 1 & [A] \\ \text{small const.} & [B] \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

$$D \cdot X = 0 \Rightarrow \det D = -\sin\alpha l = 0$$

solution: ( $A, B$  arbitrary constants)  $\infty$  bifurcated solutions

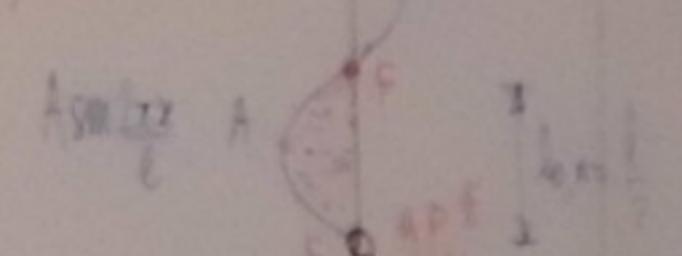
$$y(0) = B = 0$$

$\lambda = 0, \text{ small}$

$$\alpha l = n\pi, n=1, 2, \dots$$

$$(\alpha l)^2 = n^2 \pi^2$$

$$\text{Flex point: } y = 0 \Rightarrow \text{Max.}$$



$$P_{cr,n} = \frac{n^2 \pi^2 EI}{l^2}$$

$$y(x) = A \sin \frac{n\pi x}{l} = \frac{\pi^2 EI}{l^2} \frac{x^2}{n^2}$$

$$y(x) = A \cos \frac{n\pi x}{l} - B \sin \frac{n\pi x}{l}$$

$$y(x) = -\alpha^2 (A \sin \frac{n\pi x}{l} + B \cos \frac{n\pi x}{l}) = -\alpha^2 y$$

$$P_{cr,n} = P_{cr} = \frac{\pi^2 EI}{l^2}$$

The critical loads are real in the free vibration length, the distance between the supports is called the natural frequency.

The ratio of the deflection to the loading force is the stiffness coefficient. It is independent of the mass of the system.

Euler beam ( $\sqrt{f_0}$ )  $\rightarrow$  Continuous systems (static approach)  
 (Elastic bending instability of compressed beams)

B-E-N theory (bending deflection of beams)

$$\frac{\partial^2 y}{\partial x^2} = \frac{M(x)}{EI}$$

$$EI y''(x) = M(x) = -P y(x)$$

boundary conditions:

$$y(0) = 0, \quad y(l) = 0$$

$$y''(0) = 0, \quad y''(l) = 0$$

$$EI y''''(x) + P^2 y(x) = 0$$

$$y''''(x) + \frac{P^2}{EI} y(x) = 0$$

$$y''''(0) = 0, \quad y''''(l) = 0$$

By setting the b.c.s:

$$\begin{cases} y(0) = B = 0 \\ y(l) = A \sin \alpha l + B \cos \alpha l = 0 \Rightarrow A \sin \alpha l = 0 \end{cases}$$

system of b.c.s.

$$\begin{bmatrix} 0 & 1 \\ \sin \alpha l & \cos \alpha l \end{bmatrix} \begin{bmatrix} A \\ B \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

$$I \cdot x = 0 \rightarrow \det D = -\sin \alpha l = 0$$

solution: (A, B arbitrary constants)

$$y(x) = A \sin \alpha x + B \cos \alpha x$$

$$y'(x) = \alpha A \cos \alpha x - \alpha B \sin \alpha x$$

$$y''(x) = -\alpha^2 (A \sin \alpha x + B \cos \alpha x) = -\alpha^2 y(x)$$

bifurcated solutions

$$y=0 \text{ (ref conf.) equal. load } + P$$

$A=0$ ,  $\alpha = \text{arbitrary}$

$$\alpha l = n \pi, n=1, 2, 3, \dots$$

$$(x/l)^2 = n^2$$

$$\text{flex. ts} \quad y'' = 0 \Rightarrow M=0$$

$$P_{cr,n} = \frac{n^2 \pi^2 E J}{l^2}$$

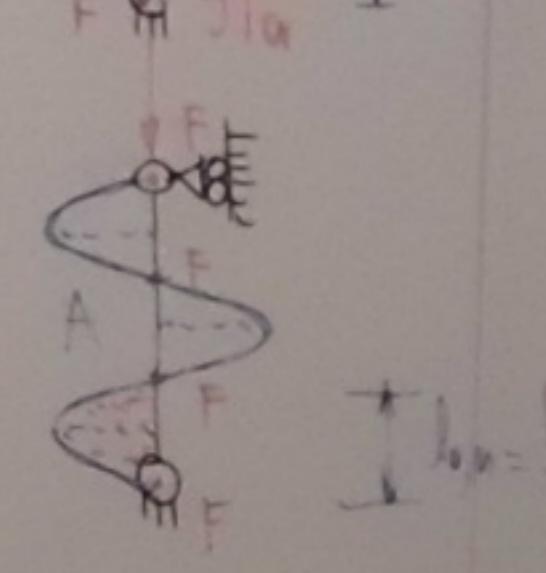
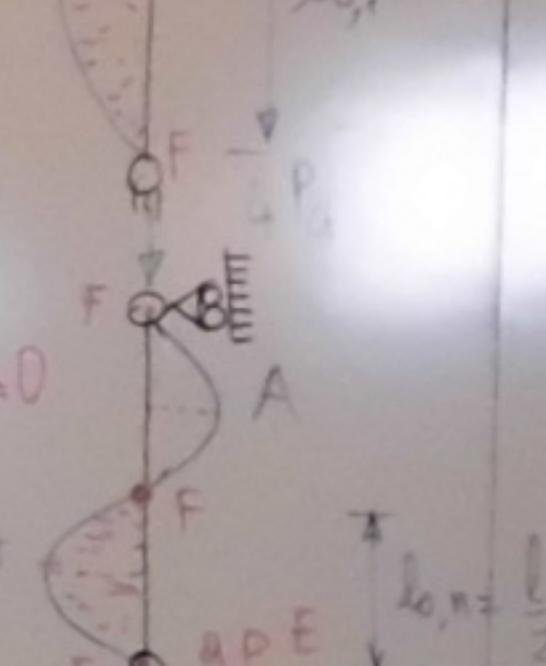
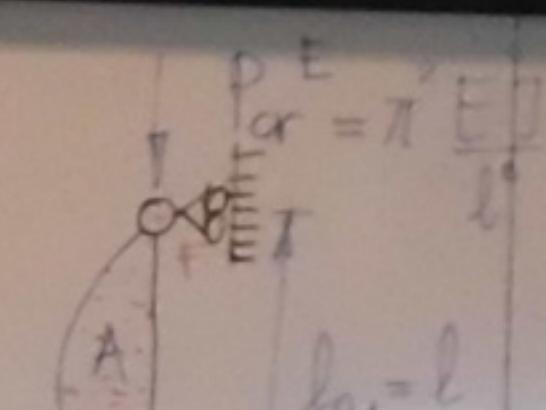
$$y(x) = \frac{\pi}{l} n \sin \frac{n \pi x}{l} = \frac{\pi^2 E J}{l^2} \frac{x}{n}$$

appear @

$$P = P_{cr,n}$$

$$P_{cr} = P_{cr,n} = \frac{\pi^2 E J}{l^2} l_{cr,n} = \frac{l}{n}$$

free inflection length



- The critical loads are ruled by the free inflection length (the distance between two successive flex points of the critical deformed shape)

$$\left\{ P_{cr,n} = \frac{\pi^2 E J}{l_{cr,n}^2} \right\}$$

- The critical load is directly proportional to the bending static stiffness  $EJ$

- It is inversely proportional to the square of the inflection length  $l_{cr,n} = \frac{l}{n}$

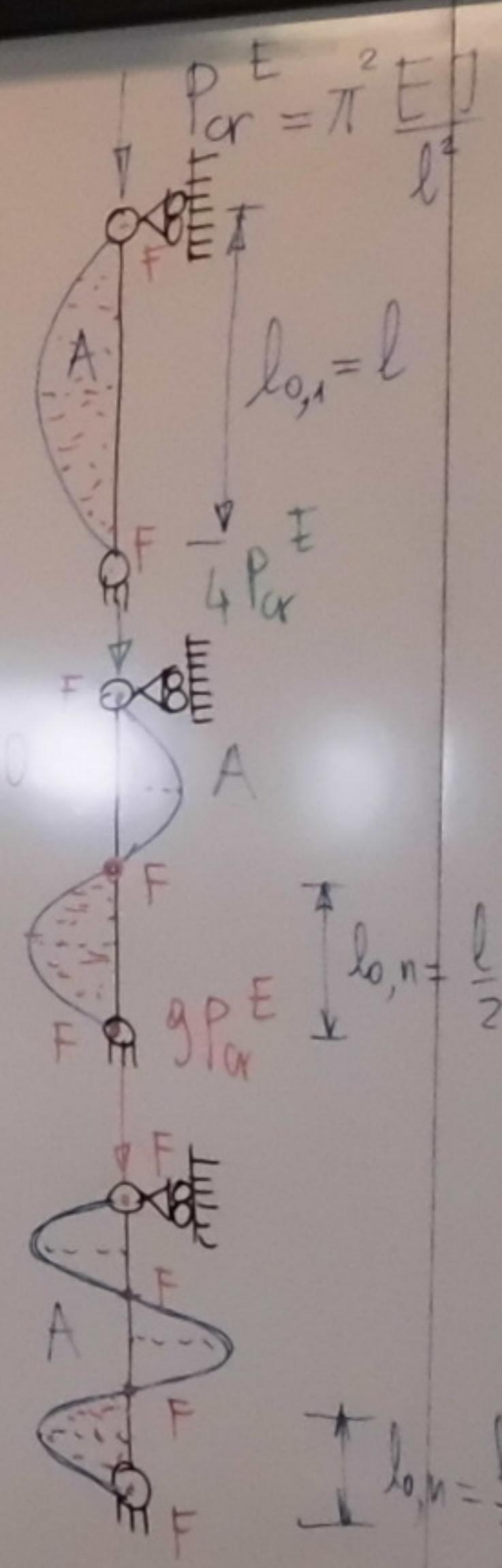
- The critical load(s) are ruled by the free inflection length (the distance between two consecutive flex points of the critical deformed shape)

as

$$P_{cr,n} = \frac{\pi^2 E}{l_{0,n}^2}$$

- The critical load is directly proportional to the bending elastic stiffness  $EJ$

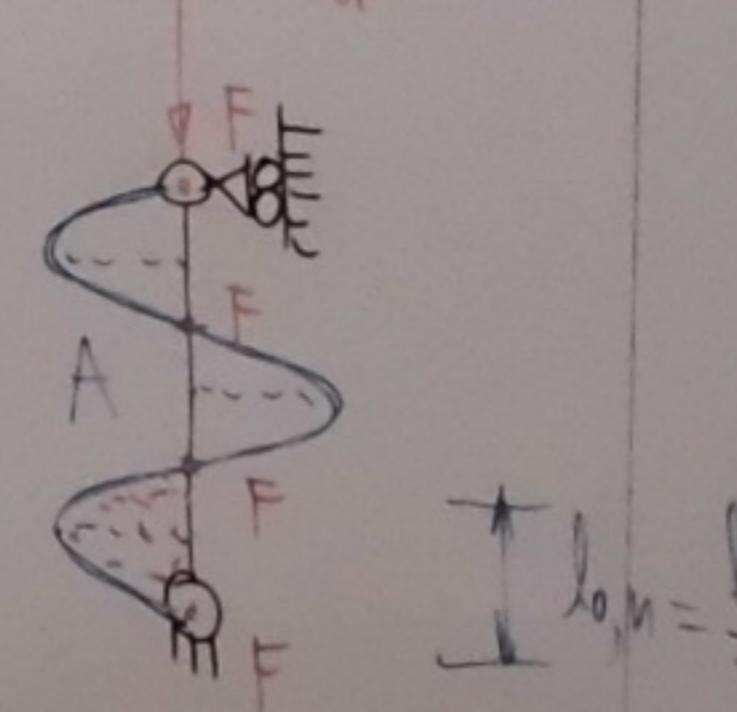
- It is indirectly proportional to the square of free inflection length  $l_{0,n} = \frac{l}{n}$



$$\text{Asm } \frac{2\pi x}{l}$$

Flex points  
 $y''=0 \Leftrightarrow x=n\frac{l}{2}$

$$\text{Asm } \frac{2\pi x}{l} \quad A \quad l_{0,n} = \frac{l}{2}$$



$$\text{Asm } \frac{3\pi x}{l} \quad l_{0,n} = \frac{l}{3}$$

$$P_{cr,n} = n^2 \frac{\pi^2 E J}{l^2}$$

$$y(x) = \frac{\pi^2 E J}{(l/n)^2} = \frac{\pi^2 E J}{l^2}$$

$$y(x) = A \sin \frac{n\pi x}{l}$$

$$P_{cr} = P_{cr,n}$$

$$P_{cr} = \frac{\pi^2 E J}{l^2} l_{0,n}$$

by setting the b's:  
 $f(0) = B = 0$   
 $y(0) = A \cdot 0 + B \cdot 0 = 0 \Rightarrow A \cdot 0 = 0$   
 $\text{and } y(l) = A \sin l + B \cos l = 0 \Rightarrow A \sin l = 0$

$$EJ \ddot{y}(x) + P y(x) = 0$$

$$\ddot{y}^2 + \frac{P}{EJ} y^2 = 0$$

homogeneous

$$y(x) = A \sin x + B \cos x$$

$$y'(x) = A \cos x - B \sin x$$

$$y(0) = A \sin 0 + B \cos 0 = B$$

$$y(l) = A \sin l + B \cos l = 0$$

solutions (Bartelsky constants)  
 $\infty$  bifurcated solutions

$$P_{cr,n} = \frac{\pi^2 E J}{l^2}$$

$$y(x) = A \sin \frac{n\pi x}{l}$$

$$A = \sqrt{\frac{E J}{l^2}} = \sqrt{\frac{\pi^2 E J}{l^2}}$$

$$P_{cr} = P_{cr,n}$$

$$P_{cr} = \frac{\pi^2 E J}{l^2} l_{0,n}$$

$$\text{free inflection length } l_{0,n}$$