

# Variational Formulation & the Galerkin Method

Dr. Konstantinos Agathos Institute of Structural Engineering ETH Zürich

17 April 2020



## **Today's Lecture Contents:**

• Introduction – reminder of previous lecture

• FE discretization

Discretized equilibrium equations

MATLAB demonstration



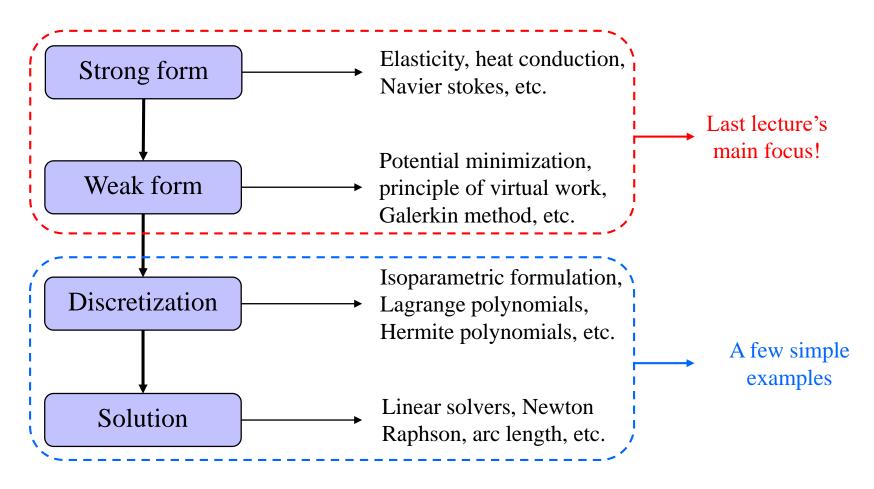
## Learning goals:

- Understanding how the weak form can be discretized using finite elements
- Understanding some of the advantages of FE discretization
- Learning some cool MATLAB tricks



### **FEM for PDEs**

A series of steps is typically followed by all FE methods:





## Strong form

General form of 2D second order partial differential equations (PDEs)

$$A(x,y)\frac{\partial^2 u}{\partial x^2} + 2B(x,y)\frac{\partial^2 u}{\partial x \partial y} + C(x,y)\frac{\partial^2 u}{\partial y^2} = \phi\left(x,y,u,\frac{\partial u}{\partial x},\frac{\partial u}{\partial y}\right)$$

Categorization:

$$B^{2} - AC = \begin{cases} <0 \rightarrow \text{elliptic} \\ =0 \rightarrow \text{parabolic} \\ >0 \rightarrow \text{hyperbolic} \end{cases}$$

Boundary conditions (BCs):

$$u(x_0, y_0) = u_0$$
 Dirichlet or essential boundary conditions

$$\frac{\partial^{m+n} u}{\partial x^m \partial y^n} = \overline{u}(x, u)$$
 Neumann of natural boundary conditions



### **FEM for PDEs**

The wide range of applicability is attributed to the fact that FEM is essentially a tool for solving partial differential equations (PDEs), for instance:

### Bernoulli-Euler beam equilibrium equations:

$$EI_z \frac{d^4 w}{dx^4} = f_y(x)$$

### 2D elasticity (Navier equations):

$$\frac{\partial \sigma_{xx}}{\partial x} + \frac{\partial \sigma_{xy}}{\partial y} + F_{x} = 0$$
$$\frac{\partial \sigma_{xy}}{\partial x} + \frac{\partial \sigma_{yy}}{\partial y} + F_{y} = 0$$

### **Laplace equation:**

$$\frac{\partial^2 \phi(x, y)}{\partial x^2} + \frac{\partial^2 \phi(x, y)}{\partial y^2} = 0$$



## Weak form

#### **Principle of stationary potential energy:**

$$\int_{0}^{L} EA \frac{du}{dx} \delta \frac{du}{dx} dx = \int_{0}^{L} \delta u F dx + \delta u(L) R$$

#### **Principle of Virtual Work:**

$$\int_{0}^{L} EA \frac{du}{dx} \frac{d\delta u}{dx} dx = \int_{0}^{L} \delta u F dx + \delta u(L) R$$

#### Galerkin method:

$$\int_{0}^{L} \frac{dw}{dx} EA \frac{du}{dx} dx = \int_{0}^{L} w F dx + w(L) R$$

©Carlos Felippa



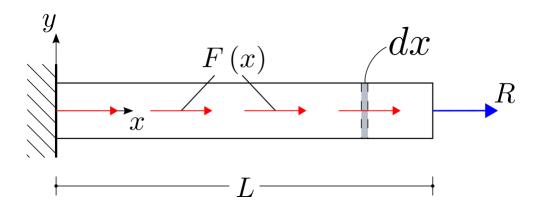
## **FEM** discretization

1D 2D 2D 3D



### Reminder – 1D bar

#### Illustrative example – 1D bar with a distributed and an end load

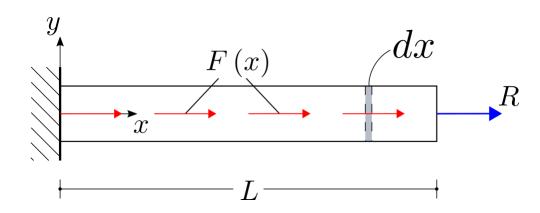


### **Assumptions:**

- Constant cross section
- Linear elastic material
- Loads applied at the centroid of the cross section
- Arbitrary distributed load



## Strong form



### **Equilibrium Equation**

$$EA\frac{d^2u}{dx^2} = -F(x)$$

### **Dirichlet (essential) boundary conditions**

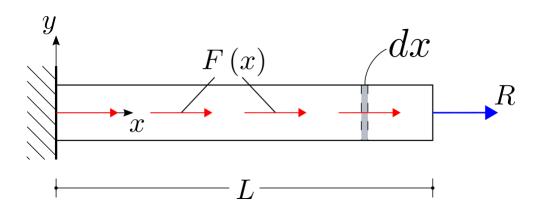
$$u(x=0)=0$$

### **Neumann (natural) boundary conditions**

$$A\sigma(x=L) = R \implies AE \frac{du}{dx}\Big|_{x=L} = R \implies \boxed{\frac{du}{dx}\Big|_{x=L}} = \frac{R}{EA}$$



## Strong form solution



Assuming no distributed load:

$$F(x) = 0$$

$$EA\frac{d^2u}{dx^2} = 0$$

$$u(x=0)=0$$

$$\frac{du}{dx}\Big|_{x=I} = \frac{R}{EA}$$

The solution should be of the form:  $u(x) = c_0 + c_1 x$ 

Equilibrium is satisfied:  $EA \frac{d^2(c_0 + c_1 x)}{dx^2} = 0$ 

From the essential B.C.:  $c_0 = 0$ 

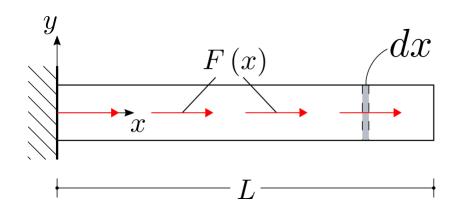
From the natural B.C.:  $c_1 = \frac{R}{EA}$ 

**Final form of the solution:** 

$$u(x) = \frac{R}{EA}x$$



## Strong form solution



Similarly, assuming a linear distributed load and no end load:

$$F(x) = ax$$
$$R = 0$$

$$EA\frac{d^2u}{dx^2} = -ax$$

$$u(x=0) = 0$$

$$\frac{du}{dx} = 0$$

The solution should be of the form:  $u(x) = c_0 + c_1 x + c_2 x^2 + c_3 x^3$ 

From the equilibrium equation:  $u(x) = c_0 + c_1 x - \frac{a}{6EA}x^3$ 

From the essential B.C.:  $c_0 = 0$ 

From the natural B.C.:  $\frac{du}{dx}\Big|_{C} = c_1 - \frac{aL^2}{2EA} = 0 \Rightarrow c_1 = \frac{aL^2}{2EA}$ 

Final form of the solution: 
$$u(x) = \frac{aL^2}{2EA}x - \frac{a}{6EA}x^3$$



## Weak form

#### **Principle of stationary potential energy:**

$$\int_{0}^{L} EA \frac{du}{dx} \delta \frac{du}{dx} dx = \int_{0}^{L} \delta u F dx + \delta u(L) R$$

#### **Principle of Virtual Work:**

$$\int_{0}^{L} EA \frac{du}{dx} \frac{d\delta u}{dx} dx = \int_{0}^{L} \delta u F dx + \delta u(L) R$$

#### Galerkin method:

$$\int_{0}^{L} \frac{dw}{dx} EA \frac{du}{dx} dx = \int_{0}^{L} w F dx + w(L) R$$



## Weak vs Strong form

#### Weak form:

$$\int_{0}^{L} \frac{dw}{dx} EA \frac{du}{dx} dx = \int_{0}^{L} w F dx + w(L)R$$

$$u(0) = 0$$

- ➤ Natural BCs are part of the weak form
- ➤ Highest derivative is of order 1
- $\triangleright$  Degree of continuity required:  $C_0$
- Equilibrium is satisfied in an integral, "weak" sense

### **Strong form:**

$$EA \frac{d^{2}u}{dx^{2}} = -F$$

$$\frac{du}{dx}\Big|_{x=L} = \frac{R}{EA}$$

$$u(x=0) = 0$$

- ➤ Natural BCs are explicitly imposed
- ➤ Highest derivative is of order 2
- $\triangleright$  Degree of continuity required:  $C_1$
- Equilibrium is satisfied everywhere in a "strong" sense



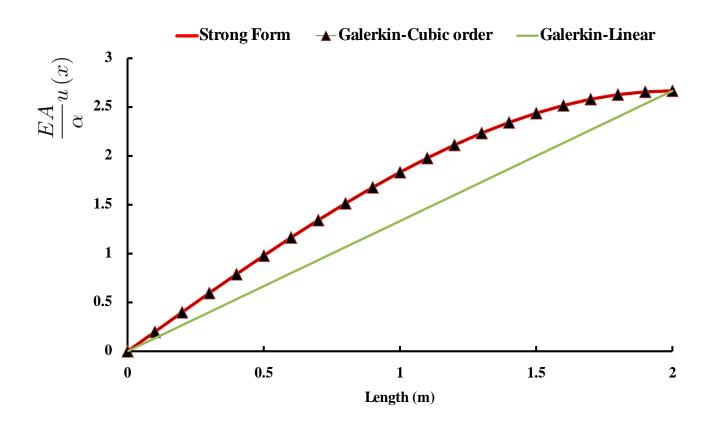
## Weak form solution - Approximate

Exact solution:

$$u(x) = \frac{aL^2}{2EA}x - \frac{a}{6EA}x^3$$

Approximate solution:

$$\overline{u}(x) = \frac{aL^2}{3EA}x$$

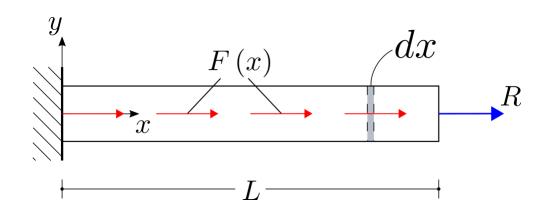




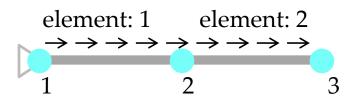
A piecewise polynomial solution is assumed

- The problem domain is divided in subdomains (elements) of simple geometry
- A polynomial solution is assumed for each subdomain
- Continuity between subdomains should be imposed





1D bar case:



• The domain is divided in subdomains

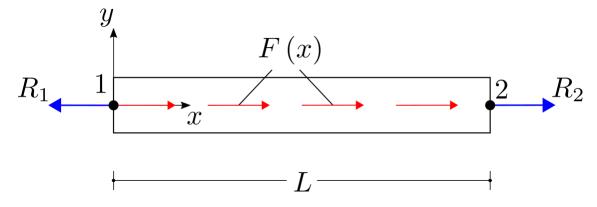
$$u(x) = a_0 + a_1 x$$

• A polynomial solution is assumed for each subdomain



For both the interpretation of results and imposition of continuity it is preferable to write displacements in terms of their nodal values. To do so:

➤ We consider a single element



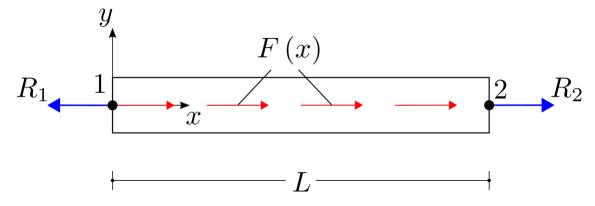
➤ We write displacements in matrix form:

$$u(x) = a_0 + a_1 x = \begin{bmatrix} 1 & x \end{bmatrix} \begin{bmatrix} a_0 \\ a_1 \end{bmatrix} = \mathbf{xa}$$



For both the interpretation of results and imposition of continuity it is preferable to write displacements in terms of their nodal values. To do so:

➤ We consider a single element



➤ We evaluate displacements at the nodes:

$$u(x=0) = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} a_0 \\ a_1 \end{bmatrix} = u_1$$

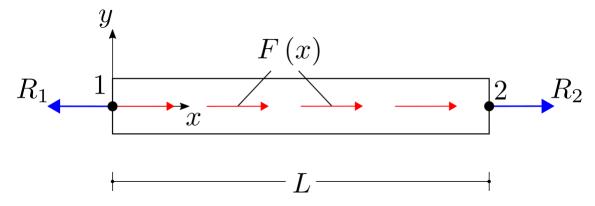
$$u(x=L) = \begin{bmatrix} 1 & L \end{bmatrix} \begin{bmatrix} a_0 \\ a_1 \end{bmatrix} = u_2$$

$$\begin{bmatrix} 1 & 0 \\ 1 & L \end{bmatrix} \begin{bmatrix} a_0 \\ a_1 \end{bmatrix} = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$



For both the interpretation of results and imposition of continuity it is preferable to write displacements in terms of their nodal values. To do so:

➤ We consider a single element



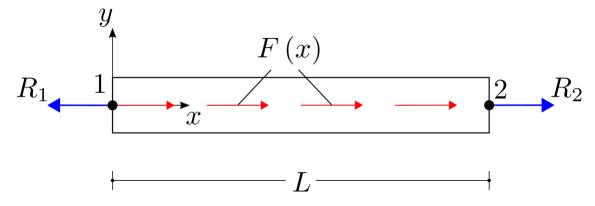
> We solve the resulting system for the polynomial coefficients:

$$\begin{bmatrix} 1 & 0 \\ 1 & L \end{bmatrix} \begin{bmatrix} a_0 \\ a_1 \end{bmatrix} = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} \Rightarrow \begin{bmatrix} a_0 \\ a_1 \end{bmatrix} = \begin{bmatrix} u_1 \\ u_2 - u_1 \\ L \end{bmatrix}$$



For both the interpretation of results and imposition of continuity it is preferable to write displacements in terms of their nodal values. To do so:

➤ We consider a single element



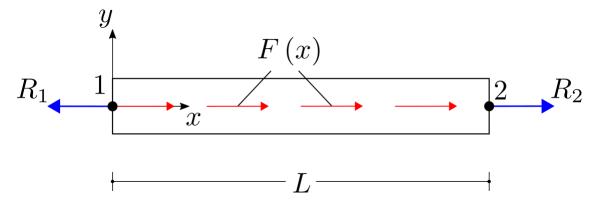
> We substitute in the initial expression

$$u(x) = \begin{bmatrix} 1 & x \end{bmatrix} \begin{bmatrix} a_0 \\ a_1 \end{bmatrix} = \begin{bmatrix} 1 & x \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 - u_1 \end{bmatrix} = u_1 + \frac{u_2 - u_1}{L} x$$



For both the interpretation of results and imposition of continuity it is preferable to write displacements in terms of their nodal values. To do so:

➤ We consider a single element



➤ We write the resulting expression in terms of the nodal displacements

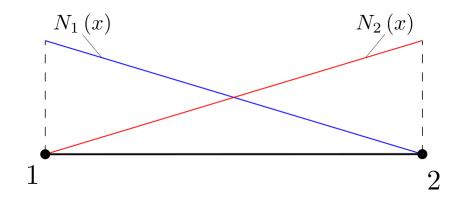
$$u(x) = u_1 + \frac{u_2 - u_1}{L}x = \begin{bmatrix} 1 - \frac{x}{L} & \frac{x}{L} \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$



The expressions resulting from the above procedure are the shape functions of a bar element:

$$u(x) = \begin{bmatrix} 1 - \frac{x}{L} & \frac{x}{L} \\ v_1 & v_2 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} N_1 & N_2 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \sum_{i=1}^2 N_i u_i$$

- ➤ They assume a value of 1 at the node they refer to
- They assume a value of 0 at all other nodes



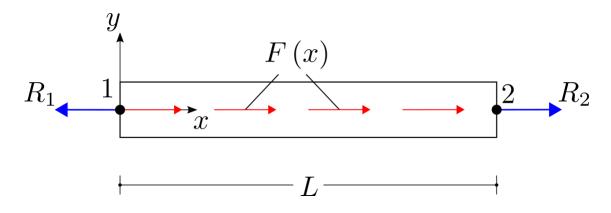


The weak form can be solved using the described FE discretization in the same way as done for a polynomial displacement assumption:

- ➤ Assume a general form for the solution
- ➤ Plug into weak form
- Obtain unknown coefficients

The unknown coefficients in this case will be the nodal displacements!





Galerkin weak form:

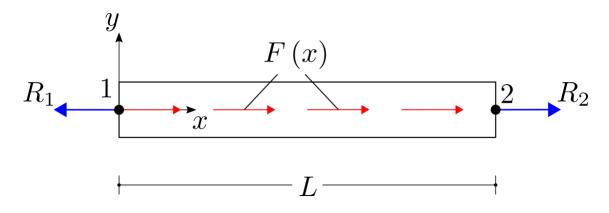
$$\int_{0}^{L} \frac{dw}{dx} EA \frac{du}{dx} dx = \int_{0}^{L} w F dx$$

FE discretization for displacements and weights:

$$u(x) = \begin{bmatrix} N_1 & N_2 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

$$w(x) = \begin{bmatrix} N_1 & N_2 \end{bmatrix} \begin{bmatrix} w_1 \\ w_2 \end{bmatrix}$$



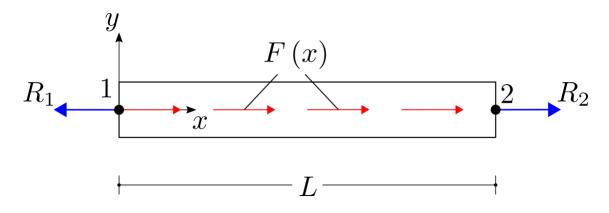


Galerkin weak form & FE shape functions:

$$\int_{0}^{L} \frac{dw}{dx} EA \frac{du}{dx} dx = \int_{0}^{L} w F dx \Leftrightarrow$$

$$\int_{0}^{L} \frac{d}{dx} \left[ \begin{bmatrix} w_{1} & w_{2} \end{bmatrix} \begin{bmatrix} N_{1} \\ N_{2} \end{bmatrix} \right] EA \frac{d}{dx} \left[ \begin{bmatrix} N_{1} & N_{2} \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \end{bmatrix} \right] dx = \int_{0}^{L} \begin{bmatrix} w_{1} & w_{2} \end{bmatrix} \begin{bmatrix} N_{1} \\ N_{2} \end{bmatrix} F dx$$



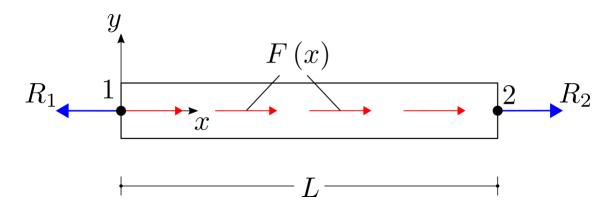


Galerkin weak form & FE shape functions:

$$\int_{0}^{L} \frac{d}{dx} \left[ \begin{bmatrix} w_{1} & w_{2} \end{bmatrix} \begin{bmatrix} N_{1} \\ N_{2} \end{bmatrix} \right] EA \frac{d}{dx} \left[ \begin{bmatrix} N_{1} & N_{2} \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \end{bmatrix} \right] dx = \int_{0}^{L} \left[ w_{1} & w_{2} \end{bmatrix} \begin{bmatrix} N_{1} \\ N_{2} \end{bmatrix} F dx \Leftrightarrow$$

$$\begin{bmatrix} w_{1} & w_{2} \end{bmatrix} \int_{0}^{L} \frac{d}{dx} \left[ \begin{bmatrix} 1 - \frac{x}{L} \\ \frac{x}{L} \end{bmatrix} \right] EA \frac{d}{dx} \left[ \begin{bmatrix} 1 - \frac{x}{L} & \frac{x}{L} \end{bmatrix} \right] dx \begin{bmatrix} u_{1} \\ u_{2} \end{bmatrix} = \begin{bmatrix} w_{1} & w_{2} \end{bmatrix} \int_{0}^{L} \begin{bmatrix} 1 - \frac{x}{L} \\ \frac{x}{L} \end{bmatrix} F dx \Leftrightarrow$$



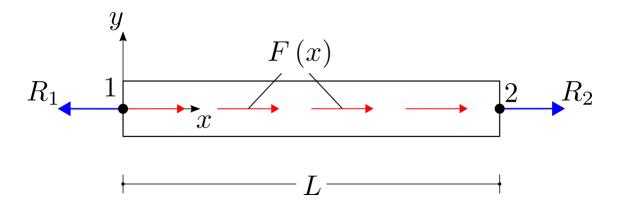


Galerkin weak form & FE shape functions:

$$\begin{bmatrix} w_1 & w_2 \end{bmatrix} \int_0^L \frac{d}{dx} \begin{bmatrix} 1 - \frac{x}{L} \\ \frac{x}{L} \end{bmatrix} EA \frac{d}{dx} \begin{bmatrix} 1 - \frac{x}{L} & \frac{x}{L} \end{bmatrix} dx \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} w_1 & w_2 \end{bmatrix} \int_0^L \begin{bmatrix} 1 - \frac{x}{L} \\ \frac{x}{L} \end{bmatrix} Fdx \Leftrightarrow$$

$$\begin{bmatrix} w_1 & w_2 \end{bmatrix} \int_0^L \begin{bmatrix} -\frac{1}{L} \\ \frac{1}{L} \end{bmatrix} EA \begin{bmatrix} -\frac{1}{L} & \frac{1}{L} \end{bmatrix} dx \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \begin{bmatrix} w_1 & w_2 \end{bmatrix} \int_0^L \begin{bmatrix} 1 - \frac{x}{L} \\ \frac{x}{L} \end{bmatrix} Fdx$$





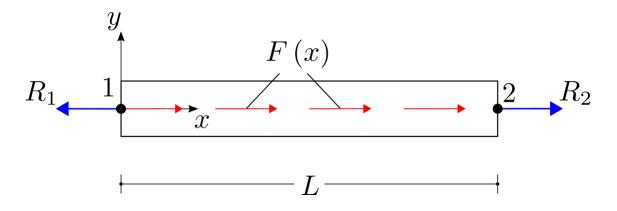
After carrying out the integrations (assuming a constant distributed load):

$$\underbrace{\frac{EA}{L} \begin{bmatrix} 1 & -1 \\ 1 & 1 \end{bmatrix}}_{\mathbf{K}} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \underbrace{\frac{FL}{2} \begin{bmatrix} 1 \\ 1 \end{bmatrix}}_{\mathbf{f}}$$

where F is the constant value of the load

The bar stiffness matrix and load vector are obtained!





After carrying out the integrations (assuming a linear distributed load):

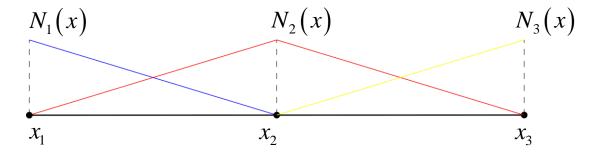
$$\underbrace{\frac{EA}{L}\begin{bmatrix} 1 & -1 \\ 1 & 1 \end{bmatrix}}_{\mathbf{K}} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} = \underbrace{\begin{bmatrix} \frac{F_1L}{3} + \frac{F_2L}{6} \\ \frac{F_2L}{3} + \frac{F_1L}{6} \end{bmatrix}}_{\mathbf{f}}$$

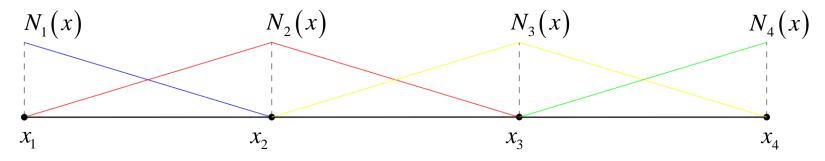
where  $F_1, F_2$  are the values of the distributed load at the nodes

The bar stiffness matrix and load vector are obtained!



Consider an assembly of multiple elements:





$$u(x) = \sum_{i=1}^{n} N_i u_i$$

$$w(x) = \sum_{i=1}^{n} N_i w_i$$

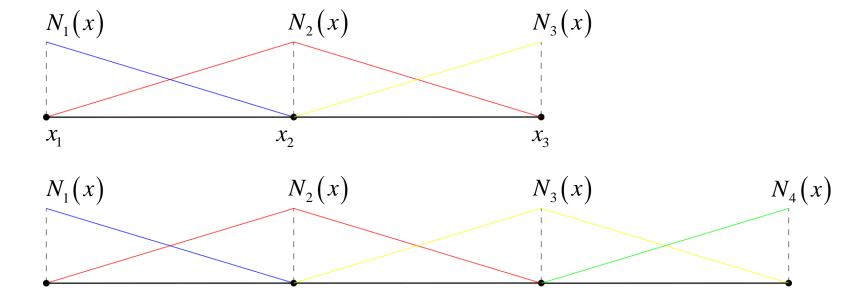
 $\mathcal{X}_{4}$ 



## FE formulation

Consider an assembly of multiple elements:

 $x_2$ 



 $\mathcal{X}_3$ 

We observe that:

$$N_i = 0 \text{ for } x < x_{i-1} \text{ or } x > x_{i+1}$$

 $\mathcal{X}_1$ 



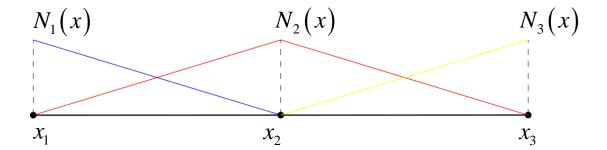
Using this observation we can write displacements and weights elementwise:

$$u^{e}(x) = [N_{e} \quad N_{e+1}] \begin{bmatrix} u_{e} \\ u_{e+1} \end{bmatrix} = [0 \quad \dots \quad N_{e} \quad N_{e+1} \quad \dots \quad 0] \begin{bmatrix} u_{e} \\ u_{e+1} \\ \vdots \\ 0 \end{bmatrix}$$

$$w^{e}(x) = [N_{e} \quad N_{e+1}] \begin{bmatrix} w_{e} \\ w_{e+1} \end{bmatrix} = [0 \quad \dots \quad N_{e} \quad N_{e+1} \quad \dots \quad 0] \begin{bmatrix} w_{e} \\ w_{e+1} \\ \vdots \\ 0 \end{bmatrix}$$



For the 2 element case:



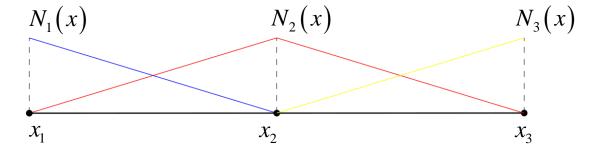
Displacements assume the form:

$$u^{1}(x) = \begin{bmatrix} N_{1} & N_{2} \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \end{bmatrix} = \begin{bmatrix} N_{1} & N_{2} & 0 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix}$$

$$u^{2}(x) = \begin{bmatrix} N_{2} & N_{3} \end{bmatrix} \begin{bmatrix} u_{2} \\ u_{3} \end{bmatrix} = \begin{bmatrix} 0 & N_{2} & N_{3} \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix}$$



For the 2 element case:

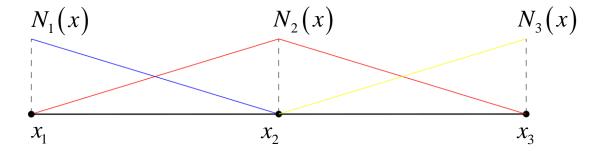


Substituting in the weak form for the two element case:

$$\int_{x_{1}}^{x_{3}} \frac{dw}{dx} EA \frac{du}{dx} dx = \int_{x_{1}}^{x_{3}} w F dx \Leftrightarrow \int_{x_{1}}^{x_{2}} \frac{dw^{1}}{dx} EA \frac{du^{1}}{dx} dx + \int_{x_{2}}^{x_{3}} \frac{dw^{2}}{dx} EA \frac{du^{2}}{dx} dx = \int_{x_{1}}^{x_{2}} w^{1} F dx + \int_{x_{2}}^{x_{3}} w^{2} F dx$$



For the 2 element case:



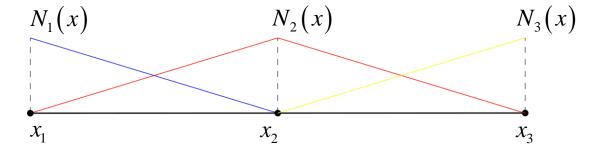
Carrying out the derivations/integrations:

$$\int_{x_{1}}^{x_{2}} \frac{dw^{1}}{dx} EA \frac{du^{1}}{dx} dx + \int_{x_{2}}^{x_{3}} \frac{dw^{2}}{dx} EA \frac{du^{2}}{dx} dx =$$

$$= \begin{bmatrix} w_{1} & w_{2} & w_{3} \end{bmatrix} \frac{EA}{L_{1}} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix} + \begin{bmatrix} w_{1} & w_{2} & w_{3} \end{bmatrix} \frac{EA}{L_{2}} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix}$$



For the 2 element case:



Carrying out the derivations/integrations:

$$\int_{x_{1}}^{x_{2}} w^{1} F dx + \int_{x_{2}}^{x_{3}} w^{2} F dx = \begin{bmatrix} w_{1} & w_{2} & w_{3} \end{bmatrix} \frac{FL_{1}}{2} \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix} + \begin{bmatrix} w_{1} & w_{2} & w_{3} \end{bmatrix} \frac{FL_{2}}{2} \begin{bmatrix} 0 \\ 1 \\ 1 \end{bmatrix}$$



Equating the two sides:

$$\underbrace{\begin{bmatrix} w_1 & w_2 & w_3 \end{bmatrix}}_{} \underbrace{\frac{EA}{L_1}}_{} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix}}_{} + \underbrace{\begin{bmatrix} w_1 & w_2 & w_3 \end{bmatrix}}_{} \underbrace{\frac{EA}{L_2}}_{} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix}}_{} = \underbrace{\begin{bmatrix} w_1 & w_2 & w_3 \end{bmatrix}}_{} \underbrace{\frac{FL_1}{2}}_{} \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix}}_{} + \underbrace{\begin{bmatrix} w_1 & w_2 & w_3 \end{bmatrix}}_{} \underbrace{\frac{FL_2}{2}}_{} \begin{bmatrix} 1 \\ 1 \\ 1 \end{bmatrix}}_{}$$

$$\frac{EA}{L_{1}} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix} + \frac{EA}{L_{2}} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix} = \frac{FL_{1}}{2} \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix} + \frac{FL_{2}}{2} \begin{bmatrix} 0 \\ 1 \\ 1 \end{bmatrix}$$



Writing the equation in more general terms:

$$\frac{EA}{L_{1}} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix} + \frac{EA}{L_{2}} \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix} = \frac{FL_{1}}{2} \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix} + \frac{FL_{2}}{2} \begin{bmatrix} 0 \\ 1 \\ 1 \end{bmatrix}$$

$$\begin{bmatrix} k_{11}^1 & k_{12}^1 & 0 \\ k_{21}^1 & k_{22}^1 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} + \begin{bmatrix} 0 & 0 & 0 \\ 0 & k_{11}^2 & k_{12}^2 \\ 0 & k_{21}^2 & k_{22}^2 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} = \begin{bmatrix} f_1^1 \\ f_2^1 \\ 0 \end{bmatrix} + \begin{bmatrix} 0 \\ f_1^2 \\ f_2^2 \end{bmatrix}$$

$$\begin{bmatrix} k_{11}^{1} & k_{12}^{1} & 0 \\ k_{21}^{1} & k_{21}^{1} + k_{11}^{2} & k_{12}^{2} \\ 0 & k_{21}^{2} & k_{22}^{2} \end{bmatrix} \begin{bmatrix} u_{1} \\ u_{2} \\ u_{3} \end{bmatrix} = \begin{bmatrix} f_{1}^{1} \\ f_{2}^{1} + f_{1}^{2} \\ f_{2}^{2} \end{bmatrix}$$

The assembly process is obtained!